MAY 1982 EDITION
GSA FPMR (41 CFR) 101-11.8

UNITED STATES GOVERNMENT

Memorandum

TO : Distribution

DATE: July 8, 1969

In reply refer to:

FROM : CF/Chief, Flight Crew Support Division

CF342-9M-133

SUBJECT: Revision A to the Apollo 11 Final Flight Plan

Enclosed is Revision A to the Apollo 11 Final Flight Plan. Enclosure 1 contains the pen and ink changes to be made on the indicated pages. Enclosure 2 contains the new pages which replace the same numbered pages in the final edition of the flight plan. Changes on the new pages are made with prestige elite type rather than gothic type so they will be readily evident.

Warren J. North

Enclosures 2

CF342:TAGuillory:jat 7-8-69



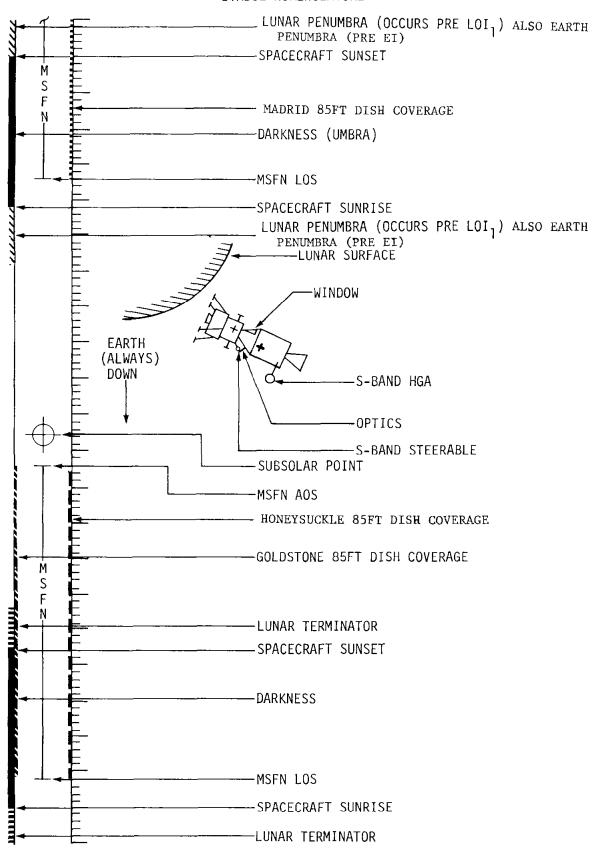
ENCLOSURE 1

MAKE THE FOLLOWING PEN AND INK CHANGES TO THE FINAL FLIGHT PLAN

PAGE	<u>CHANGE</u>
1-7	ON "LM POWERED DESCENT/HIGH GATE TO TD" FIGURE CHANGE ORDINATE AND ABSCISSA FROM "FT" TO "FT X 10^3"
1-10	ON TABLE 1-3 FIRST COLUMN CHANGE "1409 EDT" TO "1209 EDT"
	ON TABLE 1-4 CHANGE A1 LONGITUDE FROM 60.500°E TO 65.500°E
1-11	CHANGE G1 DELTA ALTITUDE FROM -001.97 TO -001.76
1-17	PARAGRAPH Ei, CHANGE TIMES: 103:30:00 TO 107:15 148:00:00 TO 151:00 154:00:00 TO 172:00
2-31 & 2-32	CHANGE TITLES FROM "PHASING" TO "ASCENT PHASING"
2-44	CHANGE 275 (+0418.1) TO 275 (+0418.5)
	CHANGE AVX, AVY, AND AVZ BLOCKS FROM
2-48	CHANGE AVR N42 BLOCKS FROM + 0 0 TO + 0
	CHANGE F/A, R/L, AND D/U BLOCKS FROM 0 0
3-ii	AT 05:25 CHANGE "S-IVB TO ORBIT CAPABILITY" TO "COI CAPABILITY"
3-1	AT 00:37 CHANGE "ECS REDUNDANT COMPONENT CHECK" TO "SEC GLYCOL LOOP CK"
	AT 00:43 IN BOTH COLUMNS CHANGE "RECORD ΔAZ CORRECTION" TO "RECORD Z TORQUE"
3-5	AT 04:07 DELETE "THRUST MONITOR - P47"
	AT 04:52 DELETE "MANEUVER TO OBSERVE SLINGSHOT"
3-6	AT 05:10 IN MCC-H COLUMN "ADD UPLINK CMC ZERO TRUNNION BIAS"
3-10	AT 11:50 IN NOTES COLUMN CHANGE THE MCC NOTE TO READ: "MCC1 WILL BE DELAYED TO MCC2 IF PROPELLANT COST IS NOT PROHIBITIVE"
3-19	AT 26:50 IN NOTES COLUMN CHANGE THE MCC NOTE TO READ: "TO AVOID EXECUTION OF MCC3, A NON FREE MCC2 WILL BE EXECUTED (SPS BURN IF THE BT >3 SEC)"

PAGE	CHANGE
3-33	AT 53:50 IN NOTES COLUMN CHANGE THE MCC NOTE TO READ: "MCC3 WILL BE DELAYED TO MCC4 IF PROPELLANT COST IS NOT PROHIBITIVE"
3-43	AT 70:50 IN NOTES COLUMN CHANGE THE MCC NOTE TO READ: "MCC4 WILL BE EXECUTED WITH THE SPS IF THE BT >3 SEC"
3-44	AT 71:25 DELETE "PRE-LOI ECS REDUNDANT COMPONENT CK ACTIVATE PRIMARY EVAPORATOR" AND ADD "SECONDARY RADIATOR LEAK CHECK"
3-51	AT 80:12 CHANGE "DO NOT TRIM" TO "TRIM X AXIS TO 1 FPS"
3-53	AT 82:13 CHANGE R 0, P 257/297, Y 0 TO R 0, P 296/303, Y 0
3-54	AT 83:20 CHANGE R 82, 350/229, Y 0 TO R 82, P 296/229, Y 0
3-61	AT 95:00 ADD "ROLL RT 40°, TO R 122, P 229, Y 0"
3-63	AT 96:48 IN MCC-H COLUMN DELETE "BLOCK DATA"
3-73 & 3-74	DELETE LOCAL HORIZONTAL PITCH OF 166.3 & 244.2 AT 106:47 AND 107:14 RESPECTIVELY
3-76	AT 109:45 IN LM COLUMNS ADD "RADIATION READOUT"
3-79	AT 112:30 IN CMP COLUMN ADD "ACTIVATE PRIM EVAPORATOR"
3-83	AT 116:40 IN LM COLUMN ADD "CONFIGURE COMM (TV-OFF)"
3-91	AT 126:00 IN MCC-H COLUMN ADD "DUMP DSE"
3-92	AT 126:07 IN LMP COLUMN CHANGE P -58 TO P +58
	AT 126:30 CHANGE CSM ATTITUDE FROM R 180, P 264/177, Y 0 TO R 0, P 239/160, Y 0
3-93	AT 128:00 ADD IN CMP COLUMN "EXT RNDZ LT - OFF" AND IN MCC-H COLUMN "DUMP DSE"
	AT 127:00 IN CMP & MCC-H COLUMNS ADD "GO/NO GO FOR PYRO ARM"
3-95	AT 130:00 IN MCC-H COLUMN ADD "DUMP DSE"
3-100	AT 134:12 DELETE "ECS REDUNDANT COMPONENTS CK"
3-101	AT 135:18 CHANGE "DAP - R1=10111 R2=11111" TO "DAP - R1=10101 R2=11111" AND MOVE TO 135:05
3-102	AT 136:15 ADD "DEACTIVATE PRIM EVAPORATOR"
3-132	AT 192:27 CHANGE "MENKAR (07) ENH" TO "MENKAR (07) EFH"

ENCLOSURE 2 SYMBOL NOMENCLATURE



INTRODUCTION

This Flight Plan has been prepared by the Flight Planning Branch, Flight Crew Support Division, with technical support by TRW Systems.

This document schedules the AS-506/CSM-107/LM-5 operations and crew activities to fulfill, when possible, the test objectives defined in the Mission Requirements, G Type Mission Lunar Landing.

The trajectory parameters used in this Flight Plan are for July 16, 1969 launch, with a 72° launch azimuth and were supplied by Mission Planning and Analysis Division as defined by the Apollo Mission G Spacecraft Operational Trajectory.

The Apollo 11 Flight Plan is under the configuration control of the Crew Procedures Control Board (CPCB). All proposed changes to this document that fall in the following categories should be submitted to the CPCB via a Crew Procedures Change Request:

- 1. Items that impose additional crew training or impact crew procedures.
- 2. Items that impact the accomplishment of detailed test objectives.
- 3. Items that result in a significant RCS or EPS budget change.
- 4. Items that result in moving major activities to a different activity day in the Flight Plan.
- 5. Items that require a change to the flight data file.

The Chief, Flight Planning Branch (FCSD) will determine what proposed changes fall in the above categories.

Mr. T. A. Guillory will act as co-ordinator for all proposed changes to the Apollo 11 Flight Plan.

Any requests for additional copies or changes to the distribution lists of this document must be made in writing to Mr. W. J. North, Chief, Flight Crew Support Division, MSC, Houston, Texas.

b. The coolant loop operation will be as follows:

(1) Launch - primary loop operation

- (2) Earth Orbit primary loop operation and secondary loop test
- (3) Post TLI deactivate both evaporators

(4) Pre LOI sec rad leak ck only.

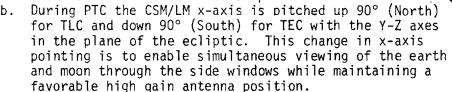
(5) At 112:30 activate primary evaporator

(6) Post TEI - deactivate primary evaporator

(7) Entry interface minus 1 hour - activate primary and secondary evaporator.

3. Guidance and Navigation

a. During lunar orbit, the CSM and LM will utilize the same landing site REFSMMAT such that the gimbal angles would be 0,0,0 at landing with the LM sitting face forward on landing site number two and the CSM over the landing site pitched up 90° from local horizontal "heads up".



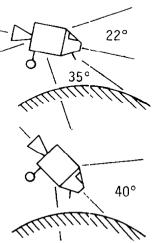
c. The <u>CSM tracking light</u> will be on continuously from undocking to landing and from LM lift-off to docking.

4. Landmark Tracking

The following ground rules were used for landmark tracking.

- a. IMU to be realigned on the dark side preceding each tracking period.
- b. $\underline{\text{MSFN}}$ is reacquired after each tracking period. The tracking data will be acquired by MSFN after all the marks have been made and while N49 ($\Delta R, \Delta V$) is displayed. MSFN will give a GO when data acquisition has been verified.
- c. The pseudo landmark tracking (Al) will be used to determine the altitude of an area in which the LM will be making altitude checks after DOI. The data will be processed during the sleep period after the trackings and relayed to the LM prior to undocking.

- d. In the docked configuration the CSM/LM approaches the landmark in an inertial hold attitude. This inertial attitude places the spacecraft 2° below the local horizontal at the 35° elevation angle point. At 35° elevation angle a pitch down of 0.3°/sec is initiated. Five marks are then taken with the time between marks a minimum of 25 seconds. (See tracking profile)
- e. In the undocked configuration the CSM approaches the landmark in ORB RATE and pitched down 22° from the local horizontal. At 35° elevation angle five marks are taken with the time between marks a minimum of 25 seconds. ORB RATE is continued throughout the marking period.
- f. In the undocked COAS tracking the CSM will approach the LM in ORB RATE heads up and pitched down 40° from the local horizontal. When the LM is centered in the COAS the CSM will initiate a variable pitch rate to keep the LM centered in the COAS.



35°

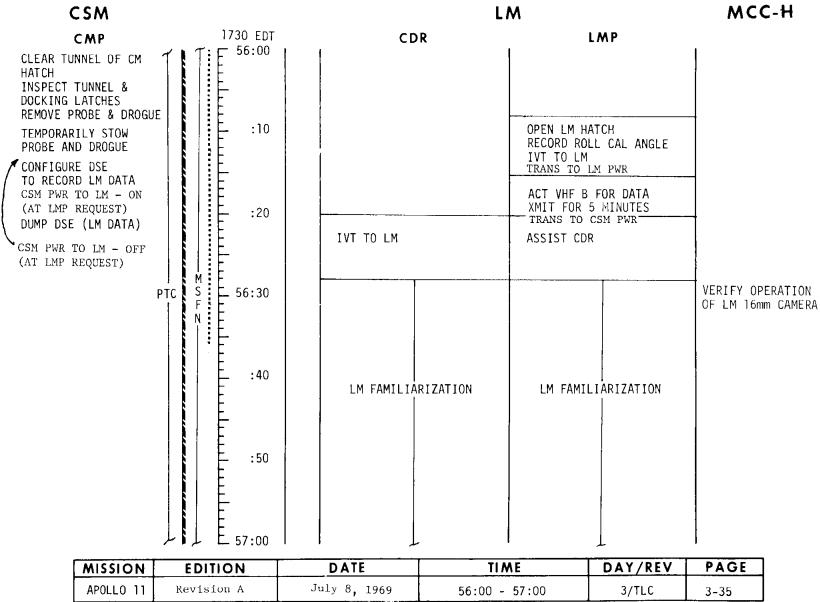
- 5. CSM/LM and CSM attitude maneuvers will normally be at a rate of 0.2°/sec or 0.5°/sec. unless other rates are required.

 NOTE: At 0.2°/sec, 15 minutes is required to maneuver 180°.

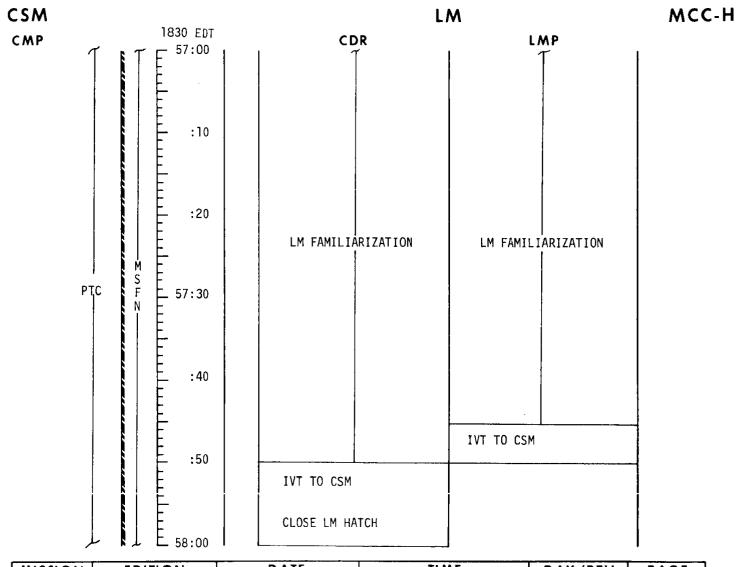
 At 0.5°/sec, 6 minutes is required to maneuver 180°.
- 6. Passive thermal control mode will be initiated after MCCl or as soon as MCCl is scrubbed and maintained throughout the mission (except in lunar orbit) until at least three hours before entry except for interruptions for midcourse corrections, communications orientation (maximum interruption of three hours). PTC will not be initiated before approximately 7:00 GFT.
- 7. Service Propulsion System All SPS burns will be initiated on Bank A except LOI1 which will be initiated on Bank B.

F. LM Notes

- 1. Entries into the LM
 - a. Three entries into the LM are scheduled in the timeline at 56:30, 81:30 and 95:52 GET respectively.



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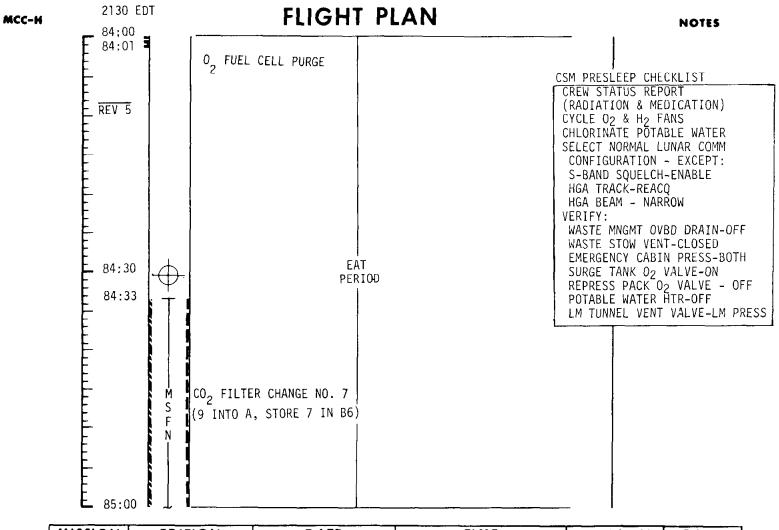


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APOLLO 11	FINAL	JULY 1, 1969	57:00 - 58:00	3/TLC	3-36

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LOI₁ BURN CHART

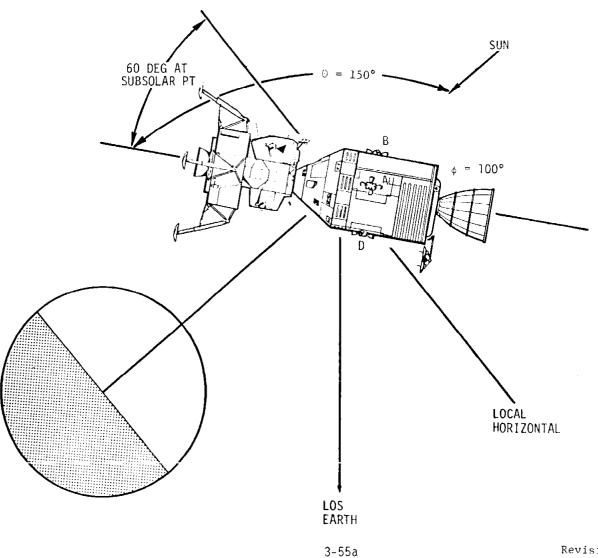
	P OR Y RATES		ATT DEVIATION		RESIDUALS		
LOI	10°/SEC TAKEOVER			BT + 10 SEC	DO NOT TRIM		
	LOI ₁ ABORT MODES						
LOI ₁ V _{GO} BT TRAJECTORY ABORT MODE				ODE			
2924.0 - 2129.0 2129.0 - 1589.0	01:50 -	HYPERBOLIC UNSTABLE	MODE MODE	(P37 BE VGO >22 II - COAST 2	HR - DPS - P37 YOND SPHERE FOR 79 AND BT <01:30 HR - 2 DPS		
1589.0 - 0	03:00 03:00 - 06:05	LUNAR ORBIT	MODE		OR STABILIZA- D WATER or CLA N AFTER		



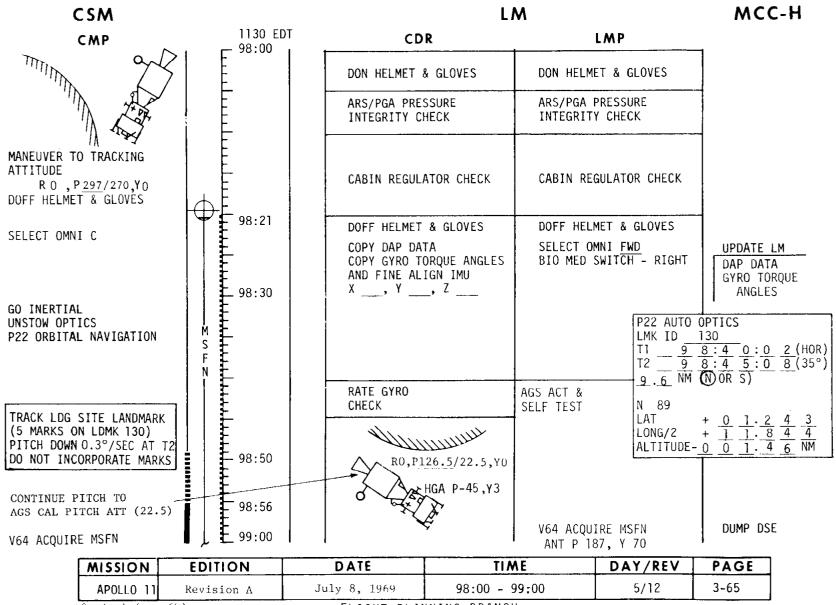
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MSC Form 29 (May 89)

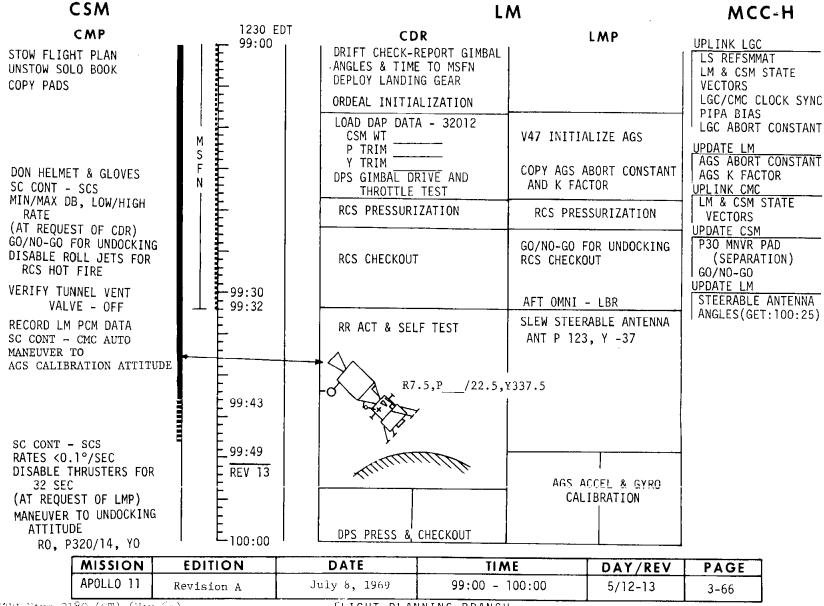
LUNAR ORBIT REST PERIOD ATTITUDE



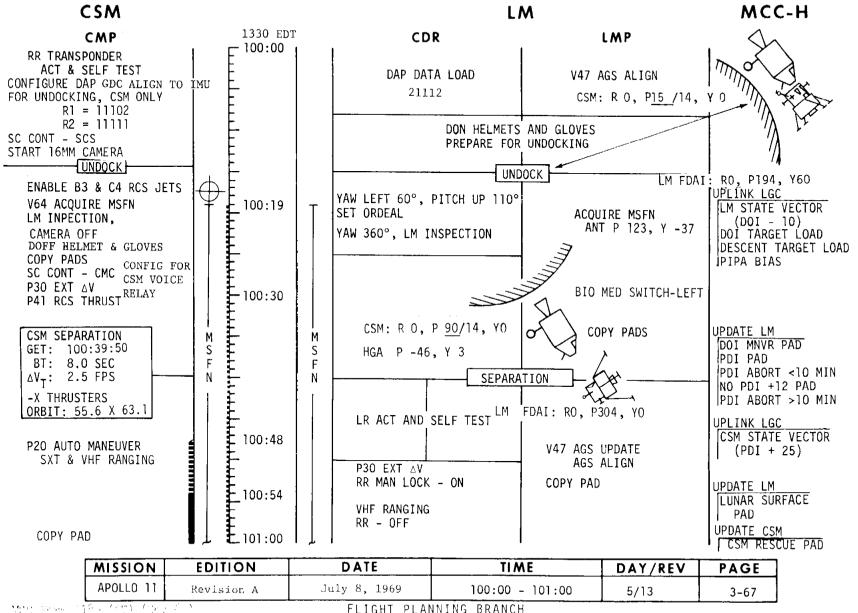
Revision A

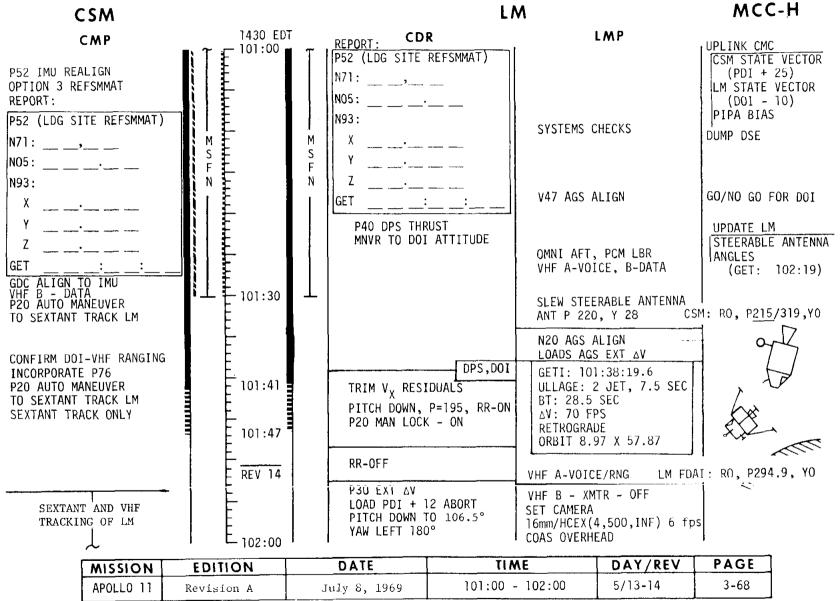


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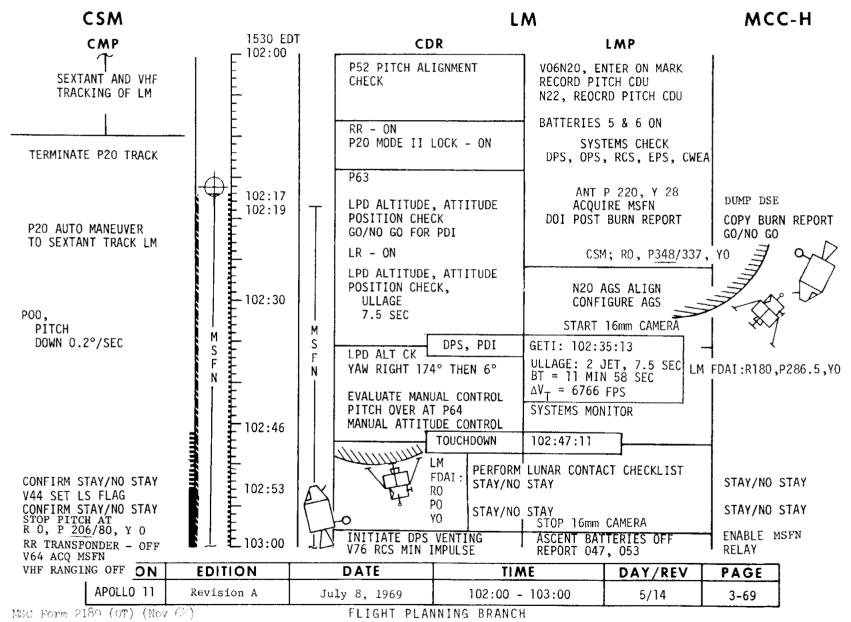


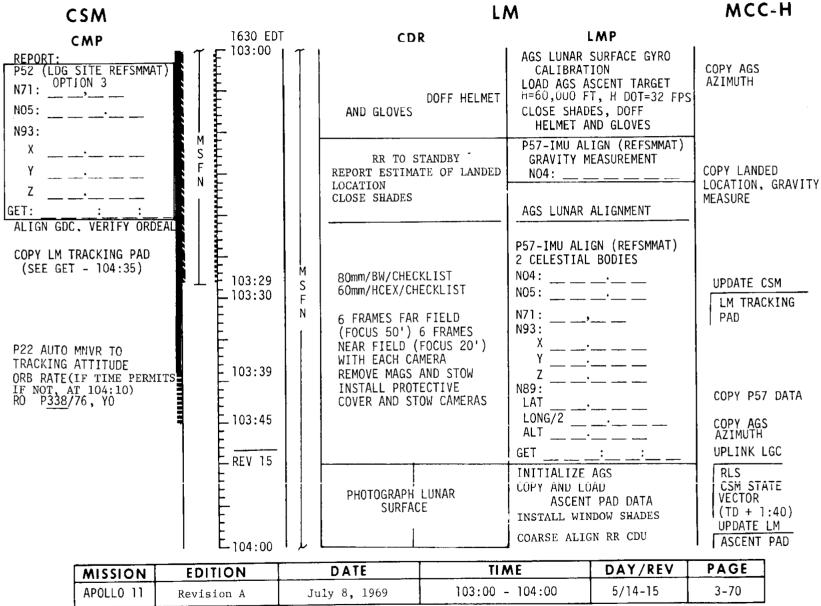
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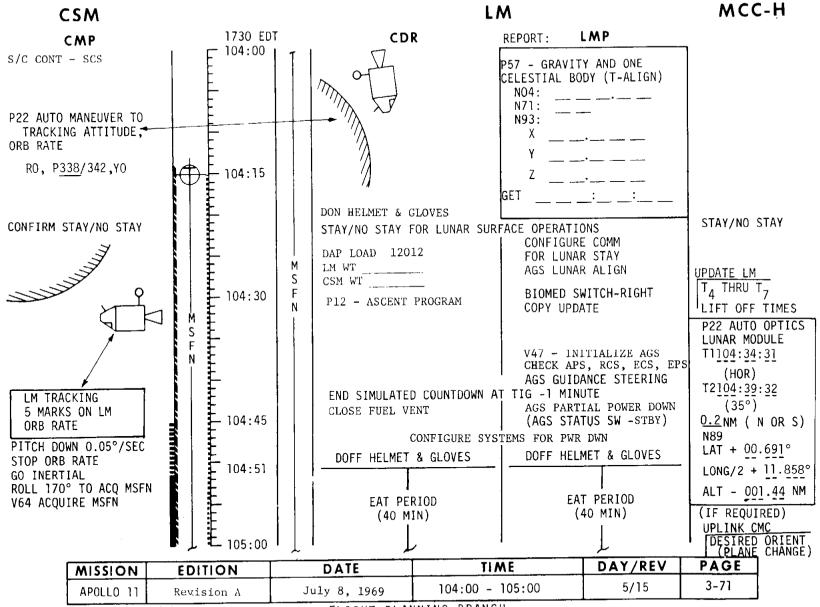




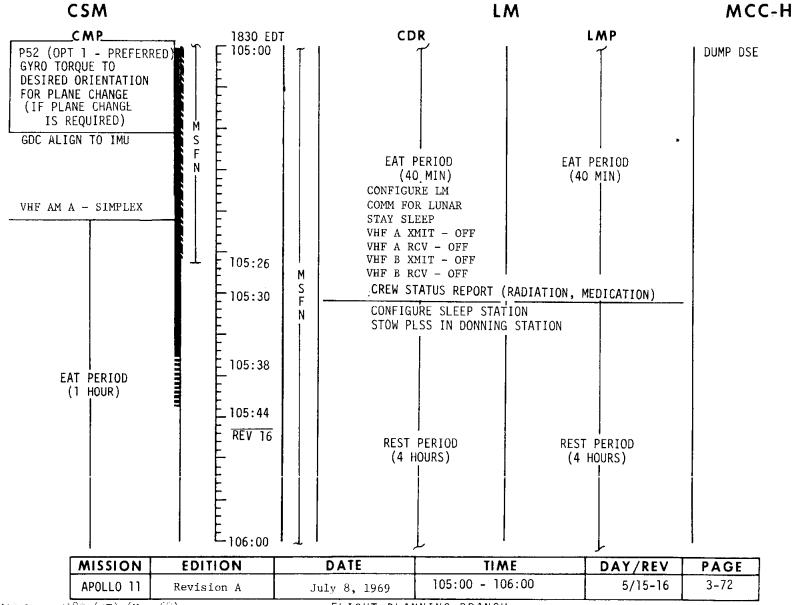
Mad Form (180 (OT) (Nov (C))



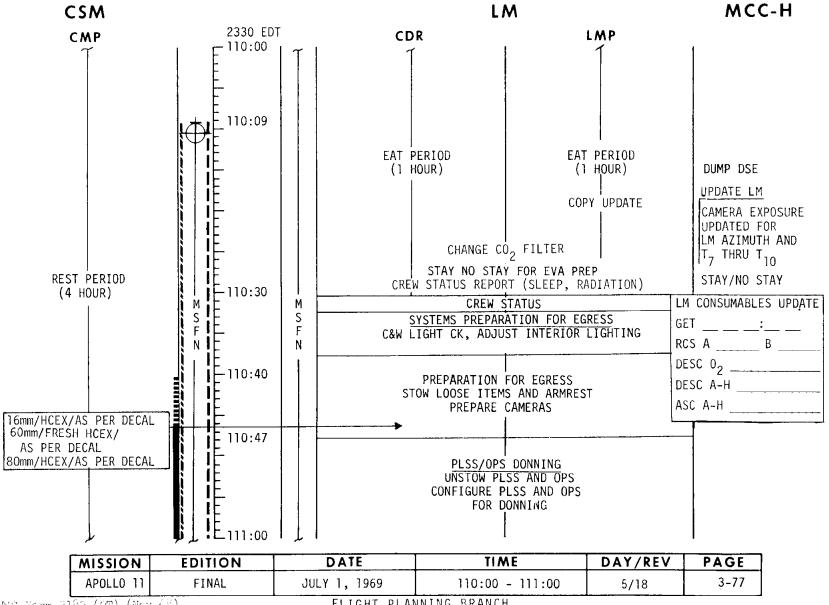




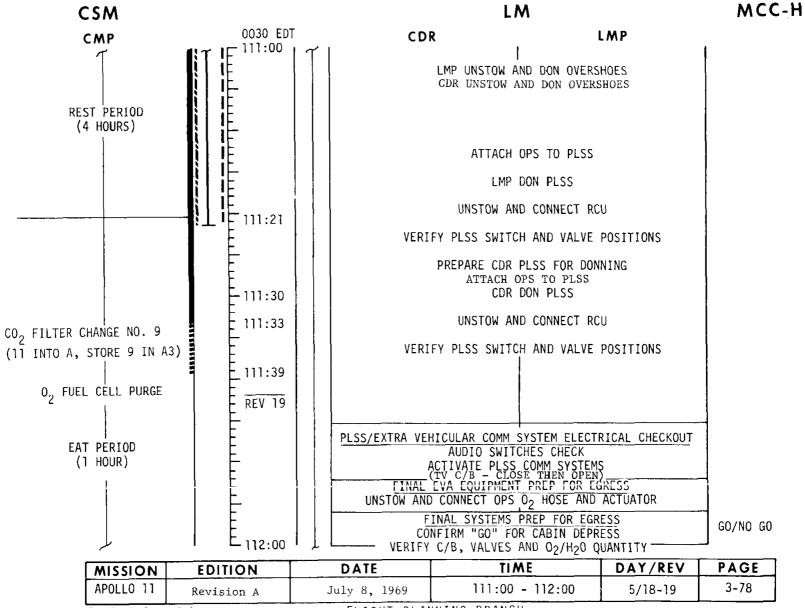
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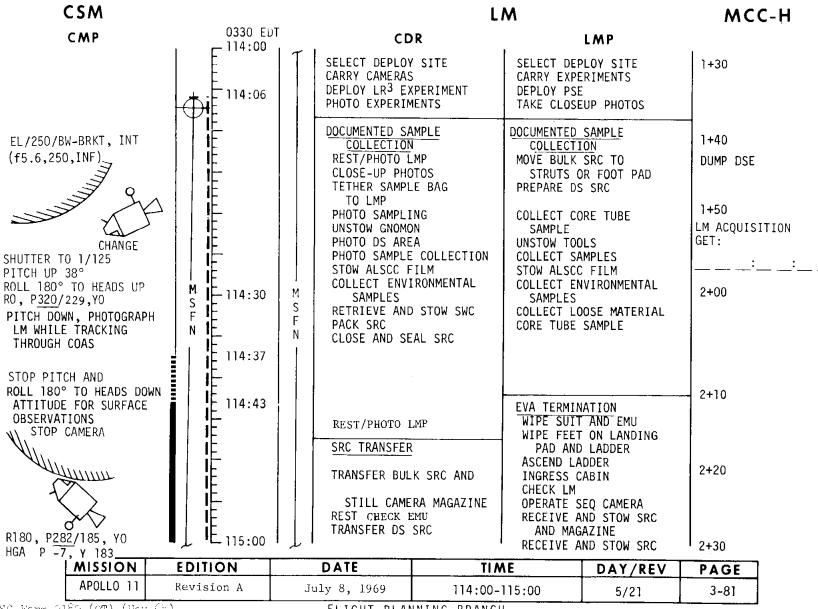
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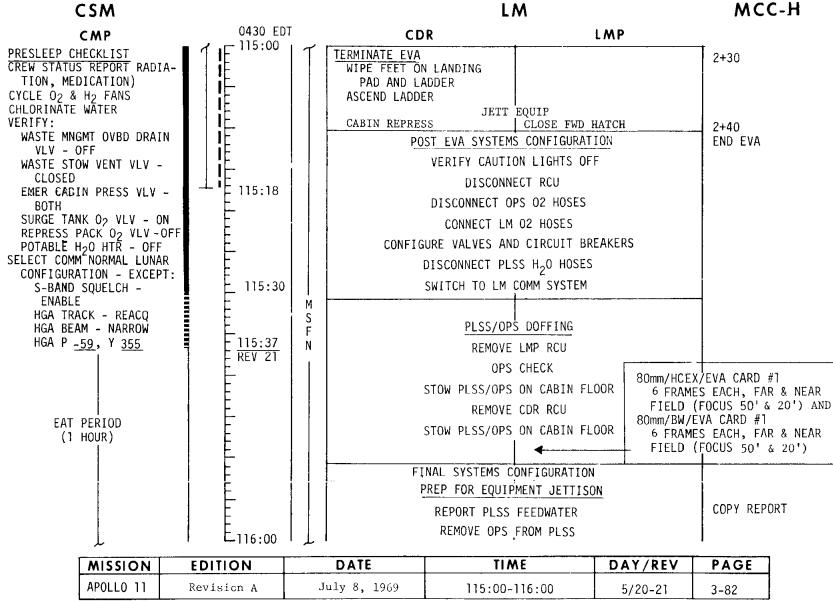
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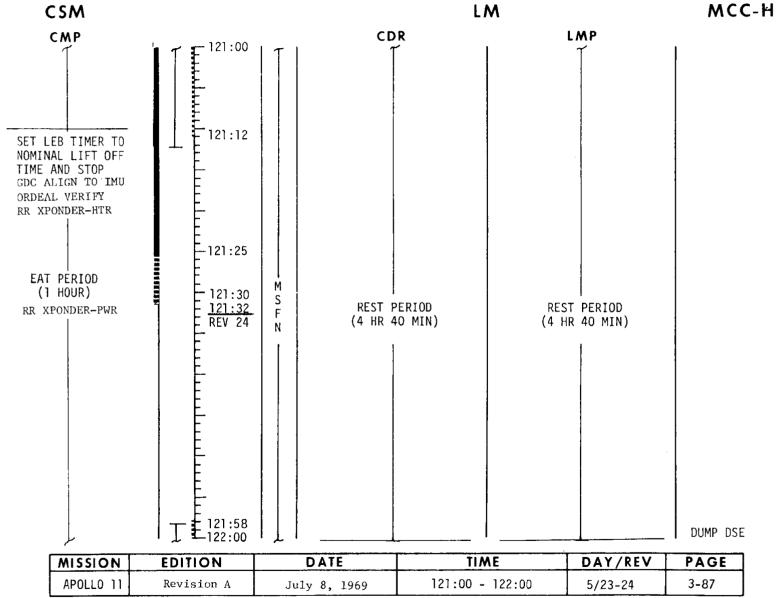
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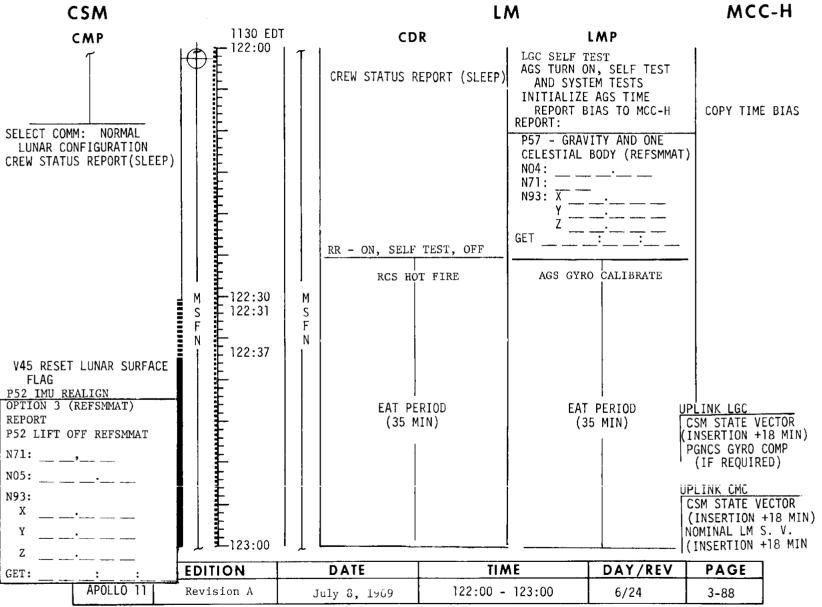


MSC Form 180 (OT) (Nov (1)



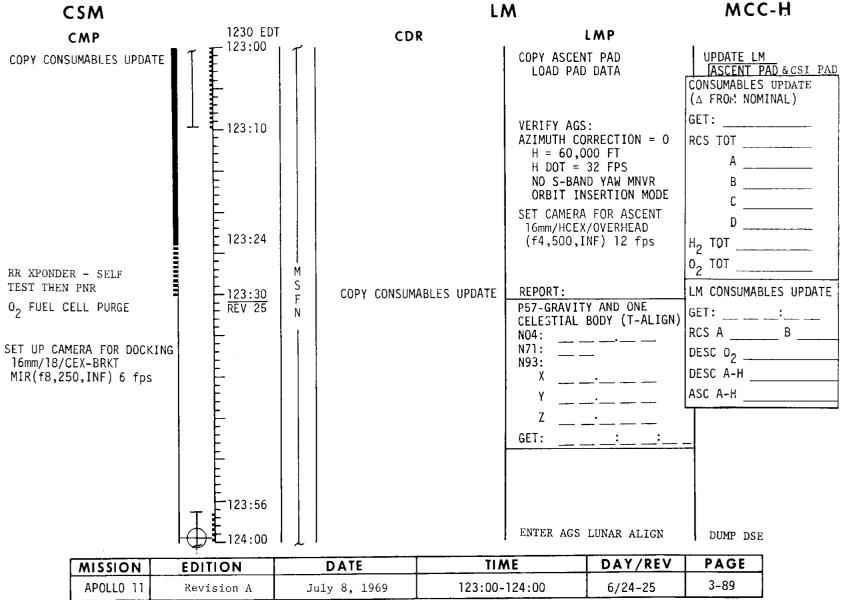
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FLIGHT PLANNING BRANCH

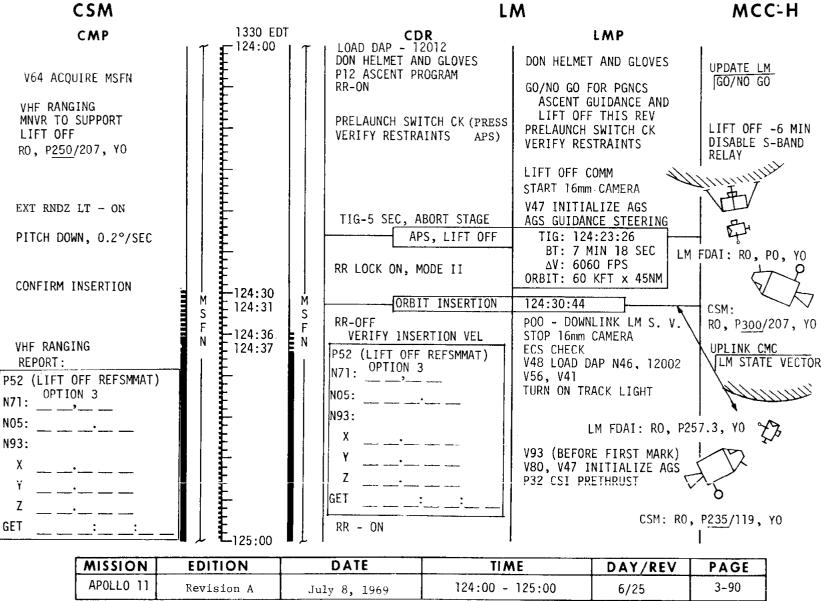


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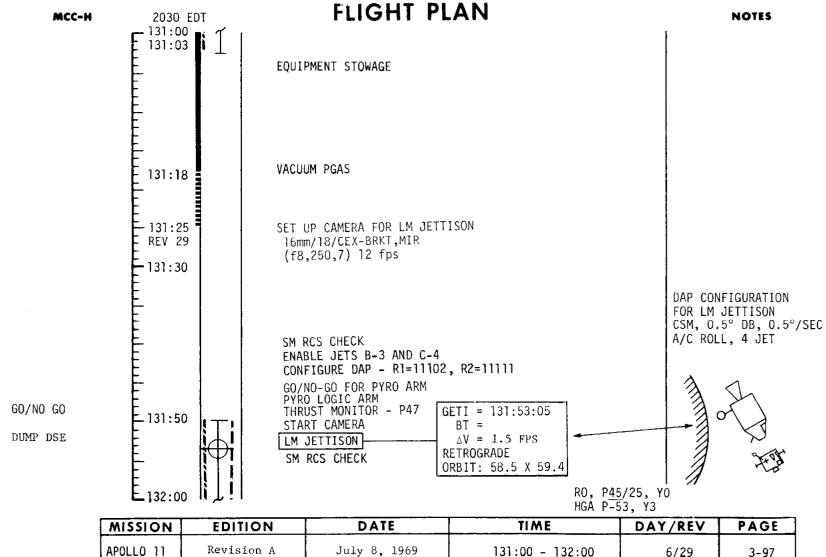
FLIGHT PLANNING BRANCH



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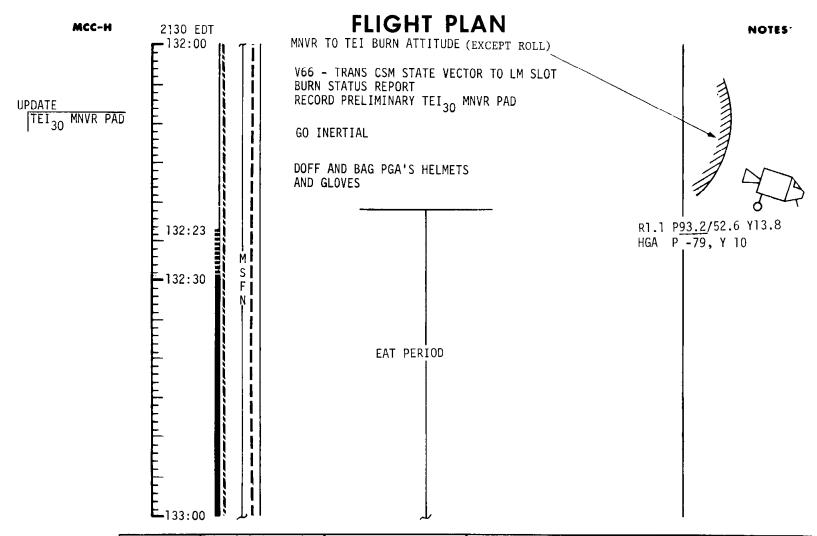


MSC Form 2189 (OT) (Nov 68)



6/29 3-97

MSC Form 29 (May 69)



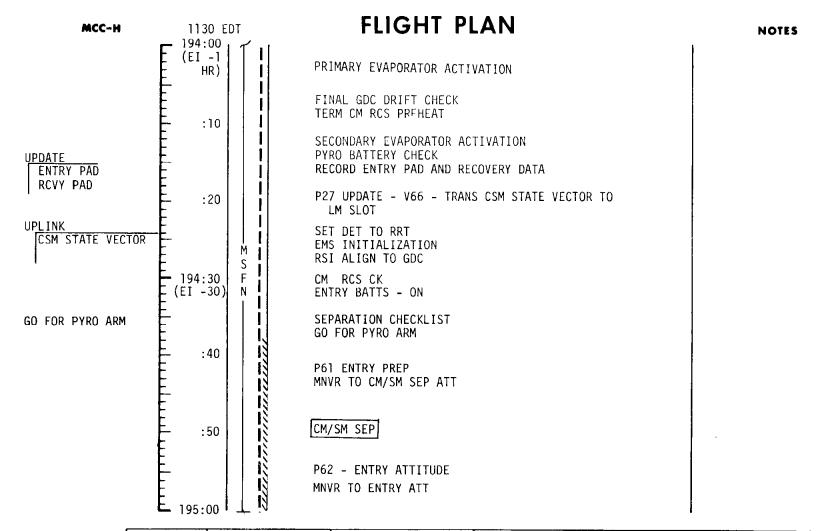
MISSION	EDITION	DATE	TIME	DAY/REV	PAGE
APOLLO 11	Revision A	July 8, 1969	132:00 - 133:00	6/29	3-98

MSC Form 29 (May 69)

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TEI BURN CHART

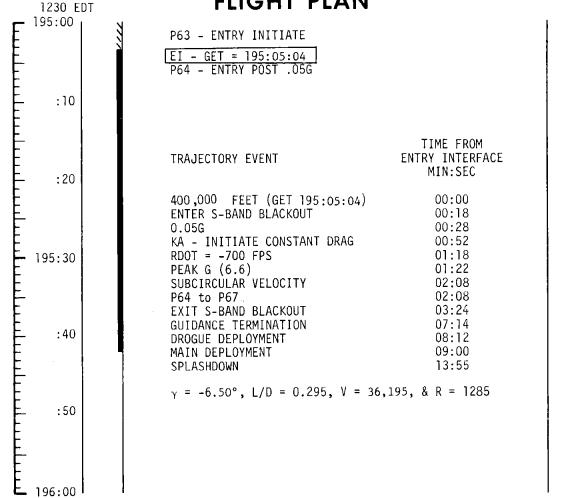
	P OR Y RATES	ATT DEVIAT		SHUTDOWN TIME	RESIDUALS		
TEI	10°/SEC TAKEOVER	$+10^{\circ}$ BT + 2 SEC & $\Delta V_{c} = -40$ FPS		TRIM X AXIS TO 0.2 FPS			
TEI ABORT MODES							
TEI V _{GO}	ВТ	TRAJECTORY	ABORT MODE				
3292.7 - 1436.0	0-01:30	LUNAR ORBIT	MODE	III - AFTER 1 F	REV		
1436.0 - 1207.0	1:30- 1:40	UNSTABLE	MODE II - 2 SPS BURNS FOR ORBIT STABILIZATION AND WATER OR CLA LANDING.				
1207.0 - Q	1:40-02:29	UNSTABLE/ HYPERBOLIC	MODE	P37 AT SE HYPERBOLI	PHERE OF INFLUENCE		



Ĺ	MISSION	EDITION	DATE	TIME	DAY/REV	PAGE
	APOLLO 11	FINAL	JULY 1, 1969	194:00 - 195:00	9/TEC	3-134

MSC Form 29 (May 69)

NOTES



MISSION	EDITION	DATE	TIME	DAY/REV	PAGE
APOLLO 11	Revision A	July 8, 1969	195:00 - 196:00	9/TEC	3-135

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