D. EYLES

PCR/PCN MIT # 477

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	DO	TR #	

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	Luminary Pr	eliminary F	Program Chang	ge Approva		
	MIT Approved PCN		ADDI		COPIES EYLES	70 :
	NASA Approved PCR			13. k M. C	(RIEGSMA. IOHNSTON	N
A. Codin	g No+ A	pplica	b)e	A. K	LUMPP.	
	Begin coding immed	diately				
			ACTION:			
\bigcirc	Do not code until					
B. GSOP	Preparation					
	Prepare GSOP revision	sions for M	DRB	- F. C	17728	
			ACTION:			
	•		-			
	Technical Committe	ee Meeting	Was Not Requ	ired		
M	Technical Committe	ee Meeting(s) held on _	LM!	Desient	Design
	. Attendees:	,	Review	hold Hende	about by	3 mont
			Dan L (raig	s. Whe	n bor	rounge (b

APOLLO SPACECRAFT SOFTWARE CONFIGURATION CONTROL BOARD

<u> </u>		PROGRAM CHANGE RE	QUEST		No. 497
100		MARILI OIA	NET TWO SHAPES		(Completed by FS
1.0	ORIGINATOR DATE:	or:G.W. Cherry 15 July 1968	1.2	ORGANIZATION: WAPPROVAL:	IIT/IL WE DATE:/5/
1.3	EFFECTIVITY: LUMINARY		1.4	TITLE OF CHANGE: I Throttle-up time	Do not delay P6
1.5	REASON(S) FOR CHANGE: During the approximately ten times more of the engine ignites ten seconds leading to the engine ignites ten seconds leading ten seconds le	critical than the	rakin engin	g phase, the thro e ignition time.	ttle-up time is For example, i
1.6	DESCRIPTION OF CHANGE: See da	ata amplification	shee	t.	national term to the second
2.0	SOFTWARE CONTROL BOARD OR FLIGHT DECISION FOR VISIBILITY IMPACT ES		2.1	APPROVED	DISAPPROVED
Wilder William	REMARKS:		2.3	SOFTWARE CONTROL F SOFTWARE BRANCH SI	
					DATE:
0	MIT VISIBILITY IMPACT EVALUATION:		3.1	SCHEDULE IMPACT:	Zero
3.2	IMPACT OF PROVIDING DETAILED EVALUATION:			STORAGE IMPACT:	4 Words
3.4	REMARKS: Coordinated with Thomas Price and Bill Tindall of NASA.			MIT COORDINATOR:	
4.0	BOARD ACTION DET	LEMENT AND PROVIDE		PROVIDE DETAILED CHANGE EVALUATION	☐ DISAPPROVED
4.2	REMARKS:		4.3	BOARD SIGN OFF:	
5.0	MIT DETAILED PROGRAM CHANGE EVALUATION	5.1 MIT COORDIN	ATOR:	and the second second second second	e constituir de la cons
5.2	MIT EVALUATION: Action: Steve Copps, Forbes L Program has already k		iegsn		inge page 15 1. Heorge W. Ch
0	SOFTWARE CONTROL BOARD DECISION OF DETAILED PROGRAM CHANGE EVALUATION	E CONTRACTOR CONTRACTO	31.	ART OP CONTINUE PLEMENTATION	DISAPPROVED OR S
6.2	REMARKS:	te established had be some her with the second some second	6.3	SOFTWARE CONTROL	

DATE:

APOLLO SPACECRAFT SOFTWARE CONFIGURATION CONTROL BOARD -DATA AMPLIFICATION SHEET -

PAGE 2 OF 2

PROGRAM CHANGE REQUEST NO.

497

PREPARED BY: G.W. Cherry

DATE: _15 July 1968

ORGANIZATION:

MIT/II

CONTINUATION SECTION (REFER TO BLOCK NUMBER AND TITLE ON PROGRAM CHANGE REQUEST FORM)

1.5 Reason(s) for Change, cont'd.

...engine ignition time but the throttle-up time still occurs at the correct (the nominal) time, throttle control should be recovered and the high gate conditions should be achieved; however, if the throttle-up time slips along with the engine ignition time by ten seconds, the high gate targets may not be achieved.

1.6 Description of Change:

Let the program reference the throttle-up time relative to the nominal engine ignition time and not the actual ignition time (when the astronaut keys in PROCEED in response to FLV99N62). Thus, if the astronaut keys in PROCEED late, the engine is ignited late but the throttle-up time is held to the nominal. The effect is to decrease the 26 second trim time. But 26 seconds is more than enough time for trimming.

Remarks:

Incidentally, if the engine fails to ignite and the Delta V Monitor calls the engine fail routine, the astronaut should not recycle to "TIG-5" by keying in ENTER. This response by the astronaut would re-define the throttle-up time (slipping it by at least 13 seconds). The only purpose recycling would serve would be to supply some more automatic ullage. A preferred procedure would be to key in PROCEED to FLV97N63 and supply more ullage translation manually (if it is needed).