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Average Power Profiles For Apollo Guidance and Navigation In The Command Module and Lunar Excursion Module (V)



by

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# AVERAGE POWER PROFILES FOR APOLLO GUIDANCE AND NAVIGATION IN THE COMMAND MODULE AND LUNAR EXCURSION MODULE

The accompanying tables summarize the average power and energy as a function of time and Apollo mission phase for the Block 2 Command Module and Lunar Excursion Module Guidance and Navigation equipments as defined in the G&N System Status Report. \*

The values reported are center value (average) estimates. They are the latest expected values and should not be taken as "not to exceed" extremes.

Transient peaks of power occur at higher levels during turn-on and slewing operations. These are not included but are estimated to consume negligible energy. Data on these peaks will be published soon.

All power and energy numbers are those drawn from the spacecraft's primary supplies assumed exactly at the nominal 28 volt level.

Two charts are given each for the Command Module and Lunar Excursion Module Guidance and Navigation Systems.

A. Power Profile

This is a tabular listing of average power against time. The column entitled mission phase contains the various phases of the nominal mission time line identified below. The mission event column denotes a particular occurrence at each listed time. For convenience, elapsed time is recorded in both hours and minutes. Associated

<sup>\*</sup>MIT/IL Report E-1142, Rev. 18, March 15, 1964 (Confidential)

with each activity (or group of recurrent activities) is a beginning and an ending time. The "X" in columns 1 through 7 marks the AGE equipment in operation during each activity. Column 8 is the power level for the operating mode of the G&N activity. Columns 9 and 10 are self-explanatory.

#### B. Power and Energy Summary

This summarizes power and energy as a function of G&N activity and equipment identification.

The column labeled "G&N Activity" describes the operation configuration of the G&N system in words and gives associated power and total time in each configuration. The first row identifies the operating elements of the G&N system as selected by the "on-off" switches. Numbers at the intersections are total average energy consumption in kilowatt hours.

## Command Module G&N

Table I is a chronological power profile and mission time-line for the Block 2 Guidance and Navigation equipment in the Apollo Command Module.

Table II is a power and energy summary of the data in Table I.

The mission time-line upon which these tables are based is a "14-day" mission listed in the enclosure to North American Aviation, Inc., letter 64MA3540, dated March 12, 1964, submitted to MIT as the required electrical power design mission.

The operating details of the G&N system throughout the Command Module mission were based upon the following:

> Alignment of IMU: Considering expected platform coarse-alignment and fine-alignment times and the time required to acquire and "mark" the necessary

alignment stars, the time of fifteen (15) minutes for alignment of the platform in midcourse was established. Similar considerations were employed to define the alignment updating time interval of five (5) minutes.

- (2) Optical Measurements: The time required for midcourse star-landmark measurements (Sextant) was defined to require ten (10) minutes. The time available for telescope tracking was assumed to be the time that the landmark was visible, i.e., nearly horizon to horizon passage. This was calculated to be five (5) minutes for Earth Orbit and thirteen (13) minutes for Lunar Orbit. The number and frequency of optical measurements were based upon guidelines established by the Space Guidance Analysis Division.
- (3) Second Computer Update: Since the Block II G&N System has two computers, a time of one (1) minute was available to update the second computer after each sextant sighting, landmark tracking, and star occultation. For similar considerations the second computer was turned on for an additional fifteen (15) minutes prior to each major maneuver and throughout the maneuver to provide emergency backup. The major maneuvers are defined as ascent, translunar injection, lunar orbit injection, transearth injection, and entry.
- (4)  $\Delta V$  Corrections: A time of fifteen (15) minutes was established for the time necessary to accomplish and confirm each of the three  $\Delta V$  corrections during transearth and translunar operations.
- (5) IMU Operation: The IMU is in operation from prelaunch until after the first midcourse correction.
   It is then off during periods of free-fall in order to conserve energy.

#### Lunar Excursion Module G&N

Table III is a chronological power profile and mission timeline for the Guidance and Navigation equipment in the Apollo Lunar Excursion Module.

Table IV is a power and energy summary of the data in Table III.

The mission timeline upon which these tables are based is that listed in the preliminary LEM Interface Control Document ICD LIS-390-2 submitted from Grumman Aircraft Engineering Corporation to MIT as the required electrical load analysis form. (The incompatibilities between this LEM mission and the former one for the command module are recognized.)

The operating details of the LEM G&N system during the LEM mission are based upon considerations similar to those for the CM G&N equipment.

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	6	at not		78.06	1, 83	0.18	0.25	0.25	16,92	0.83	0,08	0.25	1.78	0.67	0.07	0.08	0.20	0.13	0.50	0.05	
FILE	8	SHBH		78.5	348.8	268.5	575.5	405.2	78.5	348,8	268.5	575.5	305.2	575.5	495.2	575.5	405.2	305.2	575.5	495. 2	
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ont)	Hours	e ,aunch	33 23	×*. 33			<u> </u>	102.65	102.90		-		120.98	-			123.58	123.78	123.91		
E I (cont)	Minutes	Time From Launch	0761	0401				6159	6174				7259				7415	7427	7434.6		
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		Mission Phase		¢						Trans-lunar	Coast										

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PROFILE	œ	attan	405, 2	575, 5	495.2	. 1	495, 2	575, 5	405.2	575.5	495, 2	575.5	305.9		575.5	495.2	305. 2	575.5	495, 2	405.2		575.5	
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BLOCK 2 CO		G&N Activity	Inactivity & Monitor 16,00% of the time	1 Alignment 10.66% of the time	Computer Update	•	Lunar Orbit Injection	l Alignment	Inactivity & Monitor 64, 25% of the time	6 Landmark Trackings 31.40% of the time	6 Computer Updates 2.42% of the time	l Alignment	I. 33% Of the time			l Computer Update	Inactivity & Monitor	Inspect Site With Sextant	l Computer Update	Inactivity & Monitor	>	l Alignment	
(cont)	Hours	Time From Launch			124.66	124.78	124.87	124.95					129, 09	129.32	129.53	129.54	129 70		163.01	129.88	130,00	130.08	
н	Minutes	Time From La			7479.6	7486.8	7492.2	7497	чо				7745	7759	7772	7773	7782	6077	7611	7793	7800	7805	
TABLE		Mission Event		Comniete IMII	Alignment	Start Lunar	Orbit Injection Service Module	Cutoff Complete First IMU 7497	Alignment After Injecti				Complete IMU Alignment	Start Landmark Tracking	Finish Landmark Tracking	Complete AGC Update	Start Inspection of	Landing Site with SXT Finish SXT hspection	of Landing Site	Update Auc	Begin IMU	Alignment Complete IMU	Highlight
		Mission Phase	 	Coast			Injection				Lunar Orbit	Coast											

Office PCC, No     FCC No     FCC No     FCC No     FCC No     FCC No       nitor     NO     NO     NO     NO     NO       acking     X     NO     NO     NO     NO       acking     X     NO     NO     NO     NO       acking     X     N     NO     NO     NO       nitor     X     N     N     N     N       cc. M.     X     N     N     N     N       date     X     N     N     N     N       cc. M.     X     N     N     N     N       nitor     X     N     N     N     N <t< th=""><th>TABLE</th><th>nut I</th><th>(cont) es Hours</th><th>BLOCK 2 COI</th><th>VIMA V</th><th>VD M(</th><th>COMMAND MODULE</th><th>4 G&amp;</th><th>S PC</th><th></th><th>PROFIL 7</th><th>ыø</th><th>ه</th><th>0</th></t<>	TABLE	nut I	(cont) es Hours	BLOCK 2 COI	VIMA V	VD M(	COMMAND MODULE	4 G&	S PC		PROFIL 7	ыø	ه	0
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K     X     405.2     0.04       K     X     X     405.2     0.04       K     X     X     405.2     32.92     1       K     X     X     X     575.5     0.50       K     X     X     405.2     0.07		L	Perilune	Retro	x			×					0,09	0.0365
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	٥	tion the table		0.11		0.01	1. 23	8.73	0.87	0, 07	0. 33	1. 72	0.03	3, 50	0.33	1. 42	0. 21	0.01	0.69		0.08	
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(cont)	Hours	Time From Launch	167.64	167.75	167.96	167.97	169.20				179 20			184_45		186 20	186 41	12 · 001	186.42	187.11	187.19	
·	Minutes	From	10058.4	10065.0	10078	10079	10152				10752			11067		11172	11185		11186	1- 11227	11232	
TABLE		Mission Event	Begin Ascent Trans- fer to 80 n.m.	Start Landmark Tracking	Finish Landmark Tracking	Complete AGC Update	Begin Period of LEM Trackings				L.F.M Docking	0		Start First Alignment	After Star Occultation Sightings	Start Landmark	Tracking Finish Landmark	Tracking Complete AGC	Update	Begin Last IMU Align ment Before Trans- earth Injection	Complete IMU Aliznment	D
		Mission Phase			Ascent Transfer to	80 n.m.			Rendezvous, Docking, &	Transfer				Lunar Orbit								

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	ion Event	Tir From	ne Launch			JON ON A CLAR	Solute is on	A CLAR A CLAR	FORDINE S CO	Site to the site of the site o	Solution of the second	ST, BA	Stropp Bad	Restrong to C
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$ \begin{array}{                                    $	u Trans-earth on	11250.6	187.51	Computer Update		× ;		×				495.2	67 N	U. 1238
	Trans-earth to 400,000 feet		187.55	Trans-earth Injection	\$	×		×	*					0 2106
$ \begin{array}{ c c c c c c c c c c c c c c c c c c c$	st Sextant ng After Trans- Injection	11324	188.73	9 Sextant Sightings	< >				; ×	×	×	348 8	1 50	0. 5232
$ \begin{array}{ c c c c c c c c c c c c c c c c c c c$				9.33% of the time 9 Computer Updates	4	×		_	×	:		268.5	0.15	0, 0403
$ \begin{array}{ c c c c c c c c c c c c c c c c c c c$				Inactivity & Monitor 88.17% of the time			×		×			78.5	14.17	1, 1123
$ \begin{array}{ c c c c c c c c c c c c c c c c c c c$			-	1 Alignment	*			×	+	×	×		0. 25	0.1439
$ \begin{array}{ c c c c c c c c c c c c c c c c c c c$	ete IMU nent	12280	204.80	1.55% of the time	< ▶			: >		:			0 25	0 1013
$ \begin{array}{ c c c c c c c c c c c c c c c c c c c$	ete ΔV <sub>1</sub>	12303	205.05	ΔV <sub>1</sub> Inactivity & Monitor	<		;	<	;				46 04	2 6848
$ \begin{array}{ c c c c c c c c c c c c c c c c c c c$				96.48% of the time			<		<				10 ° 01	
$ \begin{array}{c c c c c c c c c c c c c c c c c c c $				8 Sextant Sightings 2.73% of the time	×				×	×	x	348.8	1. 33	0.4639
$ \begin{array}{c ccccccccccccccccccccccccccccccccccc$				8 Computer Updates 0.27% of the time		×			×				0, 13	0, 0349
$\begin{array}{ c c c c c c c c c c c c c c c c c c c$				1 Alignment 0.51% of the time	×			x		×	×	5.	0. 25	0,1439
$\Delta V_2$ 15237         253.95         Inactivity & Monitor         X         X         78.5         47.97         3.           95.84% of the time         X         X         X         X         348.8         1.67         0.           10 Sextant Sightings         X         X         X         X         348.8         1.67         0.           10 Computer Updates         X         X         X         X         0.16         0.	nent	15222	253.70	$\Delta V_2$	×			×					0, 25	0, 1013
X         X         X         348.8         1.67         0.           X         X         X         268.5         0.16         0.	tete ∆V2	15237	253, 95	Inactivity & Monitor 95, 84% of the time			×		×				47.97	3. 7656
X 268.5 0.16				10 Sextant Sightings 3, 33% of the time	×				x	×	×	348.8	1.67	0.5790
				10 Computer Updates 0.32% of the time		×			×			S	0, 16	0,0429

		<b> </b>							T		•				· [					į	1	II	TT
	0	Palitinstro		0.1439	0.4273	0, 3856	0.0347	0.0400	0. 1013	0 0103	0. 0103	0.2878	0.0248	0.0243	0 0460		0. 0445	0, 1089	0.3714	0	54.089		
	6	Strott		0.25	1. 40	0.67	0.07	0, 08	0. 25	90	90 <sup></sup> 0	0.50	0. 05	0, 06	80		0.11	0. 22	0.75	72.0	380. 30		
'ILE	8	SII BA		575.5	305.2	575.5	495.2	575.5	405.2	305 9	200. 4	575.5	495.2	405.2	2 2 2			480.2	495.2	0	1		
PROFILE	2	s <sup>3</sup> <sup>1</sup> <sup>1</sup> <sup>1</sup> <sup>1</sup> <sup>4</sup> , <sup>1</sup> <sup>0</sup> <sup>1</sup>		×		×		×				×			>	4				I	TOTAL		
POWER	ç	23 COCCAR		×		×		×				×			>	<							
N POI	5	ACIDUESS S. BC											1							1			
E G&N	4	and the sea		×	×	×	×	×	x	>	٢.	х	х	×	>	د :	×	۲.	×				
DUL		Solution O.			×					>	<												
ID MC	2	Deternon					×						x				1	×	x				
COMMAND MODULE	-	318 <sup>313</sup> 84 318 <sup>313</sup> 84 3205 <sup>43</sup> 60 3105 <sup>43</sup> 60 3105 <sup>43</sup> 60 3105 <sup>4</sup> 7 3105 <sup>4</sup> 7 31		×		×		×	×	-		×		×	>	< :	×			Į			
BLOCK 2 CON		G&N Activity		1 Augument 0.50% of the time	Inactivity & Monitor 63.06% of the time	4 Sextant Sighting 30.18% of the time	4 Computer Updates 3.15% of the time	1 Alignment 3.60% of the time	۵V <sub>3</sub>	Inactivity & Monitor	8.00% of the time	3 Sextant Sightings 66.67% of the time	3 Computer Updates 6.67% of the time	Inactivity & Monitor 8.00% of the time	1 Alignment	10.66% of the time	Inactivity & Monitor	Computer Update	Entry & Landing	Post Landing			
(cont)	Hours	ae Launch	<b>I</b>	304.00		L			300.22	306.47						307.22	307.33	307.55	00 20	ne Bue	380. 30	±	
H	Minutes	Time From Launch		18240					183/3	18388						18433	18440	18453	00101	06401	72818		
TABLE		Mission Event		Complete IMU	Augnment			Complete IMU	Alignment	Complete $\Delta V_3$						Complete IMU Alignment	Begin Final AGC Update	Begin Entry	a differ t at the construction	Соприете напалив	End Post Landing		
		Mission Phase				Trans-earth	Coast to	1991 000 001								·		Entry &	Landing	Post Landing			

	T										
		TOTAL	22.577	7.533	0. 518	2.496	17.059	2.469	l. 471	0.196	54, 089
SUMMARY	2	07)1154 07:07 a 115 0.50 f 85 1100115		1.658		0.936	1				2, 594
	9	D&C 10 10 10 10 10 10 10 10 10 10 10 10 10		0.503		0. 283	1				0.786
COMMAND MODULE G&N POWER AND ENERGY LOCK II COMMAND MODULE ENERGY CONSUMPTION (Kilowatt Hours)	S	511:01.01.922 511:00.101.92 513:00.101.2 513:001.01.2 513:001.01.2 513:001.01.927			0.170	0. 418	12.713			0.043	13. 344
WER AN PTION (Kilow	4	1111 110 110 1111 110 110 11 10 10 10 10 10 10 10 10 10 10 10 10 10 10 10 10 10 1	15. 891	3.619				2. 307	0.847		22.664
LE G&N POWER AND EN ENERGY CONSUMPTION (Kilowatt Hours)	3	6 AGC 110: 513 110: 20 10 11011:5 25 40 110111:5	1				4.346	0.162			4.508
AODULE ENE	1	TWO CONTAINS NGC ON STATES 210 NOW STO	1						0.624	0.153	0. 777
COMMAND MODU	-	sinoil sinoil	6,686	1, 523	0, 348	0.859					9, 416
TABLE II BLOCK 2 COM SUMMARY OF BLOCK 1		G& N ACTIVITY	Accomplish & confirm course corrections Inactivity & Monitor 405.2 watts 55.72 hrs.	Alignments Landmark Trackings SXT Sightings 575.5 watts	Star Occultation Sightings 178. 5 watts 2.90 hrs.	SXT Sightings (Midcourse) 348.8 watts 7.16 hrs.	Inactivity & Monitor 78.5 watts 217.31 hrs.	Inactivity & Monitor 305, 2 watts 8, 09 hrs.	Computer Updates Major Maneuver's 495, 2 watts 2, 97 hrs.	Computer Updates 268.5 watts 0.73 hrs.	TOTAL 308. 30 hours
		Operating Mode	P	В	υ	G	n	ĬL,	U	H	

	TABLE	LE III	LUNAR	EXCURSION MODULE	ION N	IODUI	1	G&N P(	POWER		PROFILE			
	Minutes	Hours		1	2	3	4	5	Ŷ	2	8	6	10	
Mission Phase	Time From L	Time From Launch	LEM G&N Activity	PGC Dff Off U. 0 Waltis	AGC Opera	owat'	IMU ate Operatio	UNIT Starts	Mo CDU CDU CDU terate	Ontur.) (A.O.Te Operate Negligible	Watts	Hours peration	Kilours Hours Consumed	
	- 300	- 2		;				\$			u u	с с ч	0000	
Pre-launch (countdown	(1) 0		Inactivity	×				×			с. яс	00. G	0. 293	
Launch and Earth Ascent	>		Inactivity	×				×			58.5	0.20	0.012	
Farth Orbit	12	0.20	Inactivity	×				×			58.5	3.00	0. 176	
E	192	3.20												
transiunar insermon & Initial Coast	200	0	Inactivity	×				×			58.5	0.58	0.034	
C.S.M. Transposition	194		Inactivity	×				×			58.5	0.50	0.029	
Jettison S IVB	257	4.28	Inactivity	×				×			58.5	0.34	0.020	
LEM Translunar	277	4.62			>	_	*		*	Mediath	e 427 7	00	0 428	
Checkout	337	5.62	Law N Checkout		<	-	•							_
Continue Translunar Trio	100		Inactivity	x				x			58.5	76.66	4.484	
Insert Into	4937	82.28	Inactivity	×				x			58.5	0.10	0, 006	
Innia Ornit	4943	82.38											0.00	
1/2 Lunar Orbit	2004	03 40	Inactivity	×				×			58. 5	1. 02	0.000	
LEM Checkout	-	01.00	G&N Checkout		×		×		×	Negligible	427.7	1. 02	0. 436	
Continue Lunar Orbit	5065	84.42	and Initial IMU Alignment		×		×		×	Negligible	427.7	0.41	0.175	
	5090	84.83					-							
Separate - 100 feet	5091 3	84 85			×		×		×		427.7	0.02	0, 009	
Orient LEM	C. 1000	00 50 U	Final IMU Alignment		×		x		x	Negligible	427.7	0, 15	0.064	
Insert Into Synchronous Orbit	0010	20. C	Descent Injection Guidance		x		x		×		4.277	0.01	0.004	
Coast in	5100.5	85.01	Inactivity		×		×		×	Negligible	427.7	2.04	0.873	
ayneni onous or ou	5223.1	87.05			-									
Coast to Pericynthion			Re-align IMU		×		×		×	Negligible	427.7	0.49	0. 210	
	5252.5	87.54												
				-		-					-	-		

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1	<u> </u>		[					
		Total	5. 113	5. 013	0, 085	l. 373	l. 075	12.659
	2	5 1 10 4 2 5 . 9 1 9 1 9 5 . (	1	Negligible		1	Negligible	Negligible
	9	4100416. 5116416. 49005. 19005. 19005. 19005. 1005. 1005. 1005. 1005. 1005. 1005. 1004. 1004. 1004. 1004. 1004. 1004. 1004. 1004. 1004. 100. 1105.	1	0, 298	0.005		1	0.303
	5	5004 51 65 65 510 51 65 65 5004 65 5004 65 5004 65 5004 65	5.113			1, 173	0.918	7.204
	4	STROUTE		3. 484	0.059			3. 543
t Hours	3	10.55 . 10.11 2.50 - 10.11 2	1		1	0.20	0.157	0, 357
Consumption in Kilowatt Hours	2		-	1, 231	0, 021	I	ł	1. 252
ergy Consumpt	-	51104 0.0	0.000					0.000
Summary of Lunar Excursion Module Energy		LEM G&N Activity	Inactivity 87,40 hours 58,5 watts	Inactivity Alignment Midcourse Measurements 11.72 hours 427.7 watts	Guidance During Major Event 0.20 hours 427.7 watts	Inactivity 20.05 hours 68.5 watts	Inactivity 15.70 hours 68.5 watts	Total
		Mode	н	II	III	N	Λ	

## E-1557

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