

12-75

[ CHANGE SHEET

FOR

MSC INTERNAL NOTE 70-FM-141 DATED SEPTEMBER 25, 1970 ;

SPACECRAFT OPERATIONAL TRAJECTORY FOR APOLLO 14

(MISSION H-3) LAUNCHED JANUARY 31, 1971,

VOLUME I - MISSION PROFILE ]

By Landing Analysis Branch, Orbital Mission Analysis Branch,  
and Lunar Mission Analysis Branch

Change 1

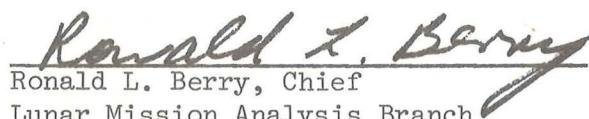
December 11, 1970



Floyd W. Bennett, Chief  
Landing Analysis Branch



Edgar C. Lineberry, Chief  
Orbital Mission Analysis Branch



Ronald L. Berry, Chief  
Lunar Mission Analysis Branch



John F. Mayer, Chief  
Mission Planning and Analysis Division

Page 1 of \_\_\_\_\_  
(with enclosures)

NOTE: A black bar in the margin indicates the area of change.

After the attached enclosures, which are replacement pages, have been inserted, place this CHANGE SHEET between the cover and title page and write on the cover, "CHANGE 1 inserted."

1. Replace page iii.

2. Add page iiia.

INDEXING DATA

DATE

OPR

#

T

PGM

SUBJECT

SIGNATOR

LOC

12-11-70

MSC

R

MPA

[ ]

MSC

079-53

CHANGE HISTORY FOR 70-FM-141

Change	Date	Description
1	12/11/70	Replace page iii and add page iiia: addition to foreword.

## FOREWORD

The trajectory for the Apollo 14 mission was computed without taking into account the weight increase caused by the addition of a third oxygen tank and a battery in the service module. These modifications are being made as a result of the Apollo 13 accident.

For all previous Apollo missions, the spacecraft operational trajectory document has been revised 2 months before launch. This permitted the flight crews and flight controllers to train on the latest available trajectory data. However, it has been determined, as recorded in MSC memorandum 70-FM-H-55, that the real-time changes have been equal to those occurring from updating the trajectory. Therefore, it is inadvisable to train to specific mission-dependent numbers. In addition, a technique for updating the ground elapsed time has been devised for Apollo 14. This will allow the ground elapsed time of the mission to match that of the flight plan, eliminating the need for extensive flight plan updates in real time. Based on these facts and because no major changes have occurred since the operational trajectory document was published, the decision has been made not to generate a revised trajectory.

The following is a list of the changes that have been identified to date.

- a. CSM weight increase of  $\approx$ 660 pounds.- This change resulted from adding the third oxygen tank, battery, and associated hardware.
- b. Three-minute change in lift-off time from 20:20 G.m.t. to 20:23 G.m.t. - This change occurred because the operational trajectory was generated before the official lift-off time was determined by MSFC. In the past, this time difference was less than a minute. New procedures are being implemented that will permit using the official lift-off time in the operational trajectory.
- c. LM descent targeted for desired landing point
- d. Preferred REFSMMAT for TEI rather than ascent REFSMMAT
- e. The CSM separation maneuver prior to landing using four jets rather than two jets
- f. S-IVB impact point moved

Change 1, December 11, 1970

g. The CSM separation maneuver attitude, prior to LM ascent stage impact, changed

The decision not to revise the operational trajectory means that the following additional documents will not be revised.

Simulator Data Pack

Flight Planning Data

Updates will be made to the consumables crew charts and to any other charts that are sensitive to the changes listed previously.

iiia

Change 1, December 11, 1970  
NASA — MSC



NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

14  
Apollo a-58

MSC-02629

VOLUME I

MSC INTERNAL NOTE NO. 70-FM-141

September 25, 1970

THE SPACECRAFT OPERATIONAL  
TRAJECTORY FOR APOLLO 14  
(MISSION H-3)

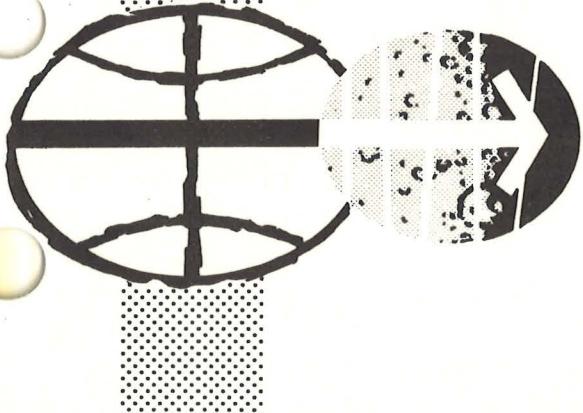
LAUNCHED JANUARY 31, 1971

VOLUME I - MISSION PROFILE

Lunar Mission Analysis Branch,  
Orbital Mission Analysis Branch,  
and  
Landing Analysis Branch

MISSION PLANNING AND ANALYSIS DIVISION

MANNED SPACECRAFT CENTER  
HOUSTON, TEXAS



MSC-02629  
Volume I

MSC INTERNAL NOTE NO. 70-FM-141

PROJECT APOLLO

THE SPACECRAFT OPERATIONAL TRAJECTORY FOR APOLLO 14  
(MISSION H-3) LAUNCHED JANUARY 31, 1971  
VOLUME I - MISSION PROFILE

By Lunar Mission Analysis Branch, Orbital Mission Analysis Branch,  
and Landing Analysis Branch

September 25, 1970

MISSION PLANNING AND ANALYSIS DIVISION  
NATIONAL AERONAUTICS AND SPACE ADMINISTRATION  
MANNED SPACECRAFT CENTER  
HOUSTON, TEXAS

Approved:

  
Ronald L. Berry, Chief  
Lunar Mission Analysis Branch

Approved:

  
Edgar C. Lineberry, Chief  
Orbital Mission Analysis Branch

Approved:

  
Floyd V. Bennett, Chief  
Landing Analysis Branch

Approved:

  
John P. Mayer, Chief  
Mission Planning and Analysis Division

## FOREWORD

The trajectory for the Apollo 14 mission was computed without taking into account the weight increase caused by the addition of a third oxygen tank and a battery in the service module. These modifications are being made as a result of the Apollo 13 accident. The weight of these modifications is approximately 754 pounds and was not incorporated into the computations because the input was frozen on July 13 and this weight was not defined for the Mission Planning and Analysis Division until August 18. A revision to this trajectory is scheduled for publication on December 1, and it will incorporate this weight change and any other changes that might occur between now and the input freeze date of October 1, 1970.

The time of day of the launch as given in this document is approximate only and will be defined exactly in the Marshall Space Flight Center Operational Launch Trajectory. The difference between the two times should not exceed 5 minutes.

## CONTENTS

Section		Page
1.0	SUMMARY AND INTRODUCTION . . . . .	1-3
	1.1 Summary . . . . .	1-5
	1.2 Introduction . . . . .	1-6
2.0	SYMBOLS AND NOMENCLATURE . . . . .	2-3
3.0	PRIMARY GUIDELINES AND CONSTRAINTS . . . . .	3-3
4.0	MISSION SUMMARY . . . . .	4-3
	4.1 Earth Launch . . . . .	4-7
	4.2 Earth Parking Orbit . . . . .	4-11
	4.3 Translunar Injection (TLI) . . . . .	4-15
	4.4 Free-return Circumlunar Trajectory . . . . .	4-19
	4.5 Posttranslunar Injection Events . . . . .	4-23
	4.6 Translunar Coast . . . . .	4-31
	4.7 Lunar Orbit Insertion (LOI) . . . . .	4-41
	4.8 Descent Orbit Insertion (DOI) . . . . .	4-49
	4.9 CSM/LM Coast From LOI to Undock . . . . .	4-53
	4.10 CSM/LM Undock to Landing . . . . .	4-57
	4.10.1 CSM/LM undock and separation . . . . .	4-57
	4.10.2 CSM circularization . . . . .	4-57
	4.10.3 Powered descent . . . . .	4-58
	4.10.4 Ullage and trim phases . . . . .	4-58
	4.10.5 Braking phase . . . . .	4-58
	4.10.6 Approach phase . . . . .	4-58
	4.10.7 Landing phase . . . . .	4-58
	4.10.8 Graphical description of the powered descent . . . . .	4-59

Section	Page
4.11 CSM Lunar Orbit Plane Change . . . . .	4-85
4.12 LM Ascent . . . . .	4-89
4.13 Rendezvous Sequence . . . . .	4-105
4.13.1 Sequence summary . . . . .	4-105
4.13.2 Tweak maneuver . . . . .	4-105
4.13.3 TPI . . . . .	4-106
4.13.4 Midcourse corrections and TPF . . . . .	4-107
4.14 LM Jettison and Separation to Transearth Injection . . . . .	4-123
4.15 Transearth Injection . . . . .	4-131
4.16 Transearth Coast . . . . .	4-139
4.17 Entry . . . . .	4-147
5.0 INPUT DATA . . . . .	5-3
6.0 REFERENCES . . . . .	6-3

TABLES

Table		Page
1.1-I	SEQUENCE OF MAJOR EVENTS . . . . .	1-8
1.1-II	LUNAR ORBIT ACTIVITIES . . . . .	1-14
1.1-III	LAUNCH WINDOW SUMMARY . . . . .	1-16
1.1-IV	TIMETABLE OF MISSION EVENTS . . . . .	1-17
1.1-V	MISSION RADAR TIMELINE	
	(a) CSM acquisition and termination during rendezvous - 0° minimum elevation . . . . .	1-19
	(b) LM acquisition and termination during descent - 0° minimum elevation . . . . .	1-66
	(c) LM acquisition and termination during ascent - 0° minimum elevation . . . . .	1-69
	(d) LM acquisition and termination during rendezvous - 0° minimum elevation . . . . .	1-70
	(e) CSM acquisition and termination during rendezvous - 3° minimum elevation . . . . .	1-78
	(f) LM acquisition and termination during descent - 3° minimum elevation . . . . .	1-123
	(g) LM acquisition and termination during ascent - 3° minimum elevation . . . . .	1-126
	(h) LM acquisition and termination during rendezvous - 3° minimum elevation . . . . .	1-127
1.1-VI	MISSION SHADOW TIMELINE	
	(a) CSM . . . . .	1-135
	(b) LM descent . . . . .	1-150
	(c) LM rendezvous . . . . .	1-151
4.5-I	SUMMARY OF EVENTS FROM TLI CUTOFF THROUGH LOX DUMP . . . . .	4-24
4.6-I	TARGET LOAD FOR RELAXED FREE-RETURN MANEUVER	
	(a) Target . . . . .	4-33
	(b) REFSMMAT . . . . .	4-33
	(c) Gimbal angles at $t_{IG}$ . . . . .	4-33

Table		Page
4.7-I	TARGET LOAD FOR LOI MANEUVER	
(a)	Target . . . . .	4-42
(b)	REFSMMAT . . . . .	4-42
(c)	Gimbal angles at $t_{IG}$ . . . . .	4-42
4.8-I	TARGET LOAD FOR LOI-2 MANEUVER	
(a)	Target . . . . .	4-50
(b)	REFSMMAT . . . . .	4-50
(c)	Gimbal angles at $t_{IG}$ . . . . .	4-50
4.10-I	DATA FOR MANUAL CSM SEPARATION MANEUVER	
(a)	Desired $\Delta V$ 's . . . . .	4-61
(b)	CSM REFSMMAT . . . . .	4-61
(c)	CSM gimbal angles at $t_{IG}$ . . . . .	4-61
4.10-II	TARGET LOAD FOR CSM CIRCULARIZATION MANEUVER	
(a)	Target . . . . .	4-62
(b)	REFSMMAT . . . . .	4-62
(c)	Gimbal angles at $t_{IG}$ . . . . .	4-62
4.10-III	TARGET LOAD FOR LM POWERED DESCENT MANEUVER	
(a)	Ignition targets for PDI (P63) . . . . .	4-63
(b)	Braking phase targets . . . . .	4-64
(c)	Approach phase targets (P64) . . . . .	4-65
(d)	Vertical descent targets (P66) . . . . .	4-65
4.11-I	TARGET LOAD FOR LUNAR ORBIT PLANE CHANGE MANEUVER	
(a)	Target . . . . .	4-86
(b)	REFSMMAT . . . . .	4-86
(c)	Gimbal angles at $t_{IG}$ . . . . .	4-86
4.12-I	LM ASCENT WEIGHT SUMMARY . . . . .	4-90
4.12-II	TARGET LOAD FOR LM ASCENT	
(a)	Target . . . . .	4-91
(b)	REFSMMAT . . . . .	4-91
(c)	Gimbal angles at $t_{IG}$ . . . . .	4-91

Table		Page
4.12-III	LM ASCENT PROPULSION PERFORMANCE SUMMARY . . . . .	4-92
4.13-I	RENDEZVOUS MANEUVER SUMMARY . . . . .	4-108
4.13-II	MSFN TRACKING AND SHADOW TIMELINE . . . . .	4-109
4.13-III	TARGET LOADS FOR TPI MANEUVER	
	(a) Target . . . . .	4-110
	(b) REFSMMAT . . . . .	4-110
	(c) Gimbal angles at $t_{IG}$ . . . . .	4-110
4.14-I	TARGET LOAD FOR CSM SEPARATION MANEUVER FOLLOWING LM JETTISON	
	(a) Target . . . . .	4-124
	(b) REFSMMAT . . . . .	4-124
	(c) Gimbal angles at $t_{IG}$ . . . . .	4-124
4.14-II	LM DEORBIT SUMMARY . . . . .	4-125
4.14-III	TARGET LOAD FOR LM DEORBIT	
	(a) Target . . . . .	4-126
	(b) REFSMMAT . . . . .	4-126
	(c) Gimbal angles at $t_{IG}$ . . . . .	4-126
4.15-I	TARGET LOAD FOR TEI MANEUVER	
	(a) Target . . . . .	4-132
	(b) REFSMMAT . . . . .	4-132
	(c) Gimbal angles at $t_{IG}$ . . . . .	4-132
4.17-I	ENTRY EVENTS SEQUENCE . . . . .	4-149
4.17-II	COMMAND MODULE MASS PROPERTIES . . . . .	4-150
4.17-III	CONDITIONS AT ENTRY INTERFACE AND TARGET POINT . . .	4-151
5-I	LAUNCH PAD REFSMMAT	
	(a) Applicable maneuvers . . . . .	5-4
	(b) Definition of REFSMMAT . . . . .	5-4
	(c) REFSMMAT components . . . . .	5-4

Table		Page
5-II	PTC REFSMMAT	
(a)	Applicable maneuvers . . . . .	5-5
(b)	Definition of REFSMMAT . . . . .	5-5
(c)	REFSMMAT components . . . . .	5-5
5-III	LANDING SITE REFSMMAT	
(a)	Applicable maneuvers . . . . .	5-6
(b)	Definition of REFSMMAT . . . . .	5-6
(c)	REFSMMAT components . . . . .	5-6
5-IV	PREFERRED (LOPC) REFSMMAT	
(a)	Applicable maneuver . . . . .	5-7
(b)	Definition of REFSMMAT . . . . .	5-7
(c)	REFSMMAT components . . . . .	5-7
5-V	LM ASCENT REFSMMAT	
(a)	Applicable maneuvers . . . . .	5-8
(b)	Definition of REFSMMAT . . . . .	5-8
(c)	REFSMMAT components . . . . .	5-8
5-VI	ENTRY REFSMMAT	
(a)	Applicable maneuvers . . . . .	5-9
(b)	Definition of REFSMMAT . . . . .	5-9
(c)	REFSMMAT components . . . . .	5-9
5-VII	SPACECRAFT WEIGHT SUMMARY . . . . .	5-10
5-VIII	ASSUMED MISSION-INDEPENDENT EXPENDABLES . . . . .	5-11
5-IX	SERVICE MODULE PROPULSION PERFORMANCE SUMMARY . . . . .	5-12
5-X	LUNAR MODULE PROPULSION PERFORMANCE SUMMARY . . . . .	5-13
5-XI	COMMAND MODULE AERODYNAMIC COEFFICIENTS . . . . .	5-14
5-XII	LUNAR LANDING AND PHOTOGRAPHY SITE POSITIONS . . . . .	5-15
5-XIII	IMU ALINEMENT CRITERIA AND REFSMMAT DEFINITIONS . . . . .	5-16

Table		Page
5-XIV	LM BRAKING SCHEDULE DURING TERMINAL RENDEZVOUS PHASE . . . . .	5-17
5-XV	MSFN STATION CHARACTERISTICS . . . . .	5-18
5-XVI	MSFN STATION IDENTIFICATION . . . . .	5-19

## FIGURES

Figure		Page
1.1-1	Schematic of the hybrid mission profile . . . . .	1-7
4.2-1	Earth parking orbit groundtrack . . . . .	4-12
4.3-1	TLI geometry . . . . .	4-16
4.5-1	Motion of the S-IVB relative to the spacecraft for the S-IVB evasive maneuver of Apollo 14	
	(a) Range versus time . . . . .	4-25
	(b) Cross range versus horizontal displacement . . . . .	4-26
	(c) Vertical displacement versus horizontal displacement . . . . .	4-29
4.6-1	Translunar coast groundtrack . . . . .	4-34
4.6-2	Time history of altitude for first 10 hours of translunar coast . . . . .	4-35
4.6-3	Time history of true anomaly for first 10 hours of translunar coast . . . . .	4-36
4.6-4	Altitude time history from translunar injection to lunar orbit insertion . . . . .	4-37
4.6-5	Time history of altitude 10 hours prior to LOI ignition . . . . .	4-38
4.7-1	LOI geometry . . . . .	4-43
4.7-2	Lunar orbit groundtracks from LOI to LM landing . . . . .	4-44
4.7-3	Time history of trajectory parameters for LOI	
	(a) Velocity, flight-path angle, and altitude versus time from LOI burn initiation . . . . .	4-45
	(b) Local horizontal pitch and yaw versus time from LOI burn initiation . . . . .	4-46
4.10-1	LM and CSM events from undocking to landing . . . . .	4-66

Figure		Page
4.10-2	Time history of LM and LM-CSM trajectory parameters from undocking to landing	
	(a) LM total velocity, altitude above landing site, and flight-path angle . . . . .	4-67
	(b) LM-CSM range, range rate, and phase angle . . . . .	4-68
4.10-3	LM-CSM relative motion from undocking to landing (curvilinear coordinate system) . . . . .	4-69
4.10-4	Powered descent profile . . . . .	4-70
4.10-5	LM powered descent groundtrack . . . . .	4-71
4.10-6	Time histories of LM and LM-CSM trajectory parameters from PDI to landing	
	(a) LM velocity, altitude, and flight-path angle . . . . .	4-72
	(b) LM-CSM range, range rate, and phase angle . . . . .	4-73
	(c) FDAI pitch angle and thrust angle from local vertical . . . . .	4-74
	(d) LM thrust magnitude . . . . .	4-75
4.10-7	Altitude above landing site versus altitude rate . . . . .	4-76
4.10-8	S-band gimbal angle profile . . . . .	4-77
4.10-9	Terrain altitude . . . . .	4-78
4.10-10	Approach phase summary . . . . .	4-79
4.10-11	Topography around the landing site Fra Mauro . . . . .	4-80
4.10-12	Landing phase summary . . . . .	4-81
4.12-1	LM ascent . . . . .	4-93
4.12-2	Vertical rise phase . . . . .	4-94
4.12-3	Orbit insertion phase . . . . .	4-95
4.12-4	LM and CSM groundtracks . . . . .	4-96

Figure		Page
4.12-5	Time histories of LM trajectory parameters from lunar lift-off to LM insertion	
	(a) LM altitude, inertial velocity, and inertial flight-path angle . . . . .	4-97
	(b) LM thrust angle and FDAI angles . . . . .	4-98
4.12-6	LM altitude versus altitude rate during powered ascent . . . . .	4-99
4.12-7	LM-CSM phase angle, range, and range rate versus time from lift-off and vertical range versus horizontal range (CSM referenced) . . . . .	4-100
4.12-8	LM-CSM range versus LM-CSM range rate during powered ascent . . . . .	4-101
4.12-9	S-band gimbal angle profile during powered ascent . . .	4-102
4.13-1	Orbital schematic of ascent through docking . . . . .	4-111
4.13-2	Parameters from LM orbit insertion through terminal rendezvous	
	(a) LM-CSM elevation angle versus time from LM orbit insertion . . . . .	4-112
	(b) CSM-LM-sun angle versus time from LM orbit insertion . . . . .	4-113
	(c) CSM lead angle versus time from LM orbit insertion . . . . .	4-114
	(d) LM-CSM range versus time from LM orbit insertion . . . . .	4-115
	(e) LM-CSM range rate versus time from LM orbit insertion . . . . .	4-116
	(f) Vertical relative range versus horizontal relative range . . . . .	4-117
4.13-3	Mission events and tracking summaries for LM-active rendezvous . . . . .	4-118
4.13-4	LM-CSM relative geometry at TPI . . . . .	4-119
4.14-1	Groundtrack from LOPC to TEI . . . . .	4-127

Figure		Page
4.15-1	Time history of trajectory parameters for transearth injection	
	(a) Velocity, flight-path angle, and altitude versus time from TEI burn initiation . . . . .	4-133
	(b) Local horizontal pitch and yaw versus time from TEI burn initiation . . . . .	4-134
4.15-2	TEI geometry . . . . .	4-135
4.16-1	Transearth coast groundtrack . . . . .	4-140
4.16-2	Altitude time history during the first 10 hours of transearth coast . . . . .	4-141
4.16-3	Altitude time history from transearth injection to entry interface . . . . .	4-142
4.16-4	Altitude time history 10 hours prior to entry interface . . . . .	4-143
4.17-1	Entry corridor . . . . .	4-152
4.17-2	Maneuver footprint and nominal groundtrack . . . . .	4-153
4.17-3	Geodetic altitude versus range-to-go . . . . .	4-154
4.17-4	CMC commanded bank angle, altitude, and altitude rate time histories . . . . .	4-155
4.17-5	Total aerodynamic heating rate and heat load time histories . . . . .	4-156
4.17-6	Time histories of velocity and flight-path angle for entry	
	(a) Relative and inertial velocity . . . . .	4-157
	(b) Relative and inertial flight-path angle . . . . .	4-157
4.17-7	Total propellant consumed from CSM separation . . . . .	4-158
4.17-8	Communications blackout . . . . .	4-159
4.17-9	Primary DSKY display VERB 06 NOUN 74 . . . . .	4-160

Figure	Page
4.17-10 DSKY displays (final phase) VERB 06 NOUN 66 . . . . .	4-161
4.17-11 Relative velocity and relative flight-path angle time histories from drogue parachute deployment . . .	4-162
4.17-12 Geodetic altitude and load factor time histories from drogue parachute deployment . . . . .	4-163
4.17-13 EMS parameters	
(a) Load factor versus inertial velocity . . . . .	4-164
(b) CMC commanded bank angle and EMS range-to-go versus inertial velocity . . . . .	4-164

1.0 SUMMARY AND INTRODUCTION

2.0 SYMBOLS AND NOMENCLATURE

3.0 PRIMARY GUIDELINES AND CONSTRAINTS

4.0 MISSION SUMMARY

4.1 EARTH LAUNCH

4.2 EARTH PARKING ORBIT

4.3 TRANSLUNAR INJECTION

4.4 FREE-RETURN CIRCUMLUNAR

4.5 POST-TRANSLUNAR INJECTION EVENTS

4.6 TRANSLUNAR COAST

4.7 LUNAR ORBIT INSERTION

4.8 DESCENT ORBIT INSERTION

4.9 CSM/LM COAST FROM LOI TO UNDOCK

4.10 CSM/LM UNDOCK TO LANDING

4.11 CSM PLANE CHANGE

4.12 LM ASCENT

4.13 RENDEZVOUS SEQUENCE

4.14 LM JETTISON AND SEPARATION TO TEI

4.15 TRANSEARTH INJECTION

4.16 TRANSEARTH COAST

4.17 ENTRY

5.0 INPUT DATA

6.0 REFERENCES

## THE SPACECRAFT OPERATIONAL TRAJECTORY FOR

APOLLO 14 (MISSION H-3)

LAUNCHED JANUARY 31, 1971

VOLUME I - MISSION PROFILE

By Lunar Mission Analysis Branch, Orbital Mission Analysis Branch,  
and Landing Analysis Branch

## 1.0 SUMMARY AND INTRODUCTION

## 1.1 Summary

In volumes I and II of the Spacecraft Operational Trajectory, a detailed description of the hybrid mission profile for Apollo 14 is presented. The nominal launch date of January 31, 1971 was used for this trajectory simulation.

Translunar injection places the spacecraft on a free-return trajectory with a high perilune altitude. The evasive maneuver is performed by the S-IVB prior to the blowdown as was done on Apollo 13. The S-IVB blowdown maneuver will place the discarded S-IVB on a lunar impact trajectory. Twenty-eight hours after TLI, an SPS transfer maneuver is performed to lower perilune to 57 n. mi. After the maneuver, the spacecraft is on a relaxed-free-return trajectory which is restricted such that an earth return would be within the DPS docked ΔV capability if an SPS failure should occur at LOI ignition.

The lunar orbit time line is similar to that of Apollo 12 but does not include a plane change after LM jettison to gain photographic coverage of selected prime sites for future landing missions. After LM jettison, there is a LM ascent stage deorbit maneuver which results in a lunar impact as was done on Apollo 12. TEI occurs at the start of revolution 35.

For the nominal mission, the launch time is  $14^{\text{h}}20^{\text{m}}$  c.s.t., and the flight azimuth is  $\sim 72^{\circ}$ . Translunar injection occurs during the second revolution over the Pacific Ocean.

The mission duration is approximately 9 days. The translunar flight time is  $80^{\text{h}}02^{\text{m}}$ . The time spent in lunar orbit is approximately  $66^{\text{h}}31^{\text{m}}$ , and the transearth flight time is  $67^{\text{h}}10^{\text{m}}$ . An inertial plot of the basic profile is shown in figure 1.1-1.

The nominal mission is summarized in the following tables.

- a. Table 1.1-I.- Sequence of major events
- b. Table 1.1-II.- Lunar orbit activities
- c. Table 1.1-III.- Launch window summary
- d. Table 1.1-IV.- Timetable of mission events
- e. Table 1.1-V.- Radar timeline
- f. Table 1.1-VI.- Shadow timeline

## 1.2 Introduction

The operational trajectory document is divided into numbered sections with index tabs. The symbols and nomenclature are presented in section 2.0, and a list of the primary mission guidelines and constraints is presented in section 3.0.

The main body of the document is section 4.0. This section is divided into subsections according to mission phases, and each subsection contains the text, tables, and figures related to the particular mission phase. In general, the information in each subsection includes a tracking timeline summary, a shadow timeline summary, target parameters, and a sequence of events table with a standard set of parameters. The propellant consumption predictions in section 4.0 are approximate and are included for completeness. The official predictions will be published in the operational consumables document.

The input data that were used to generate the operational trajectory are presented in section 5.0, which also contains the REFSMMAT's. Almost all of the input data are tabulated; however, for some data, only the source is identified. The references are listed in section 6.0.

The mission tracking timeline, shadow timeline, and more detailed trajectory data are contained in volume II of the operational trajectory document (ref. 1).

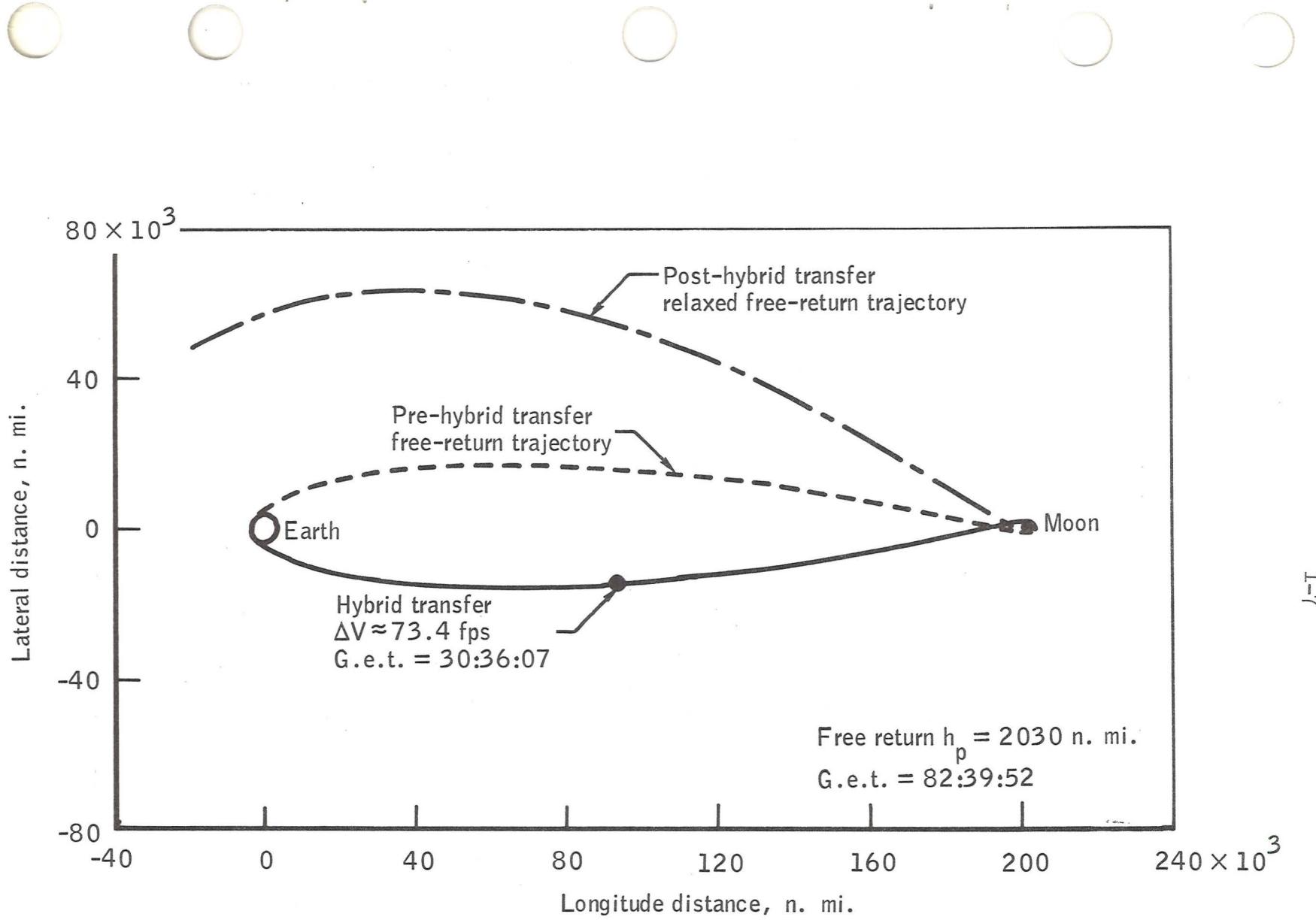


Figure 1.1-1.- Schematic of the hybrid mission profile.

TABLE 1.1-I.- SEQUENCE OF MAJOR EVENTS<sup>a</sup>

Event	Time, hr:min:sec, g.e.t	Time, hr:min:sec, c.s.t	Data summary	
Launch	00:00:00	Jan. 31, 1971 14:20:00	Flight azimuth, deg Launch complex	72 39A
EPO insertion	00:11:39.6	14:31:39.6	Geodetic latitude, deg Longitude, deg Geodetic altitude, n. mi. Velocity, fps	32.7 -53.3 103.3 25 567.7
Translunar injection Burn initiation	02:30:42.4	16:50:42.4	Geodetic latitude, deg Longitude, deg Velocity, fps Apogee altitude, n. mi. Geodetic altitude, n. mi.	-29.0 124.4 25 593.6 104.7 99.4
Burn termination	02:36:06.8	16:56:06.8	Geodetic latitude, deg Longitude, deg Burn duration, sec Plane change, deg Apogee altitude, n. mi. Geodetic altitude, n. mi.	-19.1 149.9 324.4 0.3 250 173 162.3
Post-TLI events CSM/S-IVB separation	02:51:06.8	17:11:06.8		
Docking	03:11:06.8	17:31:06.8		
CSM/LM ejection	03:56:06.8	18:16:06.8		
Evasive maneuver (performed by S-IVB)	04:19:06.8	18:39:06.8	ΔV, fps	≈10
Translunar coast, midcourse correction maneuvers MCC-1	TLI plus 9 hr	Feb. 1, 1971 01:56:06.8	Geodetic altitude, n. mi.	≈56 000
MCC-2 (hybrid transfer)	30:36:06.8	20:56:06.8	Geodetic altitude, n. mi. ΔV, fps Burn duration, sec SPS propellant used, lb Plane change, deg	118 344.5 73.4 11.1 705.7 0.2
MCC-3	LOI minus 22 hr	Feb. 3, 1971 02:58:13.6	Geodetic altitude, n. mi.	≈174 600
MCC-4	LOI minus 5 hr	19:58:13.6	Altitude above mean lunar radius, n. mi.	≈12 600
Lunar orbit insertion (LOI) Burn initiation	82:38:13.6	Feb. 4, 1971 00:58:13.6	Mass, lb Altitude above LLS, n. mi. Selenographic latitude, deg Selenographic longitude, deg Perilune altitude above LLS, n. mi. Selenographic inclination, deg Velocity, fps	96 377.7 76.4 2.2 -178.2 57.1 3.2 8094.4

<sup>a</sup>Data shown for launch vehicle events (e.g., launch, EPO insertion, TLI, etc.) are based on MSC simulations and should be considered unofficial proximate only. Official launch vehicle data are contained in the launch vehicle Operational Trajectory published by Marshall Space Flight Center.

TABLE 1.1-I.- SEQUENCE OF MAJOR EVENTS<sup>a</sup> - Continued

Event	Time, hr:min:sec, g.e.t.	Time, hr:min:sec, c.s.t.	Data summary	
Burn termination	82:44:20.2	Feb. 4, 1971 01:04:20.2	Altitude above LLS, n. mi. Selenographic latitude, deg Selenographic longitude, deg Selenographic inclination, deg Burn duration, sec Inertial burn arc, deg Plane change, deg $\Delta V$ , fps SPS propellant used, lb Velocity, fps Orbital period, hr:min:sec Perilune altitude above LLS, n. mi. Apolune altitude above LLS, n. mi.	57.7 -0.7 158.2 14.1 366.6 23.7 -11.2 2986.0 24 695.2 5490.6 02:08:29.3 57.1 170.0
S-IVB predicted lunar impact	TBD	TBD	Selenographic latitude, deg Selenographic longitude, deg	-1.7 -33.25
Descent orbit insertion (DOI) Burn initiation	86:56:57.6	05:16:57.6	Mass, lb Altitude above LLS, n. mi. Selenographic latitude, deg Selenographic longitude, deg Perilune altitude above LLS, n. mi. Apolune altitude above LLS, n. mi. Velocity, fps	71 642.6 57.3 2.4 168.6 57.1 170.0 5492.7
Burn termination	86:57:18.9	05:17:18.9	Altitude above LLS, n. mi. Selenographic latitude, deg Selenographic longitude, deg Selenographic inclination, deg Burn duration, sec Inertial burn arc, deg Plane change, deg $\Delta V$ , fps Velocity, fps SPS propellant used, lb Orbital period, hr:min:sec Perilune altitude above LLS, n. mi. Apolune altitude above LLS, n. mi.	57.2 2.2 167.5 14.1 21.4 1.1 0.0 206.6 5287.7 1452.2 01:54:12.0 9.8 58.4
First pass over Descartes	89:38:51.0	07:58:51.0		
Descent CSM/LM undock and SEP	104:27:30.7	22:47:30.7	Revolution number	12
Circularization (CSM) Burn initiation	105:46:48.5	Feb. 5, 1971 00:06:48.5	Mass, lb Selenographic latitude, deg Selenographic longitude, deg Selenographic inclination, deg Altitude above LLS, n. mi. Perilune altitude above LLS, n. mi. Velocity, fps Revolution number	35 937.7 9.7 -168.2 14.2 58.6 8.8 5280.1 12

<sup>a</sup>Data shown for launch vehicle events (e.g., launch, EPO insertion, TLI, etc.) are based on MSC simulations and should be considered unofficial and approximate only. Official launch vehicle data are contained in the launch vehicle Operational Trajectory published by Marshall Space Flight Center.

-1  
-9

TABLE 1.1-I.- SEQUENCE OF MAJOR EVENTS<sup>a</sup> - Continued

Event	Time, hr:min:sec, g.e.t.	Time, hr:min:sec, c.s.t.	Data summary	
Burn termination	105:46:52.3	00:06:52.3	ΔV, fps Burn duration, sec Altitude above LLS, n. mi. Perilune altitude above LLS, n. mi. Apolune altitude above LLS, n. mi. Velocity, fps SPS propellant consumed, lb Burn arc, deg Period, hr:min:sec	72.5 3.8 58.6 56.0 63.5 5351.0 257.2 0.2 01:58:45.2
PDI (DPS ignition time)	108:42:01.1	03:02:01.1	Altitude above LLS, ft Velocity, fps Revolution number	50 026.6 5574.5 14
High gate (P63 to P64)	108:50:35.1	03:10:35.1	Altitude above LLS, ft Velocity, fps	7496.3 464.0
Low gate	108:52:13.1	03:12:13.1	Altitude above LLS, ft Velocity, fps	682.8 84.5
Vertical descent (P64 to P65)	108:53:01.1	03:13:01.1	Altitude above LLS, ft Velocity, fps	125.1 6.7
LM landing	108:53:32.6	03:13:32.6	ΔV, fps Burn duration, min:sec DPS propellant consumed, lb Revolution number	6637.6 11:31.5 16 801.2 14
CSM first pass over LLS	108:54:18.2	03:14:18.2	Revolution number	14
CSM plane change Burn initiation	118:09:40.2	12:29:40.2	Mass, lb Selenographic latitude, deg Selenographic longitude, deg Altitude above LLS, n. mi. Perilune altitude above LLS, n. mi. Apolune altitude above LLS, n. mi. Velocity, fps Revolution number	35 585.0 -10.9 93.7 61.7 57.4 61.7 5333.8 19
Burn termination	118:09:58.5	12:29:58.5	SPS propellant consumed, lb ΔV, fps Burn duration, sec Selenographic latitude, deg Selenographic longitude, deg Altitude above LLS, n. mi. Perilune altitude above LLS, n. mi. Apolune altitude above LLS, n. mi. Plane change, deg Selenographic inclination, deg Velocity, fps Revolution number	1249.8 360.7 18.4 -11.0 92.7 61.7 57.4 61.7 3.9 12.1 5333.8 19

<sup>a</sup>Data shown for launch vehicle events (e.g., launch, EPO insertion, TLI, etc.) are based on MSC simulations and should be considered unofficial and approximate only. Official launch vehicle data are contained in the launch vehicle Operational Trajectory published by Marshall Space Flight Center.

TABLE 1.1-I.- SEQUENCE OF MAJOR EVENTS<sup>a</sup>- Continued

Event	Time, hr:min:sec, g.e.t.	Time, hr:min:sec, c.s.t.	Data summary	
Second pass over Descartes	134:21:30.6	Feb. 6, 1971 04:41:30.6	Revolution number	27
Ascent LM lift-off	142:24:29.2	12:44:29.2	Mass, lb Selenographic latitude, deg Selenographic longitude, deg Revolution number	10 747.0 -3.7 -17.5 31
LM insertion	142:31:39.9	12:51:39.9	Mass, lb $\Delta V$ , fps Burn duration, sec Latitude, deg Longitude, deg Altitude above LLS, ft Perilune altitude above LLS, ft Apolune altitude above LLS, ft	5831.5 6053.4 430.7 -1.6 -27.3 59 932.7 55 511.9 309 656.4
Rendezvous TPI	143:09:39.9	13:29:39.9	Burn duration, sec $\Delta V$ , fps Propellant used, lb Resultant $h_a/h_p$ , n. mi. Range at cutoff, n. mi. Range rate at cutoff, fps Propulsion system Revolution number	4.0 77.8 45.3 61.0/44.6 32.5 -134.3 APS 31
Braking	143:50:32.8	14:10:32.8	Burn duration, sec $\Delta V$ , fps Propellant used, lb Range at final braking, n. mi. Range rate at final braking, fps $h_a/h$ at final braking, n. mi. Propulsion system Revolution number	29.2 32.6 20.8 0.02 -0.23 60.3/58.7 LM RCS 32
Docking	144:10:00.0	14:30:00.0		
LM jettison	146:23:30.5	16:43:30.5	Selenographic latitude, deg Selenographic longitude, deg	-3.3 -21.1
CSM/LM separation Burn initiation	146:28:30.6	16:48:30.6		

<sup>a</sup>Data shown for launch vehicle events (e.g., launch, EPO insertion, TLI, etc.) are based on MSC simulations and should be considered unofficial and approximate only. Official launch vehicle data are contained in the launch vehicle Operational Trajectory published by Marshall Space Flight Center.

TABLE 1.1-I.- SEQUENCE OF MAJOR EVENTS<sup>a</sup>- Continued

Event	Time, hr:min:sec, g.e.t.	Time, hr:min:sec, c.s.t.	Data summary	
Burn termination	146:28:36.5	16:48:36.5	Mass, lb $\Delta V$ , fps Burn duration, sec Selenographic latitude, deg Selenographic longitude, deg Altitude above LLS, n. mi. Perilune altitude above LLS, n. mi. Apolune altitude above LLS, n. mi. Plane change, deg Selenographic inclination, deg Velocity, fps Revolution number	34 570.7 1.0 5.9 -0.1 -36.3 59.5 58.3 60.2 0.0 12.2 5344.7 33
LM deorbit	147:52:58.9	18:12:58.9	Mass, lb $\Delta V$ , fps Burn duration, sec Selenographic latitude, deg Selenographic longitude, deg	5277.2 183.7 77.0 -11.8 67.4
LM impact	148:20:58.1	18:40:58.1	Mass, lb Selenographic latitude, deg Selenographic longitude, deg Velocity, fps CSM revolution number	5171.1 -3.5 -19.3 5505.9 34
Transearth injection Burn initiation	149:14:50.1	19:34:50.1	Mass, lb Altitude above LLS, n. mi. Selenographic latitude, deg Selenographic longitude, deg Perilune altitude above LLS, n. mi. Selenographic inclination, deg Velocity, fps Revolution number	34 515.7 59.3 7.1 177.3 58.1 12.2 5346.4 35
Burn termination	149:17:17.3	19:37:17.3	Altitude above LLS, n. mi. Selenographic latitude, deg Selenographic longitude, deg Perilune altitude above LLS, n. mi. Selenographic inclination, deg Burn duration, sec Inertial burn arc, deg Plane change, deg $\Delta V$ , fps SPS propellant used, lb Velocity, fps	66.4 6.5 167.8 57.5 6.5 147.1 9.4 11.0 3449.5 9995.7 8503.6

<sup>a</sup>Data shown for launch vehicle events (e.g., launch, EPO insertion, TLI, etc.) are based on MSC simulations and should be considered unofficial and approximate only. Official launch vehicle data are contained in the launch vehicle Operational Trajectory published by Marshall Space Flight Center.

TABLE 1.1-I.- SEQUENCE OF MAJOR EVENTS<sup>a</sup> - Concluded

Events	Time, hr:min:sec, g.e.t.	Time, hr:min:sec, c.s.t.	Data summary
Transearth coast, midcourse correction maneuvers			
MCC-5	TEI plus 17 hr	Feb. 7, 1971 12:37:17.3	Geodetic altitude, n. mi. $\approx 175\ 500$
MCC-6	EI minus 22 hr	Feb. 8, 1971 16:46:58.7	Geodetic altitude, n. mi. $\approx 105\ 400$
MCC-7	EI minus 3 hr	Feb. 9, 1971 11:46:58.7	Geodetic altitude, n. mi. $\approx 24\ 800$
CM/SM sep	EI minus 15 min	14:31:58.7	Geodetic altitude, n. mi. 1961.9
Entry interface	216:26:58.7	14:46:58.7	Transearth coast time, hr 67.2 Inertial velocity, fps 36 170.1 Geodetic altitude, n. mi. (ft) 65.9(400 249) Inertial flight-path angle, deg -6.5 Geodetic latitude, deg -36.3 Longitude, deg 166.4 Equatorial inclination (ascending), deg 40
CM landing	216:40:53.1	15:00:53.1	Geodetic latitude, deg -27.241 Longitude, deg -171.53

<sup>a</sup>Data shown for launch vehicle events (e.g., launch, EPO insertion, TLI, etc.) are based on MSC simulations and should be considered unofficial and approximate only. Official launch vehicle data are contained in the launch vehicle Operational Trajectory published by Marshall Space Flight Center.

TABLE 1.1-II.- LUNAR ORBIT ACTIVITIES

Activity	Revolution number	Time, hr:min:, g.e.t.
LOI	Beginning of 1	82:38
DOI	Beginning of 3	86:57
Low-altitude surface observation	3	87:50
Descartes photograph	4	89:35
Begin rest period	5	91:00
End rest period	9	99:30
Begin IVT to LM	10	101:50
Undocking and separation	12	104:28
CSM circularization	12	105:47
PDI	14	108:42
Landing	14	108:54
Begin first EVA	16	113:30
Terminate first EVA	18	117:45
CSM plane change	19	118:10
Begin second EVA	27	134:12
Descartes photograph	27	134:22
Terminate second EVA	29	138:30
LM ascent	31	142:24

TABLE 1.1-II- LUNAR ORBIT ACTIVITES - Concluded

Activity	Revolution number	Time, hr:min:, g.e.t.
Docking	32	144:10
LM jettison	34	146:23
CSM separation	33	146:29
LM deorbit burn	34	147:53
LM lunar impact	34	148:21
TEI	Beginning of 35	149:17

TABLE 1.1-III.- LAUNCH WINDOW SUMMARY

Launch date . . . . .	January 31, 1971
Site . . . . .	Fra Mauro
Flight azimuth, deg <sup>a</sup> . . . . .	72 to 96
Launch time, hr:min, c.s.t. <sup>a</sup> . . . . .	14:20 to 18:10
Free-return $h_p$ , n. mi. . . . .	2030 to 1075
Hybrid transfer $\Delta V$ , fps . . . . .	73.4 to 45.0
Translunar flight time, hr:min . . . . .	80:02 to 74:50
Lunar orbit inclination, deg . . . . .	14.1
Approach azimuth at landing, deg . . . . .	-76.31
Sun elevation at landing, deg . . . . .	10.3
Goldstone landing coverage, hr:min . . . . .	1:30
Lunar surface stay time, hr . . . . .	34
Photo site . . . . .	Descartes
Total lunar orbit stay time . . . . .	67 hr, 34 revs
Transearth flight time, hr . . . . .	67
Total mission time, hr . . . . .	216:41 to 212:51

<sup>a</sup>The parameters for this phase are approximate and are presented for information only. The official source for this phase is the MSFC LV operational trajectory (ref. 2).

Event	Launch	EOI	TLI initiation	TLI cutoff	Evasive maneuver (S-IVB)	Hybrid transfer	LOI initiation	LOI cutoff	DOI	Descartes photo pass 1	CSM undock and separation	CSM circularization	PDI	LM landing	CSM first pass over LLS	CSM plane change 1	Descartes photo pass 2	LM lift-off	TPI	Braking	Docking	
Launch																						
EOI	0:11:40																					
TLI initiation	2:30:42	02:19:03																				
TLI cutoff	2:36:07	02:24:27	00:05:24																			
Evasive maneuver (S-IVB)	4:19:07	04:07:27	01:48:24	01:43:00																		
Hybrid transfer	30:36:07	30:24:27	28:05:24	28:00:00	26:17:00																	
LOI initiation	82:38:14	82:26:34	80:07:31	80:02:07	78:19:07	52:02:07																
LOI cutoff	82:44:20	82:32:41	80:13:38	80:08:13	78:25:13	52:08:13	00:06:07															
DOI	86:56:58	86:45:18	84:26:15	84:20:51	82:37:51	56:20:51	04:18:44	04:12:37		06:54:31	02:41:53											
Descartes photo pass 1	89:38:51	89:27:11	87:08:09	87:02:44	85:19:44	59:02:44	07:00:37															
CSM/LM undock and separation	104:27:31	104:15:51	101:56:48	101:51:24	100:08:24	73:51:24	21:49:17	21:43:11	17:30:33	14:48:40												
CSM circularization	105:46:48	105:35:09	103:16:06	103:10:42	101:27:42	75:10:42	23:08:35	23:02:28	18:49:51	16:07:57	01:19:18											
PDI	108:42:01	108:30:21	106:11:19	106:05:54	104:22:54	78:05:54	26:03:47	25:57:41	21:45:03	19:03:10	04:14:30	02:55:13										
LM landing	108:53:33	108:41:53	106:22:50	106:17:26	104:34:26	78:17:26	26:15:19	26:09:12	21:56:35	19:14:42	04:26:02	03:06:44	00:11:32									
CSM first pass over LLS	108:54:18	108:42:39	106:23:36	106:18:11	104:35:11	78:18:11	26:16:05	26:09:58	21:57:21	19:15:27	04:26:47	03:07:30	00:12:17	00:00:46								
CSM plane change 1	118:09:40	117:58:01	115:38:58	115:33:33	113:50:33	87:33:33	35:31:27	35:25:20	31:12:43	28:30:49	13:42:10	12:22:52	09:27:39	09:16:08	09:15:22							
Descartes photo pass 2	134:21:31	134:09:51	131:50:48	131:45:24	130:02:24	103:45:24	51:43:17	51:37:10	47:24:33	44:42:40	29:54:00	28:34:42	25:39:29	25:27:58	25:27:12	16:11:50						
LM lift-off	142:25:42	142:14:03	139:55:00	139:49:36	138:06:36	111:49:36	59:47:29	59:41:22	55:28:45	52:46:51	37:58:12	36:38:54	33:43:41	33:32:10	33:31:24	24:16:02	08:04:12					
TPI	143:09:40	142:58:00	140:38:57	140:33:33	138:50:33	112:33:33	60:31:26	60:25:20	56:12:42	53:30:49	38:42:09	37:22:51	34:27:39	34:16:07	34:15:22	25:00:00	08:48:09	00:43:57		00:40:53		
Braking	143:50:33	143:38:53	141:19:50	141:14:26	139:31:26	113:14:26	61:12:19	61:06:13	56:53:35	54:11:42	39:23:02	38:03:44	35:08:32	34:57:00	34:56:15	25:40:53	09:29:02	01:24:50				
Docking	144:10:00	143:58:20	141:39:18	141:33:53	139:50:53	113:33:53	61:31:46	61:25:40	57:13:02	54:31:09	39:42:29	38:23:11	35:27:59	35:16:27	35:15:42	26:00:20	09:48:29	01:44:18	01:00:20	00:19:27		
LM jettison	146:23:30	146:11:51	143:52:48	143:47:24	142:04:24	115:47:24	63:45:17	63:39:10	59:26:33	56:44:39	41:56:00	40:36:42	37:41:29	37:29:58	37:29:12	28:13:50	12:02:00	03:57:48	03:13:51	02:32:58	02:13:30	
CSM separation	146:28:31	146:16:51	143:57:48	143:52:24	142:09:24	115:52:24	63:50:17	63:44:10	59:31:33	56:49:40	42:01:00	40:41:42	37:46:29	37:34:58	37:34:12	28:18:50	12:07:00	04:02:48	03:18:51	02:37:58	02:18:31	
LM deorbit	147:52:59	147:41:19	145:22:16	145:16:52	143:33:52	117:16:52	65:14:45	65:08:39	60:56:01	58:14:08	43:25:28	42:06:10	39:10:58	38:59:26	38:58:41	29:43:19	13:31:28	05:27:16	04:43:19	04:02:26	03:42:59	
LM lunar impact	148:20:58	148:09:18	145:50:16	145:44:51	144:01:51	117:44:51	65:42:44	65:36:38	61:24:00	58:42:07	43:53:27	42:34:10	39:38:57	39:27:25	39:26:40	30:11:18	13:59:27	05:55:16	05:11:18	04:30:25	04:10:58	
TEI initiation	149:14:50	149:03:10	146:44:08	146:38:43	144:55:43	118:38:43	66:36:36	66:30:30	62:17:52	59:35:59	44:47:19	43:28:02	40:32:49	40:21:17	40:20:32	31:05:10	14:53:19	06:49:08	06:05:10	05:24:17	05:04:50	
TEI cutoff	149:17:17	149:05:38	146:46:35	146:41:10	144:58:10	118:41:10	66:39:04	66:32:57	62:20:20	59:38:26	44:49:47	43:30:29	40:35:16	40:23:45	40:22:59	31:07:37	14:55:47	06:51:35	06:07:37	05:26:44	05:07:17	
Entry interface	216:26:59	216:15:19	213:56:16	213:50:52	212:07:52	185:50:52	133:48:45	133:42:38	129:30:01	126:48:08	111:59:28	110:40:10	107:44:58	107:33:26	107:32:40	98:17:18	82:05:28	74:01:16	73:17:19	72:36:26	72:16:59	
Landing	216:40:59	216:29:19	214:10:17	214:04:52	212:21:52	186:04:52	134:02:45	133:56:39	129:44:01	127:02:08	112:13:28	110:54:11	107:58:58	107:47:26	107:46:41	98:31:19	82:19:28	74:15:17	73:31:19	72:50:26	72:30:59	

TABLE 1.1-IV.-- TIMETABLE OF MISSION EVENTS

cutoff	Evasive maneuver (S-IVB)	Hybrid transfer	LOI initiation	LOI cutoff	DOI	Descartes photo pass 1	CSM undock and separation	CSM circularization	PDI	LM landing	CSM first pass over LLs	CSM plane change 1	Descartes photo pass 2	LM lift-off	TPI	Braking	Docking	LM jettison	CSM separation	LM deorbit	LM lunar impact	TEI initiation	TEI cutoff	Entry interface
:43:00																								
:00:00	26:17:00																							
:02:07	78:19:07	52:02:07																						
:08:13	78:25:13	52:08:13	00:06:07																					
:20:51	82:37:51	56:20:51	04:18:44	04:12:37																				
:02:44	85:19:44	59:02:44	07:00:37	06:54:31	02:41:53																			
:51:24	100:08:24	73:51:24	21:49:17	21:43:11	17:30:33	14:48:40																		
:10:42	101:27:42	75:10:42	23:08:35	23:02:28	18:49:51	16:07:57	01:19:18																	
:05:54	104:22:54	78:05:54	26:03:47	25:57:41	21:45:03	19:03:10	04:14:30	02:55:13																
:17:26	104:34:26	78:17:26	26:15:19	26:09:12	21:56:35	19:14:42	04:26:02	03:06:44	00:11:32															
:18:11	104:35:11	78:18:11	26:16:05	26:09:58	21:57:21	19:15:27	04:26:47	03:07:30	00:12:17	00:00:46														
:33:33	113:50:33	87:33:33	35:31:27	35:25:20	31:12:43	28:30:49	13:42:10	12:22:52	09:27:39	09:16:08	09:15:22													
:45:24	130:02:24	103:45:24	51:43:17	51:37:10	47:24:33	44:42:40	29:54:00	28:34:42	25:39:29	25:27:58	25:27:12	16:11:50												
:49:36	138:06:36	111:49:36	59:47:29	59:41:22	55:28:45	52:46:51	37:58:12	36:38:54	33:43:41	33:32:10	33:31:24	24:16:02	08:04:12											
:33:33	138:50:33	112:33:33	60:31:26	60:25:20	56:12:42	53:30:49	38:42:09	37:22:51	34:27:39	34:16:07	34:15:22	25:00:00	08:48:09	00:43:57										
:14:26	139:31:26	113:14:26	61:12:19	61:06:13	56:53:35	54:11:42	39:23:02	38:03:44	35:08:32	34:57:00	34:56:15	25:40:53	09:29:02	01:24:50	00:40:53									
:33:53	139:50:53	113:33:53	61:31:46	61:25:40	57:13:02	54:31:09	39:42:29	38:23:11	35:27:59	35:16:27	35:15:42	26:00:20	09:48:29	01:44:18	01:00:20	00:19:27								
:47:24	142:04:24	115:47:24	63:45:17	63:39:10	59:26:33	56:44:39	41:56:00	40:36:42	37:41:29	37:29:58	37:29:12	28:13:50	12:02:00	03:57:48	03:13:51	02:32:58	02:13:30							
:52:24	142:09:24	115:52:24	63:50:17	63:44:10	59:31:33	56:49:40	42:01:00	40:41:42	37:46:29	37:34:58	37:34:12	28:18:50	12:07:00	04:02:48	03:18:51	02:37:58	02:18:31	00:05:00						
:16:52	143:33:52	117:16:52	65:14:45	65:08:39	60:56:01	58:14:08	43:25:28	42:06:10	39:10:58	38:59:26	38:58:41	29:43:19	13:31:28	05:27:16	04:43:19	04:02:26	03:42:59	01:29:28	01:24:28					
:44:51	144:01:51	117:44:51	65:42:44	65:36:38	61:24:00	58:42:07	43:53:27	42:34:10	39:38:57	39:27:25	39:26:40	30:11:18	13:59:27	05:55:16	05:11:18	04:30:25	04:10:58	01:57:28	01:52:28	00:27:59				
:38:43	144:55:43	118:38:43	66:36:36	66:30:30	62:17:52	59:35:59	44:47:19	43:28:02	40:32:49	40:21:17	40:20:32	31:05:10	14:53:19	06:49:08	06:05:10	05:24:17	05:04:50	02:51:20	02:46:20	01:21:51	00:53:52			
:41:10	144:58:10	118:41:10	66:39:04	66:32:57	62:20:20	59:38:26	44:49:47	43:30:29	40:35:16	40:23:45	40:22:59	31:07:37	14:55:47	06:51:35	06:07:37	05:26:44	05:07:17	02:53:47	02:48:47	01:24:18	00:56:19	00:02:27		
:50:52	212:07:52	185:50:52	133:48:45	133:42:38	129:30:01	126:48:08	111:59:28	110:40:10	107:44:58	107:33:26	107:32:40	98:17:18	82:05:28	74:01:16	73:17:19	72:36:26	72:16:59	70:03:28	69:58:28	68:34:00	68:06:01	67:12:09	67:09:41	
:04:52	212:21:52	186:04:52	134:02:45	133:56:39	129:44:01	127:02:08	112:13:28	110:54:11	107:47:26	107:46:41	98:31:19	82:19:28	74:15:17	73:31:19	72:50:26	72:30:59	70:17:29	70:12:29	68:48:00	68:20:01	67:26:09	67:23:42	00:14:00	

TABLE 1.1-V.- MISSION RADAR TIMELINE

(a) CSM acquisition and termination during rendezvous - 0° minimum elevation

VEHICLE 1 RADAR TABLE											31 JAN 71			EOI TO TLI IGNITION			LAZ 72.1					
TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA									
	HRS	MIN SEC	DAY	HRS	MIN SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN SEC	RA	DEC	AZ	ELV	X	Y	RANGE
BDA-02 X-Y USB-30	0	1 7	0	0 11	40	102	7	85	5	85	5	592	0	0 12	46	107	5	84	0	90	6	849
SHIP 1 INSERTION	0	4 20	0	0 11	40	-76	52	-38	13	-70	51	379	0	0 15	60	129	16	72	0	90	18	846
CYI-04 X-Y USB-30	0	7 6	0	0 16	34	-34	15	-73	0	-90	17	847	0	0 23	40	145	-20	113	0	90	-23	840
TAN-22 A-E FPS-16	0	5 33	0	0 37	5	84	63	-20	0	-90	70	824	0	0 42	38	-138	3	87	0	90	3	827
CRD-08 X-Y USB-30	0	5 50	0	0 52	17	98	-28	-121	0	-90	-31	833	0	0 58	7	-49	-32	126	0	90	-36	833
HSK-25 X-Y USB-85	0	6 2	0	0 59	35	156	10	-78	0	-90	-78	832	0	1 5	37	-62	37	42	0	-90	42	826
GYM-14 X-Y USB-30	0	7 3	0	1 28	19	-89	-25	-118	0	-90	-28	839	0	1 35	22	85	13	76	0	90	14	846
GDS-28 X-Y USB-85	0	4 41	0	1 29	0	-47	-51	-163	0	90	-17	840	0	1 33	41	58	-19	114	0	90	66	847
TEX-16 X-Y USB-30	0	7 3	0	1 31	12	-87	-4	-95	0	-90	-5	844	0	1 38	15	104	21	67	0	90	23	850
MIL-71 X-Y USB-30	0	6 53	0	1 35	5	-77	10	-79	0	-90	11	848	0	1 41	58	120	18	70	0	90	20	847

60

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
GBM-41 X-Y USB-30	0	6	30	0	1	35	49	-79	19	-69	0	-90	21	847	0	1	42	19	126	25	62	0	90	28	848		
BDA-02 X-Y USB-30	0	7	9	0	1	38	31	-57	4	-85	0	-90	5	850	0	1	45	40	124	-4	95	0	90	-5	846		
GTI-51 A-E TPQ-18	0	4	17	0	1	38	37	-88	49	-36	0	-90	54	849	0	1	42	54	145	47	38	0	90	52	849		
SHIP 1 INSERTION	0	7	5	0	1	42	12	-49	19	-68	0	-90	22	850	0	1	49	17	141	-6	96	0	90	-6	844		
CYI-04 X-Y USB-30	0	5	18	0	1	50	7	2	-10	-101	0	-90	-11	843	0	1	55	25	125	-57	161	0	90	-71	834		
PRE-76 A-E MPS-25	0	3	6	0	2	8	2	151	63	7	0	90	83	824	0	2	11	8	-148	27	60	0	90	30	826		
TAN-22 A-E FPS-16	0	6	50	0	2	9	11	70	15	-74	0	-90	16	827	0	2	16	2	-99	-35	127	0	90	-37	826		
CRO-08 X-Y USB-30	0	5	18	0	2	25	25	118	-33	-127	0	-90	-37	833	0	2	30	42	-35	-24	115	4	86	-25	639		
				31 JAN 71	TLI TO NFR IGNITION												LAZ	72.1									1-20
HSK-25 X-Y USB-85	0	0	51	0	2	36	7	-91	54	3	1	-89	3	1021	0	2	36	58	-72	52	15	0	-90	15	1197		
HAW-12 X-Y USB-30	12	47	58	0	2	42	34	-114	-40	-133	0	-90	-43	2344	0	15	30	32	46	30	-58	0	-90	32	75532		



## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
GWM-24 X-Y USB-30	0	4	16	0	2	44	23	-12	-15	105	0	90	-15	2748	0	2	48	39	-8	-1	91	0	90	-1	3692		
STN-00 A-E USB150	10	53	30	0	2	49	23	-79	-19	-114	0	-90	-24	3850	0	13	42	53	44	29	-52	0	-90	38	68583		
GDS-28 X-Y USB-85	10	21	12	0	2	50	12	-78	-14	-108	0	90	-72	4030	0	13	11	24	43	29	-53	0	-90	-53	66459		
GYM-14 X-Y USB-30	9	26	15	0	2	51	8	-79	-7	-97	0	-90	-7	4232	0	12	17	23	42	30	-56	0	-90	34	62706		
TEX-16 X-Y USB-30	8	22	12	0	2	54	48	-69	1	-89	0	-90	1	5011	0	11	16	60	40	30	-56	0	-90	34	58334		
MIL-71 X-Y USB-30	7	2	57	0	3	0	59	-54	8	-81	0	-90	9	6273	0	10	3	56	38	30	-55	0	-90	35	52754		
GBM-41 X-Y USB-30	6	43	27	0	3	2	18	-52	10	-79	0	-90	11	6534	0	9	45	46	37	30	-56	0	-90	34	51310		
GTI-51 A-E TPQ-18	2	2	38	0	3	7	5	-45	16	-73	0	-90	17	7457	0	5	9	43	16	32	-64	31	-56	22	23400		
BDA-02 X-Y USB-30	5	53	60	0	3	8	58	-40	13	-75	0	-90	15	7815	0	9	2	58	36	30	-54	0	-90	36	47811		
ANG-91 X-Y USB-30	4	35	33	0	3	15	50	-34	22	-67	0	-90	23	9069	0	7	51	23	31	31	-58	0	-90	32	41594		

1-21

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HFS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
SHIP 1 INSERTION	1	38	20	0	3	24	20	-24	20	-67	0	-90	23	10553	0	5	2	40	14	30	-64	15	-73	25	23392
GWM-24 X-Y USB-30	14	12	24	0	4	48	35	31	31	58	0	90	32	22557	0	19	0	58	48	30	-59	0	-90	31	87969
HSK-25 X-Y USB-85	8	31	27	0	7	47	2	40	36	44	0	-90	44	41190	0	16	18	29	47	33	-49	0	-90	-49	78480
CRO-08 X-Y USB-30	9	56	36	0	9	33	29	45	34	52	0	90	38	50310	0	19	30	5	49	31	-55	0	-90	35	89581
MAD-23 X-Y USB-85	15	57	59	0	14	38	7	50	29	50	0	-90	50	72208	1	6	36	7	55	28	-52	0	-90	-52	121732
CYI-04 X-Y USB-30	14	21	37	0	16	14	29	51	29	56	0	90	34	78248	1	6	36	7	54	28	-58	1	-89	32	121699
ACN-75 X-Y USB-30	11	24	10	0	17	42	55	53	31	59	0	90	31	83509	1	5	7	5	54	30	-60	0	-90	30	117888
BDA-02 X-Y USB-30	11	12	2	0	19	24	5	53	29	55	0	90	35	89258	1	6	36	7	55	29	-78	42	-47	9	119460
ANG-91 X-Y USB-30	10	38	46	0	19	57	20	54	30	59	0	90	31	91084	1	6	36	7	54	29	-66	35	-53	20	119803
GBM-41 X-Y USB-30	9	57	28	0	20	38	39	54	29	57	0	90	33	93318	1	6	36	7	55	29	-76	52	-37	9	119039

1-22

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
MIL-71 X-Y USB-30	9	53	4	0	20	43	3	54	29	56	0	90	34	93554	1	6	36	7	55	29	-79	55	-34	6	118949		
TEX-16 X-Y USB-30	8	41	22	0	21	54	45	54	29	57	0	90	33	97323	1	6	36	7	55	29	-80	70	-20	3	118551		
GYM-14 X-Y USB-30	7	47	24	0	22	48	43	55	29	57	0	90	33	100086	1	6	36	7	56	29	-78	82	-8	2	118379		
GDS-28 X-Y USB-85	7	46	42	0	22	49	25	55	29	54	0	-90	54	100123	1	6	36	7	56	29	-158	83	6	-3	118367		
STN-00 A-E USB150	7	32	40	0	23	3	26	55	29	53	0	90	37	100830	1	6	36	7	56	29	165	81	2	-8	118382		
HAW-12 X-Y USB-30	4	9	38	1	2	26	28	56	29	59	0	90	31	110641	1	6	36	7	57	29	70	52	36	12	119039		
GWM-24 X-Y USB-30	0	0	32	1	6	35	34	58	29	60	0	90	30	121717	1	6	36	7	58	29	60	0	90	30	121733		
	31 JAN 71			NFR TO LOI IGNITION												LAZ 72.1											
MAD-23 X-Y USB-85	0	1	11	1	6	36	18	55	28	-52	0	-90	-52	121742	1	6	37	29	55	28	-52	0	-90	-52	121801		
CYI-04 X-Y USB-30	0	3	41	1	6	36	18	54	28	-58	1	-89	32	121709	1	6	39	59	54	28	-57	0	-90	33	121904		
ANG-91 X-Y USB-30	2	42	38	1	6	36	18	54	29	-66	34	-53	20	119813	1	9	18	55	55	29	-60	0	-90	30	128369		

11-23

## VEHICLE 1 RADAR TABLE

## TRACKING TIME

## STATION ACQUISITION DATA

## STATION TERMINATION DATA

HRS MIN SEC

DAY HRS MIN SEC

RA DEC

AZ ELV

X Y RANGE

DAY HRS MIN SEC

RA DEC

AZ ELV

X Y RANGE

BDA-02 X-Y USB-30	3 37 1	1 6 36 18	55 29 -78 42 -47	9 119469	1 10 13 18	56 28 -56 0 -90 34	130508
GBM-41 X-Y USB-30	4 14 44	1 6 36 18	55 29 -76 52 -37	9 119048	1 10 51 2	56 28 -58 0 -90 32	131966
MIL-71 X-Y USB-30	4 30 16	1 6 36 18	55 29 -79 55 -34	6 118958	1 11 6 34	56 28 -57 0 -90 33	132564
TEX-16 X-Y USB-30	5 35 25	1 6 36 18	55 29 -80 70 -20	3 118559	1 12 11 43	56 28 -58 0 -90 32	135033
GYM-14 X-Y USB-30	6 30 24	1 6 36 18	56 29 -78 82 -8	2 118387	1 13 6 42	56 28 -58 0 -90 32	137079
GDS-28 X-Y USB-85	7 18 27	1 6 36 18	56 29 -157 83 6 -3	118375	1 13 54 45	57 26 -55 0 -90 -55	138838
STN-00 A-E USB150	7 47 16	1 6 36 18	56 29 165 81 2 -8	118390	1 14 23 33	57 26 -54 0 -90 36	139880
HAW-12 X-Y USB-30	9 32 41	1 6 36 18	57 29 70 53 36 12	119045	1 16 8 59	57 28 -59 0 -90 31	143618
GWM-24 X-Y USB-30	12 56 15	1 6 36 18	58 29 60 0 90 30	121738	1 19 32 33	58 28 -61 0 -90 29	150517
HSK-25 X-Y USB-85	8 43 34	1 8 27 9	58 30 52 0 -90 52	126291	1 17 10 43	58 29 -53 0 -90 -53	145743

1-24

12  
11  
10  
9  
8  
7  
6  
5  
4  
3  
2

## VEHICLE 1 RADAR TABLE

## TRACKING TIME

## STATION ACQUISITION DATA

## STATION TERMINATION DATA

	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
--	-----	-----	-----	-----	-----	-----	-----	----	-----	----	-----	---	---	-------	-----	-----	-----	-----	----	-----	----	-----	---	---	-------

CRD-08 X-Y USB-30	9	54	60	1	10	13	47	59	30	57	0	90	33	130517	1	20	8	47	58	29	-58	0	-90	32	151696
----------------------	---	----	----	---	----	----	----	----	----	----	---	----	----	--------	---	----	---	----	----	----	-----	---	-----	----	--------

MAD-23 X-Y USB-85	15	29	48	1	15	20	15	59	28	52	0	-90	52	141905	2	6	50	3	60	27	-53	0	-90	-53	170857
----------------------	----	----	----	---	----	----	----	----	----	----	---	-----	----	--------	---	---	----	---	----	----	-----	---	-----	-----	--------

CYI-04 X-Y USB-30	14	5	27	1	16	49	37	60	28	58	0	90	32	145028	2	6	55	3	60	27	-59	0	-90	31	170992
----------------------	----	---	----	---	----	----	----	----	----	----	---	----	----	--------	---	---	----	---	----	----	-----	---	-----	----	--------

ACN-75 X-Y USB-30	11	20	19	1	18	8	12	60	29	61	0	90	29	147703	2	5	28	31	60	28	-62	0	-90	28	168594
----------------------	----	----	----	---	----	---	----	----	----	----	---	----	----	--------	---	---	----	----	----	----	-----	---	-----	----	--------

SDA-02 X-Y USB-30	14	30	55	1	19	54	27	60	28	57	0	90	33	151237	2	10	25	22	61	27	-58	0	-90	32	176589
----------------------	----	----	----	---	----	----	----	----	----	----	---	----	----	--------	---	----	----	----	----	----	-----	---	-----	----	--------

ANG-91 X-Y USB-30	13	9	52	1	20	23	59	61	28	61	0	90	29	152197	2	9	33	51	60	27	-61	0	-90	29	175245
----------------------	----	---	----	---	----	----	----	----	----	----	---	----	----	--------	---	---	----	----	----	----	-----	---	-----	----	--------

GBM-41 X-Y USB-30	13	57	43	1	21	6	6	61	28	59	0	90	31	153558	2	11	3	50	61	27	-59	0	-90	31	177580
----------------------	----	----	----	---	----	---	---	----	----	----	---	----	----	--------	---	----	---	----	----	----	-----	---	-----	----	--------

MIL-71 X-Y USB-30	14	7	54	1	21	10	47	61	28	58	0	90	32	153708	2	11	18	41	61	27	-59	0	-90	31	177960
----------------------	----	---	----	---	----	----	----	----	----	----	---	----	----	--------	---	----	----	----	----	----	-----	---	-----	----	--------

TEX-16 X-Y USB-30	14	2	34	1	22	20	49	61	28	58	0	90	32	155926	2	12	23	23	61	27	-59	0	-90	31	179599
----------------------	----	---	----	---	----	----	----	----	----	----	---	----	----	--------	---	----	----	----	----	----	-----	---	-----	----	--------

GYM-14 X-Y USB-30	14	4	4	1	23	13	47	61	28	58	0	90	32	157578	2	13	17	51	61	27	-59	0	-90	31	180960
----------------------	----	---	---	---	----	----	----	----	----	----	---	----	----	--------	---	----	----	----	----	----	-----	---	-----	----	--------

1

2

3

4

5

6

7

8

9

10

11

12

## VEHICLE 1 RADAR TABLE

	TRACKING TIME				STATION ACQUISITION DATA										STATION TERMINATION DATA									
	HRS	MIN	SEC		DAY	HR	S	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HR	S	RA	DEC	AZ	ELV	X	Y	RANGE
GDS-28 X-Y USB-85	14	48	32		1	23	15	60	61	27	56	0	-90	56 157647	2	14	4	31	61	27	-56	0	-90	-56 182112
STN-00 A-E USB150	15	2	36		1	23	30	10	61	27	55	0	90	35 158085	2	14	32	45	61	27	-55	0	-90	35 182802
HAW-12 X-Y USB-30	13	32	27		2	2	46	59	62	28	60	0	90	30 163983	2	16	19	26	61	27	-61	0	-90	29 185363
GWM-24 X-Y USB-30	12	51	1		2	6	51	57	62	28	62	0	90	28 170905	2	19	42	58	62	27	-62	0	-90	28 190131
HSK-25 X-Y USB-85	8	54	33		2	8	34	14	62	29	54	0	-90	54 173659	2	17	28	48	62	28	-55	0	-90	-55 186986
CRO-08 X-Y USB-30	10	0	59		2	10	22	38	63	28	59	0	90	31 176510	2	20	23	37	62	28	-59	0	-90	31 191065
MAD-23 X-Y USB-85	15	16	6		2	15	35	55	63	27	54	0	-90	54 184329	3	6	52	1	63	26	-55	0	-90	-55 204362
CYI-04 X-Y USB-30	13	55	58		2	17	2	35	64	27	59	0	90	31 186383	3	6	58	33	63	26	-60	0	-90	30 204474
ACN-75 X-Y USB-30	11	18	43		2	18	17	32	64	27	62	0	90	28 188250	3	5	36	15	62	27	-63	0	-90	27 202752
BDA-02 X-Y USB-30	14	15	60		2	20	7	24	64	27	58	0	90	32 190719	3	10	23	23	62	26	-59	0	-90	31 208596

1  
1  
26

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
ANG-91 X-Y USB-30	13	1	24	2	20	34	17	64	27	62	0	90	28	191245	3	9	35	41	62	26	-63	0	-90	27	207720
GBM-41 X-Y USB-30	13	7	24	2	21	17	21	64	27	60	0	90	30	192217	3	10	24	45	62	26	-65	8	-82	25	208163
MIL-71 X-Y USB-30	13	1	59	2	21	22	47	64	27	59	0	90	31	192394	3	10	24	46	62	26	-66	10	-79	24	207999
TEX-16 X-Y USB-30	11	53	9	2	22	31	35	64	27	60	0	90	30	193853	3	10	24	45	62	26	-72	24	-65	17	207228
GYM-14 X-Y USB-30	10	59	54	2	23	24	49	64	27	60	0	90	30	195041	3	10	24	43	62	26	-76	35	-54	11	206630
GDS-28 X-Y USB-85	10	57	5	2	23	27	37	64	26	57	0	-90	57	195090	3	10	24	42	62	26	-85	42	-6	-48	206330
STN-00 A-E USB150	10	42	34	2	23	42	6	64	26	56	0	90	34	195405	3	10	24	40	62	26	-90	46	-44	0	206141
HAW-12 X-Y USB-30	7	28	41	3	2	55	52	64	26	61	0	90	29	199492	3	10	24	33	63	26	-67	78	-11	5	205263
GWM-24 X-Y USB-30	3	26	26	3	6	57	54	65	26	63	0	90	27	204460	3	10	24	20	64	26	65	47	41	17	206107
HSK-25 X-Y USB-85	1	51	54	3	8	32	27	64	27	56	0	-90	56	206394	3	10	24	21	64	27	36	17	-69	35	207589

1227

## VEHICLE 1 RADAR TABLE

## TRACKING TIME

## STATION ACQUISITION DATA

## STATION TERMINATION DATA

HRS MIN SEC

DAY HRS MIN SEC

RA DEC

AZ ELV

X

Y

RANGE

DAY HRS MIN SEC

RA DEC

AZ ELV

X

Y

RANGE

CRD-08  
X-Y USB-30

0 4 12

3 10 20 4 64 27 60 0 90 30 208549

3 10 24 16 64 27 60 1 89 30 208542

31 JAN 71 LOI TO DOI IGNITION

LAZ 72.1

CRD-08  
X-Y USB-30

1 24 20

3 10 56 52 64 27 56 7 81 34 207538

3 12 21 12 65 27 44 21 61 42 206942

HSK-25  
X-Y USB-85

1 24 24

3 10 56 57 63 27 30 21 -66 27 206753

3 12 21 20 64 27 11 26 -63 10 206664

GWM-24  
X-Y USB-30

1 24 24

3 10 56 60 63 26 63 54 33 15 205203

3 12 21 24 64 27 45 70 14 14 204971

HAW-12  
X-Y USB-30

1 24 28

3 10 57 15 62 26 -73 70 -19 5 204747

3 12 21 43 63 27 -75 52 -37 9 204400

STN-00  
A-E USB150

1 24 27

3 10 57 24 62 26 -85 39 -51 4 205787

3 12 21 52 63 26 -74 24 -65 14 206823

GDS-28  
X-Y USB-85

1 24 28

3 10 57 24 62 26 -81 35 -13 -54 206002

3 12 21 52 63 26 -71 19 -43 -63 207086

GYM-14  
X-Y USB-30

1 24 27

3 10 57 26 62 26 -74 28 -61 14 206347

3 12 21 53 63 26 -66 12 -77 23 207524

GBM-41  
X-Y USB-30

0 3 48

3 10 57 26 62 26 -61 1 -89 29 207914

3 11 1 14 62 26 -61 0 -90 29 207772

MIL-71  
X-Y USB-30

0 19 11

3 10 57 27 62 26 -62 4 -86 28 207745

3 11 16 39 62 26 -60 0 -90 30 207026

1-28

## VEHICLE 1 RADAR TABLE

## TRACKING TIME

## STATION ACQUISITION DATA

## STATION TERMINATION DATA

HRS MIN SEC

DAY HRS MIN SEC RA DEC AZ ELV X Y RANGE

DAY HRS MIN SEC RA DEC AZ ELV X Y RANGE

TEX-16 X-Y USB-30	1 24 25	3 10 57 27	62 26 -69 17 -72 20	206973	3 12 21 52	63 26 -61 1 -89 29	208163
CRO-08 X-Y USB-30	1 24 29	3 13 5 1	65 27 36 28 48 46	206592	3 14 29 29	66 28 17 35 22 51	206411
HSK-25 X-Y USB-85	1 24 30	3 13 5 7	64 27 0 27 -63 0	206609	3 14 29 36	65 28 -20 24 -65 -18	207006
GWM-24 X-Y USB-30	1 24 29	3 13 5 11	64 27 16 76 4 13	204864	3 14 29 40	65 27 -45 70 -15 14	205208
HAW-12 X-Y USB-30	1 24 28	3 13 5 23	63 26 -74 42 -46 12	205859	3 14 29 51	64 27 -70 25 -64 18	206979
GYM-14 X-Y USB-30	0 14 51	3 13 5 28	63 26 -62 3 -87 28	208002	3 13 20 19	63 26 -60 0 -90 30	207428
STN-00 A-E USB150	1 24 24	3 13 5 28	63 26 -69 16 -73 20	207250	3 14 29 52	65 26 -58 2 -88 32	208329
GDS-28 X-Y USB-85	1 5 26	3 13 5 29	63 26 -66 11 -65 -64	207527	3 14 10 55	64 26 -57 0 -90 -57	207470
31 JAN 71 DCI TO LM SEP IGNITION							
CRO-08 X-Y USB-30	1 5 6	3 15 15 7	65 27 4 38 6 52	206200	3 16 20 13	66 28 -13 36 -17 52	206361
HSK-25 X-Y USB-85	1 5 5	3 15 15 15	65 27 -30 20 -67 -28	207096	3 16 20 20	66 28 -42 13 -73 -41	207638

LAZ 72.1

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HR	S	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	
GWM-24 X-Y USB-30	1	5	6		3	15	15	20	65	27	-58	61	-25	15	205301		3	16	20	26	66	27	-64	48	-39	17	205880
HAW-12 X-Y USB-30	1	5	2		3	15	15	24	54	26	-67	15	-74	22	207397		3	16	20	31	66	27	-62	24	-67	28	208277
MAD-23 X-Y USB-85	0	31	31		3	15	48	37	57	26	55	0	-90	55	207138		3	16	20	8	67	26	59	5	-81	59	206111
MAD-23 X-Y USB-85	1	4	56		3	17	8	41	67	26	66	13	-60	63	207730		3	18	13	37	68	27	75	24	-31	62	207164
CRO-08 X-Y USB-30	1	4	53		3	17	8	60	66	28	-25	33	-34	49	206619		3	18	13	52	67	28	-38	25	-53	45	207101
HSK-25 X-Y USB-85	0	36	20		3	17	9	8	66	27	-50	6	-81	-50	208130		3	17	45	28	66	28	-55	0	-90	-55	207335
GWM-24 X-Y USB-30	1	4	48		3	17	9	15	66	27	-66	37	-51	19	206424		3	18	14	3	67	27	-66	23	-65	22	207235
CYI-04 X-Y USB-30	0	57	53		3	17	15	42	67	26	60	0	-90	30	208134		3	18	13	35	69	27	65	11	78	24	207923
ACN-75 X-Y USB-30	1	4	43		3	19	2	37	69	27	61	7	82	28	208265		3	20	7	20	70	27	57	20	67	31	207583
CYI-04 X-Y USB-30	1	4	42		3	19	2	44	69	27	70	21	68	19	207450		3	20	7	26	70	27	75	34	55	13	206842

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
MAD-23 X-Y USB-85	1	4	48	3	19	2	47	68	26	82	33	-12	56	206787	3	20	7	34	69	27	91	45	1	45	206331
CRO-08 X-Y USB-30	1	4	55	3	19	3	5	67	28	-47	18	-66	41	207598	3	20	7	60	68	28	-55	7	-82	35	208354
GWM-24 X-Y USR-30	0	59	42	3	19	3	14	67	27	-65	12	-77	24	207923	3	20	2	56	68	27	-62	0	-90	28	208493
ACN-75 X-Y USB-30	1	5	2	3	20	56	20	70	27	52	30	54	32	207157	3	22	1	22	71	28	42	41	38	34	206700
ANG-91 X-Y USB-30	1	5	1	3	20	56	20	70	27	62	1	89	28	208824	3	22	1	22	71	27	65	14	75	24	208125
BDA-02 X-Y USB-30	1	5	1	3	20	56	23	70	27	62	5	84	28	208530	3	22	1	24	71	27	69	17	72	20	207921
CYI-04 X-Y USB-30	1	5	1	3	20	56	23	70	27	78	44	45	8	206462	3	22	1	24	71	27	82	58	32	4	206044
MAD-23 X-Y USB-85	1	5	3	3	20	56	28	69	27	100	54	7	35	206073	3	22	1	31	70	27	115	66	11	22	205829
GRM-41 X-Y USB-30	0	21	50	3	21	39	34	71	27	60	0	90	30	207866	3	22	1	24	71	27	62	4	85	28	208695
MIL-71 X-Y USB-30	0	16	37	3	21	44	46	71	27	59	0	90	31	208062	3	22	1	23	71	27	61	3	87	29	208755

1  
31

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
GBM-41 X-Y USB-30	1	4	57	3	22	49	56	71	27	67	14	75	23	208231	3	23	54	53	72	27	71	27	62	17	207561
ANG-01 X-Y USB-30	1	4	54	3	22	49	58	71	27	67	24	64	20	207635	3	23	54	52	72	27	68	38	50	17	206992
MIL-71 X-Y USB-30	1	4	57	3	22	49	58	71	27	66	13	76	23	208302	3	23	54	54	72	27	71	25	64	17	207641
BDA-02 X-Y USB-30	1	4	55	3	22	50	0	71	27	74	27	62	14	207493	3	23	54	55	72	27	80	40	50	8	206915
ACN-75 X-Y USB-30	1	4	57	3	22	50	2	70	28	31	48	25	35	206495	3	23	54	59	71	28	11	53	8	36	206351
CYI-04 X-Y USB-30	1	4	52	3	22	50	10	70	27	86	69	21	1	205857	3	23	55	2	71	27	91	82	8	0	205719
MAD-23 X-Y USB-85	1	4	53	3	22	50	16	70	27	139	73	13	11	205765	3	23	55	9	71	27	-169	77	13	-3	205786
TEX-16 X-Y USB-30	1	0	14	3	22	54	40	71	27	60	0	90	30	208812	3	23	54	54	72	27	65	11	78	24	208439
GYM-14 X-Y USB-30	0	3	57	3	23	50	56	72	27	59	0	90	31	208883	3	23	54	53	72	27	60	1	89	30	209055
GDS-28 X-Y USB-85	0	1	8	3	23	53	50	72	27	56	0	-90	56	209045	3	23	54	58	72	27	57	0	-90	57	209093

1  
32

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
TEX-16 X-Y USB-30	1	4	41	4	0	44	3	72	27	70	21	68	19	207973	4	1	48	45	73	27	74	34	55	13	207359
GYM-14 X-Y USB-30	1	4	38	4	0	44	4	72	27	65	10	79	25	208596	4	1	48	42	73	27	70	23	66	18	207946
GDS-28 X-Y USB-85	1	4	44	4	0	44	5	72	27	63	9	-71	62	208685	4	1	48	49	73	27	71	20	-42	62	208090
MIL-71 X-Y USB-30	1	4	43	4	0	44	6	72	27	76	36	54	12	207216	4	1	48	49	73	27	80	49	41	7	206708
GBM-41 X-Y USB-30	1	4	42	4	0	44	6	72	27	75	37	52	12	207137	4	1	48	48	73	27	78	51	39	7	206637
ANG-91 X-Y USB-30	1	4	42	4	0	44	7	72	27	68	49	39	14	206627	4	1	48	48	73	28	63	62	25	12	206254
BDA-02 X-Y USB-30	1	4	45	4	0	44	7	72	27	85	50	40	3	206587	4	1	48	52	73	27	93	63	27	-1	206236
STN-00 A-E USB150	1	4	41	4	0	44	8	72	27	61	6	83	29	208850	4	1	48	49	73	27	69	17	72	20	208277
ACN-75 X-Y USB-30	1	4	51	4	0	44	14	71	28	-8	54	-6	36	206435	4	1	49	6	72	28	-28	49	-22	35	206708
CYI-04 X-Y USR-30	1	4	53	4	0	44	21	71	27	-98	87	-3	0	205787	4	1	49	14	72	27	-87	73	-17	1	206032

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
MAD-23 X-Y USB-85	1	4	51	4	0	44	23	71	27	-132	71	13	-14	205955	4	1	49	14	72	27	-108	61	10	-27	206296
GYM-14 X-Y USB-30	1	5	3	4	2	37	42	73	27	74	33	56	13	207521	4	3	42	45	74	27	78	48	43	8	206992
GDS-28 X-Y USB-85	1	5	0	4	2	37	43	73	27	76	30	-22	57	207686	4	3	42	43	74	27	83	42	-7	47	207164
STN-CO A-E USB150	1	5	1	4	2	37	44	73	27	75	26	63	13	207878	4	3	42	44	74	27	82	38	52	6	207349
TEX-16 X-Y USB-30	1	5	0	4	2	37	45	73	27	78	45	45	8	206986	4	3	42	45	74	28	82	58	32	4	206567
GRM-41 X-Y USB-30	1	5	3	4	2	37	46	73	27	81	62	28	4	206389	4	3	42	49	74	28	83	75	15	2	206170
MIL-71 X-Y USB-30	1	5	2	4	2	37	47	73	27	84	60	30	3	206444	4	3	42	49	74	28	89	73	17	0	206204
ANG-91 X-Y USB-30	1	5	5	4	2	37	48	72	28	51	72	14	11	206135	4	3	42	53	73	28	0	79	0	11	206118
BDA-02 X-Y USB-30	1	5	1	4	2	37	50	72	27	103	73	16	-4	206115	4	3	42	51	73	28	152	85	3	-5	206073
ACN-75 X-Y USB-30	1	5	4	4	2	37	58	72	28	-40	42	-36	34	207084	4	3	43	3	73	28	-50	31	-52	33	207705

134

## VEHICLE 1 RADAR TABLE

## TRACKING TIME

## STATION ACQUISITION DATA

## STATION TERMINATION DATA

HRS MIN SEC

DAY HRS MIN SEC RA DEC AZ ELV X Y RANGE

DAY HRS MIN SEC RA DEC AZ ELV X Y RANGE

CYI-04 X-Y USB-30	1 5 2	4 2 38 2	72 27 -83 62 -28 3	206360	4 3 43 4	73 27 -79 49 -41 7	206912
MAD-23 X-Y USB-85	1 5 3	4 2 38 4	72 27 -97 52 5 -38	206677	4 3 43 7	73 27 -87 41 -4 -49	207255
HAW-12 X-Y USB-30	0 13 18	4 3 29 23	74 27 60 0 90 30	208748	4 3 42 41	74 27 62 3 87 28	209312
HAW-12 X-Y USB-30	1 4 54	4 4 31 19	74 27 66 13 76 24	208832	4 5 36 13	75 27 69 26 63 19	208135
GDS-28 X-Y USB-85	1 4 54	4 4 31 27	74 27 90 52 0 38	206873	4 5 36 21	75 27 100 65 5 25	206538
GYM-14 X-Y USB-30	1 4 53	4 4 31 29	74 27 82 57 33 4	206707	4 5 36 22	75 28 86 71 19 1	206409
STN-00 A-E USB150	1 4 53	4 4 31 29	74 27 89 48 42 0	207035	4 5 36 22	75 27 100 60 29 -5	206666
TEX-16 X-Y USB-30	1 4 57	4 4 31 31	74 27 85 69 21 2	206386	4 5 36 28	75 28 88 83 7 0	206250
GBM-41 X-Y USB-30	1 4 52	4 4 31 40	73 27 77 86 4 1	206161	4 5 36 32	74 28 -81 80 -10 1	206276
MIL-71 X-Y USB-30	1 4 51	4 4 31 41	73 27 99 84 6 -1	206171	4 5 36 32	74 28 -94 82 -8 -1	206255

1-35

## VEHICLE 1 RADAR TABLE

## TRACKING TIME

## STATION ACQUISITION DATA

## STATION TERMINATION DATA

HRS MIN SEC

DAY HRS MIN SEC RA DEC AZ ELV X Y RANGE

DAY HRS MIN SEC RA DEC AZ ELV X Y RANGE

BDA-02 X-Y USB-30	1 4 53	4 4 31 43 73 27 -119 80 -9 -5 206199	4 5 36 36 74 28 -96 68 -22 -2 206476
ANG-91 X-Y USB-30	1 4 49	4 4 31 46 73 28 -45 74 -11 11 206274	4 5 36 35 74 28 -62 62 -25 13 206620
ACN-75 X-Y USB-30	1 4 47	4 4 31 59 72 28 -56 21 -65 31 208300	4 5 36 46 74 28 -60 9 -80 30 209137
MAD-23 X-Y USB-85	1 4 46	4 4 31 60 73 27 -80 31 -17 -57 207776	4 5 36 45 74 27 -71 20 -41 -63 208469
CYI-04 X-Y USB-30	1 4 46	4 4 32 1 73 27 -76 38 -51 11 207446	4 5 36 47 74 27 -71 25 -64 17 208198
HAW-12 X-Y USB-30	1 4 44	4 6 25 24 75 27 71 37 52 15 207703	4 7 30 8 76 28 73 50 39 11 207180
GYM-14 X-Y USB-30	1 4 55	4 6 25 34 75 28 91 82 8 0 206358	4 7 30 28 76 28 -91 85 -5 0 206419
GDS-28 X-Y USR-85	1 4 53	4 6 25 35 75 27 117 75 7 14 206443	4 7 30 28 76 28 178 82 8 0 206435
STN-00 A-E USB150	1 4 49	4 6 25 35 75 27 114 70 19 -8 206531	4 7 30 24 76 28 152 79 5 -10 206463
TEX-16 X-Y USB-30	1 4 51	4 6 25 40 74 28 -91 86 -4 0 206324	4 7 30 31 75 28 -85 73 -17 1 206560

36

## VEHICLE 1 RADAR TABLE

## TRACKING TIME

## STATION ACQUISITION DATA

## STATION TERMINATION DATA

HRS MIN SEC

DAY HRS MIN SEC

RA DEC

AZ ELV

X

Y

RANGE

DAY HRS MIN SEC

RA DEC

AZ

ELV

X

Y

RANGE

MIL-71 X-Y USB-30	1 4 55	4 6 25 43	74 27 -88	71 -19	1	206493	4 7 30 38	75 28 -83	58 -32	4	206931
GBM-41 X-Y USB-30	1 5 0	4 6 25 44	74 28 -82	69 -21	3	206540	4 7 30 45	75 28 -79	55 -34	6	207018
BDA-02 X-Y USB-30	1 4 57	4 6 25 48	74 27 -89	57 -33	1	206848	4 7 30 45	75 28 -82	45 -45	6	207434
ANG-91 X-Y USB=30	1 4 58	4 6 25 49	74 28 -67	51 -36	14	207055	4 7 30 47	75 28 -68	38 -50	17	207737
MAD-23 X-Y USB-85	1 5 3	4 6 25 53	74 27 -64	11 -65 -62	209046		4 7 30 56	75 27 -55	2 -87 -55	209754	
CYI-04 X-Y USB-30	1 4 58	4 6 25 56	74 27 -67	15 -74	22	208842	4 7 30 54	75 27 -61	3 -87	29	209672
GWM-24 X-Y USB-30	0 8 33	4 8 18 58	77 27 64	8 81	26	209437	4 8 27 31	77 27 64	10 79	26	208890
HAW-12 X-Y USB-30	0 8 25	4 8 19 6	76 28 73	61 28	8	206930	4 8 27 31	76 28 72	63 26	8	206446
STN-00 A-E USB150	0 8 16	4 8 19 14	76 27 -148	79 -6 -10	206569		4 8 27 31	76 27 -141	78 -8 -10	206155	
GDS-28 X-Y USB-85	0 8 15	4 8 19 15	75 27 -124	77 7 -11	206587		4 8 27 31	75 27 -119	76 7 -13	206182	

11  
37

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
GYM-14 X-Y USB-30	0	8	13	4	8	19	17	75	28	-87	74	-16	1	206635	4	8	27	31	75	28	-86	72	-18	1	206243
TEX-16 X-Y USB-30	0	8	11	4	8	19	20	75	28	-83	62	-28	3	206901	4	8	27	31	75	28	-82	60	-30	4	206531
MIL-71 X-Y USB-30	0	8	8	4	8	19	23	75	27	-79	47	-42	7	207402	4	8	27	31	75	27	-79	45	-44	8	207054
GBM-41 X-Y USB-30	0	8	5	4	8	19	26	75	27	-77	45	-45	9	207505	4	8	27	31	75	27	-76	43	-46	10	207163
BDA-02 X-Y USB-30	0	8	5	4	8	19	26	75	27	-77	34	-55	11	207981	4	8	27	31	75	27	-76	33	-57	12	207646
ANG-91 X-Y USB-30	0	8	3	4	8	19	28	75	28	-67	27	-61	20	208356	4	8	27	31	75	27	-67	25	-63	21	208035
				31 JAN 71											LAZ	72.1									
GWM-24 X-Y USB-30	0	56	28	4	8	27	37	77	27	64	10	79	26	208883	4	9	24	5	78	28	65	22	66	23	208726
HAW-12 X-Y USB-30	0	56	36	4	8	27	37	76	28	72	63	26	8	206441	4	9	24	13	77	28	65	74	14	6	206705
STN-00 A-E USB150	0	56	44	4	8	27	37	76	27	-141	78	-8	-10	206149	4	9	24	21	77	28	-111	69	-20	-8	206812
GDS-28 X-Y USB-85	0	56	47	4	8	27	37	75	27	-119	76	7	-13	206177	4	9	24	24	76	28	-101	65	5	-24	206897

1-38

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
GVM-14 X-Y USB-30	0	56	49	4	8	27	37	75	28	-86	72	-18	1	206238	4	9	24	26	76	28	-82	60	-30	4	207039
TEX-16 X-Y USB-30	0	56	51	4	8	27	37	75	28	-82	60	-30	4	206527	4	9	24	28	76	28	-78	48	-41	8	207451
MIL-71 X-Y USB-30	0	56	54	4	8	27	37	75	27	-79	45	-44	8	207050	4	9	24	31	76	28	-74	34	-55	13	208091
GBM-41 X-Y USB-30	0	56	54	4	8	27	37	75	27	-76	43	-46	10	207159	4	9	24	31	76	28	-73	31	-57	15	208223
ANG-91 X-Y USB-30	0	56	55	4	8	27	37	75	27	-67	25	-63	21	208030	4	9	24	32	76	28	-65	14	-75	24	209205
BDA-02 X-Y USB-30	0	56	57	4	8	27	37	75	27	-76	33	-57	12	207642	4	9	24	34	76	27	-71	22	-67	18	208733
31 JAN 71 CIR TO LOPC1 IGNITION LAZ 72.1																									
GVM-24 X-Y USB-30	1	11	46	4	10	11	42	78	28	65	32	55	21	208327	4	11	23	28	79	28	63	47	40	18	207716
HOK-29 X-Y USB-85	1	11	46	4	10	11	42	77	28	47	7	-79	47	209718	4	11	23	28	79	29	35	16	-71	33	209265
HAM-12 X-Y USB-30	1	11	54	4	10	11	56	77	28	30	83	3	6	206755	4	11	23	49	78	28	-62	77	-12	6	206919
GDS-28 X-Y USB-85	1	11	59	4	10	12	4	76	27	-92	55	1	-35	207324	4	11	24	3	77	28	-83	42	-8	-48	207977

L-39

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
STN-00 A-E USB150	1	11	54		4	10	12	5	76	27	-99	60	-30	-5	207193		4	11	23	59	77	28	-88	46	-44	2	207781
GYM-14 X-Y USB-30	1	11	57		4	10	12	6	76	28	-79	50	-40	7	207536		4	11	24	3	77	28	-74	35	-54	13	208289
TEX-16 X-Y USB-30	1	11	56		4	10	12	10	76	27	-75	38	-51	11	208036		4	11	24	6	77	28	-70	24	-65	18	208885
BDA-02 X-Y USB-30	1	11	51		4	10	12	12	76	27	-66	12	-76	24	209399		4	11	24	3	77	27	-57	0	-90	33	210254
GBM-41 X-Y USB-30	1	11	55		4	10	12	13	76	27	-69	21	-67	19	208896		4	11	24	8	77	27	-63	7	-82	27	209812
MIL-71 X-Y USB-30	1	11	51		4	10	12	13	76	27	-71	24	-65	18	208746		4	11	24	5	77	27	-64	10	-79	25	209644
ANG-91 X-Y USB-30	0	15	56		4	10	12	14	76	27	-62	3	-86	28	209941		4	10	28	10	76	27	-61	0	-90	29	209348
CRO-08 X-Y USB-30	0	3	48		4	11	19	34	79	28	59	0	90	31	210012		4	11	23	22	79	28	58	1	89	32	210171
CRO-08 X-Y USB-30	1	11	42		4	12	10	1	79	28	53	10	78	37	209762		4	13	21	43	80	29	42	21	61	44	209180
HSK-25 X-Y USB-85	1	11	42		4	12	10	9	78	29	25	21	-67	24	209093		4	13	21	51	79	29	10	25	-65	9	208947

1-10

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
GWM-24 X-Y USB-30	1	11	40	4	12	10	12	79	28	59	57	29	16	207453	4	13	21	52	80	28	40	70	13	15	207197
HAW-12 X-Y USB-30	1	11	40	4	12	10	29	78	28	-71	66	-22	7	207184	4	13	22	10	79	28	-73	52	-37	11	207744
GDS-28 X-Y USB-85	1	11	37	4	12	10	37	77	27	-77	32	-19	-56	208489	4	13	22	13	78	27	-69	19	-45	-62	209305
STN-00 A-E USB150	1	11	35	4	12	10	38	77	27	-81	37	-53	7	208262	4	13	22	13	78	27	-73	24	-65	16	209041
GYM-14 X-Y USB-30	1	11	34	4	12	10	39	77	27	-71	25	-64	17	208861	4	13	22	12	78	27	-65	11	-77	25	209750
TEX-16 X-Y USB-30	1	11	35	4	12	10	40	77	27	-66	14	-75	23	209489	4	13	22	15	78	27	-59	1	-89	31	210383
MIL-71 X-Y USB-30	0	6	0	4	12	10	41	77	27	-59	1	-89	31	210244	4	12	16	41	77	27	-59	0	-90	31	210007
CRO-08 X-Y USB-30	1	11	54	4	14	8	26	80	28	34	27	47	48	208921	4	15	20	20	81	29	18	34	25	52	208662
MSK-25 X-Y USB-85	1	11	57	4	14	8	34	79	29	-2	26	-64	-2	209002	4	15	20	31	80	29	-18	23	-66	-17	209231
GWM-24 X-Y USB-30	1	10	10	4	14	8	35	79	28	8	75	2	15	207201	4	15	18	45	80	28	-40	70	-13	15	207293

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
HAW-12 X-Y USB-30	1	11	60	4	14	8	48	78	28	-72	41	-48	14	208247	4	15	20	48	80	28	-69	26	-62	19	209099
GYM-14 X-Y USB-30	0	11	13	4	14	8	51	78	27	-60	2	-88	30	210370	4	14	20	4	78	27	-59	0	-90	31	209928
STN-00 A-E USB150	1	11	58	4	14	8	51	78	27	-67	15	-74	22	209608	4	15	20	50	80	27	-58	3	-86	32	210428
GDS-28 X-Y USB-85	1	2	50	4	14	8	51	78	27	-64	10	-68	-62	209886	4	15	11	42	80	27	-56	0	-90	-56	210129
CRO-08 X-Y USB-30	1	11	40	4	16	6	57	80	29	5	36	7	53	208641	4	17	18	37	81	29	-13	35	-18	53	208779
HSK-25 X-Y USB-85	1	11	41	4	16	7	4	80	29	-29	20	-68	-27	209514	4	17	18	45	81	29	-42	12	-75	-40	210068
GWM-24 X-Y USB-30	1	11	38	4	16	7	11	80	28	-55	61	-24	16	207673	4	17	18	49	81	28	-63	47	-40	18	208272
HAW-12 X-Y USB-30	1	11	36	4	16	7	21	79	28	-66	16	-73	23	209710	4	17	18	56	81	28	-61	2	-87	29	210645
MAD-23 X-Y USB-85	0	39	6	4	16	39	22	82	27	53	0	-90	53	209416	4	17	18	28	82	27	59	6	-79	58	210412
MAD-23 X-Y USB-85	1	11	54	4	18	5	7	82	27	66	14	-59	62	210051	4	19	17	1	84	27	75	26	-28	60	209451

L-42

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
CRD-08 X-Y USB-30	1	11	59	4	18	5	20	81	29	-25	32	-34	50	209034	4	19	17	19	82	29	-39	24	-55	45	209576		
HSK-25 X-Y USB-85	0	32	5	4	18	5	24	81	28	-50	5	-83	-49	210547	4	18	37	29	81	28	-54	0	-90	-54	209580		
GWM-24 X-Y USB-30	1	11	59	4	18	5	28	81	28	-65	36	-51	20	208804	4	19	17	26	82	28	-65	22	-66	23	209698		
CYI-04 X-Y USB-30	1	8	7	4	18	8	53	82	27	59	0	90	31	210676	4	19	16	60	84	28	65	12	76	24	210203		
ACN-75 X-Y USB-30	1	11	42	4	20	3	25	84	28	60	8	81	30	210583	4	21	15	7	85	28	55	22	64	32	209832		
CYI-04 X-Y USB-30	1	11	39	4	20	3	31	84	28	70	22	67	19	209745	4	21	15	10	85	28	75	36	53	12	209075		
MAD-23 X-Y USB-85	1	11	42	4	20	3	34	83	27	82	35	-12	55	209087	4	21	15	15	85	28	92	47	1	43	208585		
CRD-08 X-Y USB-30	1	11	43	4	20	3	54	82	28	-47	16	-68	41	210040	4	21	15	37	83	28	-56	4	-85	34	210854		
GWM-24 X-Y USB-30	0	55	57	4	20	3	59	82	28	-64	11	-78	26	210341	4	20	59	56	83	28	-61	0	-90	29	210296		
ACN-75 X-Y USB-30	0	8	4	4	22	1	36	85	28	50	31	52	33	209444	4	22	9	40	85	28	49	32	50	34	208962		

1  
143  
1

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
ANG-91 X-Y USB-30	0	8	2		4	22	1	38	85	28	62	3	87	28	211051		4	22	9	40	85	28	63	4	85	27	210548
BDA-02 X-Y USB-30	0	8	2		4	22	1	38	85	27	62	8	82	27	210763		4	22	9	40	85	27	63	9	80	26	210272
CYI-04 X-Y USB-30	0	7	58		4	22	1	42	85	28	78	46	43	8	208727		4	22	9	40	85	28	79	48	41	8	208258
MAD-23 X-Y USB-85	0	7	55		4	22	1	45	84	28	101	56	7	33	208357		4	22	9	40	84	28	102	58	8	31	207912

31 JAN 71 LOPCI TO LM JETT IGNITION

LAZ 72.1

ANG-91 X-Y USB-30	1	3	40		4	22	9	59	85	28	63	4	85	27	210528		4	23	13	39	86	28	65	17	71	23	210263
BDA-02 X-Y USB-30	1	3	46		4	22	9	59	85	27	63	9	80	26	210253		4	23	13	44	86	28	70	21	68	19	210082
ACN-75 X-Y USB-30	1	3	47		4	22	9	59	85	28	49	33	50	34	208944		4	23	13	45	86	28	39	43	34	35	208960
CYI-04 X-Y USB-30	1	3	50		4	22	9	59	85	28	79	48	41	8	208240		4	23	13	48	86	28	82	61	29	4	208292
MAD-23 X-Y USB-85	1	3	52		4	22	9	59	84	28	102	58	8	31	207895		4	23	13	50	85	28	120	69	11	18	208109
GBM-41 X-Y USB-30	0	40	44		4	22	32	58	85	27	59	0	90	31	209949		4	23	13	42	86	28	63	7	82	27	210835

12  
11  
10  
9  
8  
7  
6  
5  
4  
3  
2

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
MIL-71 X-Y USB-30	0	35	36	4	22	38	8	85	27	58	0	90	32	209918	4	23	13	44	86	27	62	6	83	28	210908
ANG-91 X-Y USB-30	1	11	37	5	0	0	13	86	28	67	27	61	20	209790	5	1	11	50	87	28	67	42	45	16	209131
MIL-71 X-Y USB-30	1	11	36	5	0	0	14	86	27	67	16	73	22	210446	5	1	11	50	87	28	73	30	59	15	209750
TEX-16 X-Y USB-30	1	11	32	5	0	0	15	86	27	60	2	88	30	211248	5	1	11	47	87	28	67	15	73	23	210530
GBM-41 X-Y USB-30	1	11	31	5	0	0	17	86	27	67	17	72	22	210370	5	1	11	47	87	28	72	31	58	15	209670
BDA-02 X-Y USB-30	1	11	33	5	0	0	18	86	27	75	30	59	13	209652	5	1	11	51	87	28	81	44	46	6	209062
ACN-75 X-Y USB-30	1	11	37	5	0	0	21	85	28	27	49	21	36	208771	5	1	11	58	86	28	4	54	3	36	208690
CYI-04 X-Y USB-30	1	11	37	5	0	0	25	85	28	85	72	18	2	208117	5	1	12	2	86	28	87	87	3	0	208038
MAD-23 X-Y USB-85	1	11	38	5	0	0	28	85	28	147	75	13	8	208056	5	1	12	5	86	28	-150	76	13	-7	208141
GYM-14 X-Y USB-30	0	26	48	5	0	44	58	87	27	59	0	90	31	210201	5	1	11	47	87	27	61	5	85	29	211149

11-45

## VEHICLE 1 RADAR TABLE

## TRACKING TIME

## STATION ACQUISITION DATA

## STATION TERMINATION DATA

HRS MIN SEC

DAY HRS MIN SEC

RA DEC

AZ ELV

X

Y

RANGE

DAY HRS MIN SEC

RA DEC

AZ ELV

X

Y

RANGE

GDS-28 X-Y USB-85	0 24 20	5 0 47 30	87 27	56 0 -90	56 210275	5 1 11 50	87 27	59 4 -82	59 211205
STN-00 A-E USB150	0 9 20	5 1 2 33	87 27	55 0 90	35 210948	5 1 11 53	87 27	56 1 88	34 211357
GYM-14 X-Y USB-30	1 12 1	5 1 58 23	87 27	66 14 75	23 210708	5 3 10 24	88 28	72 28	61 16 210006
TEX-16 X-Y USB-30	1 11 59	5 1 58 23	87 27	71 25 64	17 210093	5 3 10 22	88 28	76 39	50 11 209443
GBM-41 X-Y USB-30	1 12 1	5 1 58 25	87 28	76 41 48	11 209287	5 3 10 26	88 28	79 56	33 6 208784
MIL-71 X-Y USB-30	1 12 0	5 1 58 25	87 28	76 40 50	10 209362	5 3 10 26	88 28	81 54	35 5 208847
GDS-28 X-Y USB-85	1 11 58	5 1 58 26	87 27	65 12 -62	62 210807	5 3 10 24	88 27	73 25	-32 60 210156
STN-00 A-E USB150	1 11 58	5 1 58 26	87 27	63 9 79	27 210981	5 3 10 24	88 27	71 22	67 17 210346
BDA-02 X-Y USB-30	1 12 2	5 1 58 27	87 28	86 54 36	2 208776	5 3 10 29	88 28	96 68	22 -2 208447
ANG-91 X-Y USB-30	1 11 60	5 1 58 28	87 28	66 53 35	14 208814	5 3 10 27	88 28	57 67	19 12 208470

947

12  
11  
10  
9  
8  
7  
6  
5  
4  
3  
2

## VEHICLE 1 RADAR TABLE

	TRACKING TIME												STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS MIN SEC			DAY HRS MIN SEC			RA DEC			AZ ELV			X Y RANGE			DAY HRS MIN SEC			RA DEC			AZ ELV			X Y RANGE											
ACN-75 X-Y USB-30	1	12	1	5	1	58	37	86	28	-14	53	-11	36	208813		5	3	10	38	87	28	-34	45	-29	35	209186										
CYI-04 X-Y USB-30	1	12	2	5	1	58	37	86	28	-88	83	-7	0	208146		5	3	10	39	87	28	-84	68	-22	2	208471										
MAD-23 X-Y USB-85	1	12	0	5	1	58	40	86	28	-122	69	11	-18	208338		5	3	10	40	87	28	-101	57	7	-32	208760										
STN-00 A-E USB150	1	11	40	5	3	56	55	88	27	77	31	59	11	209949		5	5	8	34	89	28	86	44	46	3	209391										
GDS-28 X-Y USB-85	1	11	38	5	3	56	55	88	27	79	35	-16	54	209758		5	5	8	34	89	28	87	48	-3	42	209218										
GYM-14 X-Y USB-30	1	11	42	5	3	56	56	88	28	76	38	51	11	209589		5	5	8	38	89	28	80	53	37	6	209054										
TEX-16 X-Y USB-30	1	11	41	5	3	56	58	88	28	79	50	40	7	209093		5	5	8	38	89	28	83	65	25	3	208691										
GBM-41 X-Y USB-30	1	11	41	5	3	57	1	88	28	81	67	23	3	208563		5	5	8	42	89	28	80	82	8	2	208401										
MIL-71 X-Y USB-30	1	11	44	5	3	57	1	88	28	85	65	25	2	208611		5	5	8	45	89	28	91	80	10	0	208423										
BDA-02 X-Y USB-30	1	11	44	5	3	57	4	88	28	110	78	11	-4	208353		5	5	8	48	89	28	-139	84	-4	-5	208387										

L7-L

## VEHICLE 1 RADAR TABLE

	TRACKING TIME				STATION ACQUISITION DATA										STATION TERMINATION DATA												
	HRS	MIN	SEC		DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	
ANG-91 X-Y USB-30	1	11	45		5	3	57	6	88	28	37	76	9	11	208381		5	5	8	51	89	28	-31	77	-7	11	208456
ACN-75 X-Y USB-30	1	11	45		5	3	57	16	87	28	-44	38	-42	34	209572		5	5	9	1	88	28	-54	25	-60	32	210323
CYI-04 X-Y USB-30	1	11	46		5	3	57	17	87	28	-81	57	-33	5	208814		5	5	9	3	88	28	-77	42	-47	10	209487
MAD-23 X-Y USB-85	1	11	43		5	3	57	20	87	27	-92	48	2	-42	209138		5	5	9	3	88	27	-82	35	-11	-54	209819
HAW-12 X-Y USB-30	0	45	20		5	4	23	12	89	27	60	0	90	30	210540		5	5	8	32	90	28	64	9	80	26	211257
HAW-12 X-Y USB-30	1	11	53		5	5	55	5	90	27	67	19	70	22	210790		5	7	6	58	91	28	71	33	55	16	210069
STN-00 A-E USB150	1	11	54		5	5	55	11	89	27	93	53	36	-2	209125		5	7	7	5	90	28	108	67	22	-7	208806
GYM-14 X-Y USB-30	1	11	55		5	5	55	12	89	28	83	63	27	3	208824		5	7	7	7	90	28	88	78	12	0	208606
GDS-28 X-Y USB-85	1	11	53		5	5	55	12	89	28	94	58	2	32	208975		5	7	7	5	90	28	109	72	6	17	208705
TEX-16 X-Y USB-30	1	11	55		5	5	55	17	89	28	86	75	15	1	208569		5	7	7	12	90	28	-40	90	0	0	208537

12  
11  
10  
9  
8  
7  
6  
5  
4  
3  
2

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
MIL-71 X-Y USB-30	1	11	53	5	5	55	24	89	28	-175	89	0	-1	208449	5	7	7	17	90	28	-89	75	-15	0	208658
GBM-41 X-Y USB-30	1	11	52	5	5	55	25	89	28	-64	87	-2	1	208450	5	7	7	17	90	28	-82	73	-17	2	208697
BDA-02 X-Y USB-30	1	11	52	5	5	55	28	88	28	-104	75	-14	-4	208560	5	7	7	20	89	28	-90	61	-29	0	208974
ANG-91 X-Y USB-30	1	11	52	5	5	55	28	88	28	-55	69	-17	12	208667	5	7	7	21	89	28	-66	55	-33	14	209164
CYI-04 X-Y USB-30	1	11	48	5	5	55	40	88	27	-73	32	-57	14	210023	5	7	7	28	89	27	-68	18	-70	21	210895
MAD-23 X-Y USB-85	1	11	47	5	5	55	41	88	27	-76	26	-27	-60	210329	5	7	7	27	89	27	-66	14	-58	-62	211119
ACN-75 X-Y USB-30	1	11	46	5	5	55	42	88	28	-58	16	-72	31	210917	5	7	7	28	89	28	-62	1	-89	28	211885
HAM-12 X-Y USB-30	1	11	47	5	7	53	37	91	28	72	44	45	13	209670	5	9	5	25	92	28	73	59	30	9	209187
STN-00 A-E USB150	1	11	47	5	7	53	49	90	27	128	75	12	-9	208717	5	9	5	35	91	28	-160	80	-4	-10	208754
GDS-28 X-Y USB-85	1	11	51	5	7	53	49	90	28	138	80	8	7	208658	5	9	5	39	91	28	-132	79	7	-8	208765

1 hr

6 hr

## VEHICLE 1 F4FAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
GYM-14 X-Y USB-30	1	11	56	5	7	53	50	90	28	103	83	1	0	208604	5	9	5	46	91	28	-87	76	-14	1	208806		
TEX-16 X-Y USB-30	1	11	54	5	7	53	53	90	28	-87	79	-11	1	208661	5	9	5	47	91	28	-83	64	-26	3	209045		
GBM-41 X-Y USB-30	1	11	55	5	7	53	57	89	28	-81	62	-28	4	208997	5	9	5	52	90	28	-77	47	-42	9	209623		
MIL-71 X-Y USB-30	1	11	54	5	7	53	57	89	28	-85	64	-26	2	208932	5	9	5	51	90	28	-80	50	-40	7	209522		
BDA-02 X-Y USB-30	1	11	56	5	7	53	58	89	27	-85	51	-39	3	209365	5	9	5	55	90	27	-78	37	-53	10	210090		
ANG-91 X-Y USB-30	1	11	58	5	7	54	1	89	28	-68	44	-43	16	209618	5	9	5	59	90	28	-67	29	-59	20	210454		
MAD-23 X-Y USB-85	0	42	31	5	7	54	2	89	27	-59	6	-78	-59	211640	5	8	36	33	90	27	-53	0	-90	-53	210822		
CYI-04 X-Y USB-30	0	47	54	5	7	54	4	89	27	-64	9	-80	26	211497	5	8	41	58	90	27	-59	0	-90	31	210964		
GWM-24 X-Y USB-30	0	28	43	5	8	36	35	92	27	62	0	90	28	210821	5	9	5	18	92	28	63	6	83	27	211751		
GWM-24 X-Y USB-30	1	11	45	5	9	51	50	92	27	65	16	72	24	211246	5	11	3	35	93	28	66	31	57	21	210494		

E 50

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
HAW-12 X-Y USB-30	1	11	46	5	9	52	3	91	28	70	69	20	7	208997	5	11	3	49	92	28	39	83	5	6	208883		
STN-00 A-E USB150	1	11	44	5	9	52	15	91	27	-122	73	-14	-9	208906	5	11	3	58	92	27	-100	61	-29	-5	209296		
GDS-28 X-Y USB-85	1	11	45	5	9	52	15	91	27	-108	70	6	-19	208962	5	11	4	1	92	27	-93	57	2	-33	209423		
GYM-14 X-Y USB-30	1	11	44	5	9	52	18	91	28	-84	66	-24	2	209062	5	11	4	3	92	28	-80	51	-39	6	209629		
TEX-16 X-Y USB-30	1	11	44	5	9	52	20	90	28	-80	54	-36	6	209416	5	11	4	4	91	27	-76	39	-50	11	210123		
GBM-41 X-Y USB-30	1	11	44	5	9	52	24	90	27	-74	37	-52	12	210123	5	11	4	8	91	27	-70	22	-66	19	210979		
MIL-71 X-Y USB-30	1	11	41	5	9	52	24	90	27	-77	39	-50	10	210001	5	11	4	6	91	27	-71	25	-64	17	210830		
BDA-02 X-Y USB-30	1	11	40	5	9	52	26	90	27	-73	27	-62	15	210623	5	11	4	6	91	27	-66	14	-75	23	211483		
ANG-91 X-Y USB-30	1	11	38	5	9	52	29	90	27	-66	19	-69	23	211054	5	11	4	7	91	27	-63	4	-85	27	212018		
HSK-25 X-Y USB-85	0	43	27	5	10	20	9	92	28	54	0	-90	54	210972	5	11	3	36	93	28	48	6	-81	48	211871		

151

## VEHICLE 1 RADAR TABLE

## TRACKING TIME

## STATION ACQUISITION DATA

## STATION TERMINATION DATA

HRS MIN SEC

DAY HRS MIN SEC RA DEC AZ ELV X Y RANGE

DAY HRS MIN SEC RA DEC AZ ELV X Y RANGE

HSK-25  
X-Y USB-85

1 11 55 5 11 50 18 93 28 40 13 -73 39 211559

5 13 2 13 94 28 27 21 -67 25 211207

GWM-24  
X-Y USB-30

1 11 54 5 11 50 18 93 28 65 42 46 19 210079

5 13 2 11 94 28 60 56 30 16 209586

HAW-12  
X-Y USB-30

1 12 4 5 11 50 30 92 28 -47 82 -6 6 208960

5 13 2 34 93 28 -71 68 -21 7 209289

STN-00  
A-E USB150

1 12 1 5 11 50 37 92 27 -92 51 -39 -1 209668

5 13 2 38 93 27 -82 38 -52 6 210356

GDS-28  
X-Y USB-85

1 11 60 5 11 50 39 91 27 -86 47 -3 -43 209835

5 13 2 39 93 27 -78 33 -17 -55 210580

TEX-16  
X-Y USB-30

1 12 1 5 11 50 42 91 27 -72 29 -60 15 210673

5 13 2 43 93 27 -67 15 -74 22 211575

GYM-14  
X-Y USB-30

1 11 60 5 11 50 42 91 27 -77 41 -49 10 210107

5 13 2 42 93 27 -71 26 -63 17 210948

BDA-02  
X-Y USB-30

0 26 34 5 11 50 43 91 27 -61 5 -85 29 212061

5 12 17 17 92 27 -58 0 -90 32 211168

GBM-41  
X-Y USB-30

1 8 10 5 11 50 43 91 27 -66 13 -76 24 211584

5 12 58 54 93 27 -59 0 -90 31 212262

MIL-71  
X-Y USB-30

1 11 58 5 11 50 45 91 27 -67 15 -73 22 211423

5 13 2 43 93 27 -60 2 -88 30 212335

1-52

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HR	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HR	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
CRD-08 K-Y USB-30	0	49	50	5	12	12	15	93	28	59	0	90	31	211308	5	13	2	5	94	28	53	9	79	36	211898
CRD-08 K-Y USB-30	1	11	40	5	13	48	40	94	28	47	17	67	41	211500	5	15	0	20	95	28	35	27	48	47	211015
GWH-24 K-Y USB-30	1	11	42	5	13	48	48	94	28	50	66	19	15	209384	5	15	0	30	95	28	13	75	4	14	209283
HSK-25 X-Y USB-85	1	11	42	5	13	48	48	94	28	16	24	-65	14	211081	5	15	0	30	95	28	0	26	-64	0	211066
HAM-12 X-Y USB-30	1	11	42	5	13	49	4	93	27	-73	57	-32	9	209614	5	15	0	46	94	27	-72	42	-46	13	210298
GDS-28 K-Y USB-85	1	11	39	5	13	49	11	92	27	-73	24	-34	-61	211101	5	15	0	50	94	27	-65	11	-65	-63	211046
TEX-16 K-Y USB-30	0	30	3	5	13	49	12	92	27	-62	5	-84	28	212158	5	14	19	15	93	27	-59	0	-90	31	211258
SIN-00 A-E USB150	1	11	38	5	13	49	12	93	27	-76	29	-61	12	210849	5	15	0	49	94	27	-68	16	-73	21	211669
GWH-14 K-Y USB-30	1	11	38	5	13	49	12	92	27	-68	16	-72	22	211520	5	15	0	50	94	27	-61	3	-87	29	212425
CRD-08 K-Y USB-30	1	12	2	5	15	46	60	95	28	25	32	33	50	210818	5	16	59	2	96	28	7	37	9	53	210702

1,53

## VEHICLE I RADAR TABLE

## TRACKING TIME

## STATION ACQUISITION DATA

## STATION TERMINATION DATA

HRS MIN SEC

DAY HRS MIN SEC

PA DEC AZ ELV X Y RANGE

DAY HRS MIN SEC

RA DEC AZ ELV

X Y RANGE

HSK-25 X-Y USB-85	1 12 1	5 15 47 7 94 28 -12 25 -64 -11 211189	5 16 59 8 95 28 -28 20 -67 -26 211548
GWM-24 X-Y USB-30	1 12 0	5 15 47 9 94 28 -27 74 -7 14 209376	5 16 59 9 95 28 -55 62 -23 15 209727
HAW-12 X-Y USB-30	1 11 59	5 15 47 18 94 27 -71 32 -57 16 210846	5 16 59 17 95 27 -67 17 -71 22 211754
STN-00 A-E USB150	0 46 21	5 15 47 20 94 27 -62 7 -82 28 212216	5 16 33 41 95 27 -56 0 -90 34 211549
GDS-28 X-Y USB-85	0 15 15	5 15 47 21 94 27 -59 3 -85 -58 212499	5 16 2 36 94 27 -57 0 -90 -57 211892
MAD-23 X-Y USB-85	1 11 31	5 17 45 30 96 27 55 1 -88 55 212737	5 18 57 1 98 27 65 12 -62 63 212153
CRD-08 X-Y USB-30	1 11 37	5 17 45 41 96 28 -6 37 -8 53 210743	5 18 57 18 97 28 -24 33 -32 50 211036
HSK-25 X-Y USB-85	1 11 37	5 17 45 49 95 28 -37 15 -71 -36 211877	5 18 57 26 96 28 -49 6 -81 -49 212537
GWM-24 X-Y USB-30	1 11 40	5 17 45 52 95 27 -62 53 -34 17 210079	5 18 57 32 96 27 -66 38 -50 19 210816
HAW-12 X-Y USB-30	0 37 16	5 17 45 58 95 27 -64 7 -82 26 212365	5 18 23 14 95 27 -61 0 -90 29 211540

H  
4512  
11  
10  
9  
8  
7  
6  
5  
4  
3  
2

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
CYI-04 X-Y USB-30	1	11	59	5	19	43	31	98	27	63	7	82	26	212561	5	20	55	30	99	27	70	21	68	19	211836
MAD-23 X-Y USB-85	1	11	58	5	19	43	33	98	27	72	21	-39	63	211767	5	20	55	31	99	27	81	33	-13	56	211174
CRD-08 X-Y USB-30	1	12	2	5	19	43	52	96	28	-35	28	-47	47	211369	5	20	55	53	97	28	-47	18	-66	41	212020
GWM-24 X-Y USB-30	1	11	56	5	19	44	2	96	27	-66	27	-61	21	211391	5	20	55	58	97	27	-65	12	-76	25	212343
ACN-75 X-Y USB-30	0	32	10	5	20	23	17	99	27	62	0	90	28	211704	5	20	55	27	99	27	61	7	83	29	212650
ACN-75 X-Y USB-30	1	11	41	5	21	42	4	99	27	58	17	71	30	212132	5	22	53	45	100	27	52	30	53	32	211453
CYI-04 X-Y USB-30	1	11	40	5	21	42	7	99	27	74	31	58	14	211374	5	22	53	47	100	27	79	45	44	8	210766
MAD-23 X-Y USB-85	1	11	43	5	21	42	11	99	27	89	42	-1	48	210820	5	22	53	53	100	27	101	55	7	34	210397
CRD-08 X-Y USB-30	0	53	52	5	21	42	33	97	28	-54	9	-78	36	212530	5	22	36	25	98	27	-59	0	-90	31	212289
GWM-24 X-Y USB-30	0	8	57	5	21	42	36	97	27	-63	2	-88	27	212985	5	21	51	33	97	27	-62	0	-90	28	212644

155

## VEHICLE 1 RACAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
BDA-02 X-Y USB-30	0	36	8	5	22	17	39	99	26	58	0	90	32	211844	5	22	53	47	100	27	62	6	83	27	212826		
ANG-91 X-Y USB-30	0	6	47	5	22	46	56	100	27	62	0	90	28	212829	5	22	53	42	100	27	62	1	88	28	213108		
BDA-02 X-Y USB-30	1	11	48	5	23	40	17	100	26	68	15	74	21	212386	6	0	52	6	101	27	75	29	61	13	211701		
ANG-91 X-Y USB-30	1	11	48	5	23	40	18	100	27	65	11	77	24	212599	6	0	52	6	101	27	68	26	62	20	211825		
GBM-41 X-Y USB-30	1	11	47	5	23	40	19	100	26	61	2	88	29	213162	6	0	52	7	101	27	68	16	73	21	212425		
MIL-71 X-Y USB-30	1	11	45	5	23	40	21	100	26	60	1	89	30	213220	6	0	52	5	101	27	67	14	75	22	212500		
ACN-75 X-Y USB-30	1	11	48	5	23	40	25	100	27	44	39	41	34	211105	6	0	52	13	101	27	29	49	22	35	210747		
CYI-04 X-Y USB-30	1	11	46	5	23	40	29	100	27	82	56	34	4	210452	6	0	52	15	101	27	88	71	19	1	210130		
MAD-23 X-Y USB-85	1	11	44	5	23	40	33	99	27	113	64	11	24	210206	6	0	52	17	100	27	145	74	13	9	210071		
TEX-16 X-Y USB-30	0	3	35	6	0	48	31	101	26	60	0	90	30	213160	6	0	52	7	101	26	60	1	89	30	213308		

1-56

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
GSM-41 X-Y USB-30	0	49	44	6	1	38	46	101	27	72	25	63	16	211947	6	2	28	31	102	27	75	35	54	12	210417		
MIL-71 X-Y USB-30	0	49	44	6	1	38	47	101	26	72	24	65	17	212027	6	2	28	31	102	27	76	34	55	12	210498		
ANG-91 X-Y USB-30	0	49	43	6	1	38	47	101	27	69	37	51	17	211364	6	2	28	31	102	27	69	47	41	14	209888		
TEX-16 X-Y USB-30	0	49	43	6	1	38	47	101	26	65	10	79	24	212830	6	2	28	31	102	26	70	19	69	19	211265		
BDA-02 X-Y USB-30	0	49	42	6	1	38	48	101	27	80	38	51	8	211290	6	2	28	31	102	27	85	48	42	3	209852		
CYI-04 X-Y USB-30	0	49	33	6	1	38	57	100	27	94	81	9	-1	210038	6	2	28	31	101	27	-123	88	-2	-1	208989		
ACN-75 X-Y USB-30	0	49	33	6	1	38	58	100	27	13	54	9	35	210649	6	2	28	31	101	27	-5	55	-3	35	209609		
MAD-23 X-Y USB-85	0	49	31	6	1	38	60	100	27	-173	76	14	-2	210095	6	2	28	31	101	27	-137	72	13	-12	209149		
GYM-14 X-Y USB-30	0	47	36	6	1	40	55	101	26	60	0	90	30	213322	6	2	28	31	102	26	65	9	81	25	211892		
GDS-28 X-Y USB-85	0	44	0	6	1	44	30	101	26	57	0	-90	57	213140	6	2	28	31	102	26	63	7	-75	62	211979		

1  
2  
3  
4  
5  
6  
7  
8  
9  
10  
11  
12

## VEHICLE 1 RADAR TABLE

## TRACKING TIME

## STATION ACQUISITION DATA

## STATION TERMINATION DATA

HRS MIN SEC

DAY HRS MIN SEC RA DEC AZ ELV X Y RANGE

DAY HRS MIN SEC RA DEC AZ ELV X Y RANGE

STN-00 A-E USB150	0 28 30	6 2 0 1 101 26 56 0 90 34 212416	6 2 28 31 102 26 60 4 85 30 212144
----------------------	---------	----------------------------------	------------------------------------

31 JAN 71 LM JETT TO TEI IGNITION

LAZ 72.1

GYM-14 X-Y USB-30	0 21 56	6 2 28 36 102 26 65 9 81 25 211894	6 2 50 32 102 26 67 13 76 23 212734
----------------------	---------	------------------------------------	-------------------------------------

TEX-16 X-Y USB-30	0 21 56	6 2 28 36 102 26 70 19 69 19 211267	6 2 50 32 102 26 71 24 65 17 212111
----------------------	---------	-------------------------------------	-------------------------------------

GDS-28 X-Y USB-85	0 21 58	6 2 28 36 102 26 63 7 -75 62 211981	6 2 50 35 102 26 66 11 -65 63 212843
----------------------	---------	-------------------------------------	--------------------------------------

GBM-41 X-Y USB-30	0 21 58	6 2 28 36 102 27 75 35 54 12 210419	6 2 50 35 102 27 77 40 49 10 211293
----------------------	---------	-------------------------------------	-------------------------------------

STN-00 A-E USB150	0 21 58	6 2 28 36 102 26 60 4 85 30 212147	6 2 50 35 102 26 63 8 81 27 213018
----------------------	---------	------------------------------------	------------------------------------

MIL-71 X-Y USB-30	0 21 59	6 2 28 36 102 27 76 34 55 12 210500	6 2 50 35 102 27 77 38 51 10 211373
----------------------	---------	-------------------------------------	-------------------------------------

ANG-91 X-Y USB-30	0 22 0	6 2 28 36 102 27 69 47 41 14 209891	6 2 50 37 102 27 68 52 36 13 210800
----------------------	--------	-------------------------------------	-------------------------------------

BDA-02 X-Y USB-30	0 22 4	6 2 28 36 102 27 85 48 42 3 209855	6 2 50 41 102 27 88 53 37 1 210780
----------------------	--------	------------------------------------	------------------------------------

ACN-75 X-Y USB-30	0 22 18	6 2 28 36 101 27 -5 55 -3 35 209612	6 2 50 54 101 27 -13 54 -9 35 210746
----------------------	---------	-------------------------------------	--------------------------------------

1-58

2  
1  
0  
9  
8  
7  
6  
5  
4  
3  
2

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
CYI-04 X-Y USB-30	0	22	18	6	2	28	36	101	27	-122	88	-2	-1	208992	6	2	50	54	101	27	-98	84	-6	-1	210117		
MAD-23 X-Y USB-85	0	22	22	6	2	28	36	101	27	-137	72	13	-12	209153	6	2	50	59	101	27	-126	69	13	-17	210322		
GYM-14 X-Y USB-30	1	11	40	6	3	37	6	102	26	71	23	66	17	212257	6	4	48	45	103	26	77	37	52	11	211587		
GDS-28 X-Y USB-85	1	11	37	6	3	37	8	102	26	71	20	-41	63	212401	6	4	48	45	103	26	79	33	-16	55	211765		
TEX-16 X-Y USB-30	1	11	36	6	3	37	8	102	26	75	34	55	12	211661	6	4	48	45	103	26	80	48	41	6	211078		
STN-00 A-E USB150	1	11	36	6	3	37	9	102	26	69	17	72	20	212591	6	4	48	44	103	26	78	29	60	10	211960		
MIL-71 X-Y USB-30	1	11	39	6	3	37	10	102	26	81	49	41	6	211002	6	4	48	49	103	27	87	63	27	1	210586		
GBM-41 X-Y USB-30	1	11	39	6	3	37	11	102	27	80	50	39	7	210929	6	4	48	49	103	27	84	65	24	3	210534		
ANG-91 X-Y USB-30	1	11	39	6	3	37	13	102	27	64	62	25	11	210531	6	4	48	52	103	27	44	76	10	10	210328		
BDA-02 X-Y USB-30	1	11	41	6	3	37	14	102	26	94	63	27	-2	210524	6	4	48	55	103	26	113	77	12	-5	210321		

-159

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
ACN-75 X-Y USB-30	1	11	39	6	3	37	28	101	27	-29	50	-22	35	210938	6	4	49	7	102	27	-45	40	-40	33	211469		
CYI-04 X-Y USB-30	1	11	39	6	3	37	29	101	27	-89	73	-17	0	210283	6	4	49	8	102	26	-84	58	-32	3	210758		
MAD-23 X-Y USB-85	1	11	37	6	3	37	31	101	26	-109	61	10	-27	210559	6	4	49	8	102	26	-95	48	4	-41	211100		
				31 JAN 71	TEI TO ENTRY INTERFACE												LAZ	72.1									
HAW-12 X-Y USB-30	13	16	12	6	5	25	22	103	26	62	1	89	28	214047	6	18	41	34	102	25	-62	0	-90	28	187404		
STN-00 A-E USB150	11	26	54	6	5	25	24	103	26	83	37	53	6	212064	6	16	52	18	102	25	-57	0	-90	33	191064		
GYM-14 X-Y USB-30	10	14	22	6	5	25	25	103	26	80	45	45	7	211692	6	15	39	47	101	26	-61	0	-90	29	193473		
GDS-28 X-Y USB-85	10	59	16	6	5	25	25	103	26	84	41	-7	49	211875	6	16	24	41	102	25	-58	0	-90	-58	191988		
TEX-16 X-Y USB-30	9	20	34	6	5	25	27	103	26	83	57	33	4	211250	6	14	46	0	102	26	-61	0	-90	29	195230		
MIL-71 X-Y USB-30	8	16	11	6	5	25	29	103	26	91	72	18	0	210865	6	13	41	40	101	26	-61	0	-90	29	197334		
GBM-41 X-Y USB-30	8	1	37	6	5	25	29	103	27	86	74	16	1	210826	6	13	27	6	101	26	-61	0	-90	29	197811		

109

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HR	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HR	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
ANG-91 X-Y USB-30	6	33	32	6	5	25	32	102	27	7	80	1	10	210740	6	11	59	4	101	26	-63	0	-90	27	200684		
BDA-02 X-Y USB-30	7	22	11	6	5	25	32	102	26	147	83	4	-6	210717	6	12	47	43	101	26	-59	0	-90	31	199104		
MAD-23 X-Y USB-85	3	46	54	6	5	25	38	102	26	-89	41	-2	-49	211852	6	9	12	32	101	26	-55	0	-90	-55	206131		
CYI-04 X-Y USB-30	3	53	53	6	5	25	38	102	26	-81	50	-40	6	211496	6	9	19	32	101	26	-60	0	-90	30	205898		
ACN-75 X-Y USB-30	2	33	23	6	5	25	40	102	27	-51	33	-50	32	212248	6	7	59	3	101	27	-63	0	-90	27	208574		
GWM-24 X-Y USB-30	12	45	41	6	9	19	56	103	26	63	0	90	27	205887	6	22	5	37	102	25	-64	0	-90	26	180405		
HSK-25 X-Y USB-85	9	5	48	6	10	54	9	103	27	56	0	-90	56	202797	6	19	59	57	102	26	-57	0	-90	-57	184733		
CRO-08 X-Y USB-30	10	6	56	6	12	44	58	103	27	60	0	90	30	199181	6	22	51	54	102	26	-61	0	-90	29	178780		
PKS-96 H-D USB210	0	26	18	6	15	14	47	103	27	4	30	7	60	192555	6	15	41	6	103	27	-3	30	-5	60	191676		
MAD-23 X-Y USB-85	15	0	47	6	18	8	1	104	25	56	0	-90	56	188534	7	9	8	48	104	24	-58	0	-90	-58	155841		

T9-1

## VEHICLE I RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA										
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y
CYI-04 X-Y USB-30	13	47	58	6	19	31	54	104	25	61	0	90	29	185699	7	9	19	51	103	24	-63	0	-90	27 155403
ACN-75 X-Y USB-30	11	23	18	6	20	40	42	104	26	64	0	90	26	183341	7	8	4	1	103	25	-65	0	-90	25 158380
BDA-02 X-Y USB-30	14	9	46	6	22	37	46	104	25	60	0	90	30	179284	7	12	47	32	104	24	-62	0	-90	28 146977
ANG-91 X-Y USB-30	12	59	36	6	23	2	18	104	25	64	0	90	26	178421	7	12	1	54	104	24	-65	0	-90	25 148861
GBM-41 X-Y USB-30	13	41	4	6	23	47	15	104	25	62	0	90	28	176835	7	13	28	19	104	24	-63	0	-90	27 145275
MIL-71 X-Y USB-30	13	49	52	6	23	52	31	104	25	61	0	90	29	176649	7	13	42	23	104	24	-63	0	-90	27 144683
TEX-16 X-Y USB-30	13	45	21	7	1	2	4	105	25	62	0	90	28	174166	7	14	47	25	104	23	-63	0	-90	27 141926
GYM-14 X-Y USB-30	13	46	24	7	1	55	11	105	25	62	0	90	28	172250	7	15	41	35	104	23	-63	0	-90	27 139594
GDS-28 X-Y USB-85	14	24	18	7	2	0	15	105	25	59	0	-90	59	172066	7	16	24	33	105	23	-61	0	-90	-61 137725
STN=00 A-E USB150	14	36	1	7	2	15	29	105	25	58	0	90	32	171512	7	16	51	30	105	23	-61	0	-90	29 136541

L-62



## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
HAW-12 X-Y USB-30	13	19	9	7	5	26	35	105	25	63	0	90	27	164427	7	18	45	44	105	23	-65	0	-90	25	131426		
GWM-24 X-Y USB-30	12	43	21	7	9	29	35	106	24	65	0	90	25	155015	7	22	12	56	105	23	-67	0	-90	23	121735		
HSK-25 X-Y USB-85	9	20	55	7	10	56	27	106	25	58	0	-90	58	151523	7	20	17	21	105	24	-59	0	-90	-59	127203		
CRO-C8 X-Y USB-30	10	18	32	7	12	49	48	106	25	62	0	90	28	146875	7	23	8	21	106	24	-64	0	=90	26	119039		
PKS-96 H-D USB210	1	59	46	7	14	39	49	106	25	15	30	25	57	140528	7	16	39	35	106	25	-16	30	-26	56	135342		
MAD-23 X-Y USB-85	14	37	2	7	18	30	58	107	23	60	0	-90	60	132098	8	9	7	60	109	19	-64	0	-90	-64	86436		
CYI-04 X-Y USB-30	13	35	45	7	19	51	20	108	23	64	0	90	26	128420	8	9	27	5	109	19	-68	0	-90	22	85265		
ACN-75 X-Y USB-30	11	30	54	7	20	52	55	109	24	66	0	90	24	125541	8	8	23	49	108	21	-69	0	-90	21	89101		
BDA-02 X-Y USB-30	13	51	56	7	23	1	38	109	22	64	0	90	26	119375	8	12	53	35	111	18	-69	0	-90	21	71873		
ANG-91 X-Y USB-30	12	54	38	7	23	21	48	109	22	66	0	90	24	118384	8	12	16	26	110	19	-70	0	-90	20	74388		

1-63

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
GBM-41 X-Y USB-30	13	28	10	8	0	10	28	109	22	65	0	90	25	115974	8	13	38	38	111	17	-70	0	-90	20	68743
MIL-71 X-Y USB-30	13	35	18	8	0	16	27	109	22	65	0	90	25	115676	8	13	51	44	111	17	-70	0	-90	20	67818
TEX-16 X-Y USB-30	13	31	11	8	1	26	57	110	22	65	0	90	25	112110	8	14	58	8	112	17	-71	0	-90	19	63006
GYM-14 X-Y USB-30	13	31	35	8	2	21	19	110	22	65	0	90	25	109303	8	15	52	55	113	16	-72	0	-90	18	58867
GDS-28 X-Y USB-85	13	59	38	8	2	29	43	110	21	64	0	-90	64	108866	8	16	29	21	114	15	-72	0	-90	-72	56021
STN-00 A-E USB150	14	7	41	8	2	46	23	110	21	63	0	90	27	107993	8	16	54	4	114	14	-72	0	-90	18	54042
HAW-12 X-Y USB-30	13	11	7	8	5	55	33	112	21	67	0	90	23	97708	8	19	6	40	116	12	-77	0	-90	13	42628
GWM-24 X-Y USB-30	13	3	30	8	10	2	44	114	20	70	0	90	20	83048	8	23	6	15	126	-3	-93	0	-90	-3	15757
HSK-25 X-Y USB-85	10	55	32	8	11	11	42	114	22	63	0	-90	63	78650	8	22	7	14	123	11	-76	0	-90	-76	23665
CRO-08 X-Y USB-30	11	8	37	8	13	14	7	116	20	68	0	90	22	70451	9	0	22	44	-45	-38	132	0	90	-42	1666

L-64

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
PKS-96 H-D USB210	4	9	28	8	14	22	19	116	21	30	30	41	48	63919	8	18	31	47	117	18	-37	30	-46	44	44082		
MAD-23 X-Y USB-85	2	15	56	8	20	55	46	130	7	81	0	-90	81	31847	8	23	11	42	149	-10	103	0	90	77	14945		
TAN-22 A-E FPS-16	2	23	14	8	21	50	13	133	11	50	45	37	27	23398	9	0	13	27	-126	-25	117	0	90	-27	3837		
PRE-76 A-E MPS-25	2	11	52	8	21	57	16	135	11	61	26	60	26	23400	9	0	9	8	-148	-15	107	0	90	-17	4792		
HSK-25 X-Y USB-85	0	25	50	9	0	0	48	140	-5	-96	0	90	-84	6523	9	0	26	38	-24	-9	101	0	90	79	749		

1-65

TABLE I.I-V.- MISSION RADAR TIMELINE - Continued

(b) LM acquisition and termination during descent - 0° minimum elevation

31 JAN 71 LM FROM UNDOCK TO LANDING

## VEHICLE RADAR TABLE

## TRACKING TIME

## STATION ACQUISITION DATA

## STATION TERMINATION DATA

HRS MIN SEC

DAY HRS MIN SEC

RA DEC AZ ELV X Y RANGE

DAY HRS MIN SEC

RA DEC AZ ELV X Y RANGE

GWM-24 X-Y USB-30	0 56 33	4 8 27 31	77 27 64 10 79 26 208890	4 9 24 4 78 28 65 22 68 23 208724
HAW-12 X-Y USB-30	0 56 43	4 8 27 31	76 28 72 63 26 8 206446	4 9 24 13 77 28 65 74 14 6 206704
GYM-14 X-Y USB-30	0 56 50	4 8 27 31	75 28 -84 72 -18 1 206243	4 9 24 21 76 28 -82 60 -30 4 207034
STN-00 A-E USB-150	0 56 51	4 8 27 31	76 27 -141 78 -8 -10 206155	4 9 24 22 77 -28 -111 -69 -20 -8 206812
GDS-28 X-Y USB-85	0 56 54	4 8 27 31	75 27 -119 76 7 -13 206182	4 9 24 24 76 28 -101 65 5 -24 206896
TEX-16 X-Y USB-30	0 56 57	4 8 27 31	75 28 -82 60 -30 4 206531	4 9 24 28 76 28 -78 48 -41 8 207450
GBM-41 X-Y USB-30	0 56 58	4 8 27 31	75 27 -76 43 -46 10 207163	4 9 24 29 76 28 -73 31 -57 15 208220
MIL-71 X-Y USB-30	0 56 59	4 8 27 31	75 27 -79 45 -44 8 207054	4 9 24 30 76 28 -74 34 -55 13 208088
ANG-91 X-Y USB-30	0 57 1	4 8 27 31	75 27 -67 25 -63 21 208035	4 9 24 32 76 28 -65 14 -75 24 209203
SDA-02 X-Y USB-30	0 57 1	4 8 27 31	75 27 -76 33 -57 12 207646	4 9 24 32 76 27 -71 22 -67 18 208731

## VEHICLE RADAR TABLE

	TRACKING TIME	STATION ACQUISITION DATA												STATION TERMINATION DATA											
		HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y
GUM-24 X-Y USA-30	1 5 6	4	10	12 33	78	28	65	32	55	21	208277			4	11	17 39	79	28	63	46 41	18	207688			
HSK-25 X-Y USA-85	1 5 5	4	10	12 33	78	28	47	7	-79	47	209670			4	11	17 38	79	29	36	15 -71	35	209223			
HAW-12 X-Y USA-30	1 4 57	4	10	12 55	77	28	29	84	3	6	204705			4	11	17 53	78	28	-59	78 -10	6	206818			
STN-00 A-E USA150	1 4 59	4	10	13 7	76	27	-99	59	-30	-4	207147			4	11	18 5	77	28	-88	47 -43	1	207651			
GDS-28 X-Y USA-85	1 4 52	4	10	13 12	76	27	-92	55	1	-35	207274			4	11	18 4	77	28	-83	43 -7	-47	207837			
GVM-14 X-Y USA-30	1 4 54	4	10	13 14	76	28	-79	49	-40	7	207487			4	11	18 8	77	28	-75	36 -53	12	208144			
BDA-02 X-Y USA-30	1 4 53	4	10	13 16	76	27	-65	12	-77	24	209356			4	11	18 10	77	27	-58	1 -89	32	210110			
GUM-41 X-Y USA-30	1 4 53	4	10	13 18	76	27	-69	21	-68	20	208852			4	11	18 14	77	27	-64	9 -80	26	209656	1167		
TEX-16 X-Y USA-30	1 4 51	4	10	13 19	76	27	-75	38	-51	12	207989			4	11	18 9	77	28	-71	25 -64	18	208734			
MIL-71 X-Y USA-30	1 4 50	4	10	13 21	76	27	-71	24	-65	18	208701			4	11	18 10	77	27	-65	11 -77	25	209493			

## VEHICLE RADAR TABLE

VEHICLE	RADAR TABLE	TRACKING TIME		STATION ACQUISITION DATA										STATION TERMINATION DATA												
		HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
ANG-91 X-Y USB-30		0	14	53	4	10	13	22	76	27	-62	3	-87	28	209897	4	10	28	15	76	27	-61	0	-90	29	209333
CRO-08 X-Y USB-30		0	35	20	4	12	6	41	79	28	53	9	79	36	209726	4	12	42	1	79	28	48	15	71	40	208218
HSK-25 X-Y USB-85		0	35	13	4	12	6	48	78	29	26	21	-67	24	209038	4	12	42	1	79	29	19	23	-66	17	207735
GWM-24 X-Y USB-30		0	35	12	4	12	6	49	78	28	59	57	29	16	207406	4	12	42	1	79	28	53	63	22	16	206043
HAW-12 X-Y USB-30		0	34	53	4	12	7	8	78	28	-71	67	-22	7	207097	4	12	42	1	78	28	-72	60	-29	9	206138
STN-00 A-E USB150		0	34	45	4	12	7	16	77	27	-82	37	-52	7	208162	4	12	42	1	78	27	-78	31	-58	11	207330
GDS-28 X-Y USB-85		0	34	44	4	12	7	17	77	27	-78	33	-18	-55	208386	4	12	42	1	78	27	-74	26	-29	-59	207575
GYM-14 X-Y USB-30		0	34	44	4	12	7	17	77	27	-71	26	-63	17	208755	4	12	42	1	78	27	-68	19	-70	20	207983
TEX-16 X-Y USB-30		0	34	42	4	12	7	19	77	27	-66	15	-74	23	209380	4	12	42	1	78	27	-63	8	-81	26	208618
MIL-71 X-Y USB-30		0	9	26	4	12	7	20	77	27	-60	2	-88	30	21n137	4	12	16	46	77	27	-59	0	-90	31	209758

100

TABLE 1.1-V.- MISSION RADAR TIMELINE - Continued

(c) LM acquisition and termination during ascent - 0° minimum elevation

31 JAN 71 LM ASCENT TO INSERTION

VEHICLE	RADAR TABLE	TRACKING TIME	STATION ACQUISITION DATA										STATION TERMINATION DATA												
			HRS MIN SEC			DAY HRS MIN SEC			RA DEC		AZ ELV		X	Y	RANGE	DAY HRS MIN SEC			RA DEC		AZ ELV		X	Y	RANGE
MAD-23 X-Y USB-05		0 7 11	5	22	24	29	99	27	95	50	4	40	209365	5	22	31	40	99	27	97	51	5	39	209401	
BDA-02 X-Y USB-30		0 7 11	5	22	24	29	100	26	50	1	89	31	21	913	5	22	31	40	100	26	60	2	87	30	211923
CVI-04 X-Y USB-30		0 7 11	5	22	24	29	99	27	77	39	50	10	209814	5	22	31	40	100	27	77	41	49	10	209831	
ACN-75 X-Y USB-30		0 7 11	5	22	24	29	100	27	55	25	40	32	21	526	5	22	31	40	100	27	54	26	59	32	210537
CRO-08 X-Y USB-30		0 7 11	5	22	24	29	98	27	-58	2	-87	32	21	842	5	22	31	40	98	27	-59	1	-87	31	212009

TABLE 1.1-V.- MISSION RADAR TIMELINE - Continued

(d) LM acquisition and termination during rendezvous - 0° minimum elevation

## COAST TO RENDEZVOUS RADAR TRACKING PRIOR TO TPI

## VEHICLE 2 RADAR TABLE

VEHICLE	2	RADAR TABLE	TRACKING TIME			STATION ACQUISITION DATA								STATION TERMINATION DATA												
			HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y
CRO-08 X-Y USB=30	n	4 37	5	22	31 40	98	27	-59	1 -89	31	211999	5	22	36 17	98	27	-59	0 -90	31	212218						
MAD-23 X-Y USB=85	0	9 60	5	22	31 40	99	27	97	51 5	39	209395	5	22	41 40	100	27	98	53 6	37	209751						
BDA-02 X-Y USB=30	0	9 60	5	22	31 40	100	26	60	2 87	30	211918	5	22	41 40	100	27	61 4	85 29	212236							
CYI-04 X-Y USB=30	0	9 60	5	22	31 40	100	27	77	41 49	10	209825	5	22	41 40	100	27	78 43	47 9	210156							
ACN-75 X-Y USB=30	0	9 60	5	22	31 40	100	27	54	26 59	32	210531	5	22	41 40	100	27	53 28	56 32	210854							

## RENDEZVOUS RADAR TRACKING PRIOR TO TPI

ACN-75 X-Y USB=30	0	11 21	5	22	41 40	100	27	53	28 56	32	210854	5	22	53 1	100	27	52 30	53 32	211358						
BDA-02 X-Y USB=30	0	11 22	5	22	41 40	100	27	61	4 85	29	212236	5	22	53 2	100	27	62 6	83 28	212732						
CYI-04 X-Y USB=30	0	11 23	5	22	41 40	100	27	78	43 47	9	210156	5	22	53 3	100	27	79 45	44 8	210671						
MAD-23 X-Y USB=85	0	11 27	5	22	41 40	100	27	98	53 6	37	209751	5	22	53 7	100	27	101 55	7 34	210299						
ANG-91 X-Y USB=30	0	6 6	5	22	46 53	100	27	62	0 90	28	212755	5	22	52 59	100	27	62 1	89 28	213016						

COAST TO TPI

ULLAGE FOR TPI

TPI

T<sub>170</sub>

## COAST TO 1ST BRAKING GATE

## VEHICLE 2 RADAR TABLE

VEHICLE	TRACKING TIME	STATION ACQUISITION DATA										STATION TERMINATION DATA												
		MRS	MIN	SEC	DAY	HRS	MIN	REC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X
ANG-91 X-Y USB-30	0 8 20	5	23	40	24	100	27	65	12	77	24	212586	5	23	48	44	100	27	66	13	75	24	212060	
BDA-02 X-Y USB-30	0 8 15	5	23	40	28	100	26	68	15	74	21	212369	5	23	48	44	100	26	69	17	72	20	211860	
MIL-71 X-Y USB-30	0 8 15	5	23	40	29	100	26	60	1	89	30	213206	5	23	48	44	100	26	61	3	87	29	212695	
GBM-41 X-Y USB-30	0 8 15	5	23	40	29	100	26	61	2	88	29	213146	5	23	48	44	100	26	62	4	86	28	212634	
ACN-75 X-Y USB-30	0 8 13	5	23	40	31	100	27	44	39	41	34	211093	5	23	48	44	100	27	43	41	38	34	210813	
CYT-04 X-Y USB-30	0 8 12	5	23	40	32	100	27	82	56	34	4	210444	5	23	48	44	100	27	83	57	32	4	209970	
MAD-23 X-Y USB-85	0 8 10	5	23	40	34	99	27	113	64	11	24	210200	5	23	48	44	99	27	116	65	11	22	209951	

## COAST TO 2ND BRAKING GATE

MAD-23 X-Y USB-85	0 1 49	5	23	48	44	99	27	116	65	11	22	209751	5	23	50	33	99	27	116	65	12	22	209651
MIL-71 X-Y USB-30	0 1 49	5	23	48	44	100	26	61	3	87	29	212695	5	23	50	33	100	26	61	3	87	29	212583
GBM-41 X-Y USB-30	0 1 49	5	23	48	44	100	26	62	4	86	28	212634	5	23	50	33	100	26	62	4	85	28	212521

## VEHICLE 2 RAUAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
BDA-02 X-Y USB=30	0	1	49	5	23	48	44	100	26	69	17	72	20	211860	5	23	50	33	100	26	69	17	72	20	211748
ANG-91 X-Y USB=30	0	1	49	5	23	48	44	100	27	66	13	75	24	212060	5	23	50	33	100	27	66	14	75	23	211946
GVI-04 X-Y USB=30	0	1	49	5	23	48	44	100	27	83	57	32	4	209970	5	23	50	33	100	27	83	58	32	4	209865
ACN-75 X-Y USB=30	0	1	49	5	23	48	44	100	27	43	41	38	34	210613	5	23	50	33	100	27	43	41	38	34	210507
	1ST BRAKING MANEUVER *																								
MAD-23 X-Y USB=65	0	0	12	5	23	50	33	99	27	116	6 <sup>5</sup>	12	22	209651	5	23	50	45	99	27	116	6 <sup>5</sup>	12	22	209640
MIL-71 X-Y USB=30	0	0	12	5	23	50	33	100	26	61	3	87	29	212583	5	23	50	45	100	26	61	3	87	29	212571
GBM-41 X-Y USB=30	0	0	12	5	23	50	33	100	26	62	4	85	28	212521	5	23	50	45	100	26	62	4	85	28	212508
BDA-02 X-Y USB=30	0	0	12	5	23	50	33	100	26	69	17	72	20	211748	5	23	50	45	100	26	69	17	72	20	211735
ANG-91 X-Y USB=30	0	0	12	5	23	50	33	100	27	66	14	75	23	211946	5	23	50	45	100	27	66	14	75	23	211933
GVI-04 X-Y USB=30	0	0	12	5	23	50	33	100	27	83	58	32	4	209865	5	23	50	45	100	27	83	58	32	4	209854

1-72

## VEHICLE 2 RADAR TABLE

	TRACKING TIME HRS MIN SEC	STATION ACQUISITION DATA										STATION TERMINATION DATA											
		DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
ACN-75 X-Y USB=30	0 0 12	5	23	50	33	100	27	43	41	38	34	210507	5	23	50	45	100	27	43	41	38	34	210496
COAST TO 3RD BRAKING GATE																							
MAD-23 X-Y USB=85	0 1 1	5	23	50	45	99	27	116	65	12	22	209640	5	23	51	46	99	27	117	66	12	22	209585
MIL-71 X-Y USB=30	0 1 1	5	23	50	45	100	26	61	3	87	29	212571	5	23	51	46	100	26	62	3	86	28	212509
GBM-41 X-Y USB=30	0 1 1	5	23	50	45	100	26	62	4	85	28	212508	5	23	51	46	100	26	62	4	85	27	212446
MDA-02 X-Y USB=30	0 1 1	5	23	50	45	100	26	69	17	72	20	211735	5	23	51	46	100	26	69	17	71	20	211673
ANG-91 X-Y USB=30	0 1 1	5	23	50	45	100	27	66	14	75	23	211933	5	23	51	46	100	27	66	14	75	23	211876
GYI-04 X-Y USB=30	0 1 1	5	23	50	45	100	27	83	58	32	4	209854	5	23	51	46	100	27	83	58	32	4	209796
ACN-75 X-Y USB=30	0 1 1	5	23	50	45	100	27	43	41	38	34	210496	5	23	51	46	100	27	42	41	38	34	210437
2ND BRAKING MANEUVER *																							
MAD-23 X-Y USB=85	0 0 9	5	23	51	46	99	27	117	66	12	22	209585	5	23	51	54	99	27	117	66	12	22	209577
MIL-71 X-Y USB=30	0 0 9	5	23	51	46	100	26	62	3	86	28	212509	5	23	51	54	100	26	62	3	86	28	212500

1173

## VEHICLE 2 RADAR TABLE

	TRACKING TIME				STATION ACQUISITION DATA								STATION TERMINATION DATA													
	HRS	MIN	SEC		DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
GBM-41 X-Y USB=30	0	0	9		5	23	51	46	100	26	62	4	85	27	212446	5	23	51	54	100	26	62	4	85	27	212437
BDA-02 X-Y USB=30	0	0	9		5	23	51	46	100	26	69	17	71	20	211673	5	23	51	54	100	26	69	17	71	20	211665
ANG-91 X-Y USB=30	0	0	9		5	23	51	46	100	27	66	14	75	23	211870	5	23	51	54	100	27	66	14	75	23	211861
CYI-04 X-Y USB=30	0	0	9		5	23	51	46	100	27	83	58	32	4	209796	5	23	51	54	100	27	83	58	32	4	209788
ACN-75 X-Y USB=30	0	0	9		5	23	51	46	100	27	42	41	38	34	210437	5	23	51	54	100	27	42	41	38	34	210429
CCAST TC 4TH BRAKING GATE																										
MAD-23 X-Y USB=85	0	1	30		5	23	51	54	99	27	117	66	12	22	209577	5	23	53	24	100	27	117	66	12	21	209498
MIL-71 X-Y USB=30	0	1	30		5	23	51	54	100	26	62	3	86	28	212500	5	23	53	24	100	26	62	3	86	28	212410
GBM-41 X-Y USB=30	0	1	30		5	23	51	54	100	26	62	4	85	27	212437	5	23	53	24	100	26	63	5	85	27	212347
BDA-02 X-Y USB=30	0	1	30		5	23	51	54	100	26	69	17	71	20	211665	5	23	53	24	100	26	69	18	71	20	211575
ANG-91 X-Y USB=30	0	1	30		5	23	51	54	100	27	66	14	75	23	211861	5	23	53	24	100	27	66	14	74	23	211770

## VEHICLE 2 RADAR TABLE

	TRACKING TIME	STATION ACQUISITION DATA										STATION TERMINATION DATA												
		HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X
GVI-04 X-Y USB=30	0 1 30	5 23 51 54	100 27	83 58 32	4	209788	5 23 53 24	100 27	83 58 31	4	209704													
ACN-73 X-Y USB=30	0 1 30	5 23 51 54	100 27	42 41 38	34	210429	5 23 53 24	100 27	42 41 37	34	210345													
3RD BRAKING MANEUVER *																								
MAD-23 X-Y USB=85	0 0 4	5 23 53 24	100 27	117 66 12	21	209498	5 23 53 28	100 27	117 66 12	21	209494													
MIL-71 X-Y USB=30	0 0 4	5 23 53 24	100 26	62 3 86 28	212410		5 23 53 28	100 26	62 3 86 28	212406														
GBM-41 X-Y USB=30	0 0 4	5 23 53 24	100 26	63 5 85 27	212347		5 23 53 28	100 26	63 5 85 27	212342														
BDA-D2 X-Y USB=30	0 0 4	5 23 53 24	100 26	69 18 71 20	211575		5 23 53 28	100 26	69 18 71 20	211571														
ANG-91 X-Y USB=30	0 0 4	5 23 53 24	100 27	66 14 74 23	211770		5 23 53 28	100 27	66 14 74 23	211765														
GVI-04 X-Y USB=30	0 0 4	5 23 53 24	100 27	83 58 31 4	209704		5 23 53 28	100 27	83 58 31 4	209701														
ACN-73 X-Y USB=30	0 0 4	5 23 53 24	100 27	42 41 37 34	210345		5 23 53 28	100 27	42 41 37 34	210341														
CCAST TO 5TH BRAKING GATE																								
MAD-23 X-Y USB=85	0 1 15	5 23 53 28	100 27	117 66 12 21	209494		5 23 54 44	100 27	118 66 12 21	209429														

11-1751

## VEHICLE 2 RAUAR TABLE

	TRACKING TIME				STATION ACQUISITION DATA								STATION TERMINATION DATA													
	HRS	MIN	SEC		DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
MIL-71 X-Y USB=30	0	1	15		5	23	53	28	100	26	62	3	86	28	212406	5	23	54	44	100	26	62	4	86	28	212332
GBM-41 X-Y USB=30	0	1	15		5	23	53	28	100	26	63	5	85	27	212342	5	23	54	44	100	26	63	5	85	27	212268
BDA-02 X-Y USB=30	0	1	15		5	23	53	28	100	26	69	18	71	20	211571	5	23	54	44	100	26	70	18	71	19	211498
ANG-91 X-Y USB=30	0	1	15		5	23	53	28	100	27	66	14	74	23	211765	5	23	54	44	100	27	66	15	74	23	211690
CYI-04 X-Y USB=30	0	1	15		5	23	53	28	100	27	83	58	31	4	209701	5	23	54	44	100	27	83	59	31	3	209632
ACN-75 X-Y USB=30	0	1	15		5	23	53	28	100	27	42	41	37	34	210341	5	23	54	44	100	27	42	42	37	34	210272
4TH BRAKING MANEUVER																										
MAD-23 X-Y USB=85	0	0	4		5	23	54	44	100	27	118	66	12	21	209429	5	23	54	48	100	27	118	66	12	21	209425
MIL-71 X-Y USB=30	0	0	4		5	23	54	44	100	26	62	4	86	28	212332	5	23	54	48	100	26	62	4	86	28	212328
GBM-41 X-Y USB=30	0	0	4		5	23	54	44	100	26	63	5	85	27	212268	5	23	54	48	100	26	63	5	85	27	212264
BDA-02 X-Y USB=30	0	0	4		5	23	54	44	100	26	70	18	71	19	211498	5	23	54	48	100	26	70	18	71	19	211494

1  
76

## VEHICLE 2 RADAR TABLE

VEHICLE	TRACKING TIME	STATION ACQUISITION DATA												STATION TERMINATION DATA											
		HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y
ANG-91 X-Y USB=30	0 0 4	5 23 54 44 100 27	66 15 74 23 211690	5 23 54 48 100 27	66 15 74 23 211686																				
GVI-04 X-Y USB=30	0 0 4	5 23 54 44 100 27	83 59 31 3 209632	5 23 54 48 100 27	83 59 31 3 209629																				
ACN-75 X-Y USB=30	0 0 4	5 23 54 44 100 27	42 42 37 34 210272	5 23 54 48 100 27	42 42 37 34 210268																				
COAST TO DOCKING																									
MAD-23 X-Y USB=85	0 4 22	5 23 54 48 100 27	118 66 12 21 209425	5 23 59 10 100 27	119 67 12 20 209216																				
MIL-71 X-Y USB=30	0 4 22	5 23 54 48 100 26	62 4 86 28 212328	5 23 59 10 100 26	62 5 85 28 212087																				
GBM-41 X-Y USB=30	0 4 22	5 23 54 48 100 26	63 5 85 27 212264	5 23 59 10 100 26	63 6 84 27 212023																				
BDA-02 X-Y USB=30	0 4 22	5 23 54 48 100 26	70 18 71 19 211494	5 23 59 10 100 26	70 19 70 19 211254																				
ANG-91 X-Y USB=30	0 4 22	5 23 54 48 100 27	66 15 74 23 211686	5 23 59 10 100 27	66 15 73 23 211641																				
GVI-04 X-Y USB=30	0 4 22	5 23 54 48 100 27	83 59 31 3 209629	5 23 59 10 100 27	84 60 30 3 209408																				
ACN-75 X-Y USB=30	0 4 22	5 23 54 48 100 27	42 42 37 34 210268	5 23 59 10 100 27	41 42 36 34 210045																				

TABLE 1.1-V.- MISSION RADAR TIMELINE - Continued

(e) CSM acquisition and termination during rendezvous - 3° minimum elevation

VEHICLE	1	RADAR TABLE												LAZ 72.1												
		TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA									
		HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
BDA-02 X-Y USB-30		0	0	25	0	0	11	40	102	7	85	5	85	5	592	0	0	12	5	104	6	85	3	87	5	689
SHIP 1 INSERTION		0	3	35	0	0	11	40	-76	52	-38	13	-70	51	379	0	0	15	15	129	22	67	3	87	23	687
CYI-04 X-Y USB-30		0	5	44	0	0	17	15	-31	16	-73	3	-87	17	686	0	0	22	59	141	-19	113	3	87	-23	680
TAN-22 A-E FPS-15		0	3	41	0	0	38	1	112	67	-8	3	-69	82	664	0	0	41	42	-145	13	75	3	87	15	666
CRO-08 X-Y USB-30		0	4	2	0	0	53	11	95	-38	-131	3	-86	-41	673	0	0	57	13	-46	-43	137	3	86	-47	673
HSK-25 X-Y USB-85		0	4	22	0	1	0	25	164	15	-69	3	-82	-69	671	0	1	4	47	-73	40	33	3	-86	33	667
GYM-14 X-Y USB-30		0	5	40	0	1	29	1	-85	-24	-120	3	-87	-30	680	0	1	34	41	82	13	77	3	87	13	686
GDS-28 X-Y USB-85		0	1	58	0	1	30	21	-12	-51	176	3	87	4	679	0	1	32	19	39	-33	135	3	86	45	682
TEX-16 X-Y USB-30		0	5	38	0	1	31	54	-85	-1	-93	3	-87	-3	684	0	1	37	33	102	24	64	3	87	26	688
MIL-71 X-Y USB-30		0	5	25	0	1	35	50	-76	15	-74	3	-87	15	686	0	1	41	14	120	23	65	3	87	24	689

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HR	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HR	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
GBM-41 X-Y USB-30	0	4	55	0	1	36	36	-79	27	-62	3	-87	28	688	0	1	41	31	127	32	55	3	86	34	688
BDA-02 X-Y USB-30	0	5	47	0	1	39	13	-54	6	-85	3	-87	5	689	0	1	44	60	122	-3	95	3	87	-5	687
SHIP 1 INSERTION	0	5	40	0	1	42	54	-47	23	-66	3	-87	24	688	0	1	48	35	139	-2	94	3	87	-4	684
CYI-04 X-Y USB-30	0	3	15	0	1	51	10	13	-20	-115	3	-87	-25	681	0	1	54	24	97	-59	175	3	57	-84	676
TAN-22 A-E FPS-16	0	5	26	0	2	9	54	72	11	-77	3	-87	13	666	0	2	15	20	-102	-38	130	3	86	-40	668
CRO-08 X-Y USB-30	0	4	30	0	2	26	13	116	-41	-135	3	-86	-45	673	0	2	30	42	-35	-24	115	4	86	-25	639

31 JAN 71 TLI TO NFR IGNITION LAZ 72.1

HAW-12 X-Y USB-30	12	31	19	0	2	43	2	-110	-39	-134	3	-86	-44	2271	0	15	14	22	45	30	-59	3	-86	31	74337
STN-00 A-E USB150	10	32	22	0	2	50	7	-77	-17	-114	3	-87	-24	3835	0	13	22	29	44	29	-55	3	-86	35	67031
GDS-28 X-Y USB-85	10	0	39	0	2	50	59	-76	-13	-108	3	80	-72	4021	0	12	51	38	43	29	-55	3	-85	-55	64922
GYM-14 X-Y USB-30	9	7	24	0	2	51	58	-76	-5	-97	3	-87	-7	4235	0	11	59	23	41	30	-57	3	-86	32	61242

L-79

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
TEX-16 X-Y USB-30	8	2	48	0	2	55	53	-66	3	-89	3	-87	1	5058	0	10	58	41	40	30	-57	3	-86	32	56786
MIL-71 X-Y USB-30	6	42	17	0	3	2	34	-51	10	-80	3	-87	10	6409	0	9	44	51	37	30	-57	3	-86	33	51057
GBM-41 X-Y USB-30	6	22	47	0	3	4	1	-49	12	-78	3	-87	12	6692	0	9	26	48	36	30	-58	3	-86	32	49597
GTI-51 A-E TPQ-18	2	0	23	0	3	9	20	-42	18	-72	3	-87	18	7706	0	5	9	43	16	32	=64	31	=56	22	23400
BDA-02 X-Y USB-30	5	30	30	0	3	11	24	-37	14	-75	3	-87	15	8090	0	8	41	55	35	30	-56	3	-86	34	45854
ANG-91 X-Y USB-30	4	12	20	0	3	19	22	-30	23	-67	3	-87	23	9517	0	7	31	42	30	31	-58	3	-86	32	39613
SHIP 1 INSERTION	1	33	16	0	3	29	23	-20	22	-67	3	-87	23	11220	0	5	2	40	14	30	-64	15	-73	25	23392
GWM-24 X-Y USB-30	13	34	15	0	5	11	48	33	31	59	3	86	31	25171	0	18	46	3	48	30	-60	3	-87	30	86950
HSK-25 X-Y USB-85	7	45	45	0	8	11	11	41	36	41	3	-86	41	43172	0	15	56	56	47	33	-45	3	-86	-45	76982
CRO-08 X-Y USB-30	9	20	48	0	9	51	57	45	34	50	3	86	40	51614	0	19	12	45	49	32	-53	3	-86	37	88439

2  
11  
10  
9  
8  
7  
6  
5  
4  
3  
2

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA										
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y
MAD-23 X-Y USB-85	15	17	60	0	14	59	28	50	29	53	3 -85	53	73394	1	6	17	27	55	28	-55	3 -85	-55	120767	
CVI-04 X-Y USB-30	13	52	13	0	16	31	28	52	30	58	3	86	32	79098	1	6	23	41	54	29	-59	3 -86	31	121031
ACN-75 X-Y USB-30	10	55	8	0	17	57	36	53	31	58	3	86	32	84181	1	4	52	44	54	30	-59	3 -87	31	117077
BDA-02 X-Y USB-30	10	54	24	0	19	41	42	53	29	57	3	86	33	90052	1	6	36	7	55	29	-78	42	-47	9 119460
ANG-91 X-Y USB-30	10	23	48	0	20	12	19	54	30	60	3	87	30	91718	1	6	36	7	54	29	-66	35	-53	20 119803
GBM-41 X-Y USB-30	9	40	58	0	20	55	8	54	29	59	3	86	31	94016	1	6	36	7	55	29	-76	52	-37	9 119039
MIL-71 X-Y USB-30	9	36	13	0	20	59	53	54	29	58	3	86	32	94269	1	6	36	7	55	29	-79	55	-34	6 118949
TEX-16 X-Y USB-30	8	24	58	0	22	11	8	55	29	59	3	86	31	97990	1	6	36	7	55	29	-80	70	-20	3 118551
GYM-14 X-Y USB-30	7	30	52	0	23	5	15	55	29	59	3	86	31	100740	1	6	36	7	56	29	-78	82	-8	2 118379
GDS-28 X-Y USB-85	7	28	17	0	23	7	50	55	29	56	3 -85	56	100871	1	6	36	7	56	29	-158	83	6 -3	118367	

T-180

VEHICLE	RADAR TABLE																									
	TRACKING TIME						STATION ACQUISITION DATA										STATION TERMINATION DATA									
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	
STN-00 A-E USB150	7	13	28	0	23	22	39	55	29	56	3	86	34	101612	1	6	36	7	56	29	165	81	2	-8	118382	
HAW-12 X-Y USB-30	3	54	21	1	2	41	46	56	29	60	3	87	30	111170	1	6	36	7	57	29	70	52	36	12	119039	
ANG-91 X-Y USB=30	2	28	1	1	6	36	18	54	29	-66	34	-53	20	119813	LAZ	72.1										
BDA-02 X-Y USB-30	3	19	50	1	6	36	18	55	29	-78	42	-47	9	119469	1	9	56	8	56	28	-58	3	-86	32	129657	
GBM-41 X-Y USB-30	3	58	59	1	6	36	18	55	29	-76	52	-37	9	119048	1	10	35	17	56	28	-60	3	-87	30	131180	
MIL-71 X-Y USB-30	4	13	54	1	6	36	18	55	29	-79	55	-34	6	118958	1	10	50	12	56	28	-59	3	-86	31	131755	
TEX-16 X-Y USB-30	5	19	19	1	6	36	18	55	29	-80	70	-20	3	118559	1	11	55	36	56	28	-60	3	-87	30	134247	
GYM-14 X-Y USB-30	6	14	11	1	6	36	18	56	29	-78	82	-8	2	118387	1	12	50	29	56	28	-60	3	-86	30	136298	
GDS-28 X-Y USB-85	7	0	41	1	6	36	18	56	29	-157	83	6	-3	118375	1	13	36	59	57	28	-57	3	-84	-57	138011	
STN-00 A-E USB150	7	28	52	1	6	36	18	56	29	165	81	2	-8	118390	1	14	5	10	57	28	-57	3	-86	33	139037	

T  
28

## VEHICLE 1 RADAR TABLE

	TRACKING TIME												STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE											
HAW-12 X-Y USB-30	9	17	31	1	6	36	18	57	29	70	53	36	12	119045	1	15	53	49	57	28	-61	3	-87	29	142906											
GWM-24 X-Y USB-30	12	28	18	1	6	50	1	58	29	61	3	87	29	122139	1	19	18	20	58	28	-62	3	-87	28	149867											
HSK-25 X-Y USB-85	8	5	19	1	8	46	26	58	30	49	3	-85	49	126886	1	16	51	44	58	30	-50	3	-85	-50	144913											
CRO-08 X-Y USB-30	9	22	57	1	10	29	54	59	30	55	3	86	35	130963	1	19	52	51	58	29	-56	3	-86	34	150997											
MAD-23 X-Y USB-85	14	51	1	1	15	39	50	60	28	55	3	-85	55	142418	2	6	30	51	60	27	-56	3	-85	-56	170151											
CYI-04 X-Y USB-30	13	33	50	1	17	5	31	60	28	60	3	87	30	145396	2	6	39	21	60	27	-61	3	-87	29	170383											
ACN-75 X-Y USB-30	10	52	25	1	18	22	10	60	29	60	3	87	30	147993	2	5	14	35	60	28	-61	3	-87	29	168024											
BDA-02 X-Y USB-30	13	57	24	1	20	11	21	61	28	59	3	86	31	151609	2	10	8	45	61	27	-60	3	-87	30	175978											
ANG-91 X-Y USB-30	12	41	5	1	20	38	24	61	28	62	3	87	28	152484	2	9	19	29	60	27	-62	3	-87	28	174686											
GBM-41 X-Y USB-30	13	26	16	1	21	21	58	61	28	60	3	87	30	153883	2	10	48	13	61	27	-61	3	-87	29	176998											

1  
83

## VEHICLE 1 RADAR TABLE

## TRACKING TIME

## STATION ACQUISITION DATA

## STATION TERMINATION DATA

HRS MIN SEC

DAY HRS MIN SEC RA DEC AZ ELV X Y RANGE

DAY HRS MIN SEC RA DEC AZ ELV X Y RANGE

MIL-71 X-Y USB-30	13 35 59	1 21 26 55 61 28 60 3 87 30 154042	2 11 2 54 61 27 -61 3 -87 29 177377
TEX-16 X-Y USB-30	13 31 12	1 22 36 35 61 28 60 3 87 30 156241	2 12 7 47 61 27 -61 3 -87 29 179027
GYM-14 X-Y USB-30	13 32 26	1 23 29 44 61 28 60 3 87 30 157890	2 13 2 10 61 27 -61 3 -87 29 180391
GDS-28 X-Y USB-85	14 13 19	1 23 33 40 61 27 58 3 -84 58 158012	2 13 46 59 61 27 -59 3 -84 -59 181500
STN-00 A-E USB150	14 25 56	1 23 48 41 61 27 57 3 86 33 158473	2 14 14 37 61 27 -58 3 -86 32 182179
HAW-12 X-Y USB-30	13 2 48	2 3 1 54 62 28 61 3 87 29 164238	2 16 4 42 61 27 -62 3 -87 28 184833
GWM-24 X-Y USB-30	12 23 2	2 7 5 57 62 28 62 3 87 28 171108	2 19 28 59 62 27 -63 3 -87 27 189628
HSK-25 X-Y USB-85	8 17 35	2 8 52 37 62 29 51 3 -85 51 173969	2 17 10 13 62 28 -52 3 -85 -52 186371
CRO-08 X-Y USB-30	9 29 35	2 10 38 26 63 28 57 3 86 33 176737	2 20 8 0 62 28 -57 3 -86 33 190539
MAD-23 X-Y USB-85	14 37 9	2 15 55 19 63 27 57 3 -84 57 184611	3 6 32 29 63 26 -58 3 -84 -58 203745

T  
1  
84  
41

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA										
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y
CVI-04 X-Y USB-30	13	24	46	2	17	18	22	64	27	61	3	87	29	186572	3	6	43	8	63	26	-62	3	-87	28 203978
ACN-75 X-Y USB-30	10	52	28	2	18	30	41	64	27	62	3	87	28	188282	3	5	23	9	63	27	-62	3	-87	28 202351
BDA-02 X-Y USB-30	13	43	43	2	20	23	59	64	27	60	3	87	30	190901	3	10	7	42	62	26	-61	3	-87	29 208183
ANG-91 X-Y USB-30	12	33	21	2	20	48	31	64	27	63	3	87	27	191388	3	9	21	51	62	26	-63	3	-87	26 207255
GBM-41 X-Y USB-30	12	51	57	2	21	32	48	64	27	62	3	87	28	192376	3	10	24	45	62	26	-65	8	-82	25 208163
MIL-71 X-Y USB-30	12	46	13	2	21	38	33	64	27	61	3	87	29	192556	3	10	24	46	62	26	-66	10	-79	24 207999
TEX-16 X-Y USB-30	11	37	32	2	22	47	12	64	27	61	3	87	29	194008	3	10	24	45	62	26	-72	24	-65	17 207228
GYM-14 X-Y USB-30	10	44	15	2	23	40	28	64	27	61	3	87	29	195194	3	10	24	43	62	26	-76	35	-54	11 206630
GDS-28 X-Y USB-85	10	39	42	2	23	45	0	64	26	59	3	-84	59	195270	3	10	24	42	62	26	-85	42	-6	-48 206330
STN-00 A-E USB150	10	24	33	3	0	0	7	64	26	59	3	87	31	195593	3	10	24	40	62	26	-90	46	-44	0 206141

1185

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA										
	HRS	MIN	SEC	DAY	HR S	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HR S	MIN	SEC	RA	DEC	AZ	ELV	X	Y
HAW-12 X-Y USB-30	7	13	58	3	3	10	35	64	27	63	3	87	27	199619	3	10	24	33	63	26	-67	78	-11	5 205263
GWM-24 X-Y USB-30	3	12	36	3	7	11	43	65	26	64	3	87	26	204567	3	10	24	20	64	26	65	47	41	17 206107
HSK-25 X-Y USB-85	1	34	2	3	8	50	18	64	27	53	3	-85	53	206579	3	10	24	21	64	27	36	17	-69	35 207589
	31 JAN 71			LOI TO DOI IGNITION										LAZ	72.1									
CRO-08 X-Y USB-30	1	24	20	3	10	56	52	64	27	56	7	81	34	207538	3	12	21	12	65	27	44	21	61	42 206942
HSK-25 X-Y USB-85	1	24	24	3	10	56	57	63	27	30	21	-66	27	206753	3	12	21	20	64	27	11	26	-63	10 206664
GWM-24 X-Y USB-30	1	24	24	3	10	56	60	63	26	63	54	33	15	205203	3	12	21	24	64	27	45	70	14	14 204971
HAW-12 X-Y USB-30	1	24	28	3	10	57	15	62	26	-73	70	-19	5	204747	3	12	21	43	63	27	-75	52	=37	9 205490
STN-00 A-E USB150	1	24	27	3	10	57	24	62	26	-85	39	-51	4	205787	3	12	21	52	63	26	-74	24	-65	14 206823
GDS-28 X-Y USB-85	1	24	28	3	10	57	24	62	26	-81	35	-13	-54	206002	3	12	21	52	63	26	-71	19	-43	-63 207086
GYM-14 X-Y USB-30	1	24	27	3	10	57	26	62	26	-74	28	-61	14	206347	3	12	21	53	63	26	-66	12	-77	23 207524

12  
11  
10  
9  
8  
7  
6  
5  
4  
3  
2

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
MIL-71 X-Y USB-30	0	2	47	3	10	57	27	62	26	-62	4	-86	28	207745	3	11	0	14	62	26	-62	3	-87	28	207642		
TEX-16 X-Y USB-30	1	13	16	3	10	57	27	62	26	-69	17	-72	20	206973	3	12	10	43	63	26	-62	3	-87	28	207470		
CRO-08 X-Y USB-30	1	24	29	3	13	5	1	65	27	36	28	48	46	206592	3	14	29	29	66	28	17	35	22	51	206411		
HSK-25 X-Y USB-85	1	24	30	3	13	5	7	64	27	0	27	-63	0	206609	3	14	29	36	65	28	-20	24	-65	-18	207006		
GWM-24 X-Y USB-30	1	24	29	3	13	5	11	64	27	16	76	4	13	204864	3	14	29	40	65	27	-45	70	-15	14	205208		
HAW-12 X-Y USB-30	1	24	28	3	13	5	23	63	26	-74	42	-46	12	205859	3	14	29	51	64	27	-70	25	-64	18	206979		
STN-00 A-E USB150	1	16	6	3	13	5	28	63	26	-69	16	-73	20	207250	3	14	21	34	65	26	-59	3	-86	31	207827		
GDS-28 X-Y USB-05	0	46	21	3	13	5	29	63	26	-66	11	-65	-64	207527	3	13	51	50	64	26	-60	3	-84	-60	206648		
31 JAN 71												DOI TO LM SEP IGNITION				LAZ	72.1										
CRO-08 X-Y USB-30	1	5	6	3	15	15	7	65	27	4	38	6	52	206200	3	16	20	13	66	28	-13	36	-17	52	206361		
HSK-25 X-Y USB-85	1	5	5	3	15	15	15	65	27	-30	20	-67	-28	207096	3	16	20	20	66	28	-42	13	-73	-41	207638		

T

87

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
GWM-24 X-Y USB-30	1	5	6	3	15	15	20	65	27	-58	61	-25	15	205301	3	16	20	26	66	27	-64	48	-39	17	205880		
HAW-12 X-Y USB-30	1	1	20	3	15	15	29	64	26	-67	15	-74	22	207397	3	16	16	49	66	27	-63	3	-87	27	208032		
MAD-23 X-Y USB-85	0	11	19	3	16	8	49	57	26	57	3	-84	57	207602	3	16	20	8	67	26	59	5	-81	59	208111		
MAD-23 X-Y USB-85	1	4	56	3	17	8	41	67	26	66	13	-60	63	207730	3	18	13	37	68	27	75	24	-31	62	207164		
CRO-08 X-Y USB-30	1	4	53	3	17	8	60	66	28	-25	33	-34	49	206619	3	18	13	52	67	28	-38	25	-53	45	207101		
HSK-25 X-Y USB-85	0	16	46	3	17	9	8	66	27	-50	6	-81	-50	208130	3	17	25	54	66	27	-53	3	-85	-53	207458		
GWM-24 X-Y USB-30	1	4	48	3	17	9	15	66	27	-66	37	-51	19	206424	3	18	14	3	67	27	-66	23	-65	22	207235		
CYI-04 X-Y USB-30	0	41	20	3	17	32	15	68	26	62	3	87	28	207258	3	18	13	35	69	27	65	11	78	24	207923		
ACN-75 X-Y USB-30	1	4	43	3	19	2	37	69	27	61	7	82	28	208265	3	20	7	20	70	27	57	20	67	31	207583		
CYI-04 X-Y USB-30	1	4	42	3	19	2	44	69	27	70	21	68	19	207450	3	20	7	26	70	27	75	34	55	13	206842		

12  
11  
10  
9  
8  
7  
6  
5  
4  
3  
2  
1

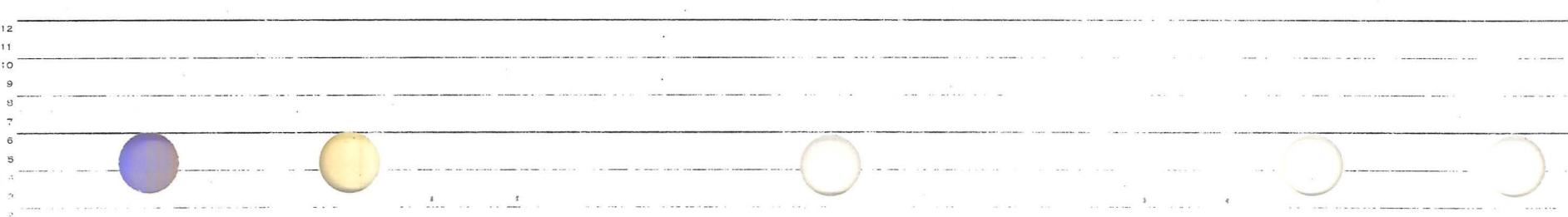
## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
MAD-23 X-Y USB-85	1	4	48	3	19	2	47	68	26	82	33	-12	56	206787	3	20	7	34	69	27	91	45	1	45	206331		
CRO-08 X-Y USB-30	1	4	55	3	19	3	5	67	28	-47	18	-66	41	207598	3	20	7	60	68	28	-55	7	-82	35	208354		
GWM-24 X-Y USB-30	0	44	48	3	19	3	14	67	27	-65	12	-77	24	207923	3	19	48	3	68	27	-63	3	-87	27	207577		
ACN-75 X-Y USB=30	1	5	2	3	20	56	20	70	27	52	30	54	32	207157	3	22	1	22	71	28	42	41	38	34	206700		
BDA-02 X-Y USB-30	1	5	1	3	20	56	23	70	27	62	5	84	28	208530	3	22	1	24	71	27	69	17	72	20	207921		
CYI-04 X-Y USB-30	1	5	1	3	20	56	23	70	27	78	44	45	8	206462	3	22	1	24	71	27	82	58	32	4	206044		
MAD-23 X-Y USB-85	1	5	3	3	20	56	28	69	27	100	54	7	35	206073	3	22	1	31	70	27	115	66	11	22	205829		
ANG-91 X-Y USB-30	0	53	0	3	21	8	21	70	27	63	3	87	27	208063	3	22	1	22	71	27	65	14	-75	24	208125		
GBM-41 X-Y USB-30	0	5	29	3	21	55	55	71	27	61	3	87	29	208453	3	22	1	24	71	27	62	4	85	28	208695		
MIL-71 X-Y USB-30	0	0	9	3	22	1	14	71	27	61	3	87	29	208749	3	22	1	23	71	27	61	3	87	29	208755		

1180

VEHICLE	1	RADAR TABLE	STATION ACQUISITION DATA												STATION TERMINATION DATA												
		TRACKING TIME	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
GBM-41 X-Y USB-30	1	4 57		3 22 49 56	71 27	67 14	75 23	208231				3 23	54 53	72 27	71 27	62 17	207561										
ANG-91 X-Y USB-30	1	4 54		3 22 49 58	71 27	67 24	64 20	207635				3 23	54 52	72 27	68 38	50 17	206992										
MIL-71 X-Y USB-30	1	4 57		3 22 49 58	71 27	66 13	76 23	208302				3 23	54 54	72 27	71 25	64 17	207641										
BDA-02 X-Y USB-30	1	4 55		3 22 50 0	71 27	74 27	62 14	207493				3 23	54 55	72 27	80 40	50 8	206915										
ACN-75 X-Y USB-30	1	4 57		3 22 50 2	70 28	31 48	25 35	206495				3 23	54 59	71 28	11 53	8 36	206351										
CYI-04 X-Y USB-30	1	4 52		3 22 50 10	70 27	86 69	21 1	205857				3 23	55 2	71 27	91 82	8 0	205719										
MAD-23 X-Y USB-85	1	4 53		3 22 50 16	70 27	139 73	13 11	205765				3 23	55 9	71 27	-169 77	13 -3	205786										
TEX-16 X-Y USB-30	0	43 43		3 23 11 11	71 27	61 3	87 29	207874				3 23	54 54	72 27	65 11	78 24	208439										
TEX-16 X-Y USB-30	1	4 41		4 0 44 3	72 27	70 21	68 19	207973				4 1	48 45	73 27	74 34	55 13	207359										
GYM-14 X-Y USB-30	1	4 38		4 0 44 4	72 27	65 10	79 25	208596				4 1	48 42	73 27	70 23	66 18	207946										

190-T



## VEHICLE 1 RADAR TABLE

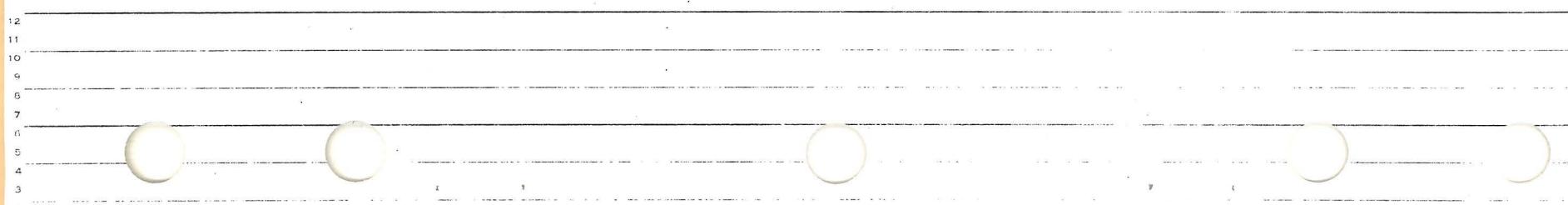
	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
GDS-28 X-Y USB-85	1	4	44	4	0	44	5	72	27	63	9	-71	62	208685	4	1	48	49	73	27	71	20	-42	62	208090		
MIL-71 X-Y USB-30	1	4	43	4	0	44	6	72	27	76	36	54	12	207216	4	1	48	49	73	27	80	49	41	7	206708		
GBM-41 X-Y USB-30	1	4	42	4	0	44	6	72	27	75	37	52	12	207137	4	1	48	48	73	27	78	51	39	7	206637		
ANG-91 X-Y USB-30	1	4	42	4	0	44	7	72	27	68	49	39	14	206627	4	1	48	48	73	28	63	62	25	12	206254		
BDA-02 X-Y USB-30	1	4	45	4	0	44	7	72	27	85	50	40	3	206587	4	1	48	52	73	27	93	63	27	-1	206236		
STN-00 A-E USB150	1	4	41	4	0	44	8	72	27	61	6	83	29	208850	4	1	48	49	73	27	69	17	72	20	208277		
ACN-75 X-Y USB-30	1	4	51	4	0	44	14	71	28	-8	54	-6	36	206435	4	1	49	6	72	28	-28	49	-22	35	206708		
CYI-04 X-Y USB-30	1	4	53	4	0	44	21	71	27	-98	87	-3	0	205787	4	1	49	14	72	27	-87	73	-17	1	206032		
MAD-23 X-Y USB-85	1	4	51	4	0	44	23	71	27	-132	71	13	-14	205955	4	1	49	14	72	27	-108	61	10	-27	206296		
GYM-14 X-Y USB-30	1	5	3	4	2	37	42	73	27	74	33	56	13	207521	4	3	42	45	74	27	78	46	43	8	206992		

1-91

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
GDS-28 X=Y USB-85	1	5	0	4	2	37	43	73	27	76	30	-22	57	207686	4	3	42	43	74	27	83	42	-7	47	207164
STN-00 A-E USB150	1	5	1	4	2	37	44	73	27	75	26	63	13	207878	4	3	42	44	74	27	82	38	52	6	207349
TEX-16 X-Y USR-30	1	5	0	4	2	37	45	73	27	78	45	45	8	206986	4	3	42	45	74	28	82	58	32	4	206567
GBM-41 X-Y USB-30	1	5	3	4	2	37	46	73	27	81	62	28	4	206389	4	3	42	49	74	28	83	75	15	2	206170
MIL-71 X=Y USB-30	1	5	2	4	2	37	47	73	27	84	60	30	3	206444	4	3	42	49	74	28	89	73	17	0	206204
ANG-91 X-Y USB-30	1	5	5	4	2	37	48	72	28	51	72	14	11	206135	4	3	42	53	73	28	0	79	0	11	206118
BDA-02 X=Y USB-30	1	5	1	4	2	37	50	72	27	103	73	16	-4	206115	4	3	42	51	73	28	152	85	3	-5	206073
ACN-75 X-Y USB-30	1	5	4	4	2	37	58	72	28	-40	42	-36	34	207084	4	3	43	3	73	28	-50	31	-52	33	207705
CYI-04 X-Y USB-30	1	5	2	4	2	38	2	72	27	-83	62	-28	3	206360	4	3	43	4	73	27	-79	49	-41	7	206912
MAD-23 X-Y USB-85	1	5	3	4	2	38	4	72	27	-97	52	5	-38	206677	4	3	43	7	73	27	-87	41	-4	-49	207255

26



## VEHICLE 1 RADAR TABLE

## TRACKING TIME

## STATION ACQUISITION DATA

## STATION TERMINATION DATA

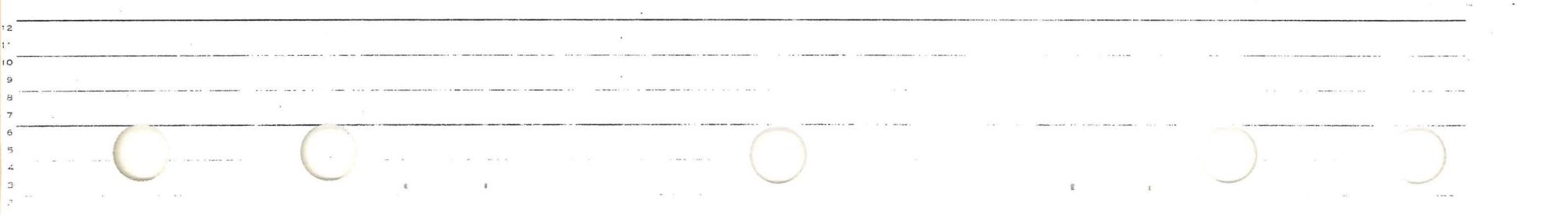
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
NAM-12 X-Y USB-30	1	4	54	4	4	31	19	74	27	66	13	76	24	208832	4	5	36	13	75	27	69	26	63	19	208135
GDS-28 X-Y USB-85	1	4	54	4	4	31	27	74	27	90	52	0	38	206873	4	5	36	21	75	27	100	65	5	25	206538
GYM-14 X-Y USB-30	1	4	53	4	4	31	29	74	27	82	57	33	4	206707	4	5	36	22	75	28	86	71	19	1	206409
STN-00 A-E USB150	1	4	53	4	4	31	29	74	27	89	48	42	0	207035	4	5	36	22	75	27	100	60	29	-5	206666
TEX-16 X-Y USB-30	1	4	57	4	4	31	31	74	27	85	69	21	2	206386	4	5	36	28	75	28	88	83	7	0	206250
GBM-41 X-Y USB-30	1	4	52	4	4	31	40	73	27	77	86	4	1	206161	4	5	36	32	74	28	-81	80	-10	1	206276
MIL-71 X-Y USB-30	1	4	51	4	4	31	41	73	27	99	84	6	-1	206171	4	5	36	32	74	28	-94	82	-8	-1	206255
BDA-02 X-Y USB-30	1	4	53	4	4	31	43	73	27	-119	80	-9	-5	206199	4	5	36	36	74	28	-96	68	-22	-2	206476
ANG-91 X-Y USB-30	1	4	49	4	4	31	46	73	28	-45	74	-11	11	206274	4	5	36	35	74	28	-62	62	-25	13	206620
ACN-75 X-Y USB-30	1	4	47	4	4	31	59	72	28	-56	21	-65	31	208300	4	5	36	46	74	28	-60	9	-80	30	209137

1-93

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
MAD-23 X-Y USB-85	1	4	46	4	4	31	60	73	27	-80	31	-17	-57	207776	4	5	36	45	74	27	-71	20	-41	-63	208469		
CYI-04 X-Y USB-30	1	4	46	4	4	32	1	73	27	-76	38	-51	11	207446	4	5	36	47	74	27	-71	25	-64	17	208198		
HAW-12 X-Y USB-30	1	4	44	4	6	25	24	75	27	71	37	52	15	207703	4	7	30	8	76	28	73	50	39	11	207180		
GYM-14 X-Y USB-30	1	4	55	4	6	25	34	75	28	91	82	8	0	206358	4	7	30	28	76	28	-91	85	-5	0	206419		
GDS-28 X-Y USB-85	1	4	53	4	6	25	35	75	27	117	75	7	14	206443	4	7	30	28	76	28	178	82	8	0	206435		
STN-00 A-E USB150	1	4	49	4	6	25	35	75	27	114	70	19	-8	206531	4	7	30	24	76	28	152	79	5	-10	206463		
TEX-16 X-Y USB-30	1	4	51	4	6	25	40	74	28	-91	86	-4	0	206324	4	7	30	31	75	28	-85	73	-17	1	206560		
MIL-71 X-Y USB-30	1	4	55	4	6	25	43	74	27	-88	71	-19	1	206493	4	7	30	38	75	28	-83	58	-32	4	206931		
GBM-41 X-Y USB-30	1	5	0	4	6	25	44	74	28	-82	69	-21	3	206540	4	7	30	45	75	28	-79	55	-34	6	207018		
BDA-02 X-Y USB-30	1	4	57	4	6	25	48	74	27	-89	57	-33	1	206848	4	7	30	45	75	28	-82	45	-45	6	207434		

476 T



## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
ANG-91 X-Y USB-30	1	4	58	4	6	25	49	74	28	-67	51	-36	14	207055	4	7	30	47	75	28	-68	38	-50	17	207737		
MAD-23 X-Y USB-85	0	55	37	4	6	25	53	74	27	-64	11	-65	-62	209046	4	7	21	30	75	27	-56	3	-85	-56	209155		
CYI-04 X-Y USB-30	1	4	21	4	6	25	56	74	27	-67	15	-74	22	208842	4	7	30	18	75	27	-61	3	-87	29	209633		
GWN-24 X-Y USB-30	0	8	33	4	8	18	58	77	27	64	8	81	26	209437	4	8	27	31	77	27	64	10	79	26	208890		
HAW-12 X-Y USB-30	0	8	25	4	8	19	6	76	28	73	61	28	8	206930	4	8	27	31	76	28	72	63	26	8	206446		
STN-00 A-E USB150	0	8	16	4	8	19	14	76	27	-148	79	-6	-10	206569	4	8	27	31	76	27	-141	78	-8	-10	206155		
GDS-28 X-Y USB-85	0	8	15	4	8	19	15	75	27	-124	77	7	-11	206587	4	8	27	31	75	27	-119	76	7	-13	206182		
GYN-14 X-Y USB-30	0	8	13	4	8	19	17	75	28	-87	74	-16	1	206635	4	8	27	31	75	28	-86	72	-18	1	206243		
TEX-16 X-Y USB-30	0	8	11	4	8	19	20	75	28	-83	62	-28	3	206901	4	8	27	31	75	28	-82	60	-30	4	206531		
MIL-71 X-Y USB-30	0	8	8	4	8	19	23	75	27	-79	47	-42	7	207402	4	8	27	31	75	27	-79	45	-44	8	207054		

1-95

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
GBM-41 X-Y USB-30	0	8	5		4	8	19	26	75	27	-77	45	-45	9	207505		4	8	27	31	75	27	-76	43	-46	10	207163
BDA-02 X-Y USB-30	0	8	5		4	8	19	26	75	27	-77	34	-55	11	207981		4	8	27	31	75	27	-76	33	-57	12	207646
ANG-91 X-Y USB-30	0	8	3		4	8	19	28	75	28	-67	27	-61	20	208356		4	8	27	31	75	27	-67	25	-63	21	208035
				31 JAN 71				LM SEP TO CIR IGNITION				LAZ		72.1													
GWM-24 X-Y USB-30	0	56	28		4	8	27	37	77	27	64	10	79	26	208883		4	9	24	5	78	28	65	22	66	23	208726
HAW-12 X-Y USB-30	0	56	36		4	8	27	37	76	28	72	63	26	8	206441		4	9	24	13	77	28	65	74	14	6	206705
STN-00 A-E USB150	0	56	44		4	8	27	37	76	27	-141	78	-8	-10	206149		4	9	24	21	77	28	-111	69	-20	-8	206812
GDS-28 X-Y USB-85	0	56	47		4	8	27	37	75	27	-119	76	7	-13	206177		4	9	24	24	76	28	-101	65	5	-24	206897
GYM-14 X-Y USB-30	0	56	49		4	8	27	37	75	28	-86	72	-18	1	206238		4	9	24	26	76	28	-82	60	-30	4	207039
TEX-16 X-Y USB-30	0	56	51		4	8	27	37	75	28	-82	60	-30	4	206527		4	9	24	28	76	28	-78	48	-41	8	207451
MIL-71 X-Y USB-30	0	56	54		4	8	27	37	75	27	-79	45	-44	8	207050		4	9	24	31	76	28	-74	34	-55	13	208091

T-1

96

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
GBM-41 X-Y USB-30	0	56	54	4	8	27	37	75	27	-76	43	-46	10	207159	4	9	24	31	76	28	-73	31	-57	15	208223
ANG-91 X-Y USB-30	0	56	55	4	8	27	37	75	27	-67	25	-63	21	208030	4	9	24	32	76	28	-65	14	-75	24	209205
BDA-02 X-Y USB-30	0	56	57	4	8	27	37	75	27	-76	33	-57	12	207642	4	9	24	34	76	27	-71	22	-67	18	208733
				31 JAN 71				CIR TO LOPC1 IGNITION						LAZ 72.1											
GWM-24 X-Y USB-30	1	11	46	4	10	11	42	78	28	65	32	55	21	208327	4	11	23	28	79	28	63	47	40	18	207716
HSK-25 X-Y USB-85	1	11	46	4	10	11	42	77	28	47	7	-79	47	209718	4	11	23	28	79	29	35	16	-71	33	209265
HAW-12 X-Y USB-30	1	11	54	4	10	11	56	77	28	30	83	3	6	206755	4	11	23	49	78	28	-62	77	-12	6	206919
GDS-28 X-Y USB-85	1	11	59	4	10	12	4	76	27	-92	55	1	-35	207324	4	11	24	3	77	28	-83	42	-8	-48	207977
STN-00 A-E USB150	1	11	54	4	10	12	5	76	27	-99	60	-30	-5	207193	4	11	23	59	77	28	-88	46	-44	2	207781
GYM-14 X-Y USB-30	1	11	57	4	10	12	6	76	28	-79	50	-40	7	207536	4	11	24	3	77	28	-74	35	-54	13	208289
TEX-16 X-Y USB-30	1	11	56	4	10	12	10	76	27	-75	38	-51	11	208036	4	11	24	6	77	28	-70	24	-65	18	208885

T

L

7

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
BDA-02 X-Y USB-30	0	54	42	4	10	12	12	76	27	-66	12	-76	24	209399	4	11	6	54	77	27	-59	3	-86	31	209181		
GBM-41 X-Y USB-30	1	11	55	4	10	12	13	76	27	-69	21	-67	19	208896	4	11	24	8	77	27	-63	7	-82	27	209812		
MIL-71 X-Y USB-30	1	11	51	4	10	12	13	76	27	-71	24	-65	18	208746	4	11	24	5	77	27	-64	10	-79	25	209644		
ANG-91 X-Y USB-30	0	1	2	4	10	12	14	76	27	-62	3	-86	28	209941	4	10	13	16	76	27	-62	3	-87	28	209906		
CRO-08 X-Y USB-30	1	11	42	4	12	10	1	79	28	53	10	78	37	209762	4	13	21	43	80	29	42	21	61	44	209180		
HSK-25 X-Y USB-85	1	11	42	4	12	10	9	78	29	25	21	-67	24	209093	4	13	21	51	79	29	10	25	-65	9	208947		
GWM-24 X-Y USB-30	1	11	40	4	12	10	12	79	28	59	57	29	16	207453	4	13	21	52	80	28	40	70	13	15	207197		
HAW-12 X-Y USB-30	1	11	40	4	12	10	29	78	28	-71	66	-22	7	207184	4	13	22	10	79	28	-73	52	-37	11	207744		
GDS-28 X-Y USB-85	1	11	37	4	12	10	37	77	27	-77	32	-19	-56	208489	4	13	22	13	78	27	-69	19	-45	-62	209305		
STN-00 A-E USB150	1	11	35	4	12	10	38	77	27	-81	37	-53	7	208262	4	13	22	13	78	27	-73	24	-65	16	209041		

T-98

## VEHICLE 1 RADAR TABLE

## TRACKING TIME

## STATION ACQUISITION DATA

## STATION TERMINATION DATA

HRS MIN SEC

DAY HRS MIN SEC RA DEC AZ ELV X Y RANGE

DAY HRS MIN SEC RA DEC AZ ELV X Y RANGE

GYM-14 1 11 34 4 12 10 39 77 27 -71 25 -64 17 208861 4 13 22 12 78 27 -65 11 -77 25 209750  
X-Y USB-30

TEX-16 0 59 28 4 12 10 40 77 27 -66 14 -75 23 209489 4 13 10 8 78 27 -60 3 -87 29 209601  
X-Y USB-30

CRD-08 1 11 54 4 14 8 26 80 28 34 27 47 48 208921 4 15 20 20 81 29 18 34 25 52 208662  
X-Y USB-30

HSK-25 1 11 57 4 14 8 34 79 29 -2 26 -64 -2 209002 4 15 20 31 80 29 -18 23 -66 -17 209231  
X-Y USB-85

GWM-24 1 10 10 4 14 8 35 79 28 8 75 2 15 207201 4 15 18 45 80 28 -40 70 -13 15 207293  
X-Y USB-30

HAW-12 1 11 60 4 14 8 48 78 28 -72 41 -48 14 208247 4 15 20 48 80 28 -69 26 -62 19 209099  
X-Y USB-30

STN-00 1 11 58 4 14 8 51 78 27 -67 15 -74 22 209608 4 15 20 50 80 27 -58 3 -86 32 210428  
A-E USB150

GDS-28 0 43 41 4 14 8 51 78 27 -64 10 -68 -62 209886 4 14 52 32 79 27 -58 3 -84 -58 209143  
X-Y USB-85

CRD-08 1 11 40 4 16 6 57 80 29 5 36 7 53 208641 4 17 18 37 81 29 -13 35 -18 53 208779  
X-Y USB-30

HSK-25 1 11 41 4 16 7 4 80 29 -29 20 -68 -27 209514 4 17 18 45 81 29 -42 12 -75 -40 210068  
X-Y USB-85

T

66

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
GWM-24 X-Y USB-30	1	11	38		4	16	7	11	80	28	-55	61	-24	16	207673		4	17	18	49	81	28	-63	47	-40	18	208272
HAW-12 X-Y USB-30	1	7	46		4	16	7	21	79	28	-66	16	-73	23	209710		4	17	15	7	81	28	-61	3	-87	29	210397
MAD-23 X-Y USB-85	0	18	11		4	17	0	17	82	27	56	3	-85	56	209639		4	17	18	28	82	27	59	6	-79	58	210412
MAD-23 X-Y USB-85	1	11	54		4	18	5	7	82	27	66	14	-59	62	210051		4	19	17	1	84	27	75	26	-28	60	209451
CRO-08 X-Y USB-30	1	11	59		4	18	5	20	81	29	-25	32	-34	50	209034		4	19	17	19	82	29	-39	24	-55	45	209576
HSK-25 X-Y USB-85	0	12	6		4	18	5	24	81	28	-50	5	-83	-49	210547		4	18	17	30	81	28	-51	3	-85	-51	210051
GWM-24 X-Y USB-30	1	11	59		4	18	5	28	81	28	-65	36	-51	20	208804		4	19	17	26	82	28	-65	22	-66	23	209698
CYI-04 X-Y USB-30	0	51	29		4	18	25	31	83	27	61	3	87	29	209709		4	19	16	60	84	28	65	12	76	24	210203
ACN-75 X-Y USB-30	1	11	42		4	20	3	25	84	28	60	8	81	30	210583		4	21	15	7	85	28	55	22	64	32	209832
CYI-04 X-Y USB-30	1	11	39		4	20	3	31	84	28	70	22	67	19	209745		4	21	15	10	85	28	75	36	53	12	209075

OUT

100

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
MAD-23 X-Y USB-85	1	11	42	4	20	3	34	83	27	82	35	-12	55	209087	4	21	15	15	85	28	92	47	1	43	208585
CRO-08 X-Y USB-30	1	11	43	4	20	3	54	82	28	-47	16	-68	41	210040	4	21	15	37	83	28	-56	4	-85	34	210854
GWM-24 X-Y USB-30	0	40	36	4	20	3	59	82	28	-64	11	-78	26	210341	4	20	44	35	82	28	-62	3	-87	28	209605
ACN-75 X-Y USB-30	0	8	4	4	22	1	36	85	28	50	31	52	33	209444	4	22	9	40	85	28	49	32	50	34	208962
BDA-02 X-Y USB-30	0	8	2	4	22	1	38	85	27	62	8	82	27	210763	4	22	9	40	85	27	63	9	80	26	210272
CYI-04 X-Y USB-30	0	7	58	4	22	1	42	85	28	78	46	43	8	208727	4	22	9	40	85	28	79	48	41	8	208258
MAD-23 X-Y USB-85	0	7	55	4	22	1	45	84	28	101	56	7	33	208357	4	22	9	40	84	28	102	58	8	31	207912
ANG-91 X-Y USB-30	0	6	28	4	22	3	12	85	28	62	3	87	28	210959	4	22	9	40	85	28	63	4	85	27	210548
TOTAL																									
31 JAN 71      LOPC1 TO LM JETT IGNITION      LAZ 72.1																									
ANG-91 X-Y USB-30	1	3	40	4	22	9	59	85	28	63	4	85	27	210528	4	23	13	39	86	28	65	17	71	23	210263
BDA-02 X-Y USB-30	1	3	46	4	22	9	59	85	27	63	9	80	26	210253	4	23	13	44	86	28	70	21	68	19	210082

12  
11  
10  
9  
8  
7  
6  
5

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
ACN-75 X-Y USB-30	1	3	47		4	22	9	59	85	28	49	33	50	34	208944		4	23	13	45	86	28	39	43	34	35	208960
CYI-04 X-Y USB-30	1	3	50		4	22	9	59	85	28	79	48	41	8	208240		4	23	13	48	86	28	82	61	29	4	208292
MAD-23 X-Y USB-85	1	3	52		4	22	9	59	84	28	102	58	8	31	207895		4	23	13	50	85	28	120	69	11	18	208109
GBM-41 X-Y USB-30	0	23	41		4	22	50	1	86	27	61	3	87	29	209948		4	23	13	42	86	28	63	7	82	27	210835
MIL-71 X-Y USB-30	0	18	17		4	22	55	27	86	27	60	3	87	30	210162		4	23	13	44	86	27	62	6	83	28	210908
ANG-91 X-Y USB-30	1	11	37		5	0	0	13	86	28	67	27	61	20	209790		5	1	11	50	87	28	67	42	45	16	209131
MIL-71 X-Y USB-30	1	11	36		5	0	0	14	86	27	67	16	73	22	210446		5	1	11	50	87	28	73	30	59	15	209750
GBM-41 X-Y USB-30	1	11	31		5	0	0	17	86	27	67	17	72	22	210370		5	1	11	47	87	28	72	31	58	15	209670
BDA-02 X-Y USB-30	1	11	33		5	0	0	18	86	27	75	30	59	13	209652		5	1	11	51	87	28	81	44	46	6	209062
ACN-75 X-Y USB-30	1	11	37		5	0	0	21	85	28	27	49	21	36	208771		5	1	11	58	86	28	4	54	3	36	208690

1102

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
CYI-04 X-Y USB-30	1	11	37	5	0	0	25	85	28	85	72	18	2	208117	5	1	12	2	86	28	87	87	3	0	208038
MAD-23 X-Y USB-85	1	11	38	5	0	0	28	85	28	147	75	13	8	208056	5	1	12	5	86	28	-150	76	13	-7	208141
TEX-16 X-Y USB-30	1	6	39	5	0	5	8	86	27	61	3	87	29	210947	5	1	11	47	87	28	67	15	73	23	210530
GYM-14 X-Y USB-30	0	9	52	5	1	1	54	87	27	60	3	87	30	210725	5	1	11	47	87	27	61	5	85	29	211149
GDS-28 X-Y USB-85	0	5	42	5	1	6	8	87	27	58	3	-84	58	210954	5	1	11	50	87	27	59	4	-82	59	211205
GYM-14 X-Y USB-30	1	12	1	5	1	58	23	87	27	66	14	75	23	210708	5	3	10	24	88	28	72	28	61	16	210006
TEX-16 X-Y USB-30	1	11	59	5	1	58	23	87	27	71	25	64	17	210093	5	3	10	22	88	28	76	39	50	11	209443
GSM-41 X-Y USB-30	1	12	1	5	1	58	25	87	28	76	41	48	11	209287	5	3	10	26	88	28	79	56	33	6	208784
MIL-71 X-Y USB-30	1	12	0	5	1	58	25	87	28	76	40	50	10	209362	5	3	10	26	88	28	81	54	35	5	208847
GDS-28 X-Y USB-85	1	11	58	5	1	58	26	87	27	65	12	-62	62	210807	5	3	10	24	88	27	73	25	-32	60	210156

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
STN-00 A-E USB150	1	11	58	5	1	58	26	87	27	63	9	79	27	210981	5	3	10	24	88	27	71	22	67	17	210346		
BDA-02 X-Y USB=30	1	12	2	5	1	58	27	87	28	86	54	36	2	208776	5	3	10	29	88	28	96	68	22	-2	208447		
ANG-91 X-Y USB-30	1	11	60	5	1	58	28	87	28	66	53	35	14	208814	5	3	10	27	88	28	57	67	19	12	208470		
ACN-75 X-Y USB-30	1	12	1	5	1	58	37	86	28	-14	53	-11	36	208813	5	3	10	38	87	28	-34	45	-29	35	209186		
CYI-04 X-Y USB-30	1	12	2	5	1	58	37	86	28	-88	83	-7	0	208146	5	3	10	39	87	28	-84	68	-22	2	208471		
MAD-23 X-Y USB-85	1	12	0	5	1	58	40	86	28	-122	69	11	-18	208338	5	3	10	40	87	28	-101	57	7	-32	208760		
STN-00 A-E USB150	1	11	40	5	3	56	55	88	27	77	31	59	11	209949	5	5	8	34	89	28	86	44	46	3	209391		
GDS-28 X-Y USB=85	1	11	38	5	3	56	55	88	27	79	35	-16	54	209758	5	5	8	34	89	28	87	48	-3	42	209218		
GYM-14 X-Y USB-30	1	11	42	5	3	56	56	88	28	76	38	51	11	209589	5	5	8	38	89	28	80	53	37	6	209054		
TEX-16 X-Y USB-30	1	11	41	5	3	56	58	88	28	79	50	40	7	209093	5	5	8	38	89	28	83	65	25	3	208691		

NOTE

## VEHICLE 1 RADAR TABLE

	TRACKING TIME										STATION ACQUISITION DATA						STATION TERMINATION DATA								
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
GBM-41 X-Y USB-30	1	11	41	5	3	57	1	88	28	81	67	23	3	208563	5	5	8	42	89	28	80	82	8	2	208401
MIL-71 X-Y USB-30	1	11	44	5	3	57	1	88	28	85	65	25	2	208611	5	5	8	45	89	28	91	80	10	0	208423
BDA-02 X-Y USB-30	1	11	44	5	3	57	4	88	28	110	78	11	-4	208353	5	5	8	48	89	28	-139	84	-4	-5	208387
ANG-91 X-Y USB-30	1	11	45	5	3	57	6	88	28	37	76	9	11	208381	5	5	8	51	89	28	-31	77	-7	11	208456
ACN-75 X-Y USB-30	1	11	45	5	3	57	16	87	28	-44	38	-42	34	209572	5	5	9	1	88	28	-54	25	-60	32	210323
CYL-04 X-Y USB-30	1	11	46	5	3	57	17	87	28	-81	57	-33	5	208814	5	5	9	3	88	28	-77	42	-47	10	209487
MAD-23 X-Y USB-85	1	11	43	5	3	57	20	87	27	-92	48	2	-42	209138	5	5	9	3	88	27	-82	35	-11	-54	209819
HAW-12 X-Y USB-30	0	28	59	5	4	39	33	89	27	62	3	87	28	210303	5	5	8	32	90	28	64	9	80	26	211257
HAW-12 X-Y USB-30	1	11	53	5	5	55	5	90	27	67	19	70	22	210790	5	7	6	58	91	28	71	33	55	16	210069
STN-00 A-E US8150	1	11	54	5	5	55	11	89	27	93	53	36	-2	209125	5	7	7	5	90	28	108	67	22	-7	208806

L-10\5

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
GYM-14 X-Y USB-30	1	11	55		5	5	55	12	89	28	83	63	27	3	208824		5	7	7	7	90	28	88	78	12	0	208606
GDS-28 X-Y USB-85	1	11	53		5	5	55	12	89	28	94	58	2	32	208975		5	7	7	5	90	28	109	72	6	17	208705
TEX-16 X-Y USB-30	1	11	55		5	5	55	17	89	28	86	75	15	1	208569		5	7	7	12	90	28	-40	90	0	0	208537
MIL-71 X-Y USB-30	1	11	53		5	5	55	24	89	28	-175	89	0	-1	208449		5	7	7	17	90	28	-89	75	-15	0	208658
GBM-41 X-Y USB-30	1	11	52		5	5	55	25	89	28	-64	87	-2	1	208450		5	7	7	17	90	28	-82	73	-17	2	208697
BDA-02 X-Y USB-30	1	11	52		5	5	55	28	88	28	-104	75	-14	-4	208560		5	7	7	20	89	28	-90	61	-29	0	208974
ANG-91 X-Y USB-30	1	11	52		5	5	55	28	88	28	-55	69	-17	12	208667		5	7	7	21	89	28	-66	55	-33	14	209164
CYI-04 X-Y USB-30	1	11	48		5	5	55	40	88	27	-73	32	-57	14	210023		5	7	7	28	89	27	-68	18	-70	21	210895
MAD-23 X-Y USB-85	1	11	47		5	5	55	41	88	27	-76	26	-27	-60	210329		5	7	7	27	89	27	-66	14	-58	-62	211119
ACN-75 X-Y USB-30	1	3	32		5	5	55	42	88	28	-58	16	-72	31	210917		5	6	59	14	89	28	-61	3	-87	29	211341

L-106

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
HAW-12 X-Y USB-30	1	11	47	5	7	53	37	91	28	72	44	45	13	209670	5	9	5	25	92	28	73	59	30	9	209187
STN-00 A-E USB150	1	11	47	5	7	53	49	90	27	128	75	12	-9	208717	5	9	5	35	91	28	-160	80	-4	-10	208754
GDS-28 X-Y USB-85	1	11	51	5	7	53	49	90	28	138	80	8	7	208658	5	9	5	39	91	28	-132	79	7	-8	208765
GYM-14 X-Y USB-30	1	11	56	5	7	53	50	90	28	103	89	1	0	208604	5	9	5	46	91	28	-87	76	-14	1	208806
TEX-16 X-Y USB-30	1	11	54	5	7	53	53	90	28	-87	79	-11	1	208661	5	9	5	47	91	28	-83	64	-26	3	209045
GBM-41 X=Y USB-30	1	11	55	5	7	53	57	89	28	-81	62	-28	4	208997	5	9	5	52	90	28	-77	47	-42	9	209623
MIL-71 X-Y USB-30	1	11	54	5	7	53	57	89	28	-85	64	-26	2	208932	5	9	5	51	90	28	-80	50	-40	7	209522
BDA-02 X-Y USB-30	1	11	56	5	7	53	58	89	27	-85	51	-39	3	209365	5	9	5	55	90	27	-78	37	-53	10	210090
ANG-91 X-Y USB-30	1	11	58	5	7	54	1	89	28	-68	44	-43	16	209618	5	9	5	59	90	28	-67	29	-59	20	210454
HAD-23 X-Y USB-85	0	20	60	5	7	54	2	89	27	-59	6	-78	-59	211640	5	8	15	2	89	27	-56	3	-85	-56	210837

1-107

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
CYI-04 X-Y USB-30	0	30	24	5	7	54	4	89	27	-64	9	-80	26	211497	5	8	24	28	89	27	-61	3	-87	29	210595		
GWM-24 X-Y USB-30	0	13	34	5	8	51	43	92	28	62	3	87	28	211203	5	9	5	18	92	28	63	6	83	27	211751		
GWM-24 X-Y USB-30	1	11	45	5	9	51	50	92	27	65	16	72	24	211246	5	11	3	35	93	28	66	31	57	21	210494		
HAW-12 X-Y USB-30	1	11	46	5	9	52	3	91	28	70	69	20	7	208997	5	11	3	49	92	28	39	83	5	6	208883		
STN-00 A-E USB150	1	11	44	5	9	52	15	91	27	-122	73	-14	-9	208906	5	11	3	58	92	27	-100	61	-29	-5	209296		
GDS-28 X-Y USB-85	1	11	45	5	9	52	15	91	27	-108	70	6	-19	208962	5	11	4	1	92	27	-93	57	2	-33	209423		
GYM-14 X-Y USB-30	1	11	44	5	9	52	18	91	28	-84	66	-24	2	209062	5	11	4	3	92	28	-80	51	-39	6	209629		
TEX-16 X-Y USB-30	1	11	44	5	9	52	20	90	28	-80	54	-36	6	209416	5	11	4	4	91	27	-76	39	-50	11	210123		
GBM-41 X-Y USB-30	1	11	44	5	9	52	24	90	27	-74	37	-52	12	210123	5	11	4	8	91	27	-70	22	-66	19	210979		
MIL-71 X-Y USB-30	1	11	41	5	9	52	24	90	27	-77	39	-50	10	210001	5	11	4	6	91	27	-71	25	-64	17	210830		

1-108

## VEHICLE 1 RADAR TABLE

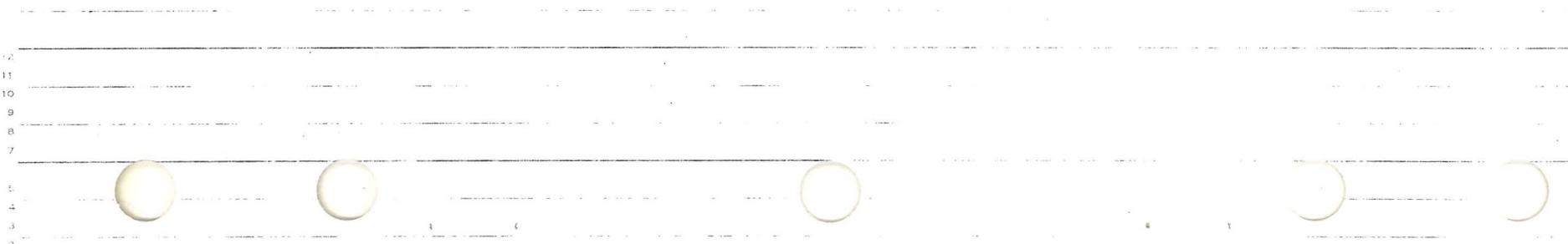
	TRACKING TIME												STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE											
BDA-02 X-Y USB-30	1	11	40	5	9	52	26	90	27	-73	27	-62	15	210623	5	11	4	6	91	27	-66	14	-75	23	211483											
ANG-91 X-Y USB-30	1	11	38	5	9	52	29	90	27	-66	19	-69	23	211054	5	11	4	7	91	27	-63	4	-85	27	212018											
HSK-25 X-Y USB-85	0	22	51	5	10	40	46	93	28	52	3	-85	52	210953	5	11	3	36	93	28	48	6	-81	48	211871											
HSK-25 X-Y USB-85	1	11	55	5	11	50	18	93	28	40	13	-73	39	211559	5	13	2	13	94	28	27	21	-67	25	211207											
GWM-24 X-Y USB-30	1	11	54	5	11	50	18	93	28	65	42	46	19	210079	5	13	2	11	94	28	60	56	30	16	209586											
HAW-12 X-Y USB-30	1	12	4	5	11	50	30	92	28	-47	82	-6	6	208960	5	13	2	34	93	28	-71	68	-21	7	209289											
STN-00 A-E USB150	1	12	1	5	11	50	37	92	27	-92	51	-39	-1	209668	5	13	2	38	93	27	-82	38	-52	6	210356											
GDS-28 X-Y USB-85	1	11	60	5	11	50	39	91	27	-86	47	-3	-43	209835	5	13	2	39	93	27	-78	33	-17	-55	210580											
TEX-16 X-Y USB-30	1	12	1	5	11	50	42	91	27	-72	29	-60	15	210673	5	13	2	43	93	27	-67	15	-74	22	211575											
GYM-14 X-Y USB-30	1	11	60	5	11	50	42	91	27	-77	41	-49	10	210107	5	13	2	42	93	27	-71	26	-63	17	210948											

16011

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA										
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y
BDA-02 X-Y USB-30	0	8	39	5	11	50	43	91	27	-61	5 -85	29	212061	5	11	59	21	91	27	-60	3 -87	30	211716	
GBM-41 X-Y USB-30	0	51	48	5	11	50	43	91	27	-66	13 -76	24	211584	5	12	42	32	92	27	-61	3 -87	29	211249	
MIL-71 X-Y USB-30	1	7	13	5	11	50	45	91	27	-67	15 -73	22	211423	5	12	57	57	92	27	-61	3 -87	29	212028	
CRO-08 X-Y USB-30	0	32	30	5	12	29	35	94	28	57	3 86	33	210897	5	13	2	5	94	28	53	9 79	36	211898	
CRO-08 X-Y USB-30	1	11	40	5	13	48	40	94	28	47	17 67	41	211500	5	15	0	20	95	28	35	27 48	47	211015	
GWM-24 X-Y USB-30	1	11	42	5	13	48	48	94	28	50	66 19	15	209384	5	15	0	30	95	28	13	75 4	14	209283	
HSK-25 X-Y USB-85	1	11	42	5	13	48	48	94	28	16	24 -65	14	211081	5	15	0	30	95	28	0	26 -64	0	211066	
HAW-12 X-Y USB-30	1	11	42	5	13	49	4	93	27	-73	57 -32	9	209614	5	15	0	46	94	27	-72	42 -46	13	210298	
GDS-28 X-Y USB-85	1	11	39	5	13	49	11	92	27	-73	24 -34	-61	211101	5	15	0	50	94	27	-65	11 -65	-63	211946	
TEX-16 X-Y USB-30	0	13	1	5	13	49	12	92	27	-62	5 -84	28	212158	5	14	2	13	93	27	-61	3 -87	29	211646	

OTTI



## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
STN-00 A-E USB150	1	11	38	5	13	49	12	93	27	-76	29	-61	12	210849	5	15	0	49	94	27	-68	16	-73	21	211669		
GYM-14 X-Y USB-30	1	11	38	5	13	49	12	92	27	-68	16	-72	22	211520	5	15	0	50	94	27	-61	3	-87	29	212425		
CRO-08 X-Y USB-30	1	12	2	5	15	46	60	95	28	25	32	33	50	210818	5	16	59	2	96	28	7	37	9	53	210702		
HSK-25 X-Y USB-85	1	12	1	5	15	47	7	94	28	-12	25	-64	-11	211189	5	16	59	8	95	28	-28	20	-67	-26	211548		
GWM-24 X-Y USB-30	1	12	0	5	15	47	9	94	28	-27	74	-7	14	209376	5	16	59	9	95	28	-55	62	-23	15	209727		
HAW-12 X-Y USB-30	1	11	59	5	15	47	18	94	27	-71	32	-57	16	210846	5	16	59	17	95	27	-67	17	-71	22	211754		
STN-00 A-E USB150	0	26	15	5	15	47	20	94	27	-62	7	-82	28	212216	5	16	13	35	94	27	-58	3	-86	32	211312		
CRO-08 X-Y USB-30	1	11	37	5	17	45	41	96	28	-6	37	-8	53	210743	5	18	57	18	97	28	-24	33	-32	50	211036		
HSK-25 X-Y USB-85	1	11	37	5	17	45	49	95	28	-37	15	-71	-36	211877	5	18	57	26	96	28	-49	6	-81	-49	212537		
GWM-24 X-Y USB-30	1	11	40	5	17	45	52	95	27	-62	53	-34	17	210079	5	18	57	32	96	27	-66	38	-50	19	210816		

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HFS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
HAW-12 X-Y USB-30	0	20	58	5	17	45	58	95	27	-64	7	-82	26	212365	5	18	6	56	95	27	-62	3	-87	28	211617
MAD-23 X-Y USB-85	1	0	46	5	17	56	15	96	27	57	3	-85	57	212094	5	18	57	1	98	27	65	12	-62	63	212153
CYI-04 X-Y USB-30	1	11	59	5	19	43	31	98	27	63	7	82	26	212561	5	20	55	30	99	27	70	21	68	19	211836
MAD-23 X-Y USB-85	1	11	58	5	19	43	33	98	27	72	21	-39	63	211767	5	20	55	31	99	27	81	33	-13	56	211174
CRD-08 X-Y USB-30	1	12	2	5	19	43	52	96	28	-35	28	-47	47	211369	5	20	55	53	97	28	-47	18	-66	41	212020
GWM-24 X-Y USB-30	1	11	56	5	19	44	2	96	27	-66	27	-61	21	211391	5	20	55	58	97	27	-45	12	-76	25	212243
ACN-75 X-Y USB-30	0	17	9	5	20	38	18	99	27	62	3	87	28	211971	5	20	55	21	99	27	40	17	65	29	212640
ACN-75 X-Y USB-30	1	11	41	5	21	42	4	99	27	58	17	71	30	212132	5	22	53	46	100	27	41	30	66	32	212452
CYI-04 X-Y USB-30	1	11	40	5	21	42	7	99	27	74	31	58	14	211374	5	22	53	47	100	27	42	45	41	18	212761
MAD-23 X-Y USB-85	1	11	43	5	21	42	11	99	27	89	42	-1	48	210820	5	22	53	53	100	27	43	56	7	34	210397

## VEHICLE 1 RADAR TABLE

## TRACKING TIME

## STATION ACQUISITION DATA

## STATION TERMINATION DATA

	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
CRO-08 X-Y USB-30	0	36	52	5	21	42	33	97	28	-54	9	-78	36	212530	5	22	19	25	98	27	-58	3	-86	32	211653
BDA-02 X-Y USB-30	0	18	1	5	22	35	46	100	26	60	3	87	30	212088	5	22	53	47	100	27	62	6	83	27	212826
BDA-02 X-Y USB-30	1	11	48	5	23	40	17	100	26	68	15	74	21	212386	6	0	52	6	101	27	75	29	61	13	211701
ANG-91 X-Y USB-30	1	11	48	5	23	40	18	100	27	65	11	77	24	212599	6	0	52	6	101	27	68	26	62	20	211825
ACN-75 X-Y USB-30	1	11	48	5	23	40	25	100	27	44	39	41	34	211105	6	0	52	13	101	27	29	49	22	35	210747
CYI-04 X-Y USB-30	1	11	46	5	23	40	29	100	27	82	56	34	4	210452	6	0	52	15	101	27	88	71	19	1	210130
MAD-23 X-Y USB-85	1	11	44	5	23	40	33	99	27	113	64	11	24	210206	6	0	52	17	100	27	145	74	13	9	210071
GBM-41 X-Y USB-30	1	6	50	5	23	45	17	100	26	62	3	87	28	212854	6	0	52	7	101	27	68	16	73	21	212425
MIL-71 X-Y USB-30	1	1	19	5	23	50	47	100	26	61	3	87	29	212570	6	0	52	5	101	27	67	14	75	22	212500
GBM-41 X-Y USB-30	0	49	44	6	1	38	46	101	27	72	25	63	16	211947	6	2	28	31	102	27	75	35	54	12	210417

CTT-L13

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
MIL-71 X-Y USB-30	0	49	44	6	1	38	47	101	26	72	24	65	17	212027	6	2	28	31	102	27	76	34	55	12	210498		
ANG-91 X-Y USB-30	0	49	43	6	1	38	47	101	27	69	37	51	17	211364	6	2	28	31	102	27	69	47	41	14	209888		
TEX-16 X-Y USB-30	0	49	43	6	1	38	47	101	26	65	10	79	24	212830	6	2	28	31	102	26	70	19	69	19	211265		
BDA-02 X-Y USB-30	0	49	42	6	1	38	48	101	27	80	38	51	8	211290	6	2	28	31	102	27	85	48	42	3	209852		
CYI-04 X-Y USB-30	0	49	33	6	1	38	57	100	27	94	81	9	-1	210038	6	2	28	31	101	27	-123	88	-2	-1	208989		
ACN-75 X-Y USB-30	0	49	33	6	1	38	58	100	27	13	54	9	35	210649	6	2	28	31	101	27	-5	55	-3	35	209609		
MAD-23 X-Y USB-85	0	49	31	6	1	38	60	100	27	-173	76	14	-2	210095	6	2	28	31	101	27	-137	72	13	-12	209149		
GYM-14 X-Y USB-30	0	31	9	6	1	57	22	101	26	62	3	87	28	212335	6	2	28	31	102	26	65	9	81	25	211892		
GDS-28 X-Y USB-85	0	25	29	6	2	3	1	101	26	60	3	-84	60	212136	6	2	28	31	102	26	63	7	-75	62	211979		
STN-00 A-E USB150	0	8	44	6	2	19	46	102	26	59	3	87	31	211999	6	2	28	31	102	26	60	4	85	30	212144		



31 JAN 71

LM JETT TO TEI IGNITION

LAZ 72.1

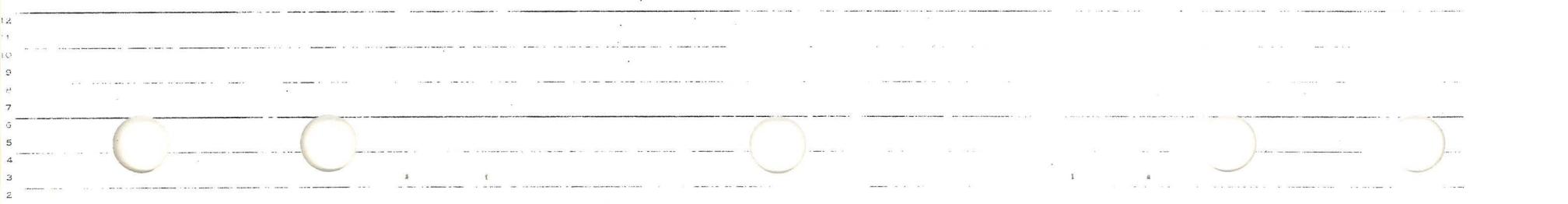
## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE		
GYM-14 X-Y USB-30	0	21	56	6	2	28	36	102	26	65	9	81	25	211894	6	2	50	32	102	26	67	13	76	23	212734		
TEX-16 X-Y USB-30	0	21	56	6	2	28	36	102	26	70	19	69	19	211267	6	2	50	32	102	26	71	24	65	17	212111		
GDS-28 X-Y USB-85	0	21	58	6	2	28	36	102	26	63	7	-75	62	211981	6	2	50	35	102	26	66	11	-65	63	212843		
GBM-41 X-Y USB-30	0	21	58	6	2	28	36	102	27	75	35	54	12	210419	6	2	50	35	102	27	77	40	49	10	211293		
STN-00 A-E USB150	0	21	58	6	2	28	36	102	26	60	4	85	30	212147	6	2	50	35	102	26	63	8	81	27	213018		
MIL-71 X-Y USB-30	0	21	59	6	2	28	36	102	27	76	34	55	12	210500	6	2	50	35	102	27	77	38	51	10	211373		
ANG-91 X-Y USB-30	0	22	0	6	2	28	36	102	27	69	47	41	14	209891	6	2	50	37	102	27	68	52	36	13	210800		
BDA-02 X-Y USB-30	0	22	4	6	2	28	36	102	27	85	48	42	3	209855	6	2	50	41	102	27	88	53	37	1	210780		
ACN-75 X-Y USB-30	0	22	18	6	2	28	36	101	27	-5	55	-3	35	209612	6	2	50	54	101	27	-13	54	-9	35	210746		
CYI-04 X-Y USB-30	0	22	18	6	2	28	36	101	27	-122	88	-2	-1	208992	6	2	50	54	101	27	-98	84	-6	-1	210117		

CONT'D

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HR	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HR	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
MAD-23 X-Y USB-85	0	22	22	6	2	28	36	101	27	-137	72	13	-12	209153	6	2	50	59	101	27	-126	69	13	-17	210322
GYM-14 X-Y USB-30	1	11	40	6	3	37	6	102	26	71	23	66	17	212257	6	4	48	45	103	26	77	37	52	11	211587
GDS-28 X-Y USB-85	1	11	37	6	3	37	8	102	26	71	20	-41	63	212401	6	4	48	45	103	26	79	33	-16	55	211765
TEX-16 X-Y USB-30	1	11	36	6	3	37	8	102	26	75	34	55	12	211661	6	4	48	45	103	26	80	48	41	6	211078
STN-00 A-E USB150	1	11	36	6	3	37	9	102	26	69	17	72	20	212591	6	4	48	44	103	26	78	29	60	10	211960
MIL-71 X-Y USB-30	1	11	39	6	3	37	10	102	26	81	49	41	6	211002	6	4	48	49	103	27	87	63	27	1	210586
GBM-41 X-Y USB-30	1	11	39	6	3	37	11	102	27	80	50	39	7	210929	6	4	48	49	103	27	84	65	24	3	210534
ANG-91 X-Y USB-30	1	11	39	6	3	37	13	102	27	64	62	25	11	210531	6	4	48	52	103	27	44	76	10	10	210328
BDA-02 X-Y USB-30	1	11	41	6	3	37	14	102	26	94	63	27	-2	210524	6	4	48	55	103	26	113	77	12	-5	210321
ACN-75 X-Y USB-30	1	11	39	6	3	37	28	101	27	-29	50	-22	35	210938	6	4	49	7	102	27	-45	40	-40	33	211469

1111  
9999

## VEHICLE 1 RADAR TABLE

## TRACKING TIME

## STATION ACQUISITION DATA

## STATION TERMINATION DATA

HRS MIN SEC

DAY HRS MIN SEC RA DEC AZ ELV X Y RANGE

DAY HRS MIN SEC RA DEC AZ ELV X Y RANGE

CYI-04 X-Y USB-30	1 11 39	6 3 37 29 101 27 -89 73 -17 0 210283	6 4 49 8 102 26 -84 58 -32 3 210758
MAD-23 X-Y USB-85	1 11 37	6 3 37 31 101 26 -109 61 10 -27 210559	6 4 49 8 102 26 -95 48 4 -41 211100
STN-00 A-E USB150	11 9 15	31 JAN 71 TEI TO ENTRY INTERFACE LAZ 72.1	6 16 34 39 102 25 -60 3 -86 30 191464
GYM-14 X-Y USB-30	9 58 57	6 5 25 25 103 26 80 45 45 7 211692	6 15 24 21 101 26 -63 3 -87 27 193801
GDS-28 X-Y USB-85	10 42 14	6 5 25 25 103 26 84 41 -7 49 211875	6 16 7 39 102 25 -61 3 -84 -61 192367
TEX-16 X-Y USB-30	9 5 12	6 5 25 27 103 26 83 57 33 4 211250	6 14 30 39 101 26 -63 3 -87 27 195553
MIL-71 X-Y USB-30	8 0 41	6 5 25 29 103 26 91 72 18 0 210865	6 13 26 9 101 26 -62 3 -87 28 197660
GBM-41 X-Y USB-30	7 46 25	6 5 25 29 103 27 86 74 16 1 210826	6 13 11 54 101 26 -63 3 -87 27 198127
ANG-91 X-Y USB-30	6 19 28	6 5 25 32 102 27 7 80 1 10 210740	6 11 44 60 101 26 -64 3 -87 26 200963
BDA-02 X-Y USB-30	7 5 55	6 5 25 32 102 26 147 83 4 -6 210717	6 12 31 27 101 26 -61 3 -87 29 199453

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HR	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HR	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
MAD-23 X-Y USB-85	3	27	43	6	5	25	38	102	26	-89	41	-2	-49	211852	6	8	53	21	101	26	-58	3	-84	-58	206586
CYI-04 X-Y USB-30	3	38	29	6	5	25	38	102	26	-81	50	-40	6	211496	6	9	4	7	101	26	-62	3	-87	28	206225
ACN-75 X-Y USB-30	2	19	46	6	5	25	40	102	27	-51	33	-50	32	212248	6	7	45	26	101	27	-63	3	-87	27	208855
HAW-12 X-Y USB-30	12	53	7	6	5	33	59	103	26	63	3	87	27	213631	6	18	27	6	102	25	-64	3	-87	26	187712
GWM-24 X-Y USB-30	12	17	57	6	9	33	58	103	26	64	3	87	26	205242	6	21	51	55	102	25	-65	3	-87	25	180702
HSK-25 X-Y USB-85	8	29	25	6	11	12	37	103	27	53	3	-85	53	202006	6	19	42	2	102	27	-54	3	-85	-54	185165
CRO-08 X-Y USB-30	9	36	19	6	13	0	18	103	27	59	3	86	31	198501	6	22	36	38	102	26	-59	3	-87	31	179136
PKS-96 H-D USB210	0	26	18	6	15	14	47	103	27	4	30	7	60	192555	6	15	41	6	103	27	-3	30	-5	60	191676
MAD-23 X-Y USB-85	14	23	46	6	18	26	53	104	25	59	3	-84	59	187722	7	8	50	38	104	24	-61	3	-84	-61	156379
CYI-04 X-Y USB-30	13	17	16	6	19	47	22	104	25	63	3	87	27	184992	7	9	4	38	103	24	-64	3	-87	26	155825

H-18

O-18

## VEHICLE 1 RADAR TABLE

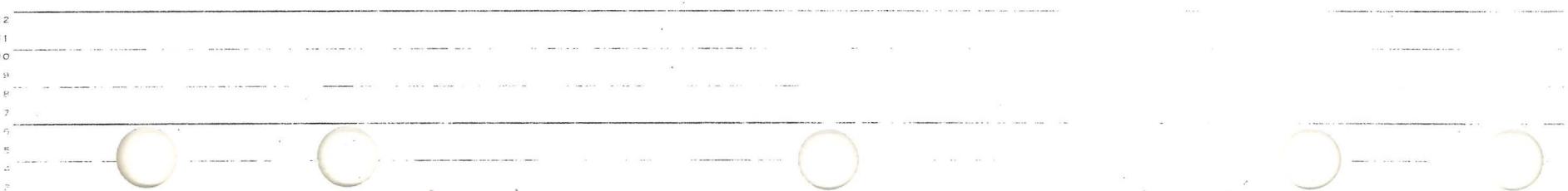
	TRACKING TIME				STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC		DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
ACN-75 X-Y USB-30	10	56	10		6	20	54	19	104	26	63	3	87	27	182692	7	7	50	29	103	25	-64	3	-87	26	158727
BDA-02 X-Y USB-30	13	37	31		6	22	54	2	104	25	62	3	87	28	178535	7	12	31	33	104	24	-64	3	-87	26	147460
ANG-91 X-Y USB-30	12	31	40		6	23	16	22	104	25	65	3	87	25	177746	7	11	48	2	104	24	-66	3	-87	24	149250
GBM-41 X-Y USB-30	13	10	48		7	0	2	30	104	25	63	3	87	26	176114	7	13	13	18	104	24	-65	3	-87	25	145723
MIL-71 X-Y USB-30	13	19	2		7	0	8	12	104	25	63	3	87	27	175911	7	13	27	13	104	24	-65	3	-87	25	145141
TEX-16 X-Y USB-30	13	14	50		7	1	17	28	105	25	63	3	87	27	173432	7	14	32	19	104	24	-65	3	-87	25	142390
GYM-14 X-Y USB-30	13	15	48		7	2	10	44	105	25	63	3	87	27	171503	7	15	26	33	104	23	-65	3	-87	25	140065
GDS-28 X-Y USB-85	13	50	29		7	2	17	22	105	25	62	3	-84	62	171264	7	16	7	51	105	23	-64	3	-83	-63	138273
STN-00 A-E USB150	14	1	8		7	2	33	10	105	25	61	3	87	29	170687	7	16	34	19	105	23	-63	3	-87	27	137116
HAW-12 X-Y USB-30	12	50	14		7	5	41	13	105	25	65	3	87	25	163692	7	18	31	27	105	23	-66	3	-87	24	131894

16TR-T

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
GWM-24 X-Y USB-30	12	16	7	7	9	43	20	106	24	66	3	87	24	154287	7	21	59	27	105	23	-67	3	-87	23	122203
HSK-25 X-Y USB-85	8	45	36	7	11	14	2	106	25	56	3	-85	55	150631	7	19	59	38	105	25	-57	3	-85	-57	127846
CRO-08 X-Y USB-30	9	48	11	7	13	4	56	107	25	61	3	87	29	146065	7	22	53	7	106	24	-62	3	-87	28	119602
PKS-96 H=D USB210	1	59	46	7	14	39	49	106	25	15	30	25	57	140528	7	16	39	35	106	25	-16	30	-26	56	135342
MAD-23 X-Y USB-85	14	1	24	7	18	49	25	108	23	62	3	-84	62	131081	8	8	50	49	109	19	-67	3	-82	-67	87301
CYI-04 X-Y USB-30	13	5	38	7	20	6	41	108	23	66	3	87	24	127527	8	9	12	19	109	20	-69	3	-87	21	85990
ACN-75 X-Y USB-30	11	4	5	7	21	6	22	109	24	66	3	87	24	124727	8	8	10	27	108	21	-68	3	-87	22	89718
BDA-02 X-Y USB-30	13	20	28	7	23	17	47	109	22	66	3	87	24	118404	8	12	38	15	111	18	-71	3	-87	19	72737
ANG-91 X-Y USB-30	12	27	6	7	23	35	47	109	22	67	3	87	23	117515	8	12	2	53	110	19	-71	3	-87	19	75114
GBM-41 X-Y USB-30	12	58	50	8	0	25	27	109	22	67	3	87	23	115045	8	13	24	17	111	18	-72	3	-87	18	69569

L=120



## VEHICLE 1 RADAR TABLE

	TRACKING TIME				STATION ACQUISITION DATA										STATION TERMINATION DATA												
	HRS	MIN	SEC		DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	
MIL-71 X-Y USB=30	13	5	14		8	0	31	50	109	22	67	3	87	23	114725		8	13	37	3	111	17	-72	3	-87	18	68674
TEX-16 X-Y USB=30	13	1	21		8	1	42	16	110	22	67	3	87	23	111143		8	14	43	37	112	17	-73	3	-87	17	63895
GYM-14 X-Y USB=30	13	1	42		8	2	36	41	110	22	67	3	87	23	108320		8	15	38	23	113	16	-73	3	-87	17	59800
GDS-28 X-Y USB=85	13	27	14		8	2	46	34	110	21	66	3	-83	66	107802		8	16	13	48	113	15	-73	3	-80	-73	57065
STN-00 A-E USB150	13	34	24		8	3	3	47	110	21	66	3	87	24	106896		8	16	38	10	114	15	-74	3	-87	16	55138
HAW-12 X-Y USB=30	12	42	46		8	6	10	6	112	21	69	3	87	21	96705		8	18	52	52	116	13	-77	3	-87	13	43709
GWM-24 X-Y USB=30	12	34	39		8	10	16	30	114	20	70	3	87	20	82004		8	22	51	9	125	-1	-91	3	-87	-1	17744
HSK-25 X-Y USB=85	10	13	50		8	11	28	55	114	21	61	3	-84	61	77349		8	21	42	45	122	13	-72	3	-80	-72	26420
CRO-08 X-Y USB=30	10	53	12		8	13	29	8	116	20	66	3	87	24	69224		9	0	22	20	-47	-41	134	3	86	-44	1590
PKS-96 H-D US8210	4	9	28		8	14	22	19	116	21	30	30	41	48	63919		8	18	31	47	117	18	-37	30	-46	44	44082

1-121

12

11

10

9

8

7

6

5

4

3

2

1

0

## VEHICLE 1 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
MAD-23 X-Y USB-85	1	25	19	8	21	29	58	132	4	87	3	-48	86	27899	8	22	55	17	145	-6	101	3	74	79	17163
TAN-22 A-E FPS-16	2	22	28	8	21	50	13	133	11	50	45	37	27	23398	9	0	12	41	-129	-25	116	3	87	-26	3833
PRE-76 A-E MPS-25	2	10	37	8	21	57	16	135	11	61	26	60	26	23400	9	0	7	53	-152	-14	105	3	87	-15	4882
HSK-25 X=Y USB-85	0	19	4	9	0	7	16	143	-8	-98	3	69	-82	5014	9	0	26	20	-25	-13	104	3	77	76	656

L-122

12  
11  
10  
9  
8  
7  
6  
5  
4  
3  
2

TABLE 1.1-V.- MISSION RADAR TIMELINE - Continued

(f) LM acquisition and termination during descent - 3° minimum elevation

31 JAN 71 LM FROM UNDOCK TO LANDING

## VEHICLE RADAR TABLE

VEHICLE	RADAR TABLE	TRACKING TIME												STATION ACQUISITION DATA						STATION TERMINATION DATA							
		DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE				
GWM-24 X-Y USB-30		0	56	38		4	8	27	31	77	27	64	10	79	26	208890		4	9	24	8	78	28	65	22	66	23 208721
HAW-12 X-Y USB-30		0	56	50		4	8	27	31	76	28	72	63	26	8	206446		4	9	24	21	77	28	65	74	34	6 206705
STN-00 A-E USB150		0	57	3		4	8	27	31	76	27	-141	78	-8	-10	206155		4	9	24	34	77	28	-111	69	-20	-8 206819
GDS-28 X-Y USB-85		0	57	5		4	8	27	31	75	27	-119	76	7	-13	206182		4	9	24	36	76	28	-101	65	5	-24 206903
GYM-14 X-Y USB-30		0	57	7		4	8	27	31	75	28	-86	72	-18	1	206243		4	9	24	38	76	28	-82	60	-30	4 207046
TEX-16 X-Y USB-30		0	57	13		4	8	27	31	75	28	-82	60	-30	4	206531		4	9	24	43	76	28	-78	48	-41	8 207463
MIL-7i X-Y USB-30		0	57	13		4	8	27	31	75	27	-79	45	-44	8	207054		4	9	24	43	76	28	-74	34	-55	13 208108
GBM-41 X-Y USB-30		0	57	13		4	8	27	31	75	27	-76	43	-46	10	207163		4	9	24	43	76	28	-73	31	-57	15 208233
BDA-02 X-Y USB-30		0	57	14		4	8	27	31	75	27	-76	33	-57	12	207646		4	9	24	45	76	27	-71	22	-67	18 208742
ANG-91 X-Y USB-30		0	57	16		4	8	27	31	75	27	-67	25	-63	21	208035		4	9	24	46	76	28	-65	14	-75	24 209217

## VEHICLE RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	FRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	FRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
HSK-25 X-Y USB-85	1	5	26		4	10	12	32	77	28	47	7	-79	47 209666		4	11	17	58	79	29	36	15	-71	35 209235
GWM-24 X-Y USB-30	1	5	26		4	10	12	32	78	28	65	32	55	21 208273		4	11	17	59	79	28	63	46	41	18 207698
HAW-12 X-Y USB-30	1	5	26		4	10	12	50	77	28	29	84	3	6 206746		4	11	18	16	78	28	-59	78	-51	6 206837
STN-00 A-E USB150	1	5	28		4	10	12	59	76	27	-99	59	-30	-4 207149		4	11	18	27	77	28	-88	47	-43	1 207672
GDS-28 X-Y USB-85	1	5	24		4	10	13	0	76	27	-92	55	1	-35 207280		4	11	18	24	77	28	-83	43	-7	-47 207856
GYM-14 X-Y USB-30	1	5	26		4	10	13	2	76	28	-79	49	-40	7 207493		4	11	18	27	77	28	-75	36	-53	12 208764
GBM-41 X-Y USB-30	1	5	24		4	10	13	7	76	27	-69	21	-68	20 208857		4	11	18	32	77	27	-64	9	-80	26 209679
TEX-16 X-Y USB-30	1	5	25		4	10	13	7	76	27	-75	38	-51	12 207994		4	11	18	32	77	28	-71	25	-64	18 206757
BDA-02 X-Y USB-30	0	53	52		4	10	13	8	76	27	-65	12	-77	24 209359		4	11	7	0	-77	27	-59	3	-87	31 209387
MIL-71 X-Y USB-30	1	5	23		4	10	13	9	76	27	-71	24	-65	18 208706		4	11	18	32	77	27	-65	11	-77	25 209515

T-1721

VEHICLE RADAR TABLE

TRACKING TIME			STATION ACQUISITION DATA												STATION TERMINATION DATA											
	HRS	MIN SEC	DAY	hrs	min	sec	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	hrs	min	sec	RA	DEC	AZ	ELV	X	Y	RANGE		
ANG-91 X-Y USB-30	0	0 0	8	4	10	13 10	76	27	-62	3 -87	28	209902		4	10	13	17	76	27	-62	3 -87	28	209897			
CRO-08 X-Y USB-30	0	35 39		4	12	6 22	79	28	53	9 79	36	209742		4	12	42	1	79	28	48	15 91	40	208218			
GWM-24 X-Y USB-30	0	35 32		4	12	6 29	78	28	59	57 29	16	207423		4	12	42	1	79	28	53	63 22	16	206043			
HSK-25 X-Y USB-85	0	35 31		4	12	6 30	78	29	26	21 -67	24	209052		4	12	42	1	79	29	19	23 -66	17	207735			
HAW-12 X-Y USB-30	0	35 11		4	12	6 50	78	28	-71	67 -22	7	207108		4	12	42	1	78	28	-72	60 -29	9	206138			
STN-00 A-E USB150	0	35 6		4	12	6 55	77	27	-82	37 -52	7	208175		4	12	42	1	78	27	-78	31 -58	11	207330			
GDS-28 X-Y USB-85	0	35 3		4	12	6 58	77	27	-78	33 -18	-55	208398		4	12	42	1	78	27	-74	26 -29	-59	207575			
GYM-14 X-Y USB-30	0	35 2		4	12	6 59	77	27	-71	26 -63	17	208766		4	12	42	1	78	27	-68	19 -70	20	207983			
TEX-16 X-Y USB-30	0	35 0		4	12	7 1	77	27	-66	15 -74	23	209391		4	12	42	1	78	27	-63	8 -81	26	208618			

TABLE 1.1-V.- MISSION RADAR TIMELINE - Continued

(g) LM acquisition and termination during ascent - 3° minimum elevation

31 JAN 71 LM ASCENT TO INSERTION

VEHICLE	RADAR TABLE																				
	TRACKING TIME			STATION ACQUISITION DATA						STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRs	MIN	SEC	RA DEC	AZ ELV	X	Y	RANGE	DAY	HRs	MIN	SEC	RA DEC	AZ ELV	X	Y	RANGE
MAD-23 X-Y USB-85	0	7	11	5	22	24	29	99 27	95 50	4	40 209365	5	22	31	40	99 27	97 51	5	39 209401		
CYL-04 X-Y USB-30	0	7	11	5	22	24	29	99 27	77 39	50	10 209814	5	22	31	40	100 27	77 41	10	209831		
ACN-75 X-Y USB-30	0	7	11	5	22	24	29	100 27	55 25	60	32 210526	5	22	31	40	100 27	54 26	39	32 210537		

L126

TABLE 1.1-V.- MISSION RADAR TIMELINE - Concluded

(h) LM acquisition and termination during rendezvous - 3° minimum elevation

## COAST TO RENDEZVOUS RADAR TRACKING PRIOR TO TPI

## VEHICLE 2 RADAR TABLE

VEHICLE	TRACKING TIME	STATION ACQUISITION DATA										STATION TERMINATION DATA												
		HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X
MAD-23 X=Y USB=85	0 9 60	5	22	31	40	99	27	97	51	5	39	209395	5	22	41	40	100	27	98	53	6	37	209751	
CYI-04 X=Y USB=30	0 9 60	5	22	31	40	100	27	77	41	49	10	209825	5	22	41	40	100	27	78	43	47	9	210156	
ACN-75 X=Y USB=30	0 9 60	5	22	31	40	100	27	54	26	59	32	210531	5	22	41	40	100	27	53	28	56	32	210854	
BDA-02 X=Y USB=30	0 6 2	5	22	35	38	100	26	60	3	87	30	212021	5	22	41	40	100	27	61	4	85	29	212236	
RENDEZVOUS RADAR TRACKING PRIOR TO TPI																								
ACN-75 X=Y USB=30	0 11 21	5	22	41	40	100	27	53	28	56	32	210854	5	22	53	1	100	27	52	30	53	32	211358	
BDA-02 X=Y USB=30	0 11 22	5	22	41	40	100	27	61	4	85	29	212236	5	22	53	2	100	27	62	6	83	28	212732	
CYI-04 X=Y USB=30	0 11 23	5	22	41	40	100	27	78	43	47	9	210156	5	22	53	3	100	27	79	45	44	8	210671	
MAD-23 X=Y USB=85	0 11 27	5	22	41	40	100	27	98	53	6	37	209751	5	22	53	7	100	27	101	55	7	34	210299	

## COAST TO TPI

## ULLAGE FOR TPI

## TPI

## COAST TO 1ST BRAKING GATE

ANG-91 X=Y USB=30	0 8 20	5	23	40	24	100	27	65	12	77	24	212586	5	23	48	44	100	27	66	13	75	24	212060
BDA-02 X=Y USB=30	0 8 15	5	23	40	28	100	26	68	15	74	21	212369	5	23	48	44	100	26	69	17	72	20	211860

L-127

## VEHICLE 2 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
ACN-75 X=Y USB=30	0	8	13	5	23	40	31	100	27	44	39	41	34	211093	5	23	48	44	100	27	43	41	38	34	210613
CYI-04 X=Y USB=30	0	8	12	5	23	40	32	100	27	82	56	34	4	210444	5	23	48	44	100	27	83	57	32	4	205974
MAD-23 X=Y USB=85	0	8	10	5	23	40	34	99	27	113	64	11	24	210200	5	23	48	44	99	27	116	65	11	22	201351
GBM-41 X=Y USB=30	0	3	27	5	23	45	17	100	26	62	3	87	28	212851	5	23	48	44	100	26	62	4	86	28	212634
COAST TO 2ND BRAKING GATE																									
MAD-23 X=Y USB=85	0	1	49	5	23	48	44	99	27	116	65	11	22	209751	5	23	50	33	99	27	116	65	12	22	209651
GBM-41 X=Y USB=30	0	1	49	5	23	48	44	100	26	62	4	86	28	212634	5	23	50	33	100	26	62	4	85	28	212521
BDA-02 X=Y USB=30	0	1	49	5	23	48	44	100	26	69	17	72	20	211860	5	23	50	33	100	26	69	17	72	20	211748
ANG-91 X=Y USB=30	0	1	49	5	23	48	44	100	27	66	13	75	24	212060	5	23	50	33	100	27	66	14	75	23	211946
CYI-04 X=Y USB=30	0	1	49	5	23	48	44	100	27	83	57	32	4	209970	5	23	50	33	100	27	83	58	32	4	209865
ACN-75 X=Y USB=30	0	1	49	5	23	48	44	100	27	43	41	38	34	210613	5	23	50	33	100	27	43	41	38	34	210507

## 1ST BRAKING MANEUVER \*

## VEHICLE 2 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
MAD=23 X=Y USB=85	0	0	12	5	23	50	33	99	27	116	65	12	22	209651	5	23	50	45	99	27	116	65	12	22	209640
GBM=41 X=Y USB=30	0	0	12	5	23	50	33	100	26	62	4	85	28	212521	5	23	50	45	100	26	62	4	85	28	212508
BDA=02 X=Y USB=30	0	0	12	5	23	50	33	100	26	69	17	72	20	211748	5	23	50	45	100	26	69	17	72	20	211735
ANG=91 X=Y USB=30	0	0	12	5	23	50	33	100	27	66	14	75	23	211946	5	23	50	45	100	27	66	14	75	23	211933
CYI=04 X=Y USB=30	0	0	12	5	23	50	33	100	27	83	58	32	4	209865	5	23	50	45	100	27	83	58	32	4	209854
ACN=75 X=Y USB=30	0	0	12	5	23	50	33	100	27	43	41	38	34	210507	5	23	50	45	100	27	43	41	38	34	210496
COAST TO 3RD BRAKING GATE																									
MAD=23 X=Y USB=85	0	1	1	5	23	50	45	99	27	116	65	12	22	209640	5	23	51	46	99	27	117	66	12	22	209585
GBM=41 X=Y USB=30	0	1	1	5	23	50	45	100	26	62	4	85	28	212508	5	23	51	46	100	26	62	4	85	27	212446
BDA=02 X=Y USB=30	0	1	1	5	23	50	45	100	26	69	17	72	20	211735	5	23	51	46	100	26	69	17	71	20	211673
ANG=91 X=Y USB=30	0	1	1	5	23	50	45	100	27	66	14	75	23	211933	5	23	51	46	100	27	66	14	75	23	211870

L-729

## VEHICLE 2 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
CYI=04 X=Y USB=30	0	1	1	5	23	50	45	100	27	83	58	32	4	209854	5	23	51	46	100	27	83	59	32	4	20979
ACN=75 X=Y USB=30	0	1	1	5	23	50	45	100	27	43	41	38	34	210496	5	23	51	46	100	27	42	41	38	34	21043
MIL=71 X=Y USB=30	0	0	59	5	23	50	47	100	26	61	3	87	29	212570	5	23	51	46	100	26	62	3	86	28	21250
2ND BRAKING MANEUVER *																									
MAD=23 X=Y USB=85	0	0	9	5	23	51	46	99	27	117	66	12	22	209585	5	23	51	54	99	27	117	66	12	22	209577
MIL=71 X=Y USB=30	0	0	9	5	23	51	46	100	26	62	3	86	28	212509	5	23	51	54	100	26	62	3	86	28	212500
GBM=41 X=Y USB=30	0	0	9	5	23	51	46	100	26	62	4	85	27	212446	5	23	51	54	100	26	62	4	85	27	212437
BDA=02 X=Y USB=30	0	0	9	5	23	51	46	100	26	69	17	71	20	211673	5	23	51	54	100	26	69	17	71	20	211665
ANG=91 X=Y USB=30	0	0	9	5	23	51	46	100	27	66	14	75	23	211870	5	23	51	54	100	27	66	14	75	23	211861
CYI=04 X=Y USB=30	0	0	9	5	23	51	46	100	27	83	58	32	4	209796	5	23	51	54	100	27	83	58	32	4	209785
ACN=75 X=Y USB=30	0	0	9	5	23	51	46	100	27	42	41	38	34	210437	5	23	51	54	100	27	42	41	38	34	210429

1130

## COAST TO 4TH BRAKING GATE

## VEHICLE 2 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
MAD-23 X-Y USB-85	0	1	30	5	23	51	54	99	27	117	66	12	22	209577	5	23	53	24	100	27	117	66	12	21	209498
MIL-71 X-Y USB-30	0	1	30	5	23	51	54	100	26	62	3	86	28	212500	5	23	53	24	100	26	62	3	86	28	212410
GBM-41 X-Y USB-30	0	1	30	5	23	51	54	100	26	62	4	85	27	212437	5	23	53	24	100	26	63	5	85	27	212347
BDA-02 X-Y USB-30	0	1	30	5	23	51	54	100	26	69	17	71	20	211665	5	23	53	24	100	26	69	18	71	20	211575
ANG-71 X-Y USB-30	0	1	30	5	23	51	54	100	27	66	14	75	23	211861	5	23	53	24	100	27	66	14	74	23	211770
CYI-04 X-Y USB-30	0	1	30	5	23	51	54	100	27	83	58	32	4	209788	5	23	53	24	100	27	83	58	31	4	209704
ACN-75 X-Y USB-30	0	1	30	5	23	51	54	100	27	92	41	38	34	210429	5	23	53	24	100	27	42	41	37	34	210345
3RD BRAKING MANEUVER *																									
MAD-23 X-Y USB-85	0	0	4	5	23	53	24	100	27	117	66	12	21	209498	5	23	53	28	100	27	117	66	12	21	209494
MIL-71 X-Y USB-30	0	0	4	5	23	53	24	100	26	62	3	86	28	212410	5	23	53	28	100	26	62	3	86	28	212406
GBM-41 X-Y USB-30	0	0	4	5	23	53	24	100	26	63	5	85	27	212347	5	23	53	28	100	26	63	5	85	27	212342

1-131

## VEHICLE 2 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
BDA=02 X=Y USB=30	0	0	4	5	23	53	24	100	26	69	18	71	20	211575	5	23	53	28	100	26	69	18	71	20	211571
ANG=91 X=Y USB=30	0	0	4	5	23	53	24	100	27	66	14	74	23	211770	5	23	53	28	100	27	66	14	74	23	211765
CYI=04 X=Y USB=30	0	0	4	5	23	53	24	100	27	83	58	31	4	209704	5	23	53	28	100	27	83	58	31	4	209701
ACN=75 X=Y USB=30	0	0	4	5	23	53	24	100	27	42	41	37	34	210345	5	23	53	28	100	27	42	41	37	34	210341
COAST TO 5TH BRAKING GATE																									
MAD=23 X=Y USB=85	0	1	15	5	23	53	28	100	27	117	66	12	21	209494	5	23	54	44	100	27	118	66	12	21	209499
MIL=71 X=Y USB=30	0	1	15	5	23	53	28	100	26	62	3	86	28	212406	5	23	54	44	100	26	62	4	86	28	212332
GBM=41 X=Y USB=30	0	1	15	5	23	53	28	100	26	63	5	85	27	212342	5	23	54	44	100	26	63	5	85	27	212268
BDA=02 X=Y USB=30	0	1	15	5	23	53	28	100	26	69	18	71	20	211571	5	23	54	44	100	26	70	18	71	19	211498
ANG=91 X=Y USB=30	0	1	15	5	23	53	28	100	27	66	14	74	23	211765	5	23	54	44	100	27	66	15	74	23	211690
CYI=04 X=Y USB=30	0	1	15	5	23	53	28	100	27	83	58	31	4	209701	5	23	54	44	100	27	83	59	31	3	209632

1-132

## VEHICLE 2 RADAR TABLE

	TRACKING TIME HRS MIN SEC	STATION ACQUISITION DATA										STATION TERMINATION DATA											
		DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
ACN-75	0 1 15	5	23	53	28	100	27	42	41	37	34	210341	5	23	54	44	100	27	42	42	37	34	210272
X=Y USB=30																							
4TH BRAKING MANEUVER																							
MAD-23	0 0 4	5	23	54	44	100	27	118	66	12	21	209429	5	23	54	48	100	27	118	66	12	21	209425
X=Y USB=85																							
MIL-71	0 0 4	5	23	54	44	100	26	62	4	86	28	212332	5	23	54	48	100	26	62	4	86	28	212328
X=Y USB=30																							
GBM-41	0 0 4	5	23	54	44	100	26	63	5	85	27	212268	5	23	54	48	100	26	63	5	85	27	212264
X=Y USB=30																							
BDA-02	0 0 4	5	23	54	44	100	26	70	18	71	19	211498	5	23	54	48	100	26	70	18	71	19	211494
X=Y USB=30																							
ANG-91	0 0 4	5	23	54	44	100	27	66	15	74	23	211690	5	23	54	48	100	27	66	15	74	23	211686
X=Y USB=30																							
CYI-04	0 0 4	5	23	54	44	100	27	83	59	31	3	209632	5	23	54	48	100	27	83	59	31	3	209629
X=Y USB=30																							
ACN-75	0 0 4	5	23	54	44	100	27	42	42	37	34	210272	5	23	54	48	100	27	42	42	37	34	210268
COAST TO DOCKING																							
MAD-23	0 4 22	5	23	54	48	100	27	118	66	12	21	209425	5	23	59	10	100	27	119	67	12	20	209216
X=Y USB=85																							
MIL-71	0 4 22	5	23	54	48	100	26	62	4	86	28	212328	5	23	59	10	100	26	62	5	85	28	212087

1-133

## VEHICLE 2 RADAR TABLE

	TRACKING TIME			STATION ACQUISITION DATA										STATION TERMINATION DATA											
	HRS	MIN	SEC	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE	DAY	HRS	MIN	SEC	RA	DEC	AZ	ELV	X	Y	RANGE
GBH-41 X=Y USB=30	0	4	22	5	23	54	48	100	26	63	5	85	27	212264	5	23	59	10	100	26	63	6	84	27	212023
BDA-02 X=Y USB=30	0	4	22	5	23	54	48	100	26	70	18	71	19	211494	5	23	59	10	100	26	70	19	70	19	211254
ANG-91 X=Y USB=30	0	4	22	5	23	54	48	100	27	66	15	74	23	211686	5	23	59	10	100	27	66	15	73	23	211441
CYI-04 X=Y USB=30	0	4	22	5	23	54	48	100	27	83	59	31	3	209629	5	23	59	10	100	27	84	60	30	3	209408
ACN-75 X=Y USB=30	0	4	22	5	23	54	48	100	27	42	42	37	34	210268	5	23	59	10	100	27	41	42	36	34	210045

1-134

TABLE 1.1-VI.- MISSION SHADOW TIMELINE

(a) CSM

31 JAN 71 EOE TO TLI IGNITION LAZ 72.1

VEH IN SUNLIGHT AT PHASE INITIATION

	TIME SPENT IN REGION							
	DAY	HRS	MIN	SEC	DAY	HRS	MIN	SEC

VEH ENTERING EARTH PENUMBRA 0 0 16 45.1

0 0 0 8.7

VEH ENTERING EARTH UMBRA 0 0 16 53.8

0 0 36 37.6

VEH ENTERING EARTH PENUMBRA 0 0 53 31.5

0 0 0 8.3

VEH ENTERING SUNLIGHT 0 0 53 39.7

0 0 51 13.0

VEH ENTERING EARTH PENUMBRA 0 1 44 52.7

0 0 0 8.7

VEH ENTERING EARTH UMBRA 0 1 45 1.4

0 0 36 36.8

VEH ENTERING EARTH PENUMBRA 0 2 21 38.1

0 0 0 8.3

VEH ENTERING SUNLIGHT 0 2 21 46.4

1-135

## LIGHTING CONDITION FOR VEH 1

31 JAN 71	TLI TO NFR IGNITION	LAZ 72.1
VEH IN SUNLIGHT	AT PHASE INITIATION	

TIME SPENT IN REGION			
DAY	HRS	MIN	SEC

31 JAN 71	NFR TO LOI IGNITION	LAZ 72.1
VEH IN SUNLIGHT	AT PHASE INITIATION	

TIME SPENT IN REGION			
DAY	HRS	MIN	SEC

VEH ENTERING LUNAR PENUMBRA	3	9	39	16.4
	0	0	0	54.2

VEH ENTERING LUNAR UMBRA	3	9	40	10.6
	0	0	55	27.1

VEH ENTERING LUNAR PENUMBRA	3	10	35	37.7
	0	0	0	13.3

VEH ENTERING SUNLIGHT	3	10	35	51.0
-----------------------	---	----	----	------

1-136

12  
11  
10  
9  
8  
7  
6  
5  
4  
3  
2



## LIGHTING CONDITION FOR VEH I

31 JAN 71 LOI TO DOI IGNITION LAZ 72.1

VEH IN SUNLIGHT AT PHASE INITIATION

	VEH ENTERING LUNAR PENUMBRA	TIME SPENT IN REGION		
		DAYS	HRS	MIN S
	VEH ENTERING LUNAR PENUMBRA	3	11	58 43.6
		0	0	0 13.8
	VEH ENTERING LUNAR UMBRA	3	11	58 57.4
		0	0	44 44.0
	VEH ENTERING LUNAR PENUMBRA	3	12	43 41.4
		0	0	0 15.7
	VEH ENTERING SUNLIGHT	3	12	43 57.0
		0	1	23 10.1
	VEH ENTERING LUNAR PENUMBRA	3	14	7 7.1
		0	0	0 11.3
	VEH ENTERING LUNAR UMBRA	3	14	7 10.5
		0	0	44 48.1
	VEH ENTERING LUNAR PENUMBRA	3	14	52 6.6
		0	0	0 5.1
	VEH ENTERING SUNLIGHT	3	14	52 11.7

11-137

## LIGHTING CONDITION FOR VEH 1

31 JAN 71 DCI TO LM SEP IGNITION LAZ 72.1

VEH IN SUNLIGHT AT PHASE INITIATION

				TIME SPENT IN REGION			
DAYS HRS MINS SECS				DAYS HRS MINS SECS			

VEH ENTERING LUNAR PENUMBRA 3 15 57 37.4

0 0 0 6.8

VEH ENTERING LUNAR UMBRA 3 15 57 44.3

0 0 48 18.4

VEH ENTERING LUNAR PENUMBRA 3 16 46 2.7

0 0 0 14.3

VEH ENTERING SUNLIGHT 3 16 46 17.1

0 1 5 35.5

VEH ENTERING LUNAR PENUMBRA 3 17 51 52.5

0 0 0 14.1

VEH ENTERING LUNAR UMBRA 3 17 52 6.7

0 0 48 .8

VEH ENTERING LUNAR PENUMBRA 3 18 40 7.5

0 0 0 11.7

VEH ENTERING SUNLIGHT 3 18 40 19.1

0 1 5 31.9

VEH ENTERING LUNAR PENUMBRA 3 19 45 51.0

0 0 0 17.1

VEH ENTERING LUNAR UMBRA 3 19 46 8.2

0 0 48 12.1

VEH ENTERING LUNAR PENUMBRA 3 20 34 20.2

1-138

## LIGHTING CONDITION FOR VEH 1

				TIME SPENT IN REGION				
	DAY	HRS	MIN	SEC	DAY	HRS	MIN	SEC
					0	0	0	9.2
VEH ENTERING SUNLIGHT	3	20	34	29.5				
					0	1	5	28.6
VEH ENTERING LUNAR PENUMBRA	3	21	39	58.0				
					0	0	0	12.0
VEH ENTERING LUNAR UMBRA	3	21	40	10.0				
					0	0	48	7.2
VEH ENTERING LUNAR PENUMBRA	3	22	28	17.2				
					0	0	0	11.7
VEH ENTERING SUNLIGHT	3	22	28	28.9				
					0	1	5	26.3
VEH ENTERING LUNAR PENUMBRA	3	23	33	55.1				
					0	0	48	30.7
VEH ENTERING LUNAR PENUMBRA	4	0	22	25.9				
					0	0	0	11.3
VEH ENTERING SUNLIGHT	4	0	22	37.1				
					0	1	5	31.6
VEH ENTERING LUNAR PENUMBRA	4	1	28	8.7				
					0	0	0	10.7
VEH ENTERING LUNAR UMBRA	4	1	28	19.4				
					0	0	48	8.7
VEH ENTERING LUNAR PENUMBRA	4	2	16	28.0				

1-139

## LIGHTING CONDITION FOR VEH 1

101-1

			TIME SPENT IN REGION		
DAYS HRS MINS SECS			DAYS HRS MINS SECS		

0 0 0 11.1

VEH ENTERING SUNLIGHT

4 2 16 39.2

0 1 5 48.0

VEH ENTERING LUNAR PENUMBRA

4 3 22 27.2

0 0 0 16.6

VEH ENTERING LUNAR UMBRA

4 3 22 43.8

0 0 47 52.6

VEH ENTERING LUNAR PENUMBRA

4 4 10 36.4

0 0 0 11.5

VEH ENTERING SUNLIGHT

4 4 10 47.9

0 1 5 29.2

VEH ENTERING LUNAR PENUMBRA

4 5 16 17.2

0 0 0 6.8

VEH ENTERING LUNAR UMBRA

4 5 16 24.0

0 0 48 19.0

VEH ENTERING LUNAR PENUMBRA

4 6 4 43.0

0 0 0 11.1

VEH ENTERING SUNLIGHT

4 6 4 54.1

0 1 5 33.2

VEH ENTERING LUNAR PENUMBRA

4 7 10 27.2

0 0 0 13.6

VEH ENTERING LUNAR UMBRA

4 7 10 40.9

0 0 48 5.4

VEH ENTERING LUNAR PENUMBRA

4 7 58 46.3

0 0 0 11.7

VEH ENTERING SUNLIGHT

4 7 58 58.0

31 JAN 71

M SEP TO CIR IGNITION

LAZ 72

VEH IN SUNLIGHT

AT PHASE INITIATION

## LIGHTING CONDITION FOR VEH 1

31 JAN 71 CIR TO LOPC1 IGNITION LAZ 72.1

VEH IN LUNAR UMBRA AT PHASE INITIATION

	TIME SPENT IN REGION							
	DAY	HRS	MIN	SEC	DAY	HRS	MIN	SEC

VEH ENTERING LUNAR PENUMBRA	4	9	52	46.0	0	0	0	11.2
-----------------------------	---	---	----	------	---	---	---	------

VEH ENTERING SUNLIGHT	4	9	52	57.2	0	1	12	39.3
-----------------------	---	---	----	------	---	---	----	------

VEH ENTERING LUNAR PENUMBRA	4	11	5	36.5	0	0	0	9.0
-----------------------------	---	----	---	------	---	---	---	-----

VEH ENTERING LUNAR UMBRA	4	11	5	45.5	0	0	45	36.9
--------------------------	---	----	---	------	---	---	----	------

VEH ENTERING LUNAR PENUMBRA	4	11	51	22.4	0	0	0	11.2
-----------------------------	---	----	----	------	---	---	---	------

VEH ENTERING SUNLIGHT	4	11	51	33.6	0	1	12	42.7
-----------------------	---	----	----	------	---	---	----	------

VEH ENTERING LUNAR PENUMBRA	4	13	4	16.3	0	0	0	11.4
-----------------------------	---	----	---	------	---	---	---	------

VEH ENTERING LUNAR UMBRA	4	13	4	27.7	0	0	45	42.1
--------------------------	---	----	---	------	---	---	----	------

VEH ENTERING LUNAR PENUMBRA	4	13	50	9.8	0	0	0	9.0
-----------------------------	---	----	----	-----	---	---	---	-----

VEH ENTERING SUNLIGHT	4	13	50	18.8	0	1	12	34.1
-----------------------	---	----	----	------	---	---	----	------

VEH ENTERING LUNAR PENUMBRA	4	15	2	52.9
-----------------------------	---	----	---	------

T-14

## LIGHTING CONDITION FOR VEH 1

	TIME SPENT IN REGION			
	DAY	HRS	MIN	SECS
VEH ENTERING LUNAR UMBRA	4	15	3	3.9
	0	0	45	38.9
VEH ENTERING LUNAR PENUMBRA	4	15	48	42.8
	0	0	0	11.3
VEH ENTERING SUNLIGHT	4	15	48	54.1
	0	1	12	32.5
VEH ENTERING LUNAR PENUMBRA	4	17	1	26.6
	0	0	0	14.2
VEH ENTERING LUNAR UMBRA	4	17	1	40.8
	0	0	45	38.9
VEH ENTERING LUNAR PENUMBRA	4	17	47	19.7
	0	0	0	10.9
VEH ENTERING SUNLIGHT	4	17	47	30.7
	0	1	12	41.2
VEH ENTERING LUNAR PENUMBRA	4	19	0	11.8
	0	0	0	11.4
VEH ENTERING LUNAR UMBRA	4	19	0	23.2
	0	0	45	38.9
VEH ENTERING LUNAR PENUMBRA	4	19	46	2.1
	0	0	0	10.9
VEH ENTERING SUNLIGHT	4	19	46	13.0

L1-12

## LIGHTING CONDITION FOR VEH 1

				TIME SPENT IN REGION				
	DAY	HRS	MIN	SEC	DAY	HRS	MIN	SEC
	0	1	12	36.3				
VEH ENTERING LUNAR PENUMBRA	4	20	58	49.2				
	0	0	0	10.9				
VEH ENTERING LUNAR UMBRA	4	20	59	.1				
	0	0	45	39.4				
VEH ENTERING LUNAR PENUMBRA	4	21	44	39.5				
	0	0	0	11.4				
VEH ENTERING SUNLIGHT	4	21	44	50.9				

1-143

## LIGHTING CONDITION FOR VEH 1

31 JAN 71      LOPC1 TO LM JETT IGNITION      LAZ 72.1  
 VEH IN SUNLIGHT      AT PHASE INITIATION

		TIME SPENT IN REGION				DAY	HRS	MIN	SEC
		DAY	HRS	MIN	SEC				
VEH ENTERING LUNAR PENUMBRA		4	22	57	41.6		0	0	0
						0	0	0	11.2
VEH ENTERING LUNAR UMBRA		4	22	57	52.9		0	0	45
						0	0	45	45.7
VEH ENTERING LUNAR PENUMBRA		4	23	43	38.6		0	0	0
						0	0	0	11.3
VEH ENTERING SUNLIGHT		4	23	43	49.9		0	1	12
						0	1	12	27.8
VEH ENTERING LUNAR PENUMBRA		5	0	56	17.7		0	0	0
						0	0	0	11.1
VEH ENTERING LUNAR UMBRA		5	0	56	28.8		0	0	45
						0	0	45	48.7
VEH ENTERING LUNAR PENUMBRA		5	1	42	17.5		0	0	0
						0	0	0	11.0
VEH ENTERING SUNLIGHT		5	1	42	28.6		0	1	12
						0	1	12	28.8
VEH ENTERING LUNAR PENUMBRA		5	2	54	57.4		0	0	0
						0	0	0	9.1
VEH ENTERING LUNAR UMBRA		5	2	55	6.5		0	0	45
						0	0	45	47.5
VEH ENTERING LUNAR PENUMBRA		5	3	40	54.0				

L-144

## LIGHTING CONDITION FOR VEH 1

				TIME SPENT IN REGION				
	DAYS	HRS	mins	SECS	DAYS	HRS	mins	SECS
					0	0	0	11.3
VEH ENTERING SUNLIGHT	5	3	41	5.3				
					0	1	12	32.2
VEH ENTERING LUNAR PENUMBRA	5	4	53	37.5				
					0	0	0	11.3
VEH ENTERING LUNAR UMBRA	5	4	53	48.9				
					0	0	45	50.8
VEH ENTERING LUNAR PENUMBRA	5	5	39	39.7				
					0	0	0	8.8
VEH ENTERING SUNLIGHT	5	5	39	48.6				
					0	1	12	25.7
VEH ENTERING LUNAR PENUMBRA	5	6	52	14.2				
					0	0	0	10.9
VEH ENTERING LUNAR UMBRA	5	6	52	25.2				
					0	0	45	49.8
VEH ENTERING LUNAR PENUMBRA	5	7	38	15.0				
					0	0	0	11.2
VEH ENTERING SUNLIGHT	5	7	38	26.2				
					0	1	12	22.0
VEH ENTERING LUNAR PENUMBRA	5	8	50	48.2				
					0	0	0	14.2
VEH ENTERING LUNAR UMBRA	5	8	51	2.3				

1-145

## LIGHTING CONDITION FOR VEH 1

97ET-1

	TIME SPENT IN REGION			
	DAYS	HRS	MINS	SECS

0	0	45	49.3
---	---	----	------

VEH ENTERING LUNAR PENUMBRA

5	9	36	51.6
---	---	----	------

0	0	0	11.1
---	---	---	------

VEH ENTERING SUNLIGHT

5	9	37	2.7
---	---	----	-----

0	1	12	30.8
---	---	----	------

VEH ENTERING LUNAR PENUMBRA

5	10	49	33.5
---	----	----	------

0	0	0	11.3
---	---	---	------

VEH ENTERING LUNAR UMPRA

5	10	49	44.9
---	----	----	------

0	0	45	49.4
---	---	----	------

VEH ENTERING LUNAR PENUMBRA

5	11	35	34.2
---	----	----	------

0	0	0	15.4
---	---	---	------

VEH ENTERING SUNLIGHT

5	11	35	49.6
---	----	----	------

0	1	12	21.5
---	---	----	------

VEH ENTERING LUNAR PENUMBRA

5	12	48	11.1
---	----	----	------

0	0	0	10.8
---	---	---	------

VEH ENTERING LUNAR UMBRA

5	12	48	21.9
---	----	----	------

0	0	45	50.4
---	---	----	------

VEH ENTERING LUNAR PENUMBRA

5	13	34	12.3
---	----	----	------

0	0	0	11.3
---	---	---	------

VEH ENTERING SUNLIGHT

5	13	34	23.7
---	----	----	------

0	1	12	29.7
---	---	----	------

VEH ENTERING LUNAR PENUMBRA

5	14	46	53.4
---	----	----	------

## LIGHTING CONDITION FOR VEH 1

				TIME SPENT IN REGION			
DAYS HRS MINS SECS				DAYS HRS MINS SECS			

0	0	0	10.8
---	---	---	------

VEH ENTERING LUNAR UMBRA	5	14	47	4.2
--------------------------	---	----	----	-----

0	0	45	45.6
---	---	----	------

VEH ENTERING LUNAR PENUMBRA	5	15	32	49.8
-----------------------------	---	----	----	------

0	0	0	10.7
---	---	---	------

VEH ENTERING SUNLIGHT	5	15	33	.5
-----------------------	---	----	----	----

0	1	12	29.2
---	---	----	------

VEH ENTERING LUNAR PENUMBRA	5	16	45	29.7
-----------------------------	---	----	----	------

0	0	0	11.3
---	---	---	------

VEH ENTERING LUNAR UMBRA	5	16	45	41.0
--------------------------	---	----	----	------

0	0	45	51.3
---	---	----	------

VEH ENTERING LUNAR PENUMBRA	5	17	31	32.3
-----------------------------	---	----	----	------

0	0	0	10.9
---	---	---	------

VEH ENTERING SUNLIGHT	5	17	31	43.2
-----------------------	---	----	----	------

0	1	12	21.0
---	---	----	------

VEH ENTERING LUNAR PENUMBRA	5	18	44	4.2
-----------------------------	---	----	----	-----

0	0	0	9.0
---	---	---	-----

VEH ENTERING LUNAR UMBRA	5	18	44	13.2
--------------------------	---	----	----	------

0	0	45	56.5
---	---	----	------

VEH ENTERING LUNAR PENUMBRA	5	19	30	9.7
-----------------------------	---	----	----	-----

0	0	0	11.4
---	---	---	------

VEH ENTERING SUNLIGHT	5	19	30	21.1
-----------------------	---	----	----	------

LHT-T

## LIGHTING CONDITION FOR VEH 1

	DAYS	HRS	MINS	SECS	TIME SPENT IN REGION			
					DAYS	HRS	MINS	SECS
					0	1	12	28.9
VEH ENTERING LUNAR PENUMBRA	5	20	42	50.0				
					0	0	0	11.2
VEH ENTERING LUNAR UMBRA	5	20	43	1.2				
					0	0	45	50.9
VEH ENTERING LUNAR PENUMBRA	5	21	28	52.1				
					0	0	0	14.3
VEH ENTERING SUNLIGHT	5	21	29	6.4				
					0	1	12	19.8
VEH ENTERING LUNAR PENUMBRA	5	22	41	26.2				
					0	0	0	11.2
VEH ENTERING LUNAR UMBRA	5	22	41	37.4				
					0	0	45	52.9
VEH ENTERING LUNAR PENUMBRA	5	23	27	30.3				
					0	0	0	11.0
VEH ENTERING SUNLIGHT	5	23	27	41.3				
					0	1	12	23.3
VEH ENTERING LUNAR PENUMBRA	6	0	40	4.6				
					0	0	0	8.8
VEH ENTERING LUNAR UMBRA	6	0	40	13.4				
					0	0	45	53.7
VEH ENTERING LUNAR PENUMBRA	6	1	26	7.1				
					0	0	0	11.4
VEH ENTERING SUNLIGHT	6	1	26	18.5				

L-148

## LIGHTING CONDITION FOR VEH 1

31 JAN 71 LM JETT TO TEI IGNITION LAZ 72.1

VEH IN SUNLIGHT AT PHASE INITIATION

				TIME SPENT IN REGION				
	DAY	HRS	MIN	SEC	DAY	HRS	MIN	SEC

VEH ENTERING LUNAR PENUMBRA 6 2 38 46.6

0 0 0 11.3

VEH ENTERING LUNAR UMBRA 6 2 38 57.9

0 0 45 51.3

VEH ENTERING LUNAR PENUMBRA 6 3 24 49.2

0 0 0 13.2

VEH ENTERING SUNLIGHT 6 3 25 2.4

0 1 12 16.5

VEH ENTERING LUNAR PENUMBRA 6 4 37 19.0

0 0 0 11.1

VEH ENTERING LUNAR UMBRA 6 4 37 30.1

31 JAN 71 TEI TO ENTRY INTERFACE LAZ 72.1

VEH IN LUNAR UMBRA AT PHASE INITIATION

				TIME SPENT IN REGION				
	DAY	HRS	MIN	SEC	DAY	HRS	MIN	SEC

VEH ENTERING LUNAR PENUMBRA 6 5 19 34.8

0 0 0 4.9

VEH ENTERING SUNLIGHT 6 5 19 39.7

164 TET

TABLE 1.1-VI.- MISSION SHADOW TIMELINE - Continued

(b) LM descent  
LIGHTING CONDITION FOR VEH I

31 JAN 71 LM FROM UNDOCK TO LANDING

VEH IN SUNLIGHT

AT PHASE INITIATION.

	DAYS HRS MINS SECS	TIME SPENT IN REGION		
		DAYS	HRS	MINS SECS
VEH ENTERING LUNAR PENUMBRA	4 9 4 34.2		0	0 0 11.7
VEH ENTERING LUNAR UMBRA	4 9 4 45.9		0	0 48 7.0
VEH ENTERING LUNAR PENUMBRA	4 9 52 52.9		0	0 0 11.8
VEH ENTERING SUNLIGHT	4 9 53 4.7		0	1 5 35.5
VEH ENTERING LUNAR PENUMBRA	4 10 58 40.2		0	0 0 10.6
VEH ENTERING LUNAR UMBRA	4 10 58 50.8		0	0 48 8.2
VEH ENTERING LUNAR PENUMBRA	4 11 46 59.0		0	0 0 10.7
VEH ENTERING SUNLIGHT	4 11 47 9.7			

T-150

TABLE 1.1-VI.- MISSION SHADOW TIMELINE - Concluded

(c) LM rendezvous

LIGHTING CONDITION FOR VEH 2

COAST TO RENDEZVOUS RADAR TRACKING PRIOR TO TPI

VEH IN PENUMBRA 00000R AT PHASE INITIATION

VEH ENTERING LUNAR UMBRA	TIME SPENT IN REGION			
	DAYS	HRS	MINS	SECS
	5	22	40	9.9

COAST TO 1ST BRAKING GATE

VEH IN PENUMBRA 00000R AT PHASE INITIATION

VEH ENTERING SUNLIGHT	TIME SPENT IN REGION			
	DAYS	HRS	MINS	SECS
	5	23	28	21.3

**2.0 SYMBOLS AND NOMENCLATURE**

## 2.0 SYMBOLS AND NOMENCLATURE

AGS	abort guidance system
AOS	acquisition of signal
APS	ascent propulsion system
C	cross-product steering gain constant
CDH	constant delta height
CDR	commander
CM	command module
CMC	command module computer
CMP	command module pilot
CSI	coelliptic sequence initiation
CSM	command and service modules
c.g.	center of gravity
DOI	descent orbit insertion
DPS	descent propulsion system
DSKY	display keyboard
ECI	earth centered inertial
EI	entry interface
EMS	entry monitor system
EPO	earth parking orbit
FTP	fixed throttle point
g.e.t.	ground elapsed time
$h_a$	apogee altitude
$h_p$	perigee altitude

IGA	inner gimbal angle
IMU	inertial measurement unit
IVT	intravehicular transfer
Jerk	time derivative of acceleration
L/D	lift-to-drag ratio
LGC	LM guidance computer
LLM	lunar landing mission
LLS	lunar landing site
LM	lunar module
LMP	lunar module pilot
LOI	lunar orbit insertion
LOS	loss of signal
LPO	lunar parking orbit
LOX	liquid oxygen
LV	launch vehicle
MC	LM midcourse correction
MCC	midcourse correction
MCI	moon centered inertial
MGA	middle gimbal angle
MNBY	mean nearest Besselian year
MSFC	Marshall Space Flight Center
MSFN	Manned Space Flight Network
OGA	outer gimbal angle
OPS	oxygen purge system
PC	plane change

PDI	powered descent initiation
PGNCS	primary guidance and navigation control subsystem
PTC	passive thermal control
RDG	position target for LM powered descent
RCS	reaction control system
REFSMMAT	transformation matrix from the basic reference coordinate system to the stable member (IMU) coordinate system
RR	rendezvous radar
RT	target vector for Lambert guidance scheme
SC	spacecraft
SPS	service propulsion system
TB7	time elapsed since TLI engine cutoff
T&D	transposition and docking
TD&E	transposition and docking and extraction
TEI	transearth injection
$t_{IG}$	time at ignition
TLI	translunar injection
Tf	$\Delta t$ from ignition time ( $t_{IG}$ ) to Lambert target vector (RT)
TPF	terminal phase finalization
TPI	terminal phase initiation
VHF	very high frequency
$\Delta V$	change in velocity caused by thrusting
$\Delta t$	change in time

$\Delta V_X$	}	components of velocity to be gained in the local vertical coordinate system
$\Delta V_Y$		
$\Delta V_Z$	}	components of unit vector in inertial coordinate system
$X_I$		
$Y_I$	}	components of unit vector in vehicle stable member system
$Z_I$		
$X_{SM}$	}	components of unit vector in vehicle stable member system
$Y_{SM}$		
$Z_{SM}$		

**3.0 PRIMARY GUIDELINES AND CONSTRAINTS**

## 3.0 PRIMARY GUIDELINES AND CONSTRAINTS

The design of the mission and the monthly and daily launch windows were based on the guidelines and constraints listed below.

- a. The primary lunar landing site is Fra Mauro. There is no backup site.
- b. Daylight launch is highly desirable.
- c. The launch windows and profiles will be designed to achieve favorable lunar lighting at the lunar landing sites. The sun elevation range of  $7^\circ$  to  $14^\circ$  is highly desirable; a range of  $18^\circ$  to  $20^\circ$  is acceptable with proper landing site approach azimuths.
- d. The flight azimuth range following launch is limited to  $72^\circ$  to  $96^\circ$ .
- e. The mission will be designed for a translunar injection over the Pacific Ocean.
- f. Two TLI opportunities will be targeted: the first on the second revolution and the second on the third revolution.
- g. TLI will be targeted to provide a free-return circumlunar trajectory.
- h. The S-IVB will perform an evasive maneuver prior to the blowdown. The blowdown will be targeted to achieve an S-IVB lunar impact.
- i. A hybrid profile will be used where necessary to provide a Pacific injection, optimum lunar lighting, 210-foot dish coverage of lunar landing, or a preferred lunar groundtrack.
- j. The hybrid transfer maneuver will be planned for TLI plus 28 hours. After the maneuver, the SC will be on a relaxed-free-return trajectory which will be restricted such that, in the case of an SPS failure at LOI ignition, an earth return is within the DPS docked  $\Delta V$  capability.
- k. An even integer number of days of flight time to the moon is highly desirable because of the crew work/rest cycle.
- l. The LOI maneuver will result in a 57- by 170-n. mi. elliptical orbit. DOI will be performed with the SPS after two revs in the 57- by 170-n. mi. orbit established by LOI. The targeting will be biased to place the spacecraft in a 50 000-foot by 60-n. mi. orbit at the time of PDI on rev 14.

m. The lunar surface stay time will be approximately 3<sup>1</sup>/<sub>2</sub> hours. The CSM lunar orbit orientation will be restricted to permit an any-orbit LM lift-off.

n. A CSM plane change will be made approximately 4.75 revolutions after LM landing to place the landing site in the orbital plane at the planned time of lift-off and to allow photography of Descartes on revolution 27.

o. The LM ascent stage will be deorbited for lunar impact.

p. The lunar orbit will be designed within the SPS capability to return to earth from any orbit including the cryogenic tank failure situation.

q. The TEI maneuver will be targeted to return as soon as possible to the prime recovery area within the available  $\Delta V$  capability and without exceeding a return inclination of 40°.

r. The earth relative entry range will be 1250 n. mi.

**4.0 MISSION SUMMARY**

## 4.0 MISSION SUMMARY

In this section, a mission profile is summarized for a mission launched January 31, 1971. Major events, spacecraft performance characteristics, and significant trajectory parameters are described in detail for each phase.

In the design of the spacecraft operational mission, the LV mission phases were simulated independently of the exact LV operational trajectory (ref. 2). The trajectory data presented here for the LV mission phases were simulated with data received from the MSFC for the Apollo 14 mission. No attempt was made to duplicate exactly the LV operational mission, and the information for the LV phases will differ from information that will be presented in the official MSFC trajectory document.

4.1 EARTH LAUNCH

4.1 Earth Launch<sup>a</sup>

The launch time for this mission was determined to provide an optimized injected payload to support two injection opportunities. The launch is summarized as follows.

Date, month, day, year . . . . .	January 31, 1971
Time, hr:min, c.s.t. . . . .	14:20
Azimuth, deg . . . . .	~ 72
Location (Cape Kennedy, complex 39A)	
Geodetic latitude, deg:min:sec . . . .	28°36'30.32"
Longitude, deg:min:sec . . . . .	279°23'45.12"

The CSM launch pad REFSMMAT is presented in table 5-I.

---

<sup>a</sup>The parameters for this phase are approximate and are presented for information only. The official source for this phase is the MSFC LV operational trajectory (ref. 2).

4.2 EARTH PARKING ORBIT

#### 4.2 Earth Parking Orbit<sup>a</sup> (EPO)

The EPO insertion conditions are as follows.

The insertion ship positioned at 28°N latitude and 49°W longitude tracks the vehicle for approximately 3 minutes after insertion for a minimum elevation angle of 5°.

The LV maintains local horizontal attitude throughout the EPO phase except for an inertial hold of approximately 10 seconds immediately after EPO insertion. The total time spent in EPO is  $2^{\text{h}}29^{\text{m}}2.8^{\text{s}}$ . The EPO ground-track from insertion through translunar injection is shown in figure 4.2-1.

<sup>a</sup>The parameters for this phase are approximate and are presented for information only. The official source for this phase is the MSFC LV operational trajectory (ref. 2).

**4.3 TRANSLUNAR INJECTION**

### 4.3 Translunar Injection<sup>a</sup> (TLI)

The TLI burn is initiated over Australia on the second revolution in earth orbit, and the burn is totally in sunlight. Four ARIA support the burn. The TLI ships are not used for this mission. The TLI burn geometry is shown in figure 4.3-1.

The TLI burn places the spacecraft on a free-return trajectory with a high perilune altitude (2030 n. mi.). The evasive maneuver after CSM/LM ejection is made with the S-IVB APS rather than with the SM SPS. The characteristics of the TLI burn are as follows.

#### Burn initiation

Time, hr:min:sec, g.e.t. . . . .	02:30:42.4
Geodetic latitude, deg . . . . .	-29.0
Longitude, deg . . . . .	124.4
Velocity, fps . . . . .	25 593.6
Apogee altitude above mean radius, n. mi. . . . .	104.7

#### Burn cutoff

Time, hr:min:sec, g.e.t. . . . .	02:36:06.8
Geodetic latitude, deg . . . . .	-19.1
Longitude, deg . . . . .	149.9
Velocity, fps . . . . .	35 598.6
Burn duration, sec . . . . .	324.4
S-IVB propellant used, lb . . . . .	157 289
Plane change, deg . . . . .	0.3
Inclination to earth-moon plane, deg . . . . .	28.1
Apogee altitude above mean radius, n. mi. . . . .	250 173

---

<sup>a</sup>The parameters for this phase are approximate and are presented for information only. The official source for this phase is the MSFC LV operational trajectory (ref. 2).

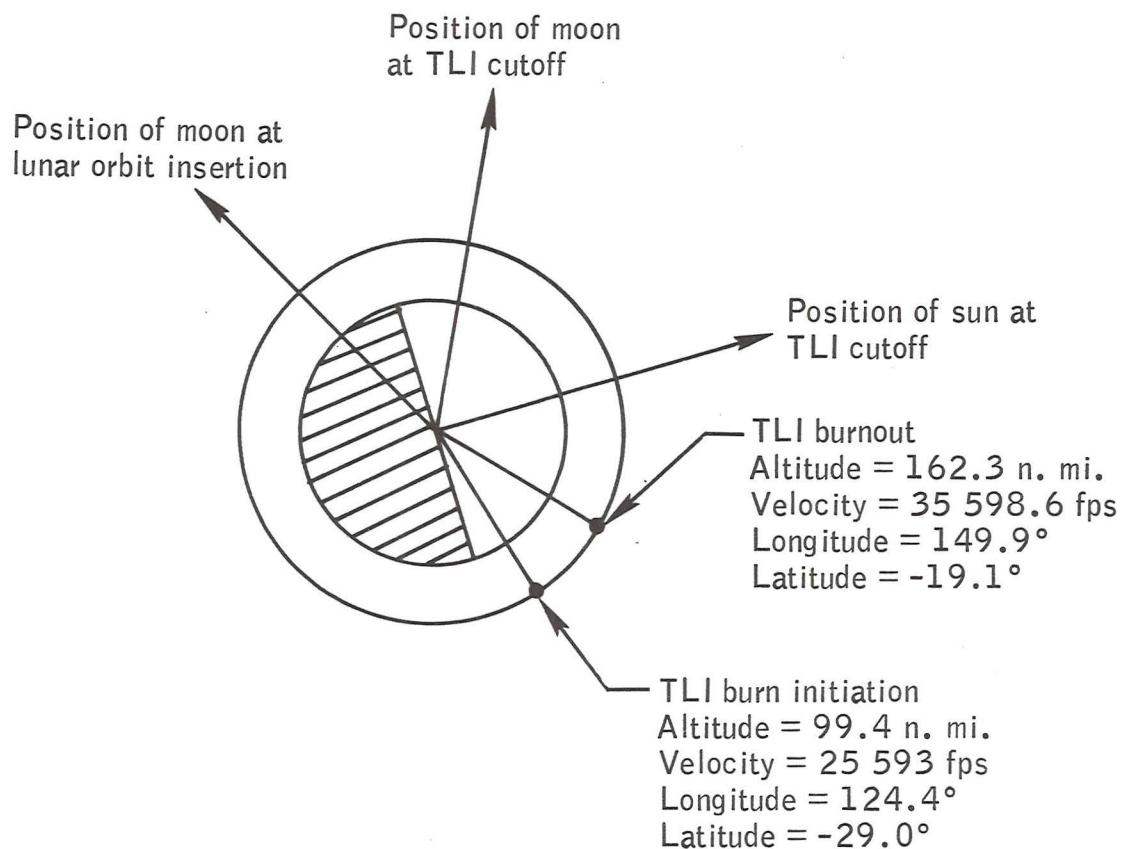


Figure 4.3-1.- TLI geometry.

4.4 FREE-RETURN CIRCUMLUNAR

#### 4.4 Free-Return Circumlunar Trajectory

The free-return trajectory of the hybrid profile, shown in figure 1.1-1, is plotted in an inertial reference frame anchored at the time of perilune. The characteristics of the trajectory are the following.

##### Perilune

Time, hr:min:sec, g.e.t. . . . .	82:39:52.2
Altitude above LLS, n. mi. . . . .	2030.9
Selenographic latitude, deg . . . . .	6.6
Selenographic longitude, deg . . . . .	-172.7

Return vacuum perigee altitude, n. mi. . . . . 20.6

##### Transit time from TLI

Burnout to entry interface, hr:min:sec . . . . . 166:51:14

##### Earth entry interface

Time, hr:min:sec, g.e.t. . . . .	169:27:21
Altitude, n. mi. . . . .	65.8
Geodetic latitude, deg . . . . .	-30.0
Longitude, deg . . . . .	117.5
Inclination, deg . . . . .	31.0

##### Landing

Time, hr:min:sec, g.e.t. . . . .	169:41:21
Geodetic latitude, deg . . . . .	-24.6
Longitude, deg . . . . .	140.2
Time prior to sunset, hr:min . . . . .	11:30

**4.5 POST-TRANSLUNAR INJECTION EVENTS**

#### 4.5 Posttranslunar Injection Events

The summary of the major events from TLI cutoff through S-IVB LOX blowdown is given in table 4.5-I. To determine the separation attitude maneuver (TB7 plus 900 sec), the sun was constrained to between  $32^\circ$  and  $90^\circ$  of the LV +X-axis. This constraint provides over-the-shoulder lighting and avoids any CSM shadow on the S-IVB for the docking phase.

Upon ground command, the S-IVB performs an APS evasive maneuver approximately 23 minutes after CSM/LM separation. The S-IVB attitude for this maneuver will be the same as the separation attitude except that the sign of the middle gimbal angle (yaw gimbal) will be reversed. The APS burn will be 10.0 fps.

The S-IVB performs a LOX dump 21 minutes 20 seconds after the APS evasive maneuver. The preliminary local horizontal attitude for this dump is pitch  $183^\circ$ , yaw  $-5^\circ$ , roll  $180^\circ$ .

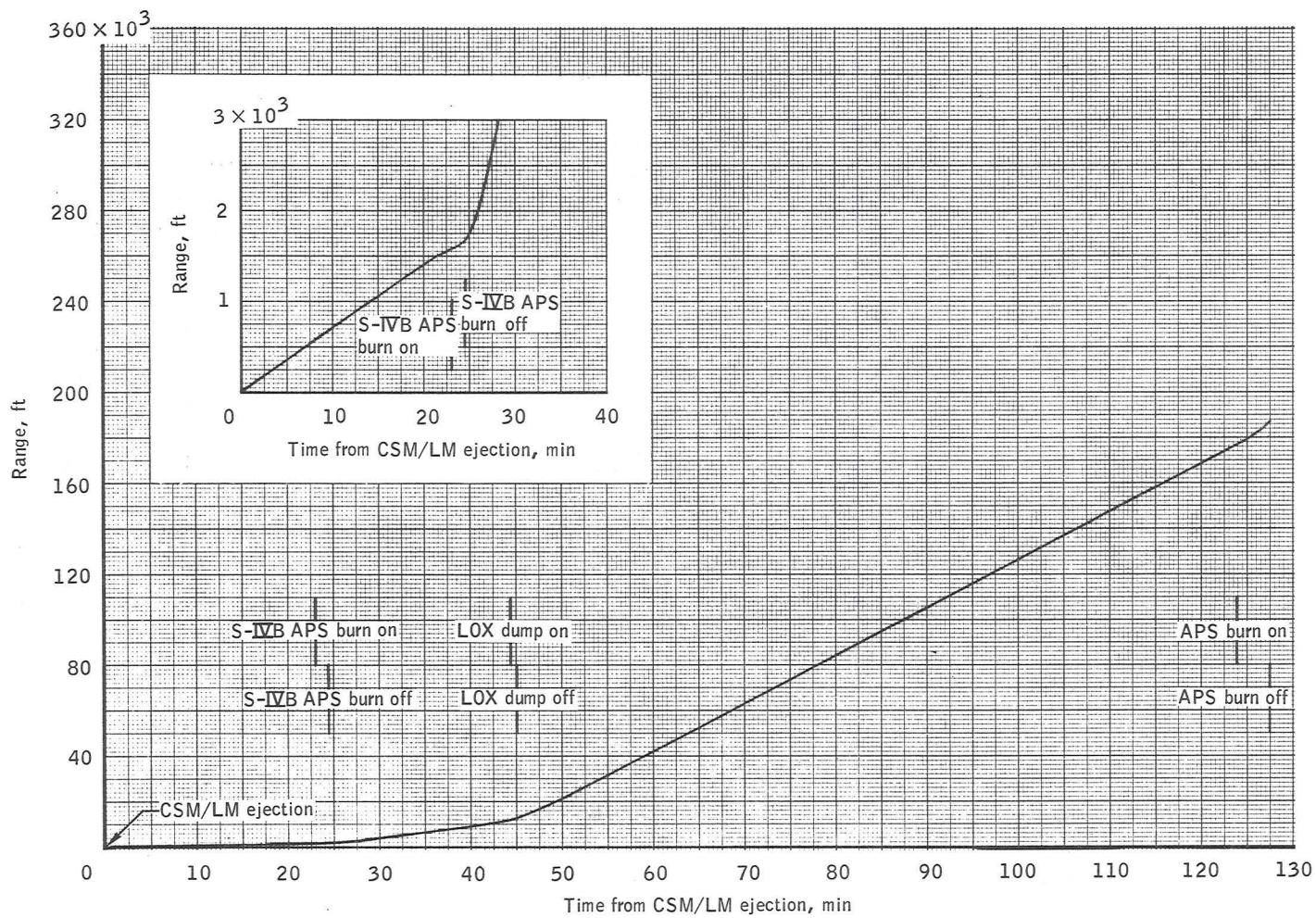
The LOX dump maneuver is designed to reduce the probability of SC re-contact with the S-IVB and also to achieve an S-IVB impact with the moon. Nominally, an S-IVB APS lunar impact burn will be executed after the APS evasive and LOX dump maneuvers (g.e.t. of approximately 6 hours) to target the S-IVB for a lunar impact at  $33^\circ\text{W}$  and  $20^\circ\text{S}$  which is near the Apollo 12 landing site.

The relative motion between the S-IVB and SC for this post-T, D, and E phase is shown in figure 4.5-1.

TABLE 4.5-I.- SUMMARY OF EVENTS FROM TLI

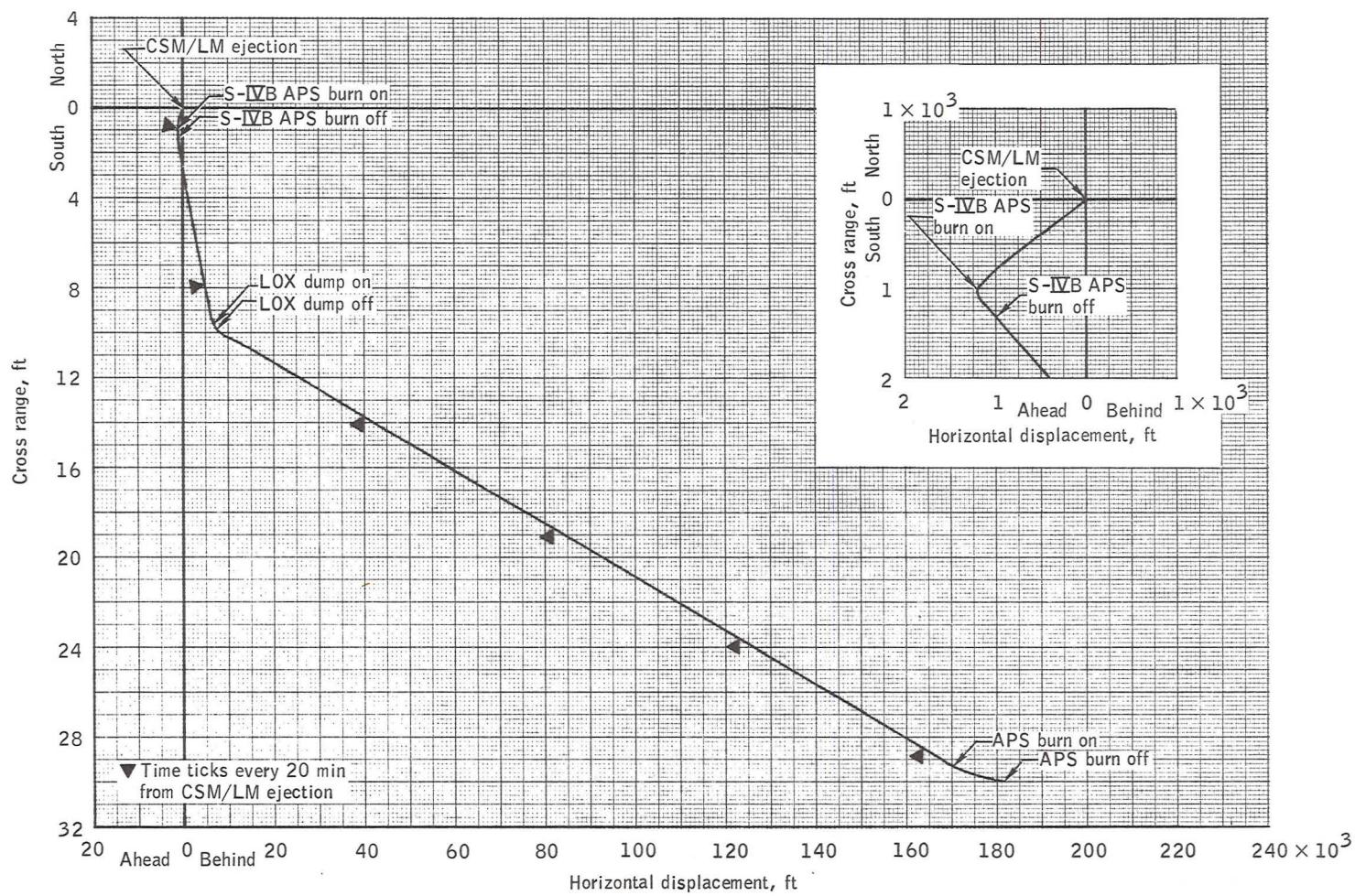
## CUTOFF THROUGH LOX DUMP

Time from TB7, hr:min:sec	Event	$\Delta V$ , fps	Comments
00:00:00	Hold cutoff attitude		
00:00:20	Command and hold local horizontal		
00:15:00	Initiate maneuver to separation attitude		Local horizontal attitude Pitch = $120^\circ$ Yaw = $-40^\circ$ Roll = $180^\circ$ 0.8 deg/sec
00:20:00	Freeze separation attitude inertially		Latest time for maneuver to be completed
00:25:00	Begin SC separation/SLA jettison	0.5	
00:25:15	AUTO maneuver to docking attitude		2.0 deg/sec
	Null rates and translation		
	Initiate 0.7-fps closing rate		4-sec RCS +X
	Null 0.7-fps closing rate	0.7	-X RCS
00:35:00	Begin dock		
01:20:00	LM/CSM undock from S-IVB	1.1	Spring ejection and 3-sec -X RCS
01:43:00	S-IVB APS	9.4	Start TB8
02:04:20	S-IVB LOX		Local horizontal attitude Pitch = $183^\circ$ Yaw = $-5^\circ$ Roll = $180^\circ$



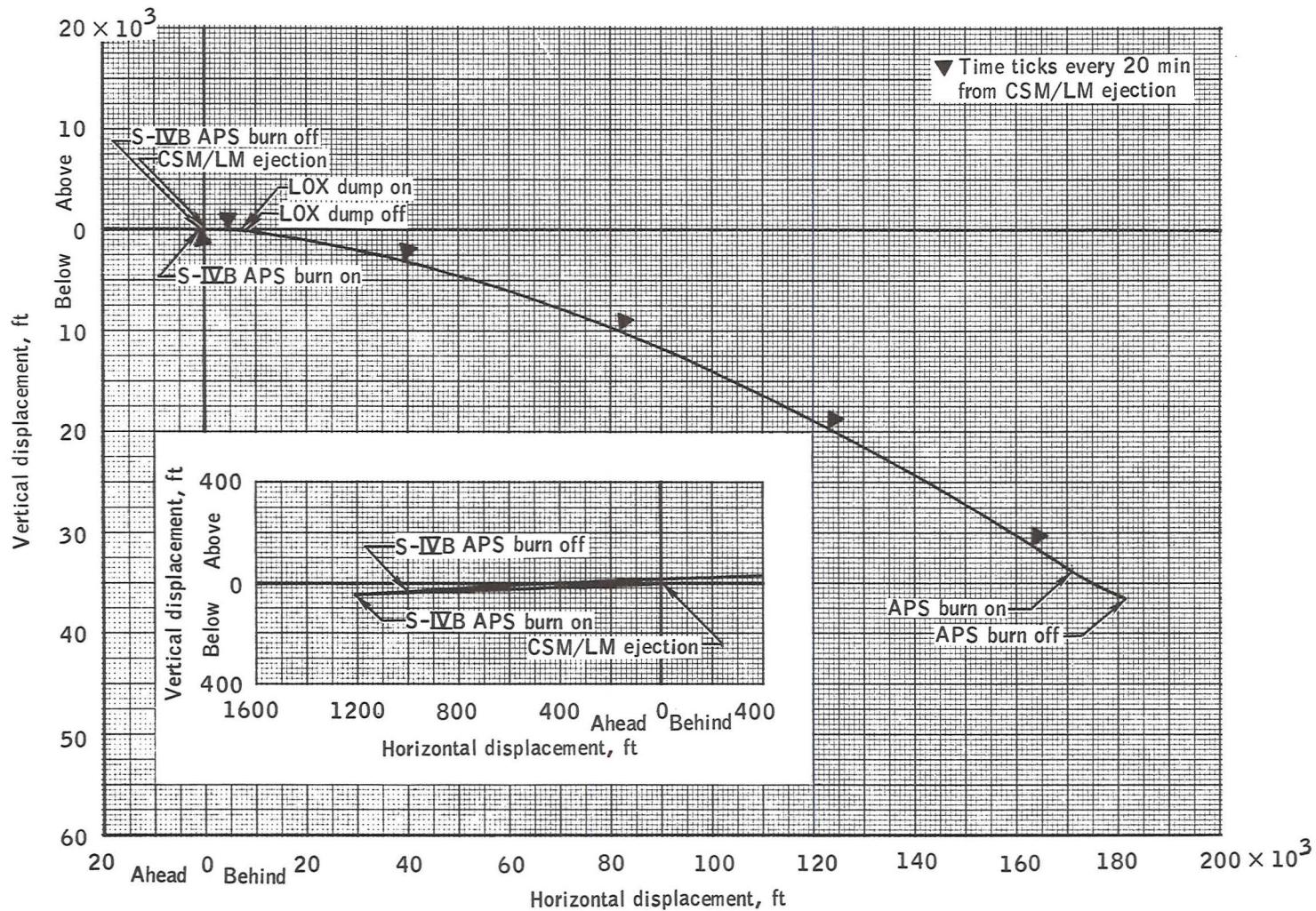
(a) Range versus time.

Figure 4.5-1.- Motion of the S-IVB relative to the spacecraft for the S-IVB evasive maneuver on Apollo 14.



(b) Cross range versus horizontal displacement.

Figure 4.5-1.- Continued.



4-27

(c) Vertical displacement versus horizontal displacement.

Figure 4.5-1.- Concluded.

4.6 TRANSLUNAR COAST

## 4.6 Translunar Coast

A groundtrack of the translunar coast phase is shown in figure 4.6-1. Time histories of altitude and true anomaly for the first 10 hours of translunar coast are provided in figures 4.6-2 and 4.6-3.

After TLI, the spacecraft is on a free-return circumlunar trajectory with a perilune altitude of 2030 n. mi. At TLI plus 28 hours, a transfer maneuver is performed which results in a relaxed-free-return translunar trajectory with a 57.1-n. mi. perilune altitude. The transfer trajectory is restricted such that, in the event an SPS maneuver could not be made, an earth return would be possible within the DPS docked  $\Delta V$  capability. It is assumed that the LM maneuver could be made at a minimum of 2 hours after perilune. The target load for this maneuver is presented in table 4.6-I.

The transfer maneuver is GNCS controlled, and external  $\Delta V$  guidance and the PTC REFSMMAT are used. The characteristics of the burn are the following.

Time, hr:min:sec, g.e.t. . . . .	30:36:07
Geodetic altitude, n. mi. . . . .	118 345
$\Delta V$ , fps . . . . .	73.4
Burn time, sec . . . . .	11.1
SPS propellant used, lb . . . . .	705.7
Plane change, deg . . . . .	0.2

Passive thermal control attitude will be maintained during most of the translunar coast phase. The PTC REFSMMAT is given in table 5-II.

Four midcourse correction maneuver points have been defined at the following times.

- a. TLI plus 9 hours (MCC-1)
  - b. TLI plus 28 hours (MCC-2) (the transfer maneuver)
  - c. LOI minus 22 hours (MCC-3)
  - d. LOI minus 5 hours (MCC-4)

The midcourse correction strategy has been defined and documented through the Apollo Data Priority Coordination Office. The maneuvers are GNCS controlled, and external  $\Delta V$  guidance is used. The PTC REFSMMAT (table 5-II) will be used for MCC-1, MCC-2, MCC-3, and MCC-4. The alignment criteria are presented in section 5.0.

The SC enters the lunar umbra approximately 53 minutes prior to LOI ignition but is back in sunlight 2 minutes before the time of ignition. Hawaii acquires approximately 6 minutes after TLI cutoff, and there is continual MSFN coverage until lunar occultation.

The duration of the translunar coast phase is 80 hours 2 minutes. Note that this duration violates the highly desirable ground rule of an even integer number of days for translunar flight to provide a more favorable crew work/rest cycle. However, a thorough review of mission design scans indicated that no even-integer day was possible without violation of other mandatory ground rules.

Geodetic altitude from translunar injection to LOI is shown in figure 4.6-4. Altitude above the lunar surface during the last 10 hours of translunar coast is shown in figure 4.6-5.

TABLE 4.6-I.- TARGET LOAD FOR RELAXED-FREE-RETURN MANEUVER

[Propulsion system: SPS; guidance: external  $\Delta V$ ]

## (a) Target

$t_{IG}$ , hr:min:sec, g.e.t. . . . .	30:36:07
$\Delta V_X$ , fps . . . . .	-5.3
$\Delta V_Y$ , fps . . . . .	4.0
$\Delta V_Z$ , fps . . . . .	73.1
Mass, lb . . . . .	97 228.8

## (b) REFSMMAT

Refer to table 5-II

(c) Gimbal angles at  $t_{IG}$ 

IGA, deg . . . . .	-12.8
MGA, deg . . . . .	-43.8
OGA, deg . . . . .	73.0

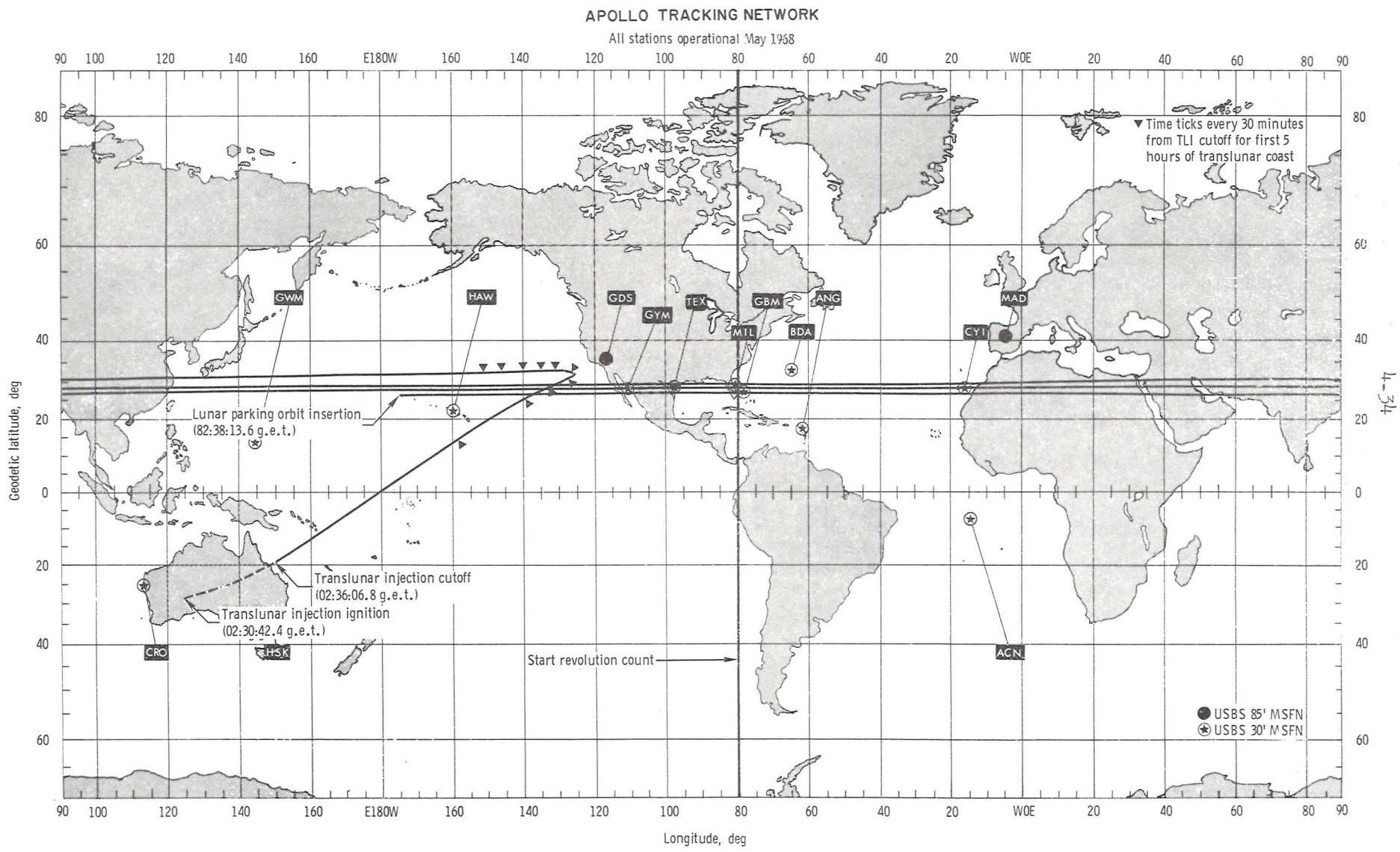


Figure 4.6-1.- Translunar coast groundtrack.

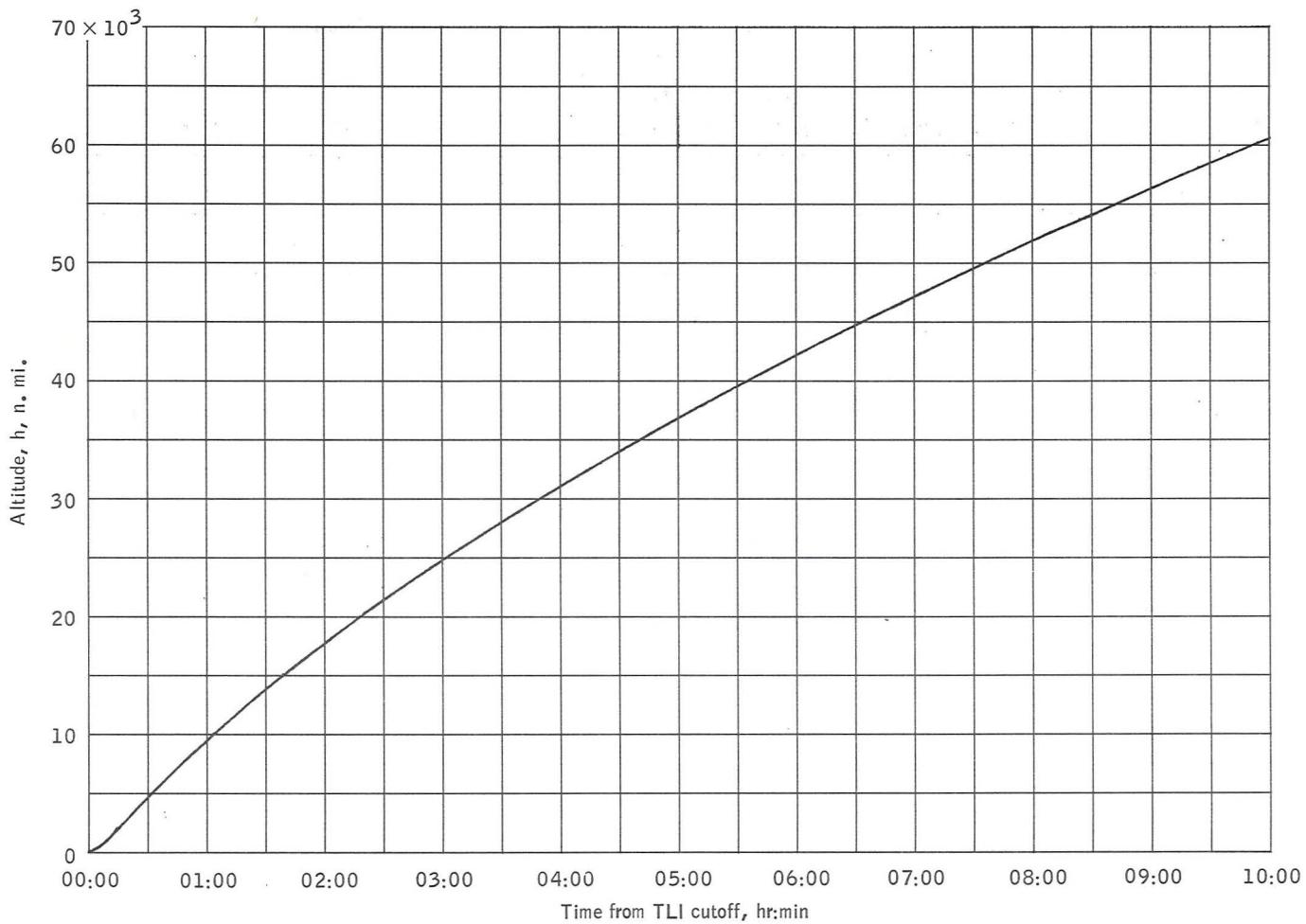


Figure 4.6-2.- Time history of altitude for first 10 hours of translunar coast.

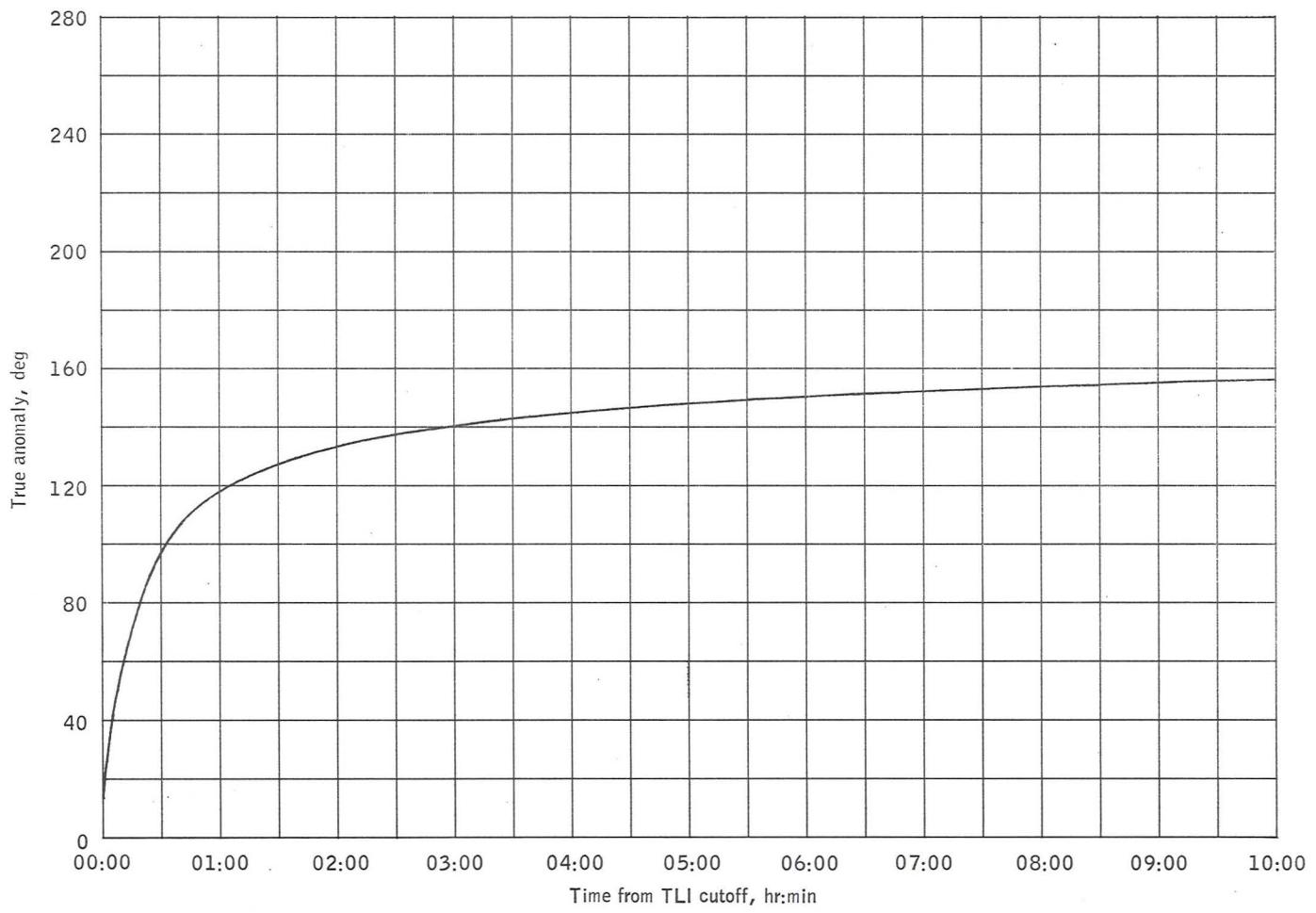


Figure 4.6-3.- Time history of true anomaly for first 10 hours of translunar coast.

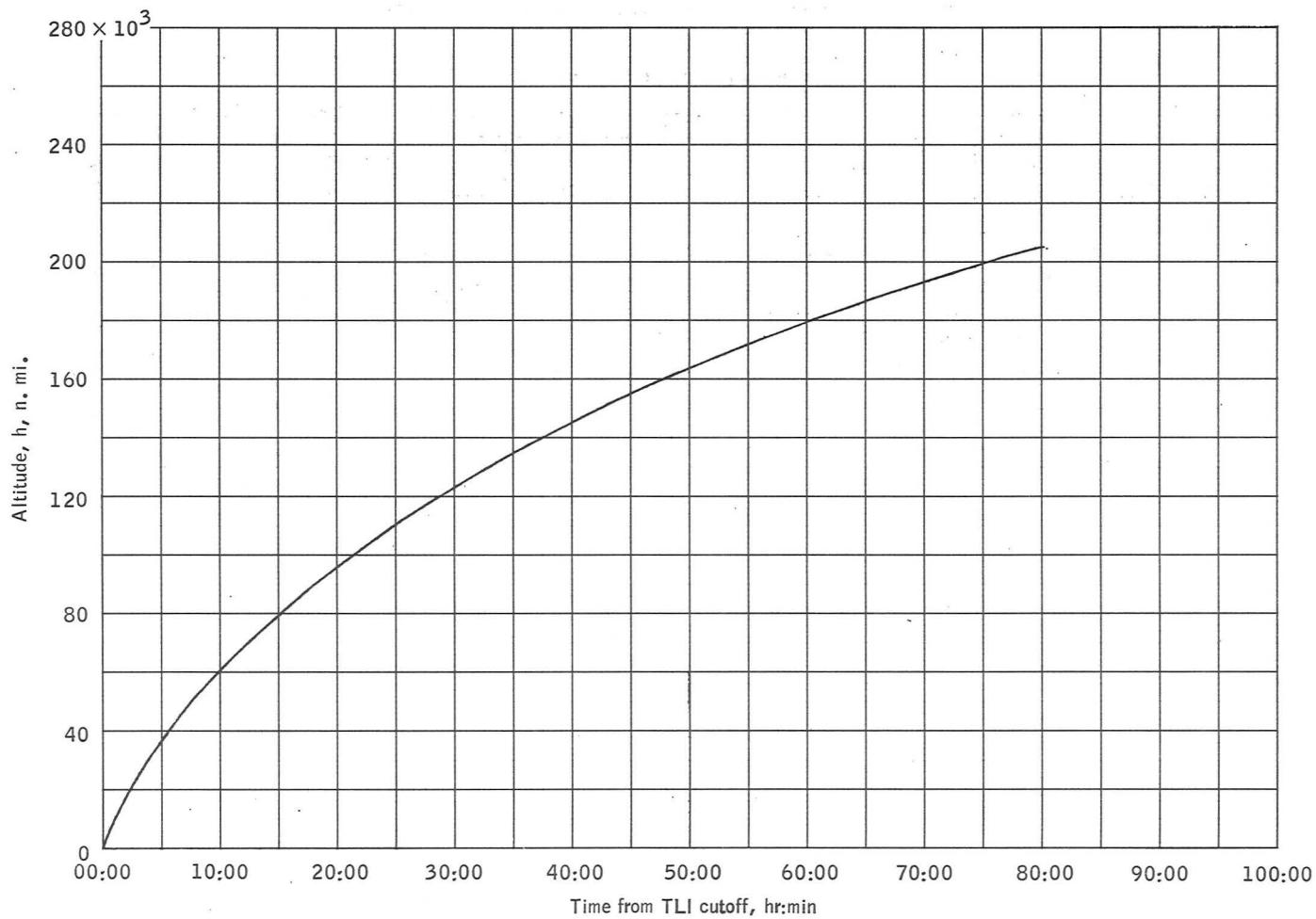


Figure 4.6-4.- Altitude time history from translunar injection to lunar orbit insertion.

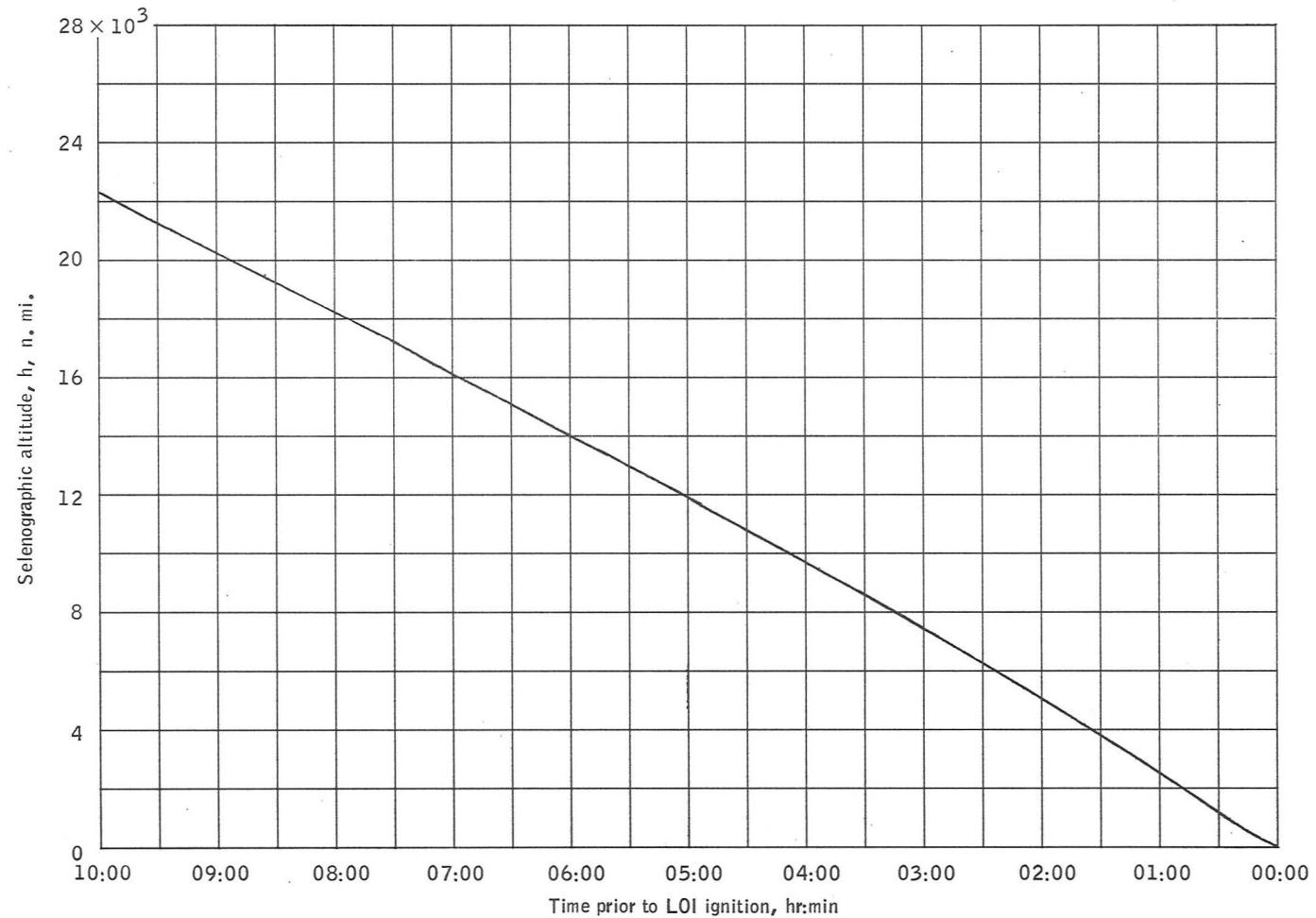


Figure 4.6-5.- Time history of altitude 10 hours prior to LOI ignition.

4.7 LUNAR ORBIT INSERTION

#### 4.7 Lunar Orbit Insertion (LOI)

The LOI maneuver is designed to insert the CSM into approximately a 57- by 170-n. mi. lunar parking orbit (LPO). The lunar orbit insertion geometry is shown in figure 4.7-1, and the lunar orbit groundtrack from LOI to LM landing is shown in figure 4.7-2. A time history of trajectory parameters during the burn is presented in figure 4.7-3. The burn was simulated with external  $\Delta V$  guidance. A description of the burn is as follows.

##### LOI burn initiation

Time, hr:min:sec, g.e.t. . . . .	82:38:13.6
Mass, lb . . . . .	96 377.7
Altitude above LLS, n. mi. . . . .	76.4
Selenographic latitude, deg . . . . .	2.2
Selenographic longitude, deg . . . . .	-178.2
Inertial velocity, fps . . . . .	8094.4
Perilune altitude above LLS, n. mi. . . . .	57.1
Selenographic inclination, deg . . . . .	3.2

##### LOI burnout - start LPO

Time, hr:min:sec, g.e.t. . . . .	82:44:20.2
Altitude above LLS, n. mi. . . . .	57.7
Selenographic latitude, deg . . . . .	-0.7
Selenographic longitude, deg . . . . .	158.2
Inertial velocity, fps . . . . .	5490.6
Selenographic inclination, deg . . . . .	14.1
Burn duration, sec . . . . .	366.6
Inertial burn arc, deg . . . . .	23.7
Plane change, deg . . . . .	-11.2
$\Delta V$ , fps . . . . .	2986.0
SPS propellant used, lb . . . . .	24 695
Orbit period, hr:min:sec . . . . .	02:08:27.4
Perilune altitude above LLS, n. mi. . . . .	57.1
Apolune altitude above LLS, n. mi. . . . .	170.0

The LOI burn parameters were computed without simulation of the SPS thrust buildup and tailoff. However, the effect of these is reflected in the burn parameters presented in the simulation data package (ref. 3).

The REFSMMAT used for the LOI burn as well as for the DOI burn is the lunar landing site alignment at the nominal lunar landing time.

The target load for the LOI burn is given in table 4.7-I. More detailed information about the burn, including reset points, navigation updates, and ignition gimbal angles, is given in reference 3.

TABLE 4.7-I.- TARGET LOAD FOR LOI MANEUVER

[Propulsion system: SPS; guidance: external  $\Delta V$ ]

## (a) Target

$t_{IG}$ , hr:min:sec, g.e.t. . . . .	82:38:13.6
$\Delta V_X$ , fps . . . . .	-2784.7
$\Delta V_Y$ , fps . . . . .	-1075.9
$\Delta V_Z$ , fps . . . . .	66.0
Mass, lb . . . . .	96 377.7

## (b) REF SMMAT

Refer to table 5-III

(c) Gimbal angles at  $t_{IG}$ 

IGA, deg . . . . .	-99.2
MGA, deg . . . . .	-33.2
OGA, deg . . . . .	-5.1

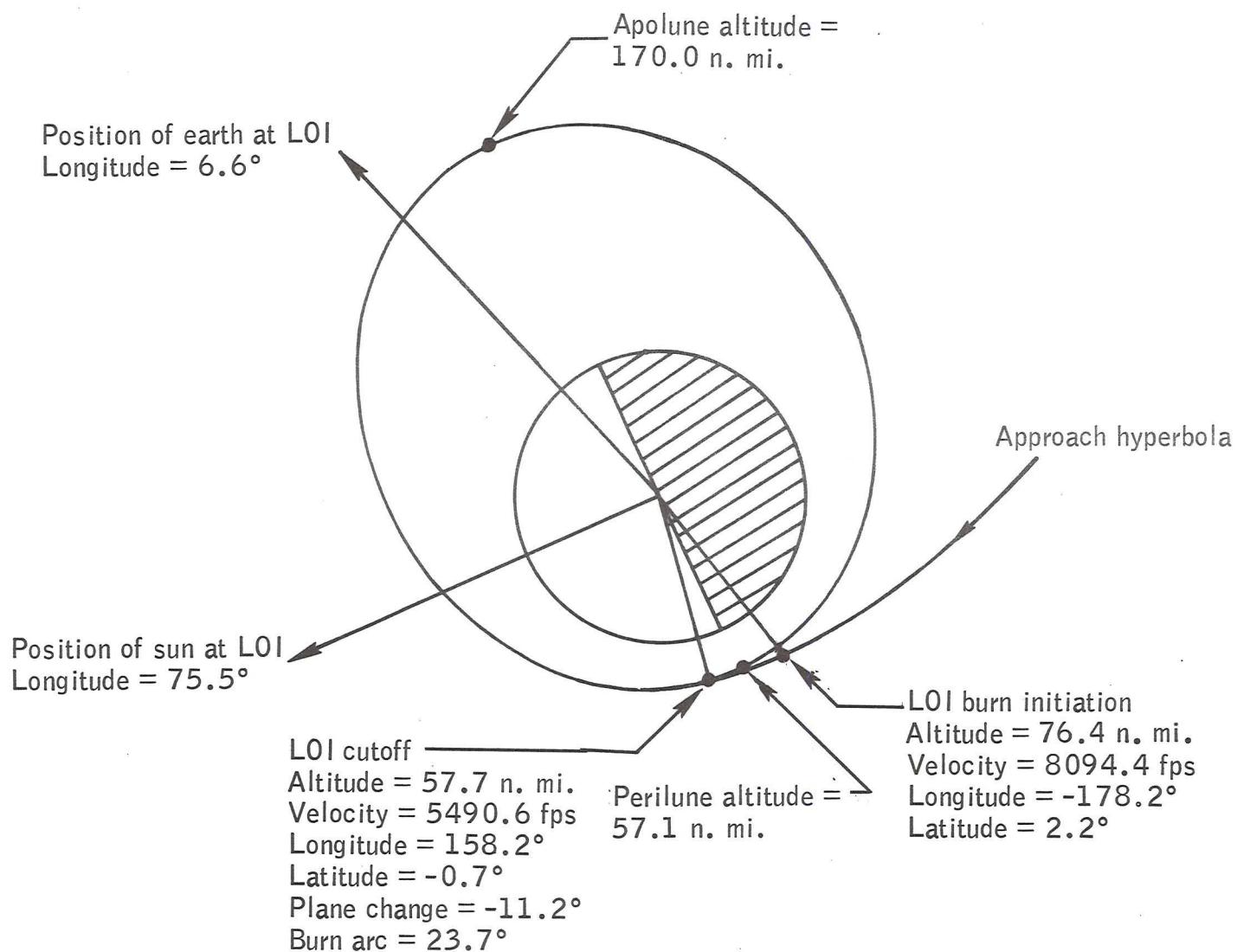


Figure 4.7-1.- LOI geometry.

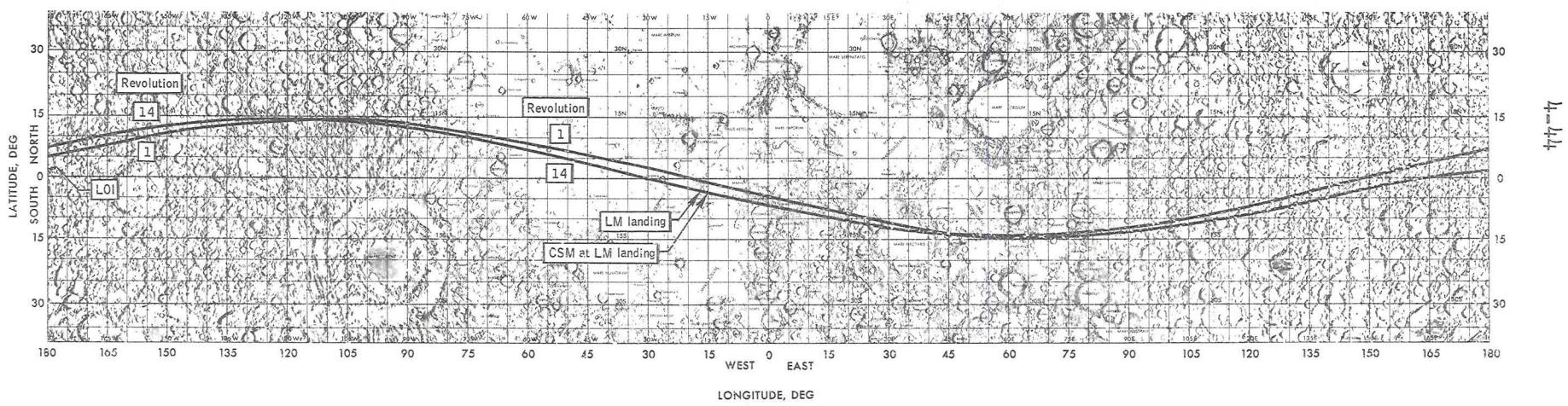
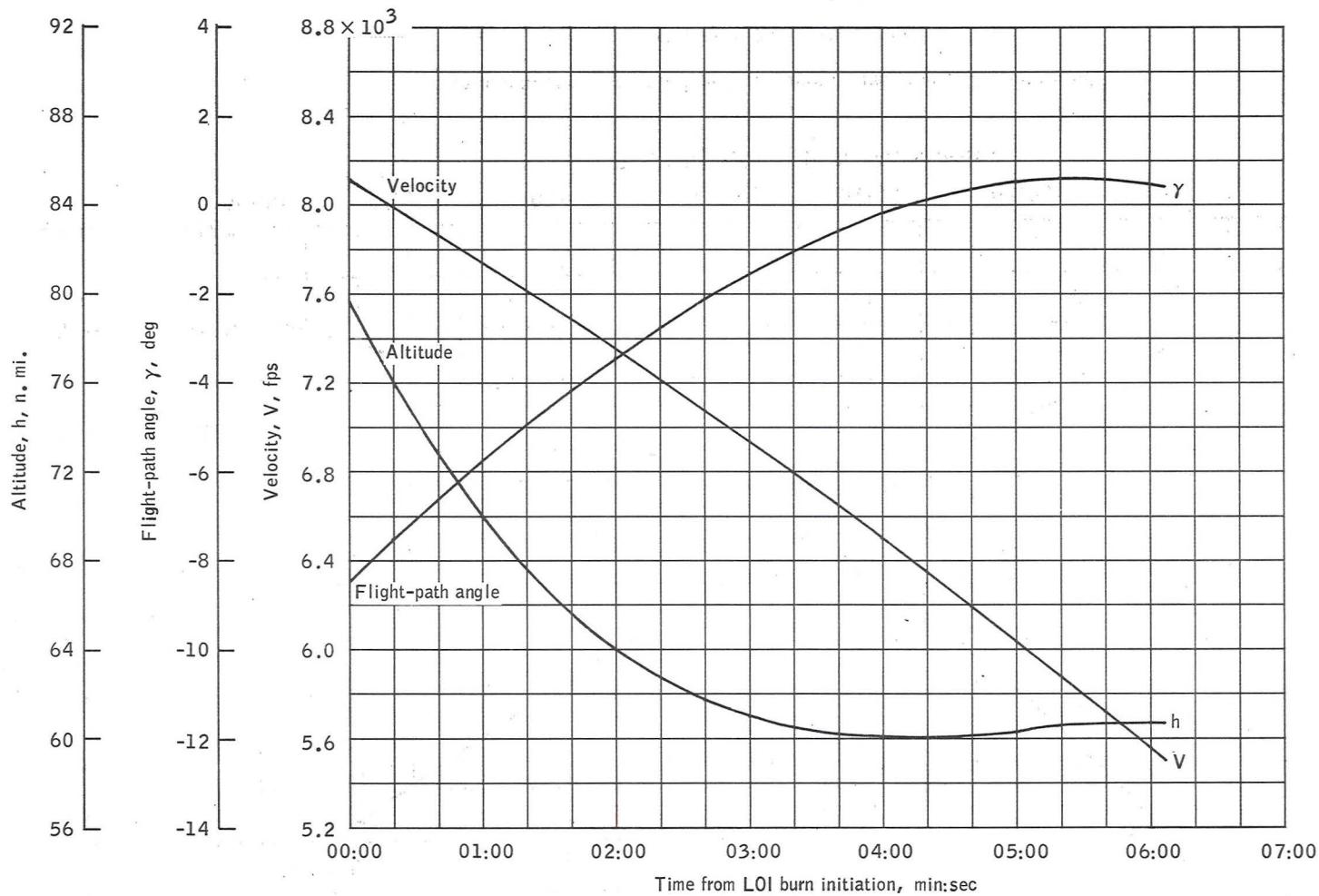
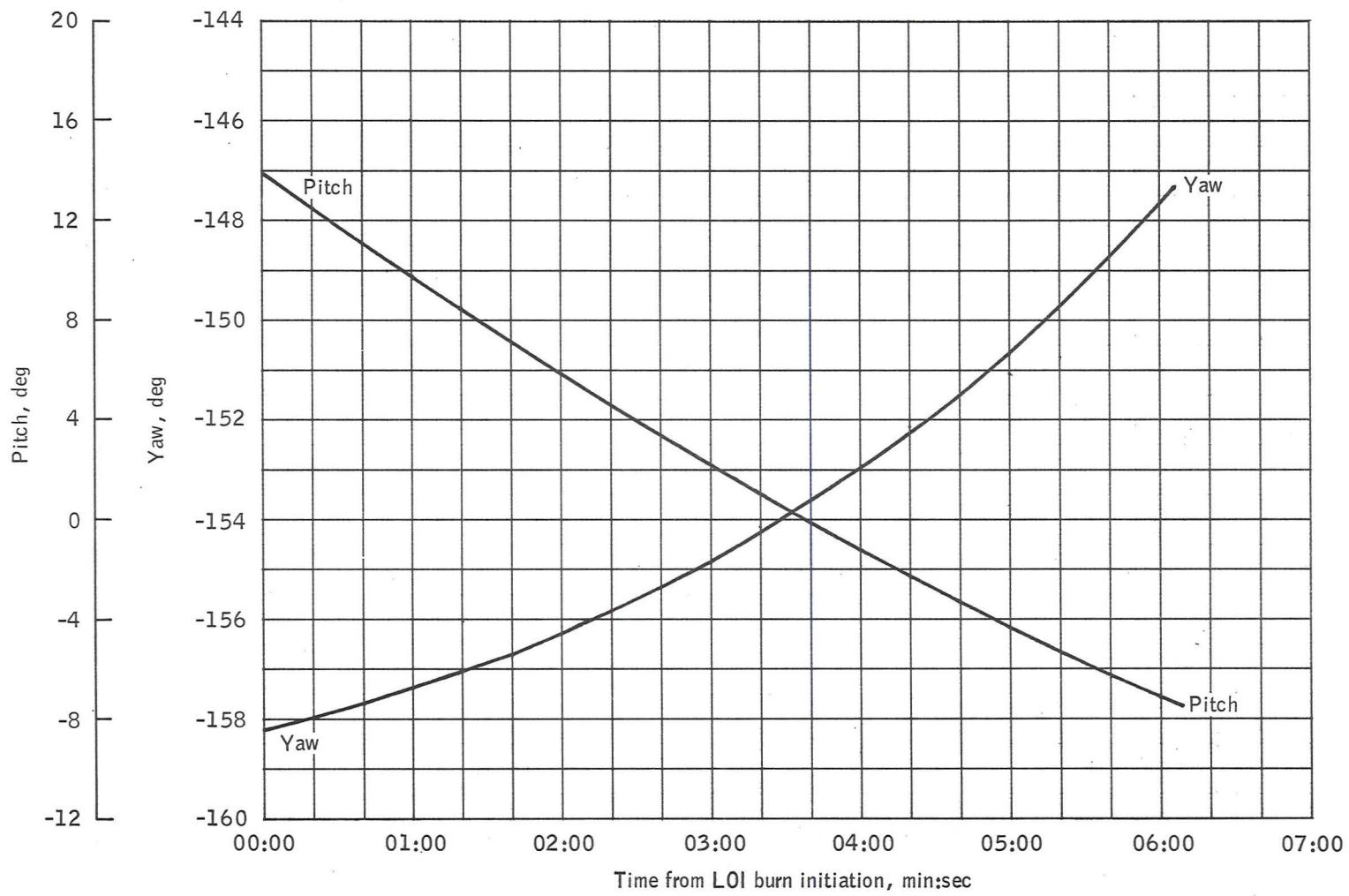


Figure 4.7-2.- Lunar orbit aroundtrack from LOI to LM landing.



(a) Velocity, flight-path angle, and altitude versus time from LOI burn initiation.

Figure 4.7-3.- Time history of trajectory parameters for LOI.



(b) Local horizontal pitch and yaw versus time from LOI burn initiation.

Figure 4.7-3.- Concluded.

**4.8 DESCENT ORBIT INSERTION**

#### 4.8 Descent Orbit Insertion (DOI)

The DOI maneuver occurs after two revolutions in the 57- by 170-n. mi. orbit. The targeting is biased to place the spacecraft in an initial 9.8- by 58.4-n. mi. lunar orbit so that the predicted lunar potential perturbation will place the spacecraft in a 50 000-foot by 60-n. mi. orbit at PDI on rev 14. The orbit is designed so that the perilune is at 2.5°W selenographic longitude at PDI. The altitudes are relative to the landing site radius.

For the DOI burn, the landing REFSMMAT (table 5-III) is used, and the CSM is oriented heads down. The burn is initiated near perilune of the second revolution in the 59- by 170-n. mi. orbit. The target load for the DOI maneuver is presented in table 4.8-I. The characteristics of the burn are the following.

##### DOI burn initiation

Time, hr:min:sec, g.e.t. . . . .	86:56:57.6
Mass, lb . . . . .	71 642.6
Altitude above LLS, n. mi. . . . .	57.3
Selenographic latitude, deg . . . . .	2.4
Selenographic longitude, deg . . . . .	168.6
Inertial velocity, fps . . . . .	5492.7
Perilune altitude above LLS, n. mi. . . . .	57.1
Apolune altitude above LLS, n. mi. . . . .	170.0

##### DOI burnout

Time, hr:min:sec, g.e.t. . . . .	86:57:18.9
Altitude above LLS, n. mi. . . . .	57.2
Selenographic latitude, deg . . . . .	2.2
Selenographic longitude, deg . . . . .	167.5
Inertial velocity, fps . . . . .	5287.7
Selenographic inclination, deg . . . . .	14.2
Burn duration, sec . . . . .	21.4
Inertial burn arc, deg . . . . .	1.1
Plane change, deg . . . . .	0.0
ΔV, fps . . . . .	206.6
SPS propellant used, lb . . . . .	1452.2
Orbit period, hr:min:sec . . . . .	01:54:00
Perilune altitude above LLS, n. mi. . . . .	9.8
Apolune altitude above LLS, n. mi. . . . .	58.4

TABLE 4.8-I.- TARGET LOAD FOR DOI MANEUVER

[Propulsion system: SPS; guidance: external  $\Delta V$ ]

## (a) Target

$t_{IG}$ , hr:min:sec, g.e.t. . . . .	86:56:57.6
$\Delta V_X$ , fps . . . . .	-205.7
$\Delta V_Y$ , fps . . . . .	0.0
$\Delta V_Z$ , fps . . . . .	19.3
Mass, lb . . . . .	71 642.6

## (b) REFSMMAT

Refer to table 5-III

(c) Gimbal angles at  $t_{IG}$ 

IGA, deg . . . . .	-90.7
MGA, deg . . . . .	-.4
OGA, deg . . . . .	-.2

4.9 CSM/LM COAST FROM LOI TO UNDOCK

## 4.9 CSM/LM Coast from LOI to Undock

The main activities in lunar orbit prior to undock include the CSM DOI maneuver at the beginning of the third revolution, photography of Descartes, and a rest period. The following is a summary of major activities.

Activity	Revolution number	Time, hr:min, g.e.t.
LOI	Beginning of 1	82:38
Landmark tracking	2	85:50
DOI	Beginning of 3	86:57
Descartes photography	4	89:35
Begin rest period	5	91:00
End rest period	9	99:30
Begin IVT to LM	10	101:50
Undock	12	104:28

4.10 CSM/LM UNDOCK TO LANDING

#### 4.10 CSM/LM Undock to Landing

The interval of the mission from CSM/LM undocking to the LM landing on the lunar surface is described in this section. The maneuvers included are CSM/LM undock and separation, in preparation for the LM descent; CSM circularization, to place the CSM in a near circular orbit; and the LM powered descent to the landing site at Fra Mauro. In figure 4.10-1, the location of these maneuvers in lunar orbit is shown with respect to the landing site, MSFN tracking, and sunlight or darkness. Also indicated are the CSM and LM positions at each maneuver.

The LM MSFN tracking summary and LM shadow data from undocking to landing are presented in tables 4.1-V and 4.1-VI, respectively.

Time histories of the LM velocity, altitude above the landing site, and flight-path angle for the time period from undocking to landing are shown in figure 4.10-2(a); major events also are indicated. Time histories of the CSM/LM range, range rate, and phase angle are shown in figure 4.10-2(b). The CSM/LM relative motion (in curvilinear coordinates) is shown in figure 4.10-3.

More details of each maneuver in the time period from undocking to landing are given in the following subsections.

**4.10.1 CSM/LM undock and separation.**- The undocking and separation maneuver is performed as a single maneuver to separate the LM and CSM in preparation for the LM descent. The maneuver occurs at 12.5°S and 88.0°E during the twelfth orbit 4.2 hours before landing. The required data for this maneuver is given in table 4.10-I.

Undocking is accomplished by simultaneously releasing a set of clamps that allows springs to push the two spacecraft apart. Immediately after undocking, the CSM performs a manual maneuver directed radially downward toward the center of the moon to achieve a total  $\Delta V$  of 1.0 fps. This maneuver provides an increase in LM-CSM separation distance.

**4.10.2 CSM circularization.**- This maneuver, which occurs at 9.7°N and 168.2°W and 79 minutes after undocking, transfers the CSM from the initial elliptical orbit to a near circular parking orbit which is necessary for two reasons: (1) landmark tracking is less difficult at higher altitudes and (2) after the powered ascent (or a descent abort), the rendezvous phasing is improved.

The maneuver is a 3.8-second external  $\Delta V$  burn performed by the SPS. The target load for the maneuver is shown in table 4.10-II.

4.10.3 Powered descent.- The powered descent maneuver (PDI) which consists of an ullage phase, trim phase, braking phase, approach phase, and the landing phase is initiated at a central angle of  $14.6^\circ$  from the landing site at Fra Mauro ( $3.67^\circ\text{S}$ ,  $17.46^\circ\text{W}$ ) and 50 000 feet above the landing site. The powered descent profile is shown in figure 4.10-4.

4.10.4 Ullage and trim phases.- A 7.5-second ullage maneuver is performed with the LM RCS to settle the DPS propellants. The DPS is then ignited and burned at the minimum throttle setting (approximately 10 percent) for 26 seconds to allow time for the DPS engine gimbal to trim through the cg. The ullage and trim are performed at the constant inertial attitude required by PGNCS guidance at DPS ignition.

4.10.5 Braking phase.- The braking phase is designed for efficient reduction of orbital velocity and, therefore, uses maximum thrust from the DPS for most of the phase; however, the DPS is throttled during the final 2 minutes of this phase for guidance control. The spacecraft is in a windows-up attitude so that the landing radar can be used; however, the crew has the option to be windows down for a maximum of 4.5 minutes after DPS ignition if they prefer to view the lunar surface. The targets for the braking phase are listed in table 4.10-III. However, these targets are not actually achieved because of increased guidance sensitivity as time to go (to achieve targets) becomes small. The guidance automatically switches from the braking phase targets to the approach phase targets when time to go becomes less than 62 seconds.

4.10.6 Approach phase.- The approach phase, which begins at an altitude of approximately 7500 feet (high gate), provides for pilot visual monitoring of the approach to the lunar surface. That is, the guidance is targeted to provide a  $30^\circ$  pitch transient at high gate to immediately bring the landing area into view in the forward window. Then the guidance provides an attitude which will maintain the landing area in the pilot's window until late in the approach phase. The quadratic acceleration guidance law is used in the approach phase as well as the braking phase. The targets for the approach phase are listed in table 4.10-III. Note that the altitude target is approximately 95 feet. However, the approach phase is operationally considered to be terminated at a time to landing of 80 seconds (low gate, altitude  $\approx 650$  ft) where the landing phase is initiated.

4.10.7 Landing phase.- The landing phase is designed to provide continued visual assessment of the landing site and to provide compatibility for pilot takeover from the automatic control. There is no change in the guidance law or targets at this point because the approach phase targets have been selected to satisfy these additional constraints. Under automatic guidance, a vertical descent is initiated when time-to-go becomes less than 12 seconds. The vertical descent portion of the landing

phase starts at an altitude of 100 feet and terminates at landing. The vertical descent guidance is a velocity nulling routine which nulls horizontal velocity while accepting the commanded vertical rate from the rate-of-descent (ROD) switch. Normally, a 3-fps rate of descent is used throughout the vertical descent.

4.10.8 Graphical description of the powered descent. - Shown in figure 4.10-5 is the LM lunar groundtrack from PDI to landing. Events indicated are PDI, high gate, and landing.

The time histories of the LM and LM-CSM trajectory parameters from PDI to landing are shown in figure 4.10-6. Major events indicated are DPS throttle recovery, high gate, low gate, and landing. In figure 4.10-6(a), time histories of the LM velocity, altitude above the landing site, and flight-path angle are shown. The time histories of the LM-CSM range, range rate, and phase angle are shown in figure 4.10-6(b). Time histories of the LM thrust angle and FDAO pitch angle are shown in figure 4.10-6(c). The thrust angle is the angle between the thrust vector and the local vertical, and the FDAO pitch angle is the angle between the X-body axis and the X-platform axis. The time histories of the commanded and actual thrust profiles are shown in figure 4.10-6(d). The two profiles are different because the DPS cannot be throttled above 63-percent thrust (100 percent is 10 500 lb of thrust). Therefore, the throttle is set at the maximum level (approximately 93 percent thrust) until the commanded value is below 57-percent thrust, then the DPS is throttled to the commanded values for the remainder of the descent. The 57-percent level allows margin for throttle operation without requiring a pulse back to FTP.

In figure 4.10-7, the LM altitude/altitude rate profile from PDI to landing is shown. Events indicated are PDI, DPS throttle recovery, high gate, low gate, and landing.

Shown in figure 4.10-8 is the S-band gimbal profile indicating the position in which the S-band antenna must be to maintain S-band communication. Also shown is the vehicle blockage pattern. Any antenna positions inside this pattern will result in loss of communication because parts of the LM hardware will reflect the signal. Events indicated are PDI, DPS throttle recovery, high gate, low gate, and landing.

To decrease the effects of the local terrain on the descent trajectory, a simple terrain model (called an a priori model) has been introduced into the guidance system. This model allows the guidance to compensate in advance for general terrain variations thus providing a smoother trajectory. Figure 4.10-9 shows the terrain for a  $-76^\circ$  approach azimuth into the landing site at Fra Mauro. The a priori model (a series of slopes) has been superimposed over the actual terrain. An enlargement of the terrain from high gate to landing has been included for better illustration.

An approach phase summary (from high gate to low gate) is presented in figure 4.10-10. The LM landing point designator (LPD), LM altitude above the landing site, and the lunar terrain altitude above the landing site as functions of surface range to the landing site are presented as figures. A table presents the LM altitude rate, thrust angle, and horizontal velocity as functions of time from high gate.

A map showing the topography around the landing site at Fra Mauro is presented in figure 4.10-11. The LM approach path is shown from a surface range of approximately 6000 feet to the landing site. Several prominent topographical features are also indicated.

A landing phase summary (from low gate to landing) is presented in figure 4.10-12. The LM LPD, LM altitude, and terrain altitude as functions of surface range are presented in figures. The altitude versus altitude rate profile along with the APS abort boundary are also presented in a figure. A table presents the LM thrust angle and horizontal velocity as functions of time from low gate.

TABLE 4.10-1.- DATA FOR MANUAL CSM SEPARATION MANEUVER

[Propulsion system: SM RCS]

(a) Desired  $\Delta V$ 's

(b) CSM REF SMMAT

Refer to table 5-III

(c) CSM gimbal angles at t<sub>TG</sub>

IGA, deg	101.7
MGA, deg	0.0
OGA, deg	0.0

TABLE 4.10-II.-- TARGET LOAD FOR CSM CIRCULARIZATION MANEUVER

[Propulsion system: SM SPS; guidance: external  $\Delta V$ ]

## (a) Target

$t_{IG}$ , hr:min:sec, g.e.t.	105:46:48.5
$\Delta V_x$ , fps	71.0999
$\Delta V_y$ , fps	0.0
$\Delta V_z$ , fps	13.9706
CSM weight at $t_{IG}$ , lb	35 937.718

## (b) REF SMMAT

Refer to table 5-III

(c) Gimbal angles at  $t_{IG}$ 

IGA, deg	109.7
MGA, deg	-1.3
OGA, deg	0.1

TABLE 4.10-III.- TARGET LOAD FOR LM POWERED DESCENT MANEUVER<sup>a</sup>

[Propulsion system: LM DPS; guidance programs: P63, P64, P66]

(a) Ignition targets for PDI (P63)

<sup>a</sup>For complete REFSSMAT, refer to table 5-III.

TABLE 4.10-III.- TARGET LOAD FOR LM POWERED  
DESCENT MANEUVER - Continued

[Propulsion system: LM DPS; guidance programs: P63, P64, P66]

(b) Braking phase targets (P63)<sup>a</sup>

Position, ft

### Velocity, fps

Acceleration, ft/sec<sup>2</sup>

Jerk, ft/sec<sup>3</sup>

<sup>a</sup>TENDBRAK = 62 sec (time-to-go at program switch).

TABLE 4.10-III.- TARGET LOAD FOR LM POWERED

## DESCENT MANEUVER - Concluded

[Propulsion system: LM DPS; guidance programs: P63, P64, P66]

(c) Approach phase targets (P64)<sup>a</sup>

## Position, ft

$R_x$	94.9191
$R_y$	0.0
$R_z$	-15.7208

## Velocity, fps

$V_x$	2.08358
$V_y$	0.0
$V_z$	0.830318

Acceleration, ft/sec<sup>2</sup>

$A_x$	0.540285
$A_y$	0.0
$A_z$	-0.235423

Jerk, ft/sec<sup>3</sup>

$J_z$	4.5092 <sup>-2</sup>
-------	----------------------

<sup>a</sup>TENDAPP = 12 sec (time-to-go at program switch).

## (d) Vertical descent targets (P66)

## Velocity, fps

$V_x$	-3.0
$V_y$	0.0
$V_z$	0.0

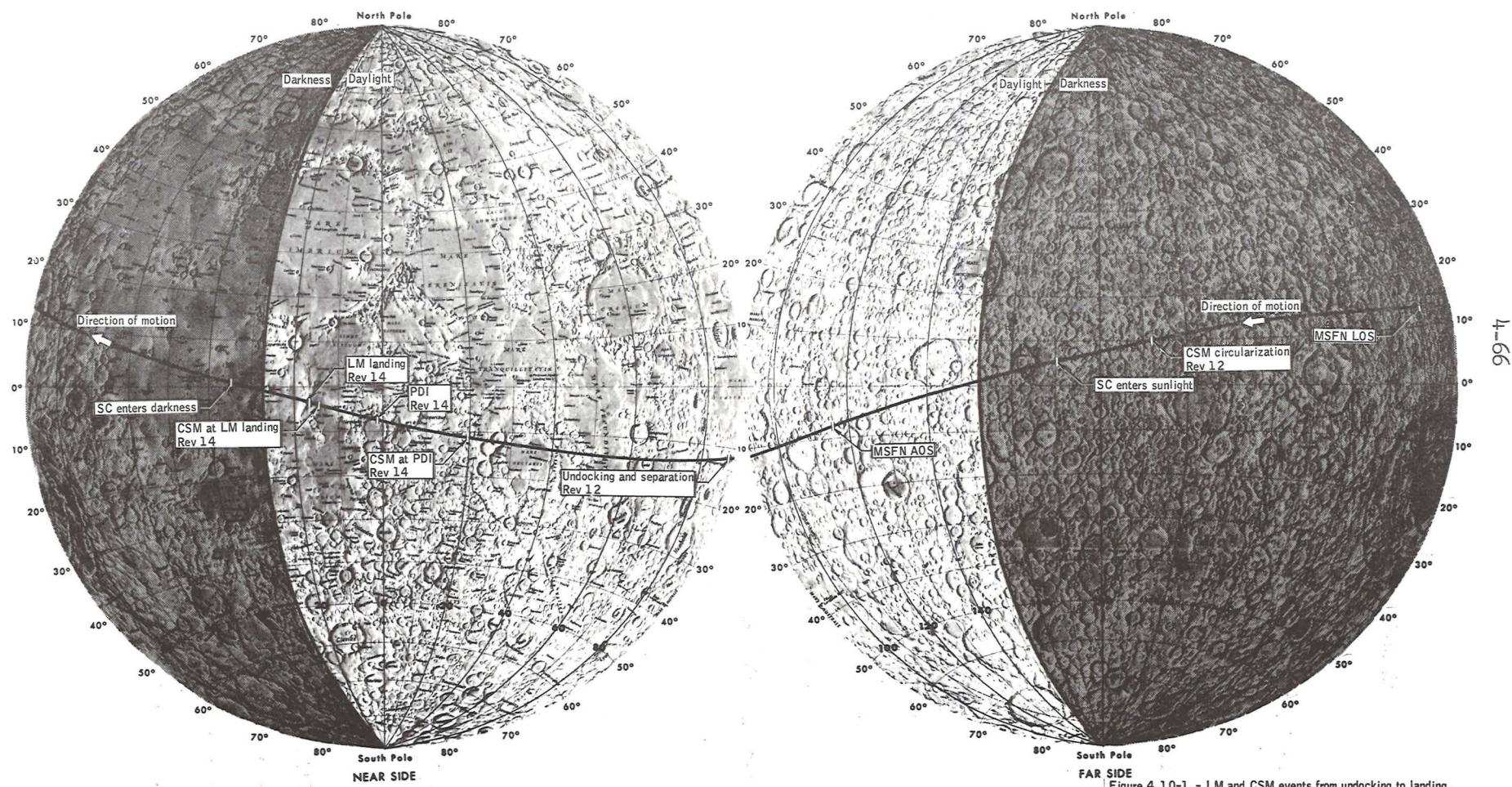
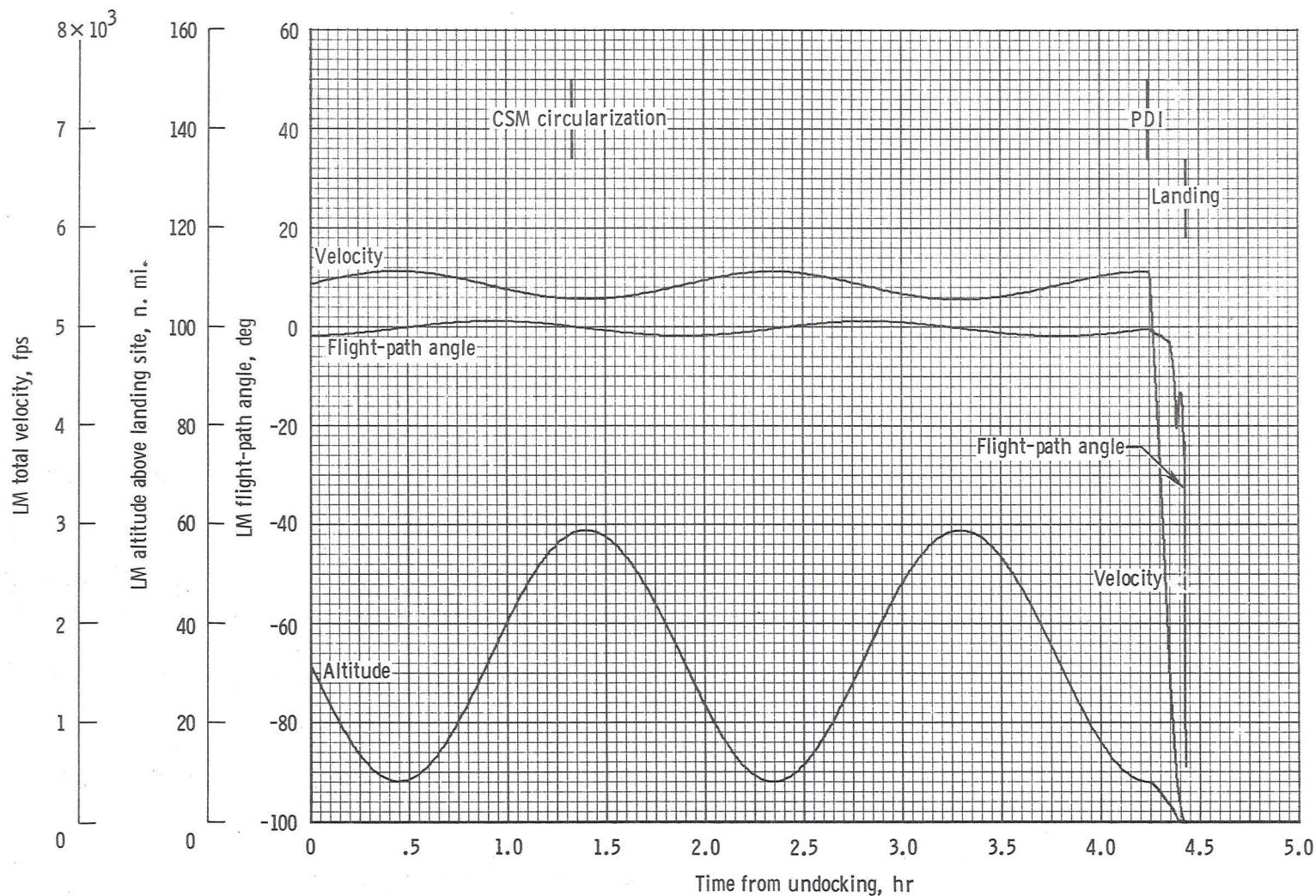
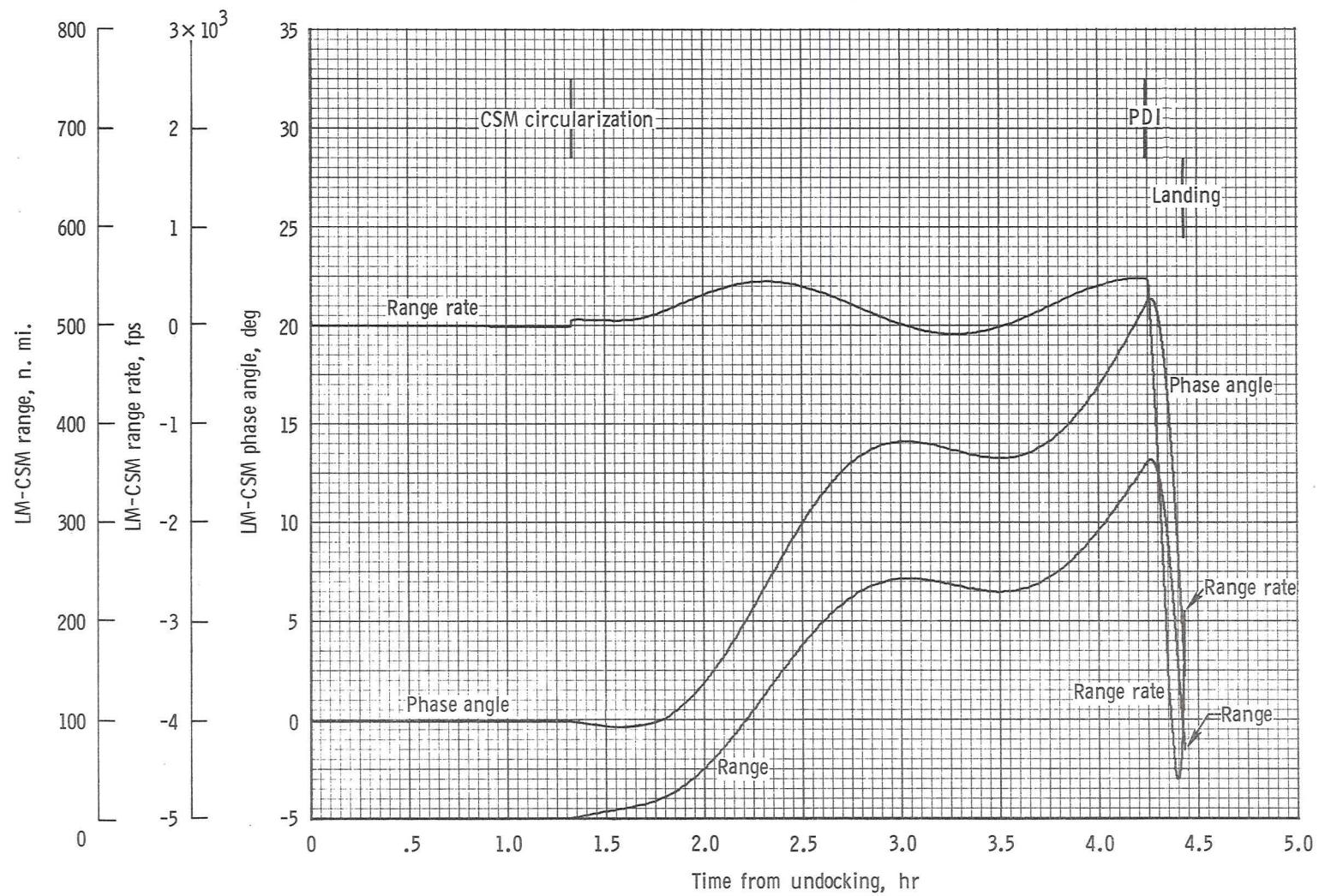


Figure 4.10-1.- LM and CSM events from undocking to landing.



(a) LM total velocity, altitude above landing site, and flight-path angle.

Figure 4.10-2. - Time history of LM and LM-CSM trajectory parameters from undocking to landing.



(b) LM-CSM range, range rate, and phase angle.

Figure 4.10-2.- Concluded.

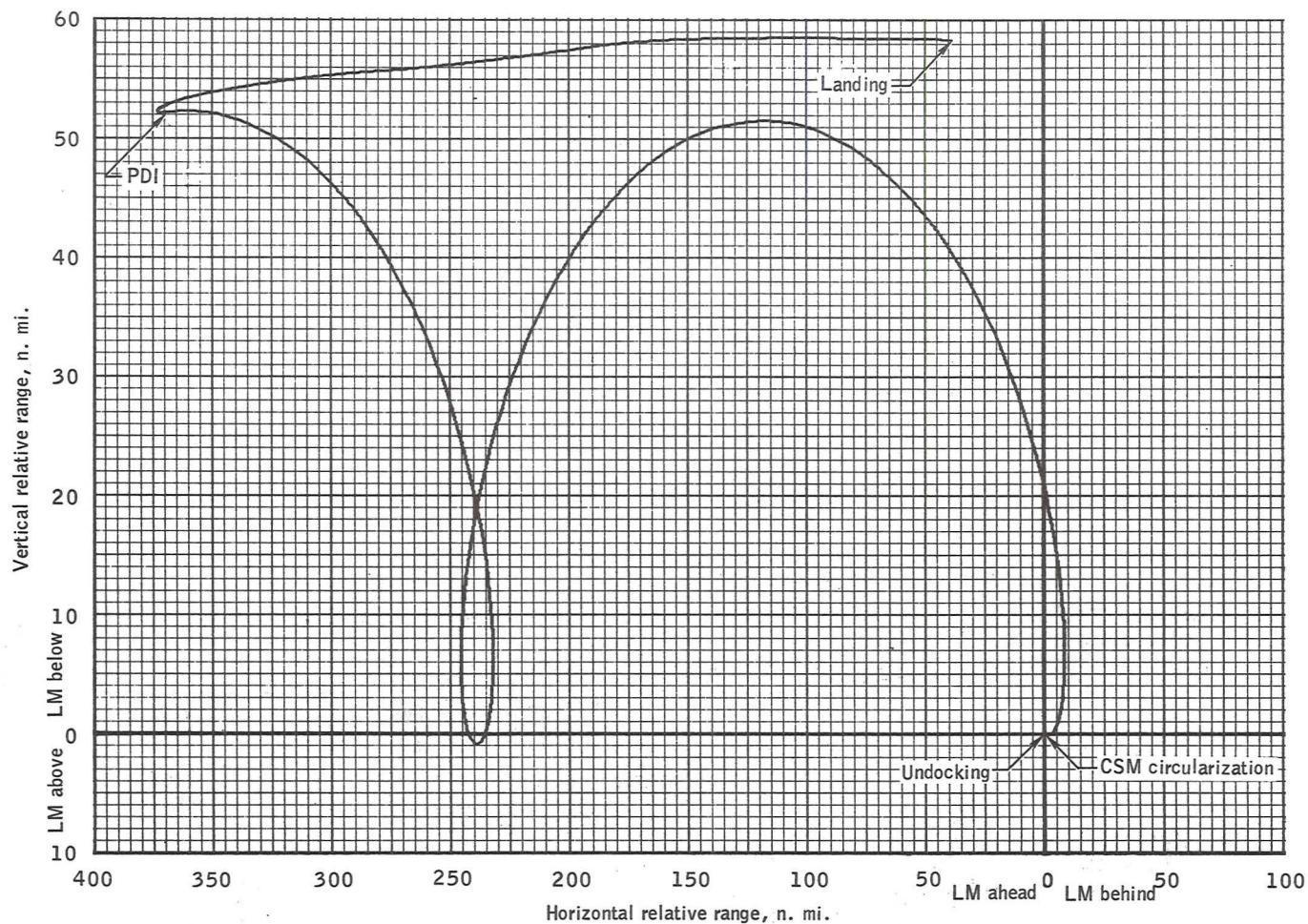
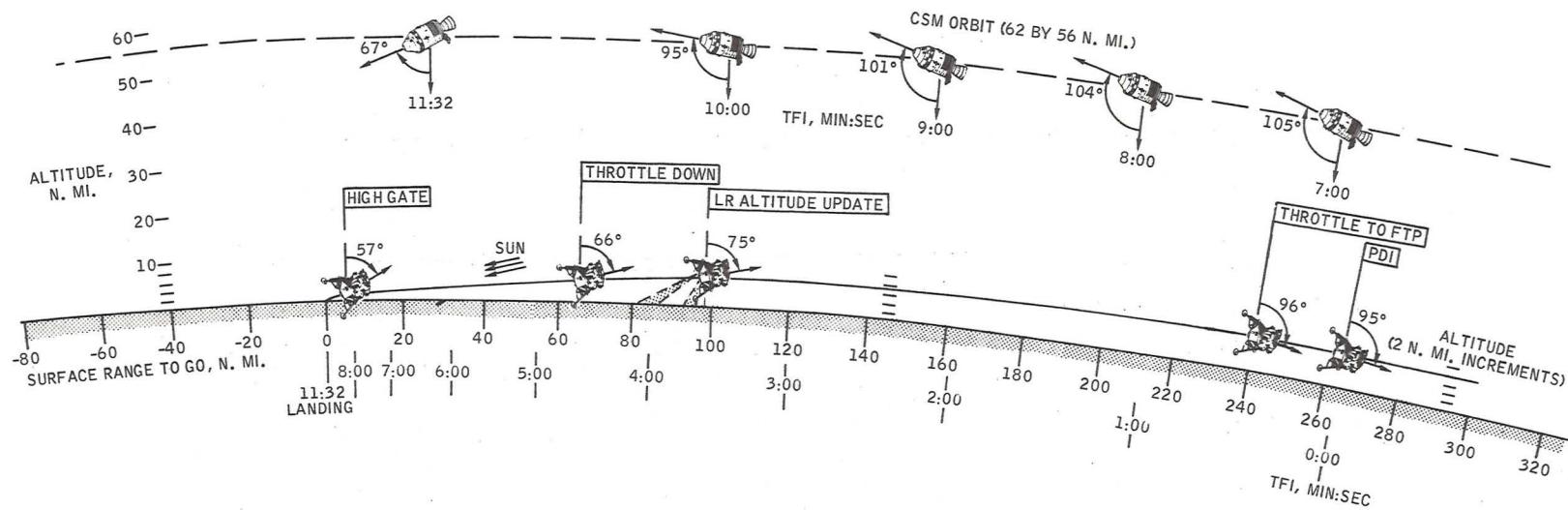


Figure 4.10-3.- LM-CSM relative motion from undocking to landing (curvilinear coordinate system).



SUMMARY					
EVENT	TFI, MIN:SEC	V <sub>x</sub> , FPS	H, FPS	H, FT	ΔV, FPS
POWERED DESCENT INITIATION	0:00	5560	-2	50 027	0
THROTTLE TO MAXIMUM THRUST	0:26	5526	-3	49 964	34
LANDING RADAR ALTITUDE UPDATE	3:36	3601	-89	37 779	1992
LANDING RADAR VELOCITY UPDATE	5:12	2500	-94	28 783	3161
THROTTLE RECOVERY	6:34	1439	-65	21 430	4289
HIGH GATE	8:34	450	-153	7 602	5436
LOW GATE	10:12	70 (82)*	-20	659	6184
LANDING	11:32	-15 (0)*	-3	5	6637

\*(HORIZONTAL VELOCITY RELATIVE TO SURFACE)

Figure 4.10-4. - Powered descent profile.

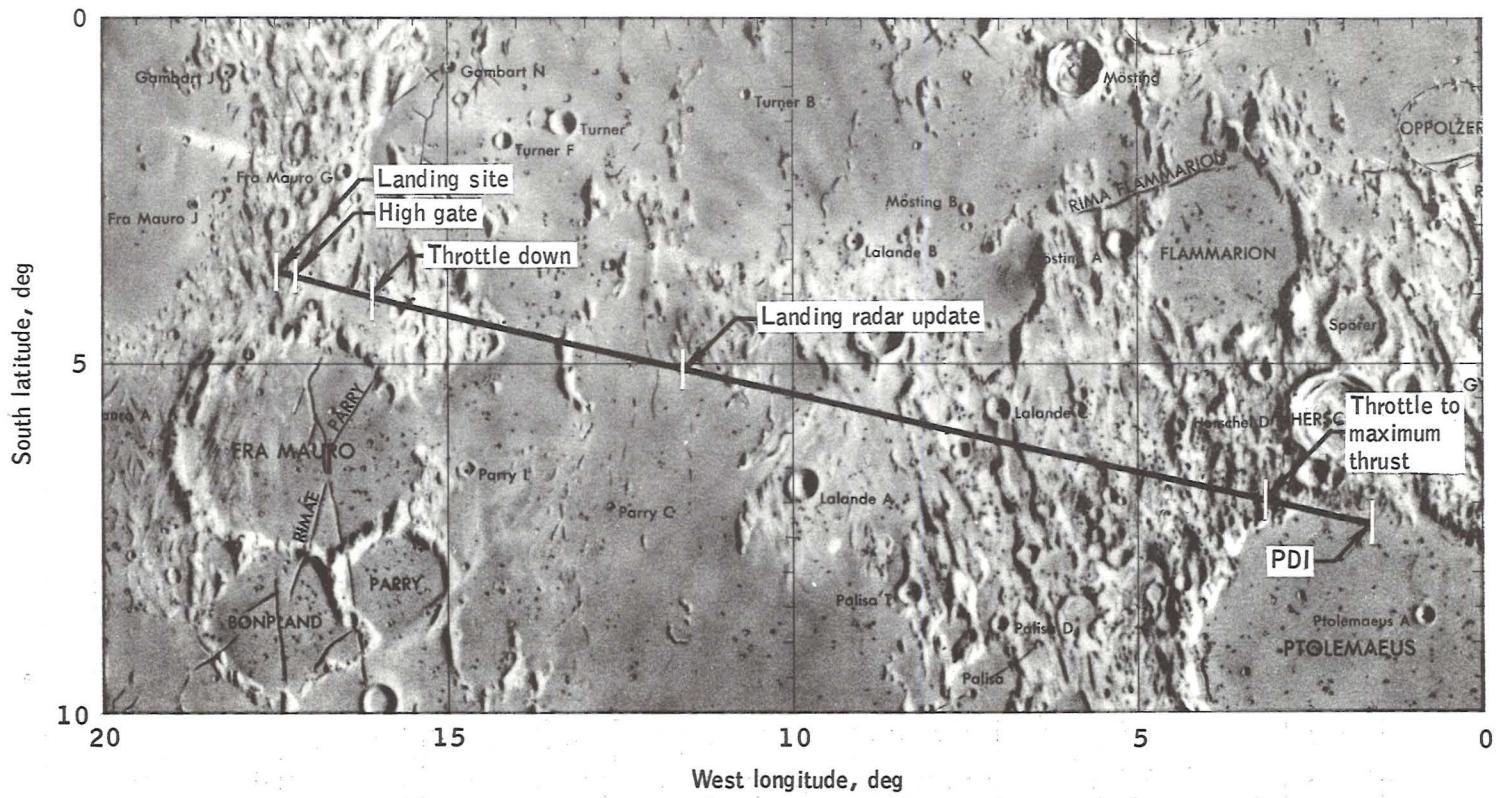
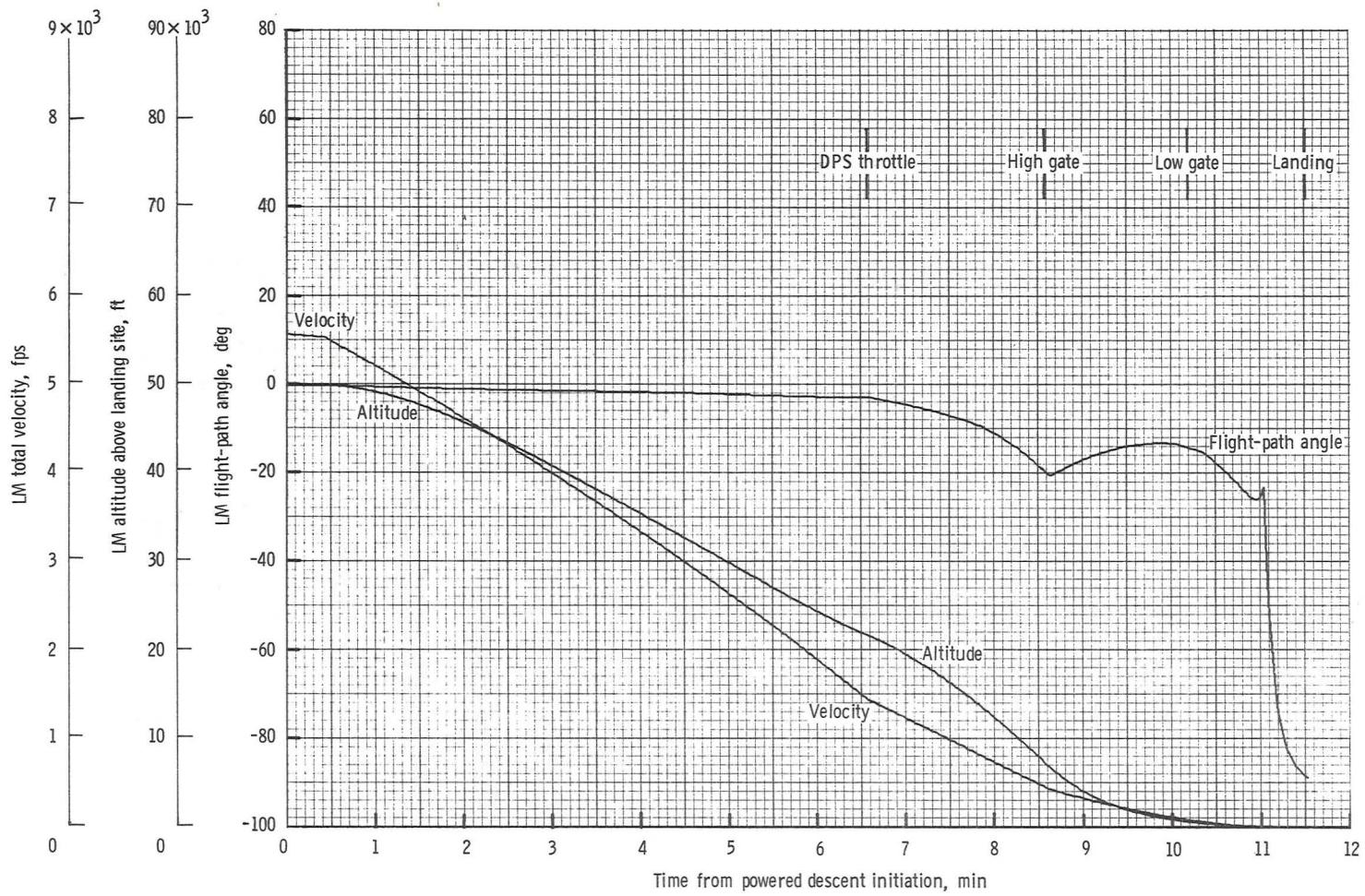
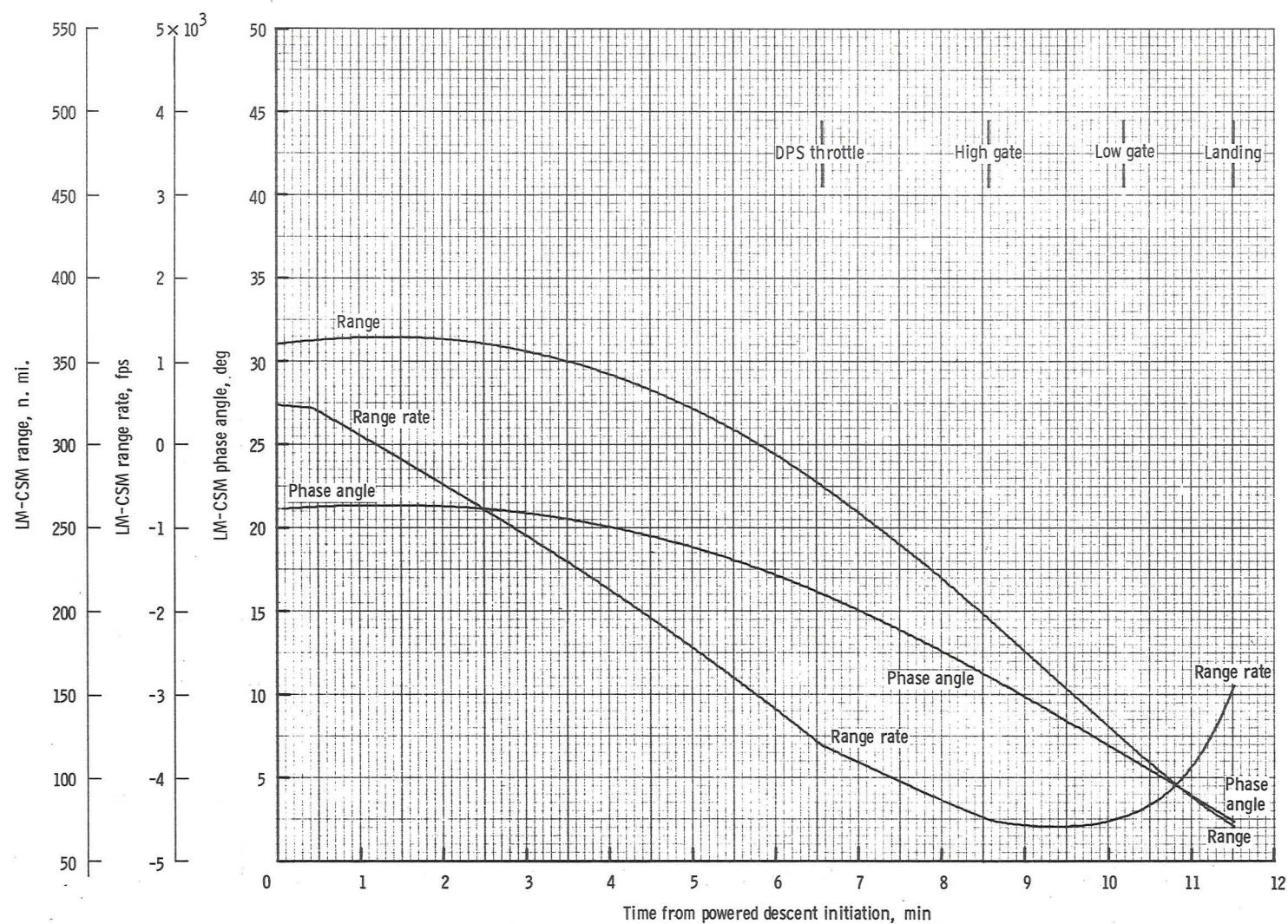


Figure 4.10-5.- LM powered descent groundtrack.



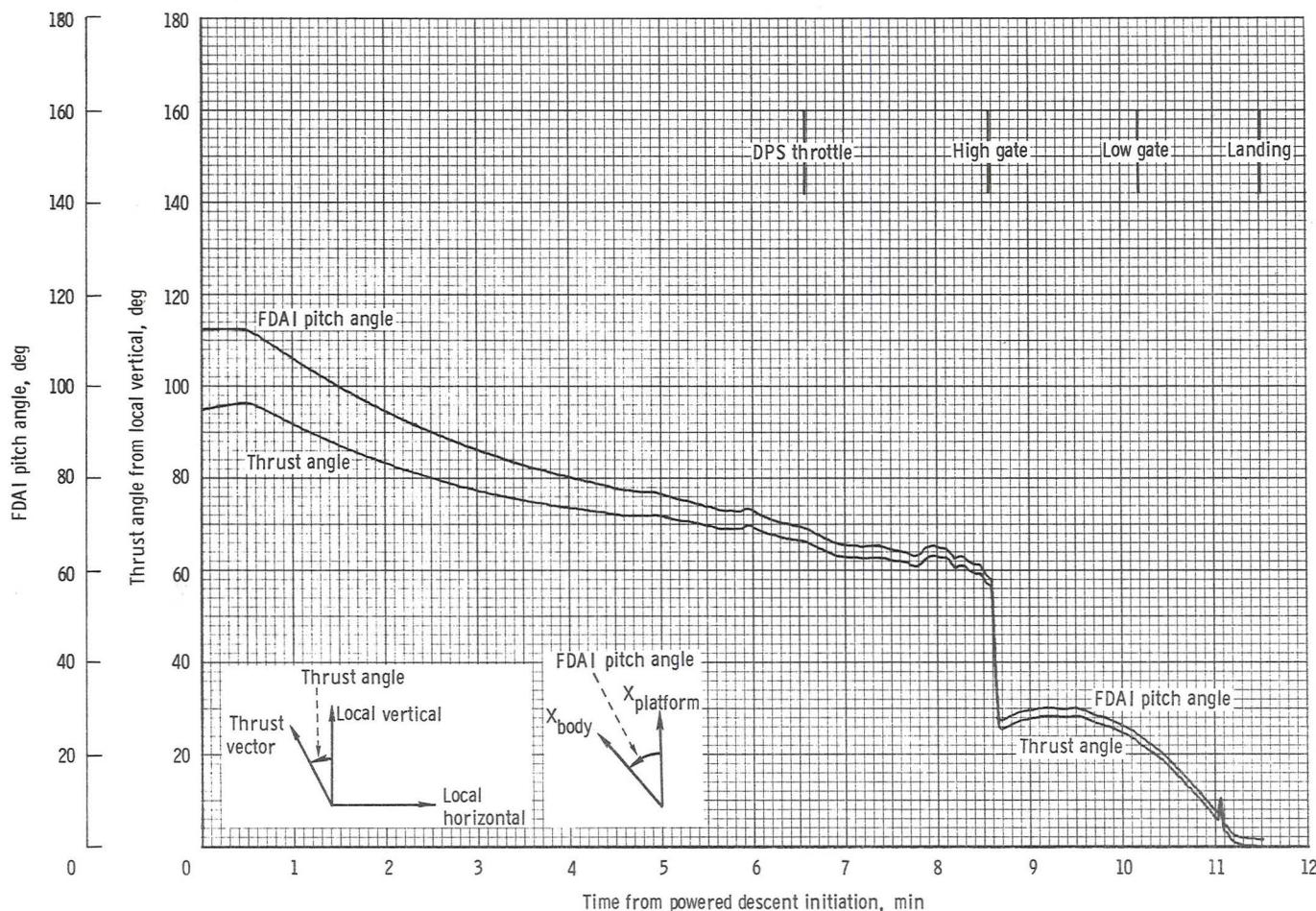
(a) LM velocity, altitude, and flight-path angle.

Figure 4.10-6. - Time histories of LM and LM-CSM trajectory parameters from PDI to landing.



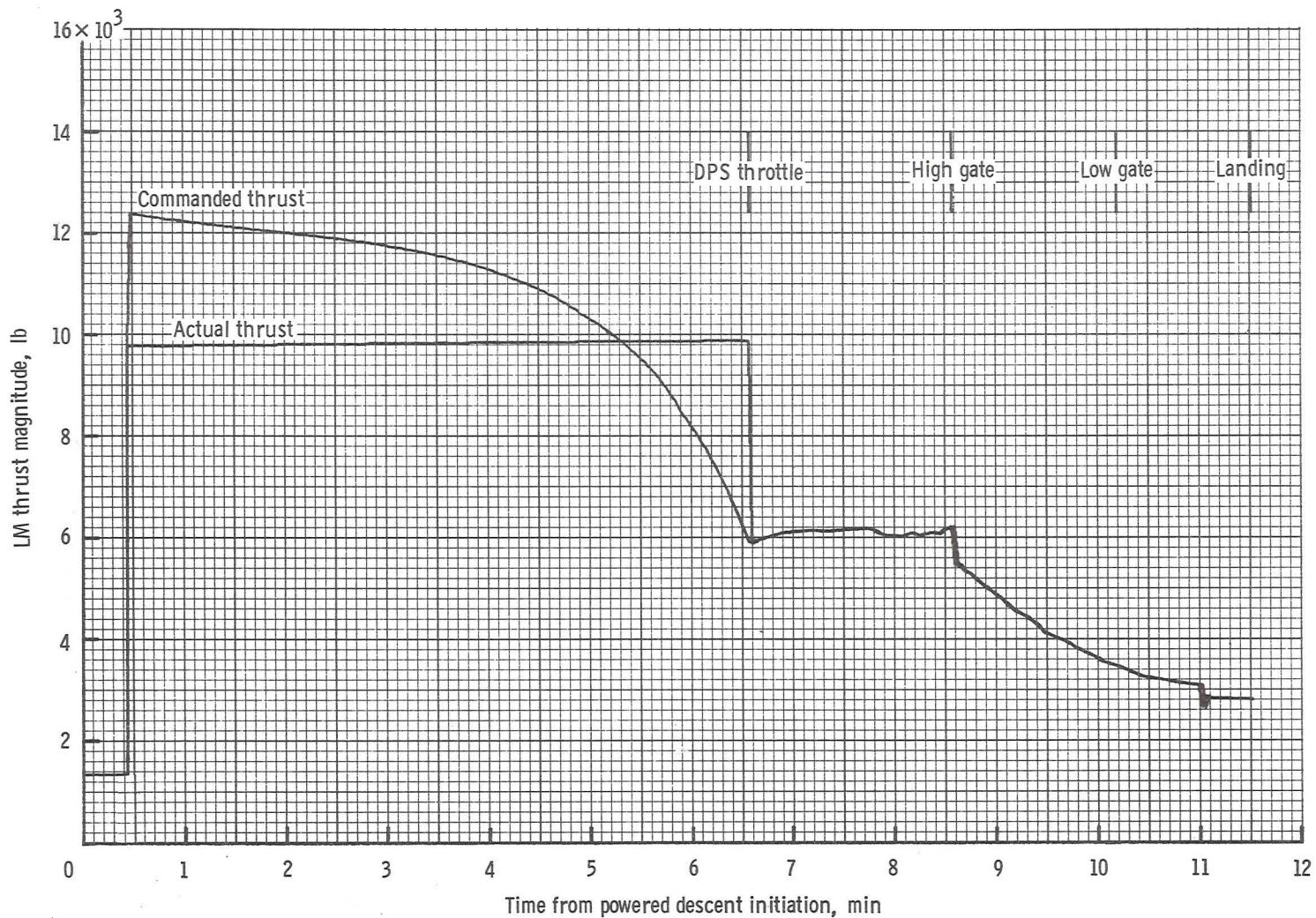
(b) LM-CSM range, range rate, and phase angle.

Figure 4.10-6.- Continued.



(c) FDAO pitch angle and thrust angle from local vertical.

Figure 4.10-6.- Continued.



(d) LM thrust magnitude.

Figure 4.10-6.- Concluded.

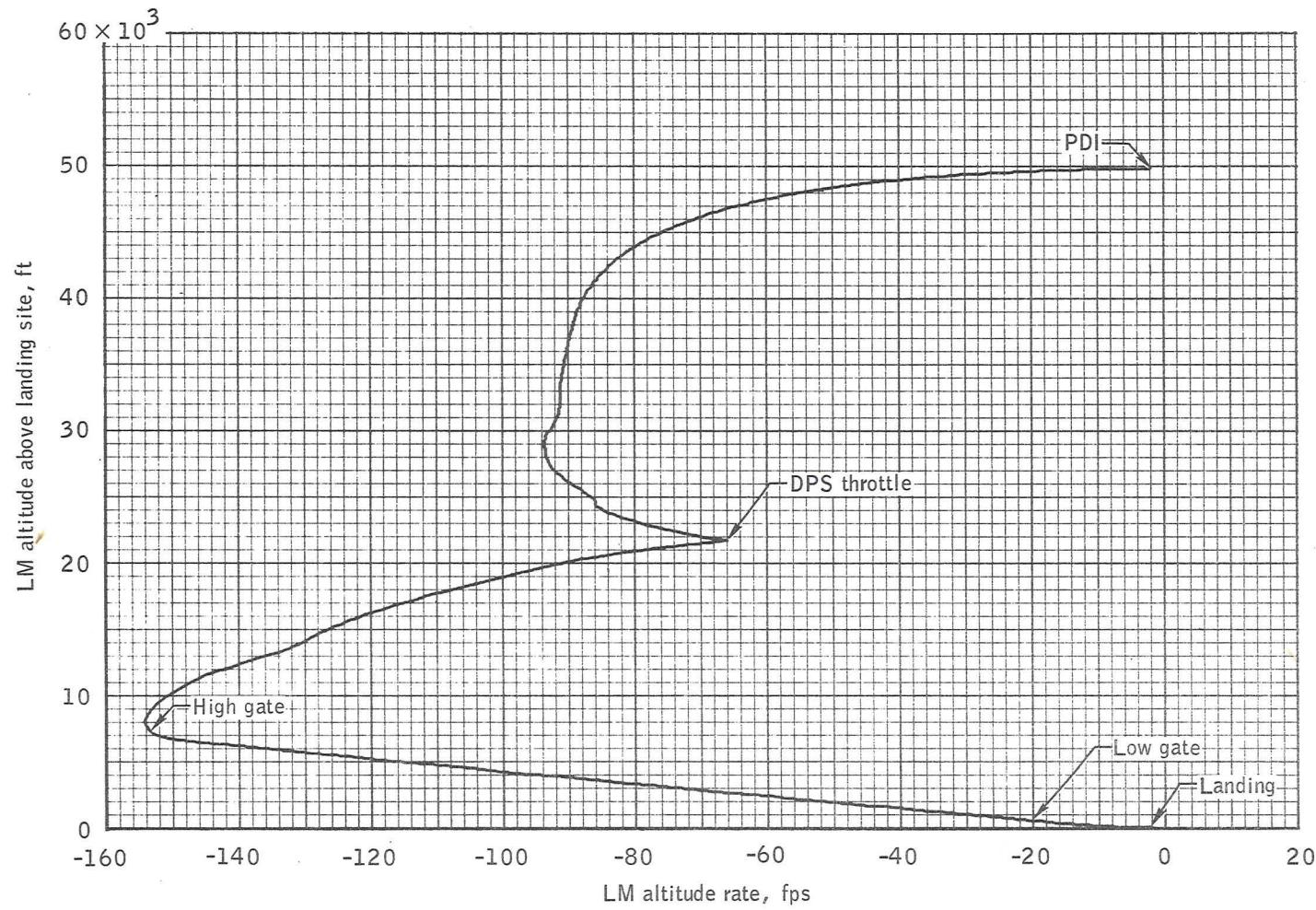


Figure 4.10-7.- Altitude above landing site versus altitude rate.

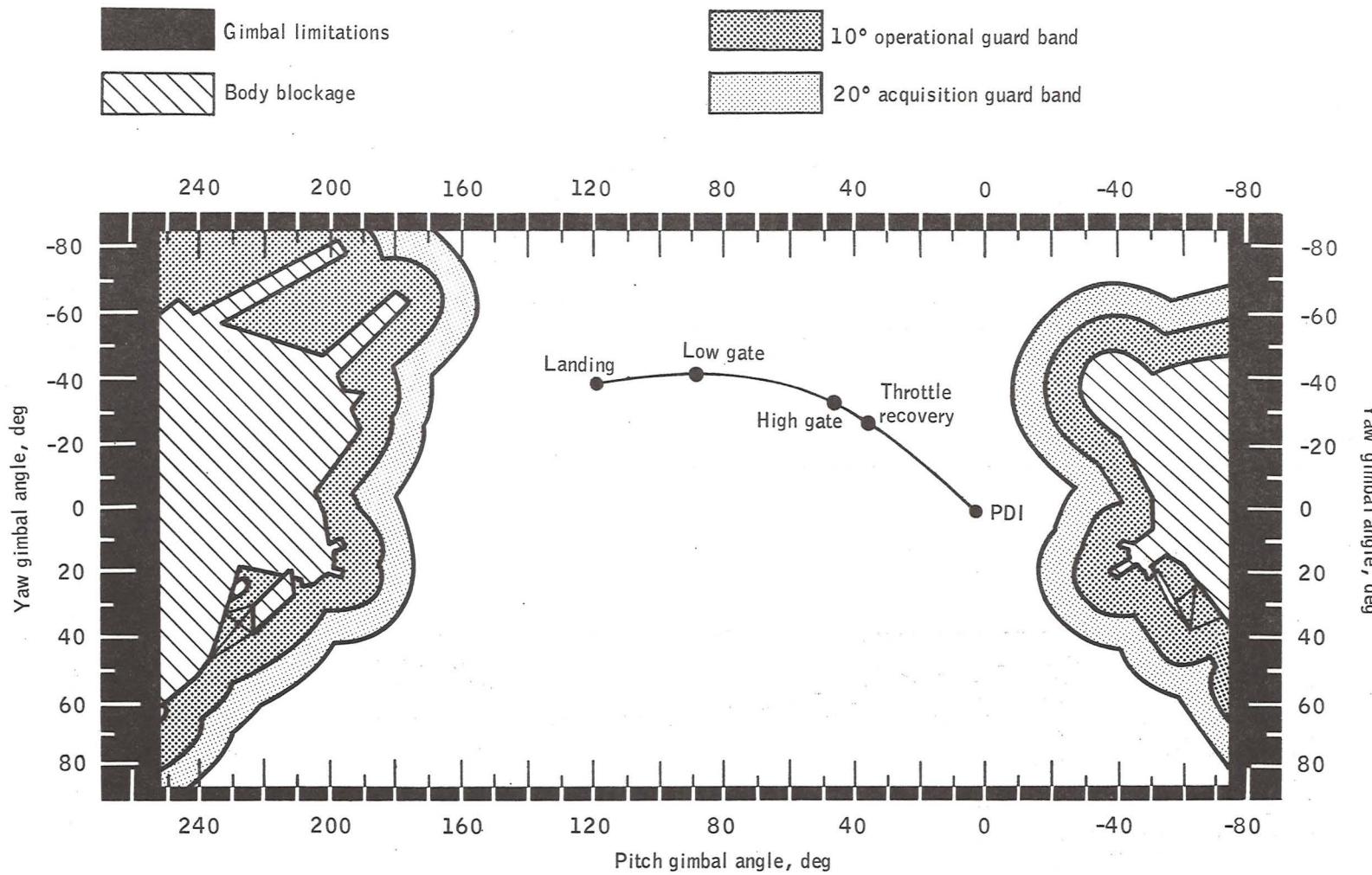


Figure 4.10-8.- S-band gimbal angle profile.

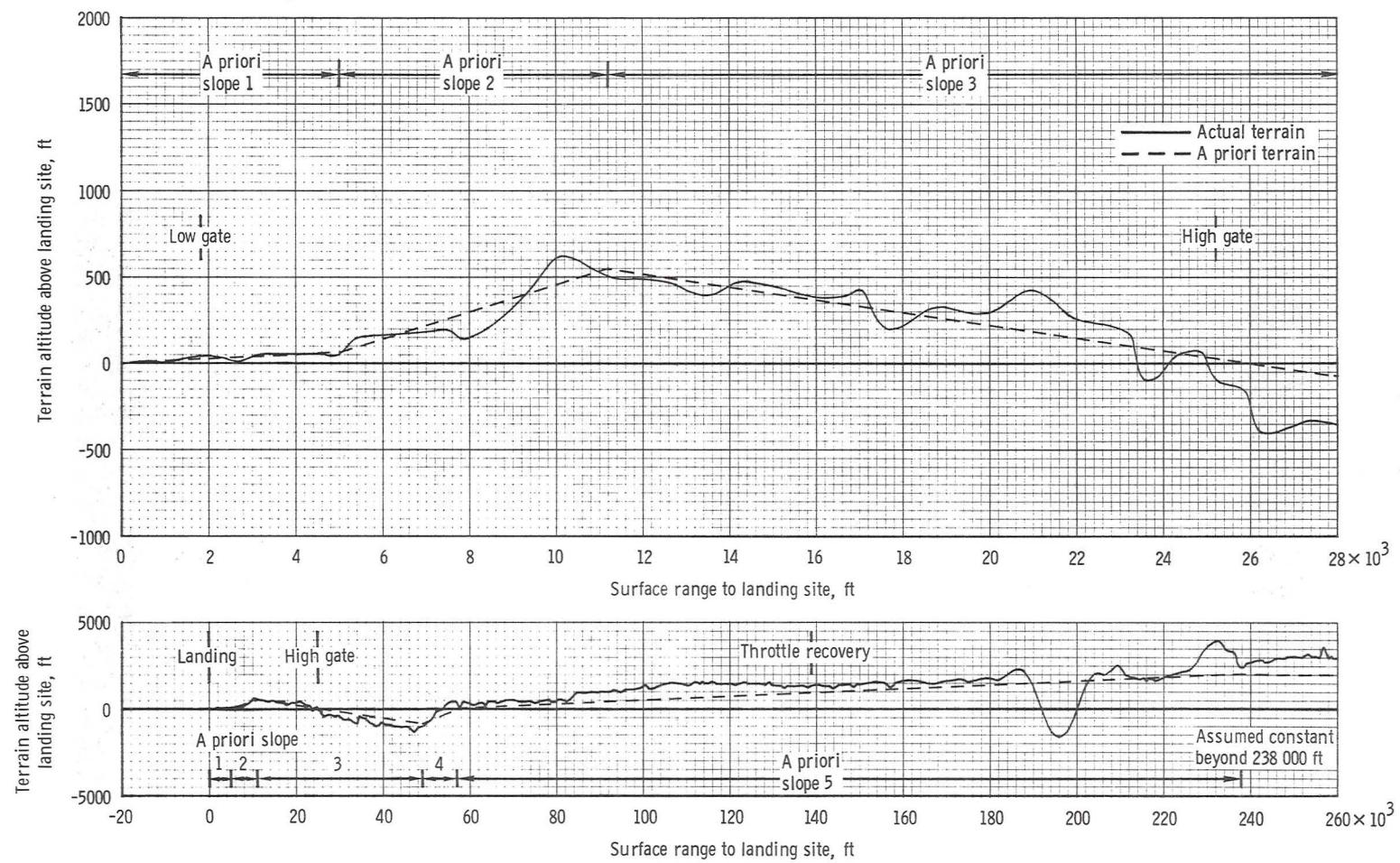


Figure 4.10-9.- Terrain altitude.

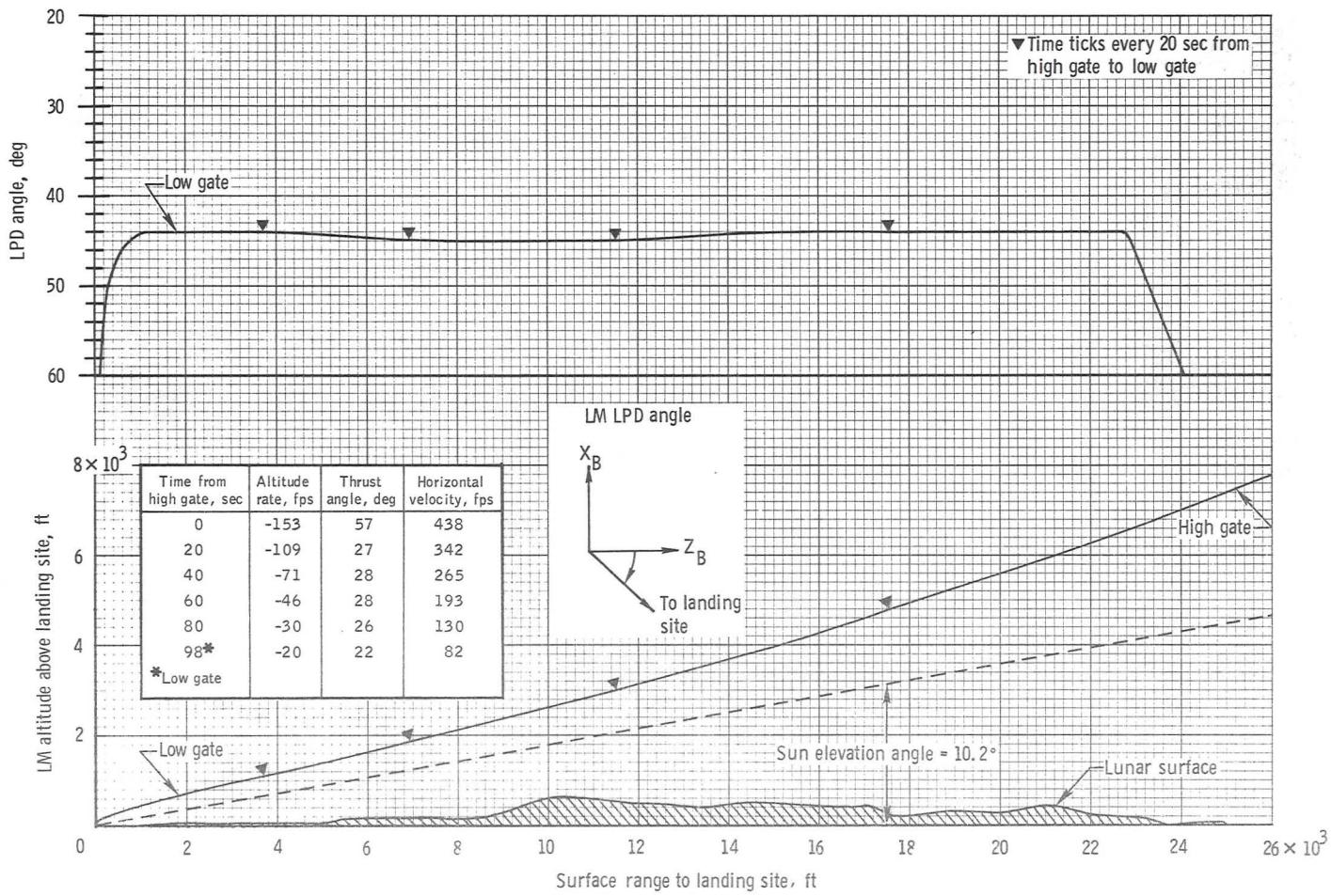


Figure 4.10-10. - Approach phase summary.

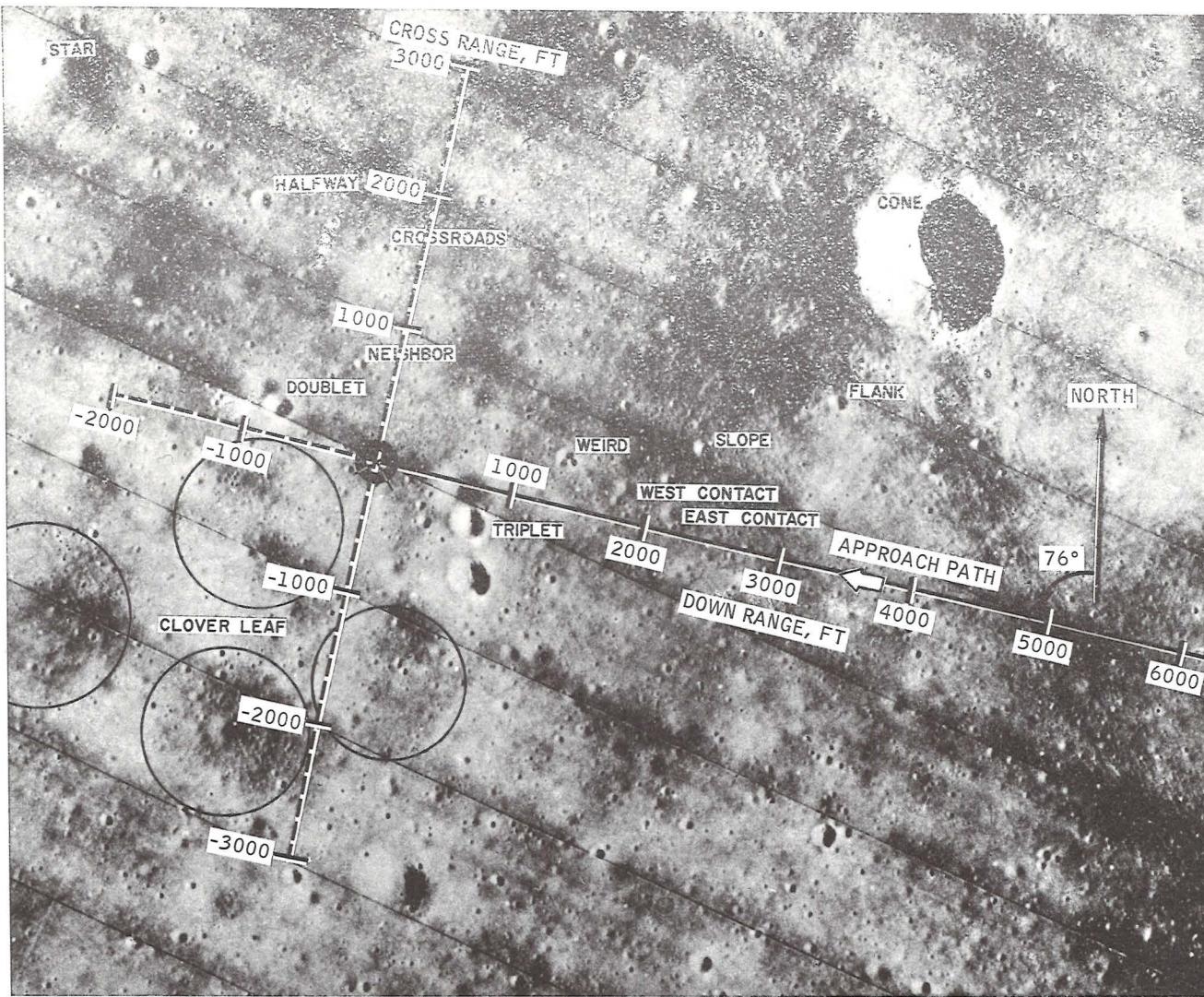
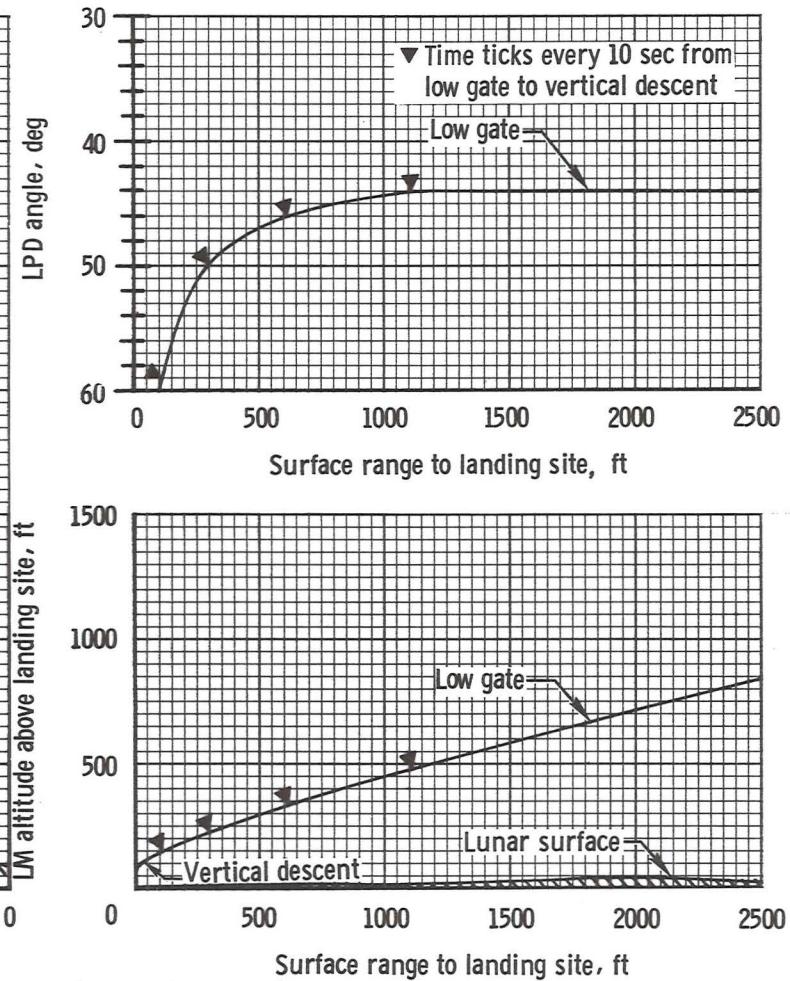
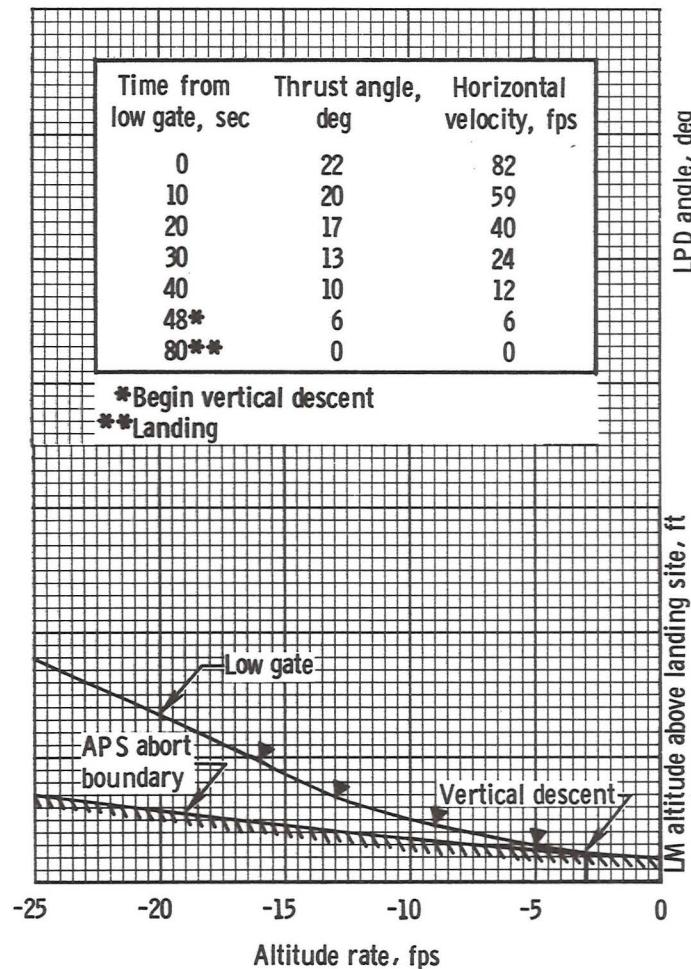


Figure 4.10-11.- Topography around the landing site at Fra Mauro.



T8-4

Figure 4.10-12. - Landing phase summary.

4.11 CSM PLANE CHANGE

## 4.11 CSM Lunar Orbit Plane Change

Approximately 35 hours 20 minutes after LOI, the CSM performs a plane change to place the LM in the CSM orbital plane at the time of LM lift-off. The maneuver is performed approximately 5.0 revolutions after LM landing. The target load for the maneuver is presented in table 4.11-I.

The burn characteristics are the following.

#### Burn initiation

Time, hr:min:sec, g.e.t. . . . .	118:09:40.2
Mass, lb . . . . .	35 585.0
Altitude above LLS, n. mi. . . . .	60.9
Selenographic latitude, deg . . . . .	-10.9
Selenographic longitude, deg . . . . .	93.7
Inertial velocity, fps . . . . .	5333.8
Selenographic inclination, deg . . . . .	14.2
Perilune altitude above LLS, n. mi. . . . .	57.4
Apolune altitude above LLS, n. mi. . . . .	61.7

#### Burn cutoff

Time, hr:min:sec, g.e.t. . . . .	118:09:58.5
Altitude above LLS, n. mi. . . . .	60.9
Selenographic latitude, deg . . . . .	-11.0
Selenographic longitude, deg . . . . .	92.7
Inertial velocity, fps . . . . .	5333.8
Selenographic inclination, deg . . . . .	12.1
Burn duration, sec . . . . .	18.4
Inertial burn arc, deg . . . . .	0.9
Plane change, deg . . . . .	3.9
$\Delta V$ , fps . . . . .	360.7
SPS propellant used, lb . . . . .	1249.8
Perilune altitude above LLS, n. mi. . . . .	57.4
Apolune altitude above LLS, n. mi. . . . .	61.7

TABLE 4.11-I.- TARGET LOAD FOR LUNAR ORBIT PLANE CHANGE MANEUVER

[Propulsion system: SPS; guidance: external  $\Delta V$ ]

(a) Target

$t_{IG}$ , hr:min:sec, g.e.t. . . . . . . . . . . . . . . . . .	118:09:40.2
$\Delta V_X$ , fps .	-12.1
$\Delta V_Y$ , fps .	360.5
$\Delta V_Z$ , fps .	0.0
Mass <sub>IG</sub> , lb .	35 585.0

(b) REFSMMAT

Refer to table 5-IV

(c) Gimbal angles at  $t_{TG}$

IGA, deg	...	0.0
MGA, deg	...	0.0
OGA, deg	...	0.0

4.12 LM ASCENT

## 4.12 LM Ascent

The LM powered ascent trajectory is an inplane launch into a 9.14-n. mi. (55 512-ft) by 50.96-n. mi. orbit about the moon. The LM lift-off time which is determined primarily by the LM-CSM phasing requirements for rendezvous occurs after approximately 33.5 hours of lunar stay time. A sketch of the powered ascent maneuver is shown in figure 4.12-1.

The LM ascent consists of two operational phases, a vertical rise phase and an orbital insertion phase. The first phase is necessary to achieve the desired altitude for terrain clearance. A description of trajectory parameters, vehicle attitudes, and the transition to the orbital insertion phase during vertical rise is shown in figure 4.12-2. For radial rates greater than 40 fps, the guidance switches to the orbital insertion phase; but, because of DAP steering lags, the pitchover does not occur until a radial rate of approximately 50 fps is achieved. Thus, the vertical rise phase is actually terminated 10 seconds after LM lift-off. During the vertical rise phase, the LM Z-body axis is rotated toward the desired azimuth which is normally in the CSM orbital plane.

The orbital insertion phase which is designed to insert the LM into the desired lunar orbit is described in figure 4.12-3. Included in this figure are orbital insertion parameters, total ascent phase performance, and onboard displays at insertion. Yaw steering is used during the orbital insertion phase if required to maneuver the LM into the CSM orbital plane or into a plane that is coplanar with the CSM orbit. Because the ascent is essentially inplane, no yaw steering is required. To avoid S-band communication blockage near the end of the orbit insertion phase, a 30° pilot yaw maneuver is initiated approximately 48 seconds into the ascent. Groundtrack information for the LM and CSM during powered ascent is shown in figure 4.12-4. The LM ascent radar coverage for a 0° minimum elevation is included in the mission radar timeline, table 1.1-V(c) (ref. 1). The radar coverage for a 3° minimum elevation is shown in volume II (ref. 1), table 1.1-V(g). The LM ascent weight summary is shown in table 4.12-I. The target load for LM ascent is presented in table 4.12-II, and the propulsion performance summary for LM ascent is shown in table 4.12-III. Because +X-axis RCS thrusting is used for attitude control, the data in table 4.12-III include the effect of both APS and RCS thrust. The APS interconnect is open during LM ascent, and all propellant is assumed to come from the APS tank.

Time histories of LM trajectory, attitude parameters, LM-CSM phasing, range, range rate and LM-CSM relative parameters are shown in figures 4.12-5, 4.12-6, and 4.12-7, respectively.

The LM-CSM range/range rate profile is shown in figure 4.12-8. The S-band gimble angle profile which shows the vehicle body blockage is shown in figure 4.12-9.

TABLE 4.12-I.- LM ASCENT WEIGHT SUMMARY

Weight of LM crew, inert weights, consumables and unusable APS propellant, lb . . . . .	5 534.8
LM APS fuel, lb . . . . .	2 002.2
LM APS oxidizer, lb . . . . .	3 210.0
Tanked APS propellant, lb . . . . .	5 212.2
Total weight of ascent stage, lb . . . . .	10 810.9
Net off-loaded weight on lunar surface, lb . . . . .	-63.9
Total ascent stage lift-off weight, lb . . . . .	10 747.0

TABLE 4.12-II.- TARGET LOAD FOR LM ASCENT

[Propulsion systems: APS, RCS; guidance:  
orbit insertion (P12)]

(a) Target

$t_{IG}$ (lift-off), hr:min:sec, g.e.t. . . . . .	142:24:29.170
$R_D$ (desired injection radius), ft . . . . .	5 757 796.5
$Y_D$ (desired injection cross-range distance), ft . . . . .	0.0
$R_D$ (desired injection radial velocity), fps . . . . .	32.0
$Y_D$ (desired injection cross-range velocity), fps . . . . .	0.0
$Z_D$ (down-range velocity for desired injection), fps . .	5 540.8
Weight at $t_{IG}$ , lb .	10 747.0

(b) REF SMMAT

Refer to table 5-V

(c) Gimbal angles at  $t_{TG}$

IGA, deg	.....	0.0
MGA, deg	.....	0.0
OGA, deg	.....	0.0

TABLE 4.12-III.- LM ASCENT PROPULSION PERFORMANCE SUMMARY

[Combined APS and RCS]

Event	Time from lift-off, sec	APS thrust, lb	APS $\dot{W}$ , lb/sec	RCS effective thrust, lb	RCS effective $\dot{W}$ , lb/sec	Total thrust, lb	Total $\dot{W}$ , lb/sec	$I_{sp}$ , sec
LM lift-off	0	3501.54	11.29	71.92	0.25	3573.46	11.54	309.58
	60	3492.37	11.25	62.24	0.22	3554.61	11.47	309.98
	120	3483.23	11.22	51.25	0.18	3534.48	11.40	310.17
	180	3475.45	11.19	39.39	0.14	3514.84	11.33	310.20
	240	3469.90	11.18	26.58	0.09	3496.48	11.27	310.17
	300	3466.96	11.18	12.30	0.04	3479.26	11.22	310.16
	360	3466.61	11.18	12.94	0.04	3479.55	11.22	310.11
	420	3468.29	11.17	25.25	0.09	3493.54	11.26	310.12
LM insertion	430.7	3468.74	11.18	29.53	0.10	3498.27	11.28	310.13

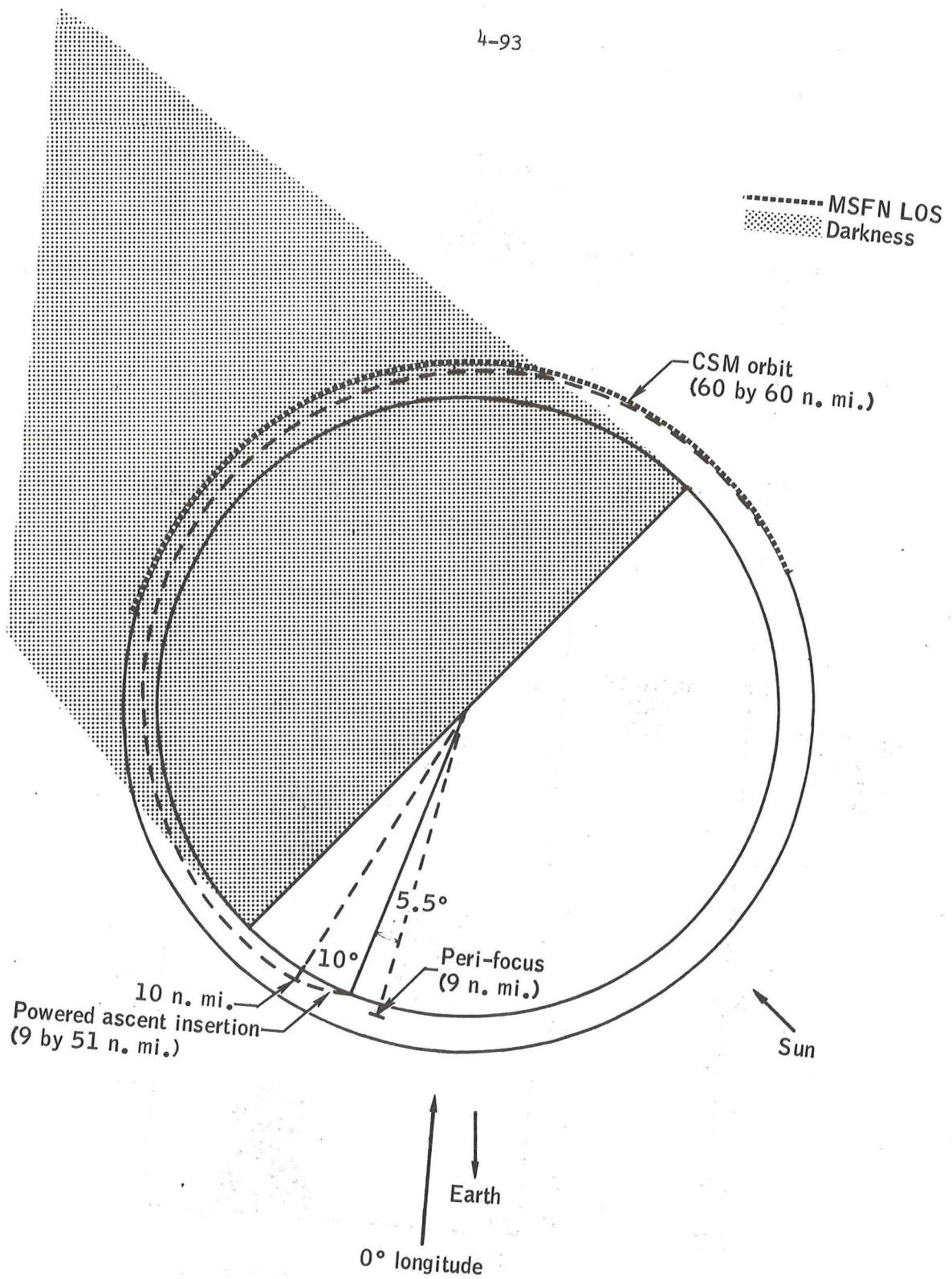


Figure 4.12-1.- LM ascent.

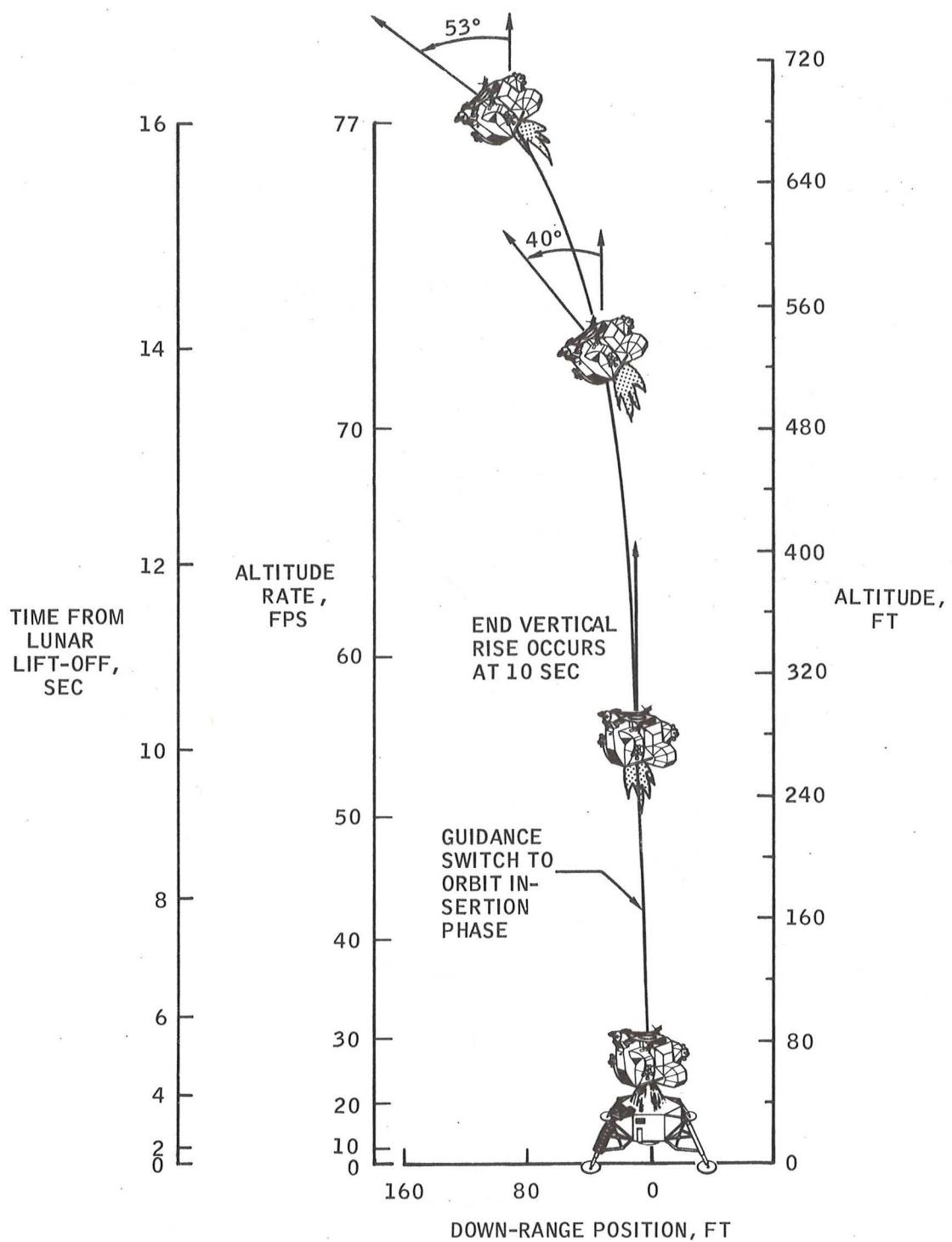
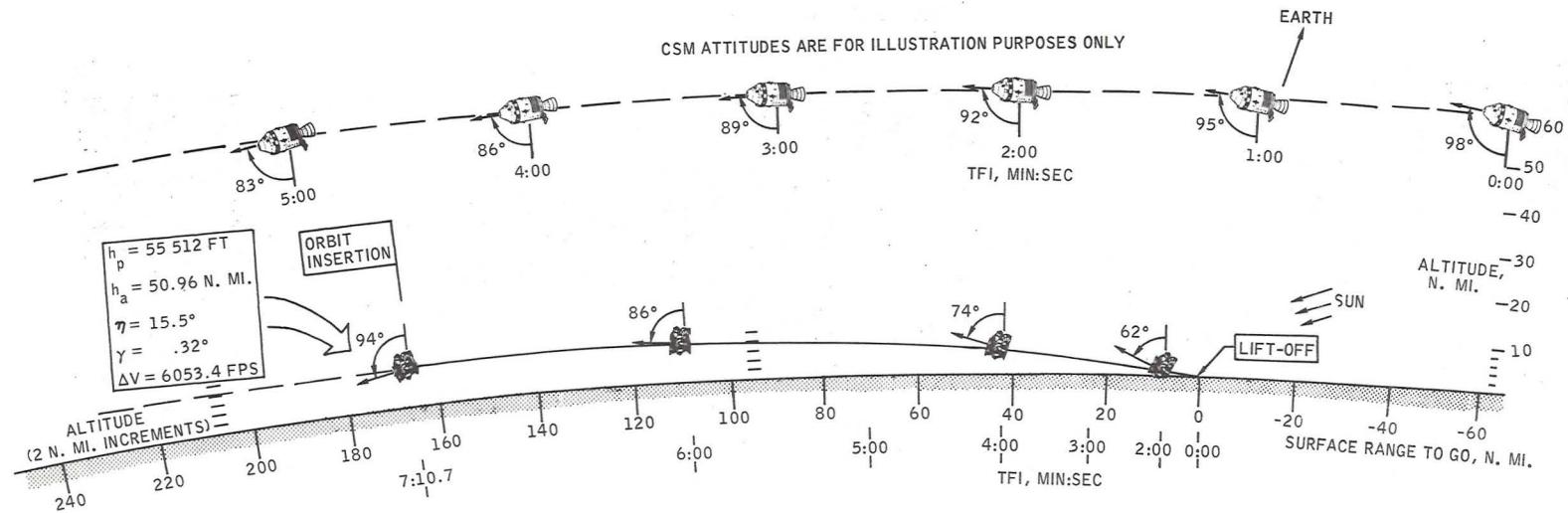


Figure 4.12-2.- Vertical rise phase.



4-95

SUMMARY							
EVENT	TFI, MIN:SEC	INERTIAL VELOCITY, FPS	ALTITUDE RATE, FPS	ALTITUDE, FT	$\Delta V$ , FPS	WEIGHT, LB	PROPELLANT, LB
LIFT-OFF	0:00	15.1	0	0	0	10 747.0	0
END OF VERTICAL RISE	0:10	55.7	54.2	269.4	107.5	10 630.8	115.4
BEGIN LM PILOT YAW	0:48	324.8	113.8	3 726.5	526.0	10 190.3	552.6
END LM PILOT YAW	1:00	435.0	126.5	5 169.0	661.6	10 051.5	690.3
	2:00	1036.0	172.5	14 279.2	1366.9	9 360.5	1376.2
	4:00	2474.5	184.8	36 727.0	2934.6	7 990.3	2736.0
	6:00	4272.9	105.9	54 969.6	4779.3	6 632.3	4083.8
ORBIT INSERTION	7:10.7	5540.6	31.3	59 932.7	6053.4	5 831.2	4878.5

Figure 4.12-3.- Orbit insertion phase.

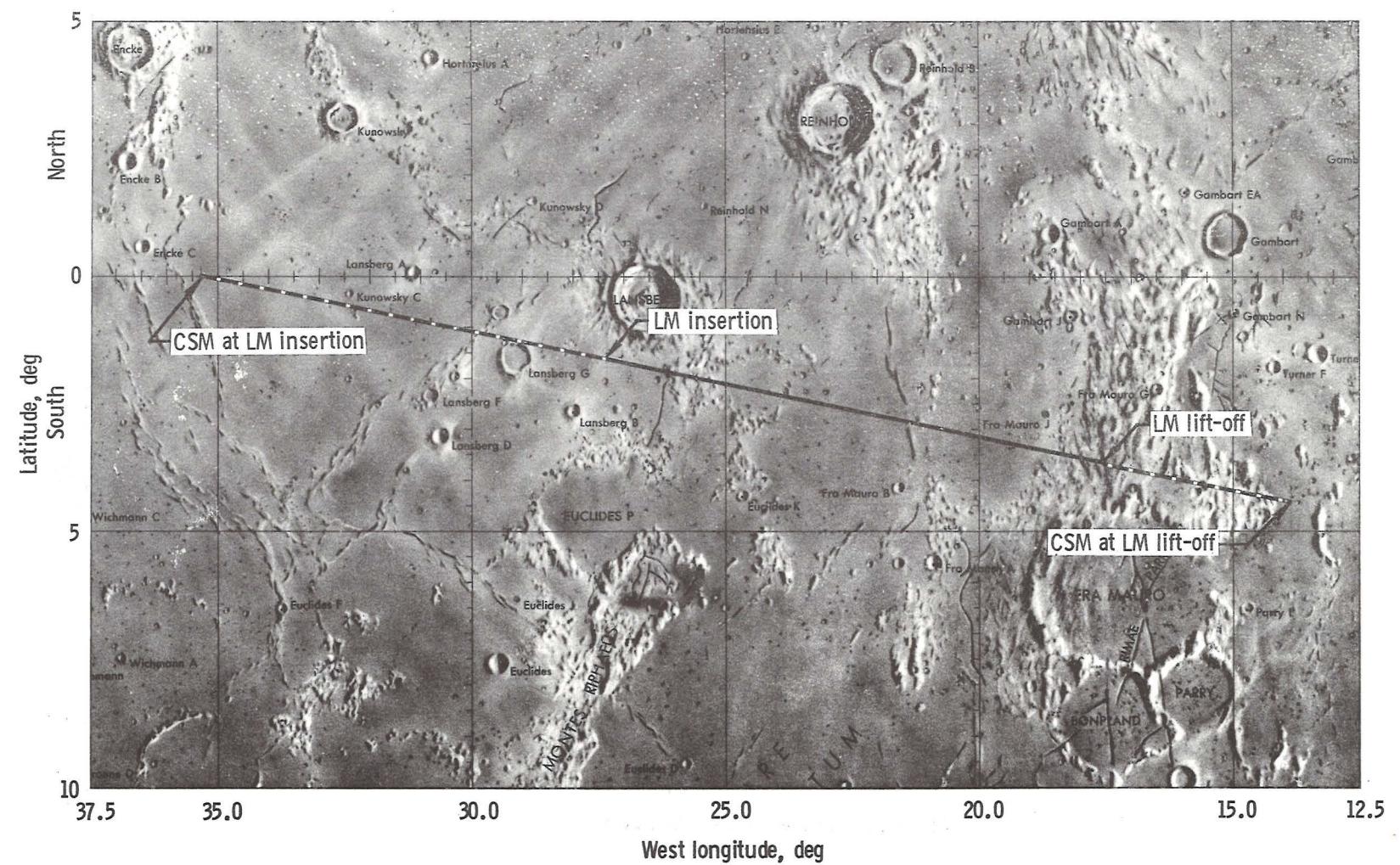
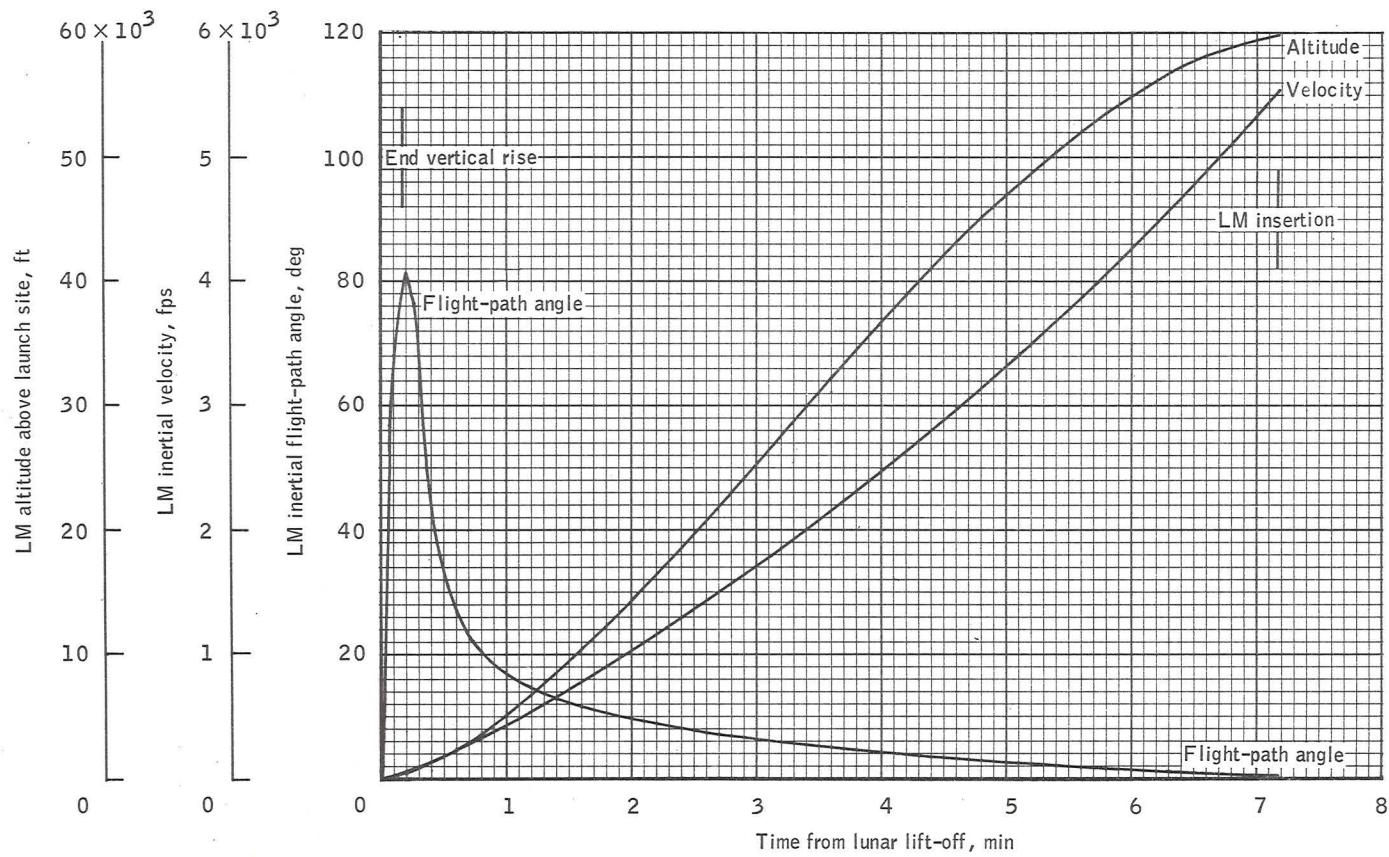
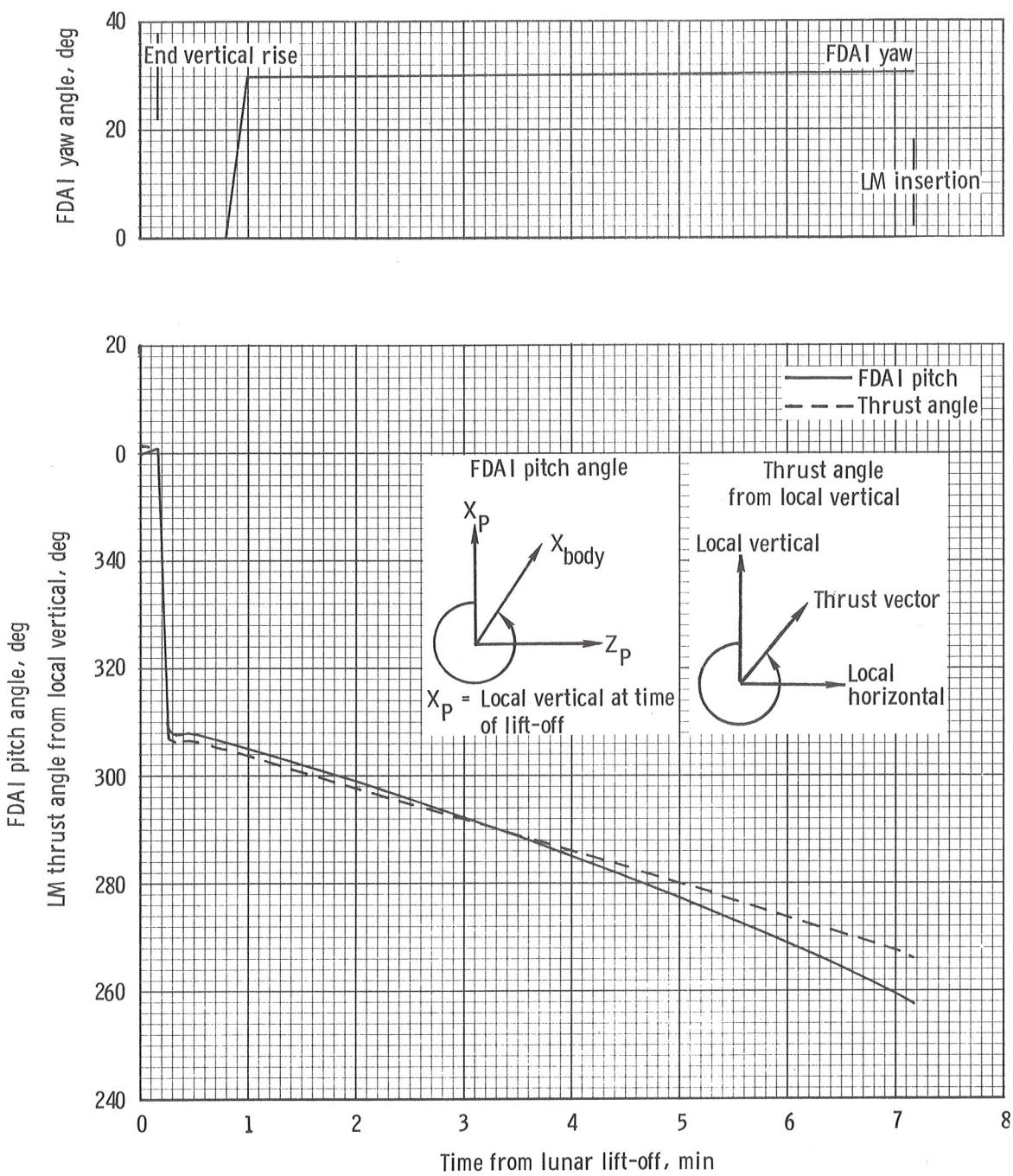


Figure 4.12-4.- LM and CSM groundtracks.



(a) LM altitude, inertial velocity, and inertial flight-path angle.

Figure 4.12-5.- Time histories of LM trajectory parameters from lunar lift-off to LM insertion.



(b) LM thrust angle and FDAO angles.

Figure 4.12-5. - Concluded.

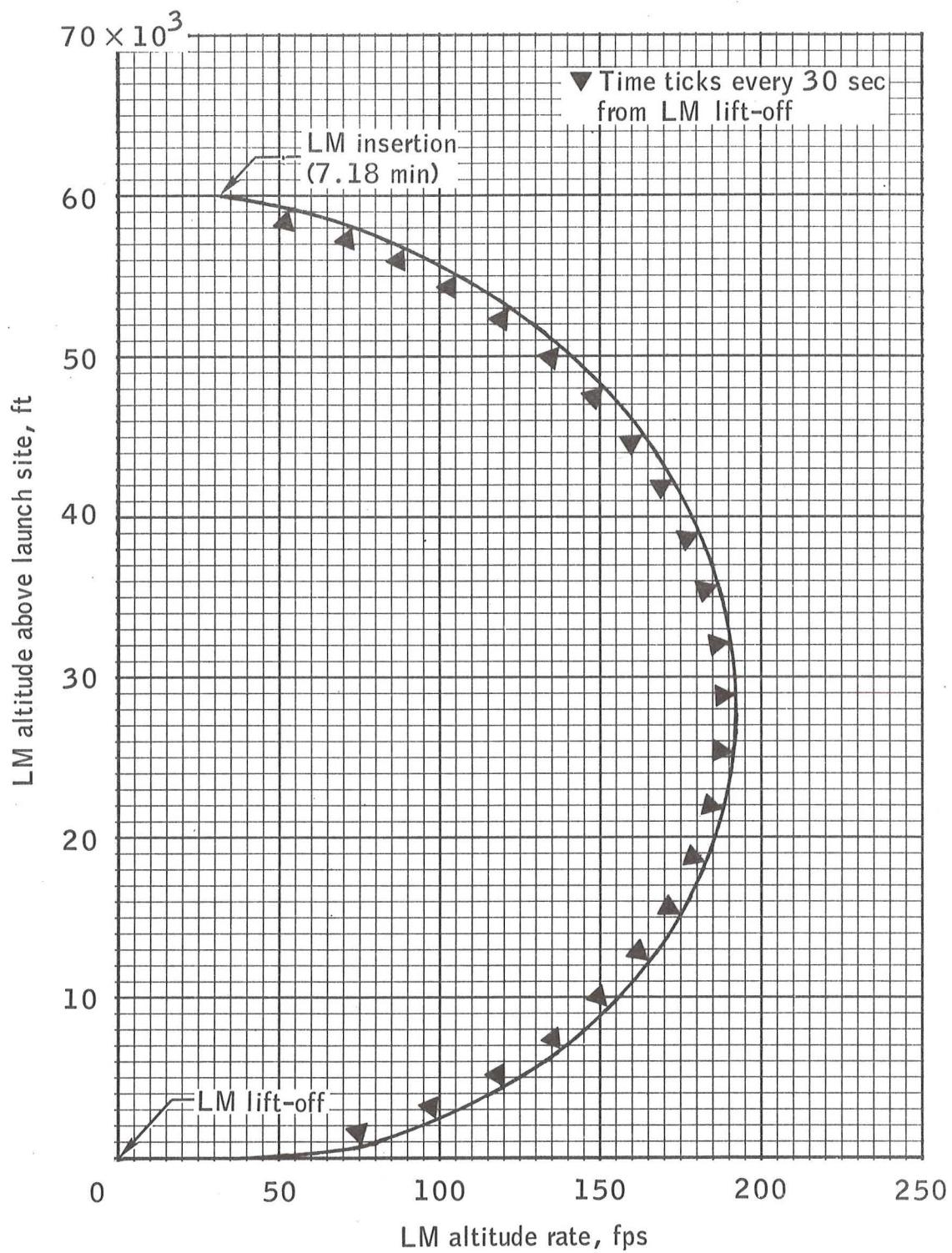


Figure 4.12-6.- LM altitude versus altitude rate during powered ascent.

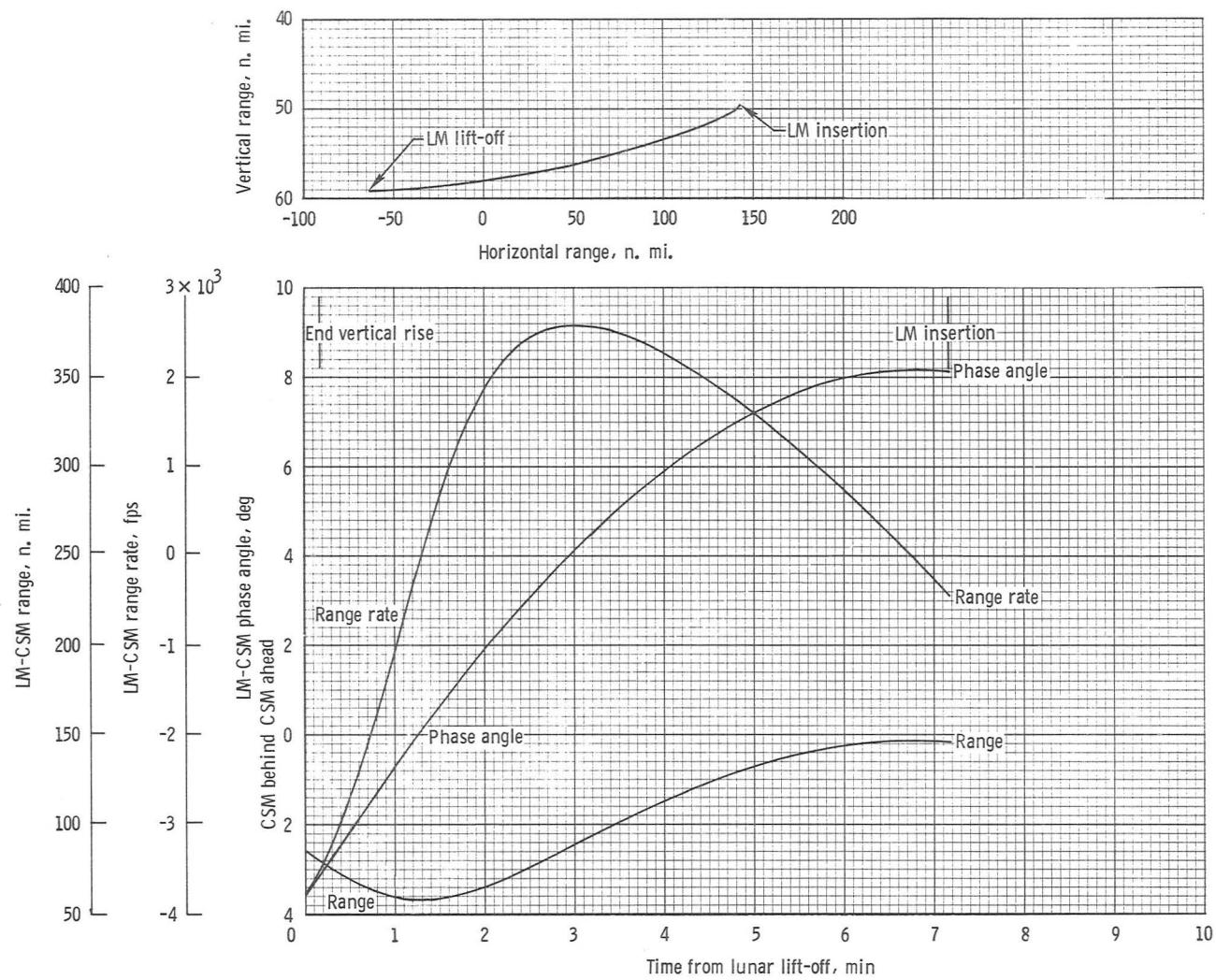


Figure 4.12-7. - LM-CSM phase angle, range, and range rate versus time from lift-off and vertical range versus horizontal range (CSM referenced).

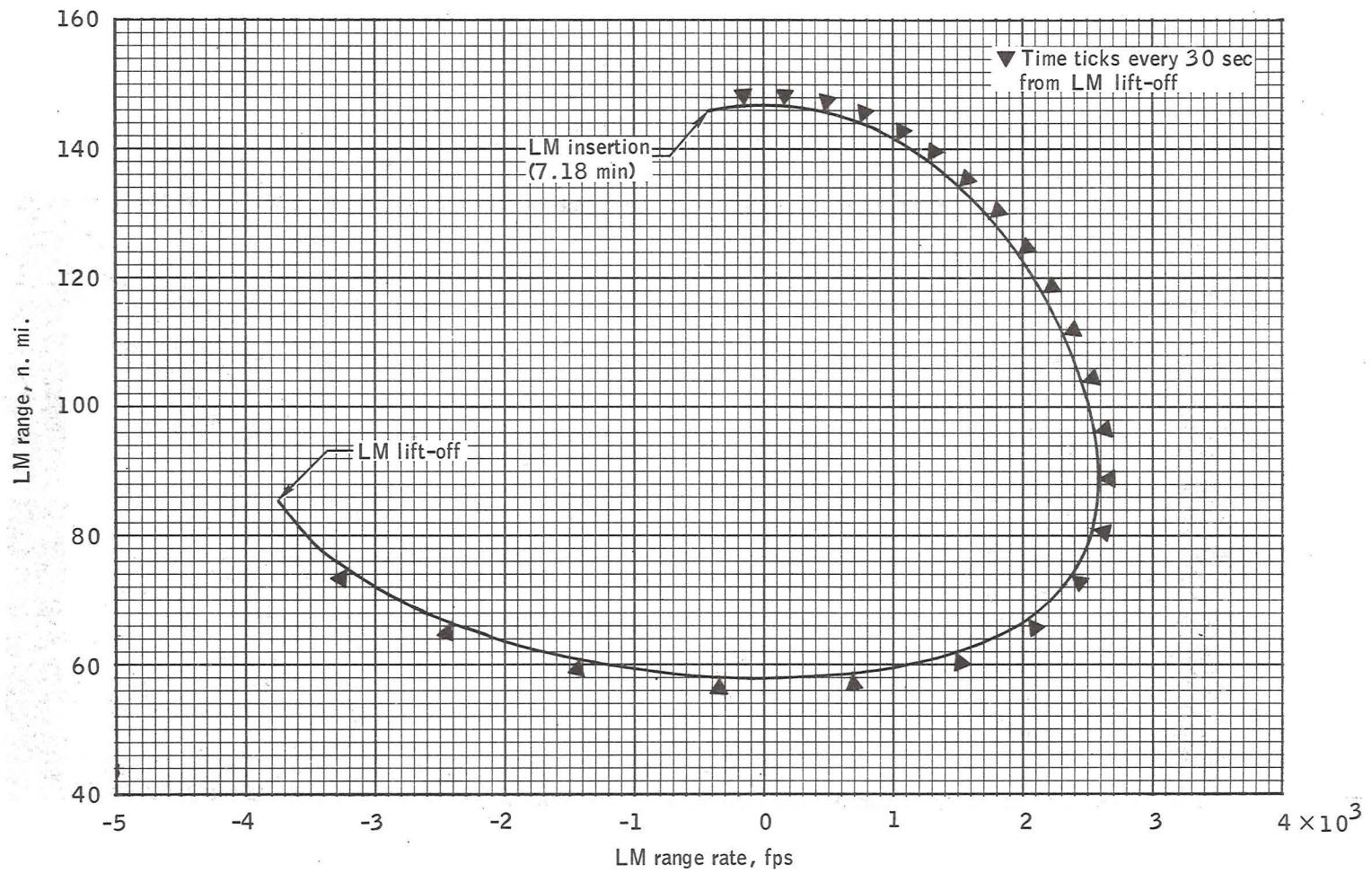


Figure 4.12-8.- LM-CSM range versus LM-CSM range rate during powered ascent.

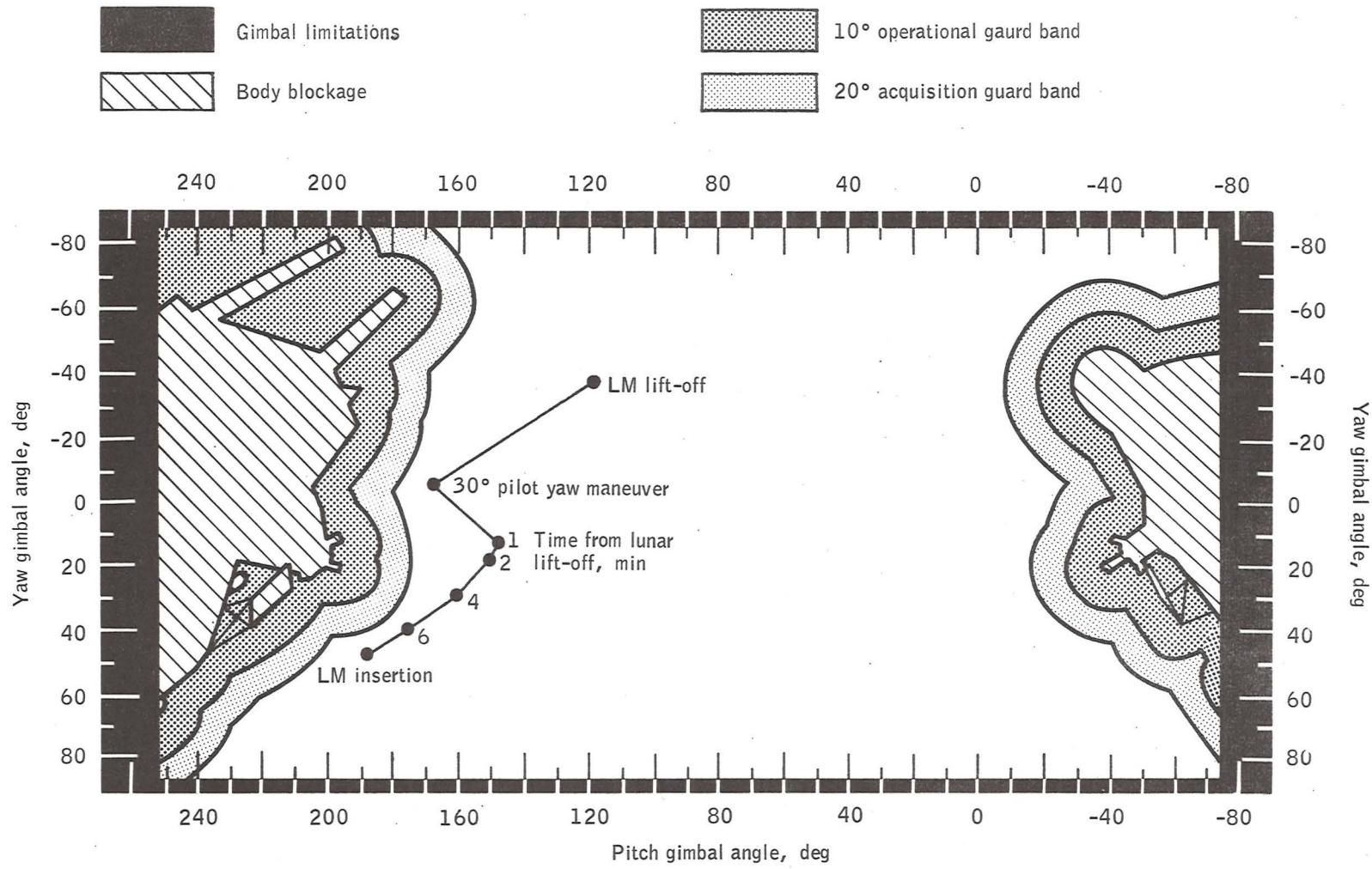


Figure 4.12-9.- S-band gimbal angle profile during powered ascent.

**4.13 RENDEZVOUS SEQUENCE**

#### 4.13 Rendezvous Sequence

4.13.1 Sequence summary.- After insertion of the LM into orbit, the LM will compute and execute the early rendezvous sequence which nominally consists of two major maneuvers, TPI and TPF. A nominally-zero tweak maneuver will be scheduled 3 minutes after insertion, and two nominally-zero midcourse correction maneuvers will be scheduled between TPI and TPF; the TPF maneuver is actually divided into several braking maneuvers. The complete maneuver sequence is presented in table 4.13-I. All maneuvers except TPI will be performed with the LM RCS system with either two jets or four jets for a prescribed quad for thrust vector control. Latest planning required TPI to be performed with the APS preceded by a four-jet RCS ullage maneuver; however, this trajectory was computed with a two-jet RCS ullage maneuver. The entire maneuver sequence from lift-off through docking requires approximately 1.8 hours.

Before detailed information is presented about each maneuver, some general mission and trajectory design requirements for the rendezvous plan are noted.

- a. TPI will occur 38 minutes after insertion.
- b. The LM will be 15 n. mi. below the CSM orbit at TPI.
- c. Braking should occur in daylight; stationkeeping should begin approximately 25 minutes after the vehicles enter sunlight.

The orbital schematic for the rendezvous sequence is presented in figure 4.13-1. The time histories of range, range rate, phase angle, CSM-LM elevation angles, and the CSM-LM-sun angles are presented in figures 4.13-2(a) through 4.13-2(e). The relative motion of the LM with respect to the CSM is shown in figure 4.13-2(f). The onboard tracking, lighting, and mission events summary are presented in figure 4.13-3 and in table 4.13-II.

4.13.2 Tweak maneuver.- A nominally-zero tweak maneuver is scheduled 3 minutes after insertion. The purpose of this maneuver is to maintain the nominal TPI offsets (both inplane and out of plane) when insertion dispersions occur. Out-of-plane dispersions will be eliminated by a tweak/TPI combination. The tweak maneuver would force a common node at TPI, and the out-of-plane velocity component would be reduced to zero at TPI. Any out-of-plane situation not eliminated by the tweak/TPI combination must be eliminated during terminal phase.

4.13.3 TPI.- The TPI maneuver executed with the APS 18.5 minutes prior to daylight is initiated 38 minutes after insertion. The insertion orbit is such that at this time the LM will be 29 n. mi. behind and 15.0 n. mi. below the CSM. The thrust direction is posigrade  $46.9^\circ$  below the local horizontal. The APS burn is preceded by a +X (two-jet) RCS ullage maneuver which is 13.5 seconds long and overlaps the main engine burn by 0.5 second. The APS burn duration is 4.0 seconds.

The Lambert targeting for this maneuver is based on a  $130^\circ$  CSM central travel angle between TPI and TPF. The target load for this maneuver is presented in table 4.13-III. The LM-CSM orbital geometry is shown in figure 4.13-4. The characteristics of the TPI maneuver are as follows.

#### Ullage maneuver (two-jet RCS, +X)

Start time, hr:min:sec, g.e.t. . . . .	143:09:26.9
Latitude, deg N . . . . .	11.2
Longitude, deg W . . . . .	149.0
Altitude above landing site, n. mi. . . . .	44.7
Phase angle, deg . . . . .	1.7
Duration, sec . . . . .	13.0
$\Delta V$ , fps . . . . .	14.4
Propellant used, lbs . . . . .	9.3

#### Main engine burn

Start time, hr:min:sec, g.e.t. . . . .	143:09:39.9
$\Delta V$ , fps . . . . .	77.8
Duration, sec . . . . .	4.0
Propellant used, lb . . . . .	45.3

#### TPI termination

Time, hr:min:sec, g.e.t. . . . .	143:09:43.9
Latitude, deg N . . . . .	11.1
Longitude, deg W . . . . .	149.9
Altitude above landing site, n. mi. . . . .	44.8
Resultant $h_a/h_p$ , n. mi. . . . .	61.0/44.6
Catchup rate, deg/min . . . . .	0.034

4.13.4 Midcourse corrections and TPF. - MC-1 and MC-2 occur 15 and 30 minutes after TPI, respectively, and are nominally-zero maneuvers. The terminal phase braking schedule performed with the two-jet -Z-thrusters is presented in table 5-XIV. Line-of-sight corrections will be made as required. Final approach and stationkeeping should occur at approximately  $5^d 23^h 54^m 47.8^s$  g.e.t., approximately 27 minutes after the vehicles enter sunlight on the back side of the moon. Docking should begin at approximately  $6^d 00^h 10^m 00^s$  g.e.t. to complete 1.0 hours of rendezvous activities.

TABLE 4.13-I.- RENDEZVOUS MANEUVER SUMMARY

Event	Time, day:hr:min:sec, g.e.t.	Propulsion system	Burn time, sec	Total $\Delta V$ , fps	Weight at ignition, lb	Propellant used <sup>a</sup> , lb	Resultant orbit <sup>b</sup> $h_a/h_p$ , n. mi.	Range at cutoff, n. mi.	Range rate at cutoff, fps
Tweak <sup>c</sup>	5:22:34:39.9	RCS	0.0	0.0	5831.5	0.0	50.3/9.1	133.0	-439.5
TPI APS burn <sup>d</sup>	5:23:09:39.9	APS	4.0	77.8	5822.2	45.3	61.0/44.6	32.5	-134.3
MC-1 <sup>c</sup>	5:23:24:39.9	RCS +Z (two-jet)	0.0	0.0	5776.9	0.0	61.0/44.6	14.9	-97.3
MC-2 <sup>c</sup>	5:23:39:39.9	RCS +Z (two-jet)	0.0	0.0	5776.9	0.0	61.0/44.6	4.5	-45.6
First braking maneuver	5:23:50:32.8	RCS -Z (two-jet)	12.0	13.4	5776.9	8.6	60.0/52.2	0.4	-19.6
Second braking maneuver	5:23:51:45.7	RCS -Z (two-jet)	8.7	9.7	5768.3	6.2	59.3/55.9	0.2	-9.8
Third braking maneuver	5:23:53:24.0	RCS -Z (two-jet)	4.3	4.8	5762.1	3.0	59.0/57.9	0.08	-4.9
Fourth braking maneuver	5:23:54:43.6	RCS -Z (two-jet)	4.2	4.7	5759.1	3.0	60.2/58.7	0.01	-0.2
Docking	6:00:10:00.0	--	--	--	5756.1	--	60.2/58.7	0.0	0.0

<sup>a</sup>Estimates for completing consumables document will present official propellant usage. Slight changes caused by rigor of analysis are expected.

<sup>b</sup>Altitudes measured above landing site.

<sup>c</sup>Nominally a zero  $\Delta V$  maneuver.

<sup>d</sup>Preceded by a 13.0 sec, two-jet, +X, RCS ullage.

TABLE 4.13-II.- MSFN TRACKING AND SHADOW TIMELINE

Event	Time, day:hr:min:sec, g.e.t.
Insertion	5:22:31:39.9
Tweak	5:22:34:39.9
Sunset	5:22:40:11.8
MSFN LOS	5:22:54:04.0
TPI (ullage on)	5:23:09:26.9
MC-1	5:23:24:39.9
Sunrise	5:23:28:09.7
MC-2	5:23:39:39.9
MSFN AOS	5:23:40:29.2
First braking maneuver	5:23:50:32.8
Second braking maneuver	5:23:51:45.7
Third braking maneuver	5:23:53:24.0
Fourth braking maneuver	5:23:54:43.6
Docking	6:00:10:00.0

TABLE 4.13-III.- TARGET LOADS FOR TPI MANEUVER<sup>a</sup>[Propulsion system: APS; guidance: external  $\Delta V$ ]

## (a) Target

$t_{IG}$ , <sup>b</sup> hr:min:sec, g.e.t.	143:09:26.9
$\Delta V_X$ , fps	63.0
$\Delta V_Y$ , fps	0.1
$\Delta V_Z$ , fps	67.3
Weight at $t_{IG}$ , lb	5831.5

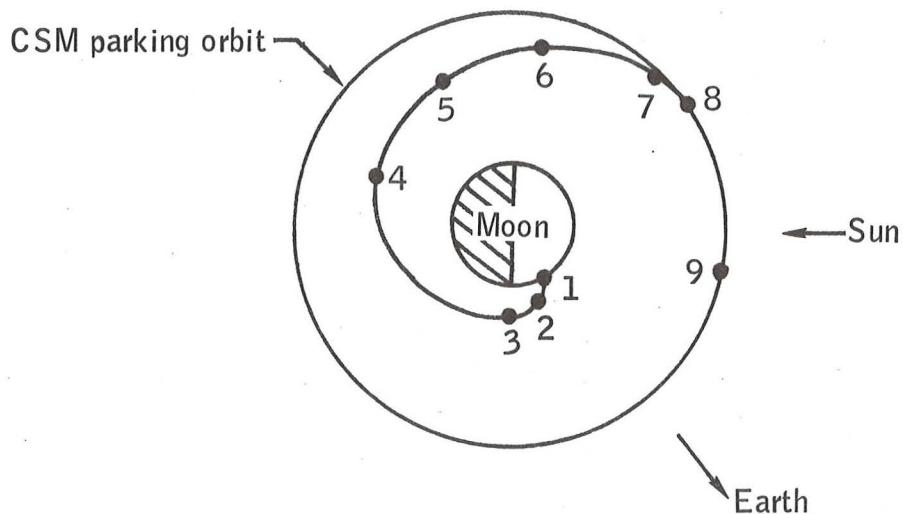
## (b) REFSSMMAT

Refer to table 5-V

(c) Gimbal angles at  $t_{IG}$ <sup>b</sup>

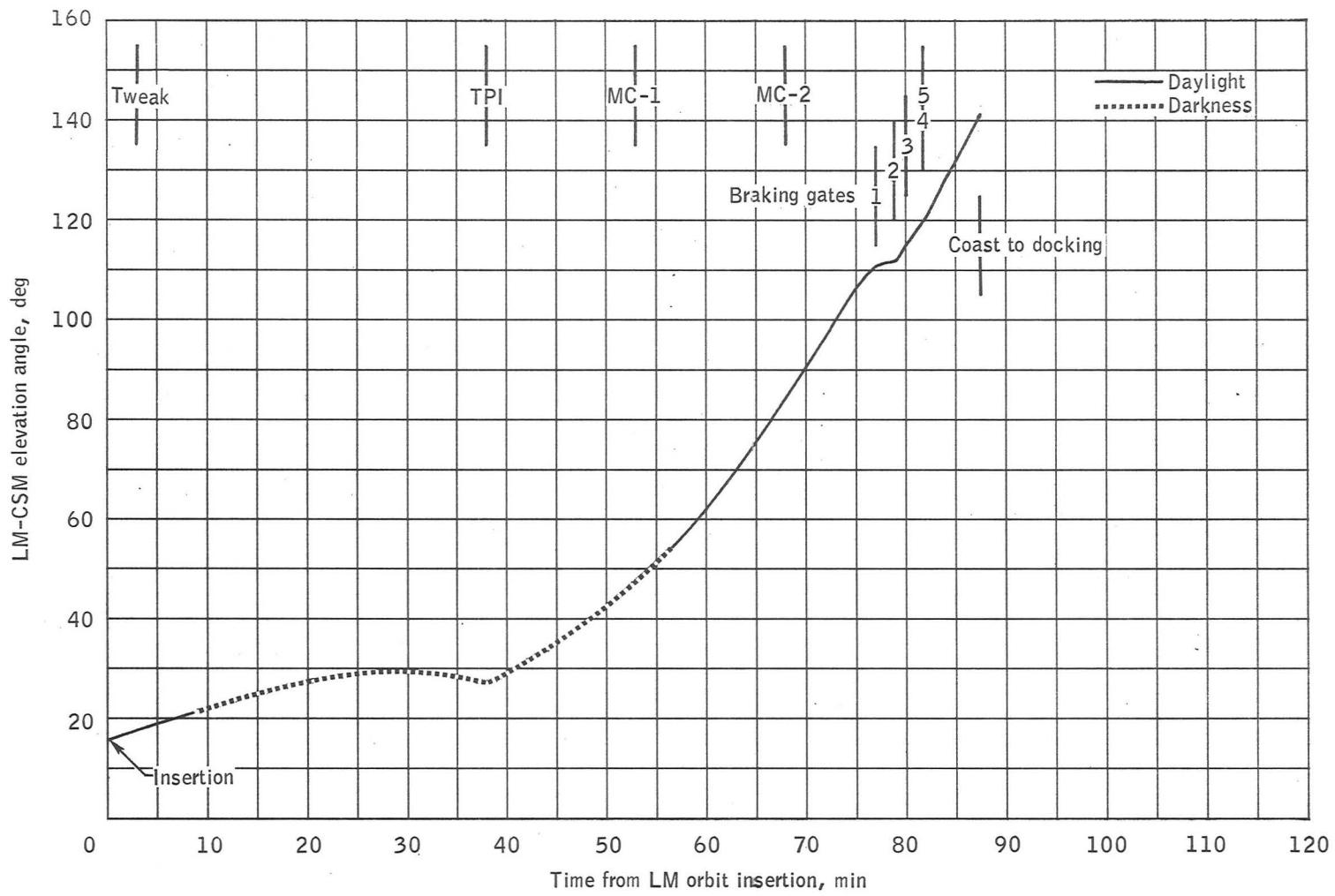
IGA, deg	92.1
MGA, deg	0.0
OGA, deg	0.0

<sup>a</sup>All targets include  $\Delta V$ 's from ullage.<sup>b</sup> $t_{IG}$  is ullage on time.



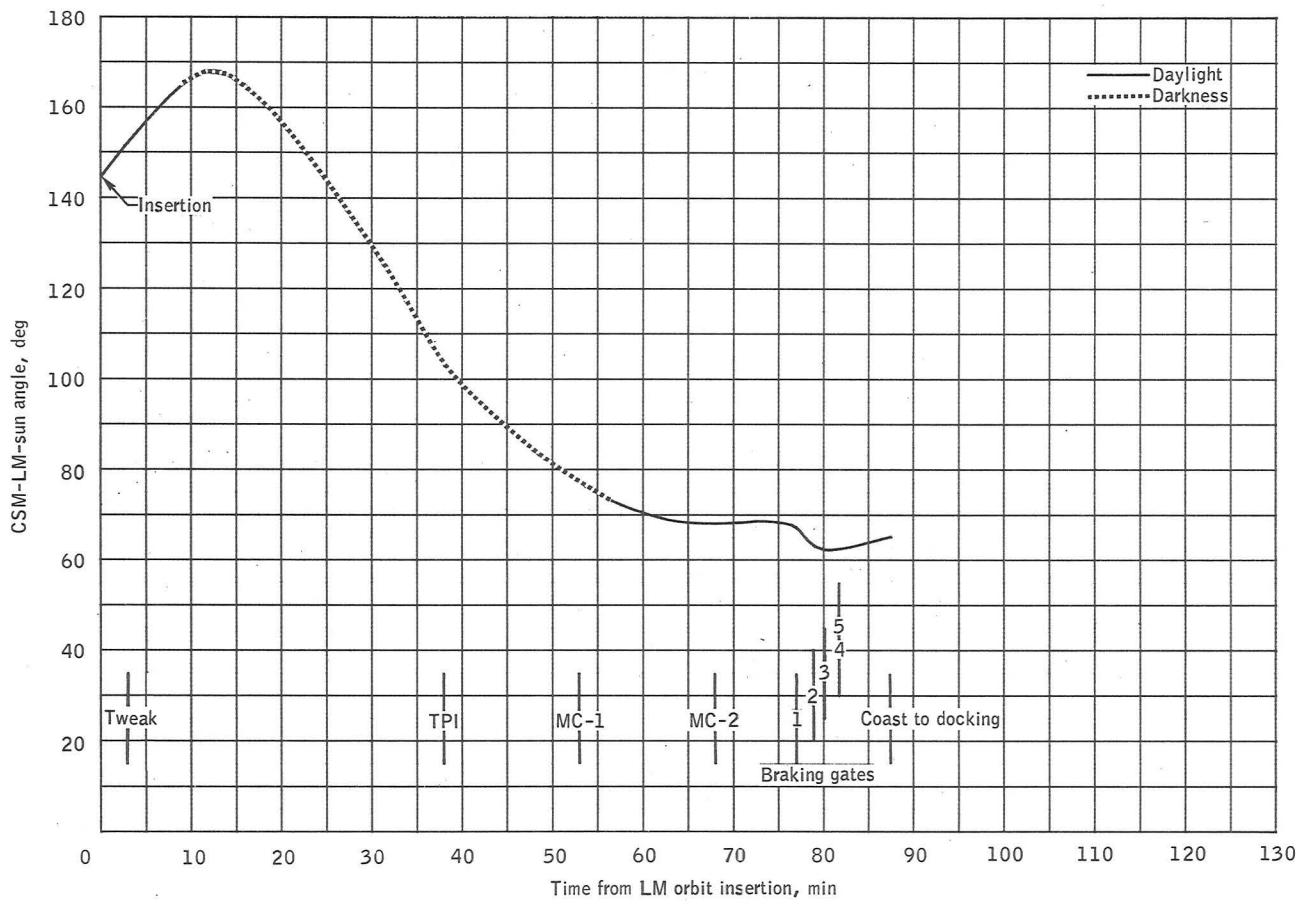
Event	Ground elapsed time, hr:min:sec
1. Lift-off	142:24:29.2
2. LM insertion	142:31:39.9
3. Tweak	142:34:39.9
4. TPI (ullage on)	143:09:26.9
5. MC-1	143:24:39.9
6. MC-2	143:39:39.9
7. Begin braking	143:50:32.8
8. Begin stationkeeping	143:54:47.8
9. Docking	144:10:00.0

Figure 4.13-1.- Orbital schematic of ascent through docking.



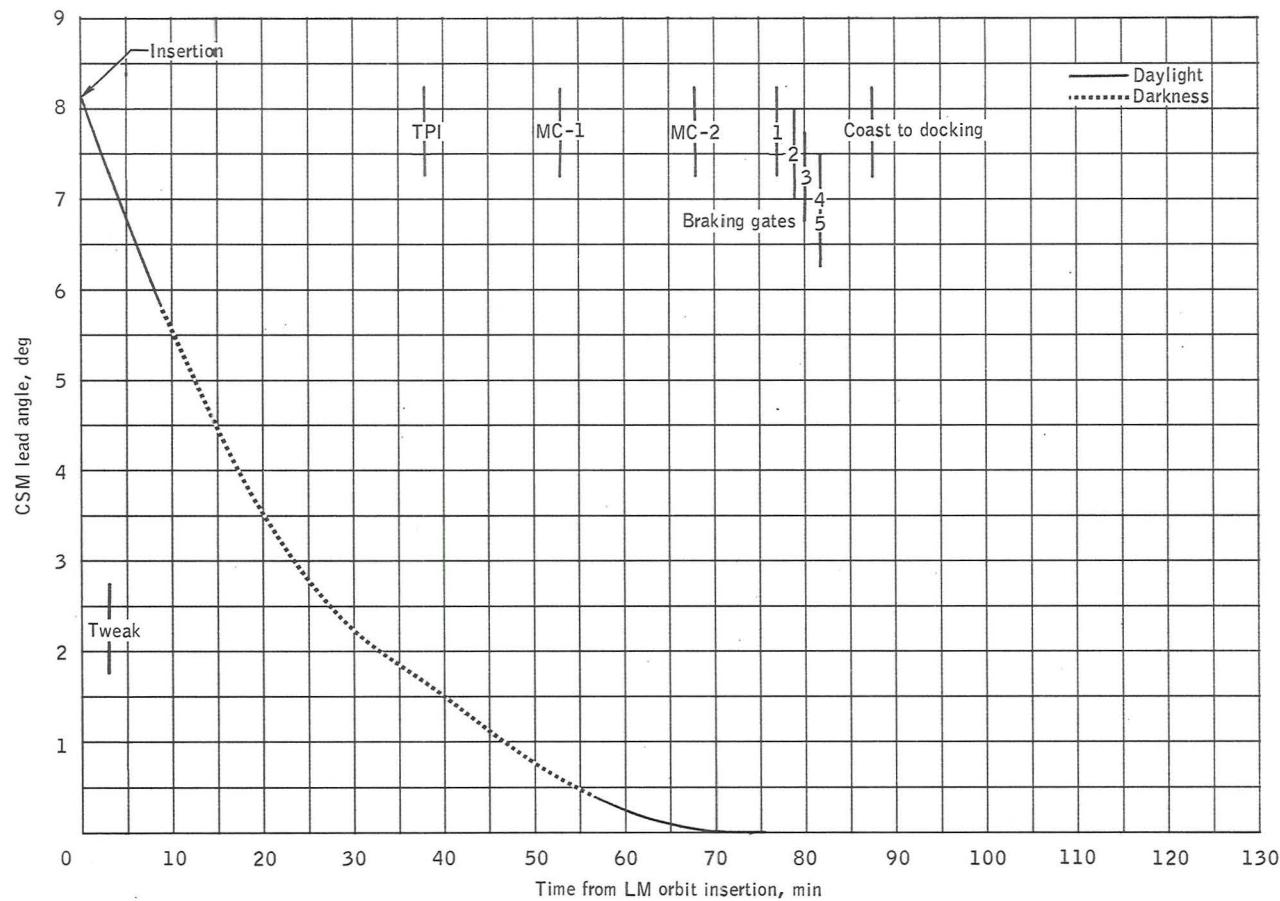
(a) LM-CSM elevation angle versus time from LM orbit insertion.

Figure 4.13-2.- Parameters from LM orbit insertion through terminal rendezvous.



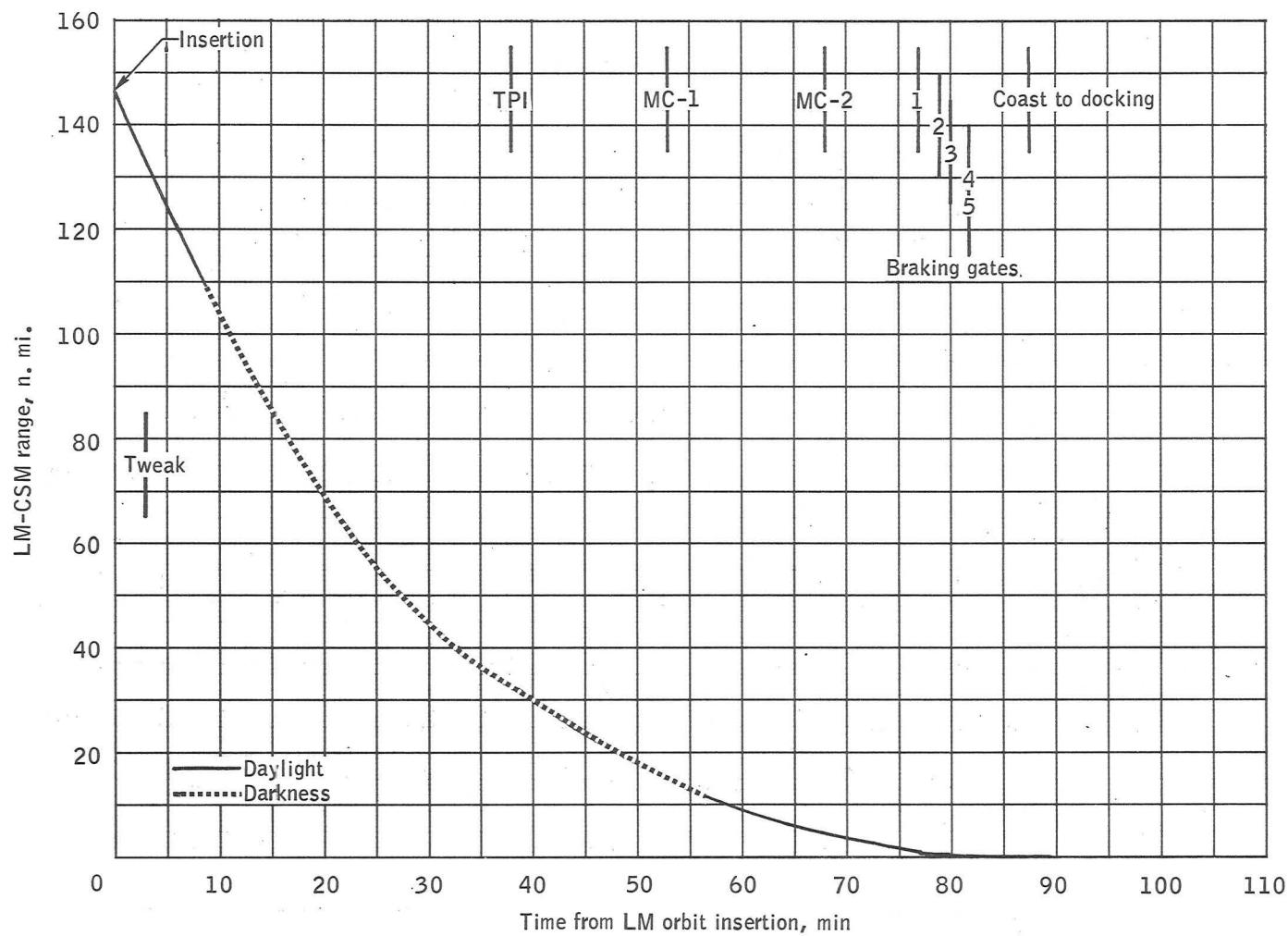
(b) CSM-LM-sun angle versus time from LM orbit insertion.

Figure 4.13-2.- Continued.



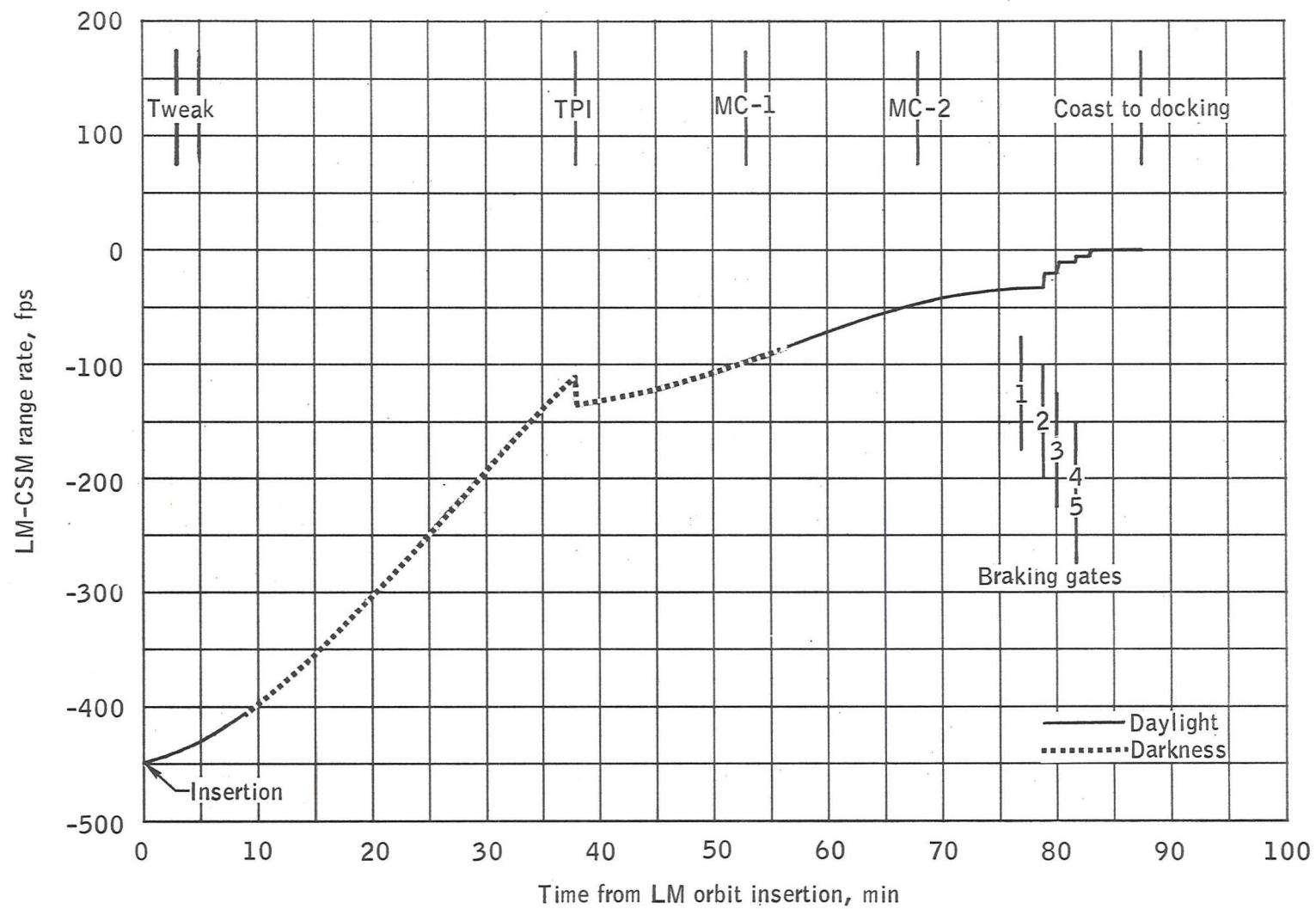
(c) CSM lead angle versus time from LM orbit insertion.

Figure 4.13-2.- Continued.



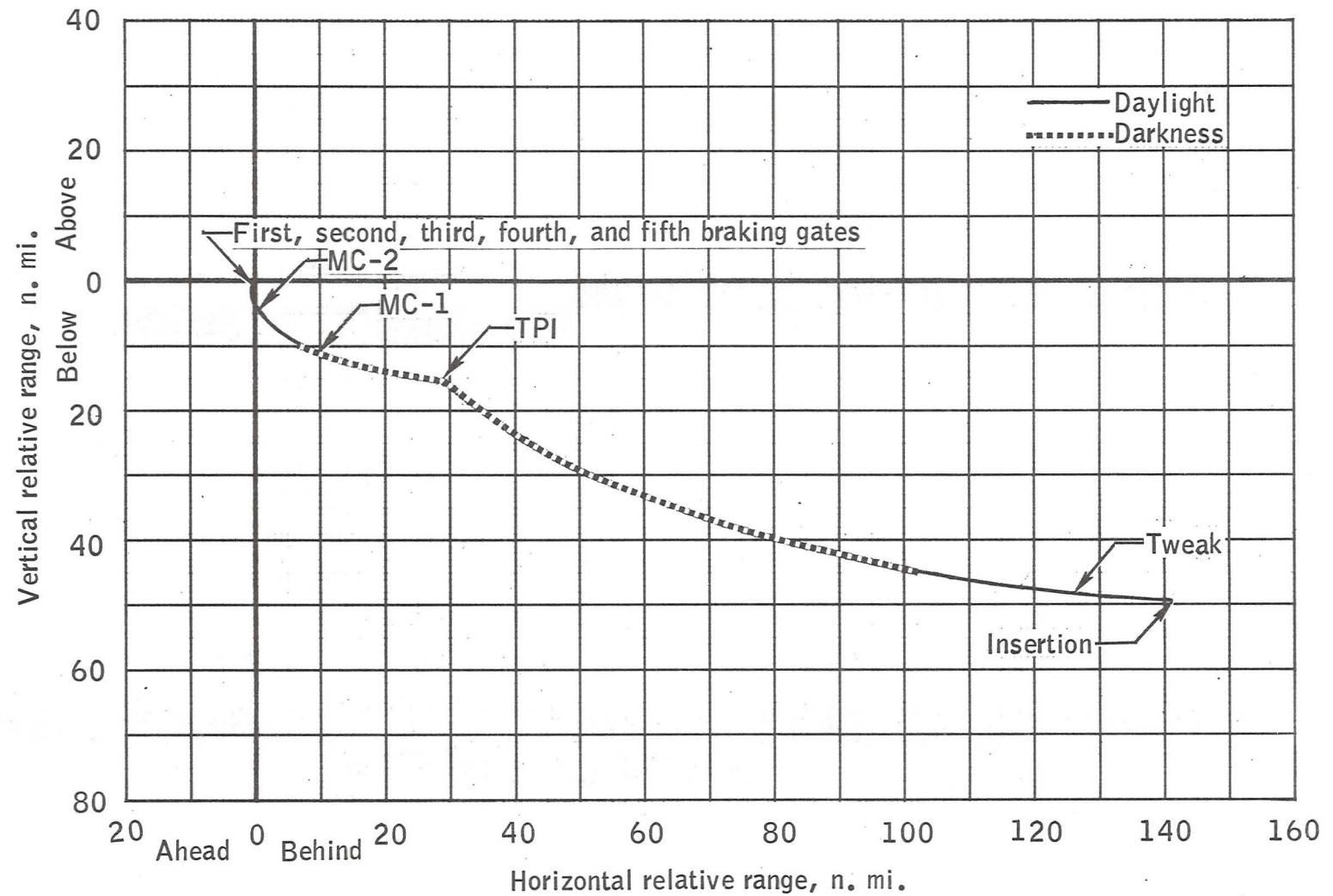
(d) LM-CSM range versus time from LM orbit insertion.

Figure 4.13-2.- Continued.



(e) LM-CSM range rate versus time from LM orbit insertion.

Figure 4.13-2.- Continued.



(f) Vertical relative range versus horizontal relative range.

Figure 4.13-2.- Concluded.

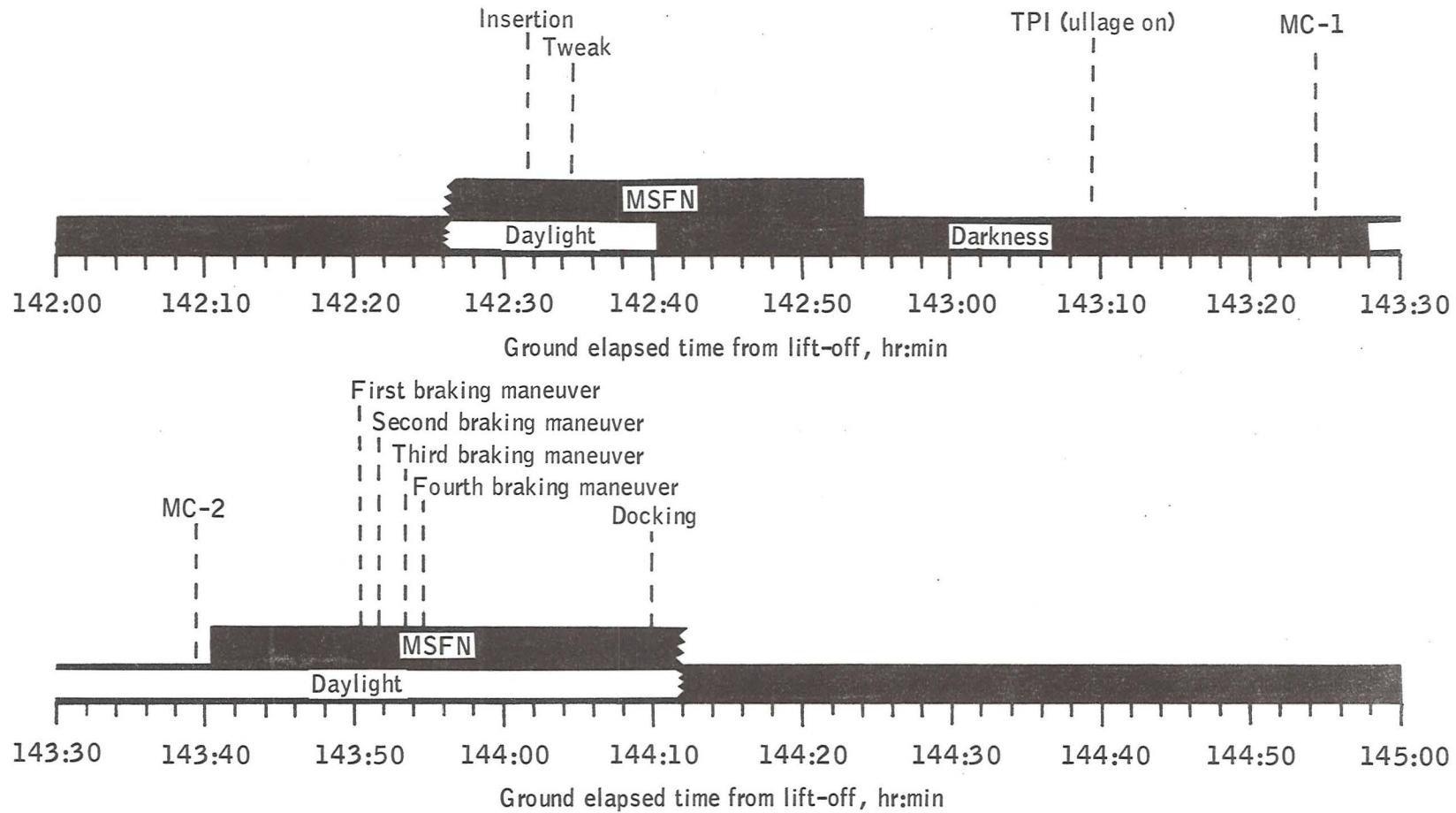


Figure 4.13-3.- Mission events and tracking summaries for LM-active rendezvous.

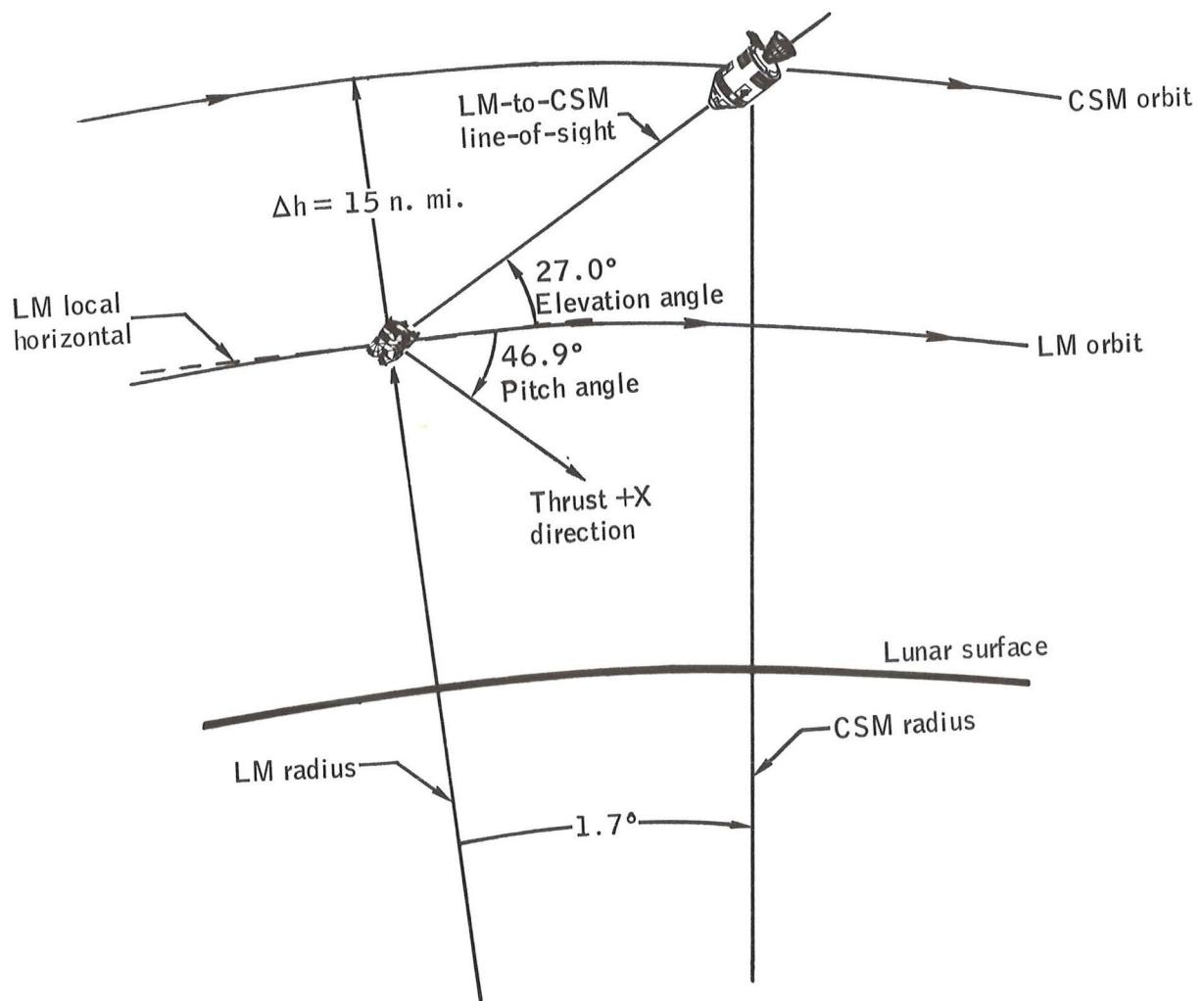


Figure 4.13-4.- LM-CSM relative geometry at TPI.

4.14 LM JETTISON AND SEPARATION TO TEI

#### 4.14 LM Ascent Stage Jettison and Separation to Transearth Injection

Docking occurs at approximately  $144^{h}10^{m}$  g.e.t. After equipment transfer and decontamination procedures, the LM ascent stage is jettisoned at  $146^{h}23^{m}30.6^{s}$  g.e.t. The CSM separates from the ascent stage by performing a +Z RCS translation of 1.0 fps in the inertial deorbit attitude. This separation maneuver occurs approximately  $270^{\circ}$  prior to the deorbit maneuver and is in a retrograde direction. The CSM target load for this separation maneuver is shown in table 4.14-I.

The discarded LM ascent stage will be targeted to impact the lunar surface at  $3.30^{\circ}$ S and  $19.27^{\circ}$ W approximately 29 n. mi. from the Apollo 14 landing site. The ascent stage deorbit burn duration will be 77.0 seconds beginning at  $147^{h}52^{m}58.9^{s}$  g.e.t. This RCS burn (+X, four-jet) will be performed at a yaw attitude of approximately  $168^{\circ}$ . The burn targets are  $\Delta V_x = -180.0$  fps and  $\Delta V_y = 36.5$  fps, which will result in a total velocity change of 183.7 fps. After the burn, the ascent stage is in a  $58.4^{\circ}$ - by  $-63.9$ -n. mi. orbit. The ascent stage will travel through a central angle of  $85^{\circ}$  before it impacts at approximately 26 minutes after the deorbit burn. At landing, the ascent stage will have a velocity of approximately 5506 fps (5500 fps horizontal and 365 fps radial). The flight-path angle of the ascent stage at impact will be approximately  $-4^{\circ}$ . The ascent stage deorbit and landing are summarized in table 4.14-II with the ascent stage target load given in table 4.14-III. The groundtrack from LOPC to TEI is shown in figure 4.14-1.

TABLE 4.14-I.- TARGET LOAD FOR CSM SEPARATION

## MANEUVER FOLLOWING LM JETTISON

[Propulsion system: CSM RCS]

(a) Target

(b) REFSMMAT

Refer to table 5-V

(c) Gimbal angles at  $t_{TG}$

IGA, deg	.....	-16.8
MGA, deg	.....	0.0
OGA, deg	.....	180.0

TABLE 4.14-II.-- LM DEORBIT SUMMARY

Event	Time, day:hr:min:sec g.e.t.	Latitude, deg	Longitude, deg	Burn duration, sec	$\Delta V$ , fps	Inertial velocity, fps	Mass, lb	Local horizontal attitudes, deg	
								Pitch	Yaw
Deorbit	6:02:23:30.6	-11.79	67.43	77.0	183.7	5351.6	5277.2	2.0	168.5
Impact	6:04:20:58.1	-3.50	-19.29	--	--	5505.9	5171.1	-96.1	168.1

TABLE 4.14-III.—TARGET LOAD FOR LM DEORBIT

### [Propulsion system LM RCS]

(a) Target

(b) .REFSMMAT

Refer to table 5-V

(c) Gimbal angles at  $t_{TG}$

IGA, deg	175.2
MGA, deg	11.4
OGA <sup>a</sup> , deg	180.4

a Will be optimized for best communications attitude

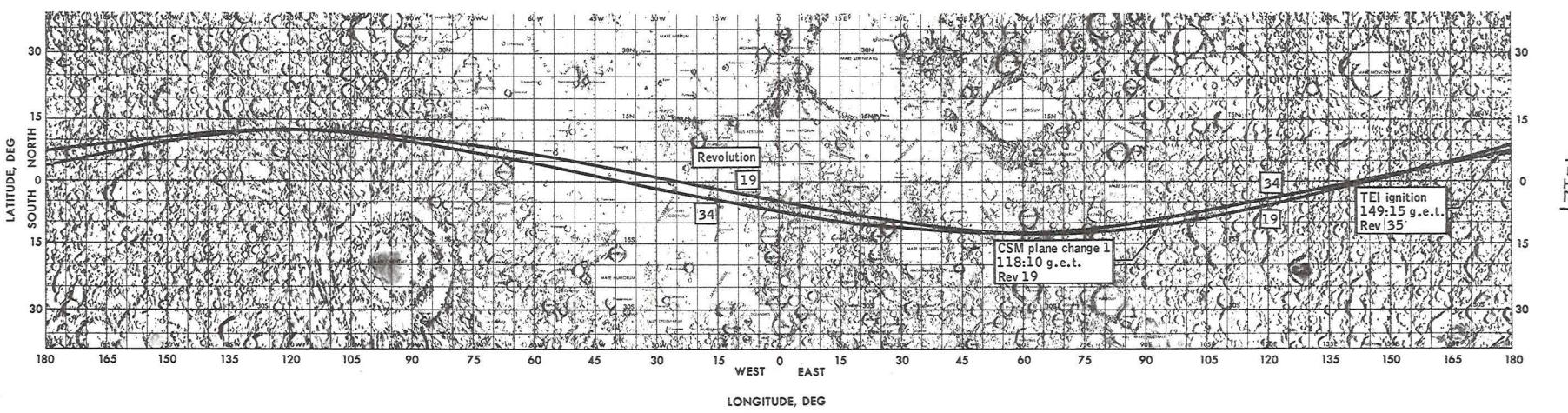


Figure 4.14-1.- Groundtrack from LOPC to TEI.

4.15 TRANSEARTH INJECTION

## 4.15 Transearth Injection

The TEI maneuver occurs  $2^d 18^h 30^m$  after LOI. The burn was targeted for a 67.2-hour transearth flight time and a  $40^\circ$  ascending return inclination relative to the earth's equator. A time history of the trajectory parameters during the burn is presented in figure 4.15-1. The transearth injection geometry is shown in figure 4.15-2.

For the TEI burn, the IMU is aligned to the ascent REFSMMAT. The target load is presented in table 4.15-I.

The characteristics of the burn are the following.

## TEI burn initiation

Time, hr:min:sec, g.e.t. . . . .	149:14:50.1
Mass, lb . . . . .	34 515.7
Altitude above LLS, n. mi. . . . .	59.3
Selenographic latitude, deg . . . . .	7.1
Selenographic longitude, deg . . . . .	177.3
Inertial velocity, fps . . . . .	5346.4
Selenographic inclination, deg . . . . .	12.2
Perilune altitude above LLS, n. mi. . . . .	58.1

## TEI burnout

Time, hr:min:sec, g.e.t. . . . .	149:17:17.3
Altitude above LLS, n. mi. . . . .	66.4
Selenographic latitude, deg . . . . .	6.5
Selenographic longitude, deg . . . . .	167.8
Inertial velocity, fps . . . . .	8503.6
Selenographic inclination, deg . . . . .	6.5
Burn duration, min:sec . . . . .	147.1
Inertial burn arc, deg . . . . .	9.4
Plane change, deg . . . . .	11.0
$\Delta V$ , fps . . . . .	3449.5
SPS propellant used, lb . . . . .	9995.7
Perilune altitude above LLS, n. mi. . . . .	57.5

TABLE 4.15-1.- TARGET LOAD FOR TEI MANEUVER

[Propulsion system: SPS; guidance: external  $\Delta V$ ]

(a) Target

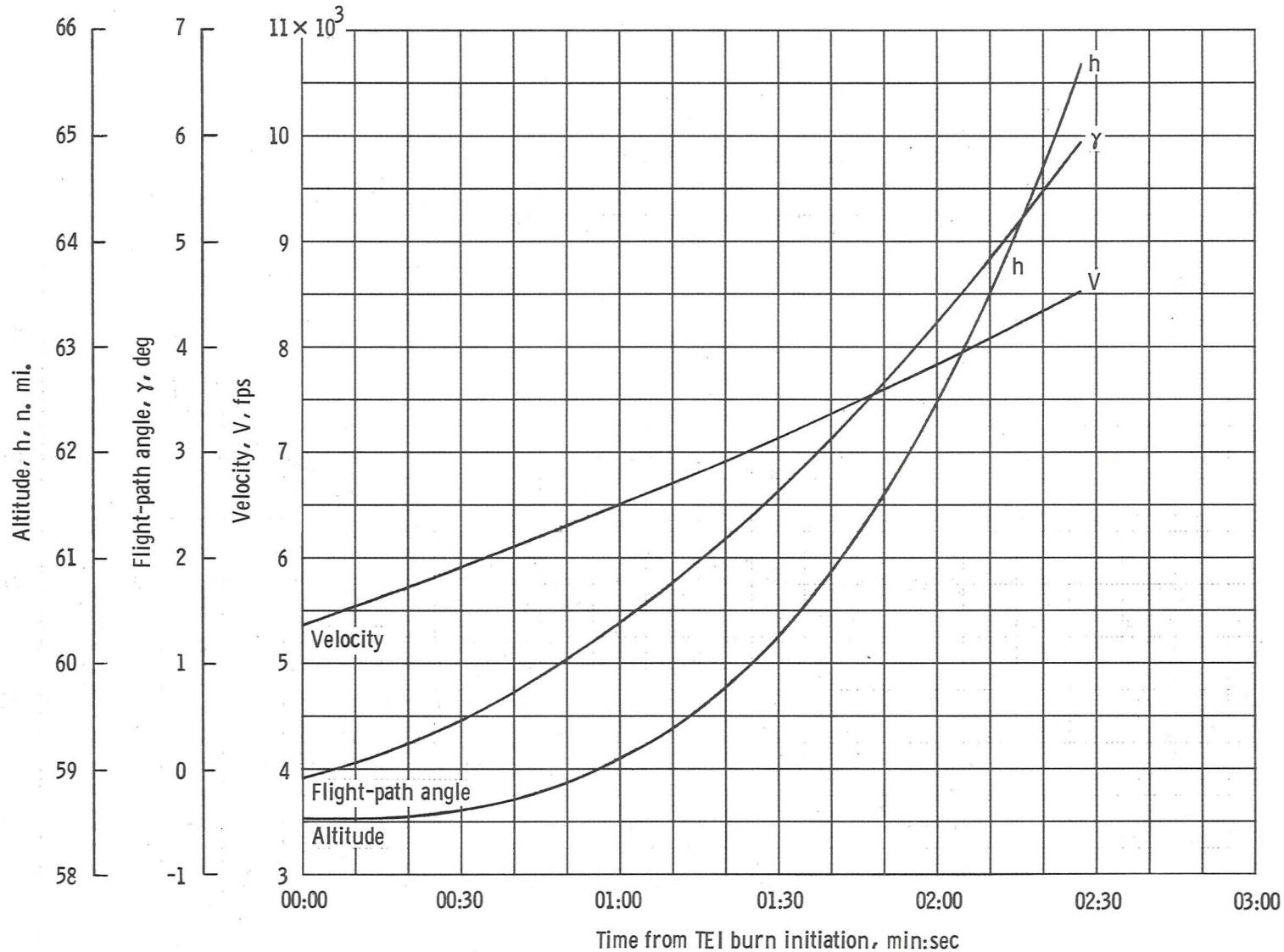
$t_{IG}$ , hr:min:sec, g.e.t. . . . . . . . . . . . . . . . . .	149:14:50.1
$\Delta V_X$ , fps .	3014.7
$\Delta V_Y$ , fps .	1622.2
$\Delta V_Z$ , fps .	-423.7
Weight, lb .	34 515.7

(b) REF SMMAT

Refer to table 5-VII

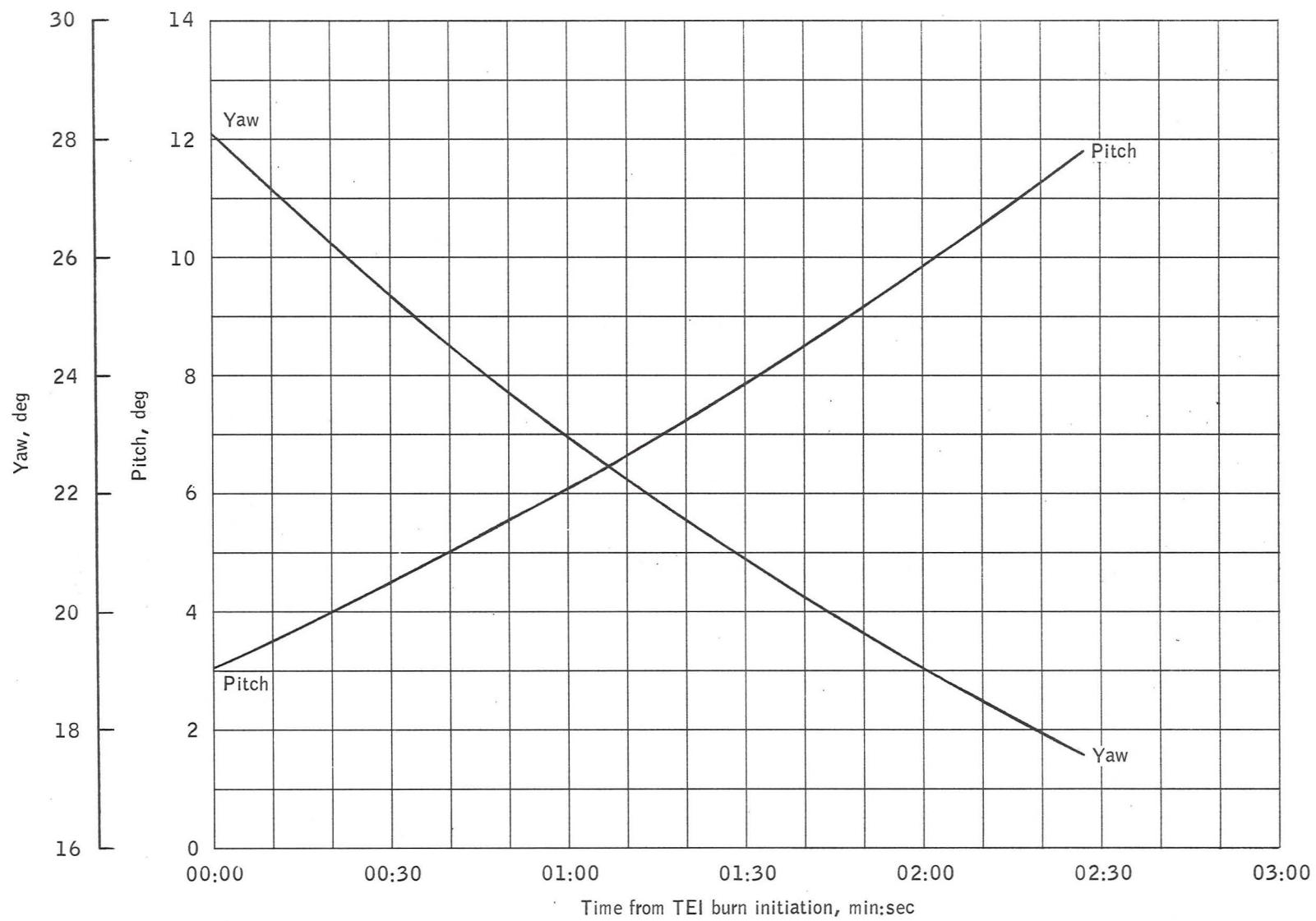
(c) Gimbal angles at  $t_{IG}$

IGA, deg	109.0
MGA, deg	29.2
OGA, deg	180.0



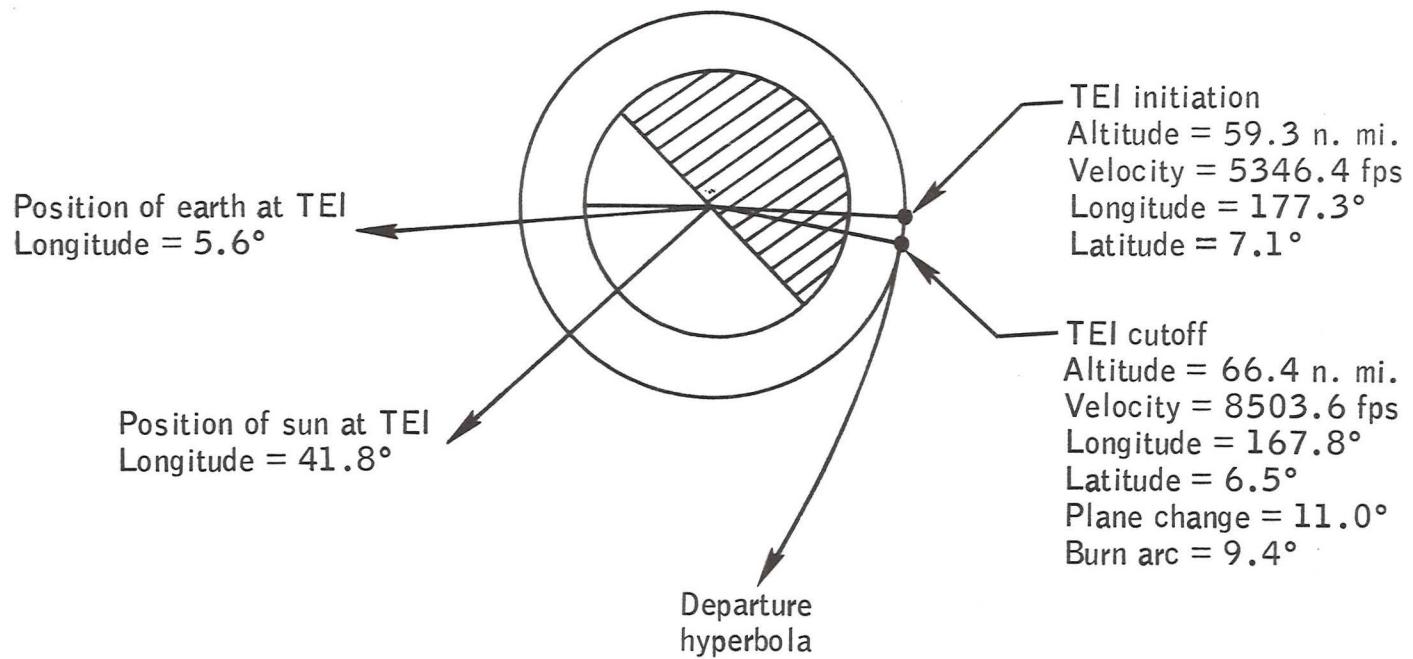
(a) Velocity, flight-path angle, and altitude versus time from TEI burn initiation.

Figure 4.15-1. - Time history of trajectory parameters for transearth injection.



(b) Local horizontal pitch and yaw versus time from TEI burn initiation.

Figure 4.15-1.- Concluded.



4-135

Figure 4.15-2.- TEI geometry.

4.16 TRANSEARTH COAST

#### 4.16 Transearth Coast

A groundtrack of the transearth coast phase is provided in figure 4.16-1. Altitude above the moon is shown for the first 10 hours of coast in figure 4.16-2. The altitude time history for TEI to entry is shown in figure 4.16-3. Three midcourse decision points have been defined for the transearth phase.

- a. MCC-5, TEI plus 17 hours
  - b. MCC-6, EI minus 22 hours
  - c. MCC-7, EI minus 3 hours

The maneuvers will be targeted for corridor control only. The midcourse strategy, which includes the threshold values for each maneuver, is contained in reference 4. Altitude is plotted against time for the last 10 hours of transearth coast in figure 4.16-4.

The SC remains in the lunar umbra during the entire TEI burn, enters sunlight approximately 5 minutes after cutoff, and remains in sunlight until entry.

Passive thermal control will be maintained during most of the trans-earth coast phase. The PTC REFSMMAT (table 5-II) will be used for MCC-5 and MCC-6; the entry REFSMMAT (table 5-VI) will be used for MCC-7. The CM/SM separation occurs approximately 15 minutes prior to entry.

The SC is acquired by MSFN approximately 11 minutes after TEI cutoff, and the last station to lose signal is Honeysuckle ( $5^{\circ}$  minimum elevation) which loses signal at  $240^{\text{h}}46^{\text{m}}$  g.e.t. or approximately 4 minutes prior to entry.

The primary trajectory parameters at entry interface are the following.

Time, hr:min:sec, g.e.t. . . . .	216:26:58.7
Transearth coast time, hr . . . . .	67.2
Inertial velocity, fps . . . . .	36 170.1
Altitude, n. mi. . . . .	65.8
Inertial flight-path angle, deg . . . . .	-6.5
Latitude, deg . . . . .	-36.3
Longitude, deg . . . . .	166.4
Equatorial inclination (ascending), deg . . . . .	40

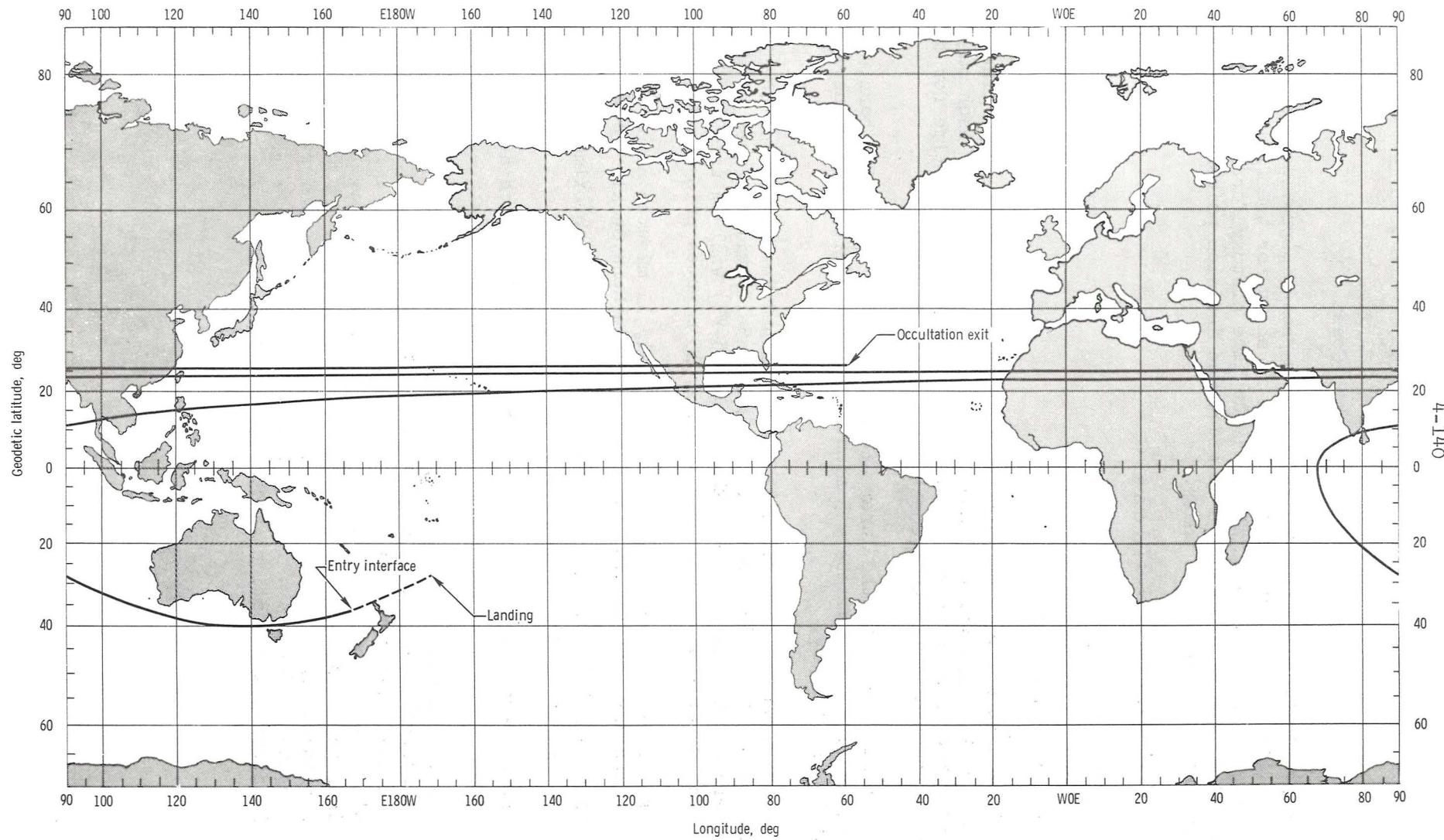


Figure 4.16-1. - Transearth coast groundtrack.

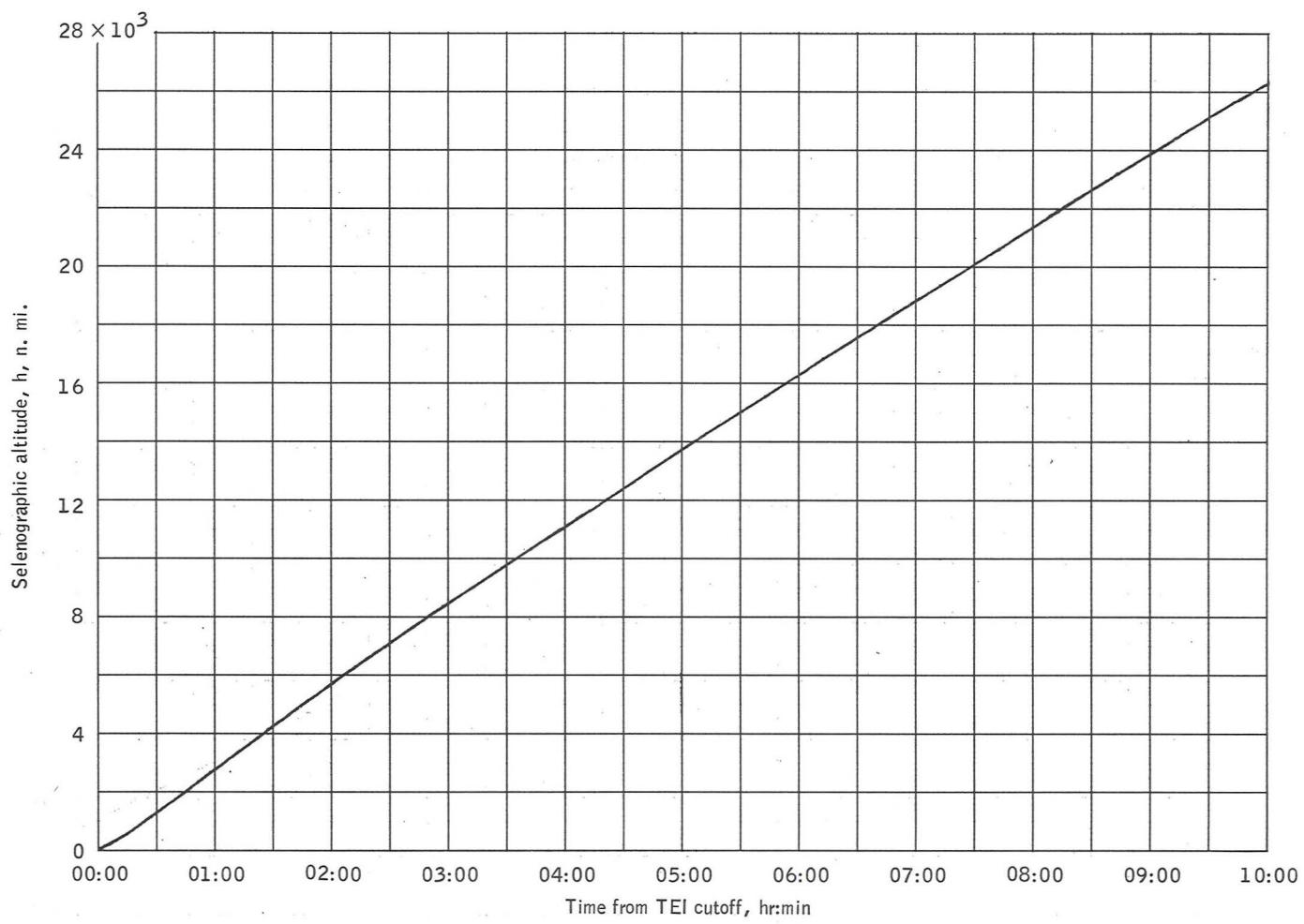


Figure 4.16-2.- Altitude time history during the first 10 hours of transearth coast.

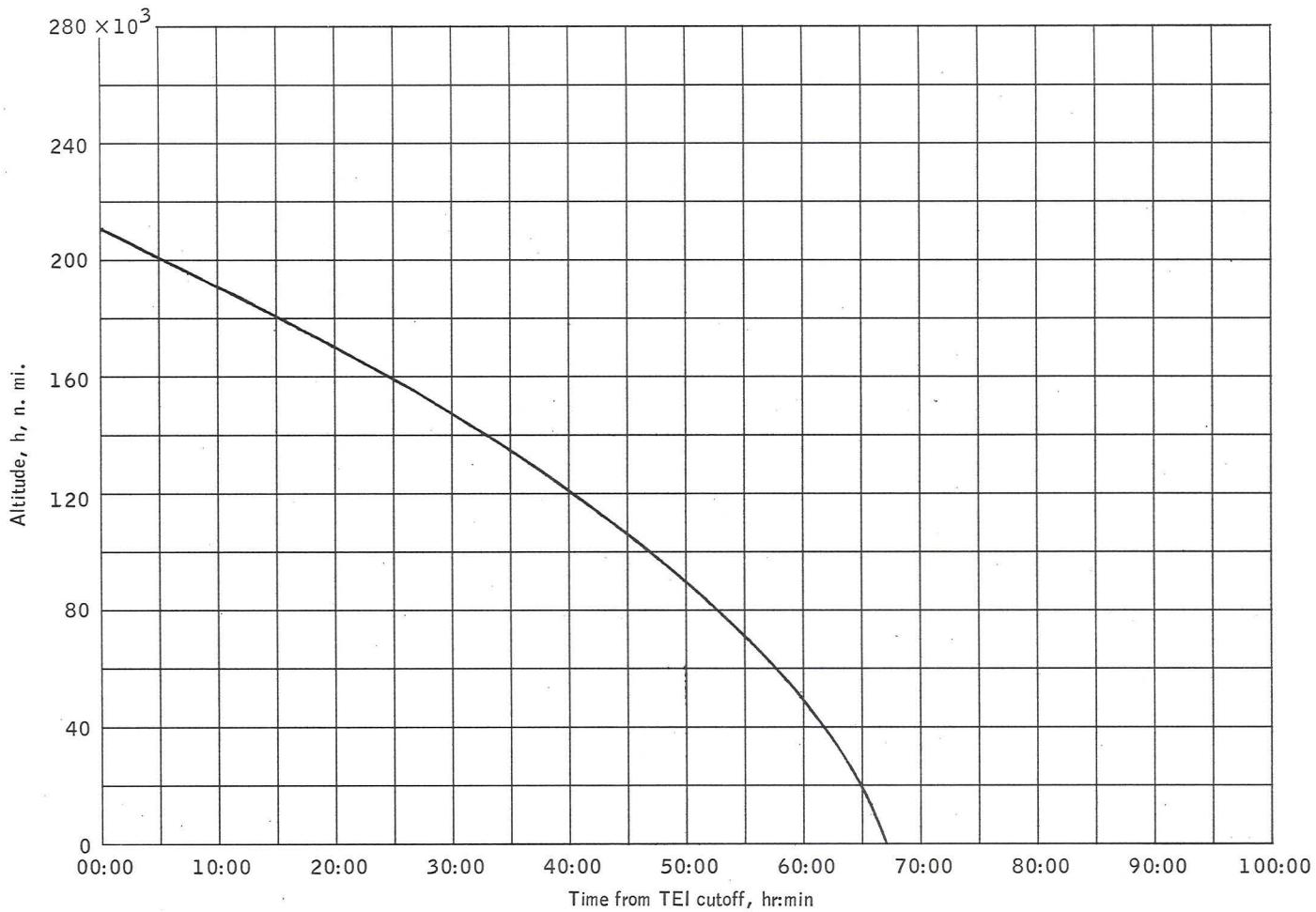


Figure 4.16-3.- Altitude time history from transearth injection to entry interface.

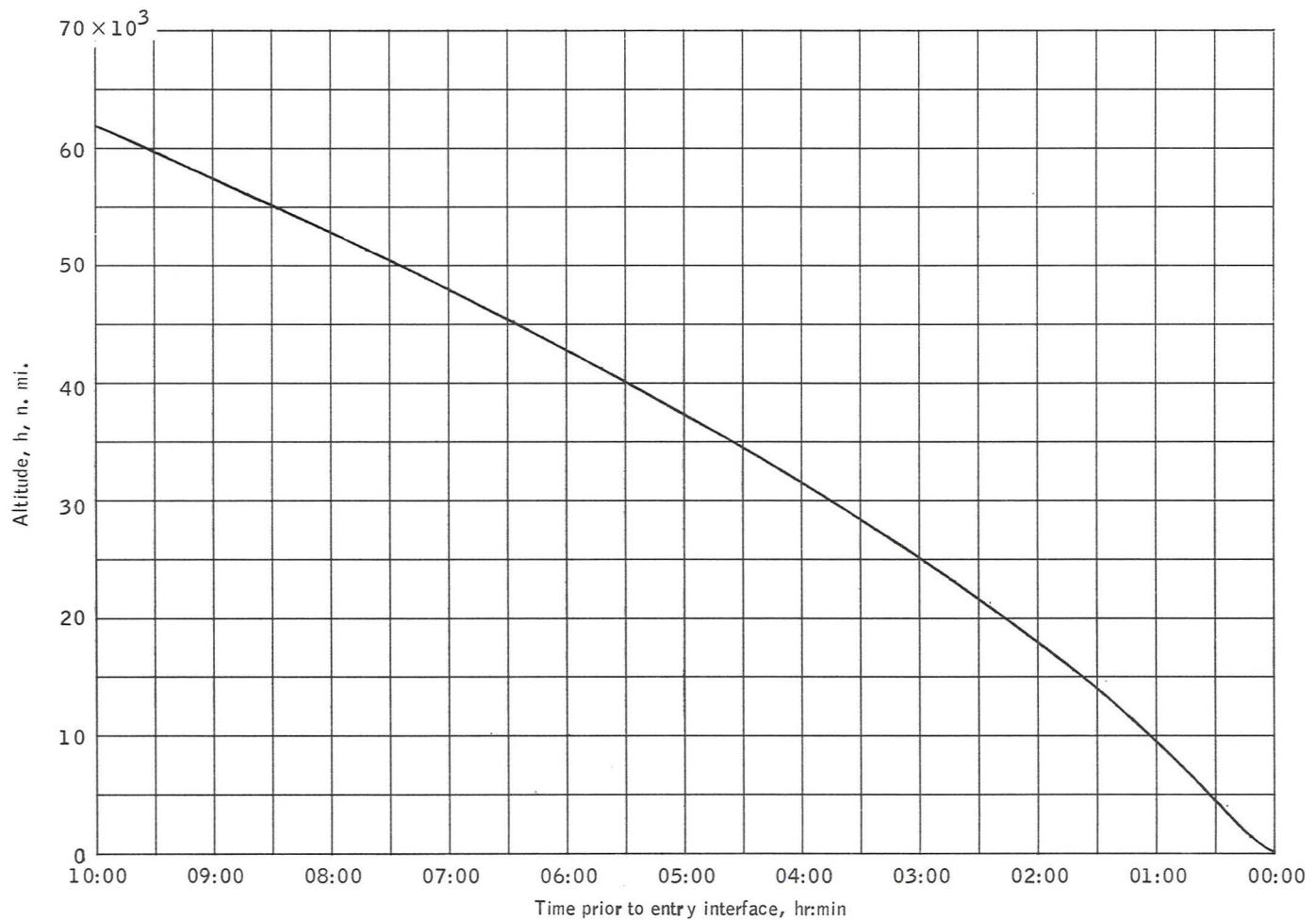


Figure 4.16-4.- Altitude time history 10 hours prior to entry interface.

4.17 ENTRY

## 4.17 Entry

The entry phase of the operational trajectory was simulated with the Apollo Reentry Simulation program with six degrees of freedom. Three-degree-of-freedom trajectories were used to determine the CM maneuver footprint. The entry ocorrider is presented in figure 4.17-1.

At the nominal EI time,  $216^{\text{h}}26^{\text{m}}59^{\text{s}}$  after lift-off, the CM is at an altitude of 400 111 feet, and the coordinates are  $36.282^{\circ}\text{S}$  geodetic latitude and  $166.356^{\circ}\text{E}$  longitude. Inertial velocity, flight-path angle, and azimuth at this point are 36 170 fps,  $6.50^{\circ}$  below the local horizontal, and  $71.461^{\circ}$ , respectively. The gimbal angles are as follows: yaw =  $0^{\circ}$ , pitch =  $152^{\circ}$ , roll =  $0^{\circ}$ .

The CM maneuver footprint and the nominal ground trace on a map of the entry area are presented in figure 4.17-2 with the footprint extended to a 3500-n. mi. entry range. The nominal landing target location is 1250-n. mi. down range from the EI position, and the coordinates of the target are  $171.5^{\circ}\text{W}$  longitude and  $27.247^{\circ}\text{S}$  geodetic latitude. A sequence of pertinent events is given in table 4.17-I and includes the periods of communications blackout which occur along the trajectory. The guidance phases are described in figure 4.17-3 which shows altitude as a function of range to the target. Time histories of the bank angle commanded by the guidance system, of the altitude rate, and of the altitude are presented in figure 4.17-4. Time histories of the total heating rate and the total heat load are presented in figure 4.17-5. The maximum total heating rate is 304.3 Btu/ft<sup>2</sup>/sec, and the total heat load is 26 542.0 Btu/ft<sup>2</sup>. Time histories for inertial and relative velocity and flight-path angles are presented in figure 4.17-6.

The CM RCS uses 12 pounds of propellant for the separation and attitude hold maneuvers before the spacecraft reaches 400 000 feet, and the RCS uses 26.9 pounds of propellant to perform the guidance commands during the remainder of the entry. A time history of the total RCS propellant consumed from separation is presented in figure 4.17-7. The altitude is plotted in relation to relative velocity in figure 4.17-8, and the boundary for S-band communication blackout is shown (ref. 5). Time histories for the primary DSKY displays, commanded bank angle, inertial velocity, and load factor are shown in figure 4.17-9. The load factor at the c.g. reaches a first maximum of 6.60g and a second maximum of 3.12g. Time histories for the final phase DSKY displays, commanded bank angle, cross-range error, and down-range error are shown in figure 4.17-10.

The drogue parachute deployment sequence begins at an altitude of 23 300 feet, which is 8 minutes 15 seconds after EI, and the two drogue parachutes are deployed 2 seconds later. At an altitude of 10 500 feet,

the low altitude baroswitch closes, and the drogue parachutes are disconnected. The three main parachutes are deployed 1 second after the baroswitch closes. The CM, which is suspended on the main parachute, lands 13 minutes 54 seconds after EI. The relative velocity and relative flight-path angle are plotted against time from drogue chute deployment in figure 4.17-11, while load factor and altitude are plotted against the same parameter in figure 4.17-12.

An EMS scroll (non-exit pattern) is presented in figure 4.17-13(a) with the reference trajectory from 0.05g superimposed upon it. This pattern has limit lines which allow the crew to monitor the entry trajectory to prevent an exit by the spacecraft from the atmosphere ( $g = 0.02$ ). The commanded bank angle and EMS range-to-go are plotted against the inertial velocity in figure 4.17-13(b).

The CM mass properties are shown in table 4.17-II. The conditions at EI are presented in table 4.17-III, and the entry REFSMMAT is presented in table 5-VIII.

TABLE 4.17-I.- ENTRY EVENTS SEQUENCE

Event	Time from lift-off, hr:min:sec	Time from 400 000 ft, min:sec
Entry	216:26:59	00:00
Enter S-band communication blackout	216:27:17	00:18
Guidance changes from P63 to P64 (load factor = 0.05g)	216:27:29	00:30
Initiate constant drag (KA)	216:27:51	00:52
Maximum heating rate	216:28:09	01:10
Guidance initiate at RDOT > -700 fps	216:28:17	01:18
Maximum load factor (FIRST)	216:28:19	01:20
Circular velocity ( $v_{CIRC}$ )	216:29:05	02:06
Change from P64 to P67	216:29:09	02:10
Exit S-band communication blackout	216:30:31	03:32
Maximum load factor (SECOND)	216:33:15	06:16
Termination of CMC guidance	216:34:15	07:16
Drogues parachute deployment	216:35:15	08:16
Main parachute deployment	216:36:05	09:05
Landing	216:40:53	13:54

6414

TABLE 4.17-II.- COMMAND MODULE MASS PROPERTIES

CM weight, lb

Entry . . . . .	12	580.8
Main chute deploy . . . . .	12	007.3
Landing . . . . .	11	357.7

Centers of gravity in Apollo  
coordinate system, in.

$X_A$ . . . . .	1039.4
$Y_A$ . . . . .	-0.3
$Z_A$ . . . . .	6.0

Moments of inertia, slug-ft<sup>2</sup>

$I_{XX}$ . . . . .	5890.0
$I_{YY}$ . . . . .	5338.0
$I_{ZZ}$ . . . . .	4798.0

Products of inertia, slug-ft<sup>2</sup>

$I_{XY}$ . . . . .	63.0
$I_{XZ}$ . . . . .	-449.0
$I_{YZ}$ . . . . .	-25.0

TABLE 4.17-III.- CONDITIONS AT ENTRY INTERFACE AND TARGET POINT

Elapsed time from launch, hr:min:sec . . . . .	216:26:59
Inertial velocity, fps . . . . .	36 170
Inertial flight-path angle, deg . . . . .	-6.50
Inertial azimuth, deg . . . . .	71.46
Spacecraft geodetic latitude, deg S . . . . .	36.282
Spacecraft longitude, deg E . . . . .	166.36
Altitude, ft . . . . .	400 111
Target geodetic latitude, deg S . . . . .	27.25
Target longitude, deg W . . . . .	171.5
LADPAD . . . . .	0.27
LODPAD . . . . .	0.207

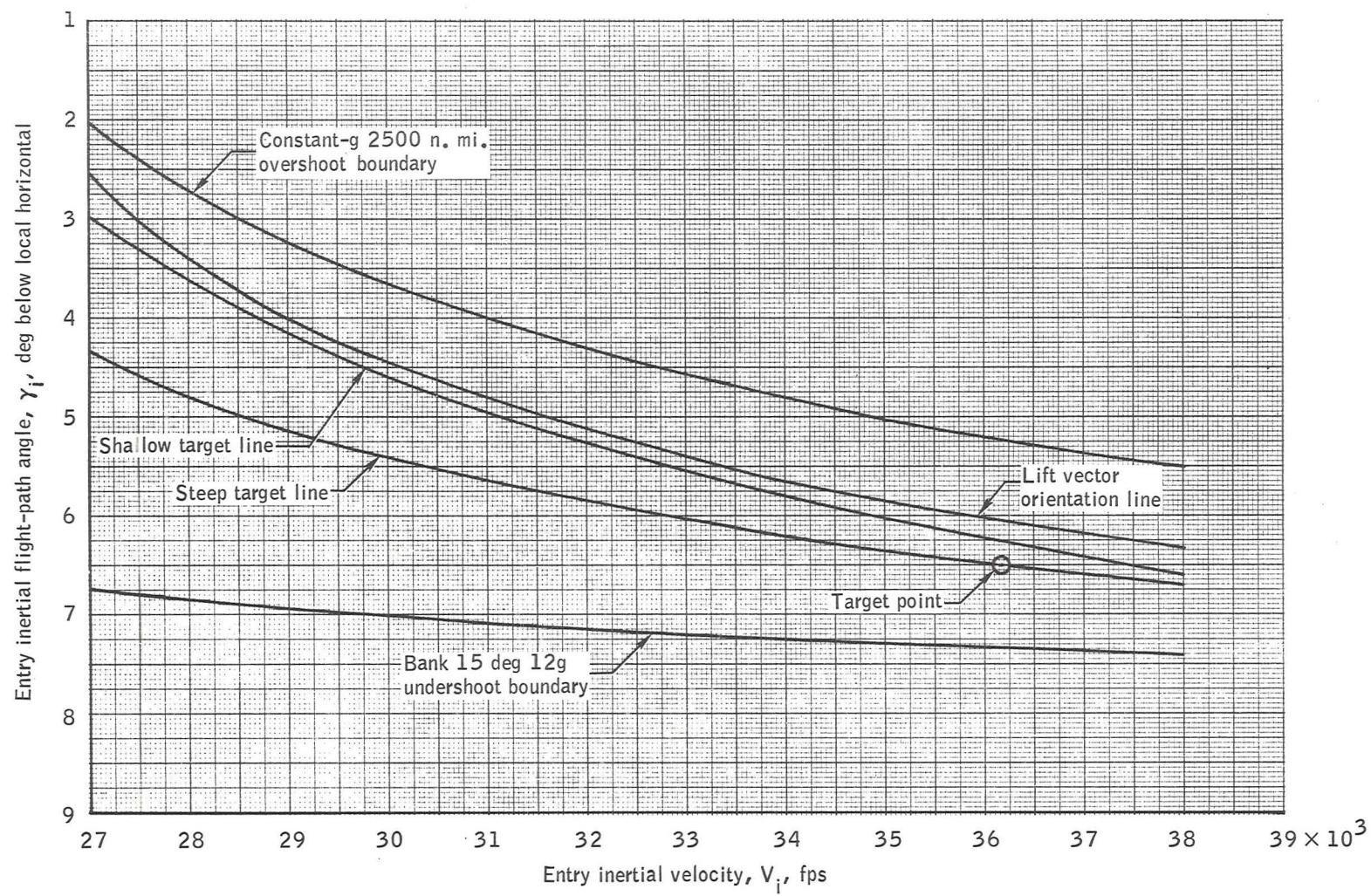


Figure 4.17-1.- Entry corridor.

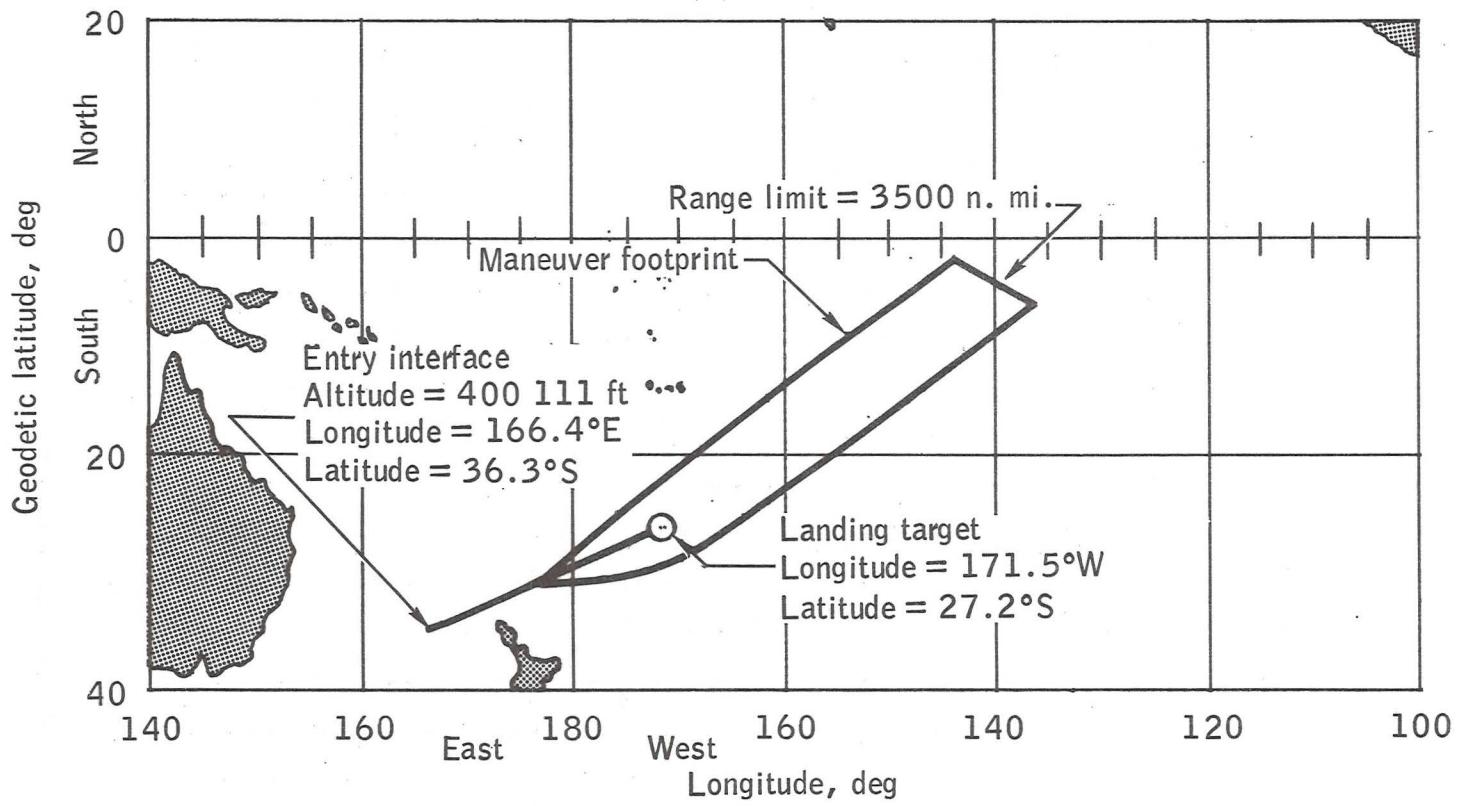


Figure 4.17-2.- Maneuver footprint and nominal groundtrack.

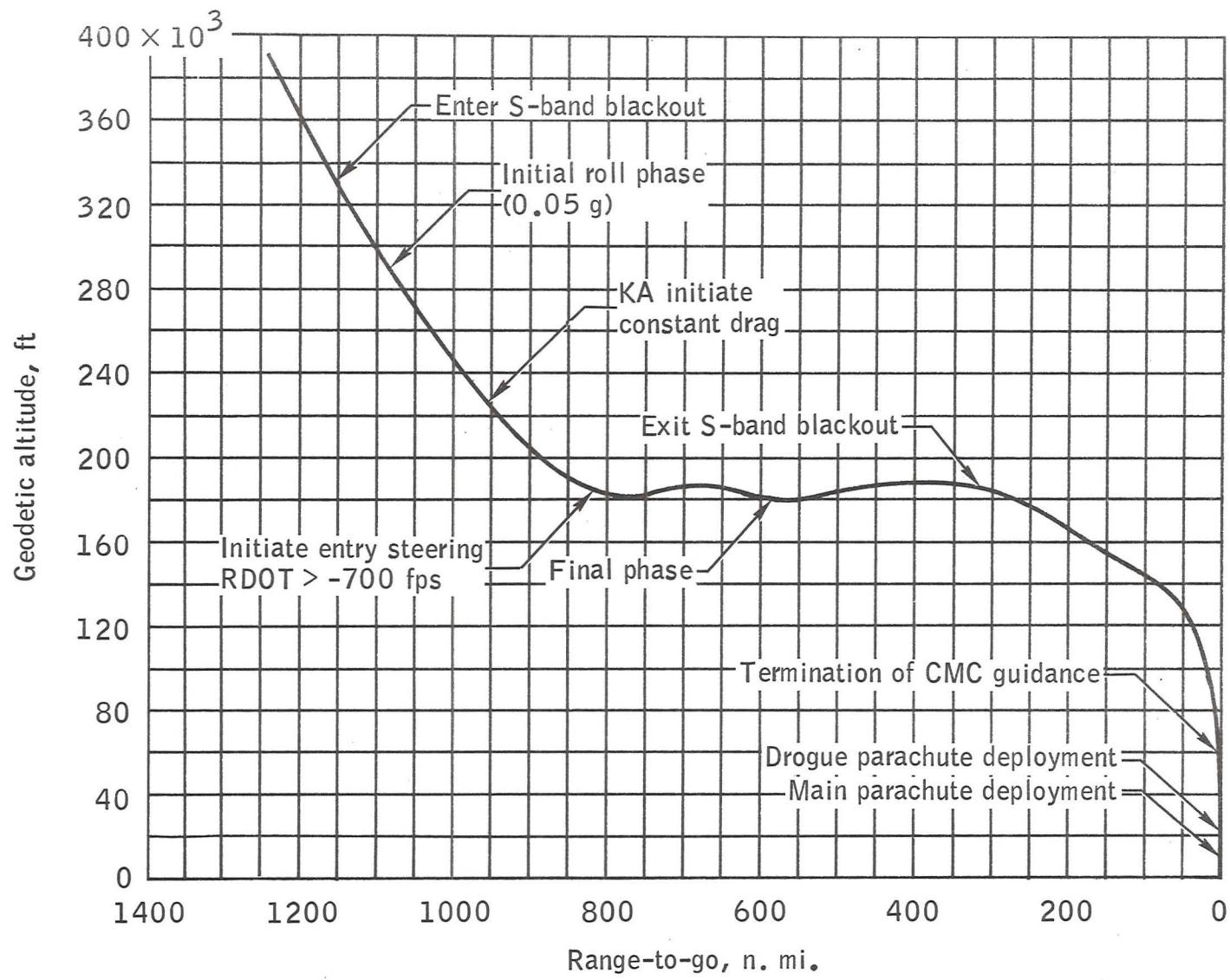


Figure 4.17-3.- Geodetic altitude versus range-to-go.

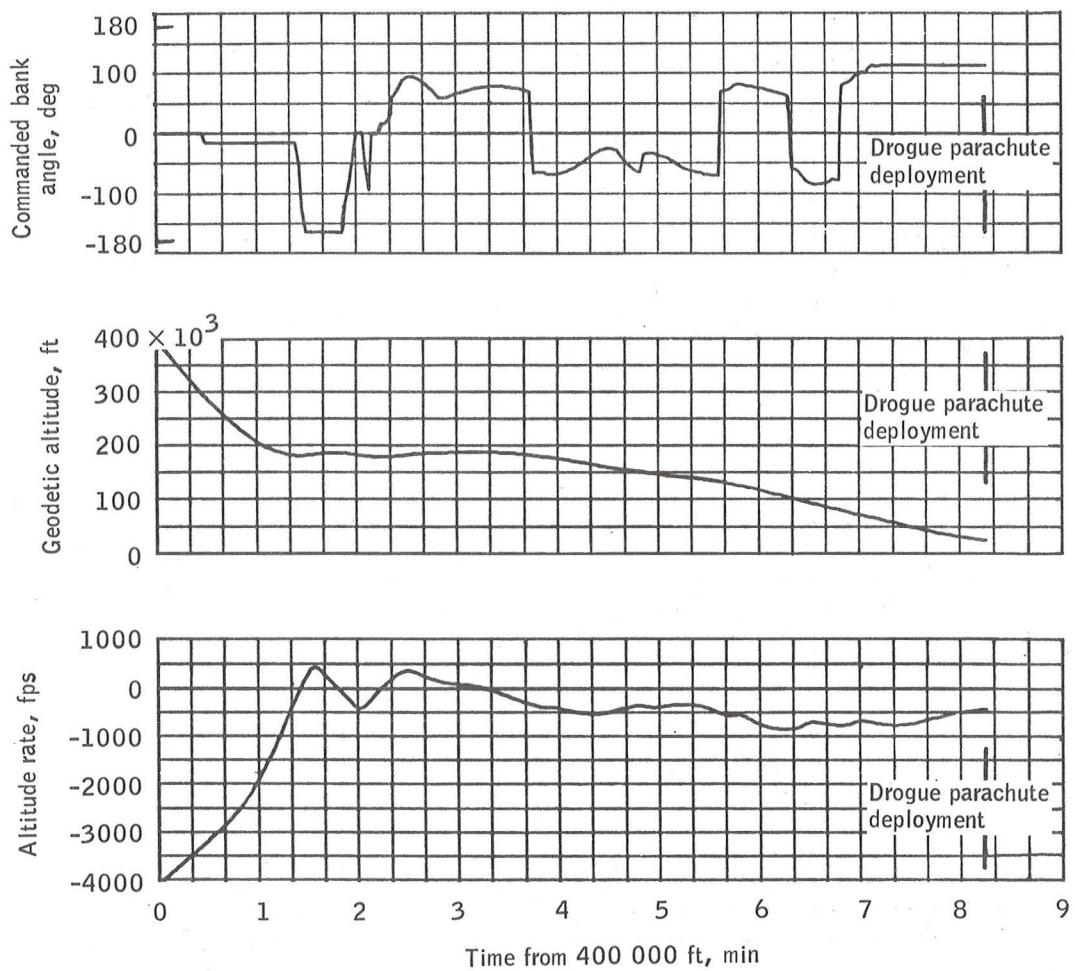


Figure 4.17-4.- CMC commanded bank angle, altitude, and altitude rate time histories.

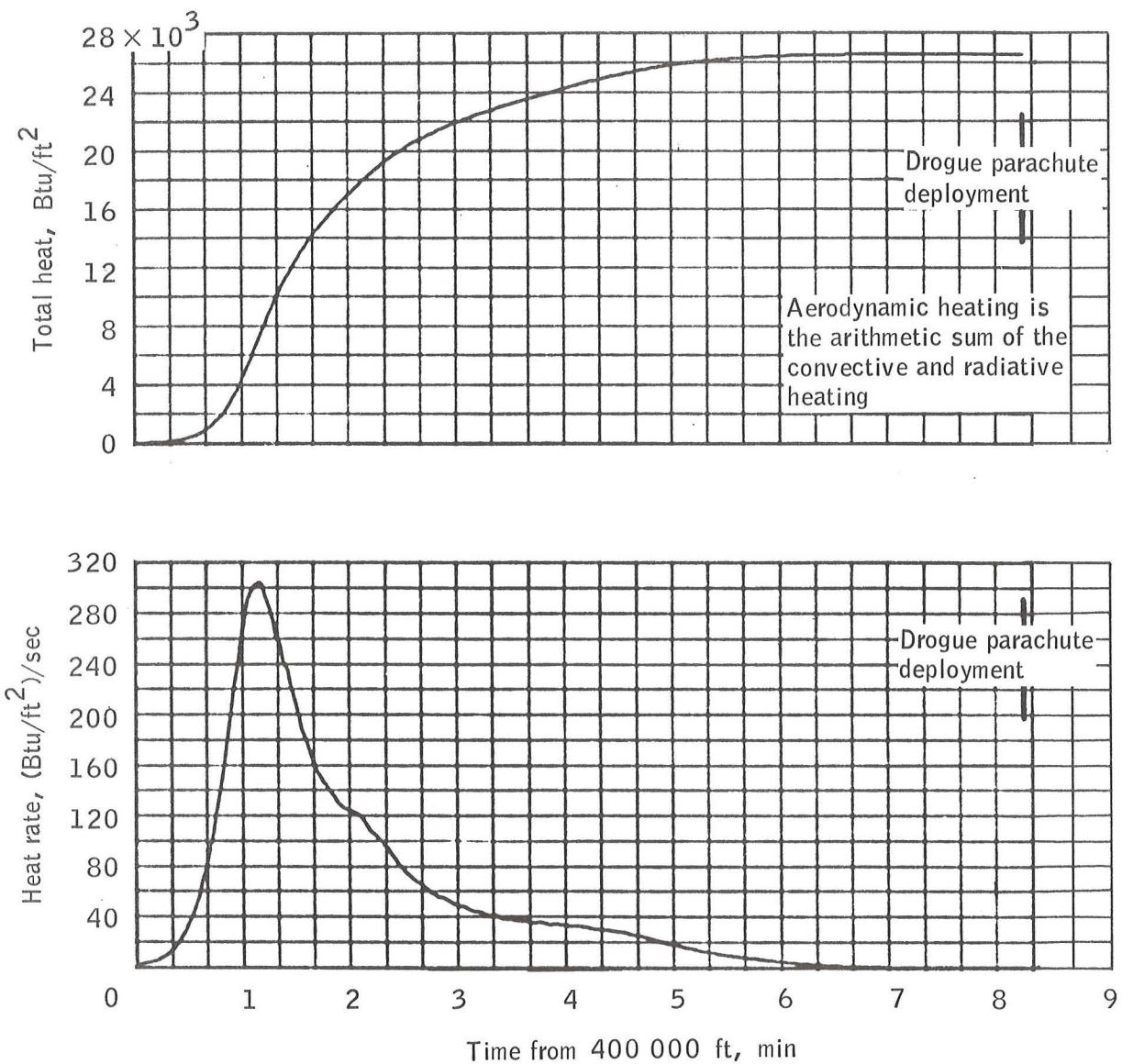
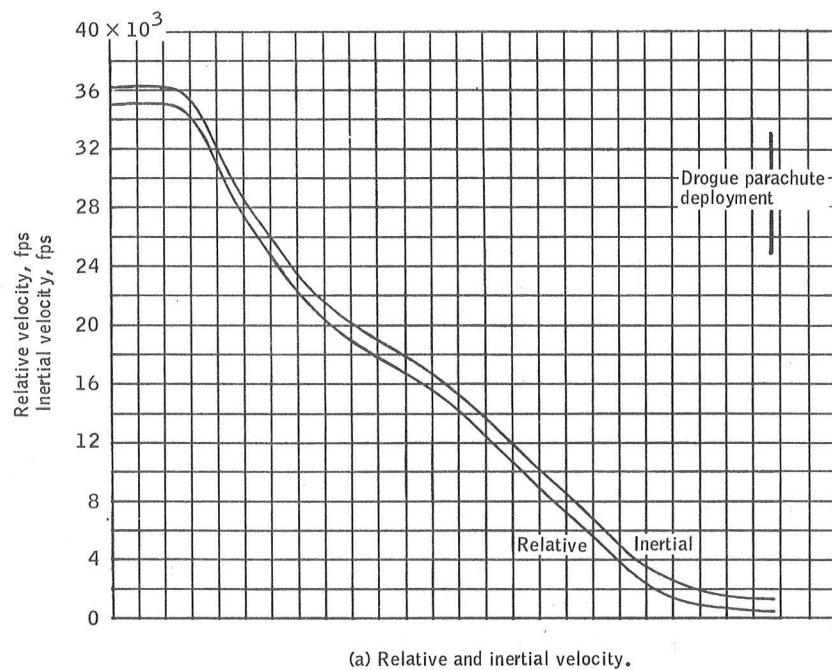
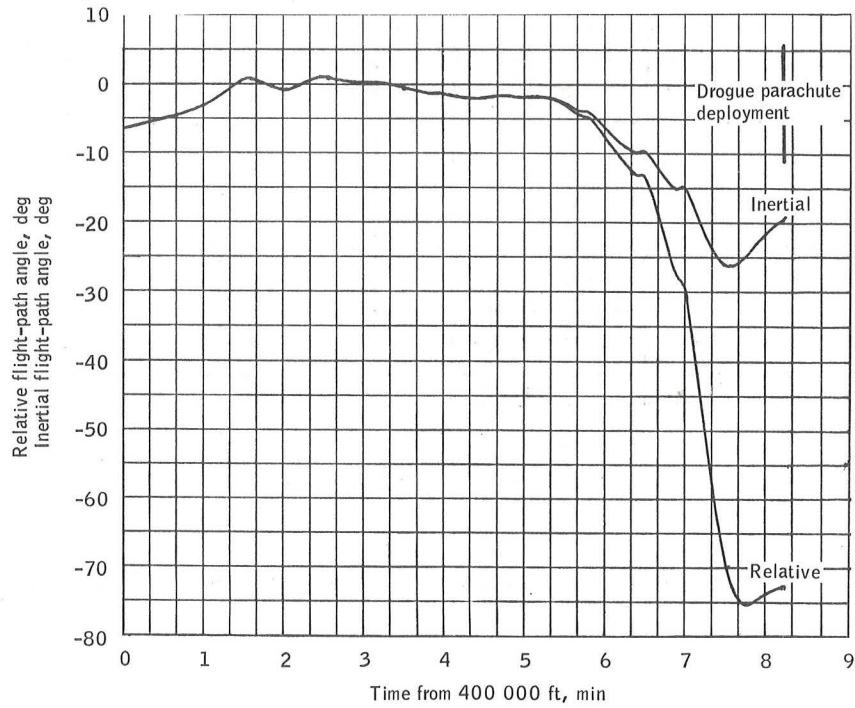


Figure 4.17-5.- Total aerodynamic heating rate and heat load time histories.



(a) Relative and inertial velocity.



(b) Relative and inertial flight-path angle.

Figure 4.17-6.- Time histories of velocity and flight-path angle for entry.

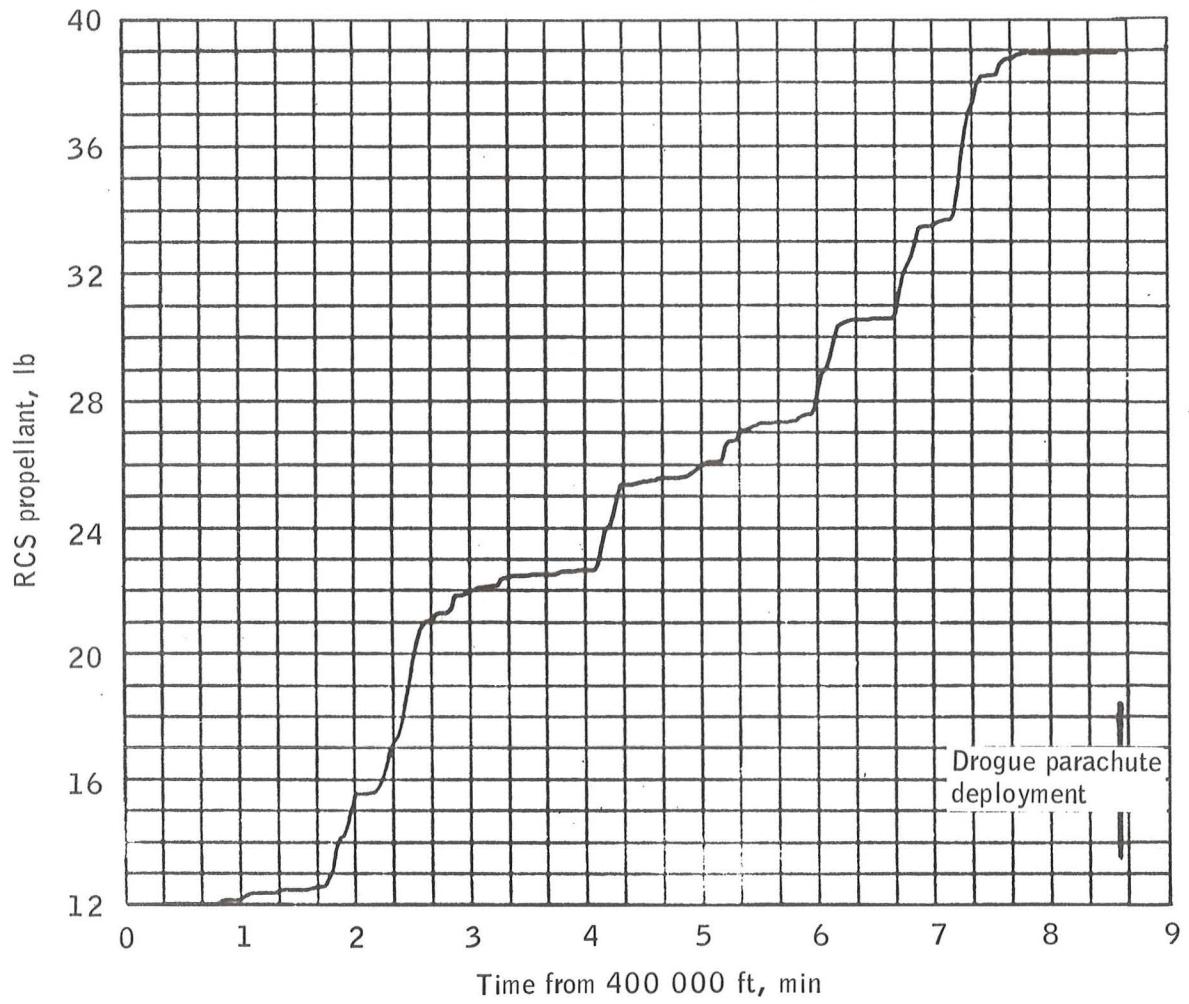


Figure 4.17-7.- Total propellant consumed from CSM separation.

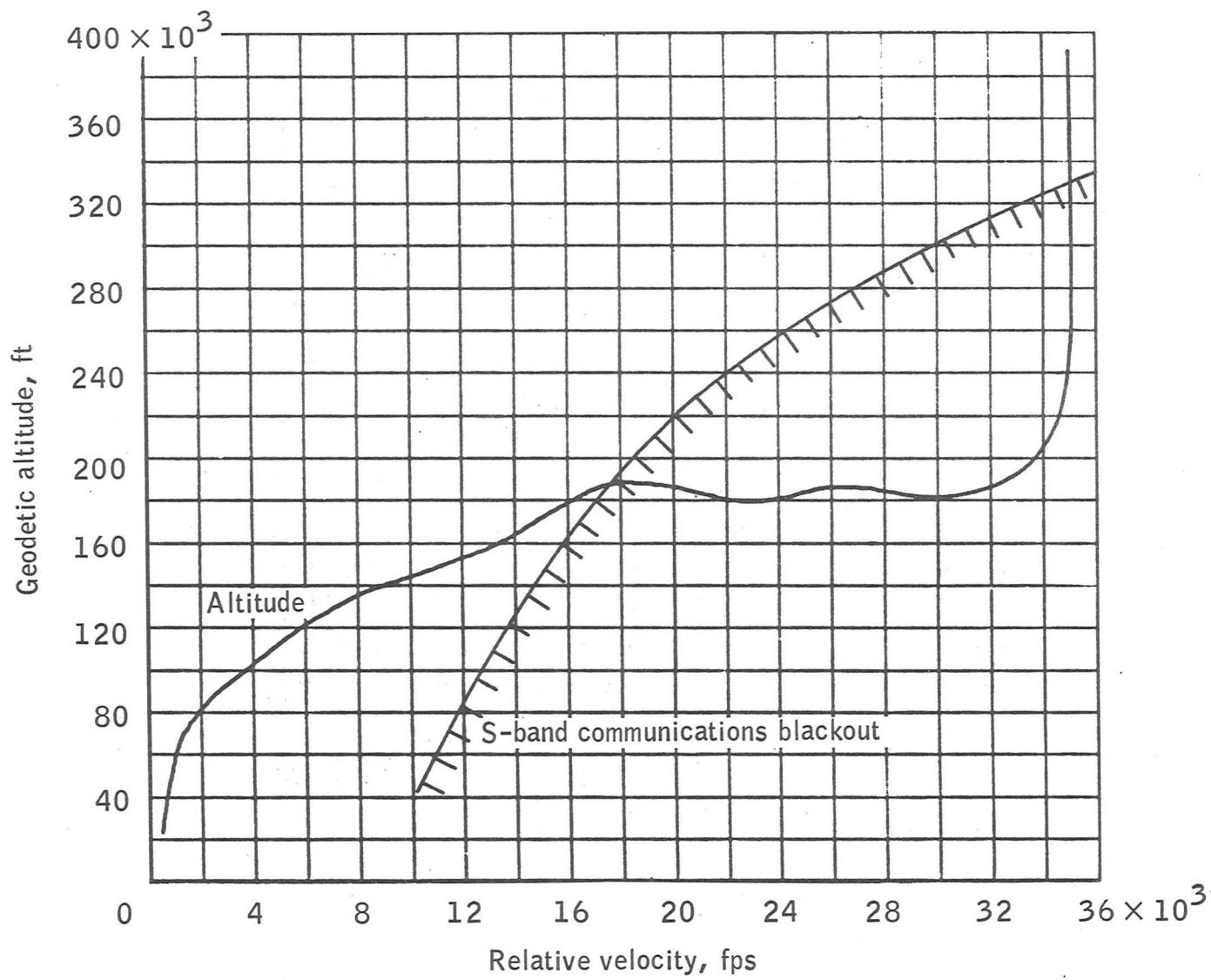


Figure 4.17-8.- Communications blackout.

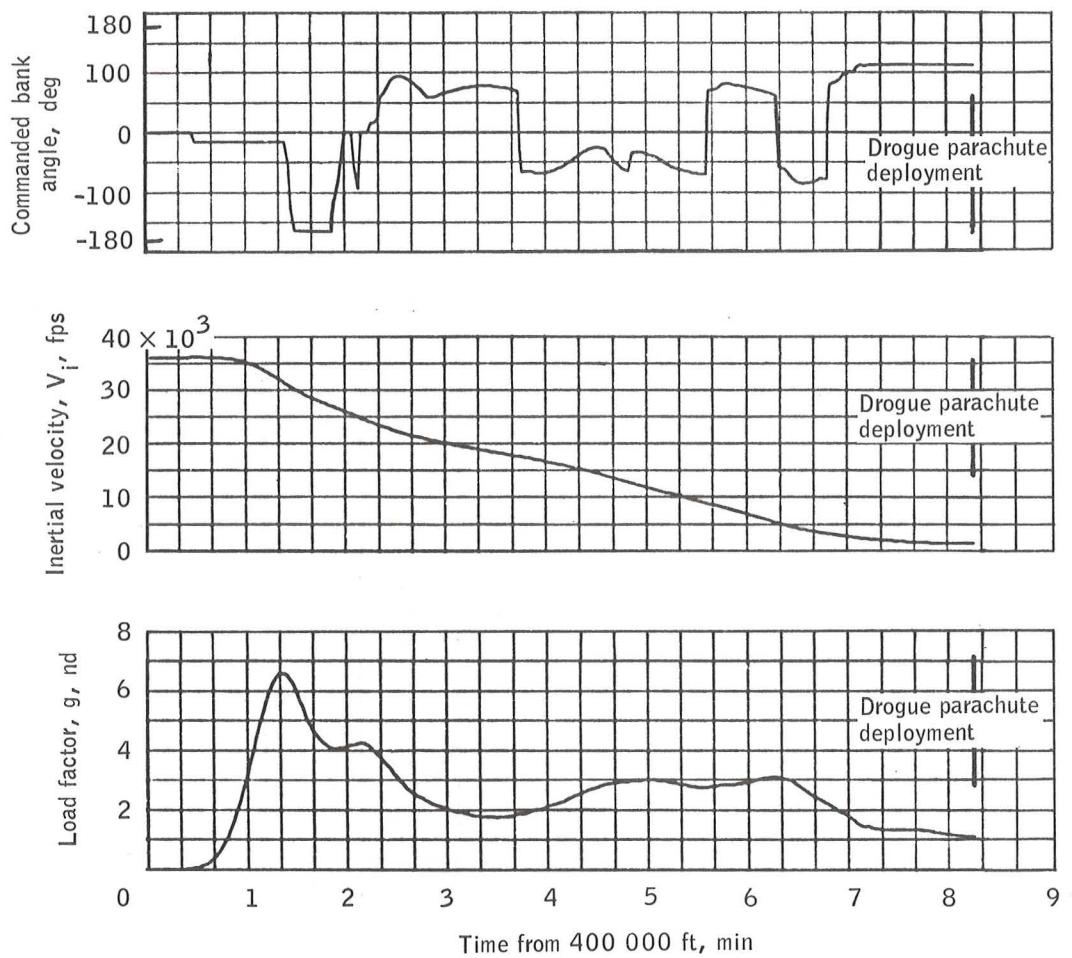


Figure 4.17-9.- Primary DSKY display VERB 06 NOUN 74.

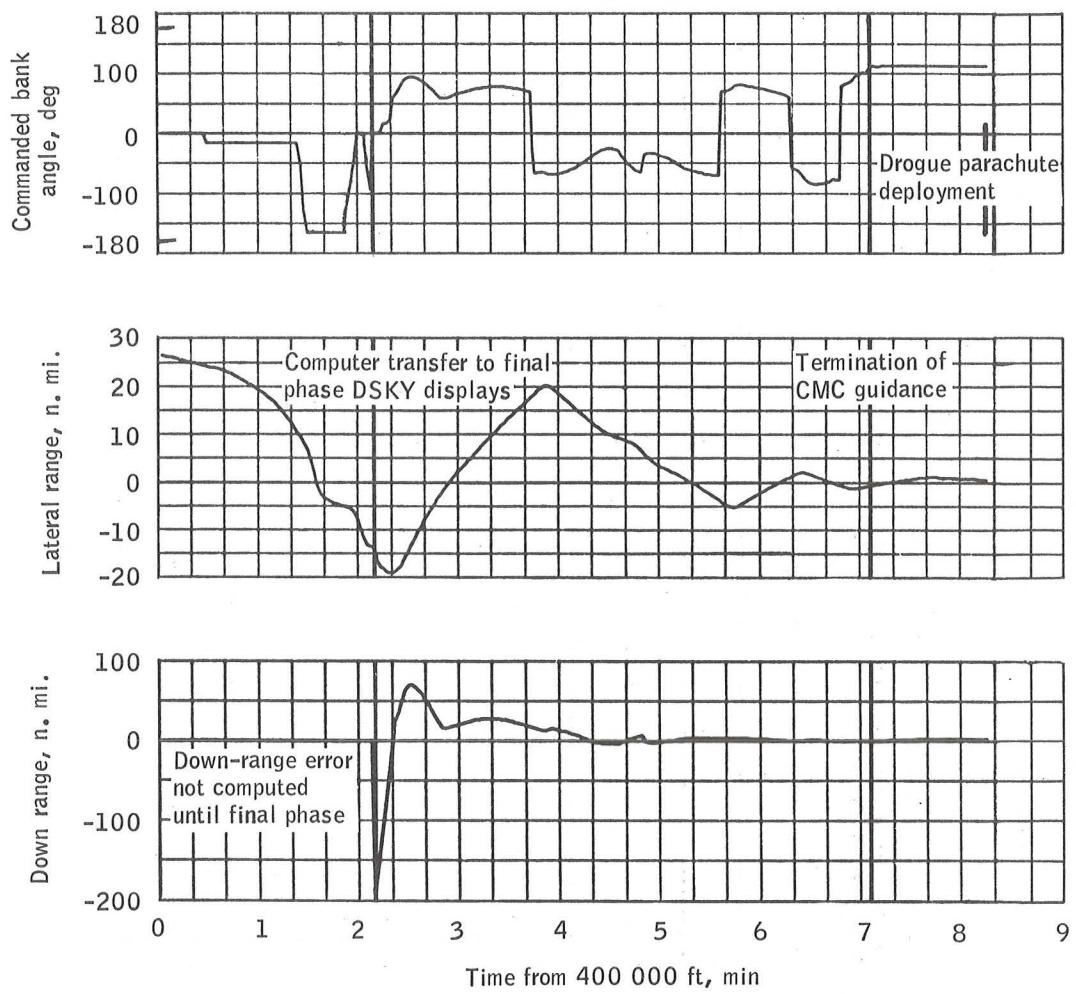


Figure 4.17-10.- DSKY displays (final phase) VERB 06 NOUN 66.

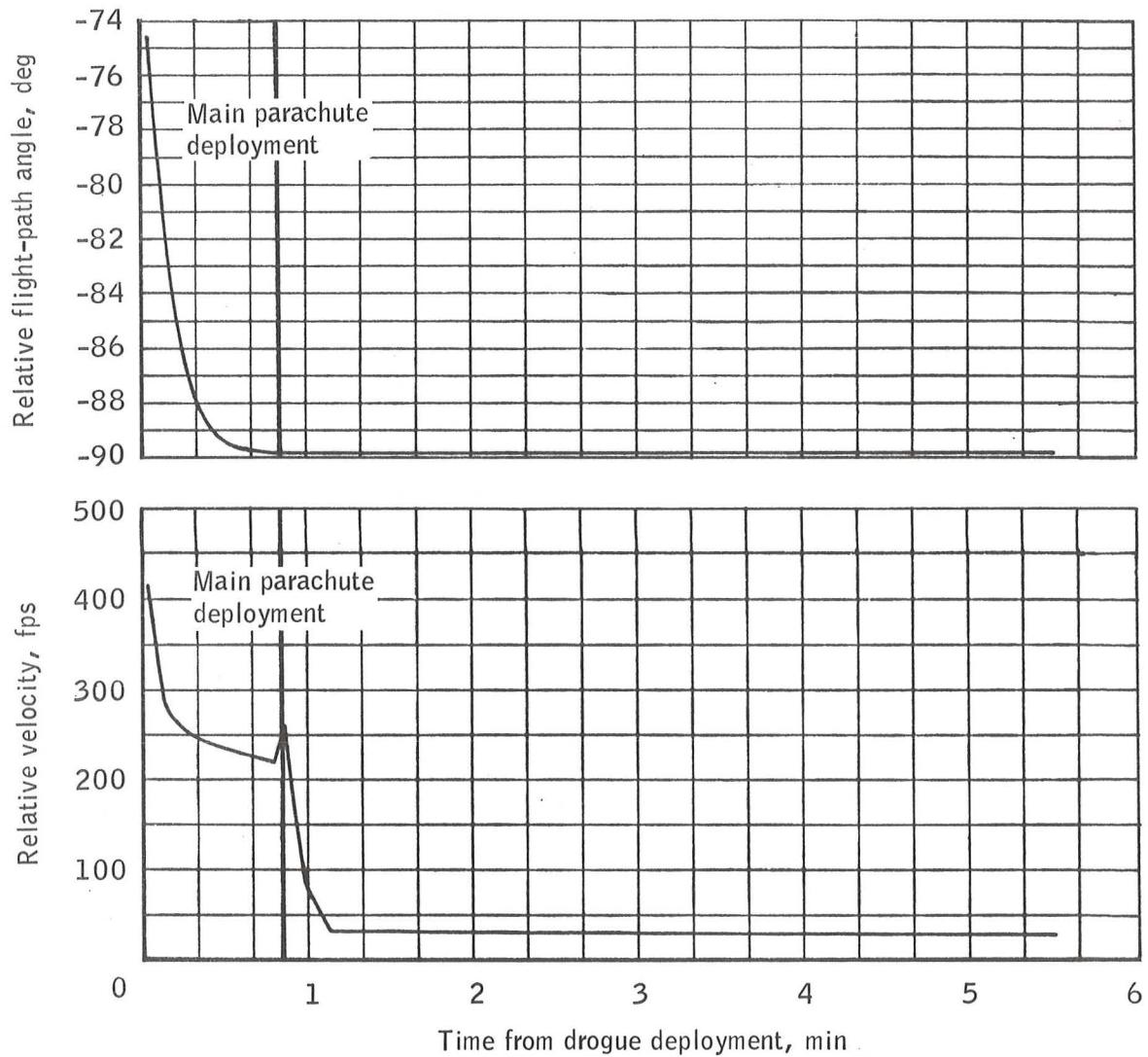


Figure 4.17-11.- Relative velocity and relative flight-path angle time histories from drogue parachute deployment.

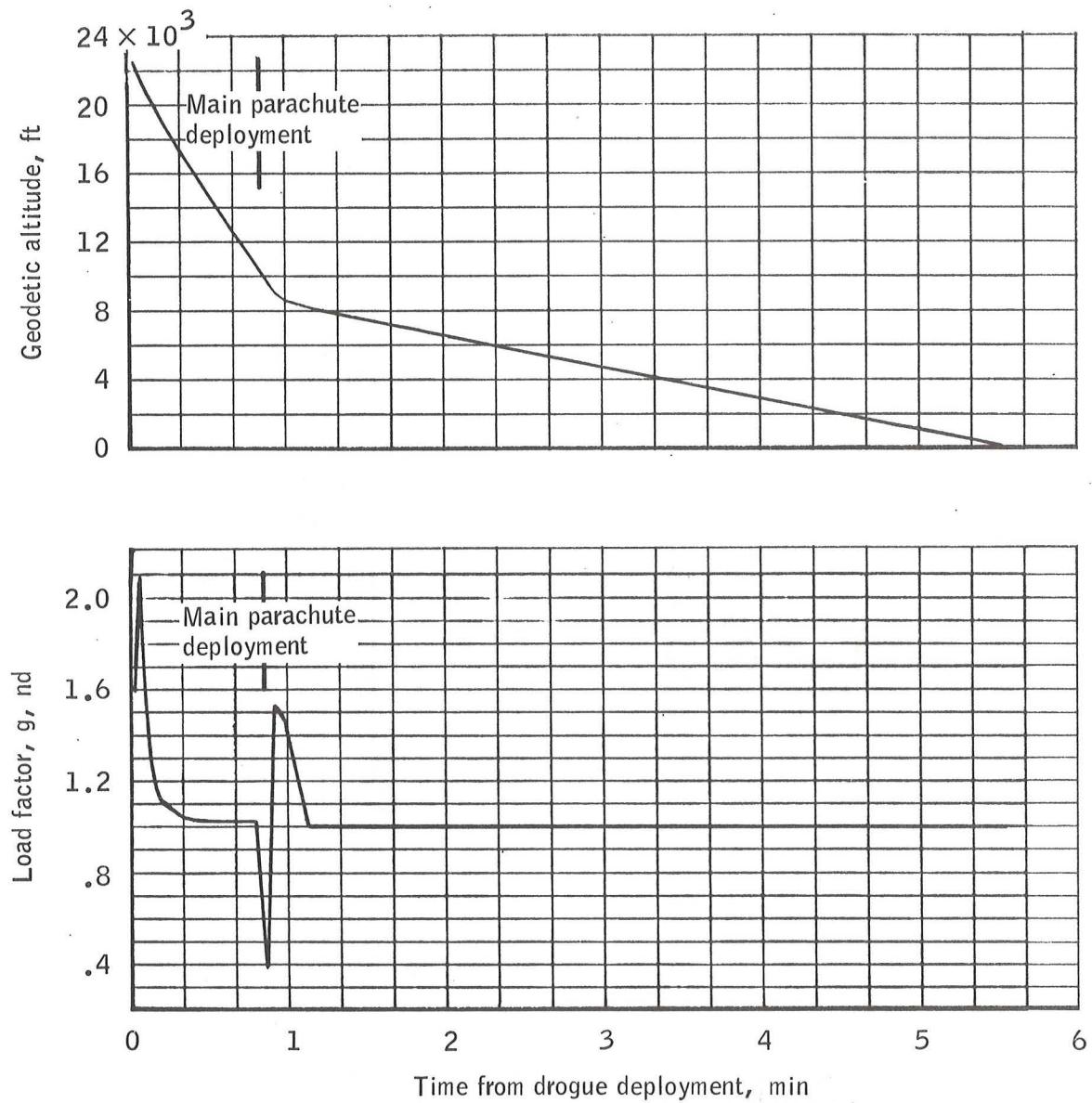
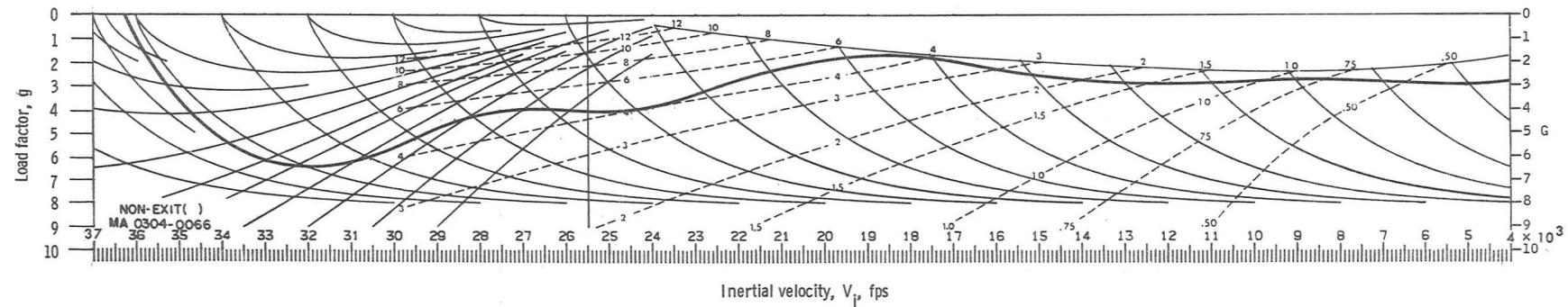
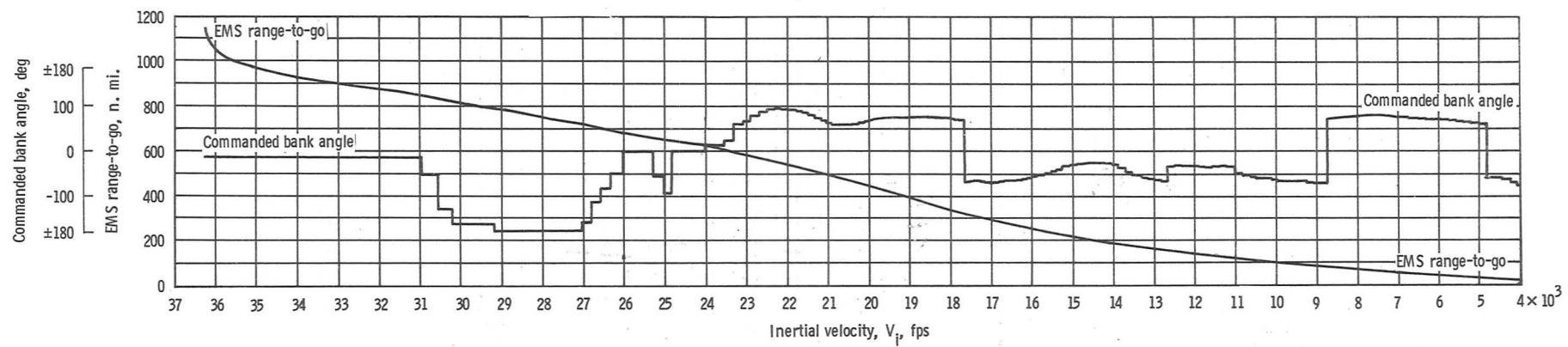


Figure 4.17-12.- Geodetic altitude and load factor time histories from drogue parachute deployment.



(a) Load factor versus inertial velocity.



(b) CMC commanded bank angle and EMS range-to-go versus inertial velocity.

Figure 4.17-13.- EMS parameters.

**5.0 INPUT DATA**

## 5.0 INPUT DATA

The following input data or reference sources were used to compute the Apollo 14 (H-3) mission operational trajectory.

Launch pad REFSMMAT . . . . .	Table 5-I
PTC REFSMMAT . . . . .	Table 5-II
Landing site REFSMMAT . . . . .	Table 5-III
Preferred (LOPC) REFSMMAT . . . . .	Table 5-IV
LM ascent REFSMMAT . . . . .	Table 5-V
Entry REFSMMAT . . . . .	Table 5-VI
Spacecraft weight summary . . . . .	Table 5-VII
Sequential mass properties . . . . .	Reference 6
Consumables weight change summary . . . . .	Reference 6
Assumed mission-independent expendables . . . . .	Table 5-VIII
Service module propulsion performance . . . . .	Table 5-IX
Command module RCS performance . . . . .	Reference 7
Lunar module propulsion performance . . . . .	Table 5-X
Command module aerodynamic parameters . . . . .	Table 5-XI
Lunar landing site positions . . . . .	Table 5-XII
IMU alignment criteria . . . . .	Table 5-XIII
Terminal rendezvous braking schedule . . . . .	Table 5-XIV
MSFN station characteristics . . . . .	Tables 5-XV and 5-XVI
Launch trajectory . . . . .	Reference 2
Mission requirements . . . . .	Reference 8
Aerodynamic heating data . . . . .	References 9 and 10
Atmosphere model . . . . .	Reference 11
Standards and constants . . . . .	Reference 12
Guidance equations and logic . . . . .	References 13 and 14

TABLE 5-I.- LAUNCH PAD REFSMMAT

## (a) Applicable maneuver

## 1. Launch

## (b) Definition of REFSMMAT

$$\begin{bmatrix} X \\ Y \\ Z \end{bmatrix}_{\text{SPACECRAFT}} * \begin{bmatrix} X_{IX} & X_{IY} & X_{IZ} \\ Y_{IX} & Y_{IY} & Y_{IZ} \\ Z_{IX} & Z_{IY} & Z_{IZ} \end{bmatrix} \begin{bmatrix} X \\ Y \\ Z \end{bmatrix}_{\text{ECI(MNBY)}}$$

PLATFROM  
STABLE  
MEMBERS

## (c) REFSMMAT components

$X_{IX} = -0.061007433$	$X_{IY} = 0.96056227$	$X_{IZ} = 0.27128992$
$Y_{IX} = 0.48156036$	$Y_{IY} = 0.26639687$	$Y_{IZ} = -0.83494452$
$Z_{IX} = -0.87428698$	$Z_{IY} = 0.079704651$	$Z_{IZ} = -0.47882091$

TABLE 5-II.- PTC REFSMMAT

## (a) Applicable maneuvers

1. MCC-1
2. Non-free-return maneuver (MCC-2)
3. Passivs thermal control
4. MCC-3
5. MCC-4

## (b) Definition of REFSMMAT

$$\begin{bmatrix} X \\ Y \\ Z \end{bmatrix}_{\text{SPACECRAFT}} = \begin{bmatrix} X_{IX} & X_{IY} & X_{IZ} \\ Y_{IX} & Y_{IY} & Y_{IZ} \\ Z_{IX} & Z_{IY} & Z_{IZ} \end{bmatrix} \begin{bmatrix} X \\ Y \\ Z \end{bmatrix}_{\text{ECI(MNBY)}}$$

PLATFORM  
STABLE  
MEMBERS

## (c) REFSMMAT components

$$X_{IX} = -0.93969262 \quad X_{IY} = -0.31378923 \quad X_{IZ} = -0.13606653$$

$$Y_{IX} = -0.34202014 \quad Y_{IY} = 0.86212879 \quad Y_{IZ} = 0.37383970$$

$$Z_{IX} = 0.0 \quad Z_{IY} = 0.39783191 \quad Z_{IZ} = -0.91745832$$

TABLE 5-III.- LANDING SITE REFSMMAT

## (a) Applicable maneuvers

1. LOI
2. CSM separation
3. CSM circularization
4. LM powered descent

## (b) Definition of REFSMMAT

$$\begin{bmatrix} X \\ Y \\ Z \end{bmatrix}_{\text{SPACECRAFT}} = \begin{bmatrix} X_{IX} & X_{IY} & X_{IZ} \\ Y_{IX} & Y_{IY} & Y_{IZ} \\ Z_{IX} & Z_{IY} & Z_{IZ} \end{bmatrix} \begin{bmatrix} X \\ Y \\ Z \end{bmatrix}_{\substack{\text{ECI(MNBY)} \text{ or} \\ \text{MCI(MNBY)}}}$$

PLATFORM  
STABLE  
MEMBERS

## (c) REFSMMAT components

$X_{IX} = -0.55709886$	$X_{IY} = -0.74626008$	$X_{IZ} = -0.36433055$
$Y_{IX} = 0.17736372$	$Y_{IY} = -0.53551514$	$Y_{IZ} = 0.82569101$
$Z_{IX} = -0.81128477$	$Z_{IY} = 0.39537251$	$Z_{IZ} = 0.43069435$

TABLE 5-IV.- PREFERRED (LOPC) REFSMMAT

## (a) Applicable maneuver

## 1. LOPC

## (b) Definition of REFSMMAT

$$\begin{bmatrix} X \\ Y \\ Z \end{bmatrix}_{\text{SPACECRAFT PLATFORM STABLE MEMBERS}} = \begin{bmatrix} X_{IX} & X_{IY} & X_{IZ} \\ Y_{IX} & Y_{IY} & Y_{IZ} \\ Z_{IX} & Z_{IY} & Z_{IZ} \end{bmatrix} \begin{bmatrix} X \\ Y \\ Z \end{bmatrix}_{\substack{\text{ECI(MNBY)} \text{ or} \\ \text{MCI(MNBY)}}}$$

## (c) REFSMMAT components

$X_{IX} = 0.13838206$	$X_{IY} = -0.48437922$	$X_{IZ} = 0.86384442$
$Y_{IX} = -0.15380724$	$Y_{IY} = -0.87216444$	$Y_{IZ} = -0.46440558$
$Z_{IX} = 0.97836278$	$Z_{IY} = -0.068600122$	$Z_{IZ} = -0.19519294$

TABLE 5-V.- LM ASCENT REFSMMAT

## (a) Applicable maneuvers

1. LM ascent
2. CSI
3. CDH
4. TPI
5. CSM separation following LM jettison
6. LM deorbit
7. TEI

## (b) Definition of REFSMMAT

$$\begin{bmatrix} X \\ Y \\ Z \end{bmatrix}_{\text{SPACECRAFT}} = \begin{bmatrix} X_{IX} & X_{IY} & X_{IZ} \\ Y_{IX} & Y_{IY} & Y_{IZ} \\ Z_{IX} & Z_{IY} & Z_{IZ} \end{bmatrix} \begin{bmatrix} X \\ Y \\ Z \end{bmatrix}_{\substack{\text{ECI(MNBY)} \\ \text{or} \\ \text{MCI(MNBY)}}}$$

PLATFORM  
STABLE  
MEMBERS

## (c) REFSMMAT components

$X_{IX} = -0.26712678$	$X_{IY} = -0.86784820$	$X_{IZ} = -0.41890668$
$Y_{IX} = 0.19308503$	$Y_{IY} = -0.47408943$	$Y_{IZ} = 0.85904448$
$Z_{IX} = -0.94411942$	$Z_{IY} = 0.14858917$	$Z_{IZ} = 0.29421047$

TABLE 5-VI.- ENTRY REFSMMAT

## (a) Applicable maneuver

## 1. Entry

## (b) Definition of REFSMMAT

$$\begin{bmatrix} X \\ Y \\ Z \end{bmatrix}_{\text{SPACECRAFT}} = \begin{bmatrix} X_{IX} & X_{IY} & X_{IZ} \\ Y_{IX} & Y_{IY} & Y_{IZ} \\ Z_{IX} & Z_{IY} & Z_{IZ} \end{bmatrix} \begin{bmatrix} X \\ Y \\ Z \end{bmatrix}_{\text{ECI(MNBY)}}$$

PLATFORM  
STABLE  
MEMBERS

## (c) REFSMMAT components

$$X_{IX} = 0.88429904 \quad X_{IY} = -0.38990236 \quad X_{IZ} = 0.25688784$$

$$Y_{IX} = 0.43228640 \quad Y_{IY} = 0.47571760 \quad Y_{IZ} = -0.76604258$$

$$Z_{IX} = 0.17647575 \quad Z_{IY} = 0.78845983 \quad Z_{IZ} = 0.58922610$$

TABLE 5-VII.- SPACECRAFT WEIGHT SUMMARY

Total CSM inert, lb . . . . .	23	230.9
CM inert, lb . . . . .	12	601.0
SM inert, lb . . . . .	10	531.9
SLA ring, lb . . . . .		98.0
Total SPS propellant tanked, lb . . . . .	40	593.7
SPS deliverable, lb . . . . .	40	354.6
SPS trapped, lb . . . . .		239.1
Total LM at earth launch, lb . . . . .	33	583.5
LM descent stage, lb . . . . .	4	775.9
LM DPS tanked, lb . . . . .	18	324.4
LM ascent stage + RCS propellant, lb . . . . .	5	271.0
LM APS tanked, lb . . . . .	5	212.2
Spacecraft at earth launch (including LES), lb . . . . .	110	386.9
Spacecraft before T&D . . . . .	101	380.3

TABLE 5-VIII.- ASSUMED MISSION-INDEPENDENT EXPENDABLES

## Mission-independent SPS budget

Translunar MCC (0 fps), lb . . . . .	0
Transearth MCC (0 fps), lb . . . . .	0

## SPS propellant allowances

Unbalance meter, lb . . . . .	100
Mean outage, lb . . . . .	59
Dispersions, lb . . . . .	426

TABLE 5-IX.- SERVICE MODULE PROPULSION PERFORMANCE SUMMARY

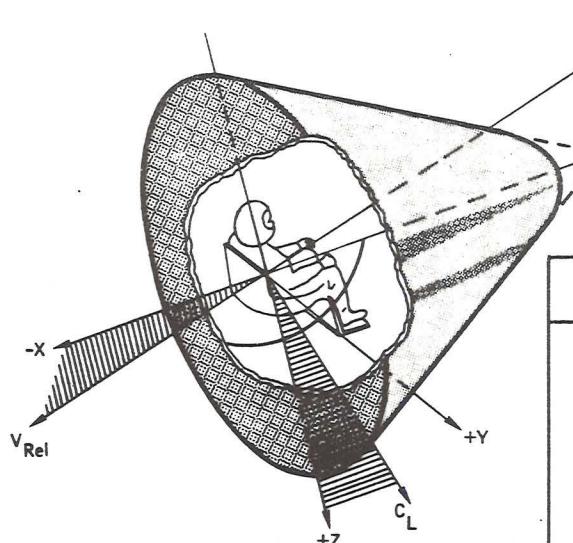
Propulsion system	$I_{sp}$ , sec	Thrust, lb	Flow rate, lb/sec
RCS <sup>a</sup>	263.58	181.87	0.69
SPS <sup>b</sup>	313.173	19 946	63.69
SPS <sup>c</sup>	313.48	21 032	67.09
SPS <sup>d</sup>	313.563	21 302	67.934

<sup>a</sup>Two-quad.<sup>b</sup>Hybrid maneuver.<sup>c</sup>Before crossover during LOI burn (258-sec burn time).<sup>d</sup>After crossover during LOI burn (258-sec burn time).

TABLE 5-X.- LUNAR MODULE PROPULSION PERFORMANCE SUMMARY

Descent				Ascent		
DPS		RCS		APS + RCS		
Thrust lb,	I <sub>sp</sub> , sec	Thrust per engine, lb	I <sub>sp</sub> , sec	Burn time, sec	Effective thrust, lb	Effective I <sub>sp</sub> , sec
1375	295.3	89.6	281.0	0	3573.5	309.6
2842	296.0			40	3561.2	309.9
3586	297.9			80	3547.9	310.1
5915	302.8			120	3534.5	310.2
6108	303.0			160	3521.3	310.2
9788	304.6			200	3508.6	310.2
9906	301.0			240	3496.5	310.2
				280	3484.9	310.2
				320	3478.5	310.1
				360	3479.5	310.1
				400	3485.3	310.1
				430.7	3498.3	310.1

TABLE 5-XI.- COMMAND MODULE AERODYNAMIC COEFFICIENTS



AERODYNAMIC COEFFICIENTS AT TRIM ANGLE OF ATTACK  
AS A FUNCTION OF MACH NUMBER

MACH	$\alpha$ , deg	$C_L$	$C_D$	L/D
0.4	167.021	0.247270	0.852535	0.290041
0.7	164.419	0.262858	0.985644	0.266687
0.9	161.786	0.319479	1.065963	0.299709
1.1	155.054	0.490841	1.171996	0.418808
1.2	155.296	0.475923	1.158027	0.410978
1.35	154.156	0.560154	1.280589	0.437419
1.65	153.441	0.547115	1.268507	0.431306
2.0	153.443	0.529746	1.278005	0.414510
2.4	153.953	0.504379	1.247470	0.404321
3.0	154.464	0.475752	1.223769	0.388760
4.0	156.378	0.438660	1.219548	0.359691
10.0	157.034	0.425825	1.229524	0.346333
29.5	160.258	0.385075	1.292907	0.297837

Center of gravity location in body coordinates

$$X_{cg} = 1039.4 \text{ in.}$$

$$Y_{cg} = -30 \text{ in.}$$

$$Z_{cg} = 6.0 \text{ in.}$$

TABLE 5-XII.- LUNAR LANDING AND PHOTOGRAPHY SITE POSITIONS

Parameter	Landing site	Photography site
Launch date . . . . .	Jan. 31, 1971	Jan. 31, 1971
Lunar site . . . . .	Fra Mauro	Descartes
Latitude, deg . . . . .	-3.6734	-8.85
Longitude, deg . . . . .	-17.4646	15.567
Altitude, n. mi. . . . .	-0.76	-0.15

TABLE 5-XIII.- IMU ALIGNMENT CRITERIA AND REFSMMAT DEFINITIONS

Applicable maneuvers	REFSMMAT	Alignment criteria (IMU) axis	Remarks
Launch TLI T&D	Launch pad (CSM only)	X-axis along launch azimuth at pad Z-axis along negative radius vector Y-axis completes right-hand system	At earth launch, the FDAO will display roll = 180° pitch = 90° yaw = 0°
MCC-1 MCC-2 (hybrid transfer) MCC-3 MCC-4	PTC (passive thermal control) (CSM only)	X-axis in the ecliptic plane and perpendicular to the earth-moon line projection in the ecliptic plane at the average time of trans-earth injection for the monthly launch window and azimuth range.  Z-axis perpendicular to the ecliptic and directed south.  Y-axis completes the right-hand system.	At the beginning of the Barbeque mode, the spacecraft will maneuver to an FDAO display of roll = 0° pitch = 90° yaw = 0°
LOI DOI PDI Landing	Landing site (CSM and LM)	X-axis along positive lunar radius vector at landing site at predicted landing time.  Z-axis in direction of flight parallel to the CSM orbital plane and perpendicular to X.  Y-axis completes the right-handed system.	At nominal touchdown, the LM FDAO will display roll = 0° pitch = 0° yaw = 0°
CSM plane changes	Preferred (CSM only)	X-axis aligned with the spacecraft X-body axis at the vehicle attitude for ignition with the thrust directed through the center of gravity.  Z-axis in the plane formed by the X-axis and position vector and directed down towards the moon.  Y-axis completes the right-hand system.	At burn ignition, the FDAO will display roll = 0° pitch = 0° yaw = 0°
LM lift-off Rendezvous TEI	Ascent (CSM and LM)	X-axis along positive lunar radius vector at landing site at predicted lift-off time.  Z-axis down range parallel to the CSM orbital plane and perpendicular to X.	At nominal lift-off time, the LM FDAO will display roll = 0° pitch = 0° yaw = 0°
MCC-5 MCC-6 MCC-7	PTC (CSM only)	Same as above PTC alignment	Same as above PTC remarks.
Entry	Entry (CSM only)	X-axis is in local horizontal plane in the direction of flight at entry interface (400 000 ft).  Z-axis is down along negative radius at entry interface.  Y-axis completes right-hand system.	At entry interface with wings level local horizontal, heat shield forward, in place, lift up, heads down, the FDAO will display roll = 0° pitch = 180° yaw = 0°

TABLE 5-XIV.- LM BRAKING SCHEDULE  
DURING TERMINAL RENDEZVOUS PHASE

Gate	Range, n. mi.	Range rate at cutoff, fps
First	0.4	-19.6
Second	0.2	-9.8
Third	0.08	-4.9
Fourth	0.01	-0.2
Fifth	0.0	0.0

TABLE 5-XV. - MSFN STATION CHARACTERISTICS

Geodetic latitude, LATR, deg  
 Longitude, LONR, deg  
 Altitude, ALTR, ft  
 Range capability, SRANGE, n. mi.

Keyhole, FTINDC: 0 = none  
 1 = north-south  
 2 = east-west

RADAR =MAD-23 X-Y USB-85, LATR = 40.454991, LONR = 355.831802
ALTR = 2552.489, SRANGE = 850000.0, FTINDC = 2.0
RADAR =HSK-25 X-Y USB-85, LATR = -35.583494, LONR = 148.978082
ALTR = 3756.555, SRANGE = 850000.0, FTINDC = 2.0
RADAR =GDS-28 X-Y USB-85, LATR = 35.341594, LONR = 243.126654
ALTR = 2976.045, SRANGE = 850000.0, FTINDC = 2.0
RADAR =PKS-96 H-D USB210, LATR = -32.998938, LONR = 148.263272
ALTR = 1227.032, SRANGE = 2100000.0, FTINDC = 0.0
RADAR =STN-00 A-E USB150, LATR = 37.409722, LONR = -122.177917
ALTR = 525.000, SRANGE = 1500000.0, FTINDC = 0.0
RADAR =MIL-71 X-Y USB-30, LATR = 28.508272, LONR = 279.306438
ALTR = -59.055, SRANGE = 300000.0, FTINDC = 1.0
RADAR =GBM-41 X-Y USB-30, LATR = 26.632857, LONR = 281.762191
ALTR = 59.055, SRANGE = 300000.0, FTINDC = 1.0
RADAR =BDA-02 X-Y USB-30, LATR = 32.351250, LONR = 295.341755
ALTR = -6.562, SRANGE = 300000.0, FTINDC = 1.0
RADAR =ANG-91 X-Y USB-30, LATR = 17.016863, LONR = 298.247038
ALTR = 111.548, SRANGE = 300000.0, FTINDC = 1.0
RADAR =CYI-04 X-Y USB-30, LATR = 27.764536, LONR = 344.366402
ALTR = 567.584, SRANGE = 300000.0, FTINDC = 1.0
RADAR =ACN-75 X-Y USB-30, LATR = -7.954794, LONR = 345.672938
ALTR = 1712.595, SRANGE = 300000.0, FTINDC = 1.0
RADAR =CRO-08 X-Y USB-30, LATR = -24.906577, LONR = 113.726049
ALTR = 49.213, SRANGE = 300000.0, FTINDC = 1.0
RADAR =GWM-24 X-Y USB-30, LATR = 13.310575, LONR = 144.736594
ALTR = 469.159, SRANGE = 300000.0, FTINDC = 1.0
RADAR =HAW-12 X-Y USB-30, LATR = 22.126307, LONR = 200.335058
ALTR = 3723.747, SRANGE = 300000.0, FTINDC = 1.0
RADAR =GYM-14 X-Y USB-30, LATR = 27.963205, LONR = 249.279004
ALTR = -72.178, SRANGE = 300000.0, FTINDC = 1.0
RADAR =TEX-16 X-Y USB-30, LATR = 27.653750, LONR = 262.621385
ALTR = -65.617, SRANGE = 300000.0, FTINDC = 1.0
RADAR =GTI-51 A-E TPQ-18, LATR = 21.462888, LONR = 288.867741
ALTR = 91.863, SRANGE = 23400.0, FTINDC = 0.0
RADAR =PRE-76 A-E MPS-25, LATR = -25.943733, LONR = 28.358286
ALTR = 5334.636, SRANGE = 23400.0, FTINDC = 0.0
RADAR =TAN-22 A-E FPS-16, LATR = -19.003019, LONR = 47.314444
ALTR = 4329.389, SRANGE = 23400.0, FTINDC = 0.0
RADAR =SHIP 1 INSEPTION, LATR = 28.000000, LONR = -49.000000
ALTR = 0.000, SRANGE = 23400.0, FTINDC = 0.0

TABLE 5-XVI.- MSFN STATION IDENTIFICATION

Ship 1	Insertion ship C-band
MIL-71	Merritt Island S-band
BDA-02	Bermuda S-band
ANG-91	Antigua S-band
CYI-04	Grand Canary S-band
ACN-75	Ascension S-band
CRO-08	Carnarvon S-band
GWM-24	Guam S-band
HAW-12	Hawaii S-band
GYM-14	Guaymas S-band
MAD-23	Madrid deep space
HSK-25	Canberra deep space
GDS-28 <sup>a</sup>	Goldstone deep space
Parks-96	Parks deep space
Tex-16	Texas S-band
STN-00	Stanford S-band
GBM-41	Grand Bahama S-band
GTI-51	Grand Turk C-band
PRE-76	Pretoria C-band
TAN-22	Tananarive C-band

<sup>a</sup>Data also applicable to MAR-95 210 ft.



## 6.0 REFERENCES

1. Lunar Mission Analysis Branch; Orbital Mission Analysis Branch; and Landing Analysis Branch: Spacecraft Operational Trajectory for Apollo 14 (Mission H-3) Launched January 31, 1971, Volume II - Trajectory Parameters. MSC IN 70-FM-128, Sept. 4, 1970.
2. Transmittal of AS-509 data tapes, MSFC memo S&E-AERO-MFT-109-70, Aug. 21, 1970.
3. The Apollo 14 (H-3) Mission Operational Trajectory Simulator Data Package. MSC memo to be published.
4. Apollo Mission Techniques, Mission F and G, Transearth Injection, Midcourse Corrections, and Entry. MSC IN SPS-040, Feb. 24, 1969, revised June 6, 1969.
5. RTCC Requirements: Revised S-band Communication Blackout Data for Apollo F and Subsequent Missions. MSC memo 69-FM23-43, Feb. 26, 1969.
6. CSM/LM Spacecraft Operational Data Book, Volume III - Mass Properties. SNS-8-D-027, Rev 2 through amendment 81, Aug. 20, 1969.
7. CSM/LM Spacecraft Operational Data Book, Volume I - CSM Data Book. SNA-8-D-027 (I) Rev 3, through amendment 43, Apr. 15, 1970.
8. Mission Requirements, SA-509/CSM-110/LM-8, H-3 Type Mission. SPD9-R-056, June, 9, 1970.
9. Ried, Robert C., Jr.: Apollo Command Module Entry Air Radiation Heating Rate. MSC memo ES5/1-2/67, Jan. 9, 1968.
10. Heating Rate Factors for Reentry Studies. MSC memo ES5/9-11/173M, Sept. 14, 1967.
11. United States Weather Bureau: U. S. Standard Atmosphere, 1962, Dec. 1962.
12. Apollo Navigation Working Group: Apollo Missions and Navigation Systems Characteristics. Technical report no. AN-1.3, Dec. 15, 1967.
13. MIT: Guidance Systems Operations Plan for Manned CM Earth Orbital and Lunar Missions Using Program COLOSSUS, Section 5, Guidance Equations, Revision 2. R577, Dec. 1968.
14. Tolin, J. W., Jr.: RTCC Requirements for Missions H and Subsequent Missions: Reentry Phase. MSC IN 69-FM-179, July 3, 1969.