

GUIDANCE, NAVIGATION AND CONTROL

Submitted by: Pyr, on 14. H. Hamilton, Date: 4-5-72

M.H.HAMILTON, DIRECTOR, MISSION PROGRAM DEVEL.
APOLLO GUIDANCE AND NAVIGATION PROGRAM
γ
Approved Date: 4-5-72
R.A.LARSON, LUMINARY PROJECT MANAGER
APOLLO GUIDANCE AND NAVIGATION PROGRAM
Approved: R. H. Batter Date: 4/5-12-
• •
R.H. BATTIN, DIRECTOR, MISSION DEVELOPMENT
APOLLO GUIDANCE AND NAVIGATION PROGRAM
$I/I = I \cap I$
Approved: Naind & Hoaf Date 50/272
Date Date Date Date Date Date Date Date
APOLLO GUIDANCE AND NAVIGATION PROGRAM
APOLLO GUIDANCE AND NAVIGATION PROGRAM
Approved: Malal & Dagan Date: 69-72
R.R. RAGAN, DEPUTY DIRECTOR
II. II. IIAGAN, DEI OIT DIRECTOR

CHARLES STARK DRAPER LABORATORY

R-567

GUIDANCE SYSTEM OPERATIONS PLAN FOR MANNED LM EARTH ORBITAL AND LUNAR MISSIONS USING PROGRAM LUMINARY 1E

SECTION 7 ERASABLE MEMORY PROGRAMS

April 1972



CHARLES STARK DRAPER
LABORATORY

CAMBRIDGE, MASSACHUSETTS, 02139

ACKNOWLEDGEMENT

This report was prepared under DSR Project 55-23890, sponsored by the Manned Spacecraft Center of the National Aeronautics and Space Administration through Contract NAS 9-4065.

GUIDANCE SYSTEM OPERATIONS PLAN FOR MANNED LM EARTH ORBITAL AND LUNAR MISSIONS USING PROGRAM LUMINARY 1E

SECTION 7 ERASABLE MEMORY PROGRAMS

Signatures appearing on this page designate approval of this document by NASA/MSC.

Approved:	Date:	4/4/72
John R. Carman Section Chief, Guidance Program Section Manned Spacecraft Center, NASA		
Approved: John E. Williams, Jr.	Date:	4/4/22
Chief, Simulation and Flight Software Branch Manned Spacecraft Center, NASA		
Approved:	Date:	4/5/72_
James C. Stokes, Jr. Chief, Flight Support Division Manned Spacecraft Center, NASA		



TABLE OF CONTENTS

NO.	TITLE
	Introduction
EMP 99	Guided RCS Translational Maneuver
EMP 100A	Backup For Failed DSKY Key Using ENG GMBL Switch
EMP 100B	Backup For Failed DSKY Key Using MODE SEL Switch
EMP 101	P47 With DAP Driving GTS
EMP 102	Software Restart
EMP 103A	Descent With Failed CDUs
*EMP 104	P20 Operation With IMU Inoperative
EMP 106	Increment AOT Detent Position
*EMP 107	For Display Of Raw LRH and H On The DSKY
EMP 108	Zero A Runaway IMU CDU and Prevent Coarse Align

Note: The Erasable Memory Programs (EMP) contained in this section are individually paginated within each EMP.

^{*} These EMPs will be supplied at a later date as change pages.



INTRODUCTION

GENERAL

Section 7 of the Guidance System Operations Plan (GSOP) describes erasable-memory programs (EMPs) designed for the guidance computers used in the command (CMC) and lunar modules (LGC). CMC programs are designated COLOSSUS 3, and the associated EMPs are identified by a three-digit number beginning with "5." LGC programs are designated LUMINARY 1E, and the associated EMPs are identified, with one exception, by a three-digit number beginning with "1." The exception is EMP 99.

The EMPs vary in complexity from a simple flagbit setting to a long and intricate logical structure. They all, however, cause the computer to behave in a way not intended in the original design of the programs; they accomplish this off-nominal behavior by some alteration of erasable memory to interface with existing fixed-memory programs to effect a desired result.

<u>CAUTION</u>.—Great care must be taken when loading or performing an EMP. An erroneous digit loaded into NOUN 26, for example, could cause indeterminate operation upon program initiation.

NOTE 1.—The EMPs described in this section should not be run simultaneously except when explicitly specified.

NOTE 2.—Level 6 performance-evaluation testing has not been performed on EMPs.

The following format is used throughout this section:

NUMBER AND NAME OF EMP.

PURPOSE.

FUNCTIONAL DESCRIPTION—a brief description of the EMP and how it interfaces with fixed-memory programs (may include a functional-flow diagram).

ASSUMPTIONS—prerequisite conditions and configurations.

RESTRICTIONS AND LIMITATIONS—conditions and operations that would interfere with, or be affected by, the EMP.

PROCEDURES—instructions for performing the EMP.

RECOVERY/TERMINATION—procedures for terminating the EMP or recovering from a hardware or software restart.

ERASABLE MEMORY—listing of memory locations (octal) and the code (mnemonic and octal) comprised by the EMP.

UPLINK-P27 format for loading the EMP into erasable memory.

JOBS AND TASKS

A number of EMPs are initiated by VERB 30 ENTR (Request Executive) or VERB 31 ENTR (Request WAITLIST). When the EMP is programed as a JOB, the activation procedures specify VERB 30 ENTR; when the EMP is programed as a TASK, the procedures specify VERB 31 ENTR. The distinction is on the basis of how the program is dispatched. A JOB carries a priority; when the JOB's priority comes up on the executive queue, the JOB is activated. A TASK differs in that it is performed as a T3-clock interrupt. The AGC WAITLIST program sets the T3 clock to overflow at a specified time; when the overflow occurs, other program activity is interrupted, and the TASK is performed.

For VERB 30 use, the JOB's priority is specified in R1 of NOUN 26. R1 of NOUN 26 also contains in the low-order digit an indication of whether or not the JOB is to be assigned a VAC area: if the low-order digit is "1," a VAC area is reserved for the JOB; if it is "0," no VAC area is reserved.

For VERB 31 use, R1 of NOUN 26 must contain the time specified to elapse (in centiseconds) between the keying of ENTR (after VERB 31) and TASK execution.

EMPs activated by VERB 30 ENTR (i.e., JOB EMPs) require NOUN 26 to be loaded as follows:

R1 = xx00y₈
where $xx_8 = \text{JOB Priority}$ y = 1 designates a VAC JOB; y = 0 designates a NOVAC JOB.R2 = xxxxx₈
where

 $xxxxx_{0}$ is the JOB starting address

R3 = xxxxx₈
where

 ${\tt xxxxx}_8$ is the BBCON, containing the fixed, super, and erasable banks associated with the JOB

EMPs activated by VERB 31 ENTR (i.e., TASK EMPs) require NOUN 26 to be loaded as for a JOB EMP, except that R1 contain not a JOB priority, but a time delay as described above:

 $R1 = xxxxx_8$ cs delay

 $R2 = xxxxx_{0}$ starting address

 $R3 = xxxxx_8$ BBCON

The BBCON is packed as follows:

F-banks $00-27_8$ are addressed independently of S-bank contents; F-banks $30-37_8$ are addressed for S-bank values of 3_8 or less, and F-banks $40-43_8$ are addressed for an S-bank value of 4_8 :

Example 1

$$BBCON = 66107_8$$

F-bank 33
$$\left. \begin{array}{c} \text{F-bank 43} \\ \text{S-bank 4} \end{array} \right\}$$
 F-bank 43 $\left. \begin{array}{c} \text{F-bank 43} \\ \text{E-bank 7} \end{array} \right.$

Example 2

$$BBCON = 66063_8$$

Example 3

BBCON = 02006_{8}

F-bank 01

S-bank unnecessary

E-bank 6

DOWNLINK

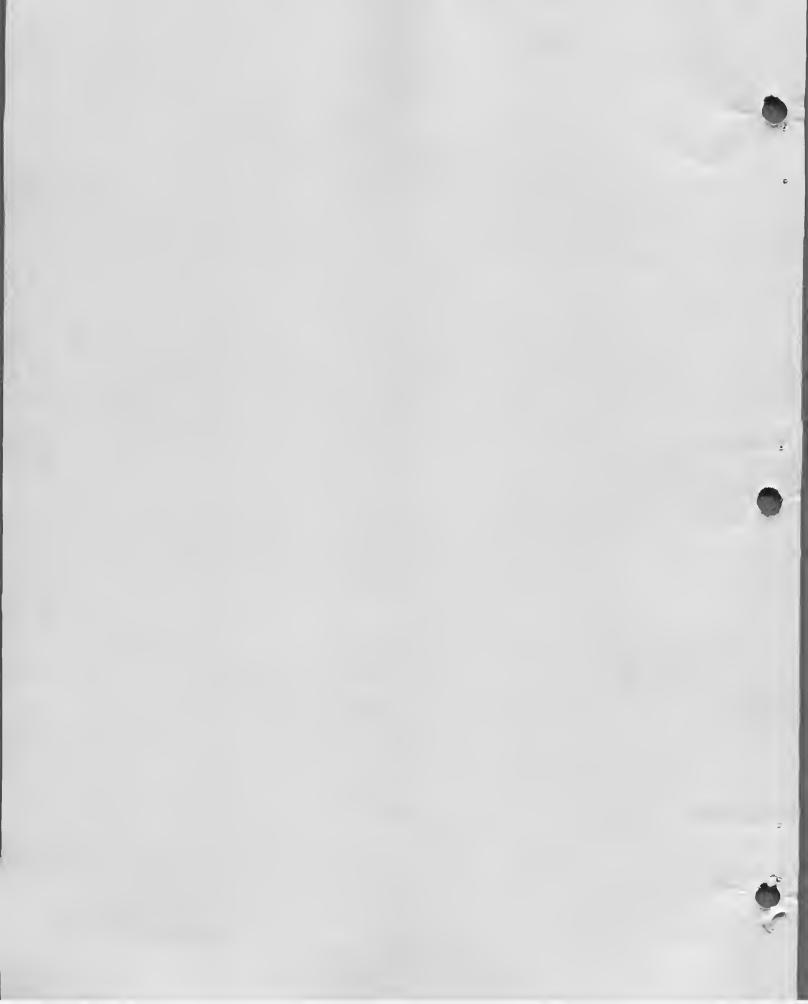
Listed below are the EMPs and the particular downlist transmitted during the operation of each ${\ensuremath{\sf EMP}}$:

COLOSSUS

EMP		Downlist
500		P22 List
501		P22 List
502		Any
503		Coast and Align List
504		Rendezvous and Prethrust List
505	P20/P23	Rendezvous and Prethrust List
	P22/P24	P22 List
	P5X	Coast and Align List
506		Rendezvous and Prethrust List
508		P22 List
509		Any
512		Powered List, Coast and Align List
513		Entry and Update List
514		Rendezvous and Prethrust List
515		Rendezvous and Prethrust List
517		Coast and Align List
518		Coast and Align List,
		Entry and Update List (during P27)
520		Rendezvous and Prethrust List,
		Coast and Align List
521		Coast and Align List
522		Any but Powered
523		Any but Powered

LUMINARY

EMP	Downlist
99 100 A & 100 B	Orbital Maneuvers List Any
101	Orbital Maneuvers List
102	Any
103A & 103B	Descent and Ascent List
104	Rendezvous and Prethrust List
106	Coast and Align List, Lunar Surface Align List
107	Descent and Ascent List
108	Any



EMP 99: GUIDED RCS TRANSLATIONAL MANEUVER

PURPOSE:

Erasable Memory Program EMP 99 provides an unmanned RCS translational thrusting capability. This capability applies to earth and to lunar orbits.

FUNCTIONAL DESCRIPTION:

When P99 is initiated (VERB 30 ENTR), the powered flight downlist is selected. (See Figure: EMP 99.) The initial thrust direction and the initial value of the VG vector is calculated by the powered-flight and navigation routines. The attitude errors are zeroed, the DAP deadband is set to 1 deg. and the attitude maneuver is requested. When the attitude maneuver has completed, the master ignition routine interrogates a register tabulation for calling the appropriate program. P99 values for this table are uplinked as part of the P99 code. The master ignition routine then branches upon these points to appropriate routines in the powered flight program, which display values of interest including time from ignition before the engine-on signal is given and time from The powered-flight program displays Δv residuals for monitor purposes. The P99 burn program is finished when P00 is selected as a response to the Δv residuals. P99 is a lead-in to the powered-flight program with the appropriate registers uplinked with the P99 code.

ASSUMPTIONS:

Program coding has been uplinked via P27 (see ERASABLE MEMORY and UPLINK).

RESTRICTIONS
AND LIMITATIONS:

The use of EMP 99 is restricted as follows:

- See RECOVERY
- 2. Use with descent stage prohibited unless the following are effected:
 - a. Content of Erasable Memory location 3405 is changed from octal 12324 to octal 12327 (TCF WANTAPS becomes TCF ULLGNOT)
 - b. NOUN 46 in R03 must be loaded to indicate descent stage, i.e., R1 Digit A = 2

3. If two-jet ullage is to be used, the following locations must be changed:

]	ECADR	Tag		Code	Octal (New)	Octal (Old)
	3734	F	2DEC	0.08896 B-7	00013	00026
	3735				14303	30605

This changes the value of thrust from 400 to 200 lb.

Also, the NOUN 46 load in R03 must indicate two-jet ullage, i.e., R1 digit B set to 0 or 1.

PROCEDURES:

- 1. Perform preliminary procedures before LM jettison (Crew):
 - a. DAP Data Load Routine (R03)
 - b. External Δv Targeting Program (P30)
 - c. GUID CONT set to PGNS
 - d. MODE CONTROL (PGNS) set to AUTO
 - e. ENG ARM set to OFF
 - f. UP DATA LINK set to DATA (to accept uplinked instructions)
- 2. Set up EMP 99 as follows (Ground):
 - Key VERB 96 ENTR to set QUITFLAG (to interrupt P00 state-vector integration)
 - b. Key VERB 5 NOUN 26 ENTR and verify address for EMP 99:

Rl 13001

R2 01420

R3 12067

- 3. Execute EMP 99 as follows:
 - a. Key VERB 30 ENTR to initiate EMP 99, which selects powered-flight downlist
 - b. Observe 99 in PROG registers

c. Observe display of desired burn attitude:

FL VERB 50 NOUN 18

R1 xxx.xx deg R
R2 xxx.xx deg P
R3 xxx.xx deg Y

NOTE. — DAP deadband is set to 1 deg for maneuver and burn; reference attitude is set to present attitude.

- d. Key VERB 33 ENTR to execute attitude maneuver to burn attitude
- e. When maneuver has completed, observe return of desired burn attitude display:

FL VERB 50 NOUN 18

R1 xxx.xx deg R
R2 xxx.xx deg P
R3 xxx.xx deg Y

- f. Key ENTR
- g. Observe display of time from ignition:

VERB 06 NOUN 40

R1 xxBxx min,sec TFI
R2 xxxx.x ft/sec VG
R3 xxxx.x ft/sec Δv

NOTE.—DSKY blanks at TIG-35. AVERAGEG Routine starts at TIG-30, and DSKY display is re-established.

h. Observe TIG:

VERB 06 NOUN 40

R1 00B00 min, sec TFI
R2 xxxx.x ft/sec VG
R3 xxxx.x ft/sec Δv

NOTE. —P42 guidance equations and DAP offset acceleration estimator enabled.

i. Monitor NOUN 40 countdown to engine cutoff.

NOTE.—NOUN 40 (R1) now contains time from engine cutoff (TFC).

j. At cutoff, observe flashing display of cutoff parameters:

FL VERB 16 NOUN 40

R1	00B00 min,sec	TFC
R2	xxxx.x ft/sec	VG
R3	xxxx.x ft/sec	Δv

NOTE. — DAP offset acceleration estimator disabled, and DAP deadband is returned to pre-EMP 99 value.

k. Key VERB 33 ENTR and observe display of burn residuals:

FL VERB 16 NOUN 85

R1	xxxx.x ft/sec	VGX
R2	xxxx.x ft/sec	VGY
R3	xxxx.x ft/sec	VGZ

NOTE. — DAP deadband is set to 0.3 deg; reference attitude is set to present attitude.

I. Key VERB 33 ENTR and observe display, "Please select new program":

FL VERB 37

NOTE. — DAP deadband is returned to pre -EMP 99 value.

m. Key 00 ENTR and observe 00 in PROG registers

NOTE. — AVERAGEG Routine turns off, and Coast/Align Downlist is selected.

RECOVERY/ TERMINATION:

- 1. Before attempting another burn, ensure that ENG ARM and ENG GMBL switches have been restored to appropriate positions.
- 2. Once EMP 99 has been loaded, new landing-radar padloads must be uplinked before attempting a lunar landing.
- 3. Once EMP 99 has been loaded, new ATIGINC/PTIGINC padloads must be uplinked before attempting transfer-phase midcourse targeting (P35/P75).
- 4. Once EMP 99 has been loaded, new AOTAZ and AOTEL padloads must be uplinked before attempting an IMU alignment.
- 5. To terminate EMP 99 at any time, key VERB 96 ENTR. The DAP deadband is returned to pre-EMP 99 value.

ECADR	Tag	<u>C</u>	ode_	Octa1
3400	SETDAPFL	TC	DOWNFLAG	05520
3401		ADRES	DRIFTDFL	00312
3402		TC	TASKOVER	05263
3404	P99WHICH	VN	0640	01450
3405		TCF	WANTAPS	12324
3406	P99IGN	TC	DOWNFLAG	05520
3407		ADRES	IDLEFLAG	00161
3410		TC	SETDAPFL	01400
3411		TCF	P40SPOT	12150
3412		DEC	2990	05656
3413		ADRES	STEERING	03667
3414		BBCON	STEERING	74066
3415		TCF	COMMON	12404
3416		TCF	IGNITION	12433
3417		TC	P99IGN	01406
3420	P99	TC	NEWMODEX	05313
3421		DEC	99	00143
3422		CAF	THREE	36266
3423		TS	DNLSTCOD	54333
3424 3425		TC RTB	INTPRET	06060 77634
3426		CADR	E/CALL S40.1	10636 56246
3427 3430 3431		GOTO	P40IN+3	77650 75202
3734 3735	F	2 DEC	0.17792 B-7	00026 30605
3736 3737	MDOT	2 DEC	0.05135 B-3	00151 05214
3740 3741	TDEC AY	2 DEC	0	00000
3742 3743	VEX	2 DEC	27 B-6	15400 00000
3455	WHICH	ADRES	P99WHICH	01404
1250	DVTHRUSH	OCT	0	00000
3515	DVCNTR	DEC	4	$00004 \\ 13001$
2371	N26/PRI	OCT	13001	
2372 2373	N26/2CAD	ADRES BBCON	P99 P99	$01420 \\ 12067$

UPLINK:

P27 uplink for loading LGC erasable memory for EMP 99 is as follows:

Load 1	Load 2	Load 3	Load 4
V71E	V71E	V72E	V 72 E
$24\mathrm{E}$	12E	$17\mathrm{E}$	$15\mathrm{E}$
$3404\mathrm{E}$	3734E	$3400\mathrm{E}$	$3455\mathrm{E}$
$1450\mathrm{E}$	26E	$5520\mathrm{E}$	$1404\mathrm{E}$
$12324\mathrm{E}$	30605E	3401E	$1250\mathrm{E}$
$5520\mathrm{E}$	$151\mathrm{E}$	312E	\mathbf{E}
$161\mathrm{E}$	$5214\mathrm{E}$	$3402\mathrm{E}$	$3515\mathrm{E}$
1400E	E	5263E	$4\mathrm{E}$
$12150\mathrm{E}$	${f E}$	3426E	$2371\mathrm{E}$
5656E	$15400\mathrm{E}$	10636E	13001E
$3667\mathrm{E}$	\mathbf{E}	$3427\mathrm{E}$	2372E
74066E	V33E	$56246\mathrm{E}$	$1420\mathrm{E}$
$12404\mathrm{E}$		3430E	2373E
12433E		$77650 \mathrm{E}$	$12067\mathrm{E}$
1406E		3431E	V33E
5313E		7 52 02E	
143E	,	V33E	
36266E			
54333E			
6060E			
$77634\mathrm{E}$			
V33E			

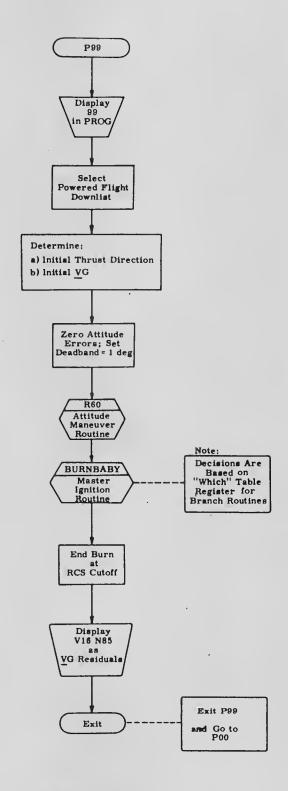


Figure: EMP99

EMP 100A: BACKUP FOR FAILED DSKY KEY USING ENG GMBL SWITCH

PURPOSE:

Erasable Memory Program EMP 100A provides a means of using the ENG GMBL switch to back up a failed DSKY key.

FUNCTIONAL DESCRIPTION:

EMP 100A is activated by VERB 31 ENTR. Once activated, sampling occurs once every 20 ms to determine whether the ENG GMBL switch has been moved from ON to OFF. (See Figure: EMP 100A.) If the ENG GMBL switch is detected first ON and then OFF, a single executive call is made to the CHARIN routine of PINBALL. In this call, EMP 100A supplies the preset keycode identifier that corresponds to the DSKY button that has failed. The effect is as though the DSKY button itself were functional and had been depressed. In fact, there is a small time lag between the instant that the ENG GMBL switch is placed in the OFF position and the receipt of the keycode by the PINBALL program, but the lag is far too small to be sensed by the crew.

ASSUMPTIONS:

EMP 100A can be used for DSKY backup when-

- 1. A DSKY key (other than PRO/STBY) has failed "open"; i.e., depressing key does not trigger an LGC KEYRUPT
- 2. The program has been previously padloaded or uplinked

NOTE. - VAC Area 5 should be checked prior to activation of EMP 100A in order to ensure that the code has not been altered.

3. High bit rate has been selected

RESTRICTIONS AND LIMITATIONS:

- 1. EMP 100A should not be activated while an erasable-memory dump (VERB 74) is in progress
- 2. EMP 100A can be used to back up only one key at a time
- 3. There is an increase in probability of a 31201 BAILOUT rstart during periods of high CPU activity. (VAC Area 5 is not available when EMP 100A is active.) This restart has no impact, however, on EMP 100A.

- 4. There is a small increase (1.46 percent) in CPU time.
- 5. If ENG GMBL switch is left in OFF position during DPS powered flight, DAP control will be impaired.
- 6. EMP 100A is deactivated by
 - a) VERB 36 ENTR (Fresh Start);
 - b) VERB 74 ENTR (Erasable-memory dump).
- 7. EMP 100A will be deactivated by a hardware restart occurring
 - a) during a small interval (<20 ms) following VERB
 37 ENTR xx ENTR, entry to P70 and P71, or POODOOs;
 - b) during antenna repositioning in P64;
 - c) between TIG -30 and turn on of ullage during a powered-flight program.

PROCEDURES:

When the prerequisite conditions exist (above), operate ${\rm EMP}$ 100 A as follows:

A. DSKY/UPLINK

1. To change or load key code (KK),

key-

VERB 21 NOUN 1 ENTR

730 ENTR

KK ENTR

- 2. To activate,
 - a) key—

VERB 5 NOUN 26 ENTR .

b) observe-

R1 00001

R2 00722

R3 10100

c) key-

VERB 31 ENTR

B. ENG GMBL SWITCH

1. Normal—

To effect keystroke, place ENG GMBL switch to OFF momentarily; then return switch to ENABLE.

NOTE.—Sample rate is 50 times per second; therefore, ENG GMBL switch can be cycled as rapidly as necessary to effect data input. If switch is left in OFF position during DPS operation, however, DAP control of the engine gimbal will be interrupted.

- 2. Operation with Engine Gimbal Failure (ENG GMBL caution light on):
 - a. Place ENG GMBL switch to OFF

<u>NOTE</u>. — An extraneous keystroke will occur. It is desirable, therefore, to effect switch position when such keystroke will have no impact.

b. When a keystroke is required, place ENG GMBL switch momentarily in ENABLE; then return to OFF.

RECOVERY/ TERMINATION:

- 1. To terminate EMP 100A, key one of the following:
 - a. VERB 74 ENTR

VERB 37 ENTR xx ENTR

or

b. VERB 36 ENTR (Fresh Start)

or

c. VERB 21 NOUN 1 ENTR
335 ENTR

3532 ENTR

VERB 37 ENTR xx ENTR.

To recover from deactivation caused by a hardware restart, key (or uplink)—
VERB 31 ENTR.

ERASABLE MEMORY:

Padloaded coding for EMP 100A is as follows:

ECADR	Tag		Code	Octal
660 661 662 663 664 665 666 667 670 671 672 673 674 675 676 677 700 701 702 703 710 711 712 713 714 715 716 717 720 721 722 723 724 725 726	VAC5USE MULTFLAG AFTGOLOC CHKPHASE	OCT CA TS CA EXTEND RAND XCH CCS TC TC CCS TC CA TC 2C ADR CA MASK TC OCT INDEX TS CCS TC TC OCT OCT OCT CCT CCA TS CA TS CA TC	0 ZERO VAC5USE BIT9 CHAN32 MULTFLAG A +2 CHKPHASE MULTFLAG CHKPHASE PRIO30 NOVAC CHARIN KEYBKUP LOW5 AFTGOLOC 400 LOCCTR MPAC PHASE1 DNPHASE2 PHASCHNG 7011 77777 722 10100 DNPHASE2 ZERO VAC5USE EPROGAD DNTMGOTO TASKOVER	00000 34746 54660 34734 00006 02032 56703 10000 00672 00712 10703 00712 34346 05063 02057 60101 30730 74337 00710 00400 50064 54154 10752 03532 05355 07011 77777 00722 10100 03532 34746 54660 30727 54335 05263
727 730	EPROGAD KEYBKUP	OCT OCT	661 KK	00661 000KK



Uplink for loading EMP 100A code by P27 is as follows:

Load 1	Load 2	Load 3	Load 4
V71E	V71E	V71E	V71E
$20\mathrm{E}$	$20\mathrm{E}$	$17\mathrm{E}$	5E
$660\mathrm{E}$	676E	$714\mathrm{E}$	2371E
${f E}$	$2057\mathrm{E}$	$5355 \mathrm{E}$	1E
34746E	60101E	7011E	722E
$54660 \mathrm{E}$	$30730 \mathrm{E}$	77777E	10100E
$34734 \mathrm{E}$	$74337 \mathrm{E}$	$722\mathrm{E}$	V33E
$6\mathrm{E}$	$710\mathrm{E}$	10100E	
2032E	400E	3532E	
56703E	$\mathbf E$	34746E	
10000E	${f E}$	$54660 \mathrm{E}$	
672E	${f E}$	$30727 \mathrm{E}$	
712E	$\mathbf E$	$54335 \mathrm{E}$	
10703E	50064E	$5263\mathrm{E}$	
712E	54154E	661E	
34346E	10752E	KKE	
5063E	3532E	V33E	
V33E	V33E	YOOL	
. 002	* 00E		

NOTE. - Load 4 is the NOUN 26 load.

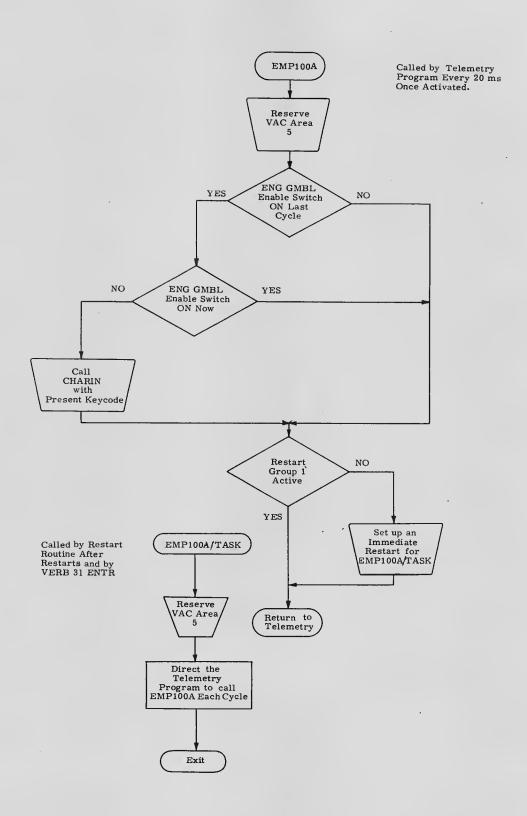


Figure: EMP100A

EMP 100B: BACKUP FOR FAILED DSKY KEY USING MODE SEL SWITCH

PURPOSE:

Erasable Memory Program EMP 100B provides a means of using the MODE SEL switch to back up a failed DSKY key.

FUNCTIONAL DESCRIPTION:

EMP 100B is activated by VERB 31 ENTR. Once activated, sampling occurs once every 20 ms to determine whether the MODE SEL switch has been moved from LDG RADAR or AGS to PGNS. (See Figure: EMP 100B.) If the MODE SEL switch is detected first at LDG RADAR or AGS and then at PGNS, a single executive call is made to the CHARIN routine of PINBALL. In this call, EMP 100B supplies the preset keycode identifier that corresponds to the DSKY button that has failed. The effect is as though the DSKY button itself were functional and had been depressed. In fact, there is a small time lag between the instant that the switch is placed in the PGNS position and the receipt of the keycode by the PINBALL program, but the lag is far too small to be sensed by the crew.

ASSUMPTIONS:

EMP 100B can be used for DSKY backup when-

- 1. A DSKY key (other than PRO/STBY) has failed "open"; i.e., depressing key does not trigger an LGC KEYRUPT
- 2. The program has been previously padloaded or uplinked

NOTE. — VAC Area 5 should be checked prior to activation of EMP 100B in order to ensure that the code has not been altered.

3. High bit rate has been selected

RESTRICTIONS AND LIMITATIONS:

- 1. EMP 100B should not be activated while an erasablememory dump (VERB 74) is in progress
- 2. EMP 100B can be used to back up only one key at a time.

- 3. There is an increase in probability of a 31201 BAILOUT restart during periods of high CPU activity. (VAC Area 5 is not available when EMP 100B is active.) This restart has no impact, however, on EMP 100B.
- 4. There is a small increase (1.46 percent) in CPU time.
- 5. EMP 100B is deactivated by
 - a) VERB 36 ENTR (Fresh Start)
 - b) VERB 74 ENTR (Erasable-memory dump)
- 6. EMP 100B will be deactivated by a hardware restart occurring
 - a) during a small interval (<20 ms) following VERB 37 ENTR xx ENTR, entry to P70 and P71, or POODOOs:
 - b) during antenna repositioning in P64;
 - c) between TIG-30 and turn on of ullage during a powered-flight program.

PROCEDURES:

When the prerequisite conditions exist (above), operate ${\rm EMP}$ 100B as follows:

A. DSKY/UPLINK

1. To change or load key code (KK),

key--

VERB 21 NOUN 1 ENTR

730 ENTR

KK ENTR

- 2. To activate,
 - a) key—

VERB 5 NOUN 26 ENTR

b) observe—

R1 00001

R2 00722

R3 10100

c) key-

VERB 31 ENTR

B. MODE SEL SWITCH

1. If the MODE SEL switch is desired at PGNS—To effect keystroke, place the switch to LDG RADAR or AGS momentarily; then return switch to PGNS.

NOTE. —Sample rate is 50 times per second; therefore, MODE SEL switch can be cycled as rapidly as necessary to effect data input.

2. If MODE SEL switch is desired at LDG RADAR or AGS—When a keystroke is required, place the switch momentarily to PGNS; then return to LDG RADAR or AGS.

RECOVERY/ TERMINATION:

1. To terminate EMP 100B either

key-

a. VERB 74 ENTR
VERB 37 ENTR xx ENTR

or

b. VERB 36 ENTR (Fresh Start)

or

c. VERB 21 NOUN 1 ENTR
335 ENTR
3532 ENTR
VERB 37 ENTR xx ENTR

To recover from deactivation caused by a hardware restart, key (or uplink)—
VERB 31 ENTR

ERASABLE MEMORY:

Padloaded coding for EMP 100B is as follows:

ECADR	Tag	Code		Octa1
ECADR 660 661 662 663 664 665 666 667 670 671 672 673 674 675 676 677 700 701 702 703 710 711 712 713 714 715 716 717 720 721 722 723 724	Tag VAC5USE MULTFLAG AFTGOLOC CHKPHASE	Code OCT CA TS CA EXTEND RAND XCH CCS TC TC CCS TC CA TC 2C ADR CA MASK TC OCT INDEX TS CCS TC TC OCT OCT OCT OCT CCT CCA TC CA TC CCS TC	0 ZERO VAC5USE BIT6 CHAN30 MULTFLAG A +2 CHKPHASE MULTFLAG CHKPHASE PRIO30 NOVAC CHARIN KEYBKUP LOW5 AFTGOLOC 0 LOCCTR MPAC PHASE1 DNPHASE2 PHASE1 DNPHASE2 PHASCHNG 7011 77777 722 10100 DNPHASE2 ZERO VAC5USE EPROGAD	00000 34746 54660 34737 00006 02030 56703 10000 00672 00712 10703 00712 34346 05063 02057 60101 30730 74337 00710 00000 50064 54154 10752 03532 05355 07011 77777 00722 10100 03532 34746 54660
725 726 727 730	EPROGAD KEYBKUP	TS TC OCT OCT	DNTMGOTO TASKOVER 661 KK	30727 54335 05263 00661 000KK

UPLINK:

Uplink for loading EMP 100B code by P27 is as follows:

Load 1	Load 2	Load 3	Load 4
V71E 20E 660E E 34746E 54660E 34737E 6E 2030E 56703E 10000E 672E 712E 10703E 712E 34346E 5063E	V71E 20E 676E 2057E 60101E 30730E 74337E 710E E E E E 50064E 54154E 10752E 3532E	V71E 17E 714E 5355E 7011E 77777E 722E 10100E 3532E 34746E 54660E 30727E 54335E 5263E 661E KKE V33E	V71E 5E 2371E 1E 722E 10100E V33E
$34346\mathrm{E}$	$10752\mathrm{E}$	KKE	

NOTE. — Load 4 is the NOUN 26 load.

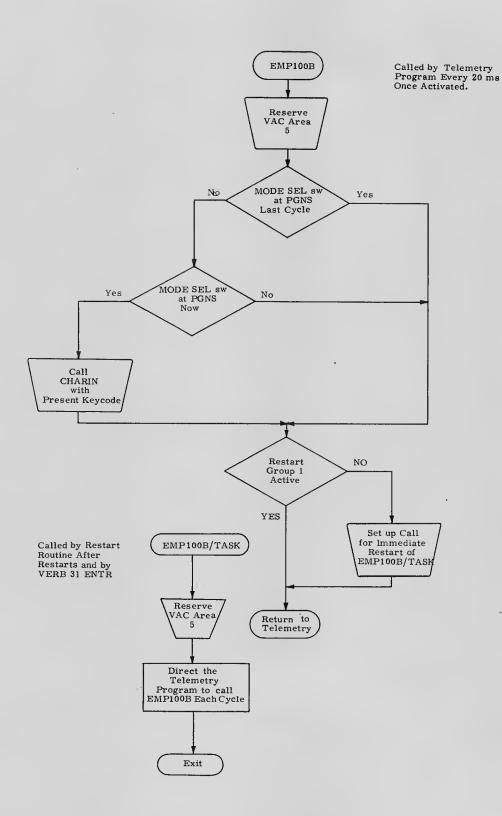


Figure: EMP100B

EMP 101: P47 WITH DAP DRIVING GTS

PURPOSE:

EMP 101 provides a means of having the PGNCS control attitude by use of the GTS during a DPS P47 burn.

FUNCTIONAL DESCRIPTION:

Two functions are provided by the PGNCS:

- Display of NOUN 83, the velocity gained in body axes, which provides the criterion for burn termination;
- 2. Control of the vehicle attitude, including the use of GTS during the burn.

The NOUN 83 display is the normal output of P47. The display does not depend on the orientation of the IMU with respect to reference coordinates, but only with respect to body coordinates, provided by the CDUs.*

For the DAP to use the GTS when the engine is thrusting—and only when it is thrusting—the DPS/APS Thrust Fail Routine (R40) is manually set up. This is done by resetting IDLEFLAG (so that P47's SERVICER jobs will execute R40) and by loading the appropriate thrust threshold into DVTHRUSH. Further, the DAP is put in the powered-flight mode by resetting DRIFTBIT in DAPBOOLS. This causes the DAP to estimate angular accelerations and to honor VERB 65 commands inhibiting RCS pitch and roll control.

P47 integrates the LM state vector to current time. If the state vector time tag is very far from LGC clock time, this integration could take a long time. To avoid having to wait, the crew manually loads the state vector time (NOUN 38) with the clock time (NOUN 65). An erroneous state vector is of no concern unless it causes a POODOO 20430 or 21204 Alarm, which forces an exit from P47 (item 4 under RE-STRICTIONS AND LIMITATIONS).

The measured ΔV vector is transformed from IMU to reference coordinates by multiplication with REFSMMAT, but the vector is then transformed back by multiplication with the inverse of REFSMMAT. Consequently, the content of REFSMMAT makes no difference to the NOUN 83 display as long as the matrix is orthogonal.

ASSUMPTIONS:

- 1. The IMU is on.
- 2. REFSMMAT contains an orthogonal matrix.
- 3. Current DAP data has been entered (R03).
- 4. The GTS has been trimmed so that the thrust vector after initial compliance will pass within 1 degree of the c.g.
- 5. The vehicle configuration can be LM-alone or LM docked to the CSM.

RESTRICTIONS AND LIMITATIONS:

- 1. DPS starting and stopping, ullage, and DPS throttling must be controlled manually.
- 2. If the LGC senses a DPS thrust failure, the DAP will stop attempting to use the GTS as long as the failure lasts, but there will be no DSKY indication other than a decrease or cessation in the incrementing of NOUN 83.
- 3. When SNUFFER is set (VERB 65) and DRIFTBIT is cleared, there will be no pitch and roll control with the RCS jets even if there is no DPS thrust. DRIFTBIT is automatically set by a major mode change, so this lock-out is terminated when P47 is exited. This inhibition does not affect manual attitude control that uses translational commands.
- 4. The LM state vector must be such as not to cause a POODOO 20430 or 21204 Alarm at the beginning of P47. A 21204 Alarm results if the state-vector time is in the past and the computed time step is zero. A 20430 Alarm results if the integration-computed acceleration exceeds program limit. A subsurface state vector or erroneous erasable constant can cause these alarms.
- 5. Attitude limit cycling will cause some fluctuating fraction of the total velocity gained to appear in the Y and Z components of NOUN 83 after the burn is terminated; for example, for a 0.5-deg DAP deadband, the fluctuation could be approximately 1 percent of the total Δv . Trimming these components will, therefore, be impossible to accomplish precisely by this display. (In the docked case, the long moment arm between the Y,Z translation jets and the center of gravity causes additional difficulty.)

PROCEDURES:

- 1. Maneuver the vehicle to the burn attitude
- 2. Coarse align IMU (optional):

Key-

VERB 41 NOUN 20 ENTR

ENTR

ENTR

ENTR

3. Synchronize CDUs; exit coarse-align mode:

Key-

VERB 40 NOUN 20 ENTR

4. Set REFSMFLG:

Key-

VERB 25 NOUN 7 ENTR

77 ENTR

10000 ENTR

1 ENTR

5. Set the threshold for the thrust monitor:

Key-

VERB 21 NOUN 1 ENTR

1250 ENTR

14 ENTR (if docked to CSM)

44 ENTR (if LM-alone)

- 6. Enter P47; load the orbital integration time tag with the LGC clock time:
 - a. Key-

VERB 37 ENTR 47 ENTR

- b. Observe 47 in PROG registers
- c. Key-

VERB 6 NOUN 65 ENTR

d. Record three components:

R1 +ooxxx. hr

R2 +oooxx. min

R3 +oxx.xx sec

e. Key-

VERB 25 NOUN 38 ENTR

(and load NOUN 65 values):

+xxx ENTR

+xx ENTR

+xxxx ENTR

7. Enable thrust monitor (reset IDLEFLAG):

Key-

VERB 25 NOUN 7 ENTR

103 ENTR

100 ENTR

ENTR

8. Put DAP in powered-flight status (reset DRIFTBIT): Key-

VERB 25 NOUN 7 ENTR

111 ENTR

200 ENTR

ENTR

9. Before ignition, ensure that the DAP is in the attitudehold mode and, if docked with the CSM, that RCS pitch and roll control will not be exercised during the burn.

NOTE.—For jet inhibition, "the burn" begins when step 8 is performed, Step 9 can be performed earlier, depending upon how the attitude is held before the burn:

- a. MODE CONT (PGNS) set to ATT HOLD
- b. Key VERB 77 ENTR
- c. Key VERB 65 ENTR (only if docked to CSM)
- 10. Null the NOUN 83 display (accumulated Δv): Key-

VERB 32 ENTR

- 11. Perform manual ullage as required
- 12. Perform manual engine start and manual throttle control

NOTE.—The initial CSM-docked throttle profile is 5-sec minimum thrust and 21 sec at 40 percent. The initial LM-alone profile is 26-sec minimum thrust.

- 13. Perform manual engine stop when NOUN 83 (R1) equals the desired Δv .
- 14. If docked, damp excessive pitch and roll rates via translation commands.
- 15. Exit P47: in the docked case, this re-allows automatic RCS pitch and roll control:

Key-

VERB 96 ENTR

16. Clear SNUFFER:

Key-

VERB 75 ENTR

RECOVERY/ TERMINATION:

There is a very small chance that the load of NOUN 38 will not "take" because the internal computations overwrite the loaded values. If the COMP ACTY light on the DSKY stays on continuously for more than 8 seconds after NOUN 38 is loaded, the load should be repeated.

If the LM state vector is such as to cause POODOO Alarm 20430 or 21204 upon entry to P47, a new state vector must be loaded. P47 can then be called again.

ERASABLE MEMORY:

NA

UPLINK:

NA



EMP 102: SOFTWARE RESTART

PURPOSE:

EMP 102 provides a means of causing a software restart

by keying VERB 31 ENTR.

FUNCTIONAL DESCRIPTION:

EMP 102 uses existing fixed program code to perform

BAILOUT and store Alarm Code 31211.

ASSUMPTIONS:

NA

RESTRICTIONS AND LIMITATIONS:

NA

PROCEDURES:

1. Key VERB 25 NOUN 26 ENTR

1 ENTR

2040 ENTR

16000 ENTR

 To effect software restart, key VERB 31 ENTR

3. Observe PROG alarm light

4. Key VERB 5 NOUN 9 ENTR to observe alarm code 31211. "Illegal interrupt of extended verb"

5. Key RSET to clear alarm

RECOVERY/ TERMINATION:

NA

ERASABLE MEMORY:

NA

UPLINK:

NA



EMP 103A: DESCENT WITH FAILED CDUs

PURPOSE:

EMP 103A provides a means of computing actual CDU values and displaying commanded CDU values should one or more CDUs fail during lunar descent or abort. Although EMP 103A could be used to back up failed CDUs in Pl2, no testing of this has been done since Pl2 has both AGS and manual backup. M.I.T. is not able to fully simulate the control configuration needed to test this EMP. EMP 103A has been used, however, on the LMS at KSC and at the Grumman simulator.

FUNCTIONAL DESCRIPTION:

Every 2 seconds, EMP 103A computes the desired gimbal angles (CDUYD and CDUZD) from the thrust axis desired by guidance and computes values for the actual gimbal angles (CDUY and CDUZ) from the measured acceleration vector (DELV). CDUYD and CDUZD are put into NOUN 87 (displayed by VERB 16), which the crew flys using the AGS autopilot. A second part of EMP 103A, executed every 20 ms, zeroes CDUX and puts the computed actual gimbal angles into the CDUY and CDUZ registers for use by the radar update and other routines—but not by the PGNCS autopilot, which must be off.

ASSUMPTIONS:

- 1. IMU is stable and correctly aligned for landing
- 2. Attitude control performed by crew in AGS ATT HOLD
- 3. EMP 103Ahas been loaded. (See ERASABLE MEMORY and UPLINK.)
- 4. High bit rate has been selected.

RESTRICTIONS AND LIMITATIONS:

- 1. Unless CDUX is okay, landing-analog-display crosspointers give velocities in the downrange and crossrange directions rather than in the body-oriented forward and lateral directions.
- 2. The P64 landing-site redesignation capability should not be enabled (PRO response to FL VERB 06 NOUN 64 is unwise) since hand-controller deflections required in flying AGS attitude hold will be interpreted as site redesignations by the PGNCS.

- 3. Use of LR to update the PGNCS state vector is not advised when a bit failure has occurred that would allow CDU values to change in large increments.
- 4. EMP 103A consumes about 5 percent of the LGC duty cycle. Failure of a CDU in the runaway mode consumes about 7.5 percent per CDU. To conserve LGC time, therefore, it is necessary that the PGNCS DAP be turned off before the erasable program is activated. (See ASSUMPTIONS.)
- 5. Hardware restart vulnerability during antenna repositioning at High Gate: Simultaneous occurrence of hardware restart and crew selection of P70/P71 at High Gate might disable EMP 103A.
- 6. EMP 103A is deactivated by either
 - a. VERB 74 ENTR

or

b. VERB 36 ENTR

- 7. EMP 103A should not be activated while an erasable-memory dump (VERB 74) is in progress.
- 8. There is some increase in the probability of a 31201/31202 BAILOUT restart during periods of high CPU activity. (VAC Area 5 and core sets 5, 6, and 7 are used to contain the EMP.)
- 9. Should a hardware or software restart (including VERB 37 ENTR xx ENTR) occur before EMP 103A activation, core sets 5, 6, and 7 and VAC area 5 should be checked to ensure that code has not been altered.

PROCEDURES:

- 1. Before PDI
 - a. GUID CONT to AGS
 - b. MODE CONTROL PGNS to OFF
 - c. Manual ullage
 - d. Manual engine on
- Between PDI and throttle up, if not sooner, key VERB
 ENTR to activate erasable program.

NOTE. - Between PDI and throttleup is the last possible moment for starting EMP 103A. If CDUZ is failed runaway, and EMP 103A is to be used for making the landing, it is better to use EMP 103 A in the first place instead of EMP 108 for preventing GLOCKMON from falsely sensing gimbal lock (CDUZ greater than 85 deg) and throwing the IMU into coarse align. This is to avoid the difficulty of switching from one EMP to the other in midstream. Although the CDUs computed by EMP 103A would not be valid between TIG-30 and ullage, the singleprecision arcsin routine used by EMP 103A cannot produce outputs greater than 81.4 deg. Therefore, CDUZ would never be computed as in gimbal lock.

3. After throttle up, key VERB 16 NOUN 87 ENTR to monitor desired CDUY and CDUZ values:

R1 xxx.xx deg CDUYD (pitch)

R2 xxx.xx 'deg CDUZD (roll)

NOTE.—NOUN 87 values are invalid before Guidance begins at throttle up.

4. Referencing NOUN 87 and the FDAI Ball, maintain attitude with AGS autopilot.

NOTE 1.—If yaw and roll are both nonzero on FDAI Ball, NOUN 87 (R1,R2) values for pitch and roll will not be equal to FDAI values. For the small roll angles (less than 3 deg) normal for landing, however, the NOUN 87 values are accurate enough to be used.

NOTE 2.—NOUN 22 cannot be used because FINDCDUW may still be providing bad data.

NOTE 3.—To determine when to enable LR state-vector updates, either depress KEY REL and observe NOUN 63 display of Delta H or ask ground to advise. (Yaw to zero before keying VERB 57 ENTR.)

RECOVERY/ TERMINATION:

To deactivate EMP 103A and disengage restart protection, key either:

- 1. VERB 74 ENTR VERB 37 ENTR xx ENTR
- 2. VERB 21 NOUN 1 ENTR
 335 ENTR
 3532 ENTR
 VERB 37 ENTR xx ENTR
- 3. VERB 36 ENTR

NOTE.—EMP 103A cannot be deactivated by methods 1 or 2 while SERVICER is running.

NOTE.—Should one or more CDUs be operating normally, it may be desirable to modify coding (*) to permit use of the good CDU data. If the only failure is CDUX, the FINDCDUW-computed values in NOUN 22 are accurate, but CDUX must still be overwritten to obtain valid radar data.

ECADR	Tag		Code		Octal
0250 *0251 0252 0253 0254 0255 0256 0257 0260 0261 0262 0263 0264 0265 0266 0267 0270 0271 0272 0273 0274 0275 0276 0277 0300 0301 0302 0303 0304 0305 0306 0307 0310 0311 0312 0313 0660	Tag	TC TS CCS TC OCT OCT OCT NOOP MASK CS AD CCS TC	Code	0300 CDUX PHASE1 0261 PHASCHNG 07011 77777 00311 10100 7776 AVGXIT 0267 A 00661 00250 0273 0676 AVGEXIT 0674 0267 0675 0270 DNTMGOTO BIT1 0263 0277 0313 ZERO VAC5USE Q 0300 TASKOVER	00300 54032 10752 00261 05355 07011 77777 00311 10100 30000 77776 41251 60267 10000 00661 00250 00675 77776 30270 54335 44744 54263 54277 54313 34746 54660 00002 005263 77776 00000
0661		TC		0300	00300
0662 0663 0664 0665 0666		TC VLOAD TC DXCH		INTPRET EXIT UNFC/2 0710 AZ	06060 77575 03252 00710 53345

0667 0670 0671 0672 0673 0674 0675 0676 0677 *0700 0701 0702 0703 0704 0705	TC VLOAD TC DXCH TC TS EXTEND DC A DXCH TC	INTPRET EXIT DELV 0710 0702 GUIDANCE AVGEXIT 0702 CDUY DNPHASE2	06060 77575 00325 00710 52703 55251 00006 30703 52034 03532
0707 0710 0711 0712	EXTEND QXCH TC	0704 INTPRET	00006 22704 06060
0713 0714	UNIT CA	EXIT MPAC +3	77456 30157
0715 0716 0717	TC CADR TS	BANKCALL SPARCSIN -1 0705	04607 61656 54705
0720 0721 0722	TC RTB	INTPRET UNIT ZEROMID	06060 53434 63671
0723 0724 0725	EXIT LXCH CS	MPAC MPAC +5	77776 22154 40161
0726 0727 0730 0731	TC CADR LXCH TC	BANKCALL ARCTRGSP 0705 0704	04607 61606 22705 00704

^{*}For individual CDU failure, load indicated location according to table.

	CDU		
LOC	X FAIL	Y FAIL	Z FAIL
251	54032	30000	30000
700	30000	54033	22034



P27 uplink (or DSKY entry in P00):

Load 1	Load 2	Load 3	Load 4	Load 5
1000				
V71E	V71E	V71E	V71E	V72E
24E	24E	24E	24E	15E
250E	272E	660E	710E	$1251\mathrm{E}$
300E	676E	\mathbf{E}	6E	3656E
*54032E	$31251\mathrm{E}$	$300\mathrm{E}$	$22704\mathrm{E}$	702E
10752E	54674E	6060E	6060E	E
$261\mathrm{E}$	$30267 \mathrm{E}$	$77575 \mathrm{E}$	77456E	703E
5355E	$675\mathrm{E}$	3252E	$30157\mathrm{E}$	\mathbf{E}
7011E	77776E	$710\mathrm{E}$	$4607\mathrm{E}$	$2371\mathrm{E}$
77777E	30270E	$53345\mathrm{E}$	$61656\mathrm{E}$	1 E
311E	$54335 \mathrm{E}$	6060E	$54705 \mathrm{E}$	2372E
10100E	44744E	$77575 \mathrm{E}$	6060E	311E
30000E	54263E	$325\mathrm{E}$	$53434\mathrm{E}$	2373E
30000E	$54277 \mathrm{E}$	710E	$63671\mathrm{E}$	\mathbf{E}
77776E	54313E	52703E	$77776 \mathrm{E}$	V33E
$41251\mathrm{E}$	34746E	\mathbf{E}	$22154\mathrm{E}$	
$60267\mathrm{E}$	$54660\mathrm{E}$	$55251\mathrm{E}$	40161E	
10000E	2E	$6\mathrm{E}$	$4607\mathrm{E}$	
$661\mathrm{E}$	300E	30703E	$61606\mathrm{E}$	
$250\mathrm{E}$	5263E	*52034E	$22705\mathrm{E}$	
$273\mathrm{E}$	$77776 \mathrm{E}$	$3532\mathrm{E}$	$704\mathrm{E}$	
V33E	V33E	V33E	V33E	

^{*}See table under ERASABLE MEMORY.



EMP 106: INCREMENT AOT DETENT POSITION

PURPOSE:

EMP 106 provides the Lunar Surface Sighting Mark Routine (R59) with a means of computing the cursor and spiral angles displayed by NOUN 79 for an advanced AOT detent position.

FUNCTIONAL DESCRIPTION:

EMP 106 redirects the PRO response to NOUN 79 to the location in R59 that increments the detent position code and recomputes the cursor and spiral angles.

ASSUMPTIONS:

- 1. IMU is on and REFSMMAT is valid.
- 2. The star selected under NOUN 70 is present in the AOT field of view for two detent positions.

RESTRICTIONS
AND LIMITATIONS:

- 1. EMP 106 procedures must be keyed into the DSKY by the operator during the NOUN 79 display in R59.
- 2. EMP 106 is self-destructing and must be reestablished with each usage.
- 3. When EMP 106 is active
 - a. A VERB 34 ENTR response to FL VERB 06 NOUN 79 causes the detent to advance the same as for a PRO response (does not terminate).
 - A VERB 32 ENTR recycles the FL VERB 06
 NOUN 79 display (does not recycle to FL VERB
 01 NOUN 70 to redefine a star).
 - c. An ENTR response causes the KEY REL light to come on and recycles the FL VERB 06 NOUN 79 display.

NOTE.—Both response a and response b destroy EMP 106. Response c does not.

PROCEDURE:

1. During the FL VERB 06 NOUN 79 display of R59, key

VERB 21 NOUN 1 ENTR 373 ENTR 32533 ENTR

2. To advance detent and destroy EMP 106, key PRO

RECOVERY/ TERMINATION:

See RESTRICTIONS AND LIMITATIONS, Item 2.

ERASABLE MEMORY:

NA

UPLINK:

NA

EMP 108: ZERO A RUNAWAY IMU CDU AND PREVENT COARSE ALIGN

PURPOSE:

EMP 108 provides a means of preventing T4RUPT from causing IMU to switch to coarse-align mode when CDUZ exceeds ±85 deg as a result of runaway CDUZ. Alternatively, EMP 108 can be used to zero CDUX or CDUY. See table under ERASABLE MEMORY.

FUNCTIONAL DESCRIPTION:

EMP 108 causes zero to be loaded into the failed CDU every 20 ms as part of the DOWNRUPT processing interrupt.

ASSUMPTIONS:

- 1. Valid data from failed CDU are not required during time EMP 108 will be activated.
- 2. High bit rate has been selected.

RESTRICTIONS AND LIMITATIONS:

- 1. EMP 108 should not be activated while an erasable-memory dump (VERB 74) is in progress.
- There is some increase in the probability of a 31201 BAILOUT restart during periods of high CPU activity.(VAC Area 5 is not available when EMP 108 is active.)This restart has no impact, however, on EMP 108.
- 3. There is a small increase (0.5 percent) in CPU time.
- 4. Should a hardware or software restart (including VERB 37) occur before EMP 108 activation, VAC Area 5 should be checked to ensure that code has not been altered.
- 5. IMU/CDU operations may not be successful.
- 6. EMP 108 has not been verified for operation with the DAP active; the DAP should be placed in the idling or minimum impulse mode.
- 7. EMP 108 is deactivated by
 - a) VERB 36 ENTR (Fresh Start);
 - b) VERB 74 ENTR (erasable-memory dump).
- 8. EMP 108 is deactivated by a hardware restart occurring
 - a) during a small interval (<20 ms) following VERB 37 ENTR xx ENTR, entry to P70 and P71, or after POODOOs;
 - b) during antenna repositioning in P64;

- c) between TIG-30 and turn on of ullage during a powered-flight program.
- 9. EMP 108 when used with the CDUZ deactivates the automatic moding to coarse align at gimbal angles greater than ±85 deg; maneuvering the vehicle into the area of a real gimbal lock will cause loss of inertial reference, with possible permanent damage to the IMU.

PROCEDURES:

1. Key VERB 5 NOUN 26 ENTR and verify contents:

R1 = 00001

R2 = 00674

R3 = 10100

To activate EMP 108, key — VERB 31 ENTR.

RECOVERY/ TERMINATION:

- 1. To deactivate EMP 108, key one of the following:
 - a. VERB 74 ENTR

VERB 37 ENTR xx ENTR VERB 40 NOUN 20 ENTR

or

b. VERB 36 ENTR

or

c. VERB 21 NOUN 1 ENTR

335 ENTR 3532 ENTR VERB 37 ENTR xx ENTR VERB 40 NOUN 20 ENTR

 To recover from deactivation caused by a hardware restart, key—
 VERB 31 ENTR



Program coding for EMP 108 is as follows:

ECADR	Tag		Code		Octal
660 661 662 *663 664 665 666 667 670 671 672 673	VAC5USE	OCT CA TS TS CCS TC TC OCT OCT ADRES OCT TC		0 ZERO VAC5USE CDUZ PHASE1 DNPHASE2 PHASCHNG 07011 77777 EPROGNIT 10100 DNPHASE2	00000 34746 54660 54034 10752 03532 05355 07011 77777 00674 10100 03532
674 675 676 677 700 701	EPROGNIT EPROGAD	CA TS CA TS TC ADRES		ZERO VAC5USE EPROGAD DNTMGOTO TASKOVER VAC5USE+1	34746 54660 30701 54335 05263 00661

 $^{^{*}}$ For individual CDU failure, load according to the following:

ECADR	CDUX	CDUY	CDUZ
663	54032	54033	54034

UPLINK:

Uplink for loading EMP 108 code by P27 is as follows:

Load 1	Load 2
V71E 24E 660E E 34746E 54660E * 10752E 3532E 5355E 7011E 77777E 674E 10100E 3532E 34746E 54660E 30701E 54335E 5263E 661E V33E	V71E 5E 2371E 1E 674E 10100E V33E

^{*}CDUZ failure. For CDUX failure, load 54032; for CDUY failure, load 54033.

LUMINARY 1E R-567

Internal Distribution List

Group 23A	D. Lutkevich Berberian Brand Gustafson Higgins Kachmar Klumpp	DL7-211 Kriegsman Levine Muller Pu Robertson	(11)
Group 23B	C. Flynn Klawsnik Nayar	DL7-221L Reed	(3)
Group 23B	C. Taylor Barnert Brodeur Cramer Hamilton	DL7-221L Lollar McCoy Ostanek Rye	(7)
Group 23B	J. Flaherty Adler Albert Berman Eyles	DL7-238A Millard Moore Volante Schulenberg	(8)
Group 23C	M. Erickson Weissman	DL7-215J	(1)
Group 23D	N. Auker Drake Dunbar Johnson Kiburz	DL7-213 Larson Metzinger Watson	(7)
Group 23H	R. Shane Cogliano Reber	DL7-272 Kossuth O'Connor	(4)
Group 23N	G. Grover Blanchard Johnson	DL11-201 Ogletree Tanner	(4)
Group 23P	B. Hwoschinsky Battin	DL7-203 Copps	(2)

Group 23P	E. Talbot Greene Stameris	DL7-252 Stubbs	(3)
Group 23S	P. Amsler Adams Edmonds Felleman Goodwin Grace Heinemann Kido Lawrence Lones	McOuat Megna O'Donnell Sheridan St. Amand Werner White Woolsey	(17)
APOLLO Lib	rary		(2)
CSDL Techni	cal Documentation Center		(2)

External Distribution List

LUMINARY

Charles Stark Draper Laboratory P. O. Box 21025 Kennedy Space Center Florida 32815 Attn: Mr. R. O'Donnell	(5)
Charles Stark Draper Laboratory Code EG/MIT Building 16 NASA Manned Spacecraft Center Houston, Texas 77058 Attn: Mr. George Silver	(3)
NASA MSC HW Building M7-409 Kennedy Space Center Florida 32815 Attn: Mr. Frank Hughes	(10)
Delco Electronics Division General Motors Corporation Milwaukee, Wisconsin 53201 Attn: Mr. J. Strikke, Dept. 42-02 Mr. W. Siarnicki, Dept. 38-02	(2)
Delco Electronics Division General Motors Corporation P. O. Box 21027 Kennedy Space Center Florida 32815 Attn: Mr. J. Kaiser	(2)
Delco Electronics Division General Motors Corporation P. O. Box 265 Bethpage, Long Island New York 11714 Attn: Mr. D. Dettmann	(1)
Delco Electronics Division General Motors Corporation P. O. Box 734 Downey, California 90241 Attn: Mr. D. Karstedt	(1)
Kollsman Instrument Corporation 575 Underhill Boulevard Syosset, Long Island New York Attn: Mr. F. McCoy	(1)

	oad for I no product (C. b. 2001). I achusetts 01776	(6)
NASA/MSFC	National Aeronautics and Space Administration George C. Marshall Space Flight Center Huntsville, Alabama 35812 Attn: J. Mack R-ASTR-5 (1) V. Buckelew S&E-AERO-MFG (1) A. Deaton R-AERO-DG (1) F. Moore R-ASTR-N (1) H. Hosenthien R-ASTR-N (1) A. McNair I-MO-R (1) D. Germany I-I/IB-E (1) R. Barraza I-V-Engineering (1)	Todd Fold
(1))	W. Chubb R-ASTR/NG (1) J. McCullough I-VE/T (1)	ATATE
NASA/MSC	National Aeronautics and Space Administration Manned Spacecraft Center APOLLO Document Control Group (BM 86) Houston, Texas 77058 Attn: Mr. J. Garman, FS6	(180 ± 1R)
(5.)	(letter of transmittal only) (letter of transmittal only) (0.0.c. decided of the content of th	Comme Material Attur
LINK)	LINK Group, GPSI SIMCOM 1740 A NASA Boulevard Houston, Texas 77058 Attn: Mr. D. Klingbell Attn: Mr. D. Klingbell Attn: Mr. D. Klingbell	Coperal " P. O. So Eugards
NASA/GSFC	National Aeronautics and Space Administration A () 1 Goddard Space Flight Center Greenbelt, Maryland 20771 Attn: Mr. Paul Pashby, Code 813 month tegror's action.	(ji 2) Stephenik Linguaga
GA	Grumman Aerospace LEM Crew Systems In the Land Bethpage, Long Island New York 11714 Attn: Mr. J. Marino	. . (.)3) 58 grav 7 (205.4) d17
(]	Mr. R. Pratt Mr. C. Tillman North American Rockwell Inc.	Males sole
NAR	Space and Information Systems Division and Advantage 12214 Lakewood Boulevard Downey, California 90241 notherogen became Attn: APOLLO Data Requirements Dept. 096-340	(4) T _i (3) T _i (4) T _i (5) T _i (5) T _i (6) T _i (7) T _i (7) T _i (7) T _i (8) T _i (9) T _i (9) T _i (10) T _i (

NASA/RASPO NAR	National Aeronautics and Space Administration Resident APOLLO Spacecraft Program Office North American Rockwell, Inc. Space and Information Systems Division 12214 Lakewood Boulevard Downey, California 90241	(1)
NASA/KSC	National Aeronautics and Space Administration Kennedy Space Center Florida 32899 Attn: Technical Document Control Office	. (3)
NASA/RASPO GE	General Electric Company APOLLO Systems P. O. Box 2500, Daytona Beach, Florida 32015 Attn: Mr. E. P. Padgett, Jr./Unit 509	(1)
NASA/HDQ	NASA Headquarters 600 Independence Avenue SW Washington, D. C. 20546 Attn: MAP-2 (4) Mission Director, Code MA (1) Robert Aller, Code MAO (1)	(6)
NASA/LEWIS	National Aeronautics and Space Administration Lewis Research Center Cleveland, Ohio 44135 Attn: Library	(2)
NASA/FRC	National Aeronautics and Space Administration Flight Research Center Edwards AFB, California 93523 Attn: Research Library	(1)
NASA/LRC	National Aeronautics and Space Administration Langley Research Center Langley AFB, Virginia 23365 Attn: Mr. A. T. Mattson	(2)
NAR/KSC	Kennedy Space Center Florida 32815 M.S.O.B. North American Mail Station ZK69 Attn: Mr. Donald Matteson	(1)
NASA/RASPO GA	National Aeronautics and Space Administration Resident APOLLO Spacecraft Program Officer Grumman Aerospace LEM Crew Systems Bethpage, Long Island New York 11714	(1)
NASA/WSMR	National Aeronautics and Space Administration Post Office Drawer MM Las Cruces, New Mexico 88001 Attn: RII4 Documentation	(. 2)

