

Space Program Operations Contract

SPACE SHUTTLE ORBITER OPERATIONAL LEVEL C FUNCTIONAL SUBSYSTEM SOFTWARE REQUIREMENTS GUIDANCE, NAVIGATION AND CONTROL PART E

NAVIGATION AIDS SUBSYSTEM OPERATING PROGRAM

PREPARED BY THE BOEING COMPANY
Space Exploration

Under Subcontract 6000132079, PDRD P1433q

December 14, 2007

DRD 1.4.3.3-q

Contract NNJ06VA01C



STS 83-0014V1-34
December 14, 2007

This page intentionally left blank.

STS 83-0014V1-34

**SPACE SHUTTLE ORBITER
OPERATIONAL
LEVEL C
FUNCTIONAL SUBSYSTEM SOFTWARE REQUIREMENTS
GUIDANCE, NAVIGATION, AND CONTROL
PART E
NAVIGATION AIDS
SUBSYSTEM OPERATING SYSTEM**

Issue Date: December 14, 2007

**NASA Contract No. NNJ06VA01C, DRD 1.4.3.3-q
USA Subcontract 6000132079, PDRD P1433q
WBS 2.4.3.2**

**The Boeing Company
Space Exploration**

This page intentionally left blank.

Prepared by

/s/ C. Bergman

C. Bergman
FSSR Documentation

/s/ H. Bui

H. Bui
G4.17 GPS SOP

/s/ J. Carter

J. Carter
G4.43 TACAN SOP
G4.44 MSBLS SOP
G4.45 RA SOP
G4.122 NAVDAD SOP

***/s/ D. Yang
for J. Pendergrass***

J. Pendergrass
G4.229 RR SOP

/s/ D. Owen

D. Owen, Product Manager
Flight Software Requirements & Verification
Flight Systems & Software Engineering

/s/ D. Bateman

D. Bateman, Manager
FSSR Documentation
Flight Systems & Software Engineering

This page intentionally left blank.

FOREWORD

The primary avionics software system (PASS) requirements for the computer programs that execute in the Shuttle general-purpose computers (GPCs) are specified in the Computer Program Development Specifications (CPDSs) and the Functional Subsystem Software Requirements (FSSR) documents. The requirements are specified at three levels

Level A CPDS – System-level Constraints and capabilities

Level B CPDS – Functional requirements

Level C FSSR – Detailed-level requirements

The Level A CPDS specifies system-level Constraints and capabilities that are not oriented toward any particular program end item. The Level B CPDS specifies system-level requirements for guidance, navigation, and control (GN&C) and detailed requirements for systems management (SM) and vehicle utility (VU). The Level C FSSR's specify the detailed requirements for GN&C and display/controls. The Level C reconfigurable requirements are specified in the systems management and payload documents. The Level A, Level B, and Level C requirements documents are listed in Section 2, Applicable Documents.

This page intentionally left blank.

DOCUMENT CHANGE RECORD

The following tabulation summarizes the change activity associated with this document.

DOCUMENT CHANGE RECORD			
Issue and Date	Change Summary/Effectivity		
OI-34 Release December 14, 2007	This baseline release for effectivity OI-34 and subs includes:		
	<u>CR</u>	<u>OI</u>	<u>Title</u>
	93017G	33	Baseline RSS99D0100 with Expanded Downlist
	93160B	32	Year End Roll Over (YERO) Reset
	93175	32	Expand Efectivity of SCR 93160B

This page intentionally left blank.

CHANGE REQUEST SUMMARY

The following tabulation lists the paragraphs, tables (T), and figures (F) that have been changed as a result of approved change requests for this release. Changes incorporated prior to this release are listed in Appendix R. Changes from approved change requests (CRs) to this document are indicated by a change bar in the outer margin of the changed page.

CHANGE REQUEST SUMMARY			
Paragraph	CR No.	OI	CR Title
N/A	93175	32	Expand Effectivity of SCR 93160B
4.17.6	93160B	32	Year End Roll Over (YERO) Reset
4.17.6-1.(T)	93160B	32	Year End Roll Over (YERO) Reset
4.17.6-2.(T)	93160B	32	Year End Roll Over (YERO) Reset
4.17.6.1.2	93160B	32	Year End Roll Over (YERO) Reset
4.17.6.10-1.(T)	93160B	32	Year End Roll Over (YERO) Reset
	93017G	33	Baseline RSS99D0100 with Expanded Downlist
4.17.6.10-2.(T)	93160B	32	Year End Roll Over (YERO) Reset
	93017G	33	Baseline RSS99D0100 with Expanded Downlist
4.43.6.5-1.(T)	93160B	32	Year End Roll Over (YERO) Reset
	93017G	33	Baseline RSS99D0100 with Expanded Downlist
4.43.6.5-2.(T)	93017G	33	Baseline RSS99D0100 with Expanded Downlist
4.44.6.3-1.(T)	93160B	32	Year End Roll Over (YERO) Reset
	93017G	33	Baseline RSS99D0100 with Expanded Downlist
4.44.6.3-2.(T)	93017G	33	Baseline RSS99D0100 with Expanded Downlist
4.45.6.2-1.(T)	93160B	32	Year End Roll Over (YERO) Reset
	93017G	33	Baseline RSS99D0100 with Expanded Downlist
4.45.6.2-2.(T)	93017G	33	Baseline RSS99D0100 with Expanded Downlist
4.122.2.4-1.(T)	93017G	33	Baseline RSS99D0100 with Expanded Downlist
4.122.2.4-2.(T)	93017G	33	Baseline RSS99D0100 with Expanded Downlist
4.229.7-1.(T)	93160B	32	Year End Roll Over (YERO) Reset
	93017G	33	Baseline RSS99D0100 with Expanded Downlist

This page intentionally left blank.

Table of Contents

<u>SECTION</u>	<u>TITLE</u>	<u>PAGE</u>
1.0	INTRODUCTION.....	1
1.1	PURPOSE	1
1.2	SCOPE.....	1
1.3	ORGANIZATION	1
2.0	APPLICABLE DOCUMENTS.....	5
2.1	LEVEL A DOCUMENTS	5
2.2	LEVEL B DOCUMENTS.....	5
2.3	LEVEL C DOCUMENTS.....	5
2.4	ICD DOCUMENTS.....	6
3.0	OVERVIEW	7
3.1	SHUTTLE OPERATIONAL PROGRAM	7
3.2	GN&C SOFTWARE REQUIREMENTS STRUCTURE	7
3.3	SOP OVERVIEW	7
3.4	SOP SUBSECTION ORGANIZATION	8
3.5	SOP FORMULATION NOTES.....	9
3.6	IMPLEMENTATION CONSTRAINTS	9
	3.6.1 Timing.....	10
4.0	DETAIL LEVEL REQUIREMENTS.....	11
	4.0.1 Traceability of Requirements	11
4.17	Global Positioning System Subsystem Operating Program	13
	4.17.1 GPS Requirements Overview	13
	4.17.1.1 GPS Functional Overview	13
	4.17.1.1.1 GPS Operating Modes.....	14
	4.17.1.1.2 GPS Digital Interfaces.....	17
	4.17.1.1.3 GPS Discrete Interfaces.....	17

Table of Contents

<u>SECTION</u>	<u>TITLE</u>	<u>PAGE</u>
4.17.1.2	GPS SOP Functional Overview	17
4.17.1.3	GPS SOP Functional Module Summary	18
4.17.1.4	Reference Documents	18
4.17.1.5	Software Protection.....	18
4.17.2	Definitions.....	19
4.17.3	Functional Requirements Traceability	26
4.17.4	GPS SOP Interfaces	26
4.17.5	Parameter Initialization.....	26
4.17.6	Detailed Requirements and Formulations.....	26
4.17.6.1	GPS Data Processing (GPS_TO_GPC)	35
4.17.6.1.1	NAV_MODE_WORD Data Processing (NAV_MODE_WORD_DATA)	37
4.17.6.1.2	NAV Message Data Processing (NAV_MESSAGE_DATA)	42
4.17.6.1.2.1	GPS Channels Status Data Processing (CHANNEL_STATUS_DATA)	49
4.17.6.1.3	INIT Message Data Processing (INIT_MESSAGE_DATA).....	56
4.17.6.1.4	TEST Message Data Processing (TEST_MESSAGE_DATA).....	58
4.17.6.2	GPS Data Quality Check Processing (GPS_QA1_CHECK).....	59
4.17.6.3	GPS Altitude Aiding Data Processing (GPS_ALT_DATA).....	60
4.17.6.4	GPC To GPS Data Processing (GPC_TO_GPS)	64
4.17.6.4.1	GPS Receiver Aiding Data Processing (RCVR_AIDING_DATA)	68
4.17.6.4.2	Initialization Data Processing (INIT_DATA).....	77
4.17.6.4.3	Nav Aiding Data Processing (NAV_AIDING_DATA).....	80
4.17.6.5	GPS Display Computations (GPS_DISP_COMP)	82
4.17.6.6	GPS SOP Downlist Computations (GPS_SOP_DL_COMP).....	86

Table of Contents

<u>SECTION</u>	<u>TITLE</u>	<u>PAGE</u>
4.17.6.7	GPS IMU Data Processing (GPS_IMU_SAVE)	89
4.17.6.8	GPS SOP Time Adjustment Uplink Processing (GPS_SOP_TIME_ADJ_UPLINK)	93
4.17.6.9	GPS SOP Time Adjustment (GPS_TIME_ADJ).....	95
4.17.6.10	GPS SOP (4.17) Principal Function Tables	127
4.43	TACAN SUBSYSTEM OPERATING PROGRAM.....	225
4.43.1	TACAN Requirements Overview	225
4.43.1.1	TACAN Functional Overview	225
4.43.1.1.1	TACAN Operating Modes.....	226
4.43.1.1.2	TACAN Digital Interfaces.....	227
4.43.1.1.3	TACAN Discrete Interfaces.....	227
4.43.1.2	TACAN SOP Functional Overview	227
4.43.1.3	TACAN Functional Module Summary.....	228
4.43.1.4	Reference Documents	228
4.43.1.5	Software Protection.....	228
4.43.2	Definitions.....	228
4.43.3	Functional Requirements Traceability	228
4.43.4	TACAN SOP Interfaces	228
4.43.5	Parameter Initialization.....	228
4.43.6	Detailed Requirements and Formulations.....	228
4.43.6.1	TACAN Self-Test Processing (TAC_TEST)	229
4.43.6.2	TACAN Data Processing (TAC_DATA).....	230
4.43.6.3	TACAN Antenna Selection Processing (TAC_ANT)	232
4.43.6.4	TACAN Channel Command Processing (TAC_CHAN).....	234
4.43.6.5	Applicable Tables for TACAN SOP.....	235
4.44	MICROWAVE SCAN BEAM LANDING SYSTEM SOP.....	271
4.44.1	MSBLS Requirements Overview.....	271
4.44.1.1	MSBLS Functional Overview.....	271

Table of Contents

<u>SECTION</u>	<u>TITLE</u>	<u>PAGE</u>
	4.44.1.1.1 MSBLS Operating Modes.	272
	4.44.1.1.2 MSBLS Digital Interfaces.	272
	4.44.1.1.3 Other MSBLS Interfaces.	273
	4.44.1.2 MSBLS SOP Overview.	273
	4.44.1.3 MSBLS SOP Functional Module Summary.	273
	4.44.1.4 Reference Documents.	273
	4.44.1.5 Software Protection.	273
	4.44.2 Definitions.	282
	4.44.3 Functional Requirements Traceability.	282
	4.44.4 Interface Parameters.	282
	4.44.5 Parameter Initialization.	282
	4.44.6 Detailed Requirements and Formulation.	282
	4.44.6.1 MSBLS Input Processing (MSB_IN).	282
	4.44.6.2 MSBLS Data Processing (MSB_DATA).	283
	4.44.6.3 Applicable Tables for MSBLS SOP.	285
4.45	RADAR ALTIMETER SUBSYSTEM OPERATING PROGRAM.	307
	4.45.1 Radar Altimeter Requirements Overview.	307
	4.45.1.1 Radar Altimeter Functional Overview.	307
	4.45.1.1.1 Radar Altimeter Operating Modes.	307
	4.45.1.1.2 Radar Altimeter Digital Interfaces.	308
	4.45.1.1.3 Radar Altimeter Discrete Interfaces.	308
	4.45.1.2 RA SOP Overview.	308
	4.45.1.3 Radar Altimeter SOP Functional Module Summary.	308
	4.45.1.4 Reference Documents.	308
	4.45.1.5 Software Protection.	308
	4.45.2 Definitions.	308
	4.45.3 Functional Requirements Traceability.	309
	4.45.4 Interface Parameters.	309

Table of Contents

<u>SECTION</u>	<u>TITLE</u>	<u>PAGE</u>
4.45.5	Radar Altimeter Initialization.....	309
4.45.6	Detailed Requirements and Formulations.....	309
4.45.6.1	RA Data Processing (RA_DATA).....	309
4.45.6.2	Applicable Tables for RA SOP.....	310
4.122	NAVIGATION-DERIVED AIR DATA SOP.....	323
4.122.1	Overview.....	323
4.122.1.1	Functional Requirements Traceability.....	323
4.122.2	NAVDAD SOP Detail Level Requirements.....	323
4.122.2.1	Navigation Derived Dynamic Pressure (DYPRS_N).	324
4.122.2.2	Navigation-Derived Equivalent Airspeed (EQAS_N).	326
4.122.2.3	Navigation-Derived Mach Number (MACH_N).....	326
4.122.2.4	Applicable Tables for NAVDAD SOP.....	327
4.229	RENDEZVOUS RADAR SOP.	343
4.229.1	Rendezvous Radar Requirements Overview.....	343
4.229.1.1	Rendezvous Radar Functional Overview.....	343
4.229.1.1.1	Rendezvous Radar Digital Interfaces.	343
4.229.1.2	Rendezvous Radar SOP Overview.....	344
4.229.1.3	Reference Documents.....	344
4.229.2	Definitions.....	344
4.229.3	Functional Requirements Traceability.	344
4.229.4	Interface Requirements.	344
4.229.5	Parameter Initialization.....	344
4.229.6	Detailed Requirements and Formulations.....	344
4.229.6.1	RR SOP.....	345
4.229.7	Applicable Tables for Rendezvous Radar SOP.....	348
4.253	Deleted.....	367
5.0	SUPPLEMENTARY MATERIAL.....	369

Table of Contents

<u>SECTION</u>	<u>TITLE</u>	<u>PAGE</u>
5.1	GLOSSARY	369
5.2	NONREQUIREMENT FLOW CHARTS/DIAGRAMS	379
APPENDIX R CHANGE AUTHORIZATION REFERENCES		381
R.1	CR REFERENCE SUMMARY	381
R.2	HISTORICAL CHANGE REQUEST SUMMARY.....	398

List of Tables

<u>TABLE</u>	<u>TITLE</u>	<u>PAGE</u>
1.3-1.	Parameter Element Description	2
4.0.1-1.	Level B to Level C Traceability	12
4.0.1-2.	SOP's FSSR/PRD Correlation.....	12
4.17.1.1.2-1.	GPC GPS I/O Data Format	19
4.17.1.1.2-2.	Message 1 Format - GPS Receiver Aiding Data	19
4.17.1.1.2-3.	Message 2 Format - Antenna Correction Data	21
4.17.1.1.2-4.	Message 3 Format - GPS Receiver Aiding data (Attitude and Altitude).....	22
4.17.1.1.2-5.	Message 4 Format - GPS Receiver Testing Data	23
4.17.1.1.2-6.	Message 5 Format - GPS Receiver Initialization Data	23
4.17.1.1.2-7.	Message 6 Format - GPS Receiver Navigation data.....	24
4.17.1.3-1.	GPS SOP Function Modules.....	25
4.17.6-1.	GPS SOP Control Input Parameters.....	32
4.17.6-2.	GPS SOP Control Output Parameters	33
4.17.6.1-1.	GPS_TO_GPC Input Parameters	36
4.17.6.1-2.	GPS_TO_GPC Output Parameters	37
4.17.6.1.1-1.	NAV_MODE_WORD_DATA Input Parameters	39
4.17.6.1.1-2.	NAV_MODE_WORD_DATA Output Parameters.....	40
4.17.6.1.1-3.	Format of NAV_MODE_WORD.....	40
4.17.6.1.2-1.	NAV_MESSAGE_DATA Input Parameters	45
4.17.6.1.2-2.	NAV_MESSAGE_DATA Output Parameters.....	46
4.17.6.1.2-3.	Format of GPS_STATUS_FOM_GDOP	47
4.17.6.1.2-4.	Definition of FOM.....	48

List of Tables

<u>TABLE</u>	<u>TITLE</u>	<u>PAGE</u>
4.17.6.1.2-5.	Format of GPS_TFOM.....	48
4.17.6.1.2-6.	Definition of TFOM	48
4.17.6.1.2.1-1.	CHANNEL_STATUS_DATA Input Parameters.....	50
4.17.6.1.2.1-2.	CHANNEL_STATUS_DATA Output Parameters.....	51
4.17.6.1.2.1-3.	Format of GPS Channel Status Word 1 (CHANNELS_STATUS(i,1))	51
4.17.6.1.2.1-4.	Format of GPS Channel Status Word 2 (CHANNELS_STATUS(i,2))	52
4.17.6.1.2.1-5.	Format of GPS Channel Status Word 3 (CHANNELS_STATUS(i,3))	52
4.17.6.1.2.1-6.	Format of GPS Channel Status Word 4 (CHANNELS_STATUS(i,4))	53
4.17.6.1.2.1-7.	Format of GPS Channel Status Word 5 (CHANNELS_STATUS(i,5))	54
4.17.6.1.2.1-8.	Format of GPS Channel Status Word 6 (CHANNELS_STATUS(i,6))	54
4.17.6.1.2.1-9.	Format of GPS Channel Status Word 7 (CHANNELS_STATUS(i,7))	55
4.17.6.1.2.1-10.	Format of GPS Channel Status Word 8 (CHANNELS_STATUS(i,8))	55
4.17.6.1.2.1-11.	Format of GPS Channel Status Word 9 (CHANNELS_STATUS(i,9))	55
4.17.6.1.3-1.	INIT_MESSAGE_DATA Input Parameters.....	56
4.17.6.1.3-2.	INIT_MESSAGE_DATA Output Parameters	56
4.17.6.1.3-3.	Definition of PPS_STATUS_WORD.....	57
4.17.6.1.4-1.	TEST_MESSAGE_DATA Input Parameters	58
4.17.6.1.4-2.	TEST_MESSAGE_DATA Output Parameters	59
4.17.6.2-1.	GPS_QA1_CHECK Input Parameters	60
4.17.6.2-2.	GPS_QA1_CHECK Output Parameters.....	60
4.17.6.3-1.	GPS_ALT_DATA Input Parameters.....	63
4.17.6.3-2.	GPS_ALT_DATA Output Parameters	64

List of Tables

<u>TABLE</u>	<u>TITLE</u>	<u>PAGE</u>
4.17.6.4-1.	GPC_TO_GPS Input Parameters	66
4.17.6.4-2.	GPC_TO_GPS Output Parameters	67
4.17.6.4.1-1.	RCVR_AIDING_DATA Input Parameters	73
4.17.6.4.1-2.	RCVR_AIDING_DATA Output Parameters.....	75
4.17.6.4.1-3.	Definition of GPS_CONTROL_WORD	76
4.17.6.4.2-1.	INIT_DATA Input Parameters	78
4.17.6.4.2-2.	INIT_DATA Output Parameters.....	79
4.17.6.4.2-3.	Definition of GPC_TO_GPS_DATA_TWO _{i,20} - Init Data Only.....	79
4.17.6.4.2-4.	Definition of GPC_TO_GPS_DATA_TWO _{i,21} - Init Data Only.....	79
4.17.6.4.3-1.	NAV_AIDING_DATA Input Parameters.....	81
4.17.6.4.3-2.	NAV_AIDING_DATA Output Parameters	81
4.17.6.5-1.	GPS_DISP_COMP Input Parameters	84
4.17.6.5-2.	GPS_DISP_COMP Output Parameters.....	85
4.17.6.5-3.	Definition of GPS_RCVR_STATUS _i word	85
4.17.6.6-1.	GPS_SOP_DL_COMP Input Parameters.....	87
4.17.6.6-2.	GPS_SOP_DL_COMP Output Parameters	87
4.17.6.6-3.	Definition of GPS_SOP_COLLECTION_WORD.....	88
4.17.6.6-4.	Definition of GPS_SOP_COLLECTION_WORD_TWO.....	88
4.17.6.7-1.	GPS_IMU_SAVE Input Parameters	91
4.17.6.7-2.	GPS_IMU_SAVE Output Parameters.....	92
4.17.6.8-1.	GPS SOP Time Adjustment Uplink Processing Input Parameters.....	93
4.17.6.8-2.	GPS SOP Time Adjustment Uplink Processing Output Parameters	94

List of Tables

<u>TABLE</u>	<u>TITLE</u>	<u>PAGE</u>
4.17.6.9-1	GPS SOP Time Adjustment Input Parameters	97
4.17.6.9-2	GPS SOP Time Adjustment Output Parameters.....	98
4.17.6.10-0.	GPS SOP Principal Function Variable Cross-Reference	127
4.17.6.10-1.	INPUT FUNCTIONAL PARAMETERS FOR GLOBAL POSITIONING SYSTEM SOP (G4.17).....	128
4.17.6.10-2.	OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17).....	133
4.17.6.10-3.	GLOBAL POSITIONING SYSTEM SOP (G4.17) I-LOADS	218
4.17.6.10-4.	GLOBAL POSITIONING SYSTEM SOP (G4.17) K-LOADS	223
4.17.6.10-5.	GLOBAL POSITIONING SYSTEM SOP (G4.17) CONSTANTS	224
4.43.1.3-1.	TACAN SOP Functional Modules	235
4.43.4-1.(1)	TACAN SOP Interface Table	236
4.43.4-1.(2)	TACAN SOP Interface Table	238
4.43.5-1.	TACAN SOP I-Load Parameters	242
4.43.6.5-1.	INPUT FUNCTIONAL PARAMETERS FOR TACTICAL AIR NAVIGATION SOP (G4.43)	254
4.43.6.5-2.	OUTPUT FUNCTIONAL PARAMETERS FROM TACTICAL AIR NAVIGATION SOP (G4.43)	257
4.43.6.5-3.	TACTICAL AIR NAVIGATION SOP (G4.43) I-LOADS.....	267
4.43.6.5-4.	TACTICAL AIR NAVIGATION SOP (G4.43) K-LOADS	268
4.43.6.5-5.	TACTICAL AIR NAVIGATION SOP (G4.43) CONSTANTS	269
4.44.1.3-1.	MSBLS SOP Functional Modules.....	274
4.44.4-1.	MSBLS SOP Interface Table.....	286

List of Tables

<u>TABLE</u>	<u>TITLE</u>	<u>PAGE</u>
4.44.5-1.	MSBLS SOP Conversion Parameters.....	293
4.44.5-2.	MSBLS SOP I-Load Parameters.....	293
4.44.6.3-1.	INPUT FUNCTIONAL PARAMETERS FOR MICROWAVE SCAN BEAM LANDING SYSTEM SOP (G4.44).....	298
4.44.6.3-2.	OUTPUT FUNCTIONAL PARAMETERS FROM MICROWAVE SCAN BEAM LANDING SYSTEM SOP (G4.44).....	300
4.44.6.3-3.	MICROWAVE SCAN BEAM LANDING SYSTEM SOP (G4.44) I-LOADS	303
4.44.6.3-4.	MICROWAVE SCAN BEAM LANDING SYSTEM SOP (G4.44) K-LOADS.....	304
4.44.6.3-5.	MICROWAVE SCAN BEAM LANDING SYSTEM SOP (G4.44) CONSTANTS.....	305
4.45.1.3-1.	RA SOP Functional Modules	310
4.45.4-1.	Radar Altimeter Interface Table	311
4.45.6.2-1.	INPUT FUNCTIONAL PARAMETERS FOR RADAR ALTIMETER SUBSYSTEM OPERATING PROGRAM (G4.45).....	317
4.45.6.2-2.	OUTPUT FUNCTIONAL PARAMETERS FROM RADAR ALTIMETER SUBSYSTEM OPERATING PROGRAM (G4.45).....	318
4.45.6.2-3.	RADAR ALTIMETER SUBSYSTEM OPERATING PROGRAM (G4.45) I-LOADS	319
4.45.6.2-4.	RADAR ALTIMETER SUBSYSTEM OPERATING PROGRAM (G4.45) K-LOADS.....	320
4.45.6.2-5.	RADAR ALTIMETER SUBSYSTEM OPERATING PROGRAM (G4.45) CONSTANTS.....	321

List of Tables

<u>TABLE</u>	<u>TITLE</u>	<u>PAGE</u>
4.122-1.	NAVDAD SOP Inputs.....	328
4.122-2.	NAVDAD SOP Outputs.....	328
4.122-3.	NAVDAD SOP Initialization Parameters	329
4.122-4.	NAVDAD SOP I-Load Parameters	330
4.122.2.4-1.	INPUT FUNCTIONAL PARAMETERS FOR NAVIGATIONALY- DERIVED AIR DATA SOP (G4.122).....	337
4.122.2.4-2.	OUTPUT FUNCTIONAL PARAMETERS FROM NAVIGATIONALY-DERIVED AIR DATA SOP (G4.122).....	338
4.122.2.4-3.	NAVIGATIONALY-DERIVED AIR DATA SOP (G4.122) I-LOADS	339
4.122.2.4-4.	NAVIGATIONALY-DERIVED AIR DATA SOP (G4.122) K-LOADS	340
4.122.2.4-5.	NAVIGATIONALY-DERIVED AIR DATA SOP (G4.122) CONSTANTS.....	341
4.229.4-1.	Rendezvous Radar SOP Interface Table	349
4.229.7-1.	INPUT FUNCTIONAL PARAMETERS FOR RENDEZVOUS RADAR SUBSYSTEM OPERATING PROGRAM (G4.229)	361
4.229.7-2.	OUTPUT FUNCTIONAL PARAMETERS FROM RENDEZVOUS RADAR SUBSYSTEM OPERATING PROGRAM (G4.229)	362
4.229.7-3.	RENDEZVOUS RADAR SUBSYSTEM OPERATING PROGRAM (G4.229) I-LOADS	363
4.229.7-4.	RENDEZVOUS RADAR SUBSYSTEM OPERATING PROGRAM (G4.229) K-LOADS	364
4.229.7-5.	RENDEZVOUS RADAR SUBSYSTEM OPERATING PROGRAM (G4.229) CONSTANTS	365
R.1-1.	CR Reference Summary.....	381

List of Tables

<u>TABLE</u>	<u>TITLE</u>	<u>PAGE</u>
R.2-1.	Historical Change Request Summary	398

This page intentionally left blank.

List of Figures

<u>FIGURE</u>	<u>TITLE</u>	<u>PAGE</u>
4.17.1.1-1.	GPS Subsystem Block Diagram.....	99
4.17.6-1.(1)	GPS_SOP (Sheet 1 of 3).....	100
4.17.6-1.(2)	GPS_SOP (Sheet 2 of 3).....	101
4.17.6-1.(3)	GPS_SOP (Sheet 3 of 3).....	102
4.17.6.1.	GPS_TO_GPC CODE	103
4.17.6.1.1.(1)	NAV_MODE_WORD_DATA (Sheet 1 of 2).....	104
4.17.6.1.1.(2)	NAV_MODE_WORD_DATA (Sheet 2 of 2).....	105
4.17.6.1.2.(1)	NAV_MESSAGE_DATA CODE (Sheet 1 of 2)	106
4.17.6.1.2.(2)	NAV_MESSAGE_DATA CODE (Sheet 2 of 2)	107
4.17.6.1.2.1.	CHANNEL_STATUS_DATA CODE.....	108
4.17.6.1.3.	INIT_MESSAGE_DATA CODE	109
4.17.6.1.4.	TEST_MESSAGE_DATA CODE.....	110
4.17.6.2.	GPS_QA1_CHECK CODE.....	111
4.17.6.3.	GPS_ALT_DATA CODE	112
4.17.6.4.	GPC_TO_GPS CODE	113
4.17.6.4.1.(1)	RCVR_AIDING_DATA CODE (Sheet 1 of 4)	114
4.17.6.4.1.(2)	RCVR_AIDING_DATA CODE (Sheet 2 of 4)	115
4.17.6.4.1.(3)	RCVR_AIDING_DATA CODE (Sheet 3 of 4)	116
4.17.6.4.1.(4)	RCVR_AIDING_DATA CODE (Sheet 4 of 4)	117
4.17.6.4.2.	INIT_DATA CODE.....	118
4.17.6.4.3.	NAV_AIDING_DATA CODE	119

List of Figures

<u>FIGURE</u>	<u>TITLE</u>	<u>PAGE</u>
4.17.6.5.(1)	GPS_DISP_COMP CODE (Sheet 1 of 2)	120
4.17.6.5.(2)	GPS_DISP_COMP CODE (Sheet 2 of 2)	121
4.17.6.6.	GPS_SOP_DL_COMP CODE	122
4.17.6.7.(1)	GPS_IMU_SAVE CODE (Sheet 1 of 2)	123
4.17.6.7.(2)	GPS_IMU_SAVE CODE (Sheet 2 of 2)	124
4.17.6.8.	GPS_SOP_TIME_ADJ_UPLINK	125
4.17.6.9	GPS_TIME_ADJ	126
4.43.1.1-1.	TACAN Subsystem Block Diagram	243
4.43.1.1.2-1.	TACAN Control Register Format	244
4.43.1.1.2-2.	TACAN Discrete Word Format	244
4.43.1.1.2-3.	TACAN-to-MDM Range Word	245
4.43.1.1.2-4.	MDM-to-GPC Range Words	246
4.43.1.1.2-5.	TACAN-to-MDM Bearing Word	247
4.43.1.1.2-6.	MDM-to-GPC Bearing Word	248
4.43.6-1.	TACAN SOP	249
4.43.6.1-1.	TAC_TEST Flow Diagram	250
4.43.6.2-1.	TAC_DATA Flow Diagram	251
4.43.6.3-1.	TAC_ANT Flow Diagram	252
4.43.6.4-1.	TAC_CHAN Flow Diagram	253
4.44.1.1-1.	Microwave Scanning Beam Landing System, Single-String Diagram	275
4.44.1.1.2-1.	MSBLS Serial Data Words (MSBLS to MDM)	276

List of Figures

<u>FIGURE</u>	<u>TITLE</u>	<u>PAGE</u>
4.44.1.1.2-2.	MSBLS Azimuth Angle Word Format (MDM to GPC).....	277
4.44.1.1.2-3.	MSBLS Elevation Angle Word Format (MDM to GPC).....	278
4.44.1.1.2-4.	MSBLS Range Word Format (MDM to GPC)	279
4.44.1.2-1.	MSBLS Data Signal Flow	281
4.44.6.1-1.	MSB_IN Flow Diagram.....	294
4.44.6.2-1.(1)	MSB_DATA Flow Diagram (Sheet 1 of 3)	295
4.44.6.2-1.(2)	MSB_DATA Flow Diagram (Sheet 2 of 3)	296
4.44.6.2-1.(3)	MSB_DATA Flow Diagram (Sheet 3 of 3)	297
4.45.1.1-1.	Radar Altimeter Subsystem Diagram	312
4.45.1.1.2-1.	RA Altitude Data Word (From RA to MDM).....	313
4.45.1.1.2-2.	RA Altitude Data Word (From MDM to GPC)	314
4.45.1.2-1.	Radar Altimeter Data Signal Flow	315
4.45.6.1-1.	RA_DATA Flow Diagram.....	316
4.122-1.	NAVDAD SOP Funtional Block Diagram.....	331
4.122-2.	NAVDAD_SOP Functional Module.....	332
4.122-3.	DYPRS_N Functional Module.....	333
4.122-4.	EQAS_N Functional Module	334
4.122-5.	MACH_N Functional Module.....	335
4.229.1.1-1.	RR Subsystem Block Diagram.....	352
4.229.1.1.1-1.	RR Range Words.....	353
4.229.1.1.1-2.	RR Roll Angle Word	353

List of Figures

<u>FIGURE</u>	<u>TITLE</u>	<u>PAGE</u>
4.229.1.1.1-3.	RR Pitch Angle Word	354
4.229.1.1.1-4.	RR Range Rate Words.....	354
4.229.1.1.1-5.	RR Roll Angle Rate Word	355
4.229.1.1.1-6.	RR Pitch Angle Rate Word	355
4.229.1.1.1-7.	RR Status Word 1	356
4.229.1.1.1-8.	RR Status Word 2	356
4.229.1.2-1.	RR SOP Signal Flow Diagram.....	357
4.229.6.1-1.(1)	RR SOP Flow Diagram (Sheet 1 of 3).....	358
4.229.6.1-1.(2)	RR SOP Flow Diagram (Sheet 2 of 3).....	359
4.229.6.1-1.(3)	RR SOP Flow Diagram (Sheet 3 of 3).....	360
5.2-1.	Major Modes and Transitions	380

1.0 INTRODUCTION

1.1 PURPOSE

The purpose of the subsystem operating program (SOP) Level C document is to specify the requirement details and to present formulations of the SOP Level B functional requirements for the orbiter GN&C flight software.

1.2 SCOPE

This document contains the requirement details and formulations for those SOP functions that are operative during Shuttle operations. The SOP functions are those specified in the GN&C Level B functional requirements document. This particular document addresses the navigation aid SOP's.

1.3 ORGANIZATION

This document is organized into the following sections:

- 1.0 Introduction
- 2.0 Applicable documents
- 3.0 Overview
- 4.0 Detail level requirements
- 5.0 Supplemental information

Section 1 defines the purpose, scope, and organization of this document. Section 2 lists the applicable documents. Section 3 provides an overview of the OFT software functions and discusses the contents of Section 4. Section 4 specifies the requirement details and presents formulation of the functional requirements for each subsystem operating program. It also contains applicable tables (Interface Definition Document (IDD) inputs/outputs, I-Loads, K-Loads, and Constants) located at the end of each Principal Function. Section 5 provides supplementary material.

The automated IDD tables provided herein define signal (parameter) interface requirements either between hardware (LRU) and software elements or between software elements and other software elements. A GN&C software element is either a sequenced principal function (PF), crew generated specialist/display function, or an OPS display function. In the event of a conflict between the IDD tables and other internal text input/output tables, the SASCB data base-controlled IDD tables take precedence.

The following table defines the data elements found in the IDD input/output, I-Load, K-Load, and Constant tables for each principal function.

Table 1.3-1. Parameter Element Description		
Title	Definition	Valid Entries
CAT	Category and Qualifier of the I-Load	Four alphanumeric characters: the first three designate the I-Load category; the last one designates the qualifier.
D	Dependency of the I-Load	D = Design M = Mission L = LRU
DESCRIPTION	Parameter description of the K-Load or Constant.	A string of ≤ 64 characters.
DESTINATION	Destination of the output parameter	Abbreviated functional titles. (** Note)
DT or DATA TYPE	Data Type	* See reference (1) for valid I-Load, K-Load and Constant data types. * See reference (2) for valid IDD data types for parameters with destination TLM. * See reference (3) for valid IDD data types for all other parameters.
ENG UNITS or UNITS	Engineering Units	* See reference (1) for valid I-Load, K-Load and Constant units. * See reference (2) for valid IDD units for parameters with destination TLM. * See reference (3) for valid IDD units for all other parameters.
FSSR NAME	Requirements parameter name	A string (with no embedded blanks) of ≤ 32 characters consisting of alphanumerics, underscores, parenthesis, commas and slashes. Parenthesis are used for array elements. Commas separate the different indices of multidimensional arrays. Slashes are used to separate multiple names used for a single MSID.
LAST CR	The last CR affecting the parameter	5 digit number with an alphabetic revision letter.
MC	Memory Configuration	00=Generic 05 = SM4 Orbit P/L 01=GN1 Ascent/GN6 RTLS 06 = P/L9 MMU 02=GN2 Orbit 07 = Not Assigned 03=GN3 Entry 08 = GN8 On-Orbit C/O 04=SM2 On-Orbit 09 = GN9 Preflight
MSID or M/S ID	Measurement/Stimulus Identification	Maximum of 10 alphanumeric characters. * See reference (3) for specifications
NOMENCLATURE	Parameter description of the IDD parameter.	String of ≤ 34 characters (including underscores and blank spaces).
PR or PREC	Precision	D, DP =Double S, SP = Single
SOURCE	Origin of the input parameter	Abbreviated functional titles.

Table 1.3-1. Parameter Element Description		
Title	Definition	Valid Entries
S	Software Utilization	C = Common B = BFS Unique P = Primary X = BFS Derived A = BFS Converted
PR FCTN	Cross reference of the principal function identifiers that are applicable to this parameter.	See Level B GN&C CPDS SS-P-0002-510
VALUE	The value of the K-Load or Constant	* See reference (1) for valid value formats.
* References:	(1) JSC 19350 I-Load Requirements Document Volume I (2) SS-P-0002-140 Level A SS Downlist/Uplink CPDS (3) JSC 18206 Shuttle Data Integration Plan Volume III MAST II/STAR Deliverable Products.	
** Note	IDD output (source) tables which reflect parameters going to input (destination) "TLM" shall not interpreted as a signal actually being downlisted. The parameter is only available for downlist in COMPOOL and does not appear in a downlist format. Parameters actually on the downlist will be designated with a destination denoting the downlist Format with a Rate Code. The Rate Code legend appears at the end of each FSSR data table. To complete the downlist information, Parent MSIDs (indicated by a 'P' in the 9 th character of the MSID) are now included in the software function output data tables in which a Child MSID already appears. A Parent MSID will have the same downlist destinations as its Child(ren) MSID(s).	

This page intentionally left blank.

2.0 APPLICABLE DOCUMENTS

2.1 LEVEL A DOCUMENTS

- SS-P-0002-140 Shuttle Downlist/Uplink Software Requirements
- SS-P-0002-150 Shuttle Launch Data Bus Software Interface Requirements
- SS-P-0002-170 Shuttle Systems Level Requirements, Software

2.2 LEVEL B DOCUMENTS

- SS-P-0002-510 Shuttle Functional Level Requirements, GN&C
- SS-P-0002-550 Shuttle Functional Level Requirements, Vehicle Utility
- SS-P-0002-580 Shuttle Functional Level Requirements, Systems Management

2.3 LEVEL C DOCUMENTS

- STS 83-0001 Operational Flight Level C, Functional Subsystem Software Requirements; Guidance, Navigation, and Control, Part A, Entry Through Landing Guidance
- STS 83-0002 Operational Flight Level C, Functional Subsystem Software Requirements; Guidance, Navigation, and Control, Part A, Guidance Ascent/RTLS
- STS 83-0003 Operational Flight Level C, Functional Subsystem Software Requirements; Guidance, Navigation, and Control, Part A, Guidance On-Orbit/Deorbit
- STS 83-0004 Operational Flight Level C, Functional Subsystem Software Requirements; Guidance, Navigation, and Control, Part B, Entry Through Landing Navigation
- STS 83-0005 Operational Flight Level C, Functional Subsystem Software Requirements; Guidance, Navigation, and Control, Part B, Navigation Ascent/RTLS
- STS 83-0006 Operational Flight Level C, Functional Subsystem Software Requirements; Guidance, Navigation, and Control, Part B, On-Orbit Navigation
- STS 83-0007 Operational Flight Level C, Functional Subsystem Software Requirements; Guidance, Navigation, and Control, Part C, Flight Control Entry – GRTLS
- STS 83-0008 Operational Flight Level C, Functional Subsystem Software Requirements; Guidance, Navigation, and Control, Part C, Flight Control Volume 1, Ascent Flight Phase, Volume 2, Ascent
- STS 83-0009 Operational Flight Level C, Functional Subsystem Software Requirements; Guidance, Navigation, and Control, Part C, Flight Control Orbit DAP
- STS 83-0010 Operational Flight Level C, Functional Subsystem Software Requirements; Guidance, Navigation, and Control, Part D, Redundancy Management

STS 83-0014V1-34
December 14, 2007

STS 83-0013	Operational Flight Level C, Functional Subsystem Software Requirements; Guidance, Navigation, and Control, Part E, Inertial Measurement Unit Subsystem Operating Program
STS 83-0014	Operational Flight Level C, Functional Subsystem Software Requirements; Guidance, Navigation, and Control, Part E, Volume 1, Navigation Aids Subsystem Operating Program, Volume 2, Star Tracker Subsystem Operating Program
STS 83-0015	Operational Flight Level C, Functional Subsystem Software Requirements; Guidance, Navigation, and Control, Part E, Subsystem Operating Programs, FC Sensor/Controller
STS 83-0016	Operational Flight Level C, Functional Subsystem Software Requirements; Guidance, Navigation, and Control, Part E, Subsystem Operating Programs, FC Effector
STS 87-0017	Operational Flight Level C, Functional Subsystem Software Requirements; Remote Manipulator System
STS 83-0020	Operational Flight Level C, Functional Subsystem Software Requirements; Displays and Controls
STS 83-0026	Operational Flight Level C, Functional Subsystem Software Requirements; Guidance, Navigation, and Control, Sequencing
JSC-19350	Shuttle Flight Software Initialization Loads <ul style="list-style-type: none">• Volume I General Requirements• Volume II Mission I-Load Requirements
JSC-19478	Payload Management, Level C Flight Software Requirements
JSC-19590	Systems Management, Level C Flight Software Requirements

2.4 ICD DOCUMENTS

ICD 3-1011-02	GPC/DEU ICD
ICD 2-1S001	Orbiter Primary Flight Software System and Backup Flight System Software Interface

In the event of a conflict between the documents referenced herein and the contents of this specification, the precedence shall be determined as defined in paragraph 1.3 of CPDS SS-P-0002-170, Volume 1, System Level Requirements, Software.

3.0 OVERVIEW

3.1 SHUTTLE OPERATIONAL PROGRAM

The Shuttle operational program represents the total computer program which operates in the orbiter general-purpose flight computer for Shuttle operations. The requirements are specified at three levels:

- Level A – System level constraints and capabilities
- Level B – Functional requirements
- Level C – Detail level requirements

The Level A specifies system-level constraints and capabilities which are not oriented toward any particular program end item. Levels B and C specify the functional- and detailed-level requirements and are program end item oriented. The Level B and C requirements are specified in separate documents for guidance, navigation, and control (GN&C); systems management (SM); and vehicle utility and data flow.

3.2 GN&C SOFTWARE REQUIREMENTS STRUCTURE

This section describes the top level GN&C software requirements structure for purposes of specifying the functional requirements for Shuttle operations. The structure is described below.

1. Operational Sequence – A collection of sequentially ordered major modes.
2. Major Modes – A collection of functions that may be scheduled by the software or that may be called by the crew. (See Figure 5.2-1.)
3. Function – A task to be performed by the software. Functions are defined as:
 - a. Principal function – A function that can only be initiated by the software. Principal functions may be ordered sequentially, run in parallel, enabled, or inhibited.
 - b. Specialist function – A function that can only be initiated by the crew.

3.3 SOP OVERVIEW

The SOP major functions consist of principal and specialist functions (as defined above). The SOP's basically provide services and interface between the GN&C LRU's and other application programs. The nature of the services and interfaces provided is indicated by the general SOP functions described below:

1. Format conversion to support input/output processing
2. Data compensation to convert data into engineering units by applying biases and scale factors
3. Data transformations and special computations as required by other application programs
4. Generation of appropriate data and interfaces to display and control (D&C)

STS 83-0014V1-34
December 14, 2007

5. BITE and failure discrete monitoring and reporting of status to redundancy management (RM) for further processing
6. Self-test execution as required
7. Moding, sequencing, and control at the individual SOP level

The SOP's do not include the specific input/output services provided by FCOS, nor do they include the data selection and failure management functions performed by other application programs.

As indicated earlier, the Level C SOP detailed requirements are specified in several documents. The following list indicates the SOP's addressed in each volume.

1. STS 83-0013 – Inertial measurement unit SOP
2. STS 83-0014 – Navigation aid SOP's, Vol. 1
 - a. Tactical air navigation SOP (TACAN)
 - b. Microwave scan beam landing system SOP (MSBLS)
 - c. Radar altimeter SOP (RA)
 - d. Navigation derived air data SOP (NAVDAD)
 - e. Rendezvous radar SOP (RR)
3. STS 83-0014 – Navigation aid SOP's, Vol. 2 Star Tracker
4. STS 83-0015 – Flight control sensor/controller SOP's
5. STS 83-0016 – Flight control effector SOP's

3.4 SOP SUBSECTION ORGANIZATION

Section 4 of this document contains the SOP requirement details and formulation. Each subsection in Section 4 is dedicated to a particular SOP. The contents of the paragraphs for a typical SOP subsection is discussed below.

A. Requirements Overview. This paragraph provides an introduction to the particular SOP. A functional overview of the applicable LRU performance characteristics, operating modes, and interfaces is presented for the purpose of defining hardware dictated requirements and constraints. An overview of the SOP is presented to indicate functions the SOP provides and its relationship to other application programs.

B. Definitions. This paragraph defines applicable coordinate systems, transformations and other supporting data that are applicable to the SOP.

C. Requirements Traceability. This paragraph contains a traceability table that correlates the detailed requirements of the SOP with the top-level functional requirements specified in the GN&C Level B CPDS.

D. Interface Parameters. This paragraph is reserved for specifying interfaces associated with the applicable SOP.

E. Parameter Initialization. This paragraph defines those parameters that require initialization and specifies their associated values. It does not include initialization processing that requires execution.

F. Detailed Requirements and Formulations. This paragraph defines the functional modules that constitute the applicable SOP. The detailed requirements and formulations are contained in this paragraph. The format and contents of each module consist of the following:

1. Name and purpose of the functional module
2. Functional requirements satisfied by the module
3. Input and output, interfaces, and I-loads that are associated with the module
4. Detailed requirements and formulations including execution rates and processing sequence
5. Special considerations that are applicable to the module
6. Tables – Inputs, outputs, I-loads
7. Flow diagrams

Subparagraphs are generally used to specify the detailed requirements of subfunctions. For each SOP, the organization of the subparagraphs is tailored to individual needs and peculiarities.

3.5 SOP FORMULATION NOTES

This section provides definitive notes for the SOP formulations contained in Section 4.

A. Computation and Coding Considerations. It should be noted that all navigation and communication aid SOP computations can be performed using single precision coding, except where double precision is specified to obtain the required accuracy. Fixed-point/floating-point conversions and computations can be implemented in the manner that most efficiently satisfies the SOP requirements.

B. Programming Deviations. It is recognized that, for the sake of programming efficiency, the detailed design and coding of the SOP software may deviate from the partitioning of functional requirements presented in the SOP functional modules. This deviation, where necessary, is acceptable provided the overall detailed design and coded program comply with the basic intent of the SOP.

3.6 IMPLEMENTATION CONSTRAINTS

The SOP design is capable of recovering from the effects of many transient-type errors. Accordingly, the software design and code shall not preclude this capability by causing a program halt, permanent discontinuance of SOP processing, or loss of protected data as a result of program check or error interrupts. Check or error interrupts shall either be prevented by software safeguards or standard fixup, and error returns shall be provided that are appropriate to the SOP function as described in Level A CPDS, Section 4.6.4. The footnotes specified in the SOP's (F1, F2, F3, etc.) refer to codes described in the Level A CPDS.

3.6.1 Timing

Timing in this document is related to error-free processing conditions. See Level A CPDS SS-P-0002-170, paragraph 4.4.2A, for timing related to error processing conditions.

4.0 DETAIL LEVEL REQUIREMENTS

The subsystem operating programs (SOP's) include all of the software required for operation of the GN&C subsystem hardware external to the data processing subsystem (DPS).

The SOP's provide necessary services and interfaces between the GN&C line replaceable units (LRU's) and other application programs. The major SOP functions are format conversion, data compensation, moding and sequencing, BITE monitoring, self-tests, and special computations. The SOP's do not include the specific input/output services provided by FCOS, nor do they include the data selection and failure management functions performed by other application programs.

This section specifies the detailed requirements for the navigation aid SOP's which are organized as follows.

Section	Title
4.17	Global Positioning System SOP
4.43	TACAN SOP
4.44	Microwave Scan Beam Landing System SOP
4.45	Radar Altimeter SOP
4.122	Navigation-Derived Air Data SOP
4.229	Rendezvous Radar SOP

Of the navigation aid SOP's listed in previous versions of this document, two are not in this book (Vol. 1): The One-Way Doppler Extractor (OWDE) SOP has been deleted since it will not be used on the Shuttle Program; the Star Tracker SOP is being issued as a separate book (Vol. 2).

The input/output functional parameters for the SOP modules, I-loads/K-loads, constants, and variable cross-reference are provided at the end of each section.

Requirements precedence in this document shall be as follows:

1. If there is a conflict between the data in the function subtables and the standard principal function tables, the principal function tables shall have precedence.
2. If there is a conflict between the data in the flow diagrams and the written requirements, the written requirements shall have precedence.

4.0.1 Traceability of Requirements

Table 4.0.1-1 shows the traceability of the software paragraphs in this FSSR with respect to the corresponding Sections of the Level B CPDS' documentation. In addition, Table 4.0.1-2 shows the traceability of the PASS GN&C principal functions located in this FSSR with respect to the applicable BFS paragraphs in the Level C PRD specifications. These tables are provided for information only.

Table 4.0.1-1.Level B to Level C Traceability			
Level B PF Number	Level B PF Name (Acronym)	Level C FSSR Section No.	Related Modules
4.17	Global Positioning System Subsystem Operating Program (GPS SOP)	4.17.6.1 4.17.6.2 4.17.6.3 4.17.6.4 4.17.6.5 4.17.6.6 4.17.6.7 4.17.6.8	GPS_TO_GPC GPS_QAI_CHECK GPS_ALT_DATA GPC_TO_GPS GPS_DISP_COMP GPS_SOP_DL_COMP GPS_IMU_SAVE GPS_SOP_TIME_ADJ_UPLINK
4.43	TACAN Subsystem Operating Program (TACAN SOP)	4.43.6.1 4.43.6.2 4.43.6.3 4.43.6.4	TAC_TEST TAC_DATA TAC_ANT TAC_CHAN
4.44	Microwave Scanbeam Landing System SOP (MSBLS SOP)	4.44.6.1 4.44.6.2	MSB_IN MSB_DATA
4.45	Radar Altimeter SOP (RA SOP)	4.45.6.1	RA_DATA
4.122	Navigation-Derived Air Data SOP (NAV DAD SOP)	4.122.2.1 4.122.2.2 4.122.2.3	DYPRS_N EOAS_N MACH_N
4.229	Rendezvous Radar SOP (RR SOP)	4.229.6.1	RR SOP

Table 4.0.1-2.SOP's FSSR/PRD Correlation					
Principal Function	FSSR Paragraph	FSSR Title	BFS PRD	PRD Paragraph	BFS Title
G4.17	4.17	Global Positioning System Subsystem Operating Program (GPS SOP)	MG038106	3.2.5.2.4	Global Positioning System Hardware Interface Program (GPS_HIP)
G4.43	4.43	TACAN Subsystem Operating Program	MG038106	3.2.5.2.3	TACAN Hardware Interface Processor (TACAN_HIP)
G4.44	4.44	Microwave Scanbeam Landing System SOP	N/A	N/A	N/A
G4.45	4.45	Radar Altimeter SOP	N/A	N/A	N/A
G4.122	4.122	Navigation-Derived Air Data SOP	MG038106	3.2.5.2.2	Navigation Derived Air Data Processor (NAVDAD_PROC)
G4.229	4.229	Rendezvous Radar SOP	N/A	N/A	N/A

4.17 Global Positioning System Subsystem Operating Program

This section defines the detailed functional requirements and formulations for the Global Positioning System (GPS) subsystem operating program (SOP). The GPS SOP defines software associated with GPS receiver command and response processing, GPS data transmittal to the GPC and associated processing, and GPC data transmittal to the GPS receiver and associated processing.

All GPS related software in the GPC is required to handle phased GPS receiver installation in the vehicle in any LRU assignment order and any number of receivers actually installed from 0 to a maximum of 3 except for Horizontal Situation display. This maximum is defined by the K-load parameter, GPS_SW_CAP.

The implementation of this principal function satisfies this requirement by processing data from 3 receivers and bypassing relevant calculations if a receiver is not actually installed on the vehicle.

4.17.1 GPS Requirements Overview

4.17.1.1 GPS Functional Overview

GPS is a space-based radio navigation system which utilizes a constellation of 24 satellites (3 spares) to provide global and continuous coverage for near earth spacecraft. GPS satellite ephemeris data, ranging codes, and time data are provided to the user vehicle receiver (or receivers) which use this data to compute highly accurate measurements of the vehicle range and Doppler from each of the GPS satellites being tracked. This information is processed in the receiver to derive the position and velocity of the user vehicle in the World Geodetic Survey 1984 (WGS84) earth-centered earth-fixed coordinate frame. The implementation of a GPS-based navigation capability on the Orbiter will consist of the following procedures:

1. Output the estimate of the Orbiter's position, velocity, and associated time tag from the GPS receiver(s) to the Orbiter GPC.
2. Output the required status of each receiver and the quality of its state estimates from the GPS receivers(s) to the Orbiter GPC.
3. Perform the appropriate coordinate transformations on and quality checks of these receiver states.
4. Select the GPS receiver state to be used by the Orbiter navigation system.
5. Provide aiding states (position, velocity, attitude and altitude) to the GPS receivers.

A single GPS subsystem is depicted in Figure 4.17.1.1-1. The major components that are required for the GPS function are identified as follows: controls on the panel, an MDM (multiplexer/demultiplexer) that contains the GPS control register, a GPS receiver, a GPS antenna and the on-board computer (GPC). The keyboard and the display electronics unit are also required for GPS receiver command and data monitoring.

4.17.1.1.1 GPS Operating Modes

The software requirements for the GPS SOP are based on the following GPS receiver operations and are provided here for information. They are also reproduced in the Display FSSR, STS-83-0020 Volume I, in the GPS Status Display section.

1. The GPS receiver operating system modes are as follows:

- OFF mode
- INIT mode
- TEST mode
- NAV mode

The GPS receiver OFF mode is controlled by a Power On Panel Switch and cannot be commanded by the crew through the GPC GPS Status Display (G6.51).

The INIT, TEST and NAV modes can be commanded by the crew through the GPC GPS Status Display (G6.51).

2. When the receiver is powered up while it is not under GPC control, the receiver will automatically mode to Power-On Self Test, to INIT, and then to NAV.

Upon power up under GPC control, the GPC receiver will perform Power-On Self Test, transition to the INIT mode, and remain in INIT until commanded to TEST or NAV.

The receiver is under GPC control when it receives valid data with a valid header from the GPC.

3. TEST mode can be commanded by the crew through execution of a display item entry on the GPS Status Display. In TEST mode, the GPS receiver performs full commanded self test using the GPS R/P Built-In-Test (BIT) capability for fault isolation and identification. After the first 6 seconds in TEST mode, the receiver will mode itself to BIT during which period (about 2 minutes) there is no output from the receiver. All GPS receiver data, including the data header, received by the GPC will be zero. When BIT is complete, the receiver will again output valid header and data (BIT FAULT LOG WORDS). The result of the self-test (pass or fail) will be in the fifth bit of word 1 in the Bit Fault Log and will be available for display to the crew. Upon completion of the commanded BIT, the receiver will remain in the TEST mode and will not automatically transition to INIT or NAV. If the receiver is commanded to INIT or NAV during the first 6 seconds of TEST mode, the receiver will not complete the commanded BIT and will transition to the new commanded mode immediately.

Note that the commanded selftest, BIT, is different than the Power-On Self Test in Item 2 above.

4. INIT mode can be commanded by the crew through execution of a display item entry on the GPS Status Display. Initialization operations occur when the receiver is in the INIT mode. The initialization mode allows the timing and frequency standard to attain stable operation as required to activate the digital interfaces with the Orbiter software. Data sent from the GPC to

GPS is used to initialize the receiver position and velocity state. Lever arm correction data and GPS measurement residual edit criteria are also collected by the receiver via the GPC to GPS data. Upon completion of INIT mode, the receiver will remain in INIT and will not automatically transition to NAV, except as indicated in (2).

5. When the receiver is commanded to INIT by the crew or as a result of moding the GPS I/O control to flight GPS I/O in G9, initialization data will be provided to the receiver and the initialization data available flag (Bit 8 of GPS_CONTROL_WORD) will be set ON at the same time the receiver is commanded to INIT.

On receipt of the INIT command from the GPC, the receiver will use the accompanying state information to perform INIT. On completion of INIT, the receiver will set the initialization complete flag to ON (Bit 17 of NAV_MODE_WORD) and sequence to the next commanded mode. Note that if the initialization data available flag is set to ON without the GPC also commanding the INIT mode, the receiver will interpret that data block from the GPC as invalid and the aiding data will not be used for initialization.

When the initialization data is sent to the receiver, the initialization data available flag and the aiding data available flag will be set to ON, and the announced reset, and filter restart, flags will be set to OFF.

6. NAV mode can be commanded by the crew through execution of a display item entry on the GPS Status Display. NAV mode occurs when the GPS receiver is in satellite tracking and navigation operation, providing position and velocity.

There are three navigation submodes: aided (INS), unaided (PVA), and Prelaunch Standby (STA).

The receiver will be in the Prelaunch Standby NAV submode when it is commanded to the NAV mode in: G9 prior to launch; in OPS 1 prior to Major Mode 102; and following transition to G9 after landing. In this nav submode, the receiver will be aided by the GPC with a position vector, a zero velocity vector and attitude aiding data. In the PASS GPS software requirements, the Prelaunch Standby submode is also referred to as the Stationary submode since it is a nav submode which can occur both prelaunch and post landing.

The receiver will be in either the INS or PVA nav submode when it is commanded to NAV in all other OPS. The INS submode occurs when there are valid aiding states provided to the receiver by the GPC. When there are no valid aiding states being provided, the receiver downmodes to the PVA nav submode. Determination of aiding depends on: rate of aiding (must be between 4 and 100 Hertz); senescence of data (< 2 seconds old); validity of data; and reasonableness checks on the aiding data. The receiver will transition from the PVA to INS NAV submodes when it receives valid aiding data over four GPC to GPS input cycles in a one second period.

7. During aided NAV submode operation, the GPS receiver navigation filter is aided by the Orbiter GPC IMU data in the form of a reference state which is computed by the Reference State Processing Principal Function (4.18). Two commands from the GPC, the first of which is controlled by the crew through GPS Status Display item entry, are provided to communicate with the GPS receiver navigation filter.

a. Filter Restart: Commands the receiver to reinitialize its state to the GPC reference state and restart its navigation filter. Upon detecting a change from an OFF to an ON state for this command, the receiver performs the necessary internal actions and sends a completion bit to the GPS SOP. The command is then turned OFF by the SOP when the GPC is notified of the task completion by the receiver. The receiver then receives the OFF command and resets its internal parameters to wait for the next OFF to ON transition for the command. Filter restart commands will not be honored by the receiver when it is in the Prelaunch Standby or PVA NAV submodes. It will be honored in the INS NAV submode.

b. Announced Reset: Commands the receiver to reconfigure its navigation algorithm due to a discontinuity (jump) in the position and velocity aiding states. This command is not crew controlled. Upon detecting a change from an OFF to an ON state for this command, the receiver performs the necessary internal actions and sends a completion bit to the GPS SOP. The command is then turned OFF by the SOP when the GPC is notified of the task completion by the receiver. The receiver then receives the OFF command and resets its internal parameters to wait for the next OFF to ON transition for the command. The minimum time between announced resets shall be greater than two seconds due to the mechanization of the GPS receiver interface messages.

8. GPC navigation covariance information is set to zero indicating GPC data is not applicable.
9. The receiver only acknowledges and enters a commanded mode which is not currently active. For example, commanding TEST mode when the receiver is already in TEST will not restart the receiver self test process. Similarly, commanding INIT when the receiver is already in the INIT mode will not restart the receiver initialization processes.
10. Deleted.
11. The GPS receiver updates its output buffers at the following rates:
 - NAV Mode: 10 Hertz if unaided, and at aiding rate for valid aiding
 - INIT Mode: 1 Hertz
 - TEST Mode: 1 Hertz except during Self Test (BIT) when it is not updated

The GPS receiver high rate navigation solution is provided to the GPC.

12. If the receiver is commanded to a new mode before it has completed its current mode functions, the receiver will complete the current mode functions and then transition to whatever mode is being commanded at that time. The only exception to this is if the receiver is commanded to INIT or NAV during the first 6 seconds of TEST mode. In this case the receiver will not complete its TEST mode functions and will transition to the new commanded mode immediately.
13. The receiver will send all zeros to the GPC for the output data from its non-used sixth channel.
14. The mode command protocol assumes the following: The mode command is sent to the receiver. Receiver gets data appropriate to that mode, with the correct headers, on the same I/O pass with the GPC.

15. The receiver will be capable of operating under conditions of no GPC-GPS I/O, no aiding data from the GPC, and with stale or static GPC aiding data during OPS transitions. In the latter case, the aiding data valid flags will be ON but the data will be stale or static.
16. During receiver selftest, the receiver will provide an I/O capability with the GPC to avoid two successive COMMFAULTS and a FCOS I/O bypass situation. During execution of this mode, the receiver will return invalid data and Header information of '0000'. Valid self test data will be sent to the GPC upon completion of selftest.
17. The navigation data validity bit indicates valid data when the following conditions are satisfied:
 - The Kalman filter estimated error is less than 1000 meters
 - The receiver is in the NAV mode
 - The receiver Performance Monitor (PM) did not detect a critical failure in the previous second
18. The receiver will be capable of receiving Baro altimeter and radar altimeter aiding data simultaneously from the GPC. The GPC software will set either the BARO altimeter data good or the radar altimeter data good to ON (both will not be on simultaneously) to indicate to the receiver which altitude aiding data has preference.
19. The receiver will be capable of deselecting satellites upon command from the GPC.
20. The receiver assumes that two sets of aiding data, (1) position and velocity and (2) attitude data have the same time tag. The time tag of the attitude angles is used by the receiver to calculate rate of change in the attitude.

4.17.1.1.2 GPS Digital Interfaces

The GPS receiver interfaces with the GPC via the serial digital multiplex/demultiplex input and output data channel. The digital I/O halfwords are 16 bits in size. There are two blocks of data sent from GPC to GPS and one block of data sent from GPS to GPC. The format of the GPS GPC I/O data interfaces are illustrated in Table 4.17.1.1.2-1. Depending on the GPS receiver mode, the data blocks take on different messages as detailed in Table 4.17.1.1.2-2 through Table 4.17.1.1.2-7. Each message content is uniquely determined by its hex Header ID. Note that messages 2 and 3 and their hex Header IDs do not have the same corresponding natural order. These data messages shall be packed appropriately into the data blocks by the GPS SOP or the GPS receivers.

4.17.1.1.3 GPS Discrete Interfaces

None.

4.17.1.2 GPS SOP Functional Overview

The GPS SOP performs the following functions:

1. Controls and monitors the operation of the GPS receiver(s).
2. Processes GPS data for use by the other GPC program elements.

3. Processes GPC data for use by the GPS receiver(s).

The GPS SOP is partitioned into 9 modules as follows:

- A. GPS Data Processing. The GPS_TO_GPC module processes the GPS data which is receiver mode dependent.
- B. GPS Data Quality Processing. The GPS_QA1_CHECK module examines the status of the GPS receivers and performs a quality check on the receiver figure of merit.
- C. GPS Altitude Aiding Processing. The GPS_ALT_DATA module performs the data processing necessary for GPS altitude aiding.
- D. GPC To GPS Processing. GPC_TO_GPS module buffers the GPC-derived commands and data which support the GPC-controlled GPS mode selection (TEST, INIT, NAV) and GPS satellite deselection, GPS receiver initialization, GPS receiver navigation filter restart, GPS receiver state announced reset, INS aiding, attitude aiding and altitude aiding.
- E. GPS Display Computations. The GPS_DISP_COMP module calculates parameters needed by the GPS STATUS (SPEC 55) display.
- F. GPS SOP Downlist Computations. The GPS_SOP_DL_COMP module prepares downlist parameters.
- G. GPS IMU Processing. The GPS_IMU_SAVE module reserves the appropriate IMU accumulated velocity counts and time tag for the current GPS states.
- H. GPS SOP Time Adjustment Uplink Processing. The GPS_SOP_TIME_ADJ_UPLINK module changes and stores the GPS SOP's GPS to GMT time adjustment in response to a command uplink of this parameter.
- I. GPS SOP Time Adjustment Module. The GPS_TIME_ADJ module computes the accumulated GPS time adjustment required to center the GPS time tag minus GPC time tag at a predetermined I-Load value.

4.17.1.3 GPS SOP Functional Module Summary

The functional modules that constitute the GPS SOP are identified in Table 4.17.1.3-1. Each of the modules, including functions and formulations, is presented in detail in Section 4.17.6.

4.17.1.4 Reference Documents

TBD

4.17.1.5 Software Protection

The software implementation of the GPS SOP shall protect against program checks and error interrupts caused by transient-type errors or bad data inputs. Specifically, the GPS SOP shall protect against all of the error conditions identified in paragraph 3.6 that are applicable to the GPS software design and code.

4.17.2 Definitions

See the glossary in Section 5.1.

Table 4.17.1.1.2-1. GPC GPS I/O Data Format				
I/O Data Block Name	Size (Halfwords)	Message Contents**		
		TEST Mode	INIT Mode	NAV Mode
GPC_TO_GPS_DATA_ONE(I)*	32	Message 1	Message 1	Message 1
GPC_TO_GPS_DATA_TWO(I)*	32	(None)***	Message 2	Message 3
GPS_TO_GPC_DATA_ONE(I)*	32	Message 4	Message 5	Message 6

* I = GPS receiver LRU number = 1 to 3
 ** Messages 1 through 6 are described in Tables 4.17.1.1.2-2 through Tables 4.17.1.1.2-7
 *** In TEST mode, this second data block to be sent to the receiver will be all zero or blank data. The header, first halfword of GPC_TO_GPS_DATA_TWO data block, contains HEX'0000' will signal the receiver to ignore the rest of the data block message. Thus the GPS receiver will only use the first data block, GPC_TO_GPS_DATA_ONE, in test mode.

Table 4.17.1.1.2-2. Message 1 Format - GPS Receiver Aiding Data					
Data Description	Halfword Number	Data Type	Units	Typical LSB	GPS SOP Related Parameters*
Header (HEX'AAAA')	1	BCH	-	-	
Control word sent to GPS receiver	2-3	HMD	-	-	GPS_CONTROL_WORD**
Aiding state time tag - GPS week from 1/6/1980	4	HXU	week	1	T_REF_WEEK_OUT(I)***
Aiding state time tag - seconds from beginning of GPS week	5-7	EMD****	sec	10 ⁻⁸	T_REF_SEC_OUT(I)***
Position aiding state - X component	8-9	FXS	ft	1/16	\bar{R} _REF_OUT (I,1)***
Position aiding state - Y component	10-11	FXS	ft	1/16	\bar{R} _REF_OUT (I,2)***
Position aiding state - Z component	12-13	FXS	ft	1/16	\bar{R} _REF_OUT (I,3)***
Velocity aiding state - X component	14-15	FXS	ft/sec	1/512	\bar{V} _REF_OUT (I,1)***
Velocity aiding state - Y component	16-17	FXS	ft/sec	1/512	\bar{V} _REF_OUT (I,2)***
Velocity aiding state - Z component	18-19	FXS	ft/sec	1/512	\bar{V} _REF_OUT (I,3)***
Spares	20-23	-	-	-	
Command discretes from Crew to deselect specific Satellites from receiver consideration	24-27	HMD	-	-	DESELECT_SAT_RCVD

Table 4.17.1.1.2-2. Message 1 Format - GPS Receiver Aiding Data					
Data Description	Halfword Number	Data Type	Units	Typical LSB	GPS SOP Related Parameters*
Square root of the sum of the position diagonals of Orbiter Nav filter covariance Matrix	28-29	FXU	ft	1	Not used by PASS S/W
Square root of the sum of the velocity diagonals of Orbiter Nav filter covariance Matrix	30	HXU	ft/sec	.01	Not used by PASS S/W
Spares	31-32	-	-	-	

* These GPC parameters are related to the fields defined in the table. Their data attributes can be found where they are defined.
 ** See Table 4.17.6.4.1-3 for detailed definition of GPS_CONTROL_WORD
 *** I = GPS receiver LRU number = 1 to 3 (Parameters without "I" are the same for all GPS receivers)
 **** Three 16-bit halfwords totaling a 48-bit fix point number.

Table 4.17.1.1.2-3. Message 2 Format - Antenna Correction Data

Data Description	Halfword Number	Data Type	Units	Typical LSB	GPS SOP Related Parameters*
Header (HEX'CCCC')	1	BCH	-	-	
Upper antenna to nav base lever arm correction in Orbiter structural coordinates	2-4	HXS	ft	1/16	GPS_UPPER_ANTENNA_NB(I)**
Lower antenna to nav base lever arm correction in Orbiter structural coordinates	5-7	HXS	ft	1/16	GPS_LOWER_ANTENNA_NB(I)**
Upper antenna to CG lever arm correction in Orbiter structural coordinates	8-10	HXS	ft	1/16	GPS_UPPER_ANTENNA_CG(I)**
Lower antenna to CG lever arm correction in Orbiter structural coordinates	11-13	HXS	ft	1/16	GPS_LOWER_ANTENNA_CG(I)**
GPS upper antenna boresight direction in Orbiter body structure coordinates	14-16	HXS	-	1/2 ¹⁵	GPS_UPPER_ANTENNA_BORESIGHT(I)**
GPS lower antenna boresight direction in Orbiter body structure coordinates	17-19	HXS	-	1/2 ¹⁵	GPS_LOWER_ANTENNA_BORESIGHT(I)**
GPS pseudo range residual edit criteria***	20	HMD	-	-	-
GPS delta range residual edit criteria***	21	HMD	-	-	-
Spares	22-32	-	-	-	

* These GPC parameters are related to the fields defined in the table. Their data attributes can be found where they are defined.
 ** I = GPS receiver LRU number = 1 to 3
 *** See Tables 4.17.6.4.2-3 and 4.17.6.4.2-4 for detailed definition

Table 4.17.1.1.2-4. Message 3 Format - GPS Receiver Aiding data (Attitude and Altitude)					
Data Description	Halfword Number	Data Type	Units	Typical LSB	GPS SOP Related Parameters*
Header (HEX'BBBB')	1	BCH	-	-	
Orbiter body roll attitude	2	HXS	rad	PI/2 ¹⁵	PHI_OUT
Orbiter body pitch attitude	3	HXS	rad	PI/2 ¹⁵	THETA_OUT
Orbiter body yaw attitude	4	HXS	rad	PI/2 ¹⁵	PSI_OUT
Spare	5-7				
BARO altitude time tag - GPS week from 1/6/1980	8	HXU	week	1	T_GPS_BARO_WEEK_OUT
BARO altitude time tag - seconds from beginning of GPS week	9-11	EMD**	sec	10 ⁻⁸	T_GPS_BARO_SEC_OUT
Radar altitude time tag - GPS week from 1/6/1980	12	HXU	week	1	T_GPS_RADALT_WEEK_OUT
Radar altitude time tag - seconds from beginning of GPS week	13-15	EMD**	sec	10 ⁻⁸	T_GPS_RADALT_SEC_OUT
BARO altitude measurement	16-17	FXS	ft	1	GPS_BARO_DATA_OUT
Radar altitude measurement	18	HXS	ft	1/2	GPS_RADALT_DATA_OUT
Spare	19-32	-	-	-	

* These GPC parameters are related to the fields defined in the table. Their data attributes can be found where they are defined.
 ** Three 16-bit halfwords totaling 48-bit fixed number.

Table 4.17.1.1.2-5. Message 4 Format - GPS Receiver Testing Data					
Data Description	Halfword Number	Data Type	Units	Typical LSB	GPS SOP Related Parameters*
Header (HEX'DDDD')	1	BCH	-	-	
GPS receiver I navigation mode word***	2-3	HMD	-	-	NAV_MODE_WORD(I)**
GPS receiver I test BIT fault log word****	4-21	HMD	-	-	BIT_FAULT_LOG_WORD (I,1 to 18)**
GPS receiver I Time Totalizing Meter (TTM)-counter indicating the total elapsed receiver operating time	22-23	FXU	hours	1	(not used)
GPS receiver I CPCI ID-last four (ASCII) characters from the computer program configuration time	24-25	ASCII	-	-	(not used)
Spare	26-32	-	-	-	

* These GPC parameters are related to the fields defined in the table. Their data attributes can be found where they are defined.
 ** I = GPS receiver LRU number = 1 to 3
 *** GPS receiver navigation mode word is defined in Table 4.17.6.1.1-3
 **** GPS receiver bit fault log word is not defined for the single string GPS configuration

Table 4.17.1.1.2-6. Message 5 Format - GPS Receiver Initialization Data					
Data Description	Halfword Number	Data Type	Units	Typical LSB	GPS SOP Related Parameters*
Header (HEX'EEEE')	1	BCH	-	-	
GPS receiver I navigation mode word***	2-3	HMD	-	-	NAV_MODE_WORD(I)**
Receiver I time tag - GPS week from 1/6/1980	4	HXU	week	1	(not used)
Receiver I time tag - seconds from beginning of GPS week	5-7	EMD****	sec	10 ⁻⁸	(not used)
Receiver I PPS status word	8	HMD	-	-	PPS_STATUS_WORD(I)**
Receiver I almanac reference week number	9	HXU	week	1	ALMANAC_REF_WEEK(I)**
Spare	10-32	-	-	-	

* These GPC parameters are related to the fields defined in the table. Their data attributes can be found where they are defined.
 ** I = GPS receiver LRU number = 1 to 3
 *** GPS receiver navigation mode word is defined in Table 4.17.6.1.1-3
 **** Three 16-bit halfwords totaling a 48-bit fix number

Table 4.17.1.1.2-7. Message 6 Format - GPS Receiver Navigation data					
Data Description	Halfword Number	Data Type	Units	Typical LSB	GPS SOP Related Parameters*
Header (HEX'FFFF')	1	BCH	-	-	
GPS receiver I navigation mode word***	2-3	HMD	-	-	NAV_MODE_WORD(I)**
Receiver I time tag - GPS week from 1/6/1980	4	HXU	week	1	T_GPS_WEEK(I)**
Receiver I time tag - seconds from beginning of GPS week	5-7	EMD§	sec	10 ⁻⁸	T_GPS_SEC(I)**
Receiver I position solution - X component	8-9	FXS	ft	1/16	\bar{R} _GPS_ECEF_SOP (I,1)**
Receiver I position solution - Y component	10-11	FXS	ft	1/16	\bar{R} _GPS_ECEF_SOP (I,2)**
Receiver I position solution - Z component	12-13	FXS	ft	1/16	\bar{R} _GPS_ECEF_SOP (I,3)**
Receiver I velocity solution - X component	14-15	FXS	ft/sec	1/512	\bar{V} _GPS_ECEF_SOP (I,1)**
Receiver I velocity solution - Y component	16-17	FXS	ft/sec	1/512	\bar{V} _GPS_ECEF_SOP (I,2)**
Receiver I velocity solution - Z component	18-19	FXS	ft/sec	1/512	\bar{V} _GPS_ECEF_SOP (I,3)**
Receiver I status, FOM and GDOP word****	20	HMD	-	-	GPS_STATUS_FOM_GDOP(I)**
Receiver I PPS status word§§	21	HMD	-	-	PPS_STATUS_WORD(I)**
Receiver I channel status word*****	22-30	HMD	-	-	CHANNELS_STATUS(I,K)**
Receiver I time figure of merit	31	HMD	-	-	TFOM(I)**
Receiver I age of aiding data	32	HXS	msec	1	GPS_AIDING_DATA_AGE_DL (I)**

* These GPC parameters are related to the fields defined in the table. Their data attributes can be found where they are defined.

** I = GPS receiver LRU number = 1 to 3, K = 1 to 9

*** GPS receiver navigation mode word is defined in Table 4.17.6.1.1-3

**** GPS receiver status, FOM and GDOP word is defined in Table 4.17.6.1.2-3

***** GPS receiver channel status words are defined in Table 4.17.6.1.2.1-3 through Table 4.17.6.1.2.1-11

§ Three 16-bit halfwords totaling a 48-bit fix point number

§§ GPS receiver PPS status words are defined in Table 4.17.6.1.3-3

Table 4.17.1.3-1. GPS SOP Function Modules

Module Name	Applicable Paragraph	Module Description
GPS_TO_GPC	4.17.6.1	GPS data processing
GPS_QA1_CHECK	4.17.6.2	GPS data quality check
GPS_ALT_DATA	4.17.6.3	Altitude aiding data processing
GPC_TO_GPS	4.17.6.4	Output data processing to GPS receivers
GPS_DISP_COMP	4.17.6.5	GPS displays computations
GPS_SOP_DL_COMP	4.17.6.6	GPS SOP downlist computations
GPS_IMU_SAVE	4.17.6.7	IMU data processing
GPS_SOP_TIME_ADJ_UPLINK	4.17.6.8	GPS to GMT time adjustment uplink processing
GPS_TIME_ADJ	4.17.6.9	GPS SOP time adjustment

4.17.3 Functional Requirements Traceability

The detailed requirements and formulations contained herein satisfy the GN&C, Level B, CPDS functional requirements for the GPS SOP.

4.17.4 GPS SOP Interfaces

This section defines the interfaces associated with the GPS SOP. Inputs and outputs of GPS SOP are contained in the IDD Tables 4.17.6.10-1 and 4.17.6.10-2.

4.17.5 Parameter Initialization

This section defines the I-load, K-load and Constant parameters associated with the GPS SOP. Table 4.17.6.10-3 contains the I-load parameters. Table 4.17.6.10-4 contains the K-load parameters. Table 4.17.6.10-5 contains the Constant parameters.

4.17.6 Detailed Requirements and Formulations

This section defines the GPS_SOP functional modules in sufficient detail to create a functional design and generate computer code. Figure 4.17.6-1 presents a flow diagram of the GPS SOP. Each module defines the functional requirements, interface requirements, detailed requirements and formulations, and special considerations if applicable. A flow diagram is also included for each module; however, the flow diagrams are supplementary material and are not intended to convey requirements or to place restrictions on functional design or coding. Tables 4.17.6-1 and 4.17.6-2 contain the input and output parameters for the GPS SOP control function.

The GPS SOP shall function as follows:

1. A homogeneous set of GPS data shall be read into the GPC prior to each execution of the GPS_TO_GPC module, which performs the required data format conversion and scaling. Table 4.17.1.1.2-5 through Table 4.17.1.1.2-7 list the three possible input message data from each of the available GPS receivers. It is not required that the different receivers have the same time tag (i.e. be time homogeneous with each other). However, data from each individual receiver is required to be time homogeneous.

SNAP GPS (GPS_TO_GPC_DATA_ONE₁, GPS_TO_GPC_DATA_ONE₂,
GPS_TO_GPC_DATA_ONE₃)

2. The selected accumulated sensed velocities and the associated time tag shall be snapped from IMU RM for use by the GPS Receiver State Processing principal function.

SNAP IMU (\bar{V} _IMU_GPS_SOP, T_IMU_GPS_SOP, REFS_RDY)

3. The GPS commfault flags (GPS_COMMFAULT₁), GPS bypass flags (GPS_BYPASS₁), and GPS transaction status words (GPS_TSW₁) shall be snapped from system software.

SNAP FCOS (GPS_COMMFAULT₁, GPS_COMMFAULT₂, GPS_COMMFAULT₃,
GPS_BYPASS₁, GPS_BYPASS₂, GPS_BYPASS₃, GPS_TSW₁, GPS_TSW₂,
GPS_TSW₃)

4. Only in Major Mode 304, 305, 602 and 603, Barometric altimeter data is also collected and stored. This includes the Shuttle's barometric altitude, a flag indicating if the Shuttle is in the MACH jump region, its data good flag and the associated time tag, to be adjusted by a premission-loaded time bias. Baro data shall be used for altitude-aiding the GPS receivers.

SNAP BARO (Q_BARO, MACH_JUMP, BARO_DG, T_BARO)

5. Only in Major Mode 305 and 603, Radar altimeter data, consisting of the altitude of the Shuttle, the data good flag, and the time tag (which will subsequently be adjusted by a premission-loaded time bias) shall be collected for altitude-aiding the GPS receivers.

SNAP RADALT (RADALT_HEIGHT, RADALT_DG, T_RADALT)

6. Commands and data from crew via GPS STATUS display are collected. These include command to restart GPS navigation filter, command to mode GPS receiver to 'TEST', 'INIT' or 'NAV', GPS I/O designation, crew deselected satellite list, crew deselected GPS receiver, and crew enabled GPS time adjustment capability. Commands from crew via Horizontal Situation display are also collected. These include command to enable GPS radar altimeter aiding and BARO auto/inhibit/force command.

SNAP GPS_STATUS (GPS_IO_CNTL, FILT_RESTART_RCVD₁,
FILT_RESTART_RCVD₂, FILT_RESTART_RCVD₃,
GPS_MODE_CMD_RCVD₁, GPS_MODE_CMD_RCVD₂,
GPS_MODE_CMD_RCVD₃, CREW_DESELECT_SAT_RCVD,
CREW_DES_RCVR_SOP₁, CREW_DES_RCVR_SOP₂,
CREW_DES_RCVR_SOP₃, GPS_TIME_ADJ_ENBL)

SNAP HSD (GPS_RA_RCVD, BARO_AIF)

Note that GPS_RA_RCVD can only be set by the single GPS configuration.

7. In all OPS modes except OPS9, a time homogeneous set of Reference States and associated flags from the GPS Reference State Proc principal function shall be collected here for aiding the GPS receivers.

SNAP REFERENCE STATE (T_CURRENT_GPS_REF, \bar{R} _REF, \bar{V} _REF,
AIDING_DATA_VALID₁, AIDING_DATA_VALID₂, AIDING_DATA_VALID₃,
GPS_ANNCD_RESET_SET₁, GPS_ANNCD_RESET_SET₂,
GPS_ANNCD_RESET_SET₃, INIT_STATE_SET₁, INIT_STATE_SET₂,
INIT_STATE_SET₃, GPS_FILT_RESTART₁, GPS_FILT_RESTART₂,
GPS_FILT_RESTART₃, GPS_CMD_RESET₁, GPS_CMD_RESET₂,
GPS_CMD_RESET₃, GPS_ANNCD_RESET_TIME_ADJ)

8. The GPS_SOP_TIME_ADJ_UPLINK module is called. This module updates the GPS to GMT time adjustment with a ground update when required.
9. For each of the receivers, the GPS SOP shall perform the following operations, if two-way communication is enabled, GPS_IO_CNTL = 2:

- 9.1 If the flag, `FILT_RESTART_RCVDi`, is ON indicating Crew command to restart GPS navigation filter and the flag, `GPS_FILT_RESTARTi`, is OFF indicating that the GPS filter restart data is not prepared and ready, then the request flag, `FILT_RESTART_REQUESTi`, shall be set ON indicating a request to the GPS Reference State Proc principal function to determine whether a GPS filter restart can currently be commanded, and if so, command the filter restart.

IF `FILT_RESTART_RCVDi` = ON AND `GPS_FILT_RESTARTi` = OFF
THEN `FILT_RESTART_REQUESTi` = ON

- 9.2 From GPS Reference State Proc principal function, the flag, `GPS_FILT_RESTARTi`, is checked next. If it is ON indicating that the GPS filter restart is commanded, and its previous state, `GPS_FILT_RESTART_PREVi`, is OFF, then the clocktime is saved for use by `NAV_MODE_WORD_DATA` module in determining the logic to refresh the filter restart command field on the `GPS_STATUS` display, and the crew command to do filter restart, `GPS_FILT_RESTART_CMDi`, shall be set OFF.

IF `GPS_FILT_RESTARTi` = ON AND `GPS_FILT_RESTART_PREVi` = OFF
THEN `T_FRi` = CLOCKTIME
`GPS_FILT_RESTART_CMDi` = OFF

- 9.3 The Crew commanded GPS mode, `GPS_MODE_CMD_RCVDi`, is checked next. If it is not INIT command, then the command shall be honored on this GPS output pass and the `GPS_INIT_REQUEST_SOPi` flag shall be set to OFF. Otherwise, the `GPS_INIT_REQUEST_SOPi` flag shall be set ON requesting initialization from GPS Reference State Proc principal function, if the current GPS mode, `GPS_MODE_CURRENTi`, is not already in 'INIT' and the initialization is not already commanded.

IF `GPS_MODE_CMD_RCVDi` not = INIT THEN
`GPS_MODE_CMD_SOPi` = `GPS_MODE_CMD_RCVDi`
`GPS_INIT_REQUEST_SOPi` = OFF
ELSE
IF `GPS_MODE_CURRENTi` NOT = 'INIT' AND `INIT_STATE_SETi` = OFF THEN
`GPS_INIT_REQUEST_SOPi` = ON

- 9.4 When the initialization is commanded from GPS Reference State Proc principal function as indicated by the flag, `INIT_STATE_SETi`, being ON, the GPS SOP commanded GPS mode will be INIT.

IF `INIT_STATE_SETi` = ON THEN
`GPS_MODE_CMD_SOPi` = INIT

- 9.5 The `GPS_FILT_RESTARTi` command is stored as follows:

`GPS_FILT_RESTART_PREVi` = `GPS_FILT_RESTARTi`

10. For each of the receivers, the GPS SOP shall perform the following operations:

- 10.1 If 6.25 Hz input is enabled (GPS_IO_CNTL = 1); or if both flight I/O has been enabled and the GPS bypass flag is ON (GPS_IO_CNTL = 2 and GPS_BYPASS_i = ON); or if the GPS command reset flag is ON (GPS_CMD_RESET_i = ON), then perform the following:

```
GPS_INIT_REQUEST_SOPi = OFF
INIT_STATE_SETi = OFF
GPS_FILT_RESTARTi = OFF
FILT_RESTART_REQUESTi = OFF
GPS_FILT_RESTART_CMDi = OFF
GPS_ANNCD_RESET_SETi = OFF
GPS_CMD_RESETi = OFF
```

IF GPS_IO_CNTL = 1, then also check the GPS transaction status word GPS_TSW₁ to determine if it is non-zero. If so, set GPS_COMMFAULT₁ = ON; otherwise set GPS_COMMFAULT₁ = OFF.

```
IF GPS_TSW1 ≠ '0000'
    GPS_COMMFAULT1 = ON
ELSE
    GPS_COMMFAULT1 = OFF
ENDIF
```

- 10.2 Set the FOM_GDOP_STATIC_i flag to on. This flag will subsequently be reset to OFF when the FOM and GDOP are determined to be valid for display.

```
FOM_GDOP_STATICi = ON
```

- 10.3 Check the commfault flag, GPS_COMMFAULT_i. If no communications fault exists, then GPS_TO_GPC is executed to process the incoming GPS data.

```
IF GPS_COMMFAULTi = OFF THEN
    EXECUTE: GPS_TO_GPC CODE
```

11. If the GPS Time Adjustment capability is enabled (GPS_TIME_ADJ_ENBL = ON) then the GPS Time Adjustment computation shall be performed according to information available in section 4.17.6.9 (GPS_TIME_ADJ). Otherwise the accumulated GPS time adjustment (T_GPS_TIME_ADJ_ACC) shall be set to zero. In addition, the flag indicating an announced reset for the GPC-to-GPS time adjustment computation (GPS_ANNCD_RESET_TIME_ADJ) shall be set to OFF.

```
IF (GPS_TIME_ADJ_ENBL = ON) THEN
    EXECUTE: GPS_TIME_ADJ
ELSE
    T_GPS_TIME_ADJ_ACC = 0
    GPS_ANNCD_RESET_TIME_ADJ = OFF
END IF
```

12. If two-way communication is enabled, GPS_IO_CNTL = 2, perform the following:

- 12.1 The GPS SOP shall execute the GPS_ALT_DATA module to set up data for GPS receiver altitude aiding.
- 12.2 The GPS SOP shall execute the GPC_TO_GPS module, which prepares data to be transferred to GPS receivers.
13. If the flag, REFS_RDY, is ON indicating IMU data is available and major mode is not 901, then the GPS SOP shall execute the GPS_IMU_SAVE module, which stores the set of accumulated IMU velocity counts that are most appropriate to the latest GPS state vectors. The appropriateness of the selection of IMU velocity counts is explained in detail in section 4.17.6.7.
14. The GPS SOP shall execute the GPS_DISP_COMP module, which processes GPS data for display.
15. The GPS SOP shall execute the GPS_SOP_DL_COMP module, which prepares GPS downlist data.

At each OPS transition except transition between OPS1 and OPS6 the following variables shall be initialized:

- a. The current GPS mode indicator, GPS_MODE_CURRENT, shall be initialized to blank. The actual operating mode shall be determined via examining the NAV_MODE_WORD by GPS SOP.
- b. The GPS filter restart request flag, FILT_RESTART_REQUEST, shall be set OFF indicating no request to the GPS Reference State Proc principal function to command a GPS filter restart.
- c. The GPS initialization state request flag, GPS_INIT_REQUEST_SOP, shall be initialized to OFF indicating no request to the GPS Reference State Proc principal function to command an initialization.
- d. The GPS initialization state ready flag, INIT_STATE_SET, shall be initialized to OFF indicating that GPS initialization is not commanded.

Prior to the first pass of the GPS SOP, (a) the GPC to GPS and GPS to GPC time reference conversion parameter, T_GPS_OFFSET_SECS, and (b) the number of GPS LRU's installed, NUM_GPS_INSTALLED, shall be initialized as follows:

$$T_GPS_OFFSET_SECS = (T_GPS_OFFSET_DAYS \ 86400) - 86400$$

$$NUM_GPS_INSTALLED = GPS_INSTALLED(1) + GPS_INSTALLED(2) \\ + GPS_INSTALLED(3)$$

For a year-end rollover event, T_GPS_OFFSET_SECS will be incremented by the number of seconds in a year. [Note: A detailed description of this processing can be found in section 4.1.1 Deorbit/Landing Navigation Sequencer (4.123) in STS83-0004, the Entry Through Landing Navigation FSSR.]

T_GPS_OFFSET_SECS and NUM_GPS_INSTALLED shall be preserved across all GN&C OPS transitions.

The GPS SOP is unique among shuttle GN&C applications in that GPS receiver input and output processing is initiated by a timer different from the one that initiates the application and because it is both crew and I-load configurable.

During OPS 1/6, 2, 3, and 8, if GPS_IO_CNTL = 2 (its value in these OPS when at least one GPS receiver is installed), then the GPS I/O, specified in the GPS support section of the SSW Level-A CPDS (SS-P-0002-170), shall be invoked by GPS Input and Output Processes to provide receiver inputs at 1.04 Hz and receiver outputs at 6.25 Hz. Otherwise, if GPS_IO_CNTL = 0 (its value in these OPS when no receivers are installed), then no GPS I/O shall be requested by the GPS Input and Output Processes. GPC TO GPS output shall follow GPS TO GPC input on cycles on which both the input and output occur. The GPS I/O Processes shall execute independent of the GPS SOP and shall be phased within the I/O profile so as to maximize available system I/O performance. However, the various SOP modules shall be sequenced so as to process GPS TO GPC inputs within 240 milli-seconds of their acquisition and provide SOP generated GPC TO GPS outputs that are no more than 240 milli-seconds old. Note that these times are maximums and can not both be maximized so the total of the two times will not exceed 320 ms in any given cycle.

During OPS 9 the following three crew selectable GPS I/O configurations shall be provided as determined by the setting of GPS_IO_CNTL by Item 10 on SPEC 55 (the GPS Status Display):

GPS_IO_CNTL = 0:	disabled GPS receiver I/O (no input and no output)
GPS_IO_CNTL = 1:	6.25 Hz non-commfaultable GPS to GPC input only
GPS_IO_CNTL = 2:	flight GPS I/O of 1.04 Hz GPS TO GPC input and 6.25 Hz GPC TO GPS output (as specified in the GPS Support section of the SSW Level-A CPDS (SS-P-0002-170)).

GPS I/O shall be initially disabled (GPS_IO_CNTL = 0) in OPS 9. Upon transition to OPS 1 and provided that at least one GPS receiver is installed, flight GPS I/O (GPS_IO_CNTL = 2) shall be enabled. Only the flight and disabled GPS I/O configurations shall be supported in the flight OPS. Upon transition to OPS 9 from any other GN&C OPS and provided that at least one GPS receiver is installed, flight GPS I/O (GPS_IO_CNTL = 2) shall be in effect until reconfigured by crew input. During GNC OPS 9, the GPS/GPC I/O process shall suspend operations when the GPS_IO_CNTL is set to 0. If in GNC OPS 9 and GPS_IO_CNTL is set to 1 or 2, the GPS/GPC I/O process shall operate at the rates specified above.

When GPS_IO_CNTL = 1 in OPS 9, the GPS TO GPC input process shall request, on an installed LRU basis, 6.25 Hz non-commfaultable GPS to GPC input. Further, the GPS to GPC input process shall examine the FCOS returned input transaction status word and either clear or set the appropriate GPS_COMMFAULT(I) bits to either enable or disable processing of the input data by the downstream GPS SOP modules. Specifically, if the FCOS returned transaction status word, GPS_TSW(I) for GPS LRU I (where I = 1, 2, or 3), is non-zero, then the receiver commfault bit, GPS_COMMFAULT(I), shall be set. Otherwise, the GPS_COMMFAULT(I) bit shall be cleared. When GPS_IO_CNTL is 0 or 2 the GPS_COMMFAULT(I) bits shall not be cleared or set by the GPS TO GPC input process (this is an FCOS function when GPS_IO_CNTL = 2).

During OPS 9 when GPC to GPS outputs are disabled (GPS_IO_CNTL = 1 or 0) the ground is free to command the GPS receivers using the LDB/TCS operators (see SS-P-0002-150 section 5.3.4 Operator Code 3 (Issue) for example) or via some other GPS receiver test stimuli.

At the first FCOS output to the GPS receivers following an OPS transition, and at succeeding FCOS outputs until the GPS SOP executes, the following requirements shall be met. Note that the following requirements are also applicable when GPS_IO_CNTL transitions to two while in OPS 9:

The default values of both 32 halfword output of the GPS SOP output buffers shall remain as set before the OPS transition, except for the following:

1. The default values of the mode commands (Bit 0, 1, 30 and 31 of GPC_TO_GPS_DATA_ONE_{i,2 TO 3}, I = 1 to 3) at OPS transitions shall insure the default mode command requirements as specified in the Shuttle Display and Controls FSSR STS 83-0020 for the software release for which this FSSR applies. The default values of the major mode (Bit 17 to 27 of GPC_TO_GPS_DATA_ONE_{i,2 TO 3}, I = 1 to 3) shall be that of the new OPS. All other bits in GPC_TO_GPS_DATA_ONE_{i,2 TO 3} shall be set to '0' at all OPS transition except transition between OPS 1 and OPS 6.
2. The header label (first 16 bit halfword) in the first 32 halfword output buffer shall have the value 'AAAA'. Note that the header label (first 16 bit halfword) in the second 32 halfword output buffer will prevent the receiver from processing incorrect mode data in the event that the default mode setting at the OPS transition is different from the mode setting at exit from the previous OPS.

There are no specific requirements on the GPC TO GPS output buffers generated by GPS SOP during the period following an OPS transition and prior to the execution of principal functions which provide data to GPS SOP. It is assumed that the data parameters provided to the GPS SOP and used to generate the GPC TO GPS output buffers are appropriately initialized at OPS transition by the appropriate principal functions. The proper initialization is required in order to preclude GPS SOP from forming erroneous output buffers.

Table 4.17.6-1. GPS SOP Control Input Parameters

Description	Symbol	Source	Type	Prec	Range	Units	Rate
Flag indicating valid aiding reference state for GPS receiver I	AIDING_DATA_VALID(I)**	*	D	-	ON/OFF	-	1.04
BARO auto/inhibit/force switch	BARO_AIF	*	D	-	ON/OFF	-	1.04
Pressure altitude data good indicator	BARO_DG	*	D	-	ON/OFF	-	1.04
Current computer clock time (GMT)	CLOCKTIME	*	F	DP	-	SEC	1.04
Flag indicating filter restart states have been made available for GPS receiver I (snapped copy of GPS_FILT_RESTART_CMD(I)**)	FILT_RESTART_RCVD(I)**	*, NAV_MODE_WORD_DATA	D	-	ON/OFF	-	1.04
Flag indicating GPS announced reset of receiver I is commanded	GPS_ANNCD_RESET_SET(I)**	*	D	-	ON/OFF	-	6.25
Flag indicating a GPS command reset is required due to a GPS command time-out	GPS_CMD_RESET(I)**	*	D	-	ON/OFF	-	6.25
Flag indicating communication fault status for GPS receiver I	GPS_COMMFAULT(I)**	*	D	-	ON/OFF	-	1.04
Flag indicating GPS filter restart of receiver I is commanded	GPS_FILT_RESTART(I)**	*	D	-	ON/OFF	-	6.25
Designation of GPS I/O to be performed	GPS_IO_CNTL	*	I	S	0,1,2	-	0.52
Crew requested mode for GPS receiver I	GPS_MODE_CMD_RCVD(I)**	*	I	S	TEST(0) INIT(1) NAV(2)	-	0.52
Crew initiated radar altimeter aiding flag	GPS_RA_RCVD	*	D	-	ON/OFF	-	6.25
GPS receiver I to GPC data buffer one	GPS_TO_GPC_DATA_ONE(I)**	*	D	-	-	-	1.04

Table 4.17.6-1. GPS SOP Control Input Parameters

Description	Symbol	Source	Type	Prec	Range	Units	Rate
Crew deselected GPS receiver	CREW_DES_RCVR_SOP(I)**	*	D	-	ON/OFF	-	0.52
GPS time adjustment enable ind.	GPS_TIME_ADJ_ENBL	*	D	-	ON/OFF	-	0.52
Flag indicating an announced reset for the GPC-to-GPS time adjustment computation	GPS_ANNCD_RESET_TIME_ADJ	*	D	-	ON/OFF	-	6.25
Flag indicating initialization of receiver I is commanded	INIT_STATE_SET(I)**	*	D	-	-	-	6.25
Flag indicating (ON) that the shuttle is in the MACH jump region	MACH_JUMP	*	D	-	ON/OFF	-	1.04
Major Mode indicator	MM_CODE	*	I	S	-	-	6.25
ADS corrected pressure altitude	Q_BARO	*	F	S	-	FT	1.04
Current GPS reference state position	R_REF	*	V	DP	-	FT	6.25
Radar Altimeter data good indicator	RADALT_DG	*	D	-	ON/OFF	-	6.25
Radar Altimeter measurement	RADALT_HEIGHT	*	F	S	-	FT	6.25
Status flag for IMU reference matrix computation	REFS_RDY	*	D	-	ON/OFF	-	-
Time tag for ADS corrected pressure altitude	T_BARO	*	F	DP	-	SEC	1.04
Time tag for current GPS reference state	T_CURRENT_GPS_REF	*	F	DP	-	SEC	6.25
Number of seconds from 1/6/80 to midnight between 12/30 and 12/31 of year prior to launch year	T_GPS_OFFSET_SECS	*	F	DP	-	SEC	on Demand
Time tag of current accumulated IMU velocity counts	T_IMU_GPS_SOP	*	F	DP	-	SEC	1.04/ 6.25
Time tag for radar altimeter data	T_RADALT	*	F	DP	-	SEC	6.25
Current accumulated IMU velocity counts	V_IMU_GPS_SOP	*	V	DP	-	FT/SEC	1.04/ 6.25
Current GPS reference state velocity	V_REF	*	V	DP	-	FT/SEC	6.25
Flag indicating GPS receiver I is to be bypassed	GPS_BYPASS(I)**	*	D	-	ON/OFF	-	1.04
Crew deselect satellite word K	CREW_DESELECT_SAT_RCVD(K)***	*	HMD	-	-	-	6.25
GPS transaction status words	GPS_TSW(I)**	*	HMD	-	-	-	6.25

* See principal function input list in Table 4.17.6.10-1

** I = 1,2,3 (1 to GPS_SW_CAP)

*** K = 1 to 4

Table 4.17.6-2. GPS SOP Control Output Parameters

Description	Symbol	Destination	Type	Prec	Range	Units	Rate
Flag indicating the request for a GPS filter restart	FILT_RESTART_REQUEST(I)**	*	D	-	ON/OFF	-	6.25
Flag indicating static FOM/GDOP data	FOM_GDOP_STATIC(I)**	*	D	-	ON/OFF	-	6.25
Flag indicating GPS announced reset of receiver I is commanded	GPS_ANNCD_RESET_SET(I)**	*	D	-	ON/OFF	-	6.25

Table 4.17.6-2. GPS SOP Control Output Parameters

Description	Symbol	Destination	Type	Prec	Range	Units	Rate
Flag indicating GPS filter restart of receiver I is commanded	GPS_FILT_RESTART(I)**	*	D	–	ON/OFF	–	6.25
Flag indicating the request for a GPS receiver initialization	GPS_INIT_REQUEST_SOP(I)**	*, RCVR_AIDING_DATA	D	–	ON/OFF	–	6.25
GPS SOP commanded GPS mode for GPS receiver I	GPS_MODE_CMD_SOP(I)**	RCVR_AIDING_DATA, GPC_TO_GPS	I	S	0,1,2	–	6.25
GPS receiver I current operating mode indicator	GPS_MODE_CURRENT(I)**	*, GPS_TO_GPC	I	S	0,1,2	–	6.25
Flag indicating initialization of receiver I is commanded	INIT_STATE_SET(I)**	*	D	–	ON/OFF	–	6.25
Crew deselected GPS receiver	CREW_DES_RCVR_SOP(I)**	GPS_TIME_ADJ	D	–	ON/OFF	–	6.25
GPS time adjustment enable ind	GPS_TIME_ADJ_ENBL	GPS_SOP_DL_COMP	D	–	ON/OFF	–	6.25
Accumulated GPC-to-GPS time adjustment	T_GPS_TIME_ADJ_ACC	GPS_TIME_ADJ	HXS	–	–	MSEC	6.25
Flag indicating an announced reset for the GPC-to-GPS time adjustment computation	GPS_ANNCD_RESET_TIME_ADJ	GPS_TIME_ADJ	D	–	ON/OFF	–	6.25
Number of seconds from 1/6/80 to midnight between 12/30 and 12/31 of year prior to launch year	T_GPS_OFFSET_SECS	*, NAV_MESSAGE_DATA, GPS_ALT_DATA, RCVR_AIDING_DATA	I	DP	–	SEC	–
Time indicating the beginning of GPS I receiver filter restart process	T_FR(I)**	NAV_MODE_WORD_DATA	F	DP	–	SEC	6.25
Number of GPS receivers installed	NUM_GPS_INSTALLED	*	HXS	–	0,1,3	–	–

* See principal function output list in Table 4.17.6.10-2

** I = 1,2,3 (1 to GPS_SW_CAP)

4.17.6.1 GPS Data Processing (GPS_TO_GPC)

The purpose of the GPS_TO_GPC functional module is to collect data from each GPS receiver and convert it into a form which is usable by the GPS Receiver State Processing principal function, the GPS Reference State Processing principal function and the Orbiter display processing function.

A. Functional Requirements. The GPS_TO_GPC module is required to perform the following functions:

1. Parse GPS data (GPS_TO_GPC_DATA_ONE) according to GPS receiver's operating mode.
2. Compute GPS parameters such as time, position and velocity vectors by performing the required scaling and conversion to obtain floating point data in the appropriate units.

B. Interface Requirements. The input and output parameters associated with the GPS_TO_GPC functional modules are contained in Table 4.17.6.1-1 and 4.17.6.1-2. I-load parameters are contained in Table 4.17.6.10-3. The content of the data block GPS_TO_GPC_DATA_ONE is dependent upon the GPS receiver mode. Table 4.17.1.1.2-5 through Table 4.17.1.1.2-7 define the three possible data message formats for GPS_TO_GPC_DATA_ONE.

C. Detailed Requirements and Formulations

The following operations shall be performed by GPS_TO_GPC module for each receiver. Figure 4.17.6.1 through Figure 4.17.6.1.4 present the flow diagrams of the GPS_TO_GPC module.

1. Deleted
2. If the header (word 1 of the GPS_TO_GPC_DATA_ONE buffer) is HEX 'DDDD' or HEX 'EEEE' or HEX 'FFFF' steps 3 through 5 below shall be executed only. Otherwise, step 6 is executed.
3. Extract NAV_MODE_WORD from word 2 and 3 of the GPS_TO_GPC_DATA_ONE_i buffer.

$$\text{NAV_MODE_WORD}_i = \text{GPS_TO_GPC_DATA_ONE}_{i,(2\text{ TO }3)}$$

4. Execute NAV_MODE_WORD_DATA code to parse the NAV_MODE_WORD_i. This module is detailed in section 4.17.6.1.1. The flag ILLEGAL_MODE is set in this function and is checked next to determine whether to process the rest of the GPS data buffer.
5. If ILLEGAL_MODE flag is OFF, indicating that the GPS mode as decoded from bits 1, 2, 30 and 31 of NAV_MODE_WORD_i is valid, then Steps 5.1 through 5.3 are executed.

- 5.1 GPS Status Display blanking flags are all turned ON. Section 4.17.6.1.2.1 will turn the flags OFF when the GPS receiver indicates state three or state five tracking and GPS_MODE_CURRENT is NAV.

$$\text{SATS_BLANKING_FLAG}_{i,1\text{ to }5} = \text{ON}$$

- 5.2 If GPS_MODE_CURRENT_i is 'NAV', then GPS_TO_GPC_DATA_ONE_i is extracted according to the NAV message format, Message 6, as defined in Table 4.17.1.1.2-7. This process is accomplished by executing the NAV_MESSAGE_DATA subfunction which is detailed in Section 4.17.6.1.2. Set GPS_BIT_DISP_ENABLE_i = OFF.

Else if GPS_MODE_CURRENT_i is 'INIT', then GPS_TO_GPC_DATA_ONE_i is extracted according to the INIT message format, Message 5, as defined in Table 4.17.1.1.2-6. This process is accomplished by executing the INIT_MESSAGE_DATA subfunction which is detailed in Section 4.17.6.1.3. Set GPS_BIT_DISP_ENABLE_i = OFF.

Else if GPS_BIT_SOP_i is ON, then GPS_TO_GPC_DATA_ONE_i is extracted according to the TEST message format, Message 4, as defined in Table 4.17.1.1.2-5. This process is accomplished by executing the TEST_MESSAGE_DATA subfunction which is detailed in Section 4.17.6.1.4. Set GPS_BIT_DISP_ENABLE_i = ON.

5.3 Set GPS_BIT_SOP_i to OFF.

6. If the current GPS mode is 'TEST' and the input header is zero, then the flag, GPS_BIT_SOP_i, shall be set ON indicating GPS receiver i BIT (Built In Test) is in progress.

D. Special Considerations

GPS_BIT_DISP_ENABLE_i shall be initialized to OFF after all OPS transitions except after OPS 1 to OPS 6.

The SATS_BLANKING_FLAG_{i,6} shall be initialized to ON.

Table 4.17.6.1-1. GPS_TO_GPC Input Parameters							
Description	Name	Input Source	Type	Precision	Range	Units	Rate (Hz)
GPS receiver I to GPC data buffer one	GPS_TO_GPC_DATA_ONE(I)**	*	-	-	-	-	1.04
GPS receiver I current operating mode indicator	GPS_MODE_CURRENT(I)**	NAV_MODE_WORD_DATA, GPS_SOP	I	S	TEST(0) INIT(1) NAV(2)	-	1.04
Flag indicating illegal GPS operating mode	ILLEGAL_MODE	NAV_MODE_WORD_DATA	D	-	ON/OFF	-	1.04
* See principal function input list in Table 4.17.6.10-1							
** I = 1,2,3							

Table 4.17.6.1-2. GPS_TO_GPC Output Parameters							
Description	Name	Destination	Type	Precision	Range	Units	Rate (Hz)
GPS receiver I Nav Mode Word	NAV_MODE_WORD(I)**	NAV_MODE_WORD_DATA	-	-	-	-	6.25
GPS BIT DISPLAY ENABLE Flag	GPS_BIT_DISP_ENABLE(I)**	GPS_DISP_COMP	D	-	ON/OFF	-	6.25
GPS receiver I self test in progress flag	GPS_BIT_SOP(I)**	GPS_DISP_COMP GPS_SOP_DL_COMP	D	-	ON/OFF	-	6.25
Satellite blanking flag for GPS Status Display	SATS_BLANKING_FLAG(I,K)**	CHANNEL_STATUS_DATA, *	-	-	-	-	1.04

** I = 1,2,3, K= 1 to 6

4.17.6.1.1 NAV_MODE_WORD Data Processing (NAV_MODE_WORD_DATA)

The purpose of the NAV_MODE_WORD_DATA code is to parse the NAV_MODE_WORD according to the format as defined in Table 4.17.6.1.1-3.

A. Functional Requirements. The NAV_MODE_WORD_DATA code is required to perform the following functions:

1. Parse NAV_MODE_WORD according to the format defined in Table 4.17.6.1.1-3.

B. Interface Requirements. The input and output parameters associated with the NAV_MODE_WORD_DATA module are contained in Table 4.17.6.1.1-1 and 4.17.6.1.1-2.

C. Detailed Requirements and Formulations

The following operations shall be performed by the NAV_MODE_WORD_DATA code for each receiver. Figure 4.17.6.1.1 presents the flow diagram of the NAV_MODE_WORD_DATA code. Table 4.17.6.1.1-3 defines the format of the NAV_MODE_WORD parent word.

1. The GPS data validity flag is set according to the bit 0 of NAV_MODE_WORD_i

```
IF BIT 0 of NAV_MODE_WORDi = 1 THEN
    GPS_NAV_DATA_VALIDi = ON
ELSE
    GPS_NAV_DATA_VALIDi = OFF
```

2. The current GPS receiver operation mode is determined first by examining bits 1, 2, 30 and 31 of NAV_MODE_WORD_i. Table 4.17.6.1.1-3 lists the definition for GPS_MODE_CURRENT_i assignment. The flag ILLEGAL_MODE is set OFF indicating that the GPS_MODE_CURRENT_i takes on either 'NAV' or 'INIT' or 'TEST'. Otherwise, it is set to ON.

```
ILLEGAL_MODE = OFF
```

```

IF BIT 1 and 2 of NAV_MODE_WORDi = (1,0) and
  BIT 30 and 31 of NAV_MODE_WORDi = (0,1)
  THEN GPS_MODE_CURRENTi = 'NAV'

```

ELSE

```

IF BIT 1 and 2 of NAV_MODE_WORDi = (0,1) and
  BIT 30 and 31 of NAV_MODE_WORDi = (1,0)
  THEN GPS_MODE_CURRENTi = 'INIT'

```

ELSE

```

IF BIT 1 and 2 of NAV_MODE_WORDi = (0,0) and
  BIT 30 and 31 of NAV_MODE_WORDi = (1,1)
  THEN GPS_MODE_CURRENTi = 'TEST'
ELSE ILLEGAL_MODE = ON

```

3. Steps 3.1 to 3.6 are executed if a valid receiver mode was detected, as indicated by ILLEGAL_MODE set to OFF.

- 3.1 GPS stationary mode flag is extracted as follows:

```

IF BIT 4 of NAV_MODE_WORDi = 1 THEN
  STATIONARY_MODE_SOPi = ON
ELSE
  STATIONARY_MODE_SOPi = OFF

```

The STATIONARY_MODE_SOP_i, being on, indicates that the GPS receiver *i* is in STATIONARY mode during prelaunch.

- 3.2 GPS NAV filter mode is determined from bits 13 and 14 of NAV_MODE_WORD_i as follows:

```

IF GPS_MODE_CURRENTi NOT = 'NAV' THEN
  GPS_NAV_SUBMODEi = 'Ø'
ELSE
  IF BIT 13 of NAV_MODE_WORDi = 0 and BIT 14 of NAV_MODE_WORDi = 0
  THEN
    GPS_NAV_SUBMODEi = 'PVA'
  ELSE
    IF BIT 13 of NAV_MODE_WORDi = 0 and BIT 14 of
    NAV_MODE_WORDi = 1
    THEN
      GPS_NAV_SUBMODEi = 'INS'
    ELSE
      GPS_NAV_SUBMODEi = 'Ø'

```

- 3.3 If bit 16 of the NAV_MODE_WORD_i is ON, indicating that the GPS receiver has acknowledged the receipt of announced reset states and accomplished the announced reset process, then the flag GPS_ANNCD_RESET_SET_i will be set to OFF.

```

IF BIT 16 of NAV_MODE_WORDi = 1 THEN
  GPS_ANNCD_RESET_SETi = OFF

```

- 3.4 If bit 17 of the NAV_MODE_WORD_i is ON, indicating that the GPS receiver has acknowledged the receipt of initialization states and accomplished the initialization process, then the flag INIT_STATE_SET_i will be set to OFF.

IF BIT 17 of NAV_MODE_WORD_i = 1 THEN
INIT_STATE_SET_i = OFF

- 3.5 If bit 19 of the NAV_MODE_WORD_i is ON, indicating that the GPS receiver has acknowledged the receipt of initialization states for GPS NAV filter restart and accomplished NAV filter restart process and GPS_FILT_RESTART_i has been issued for more than an I-load time, DT_FR_DISP_REFRESH, then the flag GPS_FILT_RESTART_i will be set to OFF.

IF BIT 19 of NAV_MODE_WORD_i = 1 and (CLOCKTIME – T_FR_i) >
DT_FR_DISP_REFRESH
THEN
GPS_FILT_RESTART_i = OFF

D. Special Considerations:

The GPS_MODE_CURRENT_i (i = 1 to 3) shall be initialized to ‘Ø’, during all OPS transitions except between OPS1 and OPS6.

Table 4.17.6.1.1-1. NAV_MODE_WORD_DATA Input Parameters

Description	Name	Input Source	Type	Precision	Range	Units	Rate (Hz)
GPS receiver I Nav Mode Word	NAV_MODE_WORD(I)**	GPS_TO_GPC	–	–	–	–	1.04
Current clock computer time GMT	CLOCKTIME	*	F	DP	–	–	1.04
Time indicating the beginning of GPS I receiver Filter Restart process	T_FR(I)**	GPS_SOP	F	DP	–	–	1.04

* See principal function input list in table 4.17.6.10-1
** I = 1,2,3

Table 4.17.6.1.1-2. NAV_MODE_WORD_DATA Output Parameters							
Description	Name	Destination	Type	Precision	Range	Units	Rate (Hz)
Flag indicating valid GPS receiver I data	GPS_NAV_DATA_VALID(I)**	NAV_MESSAGE_DATA, GPS_DISP_COMP, GPS_TIME_ADJ	D	–	ON/OFF	–	6.25
Flag indicating GPS filter restart of receiver I is commanded	GPS_FILT_RESTART(I)**	RCVR_AIDING_DATA,*	D	–	ON/OFF	–	6.25
GPS receiver I current operating mode indicator	GPS_MODE_CURRENT(I)**	GPS_TO_GPC, GPS_DISP_COMP, RCVR_AIDING_DATA,* , GPS_TIME_ADJ	CHAR	–	TEST/INIT/NAV/‘Ø’	–	6.25
GPS receiver I navigation submode indicator	GPS_NAV_SUBMODE(I)**	GPS_DISP_COMP	CHAR	–	PVA/INS/‘Ø’	–	6.25
Flag indicating illegal GPS operating mode	ILLEGAL_MODE	GPS_TO_GPC	D	–	ON/OFF	–	6.25
Flag indicating initialization of receiver I is commanded	INIT_STATE_SET(I)**	RCVR_AIDING_DATA,*	D	–	ON/OFF	–	6.25
Flag indicating GPS announced reset of receiver I is commanded	GPS_ANNC_RESET_SET(I)**	RCVR_AIDING_DATA,*	D	–	ON/OFF	–	6.25
Flag indicating GPS receiver I in stationary mode	STATIONARY_MODE_SOP(I)**	GPS_DISP_COMP	D	–	ON/OFF	–	6.25

* See principal function output list in Table 4.17.6.10-2
** I = 1,2,3

Table 4.17.6.1.1-3. Format of NAV_MODE_WORD			
Bit Parameter	Bit no.	Bit value	Meaning
Nav data Validity flag	0	1	Valid*
Actual GPS Receiver Mode	1 - 2	<u>1</u> <u>2</u>	
		0 0	Test mode
		0 1	Initialization Mode
		1 0	Navigation Mode
		(all others)	(illegal mode)

Table 4.17.6.1.1-3. Format of NAV_MODE_WORD			
Bit Parameter	Bit no.	Bit value	Meaning
BIT in Progress	3	0	BIT in progress**
Prelaunch Standby Mode Bit	4	1	Standby mode during prelaunch
Init Data Request	5	1	Need init data
Almanac downloaded	6	1	Almanac downloaded
PPS or SPS Mode	7	1	PPS operation
		0	SPS operation
UTC Valid	8	1	UTC Valid
		0	UTC not valid
Lever arm correction	9	1	Lever arm correction used
		0	Lever arm correction not used
Spares	10 - 12		
GPS Filter Mode	13 - 14	<u>13</u> <u>14</u>	
		0 0	PVA Mode
		0 1	INS Mode
		1 0	Not Used
		1 1	Not Used
Spare	15		
Announced reset complete	16	1	Announced reset complete
Initialization states complete	17	1	Initialization states set
IMU reconfiguration complete	18	1	IMU reconfiguration complete (Not used)
Filter restart complete	19	1	Filter restart process complete
Receiver self-reset	20	1	Receiver autonomously reset
Spares	21 - 29	–	Not used
Compliment of Bit 1,2	30 - 31	<u>30</u> <u>31</u>	
		1 1	Test mode
		1 0	Initialization mode
		0 1	Navigation mode

* The Navigation Data Validity flag indicates valid NAV data when all of the following are true:
1. GPS receiver Kalman Filter estimated position uncertainty is less than 1000 meters.
2. The receiver is in NAV mode.
3. The receiver's performance monitor (PM) did not detect a critical failure in the previous second.

** Not applicable

4.17.6.1.2 NAV Message Data Processing (NAV_MESSAGE_DATA)

The purpose of the NAV_MESSAGE_DATA code is to parse the GPS_TO_GPC_DATA_ONE according to the 'NAV' message format, Message 6, as defined in Table 4.17.1.1.2-7.

A. Functional Requirements. The NAV_MESSAGE_DATA code is required to perform the following function:

1. Parse GPS data (GPS_TO_GPC_DATA_ONE) according to the 'NAV' message format.

B. Interface Requirements. The input and output parameters associated with the NAV_MESSAGE_DATA modules are contained in Table 4.17.6.1.2-1 and 4.17.6.1.2-2. I-load parameters are contained in Table 4.17.6.10-3.

C. Detailed Requirements and Formulations

The following operations shall be performed by the NAV_MESSAGE_DATA code for each receiver. Figure 4.17.6.1.2 presents the flow diagram of the NAV_MESSAGE_DATA code. Table 4.17.6.1.2-3 defines the format of the parent word, GPS_STATUS_FOM_GDOP. The definition of values taken on by FOM_NEW (figure of merit) is described in Table 4.17.6.1.2-4. Table 4.17.6.1.2-5 defines the format of the parent word, GPS_TFOM. The definition of values taken on by TFOM (time figure of merit) is described in Table 4.17.6.1.2-6.

1. Raw navigation related GPS data is parsed for downlist as follows:

$$\begin{aligned} \text{GPS_WEEK_DL}_i &= \text{GPS_TO_GPC_DATA_ONE}_{i,4} \\ \text{GPS_SEC_DL}_i &= \text{GPS_TO_GPC_DATA_ONE}_{i,5 \text{ to } 7} \\ \text{GPS_ECEF_POS_X_DL}_i &= \text{GPS_TO_GPC_DATA_ONE}_{i,8 \text{ to } 9} \\ \text{GPS_ECEF_POS_Y_DL}_i &= \text{GPS_TO_GPC_DATA_ONE}_{i,10 \text{ to } 11} \\ \text{GPS_ECEF_POS_Z_DL}_i &= \text{GPS_TO_GPC_DATA_ONE}_{i,12 \text{ to } 13} \\ \text{GPS_ECEF_VEL_X_DL}_i &= \text{GPS_TO_GPC_DATA_ONE}_{i,14 \text{ to } 15} \\ \text{GPS_ECEF_VEL_Y_DL}_i &= \text{GPS_TO_GPC_DATA_ONE}_{i,16 \text{ to } 17} \\ \text{GPS_ECEF_VEL_Z_DL}_i &= \text{GPS_TO_GPC_DATA_ONE}_{i,18 \text{ to } 19} \\ \text{GPS_STATUS_FOM_GDOP}_i &= \text{GPS_TO_GPC_DATA_ONE}_{i,20} \\ \text{PPS_STATUS_WORD}_i &= \text{GPS_TO_GPC_DATA_ONE}_{i,21} \\ \text{GPS_TFOM}_i &= \text{GPS_TO_GPC_DATA_ONE}_{i,31} \\ \text{GPS_AIDING_DATA_AGE_DL}_i &= \text{GPS_TO_GPC_DATA_ONE}_{i,32} \\ \text{GPS_AOA}_i &= \text{GPS_AIDING_DATA_AGE_DL}_i \end{aligned}$$

2. GPS channel status data is obtained by executing the CHANNEL_STATUS_DATA code as described in Section 4.17.6.1.2.1.
3. Raw GPS time tag consists of two components, T_GPS_WEEK and T_GPS_SEC, which are the GPS week number from January 6, 1980 and the seconds from the beginning of the GPS week respectively. These variables are extracted from the GPS_TO_GPC_DATA_ONE_i with appropriate scaling as defined in Table 4.17.1.1.2-7.

$$\begin{aligned} \text{T_GPS_SEC} &= \text{SCALAR}(\text{INTEGER}(\text{GPS_SEC_DL}_i)) * \text{SF_TIME} \\ \text{T_GPS_WEEK} &= \text{INTEGER}(\text{GPS_WEEK_DL}_i) \end{aligned}$$

The GPS time tag in GMT seconds is then formed by adding T_GPS_SEC, converting T_GPS_WEEK to seconds, subtracting the leap second correction, the time offset, the GPS to GMT time adjustment, and the accumulated GPC-to-GPS time adjustment as follows:

$$T_GPS_CURRENT_i = T_GPS_SEC + T_GPS_WEEK*7*SEC_PER_DAY - \\ LEAP_SEC_CORR - T_GPS_OFFSET_SECS - \\ T_GPS_TO_GMT_ADJ - (T_GPS_TIME_ADJ_ACC/1000)$$

where T_GPS_OFFSET_SECS is the number of seconds from January 6, 1980 to midnight (12 a.m.) between December 30 and 31 of the calendar year prior to the launch year.

If new GPS state is not available as indicated by the stale T_GPS_SEC value, perform step 3.1. Otherwise, continue with step 3.2 to determine the status of the newly read GPS Nav data.

IF T_GPS_SEC = T_GPS_SEC_PREV_i THEN

3.1 Set the FOM_GDOP_STATIC_i flag to OFF to indicate valid FOM and GDOP data in between GPS input updates.

FOM_GDOP_STATIC_i = OFF

3.2 Else if the GPS nav data validity flag (GPS_NAV_DATA_VALID_i) is ON and the GPS time is within I-load value (GPS_QA_TIME_THRESH) from current Orbiter clocktime and either the previous GPS state has been IMU-bracketed as indicated by IMU_BRACKET_NEW_i flag being ON or the IMU buffer is not full as indicated by GPS_IMU_BUFFER_FULL flag being OFF, then the NAV message input data processing shall continue as detailed in paragraph 3.2.1 to 3.2.9 below.

ELSE IF (GPS_NAV_DATA_VALID_i = ON and
(ABS(CLOCKTIME - T_GPS_CURRENT_i) < GPS_QA_TIME_THRESH) and
(IMU_BRACKET_NEW_i = ON or GPS_IMU_BUFFER_FULL = OFF)) THEN:

3.2.1 If T_GPS_CURRENT_i is greater than the time of the last parsed NAV message (T_GPS_NEW_i) by more than LAST_READ_DT_LIM I-load, then reset FOM_GOOD_CNT_i to 1.

IF (T_GPS_CURRENT_i - T_GPS_NEW_i) > LAST_READ_DT_LIM THEN
FOM_GOOD_CNT_i = 1

3.2.2 The IMU_BRACKET_NEW_i flag is set OFF indicating that the new GPS state is not yet determined to be IMU-bracketed.

IMU_BRACKET_NEW_i = OFF

3.2.3 The new GPS time tag is copied from the current GPS time tag.

T_GPS_NEW_i = T_GPS_CURRENT_i

3.2.4 The GPS position and velocity state output is extracted from GPS_TO_GPC_DATA_ONE_i and scaled as follows:

```

R_GPS_ECEF_NEWi,1 =
    SCALAR(INTEGER(GPS_ECEF_POS_X_DLi))*SF_POS
R_GPS_ECEF_NEWi,2 =
    SCALAR(INTEGER(GPS_ECEF_POS_Y_DLi))*SF_POS
R_GPS_ECEF_NEWi,3 =
    SCALAR(INTEGER(GPS_ECEF_POS_Z_DLi))*SF_POS

V_GPS_ECEF_NEWi,1 =
    SCALAR(INTEGER(GPS_ECEF_VEL_X_DLi))*SF_VEL
V_GPS_ECEF_NEWi,2 =
    SCALAR(INTEGER(GPS_ECEF_VEL_Y_DLi))*SF_VEL
V_GPS_ECEF_NEWi,3 =
    SCALAR(INTEGER(GPS_ECEF_VEL_Z_DLi))*SF_VEL
    
```

3.2.5 Derive GDOP and Figure of merit (FOM) values from GPS_STATUS_FOM_GDOP word as defined in Table 4.17.6.1.2-3. Set the FOM_GDOP_STATIC_i flag OFF to indicate valid FOM and GDOP data.

```

GDOP_NEWi = INTEGER (GPS_STATUS_FOM_GDOPi, 7 to 10)
FOM_NEWi = INTEGER (GPS_STATUS_FOM_GDOPi, 11 to 14)
FOM_GDOP_STATICi = OFF
    
```

3.2.6 If the FOM is less than one or greater than nine, then set the FOM to nine

```

IF (FOM_NEWi < 1 or FOM_NEWi > 9) THEN
    FOM_NEWi = 9
    
```

3.2.7 The T_GPS_SEC is saved into T_GPS_SEC_PREV_i which is used in paragraph 2 above to determine a change in GPS input buffer.

3.2.8 A first level quality assessment on the GPS NAV data collected is performed by executing the GPS_QA1_CHECK code as described in Section 4.17.6.2

3.2.9 If the Major Mode is 901, 601, or is less than 200 then store the FOM and the fail FOM flag for display.

```

IF((MM_CODE = 901) OR (MM_CODE = 601) OR (MM_CODE < 200)) THEN
    FOM_NAVi = FOM_NEWi
    GPS_FAIL_FOM_DISPi = GPS_FAIL_FOM_NEWi
    
```

D. Special Considerations:

The variables T_GPS_NEW_i and T_GPS_SEC_PREV_i (i = 1 to 3) shall be initialized to 0.

Table 4.17.6.1.2-1. NAV_MESSAGE_DATA Input Parameters

Description	Name	Input Source	Type	Precision	Range	Units	Rate (Hz)
Clock computer (GMT)	CLOCKTIME	*	F	DP	–	sec	6.25
Flag indicating that the IMU buffer is full	GPS_IMU_BUFFER_FULL	GPS_IMU_SAVE	D	–	ON/OFF	–	6.25
Flag indicating GPS NAV data validity	GPS_NAV_DATA_VALID(I)**	NAV_MODE_WORD_DATA	D	–	ON/OFF	–	1.04
GPS receiver I to GPC data buffer one	GPS_TO_GPC_DATA_ONE(I)**	*	–	–	–	–	1.04
Flag indicating GPS data has been time-bracketed by IMU data	IMU_BRACKET_NEW(I)**	GPS_IMU_SAVE	D	–	ON/OFF	–	6.25
Number of seconds from 1/6/80 to midnight between 12/30 and 12/31 of year prior to the launch year	T_GPS_OFFSET_SECS	GPS_SOP	I	DP	–	–	1.04
GPS to GMT time adjustment	T_GPS_TO_GMT_ADJ	GPS_SOP_TIME_ADJ_UPLINK	F	DP	–	sec	–
Accumulated GPC-to-GPS time adjustment	T_GPS_TIME_ADJ_ACC	GPS_TIME_ADJ	HXS	–	–	msec	6.25
Major Mode Indicator	MM_CODE	*	I	S	–	–	6.25
Flag indicating GPS failed FOM threshold	GPS_FAIL_FOM_NEW _i	GPS_QA1_CHECK	D	–	ON/OFF	–	6.25

* See principal function input list in Table 4.17.6.10-1

** I = 1,2,3 (1 to GPS_SW_CAP)

Table 4.17.6.1.2-2. NAV_MESSAGE_DATA Output Parameters							
Description	Name	Destination	Type	Precision	Range	Units	Rate (Hz)
Flag indicating GPS data has been time-bracketed by IMU data	IMU_BRACKET_NEW(I)**	GPS_IMU_SAVE	D	–	ON/OFF	–	6.25
Raw GPS week number	GPS_WEEK_DL(I)**	*	HXU	–	–	–	6.25
Raw GPS time (seconds portion)	GPS_SEC_DL(I)**	*	BSU	–	–	–	6.25
Raw GPS ECEF x position component	GPS_ECEF_POS_X_DL(I)**	*	FXS	–	–	–	6.25
Raw GPS ECEF y position component	GPS_ECEF_POS_Y_DL(I)**	*	FXS	–	–	–	6.25
Raw GPS ECEF z position component	GPS_ECEF_POS_Z_DL(I)**	*	FXS	–	–	–	6.25
Raw GPS ECEF x velocity component	GPS_ECEF_VEL_X_DL(I)**	*	FXS	–	–	–	6.25
Raw GPS ECEF y velocity component	GPS_ECEF_VEL_Y_DL(I)**	*	FXS	–	–	–	6.25
Raw GPS ECEF z velocity component	GPS_ECEF_VEL_Z_DL(I)**	*	FXS	–	–	–	6.25
Raw GPS aiding data age	GPS_AIDING_DATA_AGE_DL(I)**	*	HXS	–	–	–	6.25
GPS receiver I STATUS, FOM and GDOP word	GPS_STATUS_FOM_GDOP(I)**	GPS_DISP_COMP, *	BSU	–	–	–	6.25
Current GPS time figure of merit	GPS_TFOM(I)**	GPS_SOP_DL_COMP	I	S	–	–	6.25
Time tag of “NEW” GPS receiver I data	T_GPS_NEW(I)**	GPS_IMU_SAVE, *	F	DP	–	sec	6.25
Position vector of “NEW” GPS I data	R_GPS_ECEF_NEW(I)**	GPS_IMU_SAVE, *	V	DP	–	ft	6.25
Velocity vector of “NEW” GPS I data	V_GPS_ECEF_NEW(I)**	GPS_IMU_SAVE, *	V	S	–	ft/sec	6.25
Figure of merit associated with “NEW” GPS I data	FOM_NEW(I)**	GPS_IMU_SAVE, GPS_QA1_CHECK	I	S	0-9	–	6.25
GDOP value associated with “NEW” GPS receiver I data	GDOP_NEW(I)**	*	I	S	1-15	–	6.25
NAV FOM for receiver I	FOM_NAV(I)**	*	I	S	0-9	–	6.25
Age-of-aiding data from GPS receiver	GPS_AOA(I)**	GPS_TIME_ADJ	HXS	–	–	msec	6.25

Table 4.17.6.1.2-2. NAV_MESSAGE_DATA Output Parameters							
Description	Name	Destination	Type	Precision	Range	Units	Rate (Hz)
GPS I FOM threshold fail flag	GPS_FAIL_FOM_DISP(I)**	*	D	–	ON/OFF	–	6.25
Flag indicating static FOM/GDOP data	FOM_GDOP_STATIC(I)**	*	D	–	ON/OFF	–	6.25
Counter indicating the number of cycles FOM has been good	FOM_GOOD_CNT(I)**	GPS_QA1_CHECK	I	S	–	–	6.25
* See principal function output list in Table 4.17.6.10-2 ** I = 1,2,3 (1 to GPS_SW_CAP)							

Table 4.17.6.1.2-3. Format of GPS_STATUS_FOM_GDOP				
Bit Parameter	Bit no.	Data Type	LSB Location	Meaning
Spare	0 - 1	–	–	Not used
Incorporating < 4 meas	2	Binary	2	1 = Rcvr incorporated less than 4 meas
RPU fail	3	Binary	3	1 = Receiver processing unit failure
Spare	4	–	–	Not used
Battery low	5	Binary	5	1 = Receiver battery low
Spare	6	–	–	Not used
GDOP	7 - 10	BSU	10	Integer (4 bit) indication of Geometric Dilution of Precision
FOM	11 - 14	BSU	14	Integer (4 bit) indication of Estimated position error
Spare	15	–	–	Not used

Table 4.17.6.1.2-4. Definition of FOM	
FOM (Figure of Merit)	Estimated Position Error (Three Dimensional RSS)
1	0 - 25 meters
2	26 - 50 meters
3	51 - 75 meters
4	76 - 100 meters
5	101 - 200 meters
6	201 - 500 meters
7	501 - 1000 meters
8	1001 - 5000 meters
9	Greater than 5000 meters

Table 4.17.6.1.2-5. Format of GPS_TFOM				
Bit Parameter	Bit no.	Data Type	LSB Location	Meaning
Spare	0 - 11	–	–	Not used
TFOM	12 - 15	BSU	15	Integer (4 bit) indication of Estimated time error

Table 4.17.6.1.2-6. Definition of TFOM	
TFOM (Time Figure of Merit)	Estimated Time Error
0	Proper/Nominal Operation
1	< 1 nanosecond
2	1 to 10 nanoseconds
3	10 to 100 nanoseconds
4	100 to 1000 nanoseconds
5	1 to 10 microseconds
6	10 to 100 microseconds
7	100 to 1000 microseconds
8	1 to 10 milliseconds
9	> 10 milliseconds

4.17.6.1.2.1 GPS Channels Status Data Processing (CHANNEL_STATUS_DATA)

The purpose of the CHANNEL_STATUS_DATA code is to parse the CHANNELS_STATUS words according to the formats as defined in Table 4.17.6.1.2.1-3 through Table 4.17.6.1.2.1-11.

A. Functional Requirements. The CHANNEL_STATUS_DATA code is required to perform the following functions:

1. Retrieve GPS channel status data (CHANNELS_STATUS) from the buffer GPS_TO_GPC_DATA_ONE according to the format defined in Table 4.17.1.1.2-7.
2. Parse GPS channel status data (CHANNELS_STATUS) according to the format defined in Table 4.17.6.1.2.1-3 through Table 4.17.6.1.2.1-11.

B. Interface Requirements. The input and output parameters associated with the CHANNEL_STATUS_DATA modules are contained in Table 4.17.6.1.2.1-1 and 4.17.6.1.2.1-2.

C. Detailed Requirements and Formulations

The following operations shall be performed by the CHANNEL_STATUS_DATA code for each receiver. Figure 4.17.6.1.2.1 presents the flow diagram of the CHANNEL_STATUS_DATA code. Tables 4.17.6.1.2.1-3 through 4.17.6.1.2.1-11 define the format of the parent word CHANNEL_STATUS used in this module.

1. GPS Channel status information is extracted from word 22 to 30 of the GPS_TO_GPC_DATA_ONE as follows:

```
DO FOR k = 1 to 8
    CHANNELS_STATUSi,k = GPS_TO_GPC_DATA_ONEi,(k+21)
```

2. Parameters are extracted from CHANNELS_STATUS_i according to the format defined in Table 4.17.6.1.2.1-3 through Table 4.17.6.1.2.1-11.

Current satellite numbers (SATS_TRACKED_{i,j}) tracked by all five channels operating in State 5¹ or State 3² are obtained from bit 10 to 15 of CHANNELS_STATUS_{i,j} as follows:

```
IF BIT 4 TO 6 of CHANNELS_STATUSi,1 = BIT('101') or = BIT('011')
    SATS_TRACKEDi,1 = INTEGER(BITS 10 to 15 of CHANNELS_STATUSi,1)
    SAT_BLANKING_FLAGi,1 = OFF
ELSE
    SATS_TRACKEDi,1 = 0
IF BIT 4 TO 6 of CHANNELS_STATUSi,3 = BIT('101') or = BIT('011')
    SATS_TRACKEDi,2 = INTEGER(BITS 10 to 15 of CHANNELS_STATUSi,3)
    SAT_BLANKING_FLAGi,2 = OFF
ELSE
    SATS_TRACKEDi,2 = 0
IF BIT 4 TO 6 of CHANNELS_STATUSi,5 = BIT('101') or = BIT('011')
    SATS_TRACKEDi,3 = INTEGER(BITS 10 to 15 of CHANNELS_STATUSi,5)
    SAT_BLANKING_FLAGi,3 = OFF
ELSE
    SATS_TRACKEDi,3 = 0
```

```

IF BIT 4 TO 6 of CHANNELS_STATUSi,7 = BIT('101') or = BIT('011')
    SATS_TRACKEDi,4 = INTEGER(BITS 10 to 15 of CHANNELS_STATUSi,7)
    SAT_BLANKING_FLAGi,4 = OFF
ELSE
    SATS_TRACKEDi,4 = 0
IF BIT 4 TO 6 of CHANNELS_STATUSi,8 = BIT('101') or = BIT('011')
    SATS_TRACKEDi,5 = INTEGER(BITS 10 to 15 of CHANNELS_STATUSi,8)
    SAT_BLANKING_FLAGi,5 = OFF
ELSE
    SATS_TRACKEDi,5 = 0
    
```

1. Under state 5 operation, GPS receiver has both code and carrier tracking.
2. Under state 3 operation, GPS receiver has only code loop tracking.

D. Special Consideration:

The following two parameters shall be initialized to zero:

SATS_TRACKED_{i,6}
 CHANNELS_STATUS_{i,9} for i = 1, 2, 3

Table 4.17.6.1.2.1-1. CHANNEL_STATUS_DATA Input Parameters							
Description	Name	Input Source	Type	Precision	Range	Units	Rate (Hz)
GPS receiver I to GPC data buffer one	GPS_TO_GPC_DATA_ONE(I)**	*	-	-	-	-	1.04
Satellite blanking flag for GPS Status Display	SATS_BLANKING_FLAG(I,K)**	GPS_TO_GPC	-	-	-	-	1.04
* See principal function input list in Table 4.17.6.10-1 ** I = 1,2,3 (1 to GPS_SW_CAP), K = 1 to 6							

Table 4.17.6.1.2.1-2. CHANNEL_STATUS_DATA Output Parameters							
Description	Name	Destination	Type	Precision	Range	Units	Rate (Hz)
Channel status words for GPS Receiver I	CHANNELS_STATUS(I,J)**	*	-	-	-	-	1.04
Satellite number tracked by all six channels of GPS receiver I	SATS_TRACKED(I,K)**	*	I	S	0 - 63	-	1.04
Satellite blanking flag for GPS Status Display	SATS_BLANKING_FLAG(I,K)**	*	-	-	-	-	1.04

* See principal function output list in Table 4.17.6.10-2
** I = 1,2,3, K = 1 to 6, J = 1 to 9 (I = 1 to GPS_SW_CAP)

Table 4.17.6.1.2.1-3. Format of GPS Channel Status Word 1 (CHANNELS_STATUS(i,1))				
Bit Parameter	Bit no.	Data Type	LSB Location	Meaning
Channel 1 Fault	0	Binary	0	1 = Fault
Channel 1 Antenna	1	Binary	1	1 = Upper; 0 = Lower
Channel 1 Frequency	2	Binary	2	1 = L2; 0 = L1
Channel 1 Ephemeris	3	Binary	3	1 = Ephemeris downloaded
Channel 1 State	4 - 6	UFXP	6	Integer (3 bit) indication of channel operating state
Channel 1 Hardware Channel No.	7 - 9	UFXP	9	Integer (3 bit) indication of receiver physical channel
Channel 1 Satellite Number	10 - 15	UFXP	15	Integer (6 bit) indication of PRN number of satellite being tracked

Table 4.17.6.1.2.1-4. Format of GPS Channel Status Word 2 (CHANNELS_STATUS(i,2))				
Bit Parameter	Bit no.	Data Type	LSB Location	Meaning
Channel 1 Signal to Noise Ratio, C/No	0 - 5	UFXP	5	Integer (6 bit) indication of signal to noise ratio with LSB to be 1 dB Hz
Channel 1 Code Tracking Status	6 - 7	UFXP	7	Integer (2 bit) indication of the type of tracking code 0 = P Code, 2 = C/A Code 1 = Y Code
Channel 2 Signal to Noise Ratio, C/No	8 - 13	UFXP	13	Integer (6 bit) indication of signal to noise ratio with LSB to be 1 dB Hz
Channel 2 Code Tracking Status	14 - 15	UFXP	15	Integer (2 bit) indication of the type of tracking code 0 = P Code, 2 = C/A Code 1 = Y Code

Table 4.17.6.1.2.1-5. Format of GPS Channel Status Word 3 (CHANNELS_STATUS(i,3))				
Bit Parameter	Bit no.	Data Type	LSB Location	Meaning
Channel 2 Fault	0	Binary	0	1 = Fault
Channel 2 Antenna	1	Binary	1	1 = Upper; 0 = Lower
Channel 2 Frequency	2	Binary	2	1 = L2; 0 = L1
Channel 2 Ephemeris	3	Binary	3	1 = Ephemeris downloaded
Channel 2 State	4 - 6	UFXP	6	Integer (3 bit) indication of channel operating state
Channel 2 Hardware Channel No.	7 - 9	UFXP	9	Integer (3 bit) indication of receiver physical channel
Channel 2 Satellite Number	10 - 15	UFXP	15	Integer (6 bit) indication of PRN number of satellite being tracked

Table 4.17.6.1.2.1-6. Format of GPS Channel Status Word 4 (CHANNELS_STATUS(i,4))				
Bit Parameter	Bit no.	Data Type	LSB Location	Meaning
Channel 3 Signal to Noise Ratio, C/No	0 - 5	UFXP	5	Integer (6 bit) indication of signal to noise ratio with LSB to be 1 dB Hz
Channel 3 Code Tracking Status	6 - 7	UFXP	7	Integer (2 bit) indication of the type of tracking code 0 = P Code, 2 = C/A Code 1 = Y Code
Channel 4 Signal to Noise Ratio, C/No	8 - 13	UFXP	13	Integer (6 bit) indication of signal to noise ratio with LSB to be 1 dB Hz
Channel 4 Code Tracking Status	14 - 15	UFXP	15	Integer (2 bit) indication of the type of tracking code 0 = P Code, 2 = C/A Code 1 = Y Code

Table 4.17.6.1.2.1-7. Format of GPS Channel Status Word 5 (CHANNELS_STATUS(i,5))				
Bit Parameter	Bit no.	Data Type	LSB Location	Meaning
Channel 3 Fault	0	Binary	0	1 = Fault
Channel 3 Antenna	1	Binary	1	1 = Upper; 0 = Lower
Channel 3 Frequency	2	Binary	2	1 = L2; 0 = L1
Channel 3 Ephemeris	3	Binary	3	1 = Ephemeris downloaded
Channel 3 State	4 - 6	UFXP	6	Integer (3 bit) indication of channel operating state
Channel 3 Hardware Channel No.	7 - 9	UFXP	9	Integer (3 bit) indication of receiver physical channel
Channel 3 Satellite Number	10 - 15	UFXP	15	Integer (6 bit) indication of PRN number of satellite being tracked

Table 4.17.6.1.2.1-8. Format of GPS Channel Status Word 6 (CHANNELS_STATUS(i,6))				
Bit Parameter	Bit no.	Data Type	LSB Location	Meaning
Channel 5 Signal to Noise Ratio, C/No	0 - 5	UFXP	5	Integer (6 bit) indication of signal to noise ratio with LSB to be 1 dB Hz
Channel 5 Code Tracking Status	6 - 7	UFXP	7	Integer (2 bit) indication of the type of tracking code 0 = P Code, 2 = C/A Code 1 = Y Code
Channel 1 Pseudo Range Measurement Edit	8	Binary	8	1 = Edit
Channel 2 Pseudo Range Measurement Edit	9	Binary	9	1 = Edit
Channel 3 Pseudo Range Measurement Edit	10	Binary	10	1 = Edit
Channel 4 Pseudo Range Measurement Edit	11	Binary	11	1 = Edit
Channel 1 Delta Range Measurement Edit	12	Binary	12	1 = Edit
Channel 2 Delta Range Measurement Edit	13	Binary	13	1 = Edit
Channel 3 Delta Range Measurement Edit	14	Binary	14	1 = Edit
Channel 4 Delta Range Measurement Edit	15	Binary	15	1 = Edit

Table 4.17.6.1.2.1-9. Format of GPS Channel Status Word 7 (CHANNELS_STATUS(i,7))				
Bit Parameter	Bit no.	Data Type	LSB Location	Meaning
Channel 4 Fault	0	Binary	0	1 = Fault
Channel 4 Antenna	1	Binary	1	1 = Upper; 0 = Lower
Channel 4 Frequency	2	Binary	2	1 = L2; 0 = L1
Channel 4 Ephemeris	3	Binary	3	1 = Ephemeris downloaded
Channel 4 State	4 - 6	UFXP	6	Integer (3 bit) indication of channel operating state
Channel 4 Hardware Channel No.	7 - 9	UFXP	9	Integer (3 bit) indication of receiver physical channel
Channel 4 Satellite Number	10 - 15	UFXP	15	Integer (6 bit) indication of PRN number of satellite being tracked

Table 4.17.6.1.2.1-10. Format of GPS Channel Status Word 8 (CHANNELS_STATUS(i,8))				
Bit Parameter	Bit no.	Data Type	LSB Location	Meaning
Channel 5 Fault	0	Binary	0	1 = Fault
Channel 5 Antenna	1	Binary	1	1 = Upper; 0 = Lower
Channel 5 Frequency	2	Binary	2	1 = L2; 0 = L1
Channel 5 Ephemeris	3	Binary	3	1 = Ephemeris downloaded
Channel 5 State	4 - 6	UFXP	6	Integer (3 bit) indication of channel operating state
Channel 5 Hardware Channel No.	7 - 9	UFXP	9	Integer (3 bit) indication of receiver physical channel
Channel 5 Satellite Number	10 - 15	UFXP	15	Integer (6 bit) indication of PRN number of satellite being tracked

Table 4.17.6.1.2.1-11. Format of GPS Channel Status Word 9 (CHANNELS_STATUS(i,9))				
Bit Parameter	Bit no.	Data Type	LSB Location	Meaning
	0 - 15	–	–	Not Applicable

4.17.6.1.3 INIT Message Data Processing (INIT_MESSAGE_DATA)

The purpose of the INIT_MESSAGE_DATA code is to parse the GPS_TO_GPC_DATA_ONE according to the 'INIT' message format, Message 5, as defined in Table 4.17.1.1.2-6.

A. Functional Requirements. The INIT_MESSAGE_DATA code is required to perform the following functions:

1. Parse GPS data (GPS_TO_GPC_DATA_ONE) according to the 'INIT' message format.

B. Interface Requirements. The input and output parameters associated with the INIT_MESSAGE_DATA modules are contained in Table 4.17.6.1.3-1 and 4.17.6.1.3-2. I-load parameters are contained in Table 4.17.6.10-3.

C. Detailed Requirements and Formulations

The following operations shall be performed by the INIT_MESSAGE_DATA code for each receiver. Figure 4.17.6.1.3 presents the flow diagram of the INIT_MESSAGE_DATA code. Table 4.17.6.1.3-3 defines the format for the parent word PPS_STATUS_WORD.

1. The PPS status word and Almanac reference week are derived as follows:

$$\begin{aligned} \text{PPS_STATUS_WORD}_i &= \text{GPS_TO_GPC_DATA_ONE}_{i,8} \\ \text{ALMANAC_REF_WEEK}_i &= \text{GPS_TO_GPC_DATA_ONE}_{i,9} \end{aligned}$$

Table 4.17.6.1.3-1. INIT_MESSAGE_DATA Input Parameters							
Description	Name	Input Source	Type	Precision	Range	Units	Rate (Hz)
GPS receiver I to GPC data buffer one	GPS_TO_GPC_DATA_ONE(I)**	*	-	-	-	-	1.04
* See principal function input list in Table 4.17.6.10-1 ** I = 1,2,3 (1 to GPS_SW_CAP)							

Table 4.17.6.1.3-2. INIT_MESSAGE_DATA Output Parameters							
Description	Name	Destination	Type	Precision	Range	Units	Rate (Hz)
GPS receiver I PPS status	PPS_STATUS_WORD(I)**	*	D	-	-	-	6.25
GPS receiver I almanac week number	ALMANAC_REF_WEEK(I)**	*	I	S	-	-	6.25
* See principal function output list in Table 4.17.6.10-2 ** I = 1,2,3 (1 to GPS_SW_CAP)							

Table 4.17.6.1.3-3. Definition of PPS_STATUS_WORD			
Bit Parameter	Bit no.	Bit value	Meaning*
Daily Key In Use (Verified)	0	1	CKw User: Key for the current day has been verified against SV subframe data. GUK User: Key for the current day has been obtained from and verified against the SV subframe data.
		0	CKw User: The receiver does not contain a verified key for the current day. GUK User: The receiver does not contain a verified key for the current day.
Daily Key In Use (Unverified)	1	1	CKw User: Key entered is in use but has not been verified against SV subframe data. GUK User: The receiver is waiting for subframe data and does not contain a key for the current day.
		0	CKw User: Key for the current day has been verified or the key for the current day is incorrect or not loaded GUK User: Key for the current day has been obtained and verified or the GUK issue number is invalid or expired.
Daily Key In Use (Incorrect)	2	1	CKw User: The receiver does not contain a key for the current day. The verification process will not be attempted if the key loaded is not for the current week based on receiver time. GUK User: The issue number is invalid. The receiver does not contain a key for the current day.
		0	CKw User: Set when the key for the current day is entered or updated. GUK User: Set when a new GUK is entered.
Key Failed Parity	3	1	If any key entered contains a parity error.
		0	If new keys are entered that pass the parity test.
GUK User:	4	1	Receiver using GUVs.
2 Hour Alert	5	1	Indicating that there is 2 hours or less before expiration of the current key if the subsequent key is not available.
Insufficient Key	6	1	Indicating that insufficient consecutive keys have been loaded to accomplish the Mission Duration.
Erase Failure	7	1	Indicating that the receiver failed to erase or zeroize a portion of memory.
Receiver Contains Keys	8	1	Key entry successful.
Mission Duration	9 - 15	0 - 44	Representing the number of consecutive days for which the receiver has keys (from the current day) Bit 9 is MSB and bit 15 is LSB.
<p>* CKw - Crypto Key Weekly GUK - Group Unique Key GUV - Group Unique Variable</p>			

4.17.6.1.4 TEST Message Data Processing (TEST_MESSAGE_DATA)

The purpose of the TEST_MESSAGE_DATA code is to parse the GPS_TO_GPC_DATA_ONE according to the ‘TEST’ message format, Message 4, as defined in Table 4.17.1.1.2-5.

A. Functional Requirements. The TEST_MESSAGE_DATA code is required to perform the following functions:

1. Parse GPS data (GPS_TO_GPC_DATA_ONE) according to the ‘TEST’ message format.

B. Interface Requirements. The input and output parameters associated with the TEST_MESSAGE_DATA modules are contained in Table 4.17.6.1.4-1 and 4.17.6.1.4-2. I-load parameters are contained in Table 4.17.6.10-3.

C. Detailed Requirements and Formulations

The following operations shall be performed by the TEST_MESSAGE_DATA code for each receiver. Figure 4.17.6.1.4 presents the flow diagram of the TEST_MESSAGE_DATA code.

1. The GPS self-test results as recorded in BIT_FAULT_LOG_WORD’s are extracted from GPS_TO_GPC_DATA_ONE locations defined in Table 4.17.1.1.2-5 TEST message format.

For k = 1 to 18

$$\text{BIT_FAULT_LOG_WORD}_{i,k} = \text{GPS_TO_GPC_DATA_ONE}_{i,(k+3)}$$

2. The self-test fail flag is extracted from BIT_FAULT_LOG_WORD_{i,1} as follows:
IF BIT 4 of BIT_FAULT_LOG_WORD_{i,1} = 1 THEN GPS_FAIL_ST_FLAG_i = ON
ELSE GPS_FAIL_ST_FLAG_i = OFF

D. Special Considerations

The GPS_FAIL_ST_FLAG_i (i=1,2,3) flags shall be initialized to OFF, and shall retain their current values across all OPS transitions.

Table 4.17.6.1.4-1. TEST_MESSAGE_DATA Input Parameters							
Description	Name	Input Source	Type	Precision	Range	Units	Rate (Hz)
GPS receiver I to GPC data buffer one	GPS_TO_GPC_DATA_ONE(I)**	*	–	–	–	–	1.04
* See principal function input list in Table 4.17.6.10-1 ** I = 1,2,3 (1 to GPS_SW_CAP)							

Table 4.17.6.1.4-2. TEST_MESSAGE_DATA Output Parameters							
Description	Name	Destination	Type	Precision	Range	Units	Rate (Hz)
GPS receiver I test log words	BIT_FAULT_LOG_WORD(I,K)**	*	-	-	-	-	6.25
GPS receiver I fail self test flag	GPS_FAIL_ST_FLAG(I)	GPS_SOP_DL_COMP GPS_DISP_COMP	D	-	ON/OFF	-	6.25
* See principal function output list in Table 4.17.6.10-2 ** I = 1,2,3 K = 1 to 18 (I = 1 to GPS_SW_CAP)							

4.17.6.2 GPS Data Quality Check Processing (GPS_QA1_CHECK)

The purpose of the GPS data quality check function is to provide the first level of redundancy management for GPS states based on the figures of merit provided by the receivers.

A. Functional Requirements. The GPS_QA1_CHECK module shall perform the following function:

1. Determine when the receiver figure of merit passes the first level quality assessment check (QA1). Passing QA1 requires the receiver figure of merit to be less than or equal to the I-loaded threshold value for an I-loaded number of consecutive cycles.

B. Interface Requirements. The input and output parameters associated with the GPS_QA1_CHECK module are contained in Table 4.17.6.2-1 and 4.17.6.2-2. I-load parameters are contained in Table 4.17.6.10-3.

C. Detailed Requirements and Formulations. The following operations shall be performed by the GPS_QA1_CHECK module. Figure 4.17.6.2 presents a flow diagram of the GPS_QA1_CHECK functional requirements.

- 1.0 If the GPS figure of merit (FOM_NEW_i) is less than or equal to the I-loaded threshold value (MAX_FOM_ACCEPT_PFS), then execute step 1.1. Otherwise, set the flag GPS_FAIL_FOM_NEW_i to ON and reset FOM_GOOD_CNT_i to 1.

IF FOM_NEW_i ≤ MAX_FOM_ACCEPT_PFS THEN

Execute step 1.1

ELSE

GPS_FAIL_FOM_NEW_i = ON

FOM_GOOD_CNT_i = 1

- 1.1 If FOM_GOOD_CNT_i is greater than or equal to the I-loaded minimum number of consecutive cycles (FOM_GOOD_COUNT_MIN), then reset GPS_FAIL_FOM_NEW_i to OFF. Otherwise, set GPS_FAIL_FOM_NEW_i to ON and increment FOM_GOOD_CNT_i by 1.

```

IF FOM_GOOD_CNTi ≥ FOM_GOOD_COUNT_MIN THEN
    GPS_FAIL_FOM_NEWi = OFF
ELSE
    GPS_FAIL_FOM_NEWi = ON
    FOM_GOOD_CNTi = FOM_GOOD_CNTi + 1

```

D. Special Considerations. FOM_GOOD_CNT_i shall be initialized to 1 after all OPS transitions except between OPS 1 and OPS 6.

Table 4.17.6.2-1. GPS_QA1_CHECK Input Parameters							
Description	Name	Input Source	Type	Precision	Range	Units	Rate (Hz)
Counter indicating the number of cycles FOM has been good	FOM_GOOD_CNT(I)**	NAV_MESSAGE_DATA	I	S	–	–	1.04
Figure of merit associated with “NEW” GPS I data	FOM_NEW(I)**	NAV_MESSAGE_DATA	I	S	–	–	1.04
** I = 1,2,3 (1 to GPS_SW_CAP)							

Table 4.17.6.2-2. GPS_QA1_CHECK Output Parameters							
Description	Name	Destination	Type	Precision	Range	Units	Rate (Hz)
Flag indicating GPS failed FOM threshold	GPS_FAIL_FOM_NEW(I)**	GPS_IMU_SAVE, NAV_MESSAGE_DATA	D	–	ON/OFF	–	6.25
** I = 1,2,3 (1 to GPS_SW_CAP)							

4.17.6.3 GPS Altitude Aiding Data Processing (GPS_ALT_DATA)

The purpose of the GPS altitude aiding data processing function is to process data from the air data system and/or radar altimeter for eventual altitude aiding to the GPS receivers at 6.25 Hz rate.

A. Functional Requirements. The GPS_ALT_DATA module is required to perform the following functions:

1. Determine if BARO and radar altimeter altitude data is valid. If they are good then they shall be collected and prepared as altitude aiding data to all GPS receivers.
2. Collect, store and process data from ADTA SOP and RA SOP.

B. Interface Requirements. The input and output parameters associated with GPS_ALT_DATA module are contained in Tables 4.17.6.3-1 and 4.17.6.3-2. I-load parameters are contained in Table 4.17.6.10-3.

C. Detailed Requirements and Formulations. The following operations shall be performed by the GPS_ALT_DATA module. Figure 4.17.6.3 presents a flow diagram of the GPS_ALT_DATA functional requirements.

1. In order to avoid unnecessary computation, the following step 2 and step 3 are performed only if BARO time tag is changed, indicating new BARO data is received.

IF T_BARO NOT = T_BARO_PREV THEN do step 2 and 3 below.

2. A five part test shall be made to determine if the GPS_BARO_DG_OUT flag shall be set ON or OFF. If the Major Mode is not 304, not 305, not 602 and not 603 or BARO_AIF flag is in 'INHIBIT' or the pressure altitude data good flag, BARO_DG, is OFF or the Mach jump flag, MACH_JUMP, is ON or the earth relative vehicle velocity magnitude, REL_VEL_MAG, is greater than the maximum velocity for BARO processing, RV_MAX, or the altitude of the vehicle wheels, ALT_WHEELS, is less than the BARO processing cutoff low altitude, ALT_BARO_CUTOFF_LOW, then the GPS_BARO_DG_OUT flag is set to OFF, otherwise GPS_BARO_DG_OUT is set to ON and the following altitude aiding information is computed from air data below.

The time tag and pressure corrected altitude from the ADTA SOP shall be processed as follows:

TOTAL_SEC = T_BARO + LEAP_SEC_CORR + T_GPS_OFFSET_SECS –
TIME_BIAS_BARO + T_GPS_TO_GMT_ADJ +
(T_GPS_TIME_ADJ_ACC/1000)
T_GPS_BARO_SEC_OUT = MOD(TOTAL_SEC, (7*SEC_PER_DAY))
GPS_BARO_DATA_OUT = Q_BARO
T_GPS_BARO_WEEK_OUT = FLOOR(TOTAL_SEC/(7*SEC_PER_DAY))

3. The new BARO time is saved into previous buffer for use in step 1 above on the next cycle.

T_BARO_PREV = T_BARO

4. If the crew has not enabled radar altimeter aiding, GPS_RA_RCVD = OFF (via Item 46 on the GPS single string configuration HSD (SPEC 50)) or the Major mode is not 305 and not 603 or vehicle altitude (ALT_WHEELS) is outside the I-loaded ranges of operation, ALT_RA_CUTOFF_HIGH and ALT_RA_CUTOFF_LOW, then the data good flag for the GPS radar altimeter aiding, GPS_RADALT_DG_OUT, will be set OFF. Otherwise, the data good flag for the GPS radar altimeter aiding will be set to the value of the data good flag, RADALT_DG, from the radar altimeter SOP. If the data good flag, RADALT_DG, is ON then the altitude information is computed from radar data below and the BARO data good flag, GPS_BARO_DG_OUT, shall be set OFF.

Note, GPS_RA_RCVD can only be set by the single string HSD.

Note, radar altimeter aiding takes precedence over BARO data aiding.

The time tag and data from the radar altimeter SOP shall be processed as follows:

```
IF RADALT_DG = ON THEN
    TOTAL_SEC = T_RADALT + LEAP_SEC_CORR + T_GPS_OFFSET_SECS -
        TIME_BIAS_RADALT + T_RA_GPS_DELAY + T_GPS_TO_GMT_ADJ +
        (T_GPS_TIME_ADJ_ACC/1000)
    T_GPS_RADALT_SEC_OUT = MOD(TOTAL_SEC, (7*SEC_PER_DAY))
    GPS_RADALT_DATA_OUT = (RADALT_HEIGHT + DELH_RA_GPS +
        H_DOT_ELLIPSOID*T_RA_GPS_DELAY)
    T_GPS_RADALT_WEEK_OUT = FLOOR(TOTAL_SEC/(7*SEC_PER_DAY))
    GPS_BARO_DG_OUT = OFF
```

D. Special Considerations:

The flags GPS_BARO_DG_OUT and GPS_RADALT_DG_OUT, shall be initialized to OFF.

The parameter, T_BARO_PREV, shall be initialized to 0.

The I-load ALT_RA_CUTOFF_HIGH is also used to disable radar altimeter aiding on the Horizontal Situation Display by having a value less than or equal to zero. Radar altimeter aiding can only be enabled on the single string HSD.

Table 4.17.6.3-1. GPS_ALT_DATA Input Parameters

Description	Name	Input Source	Type	Precision	Range	Units	Rate (Hz)
ADS corrected pressure altitude	Q_BARO	*	F	S	–	ft	1.04
ADS corrected pressure altitude time tag	T_BARO	*	F	DP	–	sec	1.04
Pressure altitude data good	BARO_DG	*	D	–	ON/ OFF	–	1.04
Flag indicating (ON) that the Shuttle is in the MACH jump region	MACH_JUMP	*	D	–	ON/ OFF	–	1.04
Magnitude of vehicle relative velocity	REL_VEL_MAG	*	F	S	–	ft/sec	1.04
Altitude to rear wheels	ALT_WHEELS	*	F	S	–	ft	6.25
Altitude Rate	H_DOT_ELLIPSOID	*	F	S	–	ft/sec	6.25
Radar Altitude Data Good	RADALT_DG	*	D	–	ON/ OFF	–	6.25
Crew initiated radar altimeter aiding flag	GPS_RA_RCVD	*	D	–	ON/ OFF	–	6.25
Radar Altitude measurement	RADALT_HEIGHT	*	F	S	–	ft	6.25
Radar Altitude time tag	T_RADALT	*	F	DP	–	sec	6.25
Number of seconds from 1/6/80 to midnight between 12/30 and 12/31 of year prior to launch year	T_GPS_OFFSET_SECS	GPS_SOP	I	DP	–	sec	–
BARO auto/inhibit/force switch	BARO_AIF	*	I	S	(0)AUTO/ (1)INHIBIT/ (2)FORCE	–	1.04
Major mode code number	MM_CODE	*	I	S	–	–	6.25
GPS to GMT time adjustment	T_GPS_TO_GMT_ADJ	GPS_SOP_TIME_ADJ_UPLINK	F	DP	–	sec	–
Accumulated GPC-to-GPS time adjustment	T_GPS_TIME_ADJ_ACC	GPS_TIME_ADJ	HXS	–	–	msec	6.25

* See principal function input list in Table 4.17.6.10-1.

Table 4.17.6.3-2. GPS_ALT_DATA Output Parameters

Description	Name	Output Destination	Type	Precision	Range	Units	Rate (Hz)
Baro Data Good	GPS_BARO_DG_OUT	RCVR_AIDING_DATA	D	–	ON/OFF	–	6.25
Baro Measurement	GPS_BARO_DATA_OUT	NAV_AIDING_DATA	F	S	–	ft	6.25
Baro Time Tag (Weeks)	T_GPS_BARO_WEEK_OUT	NAV_AIDING_DATA	I	S	–	week	6.25
Baro Time Tag (Seconds)	T_GPS_BARO_SEC_OUT	NAV_AIDING_DATA	F	DP	–	sec	6.25
Radar Altitude Data Good	GPS_RADALT_DG_OUT	RCVR_AIDING_DATA	D	–	ON/OFF	–	6.25
Radar Altitude measurement	GPS_RADALT_DATA_OUT	NAV_AIDING_DATA	F	S	–	ft	6.25
Radar Altitude time tag (Weeks)	T_GPS_RADALT_WEEK_OUT	NAV_AIDING_DATA	I	S	–	week	6.25
Radar Altitude time tag (Seconds)	T_GPS_RADALT_SEC_OUT	NAV_AIDING_DATA	F	DP	–	sec	6.25

4.17.6.4 GPC To GPS Data Processing (GPC_TO_GPS)

The purpose of the GPC to GPS data processing function is to format, scale and convert data from the GPS SOP for the GPS receivers. This function is required to form the appropriate outgoing messages according to the GPS receiver mode. The Receiver Aiding Data message (GPS Control Word and aiding state vector) is always output to the GPS receiver. The Receiver Aiding Data message (Attitude and Altitude aiding data) is output whenever the GPS receiver is in NAV mode. The Antenna Correction Data message (antenna lever arm/boresight data and residual edit criteria) is output whenever the GPS receiver is in INIT mode.

A. Functional Requirements. The GPC_TO_GPS module is required to perform the following functions:

1. Collect data to be processed and sent to GPS receivers.
2. Data scaling and conversion into format suitable for GPS receivers.
3. Packing of GPS control word.

B. Interface Requirements. The input and output parameters associated with GPC_TO_GPS module are contained in Tables 4.17.6.4-1 and 4.17.6.4-2. I-load parameters are contained in Table 4.17.6.10-3.

C. Detailed Requirements and Formulations. The following operations shall be performed by the GPC_TO_GPS module. Figure 4.17.6.4 through 4.17.6.4.3 present flow diagrams of the GPC_TO_GPS functional requirements.

1. If the Major Mode is not 901, and one of the aiding data valid flags is ON:

MM_CODE NOT = 901
and
(AIDING_DATA_VALID₁ = ON
or
AIDING_DATA_VALID₂ = ON
or
AIDING_DATA_VALID₃ = ON)

The M50 to WGS84 coordinate transformation matrix for the time of the GPS reference states is computed.

$$M_M50_TO_WGS84 = \text{TRANPOSE}(\text{EARTH_FIXED_TO_M50_COORD} \\ (\text{T_CURRENT_GPS_REF}) * M_WGS84_TO_EF)$$

The reference state in WGS84 coordinates is computed and stored in intermediate variables.

$$\bar{R}_REF_ECEF = M_M50_TO_WGS84 \bar{R}_REF$$

$$\bar{V}_REF_ECEF = M_M50_TO_WGS84 \bar{V}_REL(\bar{V}_REF, \bar{R}_REF)$$

2. The GPC_TO_GPS module shall pack the GPC_TO_GPS_DATA_ONE and GPC_TO_GPS_DATA_TWO buffers for each of the receivers in the format as defined by the Table 4.17.1.1.2-1 through 4.17.1.1.2-4.
 - 2.1 Independent of the GPS operating mode, the GPC_TO_GPS module shall execute the RCVR_AIDING_DATA code to pack the GPC_TO_GPS_DATA_ONE_i buffer with state aiding data to the GPS receivers. The RCVR_AIDING_DATA code is described in detail in Section 4.17.6.4.1.
 - 2.2 Depending on the GPS SOP commanded GPS mode of each GPS receiver, data is packed into the buffers differently. If GPS_MODE_CMD_SOP_i is in 'NAV' mode then the GPC_TO_GPS module will execute NAV_AIDING_DATA code to pack the GPC_TO_GPS_DATA_TWO_i buffer with attitude and altitude aiding data to the GPS receivers. The NAV_AIDING_DATA code is described in detail in Section 4.17.6.4.3.
 - 2.3 If GPS_MODE_CMD_SOP_i is in 'INIT' mode then the GPC_TO_GPS module will execute INIT_DATA code to pack the GPC_TO_GPS_DATA_TWO_i buffer with antenna location, boresight direction information, and residual edit criteria to the GPS receivers. The INIT_DATA code is described in detail in Section 4.17.6.4.2.
 - 2.4 If GPS_MODE_CMD_SOP_i is neither 'INIT' or 'NAV' mode, the GPC_TO_GPS_DATA_TWO_i buffer will be zeroed.

D. Special Considerations:

1. The EARTH_FIXED_TO_M50_COORD coordinate transformation function is defined in the Level C Functional Subsystem Software Requirements Guidance, Navigation, and Control Part B, Ascent/RTLS Navigation, Section 4.4.2, Coordinate Transformations.

2. The V_REL function is defined in the Level C Functional Subsystem Software Requirements Guidance, Navigation, and Control Part B, Entry Navigation.

Table 4.17.6.4-1. GPC_TO_GPS Input Parameters							
Description	Name	Input Source	Type	Precision	Range	Units	Rate (Hz)
Flag indicating (ON) valid aiding reference states	AIDING_DATA_VALID(I)**	*	D	–	ON/OFF	–	6.25
Current GPS reference position state	\bar{R} _REF	*	V	DP	–	ft	6.25
Current GPS reference velocity state	\bar{V} _REF	*	V	DP	–	ft/sec	6.25
Time tag for GPS reference states	T_CURRENT_GPS_REF	*	F	DP	–	sec	6.25
GPS SOP commanded GPS Mode	GPS_MODE_CMD_SOP(I)**	GPS_SOP	I	S	0(TEST) 1(INIT) 2(NAV)	–	1.04
Transformation matrix from mean equator and equinox of 1950 to true equator and Greenwich Meridian of date	M_M50TOEF_AT_EPOCH	*	M	DP	±1.0	–	6.25
Major Mode Code	MM_CODE	*	I	S	–	–	6.25
Time tag for M_M50TOEF_AT_EPOCH	T_EPOCH	*	F	DP	–	–	6.25
* See principal function input list in Table 4.17.6.10-1 ** I = 1,2,3 (I = 1 to GPS_SW_CAP)							

Table 4.17.6.4-2. GPC_TO_GPS Output Parameters

Description	Name	Destination	Type	Precision	Range	Units	Rate (Hz)
GPC to GPS receiver I data buffer two	GPC_TO_GPS_DATA_TWO(I)**	*	-	-	-	-	6.25
M50 to WGS84 coordinate transformation	M_M50_TO_WGS84	RCVR_AIDING_DATA	M	DP	-	-	6.25
Current GPS reference position state in WGS84 coordinates	\bar{R} _REF_ECEF	RCVR_AIDING_DATA	V	DP	-	ft	6.25
Current GPS reference velocity state in WGS84 coordinates	\bar{V} _REF_ECEF	RCVR_AIDING_DATA	V	DP	-	ft/sec	6.25

* See principal function output list in Table 4.17.6.10-2

** I = 1,2,3 (I = 1 to GPS_SW_CAP)

4.17.6.4.1 GPS Receiver Aiding Data Processing (RCVR_AIDING_DATA)

The purpose of the receiver aiding data processing function is to format, scale and convert data for packing the GPC_TO_GPS_DATA_ONE buffer which provides receiver with control word, state aiding, and crew manual satellite deselection.

A. Functional Requirements. The RCVR_AIDING_DATA code is required to perform the following functions:

1. Packing of GPS control word.
2. Collection of state and attitude aiding data to be processed.
3. Data scaling and conversion into format suitable for GPS receivers.

B. Interface Requirements. The input and output parameters associated with RCVR_AIDING_DATA module are contained in Table 4.17.6.4.1-1 and 4.17.6.4.1-2. I-load parameters are contained in Table 4.17.6.10-3.

C. Detailed Requirements and Formulations.

The following operations shall be performed by the RCVR_AIDING_DATA module. Figure 4.17.6.4.1 presents the flow diagram of the RCVR_AIDING_DATA functional requirements. Table 4.17.6.4.1-3 defines the format of the parent word GPS_CONTROL_WORD used in this module.

1. The GPS_CONTROL_WORD shall be packed as follows according to the definitions listed in Table 4.17.6.4.1-3.

- 1.1 GPS_MODE_CMD_SOP_i will be used to set bit 0, 1, 30 and 31 of GPS_CONTROL_WORD.

IF GPS_MODE_CMD_SOP_i = NAV THEN

BIT 0 and 1 of GPS_CONTROL_WORD = (1,0) and

BIT 30 and 31 of GPS_CONTROL_WORD = (0,1)

ELSE

IF GPS_MODE_CMD_SOP_i = INIT THEN

BIT 0 and 1 of GPS_CONTROL_WORD = (0,1) and

BIT 30 and 31 of GPS_CONTROL_WORD = (1,0)

ELSE

IF GPS_MODE_CMD_SOP_i = TEST THEN

BIT 0 and 1 of GPS_CONTROL_WORD = (0,0) and

BIT 30 and 31 of GPS_CONTROL_WORD = (1,1)

- 1.2 Bits 2 through 15 of GPS_CONTROL_WORD will be set as follows:

IF GPS_ANNC_D_RESET_SET_i = ON THEN

BIT 2 of GPS_CONTROL_WORD = 1

ELSE

BIT 2 of GPS_CONTROL_WORD = 0

IF GPS_BARO_DG_OUT = ON and GPS_MODE_CMD_SOP_i = 'NAV' THEN
 BIT 3 of GPS_CONTROL_WORD = 1

ELSE
 BIT 3 of GPS_CONTROL_WORD = 0

IF GPS_RADALT_DG_OUT = ON and GPS_MODE_CMD_SOP_i = 'NAV' THEN
 BIT 4 of GPS_CONTROL_WORD = 1

ELSE
 BIT 4 of GPS_CONTROL_WORD = 0

IF AIDING_DATA_VALID_i = ON THEN
 BIT 6 of GPS_CONTROL_WORD = 1

ELSE
 BIT 6 of GPS_CONTROL_WORD = 0

IF INIT_STATE_SET_i = ON THEN
 BIT 8 of GPS_CONTROL_WORD = 1

ELSE
 BIT 8 of GPS_CONTROL_WORD = 0

IF GPS_MODE_CMD_SOP_i = 'INIT' THEN
 BIT 10 of GPS_CONTROL_WORD = 1

ELSE
 BIT 10 of GPS_CONTROL_WORD = 0

IF GPS_FILT_RESTART_i = ON THEN
 BIT 15 of GPS_CONTROL_WORD = 1

ELSE
 BIT 15 of GPS_CONTROL_WORD = 0

2. GPC major mode number is packed into bits 17 to 27 of GPS_CONTROL_WORD.

 BIT 17 to 27 of GPS_CONTROL_WORD = BINARY(MM_CODE)

3. The major mode shall be checked to monitor whether it is in OPS9 post OPS305 or OPS603.

 IF (MM_CODE = 901 and (MM_CODE_PREV = 305 or MM_CODE_PREV = 603))
 THEN OPS9_ROLLOUT_COMPLETE = ON

 where OPS9_ROLLOUT_COMPLETE shall be initialized to OFF.

The current major mode shall also be saved for next pass.

 MM_CODE_PREV = MM_CODE

4. State and attitude aiding data are collected next. During GN&C OPS9 prior to OPS1, the aiding states to the GPS receivers are the GROUND IMU DISP attitude and aiding states which are all time tagged to the most current Clocktime. Bit 6 of the GPS_CONTROL_WORD shall be set to 1 to indicate to receiver valid aiding states. The ATT_AIDING_DATA_VALID_i flag is also set on. During OPS1 prior to valid aiding states available from GPS Reference State Proc principal function, aiding states are those used last in

OPS9 prelaunch time tagged to the current Clocktime. The bit 6 of the GPS_CONTROL_WORD is set to 1 and ATT_AIDING_DATA_VALID_i flag shall be set to ON. During OPS9 post OPS305 or OPS603, the previous position and attitude aiding state shall continuously be used for aiding, while the velocity aiding state shall be a zero vector indicating that vehicle is stopped on the runway and time tagged with the current Clocktime.

IF (MM_CODE = 901 and OPS9_ROLLOUT_COMPLETE = OFF) THEN

T_REF_i = CLOCKTIME

BIT 6 of GPS_CONTROL_WORD = 1

ATT_AIDING_DATA_VALID_i = ON

IF GPS_INIT_REQUEST_SOP_i = ON THEN

INIT_STATE_SET_i = ON

GPS_INIT_REQUEST_SOP_i = OFF

ELSE

IF (MM_CODE = 101 and AIDING_DATA_VALID_i = OFF) THEN

T_REF_i = CLOCKTIME

BIT 6 of GPS_CONTROL_WORD = 1

ATT_AIDING_DATA_VALID_i = ON

IF GPS_INIT_REQUEST_SOP_i = ON THEN

INIT_STATE_SET_i = ON

GPS_INIT_REQUEST_SOP_i = OFF

ELSE

IF OPS9_ROLLOUT_COMPLETE = ON THEN

T_REF_i = CLOCKTIME

\bar{V} _REF_OUT_i = (0,0,0)

IF GPS_INIT_REQUEST_SOP_i = ON THEN

INIT_STATE_SET_i = ON

GPS_INIT_REQUEST_SOP_i = OFF

ELSE

do paragraph 4.1 and 4.2 below

- 4.1 During all other OPS phases under which the flag AIDING_DATA_VALID_i is ON indicating valid aiding states available, the state aiding data comes from the reference state which is computed in the GPS Reference State Proc principal function. The reference state shall be transformed to the WGS84 (World Geodetic Survey 1984) earth-fixed frame as shown in Section 4.17.6.4.

IF AIDING_DATA_VALID_i = ON THEN

T_REF_i = T_CURRENT_GPS_REF

\bar{R} _REF_OUT_i = \bar{R} _REF_ECEF

\bar{V} _REF_OUT_i = \bar{V} _REF_ECEF

Regardless of the value of AIDING_DATA_VALID_i, do paragraph 4.2 below.

- 4.2 The attitude aiding states are collected from ATT_PROC principal function, if the topodetic Euler angles availability flag, EULER_GPS_AVAIL, is ON.

```

IF EULER_GPS_AVAIL = ON THEN
    PHI_OUT = PHI
    PSI_OUT = PSI
    THETA_OUT = THETA
    ATT_AIDING_DATA_VALIDi = ON
ELSE
    ATT_AIDING_DATA_VALIDi = OFF

```

The attitude aiding parameters collected and computed above shall only be packed into GPC_TO_GPS_DATA_TWO_i buffer when the receiver is in NAV mode. This is done in NAV_AIDING_DATA code described in Section 4.17.6.4.3.

5. Compute the week and second components of the time to be associated with the aiding states.

```

TOTAL_SEC = T_REFi + LEAP_SEC_CORR + T_GPS_OFFSET_SECS +
            T_GPS_TO_GMT_ADJ + (T_GPS_TIME_ADJ_ACC/1000)
T_REF_WEEK_OUT = FLOOR(TOTAL_SEC / (7*SEC_PER_DAY))
T_REF_SEC_OUT = MOD(TOTAL_SEC, (7*SEC_PER_DAY))

```

6. Bit 16 of GPS_CONTROL_WORD is set according to ATT_AIDING_DATA_VALID_i flag.

```

IF ATT_AIDING_DATA_VALIDi = ON and GPS_MODE_CMD_SOPi = 'NAV'
    THEN
        BIT 16 of GPS_CONTROL_WORD = 1
    ELSE
        BIT 16 of GPS_CONTROL_WORD = 0

```

7. The 24th through 27th halfwords of the GPC_TO_GPS_DATA_ONE buffer contain the crew deselect satellite status.

```

GPC_TO_GPS_DATA_ONEi,24 = CREW_DESELECT_SAT_RCVD1
GPC_TO_GPS_DATA_ONEi,25 = CREW_DESELECT_SAT_RCVD2
GPC_TO_GPS_DATA_ONEi,26 = CREW_DESELECT_SAT_RCVD3
GPC_TO_GPS_DATA_ONEi,27 = CREW_DESELECT_SAT_RCVD4

```

8. Finally, the parameters collected and computed above are scaled and packed into GPC_TO_GPS_DATA_ONE_i buffer forming the Message 1 as defined in Table 4.17.1.1.2-2.

```

GPC_TO_GPS_DATA_ONEi,1 = HEX('AAAA')
GPC_TO_GPS_DATA_ONEi,(2 to 3) = GPS_CONTROL_WORD
GPC_TO_GPS_DATA_ONEi,4 = BINARY(T_REF_WEEK_OUT)
GPC_TO_GPS_DATA_ONEi,(5 to 7) = BINARY(INTEGER(T_REF_SEC_OUT /
                                           SF_TIME))
GPC_TO_GPS_DATA_ONEi,(8 to 9) = BINARY(INTEGER(R_REF_OUTi,1 / SF_POS))
GPC_TO_GPS_DATA_ONEi,(10 to 11) = BINARY(INTEGER(R_REF_OUTi,2 / SF_POS))
GPC_TO_GPS_DATA_ONEi,(12 to 13) = BINARY(INTEGER(R_REF_OUTi,3 / SF_POS))
GPC_TO_GPS_DATA_ONEi,(14 to 15) = BINARY(INTEGER(V_REF_OUTi,1 / SF_VEL))
GPC_TO_GPS_DATA_ONEi,(16 to 17) = BINARY(INTEGER(V_REF_OUTi,2 / SF_VEL))
GPC_TO_GPS_DATA_ONEi,(18 to 19) = BINARY(INTEGER(V_REF_OUTi,3 / SF_VEL))

```

D. Special Considerations:

1. OPS9_ROLLOUT_COMPLETE flag shall be initialized to OFF. The previous major mode indicator MM_CODE_PREV shall be a major-function-based parameter and initialized to any number other than 305 and 603.
2. \bar{R} _REF_OUT_i, \bar{V} _REF_OUT_i, PHI_OUT, PSI_OUT, THETA_OUT, ATT_AIDING_DATA_VALID, and AIDING_DATA_VALID are required to remain the same across the following OPS transitions, OPS1 to OPS6, OPS3 to OPS9, OPS6 to OPS9 and OPS9 to OPS1. The attitude data carried across the transition from OPS6 to OPS9 and OPS3 to OPS9 shall be used for attitude aiding post rollout in OPS9.
3. The GPC_TO_GPS_DATA_ONE_{i,(20 to 32)} shall be initialized to 0.
4. Bit 5 of GPS_CONTROL_WORD shall be initialized to zero at all OPS transitions except OPS1 to OPS6. This is required to insure that the GPS receiver differential navigation capability is not enabled. The Shuttle GPS implementation does not support this capability.

Bit 7 of GPS_CONTROL_WORD shall be initialized to 1; this enables the crew satellite de-selection capability.

Bit 9 of GPS_CONTROL_WORD shall be initialized to 0.
5. \bar{V} _REF_OUT_i shall be initialized to (0,0,0) upon entry into OPS 9.

Table 4.17.6.4.1-1. RCVR_AIDING_DATA Input Parameters

Description	Name	Input Source	Type	Precision	Range	Units	Rate (Hz)
Flag indicating valid aiding reference state	AIDING_DATA_VALID(I)**	*	D	–	ON/OFF	–	1.04
Current Clock Computer Time (GMT)	CLOCKTIME/CLOCK	*	F	DP		sec	1.04
BARO Data Good	GPS_BARO_DG_OUT	GPS_ALT_DATA	D	–	ON/OFF	–	1.04
RADALT Data Good	GPS_RADALT_DG_OUT	GPS_ALT_DATA	D	–	ON/OFF	–	6.25
GPS filter restart flag	GPS_FILT_RESTART(I)**	*, NAV_MODE_WORD_DATA	D	–	ON/OFF	–	1.04
GPS SOP commanded GPS mode	GPS_MODE_CMD_SOP(I)**	GPS_SOP	I	S	0(TEST) 1(INIT) 2(NAV)	–	1.04
Flag indicating GPS receiver I requesting Initialization State	GPS_INIT_REQUEST_SOP(I)**	NAV_MODE_WORD_DATA, GPS_SOP	D	–	ON/OFF	–	1.04
GPS reference state init flag	INIT_STATE_SET(I)**	*, NAV_MODE_WORD_DATA	D	–	ON/OFF	–	1.04
Major mode code	MM_CODE	*	I	S	–	–	1.04
GPS reference state announced reset flag	GPS_ANNC_RESET_SET(I)**	*, NAV_MODE_WORD_DATA	D	–	ON/OFF	–	1.04
Current GPS reference position state in WGS84 coordinates	R_REF_ECEF	GPC_TO_GPS	V	DP	–	ft	6.25
Time tag for current GPS reference state	T_CURRENT_GPS_REF	*	F	DP	–	sec	6.25
Current GPS reference velocity state in WGS84 coordinates	V_REF_ECEF	GPC_TO_GPS	V	DP	–	ft/sec	6.25
Topodetic Euler angles availability flag	EULER_GPS_AVAIL	*	D	–	ON/OFF	–	6.25
Body roll attitude euler angle	PHI	*	F	S	–	deg	6.25
Body yaw attitude euler angle	PSI	*	F	S	–	deg	6.25
Body pitch attitude euler angle	THETA	*	F	S	–	deg	6.25

Table 4.17.6.4.1-1. RCVR_AIDING_DATA Input Parameters							
Description	Name	Input Source	Type	Precision	Range	Units	Rate (Hz)
GPS receiver I current operating mode indicator	GPS_MODE_CURRENT(I)**	NAV_MODE_WORD_DATA	CHAR	–	TEST/INIT/NAV/‘ δ ’	–	1.04
Number of seconds from 1/6/80 to midnight between 12/30 and 12/31 of year prior to launch year	T_GPS_OFFSET_SECS	GPS_SOP	I	DP	–	sec	–
Crew deselect satellite word 1	CREW_DESELECT_SAT_RCVD(1)	*	HMD	–	–	–	6.25
Crew deselect satellite word 2	CREW_DESELECT_SAT_RCVD(2)	*	HMD	–	–	–	6.25
Crew deselect satellite word 3	CREW_DESELECT_SAT_RCVD(3)	*	HMD	–	–	–	6.25
Crew deselect satellite word 4	CREW_DESELECT_SAT_RCVD(4)	*	HMD	–	–	–	6.25
X-comp of GPS I ECEF position	R_REF_OUT(I,1)**	*	F	DP	–	ft	6.25
Y-comp of GPS I ECEF position	R_REF_OUT(I,2)**	*	F	DP	–	ft	6.25
Z-comp of GPS I ECEF position	R_REF_OUT(I,3)**	*	F	DP	–	ft	6.25
GPS to GMT time adjustment	T_GPS_TO_GMT_ADJ	GPS_SOP_TIME_ADJ_UPLINK	F	DP	–	sec	–
Accumulated GPC-to-GPS time adjustment	T_GPS_TIME_ADJ_ACC	GPS_TIME_ADJ	HXS	–	–	msec	6.25

* See principal function input list in Table 4.17.6.10-1
 ** I = 1,2,3, (I = 1 to GPS_SW_CAP)

Table 4.17.6.4.1-2. RCVR_AIDING_DATA Output Parameters

Description	Name	Destination	Type	Precision	Range	Units	Rate (Hz)
GPC to GPS receiver I data buffer one	GPC_TO_GPS_DATA_ONE(I)**	*	-	-	-	-	6.25
Body roll attitude euler angle	PHI_OUT	NAV_AIDING_DATA	F	S	-	Deg	6.25
Body yaw attitude euler angle	PSI_OUT	NAV_AIDING_DATA	F	S	-	Deg	6.25
Body pitch attitude euler angle	THETA_OUT	NAV_AIDING_DATA	F	S	-	Deg	6.25
Rollout complete flag	OPS9_ROLLOUT_COMPLETE	GPS_DISP_COMP	D	-	ON/OFF	-	6.25

* See principal function output list in Table 4.17.6.10-2

** I = 1,2,3 (1 to GPS_SW_CAP)

Table 4.17.6.4.1-3. Definition of GPS_CONTROL_WORD

Bit Parameter Description	Bit no.	Bit Value	Meaning	Related Parameter (Source)
Command GPS rcvr to be in TEST, INIT or NAV mode	0,1	0 0 0 1 1 0	Command TEST mode Command INIT mode Command NAV mode	GPS_MODE_CMD_SOP (GPS SOP)
Flag indicating position and velocity vectors available for receiver announced reset	2	1	Aiding states are valid for GPS receiver announced reset	GPS_ANNCED_RESET_SET (GPS REF STATE PROC)
Flag indicating BARO data good	3	1	Valid Baro aiding data	GPS_BARO_DG_OUT (GPS SOP)
Flag indicating radar data good	4	1	Valid Radar aiding data	GPS_RADALT_DG_OUT (GPS SOP)
Command GPS rcvr to operate in DIFFERENTIAL mode	5	1	Command GPS to do DIFFERENTIAL mode	Not applicable
Flag indicating valid aiding data	6	1	Valid state aiding data	AIDING_DATA_VALID (GPS REF STATE PROC, GPS SOP)
Satellite de-selection command	7	1	Crew de-select GPS satellites	Not applicable
Flag indicating position and velocity vectors ready for GPS receiver initialization	8	1	Aiding states are valid for GPS receiver initialization	INIT_STATE_SET (GPS REF STATE PROC)
Flag indicating position and velocity vectors available for receiver state reset due to IMU reconfiguration	9	1	Not used	
Flag indicating lever arm correction data available	10	1	GPC to GPS data buffer 2 contains valid lever arm and antenna boresight data	GPS_MODE_CMD_SOP (GPS SOP)
Spare	11 - 14		not used	
Flag indicating crew's request for GPS filter to restart	15	1	Crew command GPS receiver to restart (reinitialize) navigation filter	GPS_FILT_RESTART (GPS REF STATE PROC)
Flag indicating valid attitude data	16	1	GPC to GPS buffer 2 contains valid attitude aiding data	ATT_AIDING_DATA_VALID (GPS SOP)
Major mode	17 - 27		Major mode indicator	MM_CODE
Spare	28 - 29		not used	
Compliments of bit 0 and 1	30 - 31	1 1 1 0 0 1	Command TEST mode Command INIT mode Command NAV mode	GPS_MODE_CMD_SOP (GPS SOP)

4.17.6.4.2 Initialization Data Processing (INIT_DATA)

The purpose of this function is to format, scale and convert initialization data to pack the GPC_TO_GPS_DATA_TWO buffer. The receiver initialization data include all antenna location and boresight information, as well as GPS measurement residual edit criteria.

A. Functional Requirements. The INIT_DATA code is required to perform the following functions:

1. Collection of GPS antenna location and boresight data to be processed.
2. Data scaling and conversion into format suitable for GPS receivers.
3. Collect and convert GPS pseudo range and delta range residual edit criteria.

B. Interface Requirements. The input and output parameters associated with INIT_DATA module are contained in Table 4.17.6.4.2-1 and 4.17.6.4.2-2. I-load parameters are contained in Table 4.17.6.10-3.

C. Detailed Requirements and Formulations. The following operations shall be performed by the INIT_DATA module. Figure 4.17.6.4.2 presents the flow diagram of the INIT_DATA functional requirements.

1. The GPS antenna offset corrections and antenna boresight directions are scaled and packed into GPC_TO_GPS_DATA_TWO buffer forming the Message 2 as defined in Table 4.17.1.1.2-3.

```

GPC_TO_GPS_DATA_TWOi,1 = HEX('CCCC')
GPC_TO_GPS_DATA_TWOi,2 =
    BINARY(INTEGER (GPS_UPPER_ANTENNA_NBi,1 / SF_ANT))
GPC_TO_GPS_DATA_TWOi,3 =
    BINARY(INTEGER (GPS_UPPER_ANTENNA_NBi,2 / SF_ANT))
GPC_TO_GPS_DATA_TWOi,4 =
    BINARY(INTEGER (GPS_UPPER_ANTENNA_NBi,3 / SF_ANT))
GPC_TO_GPS_DATA_TWOi,5 =
    BINARY(INTEGER (GPS_LOWER_ANTENNA_NBi,1 / SF_ANT))
GPC_TO_GPS_DATA_TWOi,6 =
    BINARY(INTEGER (GPS_LOWER_ANTENNA_NBi,2 / SF_ANT))
GPC_TO_GPS_DATA_TWOi,7 =
    BINARY(INTEGER (GPS_LOWER_ANTENNA_NBi,3 / SF_ANT))
GPC_TO_GPS_DATA_TWOi,8 =
    BINARY(INTEGER (GPS_UPPER_ANTENNA_CGi,1 / SF_ANT))
GPC_TO_GPS_DATA_TWOi,9 =
    BINARY(INTEGER (GPS_UPPER_ANTENNA_CGi,2 / SF_ANT))
GPC_TO_GPS_DATA_TWOi,10 =
    BINARY(INTEGER (GPS_UPPER_ANTENNA_CGi,3 / SF_ANT))
GPC_TO_GPS_DATA_TWOi,11 =
    BINARY(INTEGER (GPS_LOWER_ANTENNA_CGi,1 / SF_ANT))
GPC_TO_GPS_DATA_TWOi,12 =
    BINARY(INTEGER (GPS_LOWER_ANTENNA_CGi,2 / SF_ANT))
GPC_TO_GPS_DATA_TWOi,13 =
    BINARY(INTEGER (GPS_LOWER_ANTENNA_CGi,3 / SF_ANT))
GPC_TO_GPS_DATA_TWOi,14 =

```

BINARY(INTEGER (GPS_UPPER_ANTENNA_BORESIGHT_{i,1} / SF_BORE))
 GPC_TO_GPS_DATA_TWO_{i,15} =
 BINARY(INTEGER (GPS_UPPER_ANTENNA_BORESIGHT_{i,2} / SF_BORE))
 GPC_TO_GPS_DATA_TWO_{i,16} =
 BINARY(INTEGER (GPS_UPPER_ANTENNA_BORESIGHT_{i,3} / SF_BORE))
 GPC_TO_GPS_DATA_TWO_{i,17} =
 BINARY(INTEGER (GPS_LOWER_ANTENNA_BORESIGHT_{i,1} / SF_BORE))
 GPC_TO_GPS_DATA_TWO_{i,18} =
 BINARY(INTEGER (GPS_LOWER_ANTENNA_BORESIGHT_{i,2} / SF_BORE))
 GPC_TO_GPS_DATA_TWO_{i,19} =
 BINARY(INTEGER (GPS_LOWER_ANTENNA_BORESIGHT_{i,3} / SF_BORE))

2. GPC_TO_GPS_DATA_TWO_{i,20} shall be packed with GPS pseudo range residual edit criteria according to the format defined in Table 4.17.6.4.2-3.

BIT 0 of GPC_TO_GPS_DATA_TWO_{i,20} = GPS_PR_EDIT_MAX_HOLD
 BIT 1 to 5 of GPC_TO_GPS_DATA_TWO_{i,20} =
 BINARY(GPS_PR_EDIT_MAX_STEP)
 BIT 6 to 10 of GPC_TO_GPS_DATA_TWO_{i,20} =
 BINARY(GPS_PR_EDIT_STEP_SIZE)
 BIT 11 to 15 of GPC_TO_GPS_DATA_TWO_{i,20} =
 BINARY(GPS_PR_EDIT_THRESHOLD)

3. GPC_TO_GPS_DATA_TWO_{i,21} shall be packed with GPS delta range residual edit criteria according to the format defined in Table 4.17.6.4.2-4.

BIT 0 of GPC_TO_GPS_DATA_TWO_{i,21} = GPS_DR_EDIT_MAX_HOLD
 BIT 1 to 5 of GPC_TO_GPS_DATA_TWO_{i,21} =
 BINARY(GPS_DR_EDIT_MAX_STEP)
 BIT 6 to 10 of GPC_TO_GPS_DATA_TWO_{i,21} =
 BINARY(GPS_DR_EDIT_STEP_SIZE)
 BIT 11 to 15 of GPC_TO_GPS_DATA_TWO_{i,21} =
 BINARY(GPS_DR_EDIT_THRESHOLD)

4. GPC_TO_GPS_DATA_TWO_{i,(22 to 32)} = 0

Table 4.17.6.4.2-1. INIT_DATA Input Parameters

Description	Name	Input Source	Type	Precision	Range	Units	Rate (Hz)
(NONE)							

Table 4.17.6.4.2-2. INIT_DATA Output Parameters							
Description	Name	Destination	Type	Precision	Range	Units	Rate (Hz)
GPC to GPS receiver I data buffer two	GPC_TO_GPS_DATA_TWO(I)**	*	-	-	-	-	6.25
* See principal function output list in Table 4.17.6.10-2 ** I = 1,2,3 (1 to GPS_SW_CAP)							

Table 4.17.6.4.2-3. Definition of GPC_TO_GPS_DATA_TWO _{i,20} - Init Data Only				
Parameter Description	Bit no.	LSB Location	Meaning	Related Parameter (Source)
PR Test Max Level Hold	0	-	Flag value of 1 indicates a maximum pseudo range edit threshold is in effect	GPS_PR_EDIT_MAX_HOLD
PR Test Max Step Number	1 - 5	5	Number of steps (cycles) to reach the maximum pseudo range edit threshold	GPS_PR_EDIT_MAX_STEP
PR Test Step Size	6 - 10	10	Size of the pseudo range edit threshold increment (sigma) per step (cycle)	GPS_PR_EDIT_STEP_SIZE
PR Test Threshold	11 - 15	15	Minimum pseudo range edit threshold	GPS_PR_EDIT_THRESHOLD

Table 4.17.6.4.2-4. Definition of GPC_TO_GPS_DATA_TWO _{i,21} - Init Data Only				
Parameter Description	Bit no.	LSB Location	Meaning	Related Parameter (Source)
DR Test Max Level Hold	0	-	Flag value of 1 indicates a maximum delta range edit threshold is in effect	GPS_DR_EDIT_MAX_HOLD
DR Test Max Step Number	1 - 5	5	Number of steps (cycles) to reach the maximum delta range edit threshold	GPS_DR_EDIT_MAX_STEP
DR Test Step Size	6 - 10	10	Size of the delta range edit threshold increment (sigma) per step (cycle)	GPS_DR_EDIT_STEP_SIZE
DR Test Threshold	11 - 15	15	Minimum delta range edit threshold	GPS_DR_EDIT_THRESHOLD

4.17.6.4.3 Nav Aiding Data Processing (NAV_AIDING_DATA)

The purpose of the nav aiding data processing function is to format, scale and convert data to pack the GPC_TO_GPS_DATA_TWO buffer which provides GPS receivers with vehicle attitude and altitude aiding information. The collection of the attitude and altitude parameters are done in RCVR_AIDING_DATA code which is described in Section 4.17.6.4.1.

A. Functional Requirements. The NAV_AIDING_DATA code is required to perform the following functions:

1. Data scaling and conversion of attitude and altitude information into format suitable for GPS receivers.

B. Interface Requirements. The input and output parameters associated with NAV_AIDING_DATA module are contained in Table 4.17.6.4.3-1 and 4.17.6.4.3-2. I-load parameters are contained in Table 4.17.6.10-3.

C. Detailed Requirements and Formulations. The following operations shall be performed by the NAV_AIDING_DATA module. Figure 4.17.6.4.3 presents the flow diagram of the NAV_AIDING_DATA functional requirements.

The attitude and altitude aiding data collected and computed in RCVR_AIDING_DATA module is scaled and packed into GPC_TO_GPS_DATA_TWO buffer forming the Message 3 as defined in Table 4.17.1.1.2-4.

```

GPC_TO_GPS_DATA_TWOi,1 = HEX('BBBB')
GPC_TO_GPS_DATA_TWOi,2 = BINARY(INTEGER (PHI_OUT/SF_ATT))
GPC_TO_GPS_DATA_TWOi,3 = BINARY(INTEGER (THETA_OUT/SF_ATT))
GPC_TO_GPS_DATA_TWOi,4 = BINARY(INTEGER (PSI_OUT/SF_ATT))
GPC_TO_GPS_DATA_TWOi,(5 to 7) = 0
GPC_TO_GPS_DATA_TWOi,8 = BINARY(T_GPS_BARO_WEEK_OUT)
GPC_TO_GPS_DATA_TWOi,(9 to 11) = BINARY(INTEGER
(T_GPS_BARO_SEC_OUT/SF_TIME))
GPC_TO_GPS_DATA_TWOi,12 = BINARY(T_GPS_RADALT_WEEK_OUT)
GPC_TO_GPS_DATA_TWOi,(13 to 15) = BINARY(INTEGER
(T_GPS_RADALT_SEC_OUT/SF_TIME))
GPC_TO_GPS_DATA_TWOi,(16 to 17) = BINARY(INTEGER
(GPS_BARO_DATA_OUT/SF_BARO))
GPC_TO_GPS_DATA_TWOi,18 = BINARY(INTEGER
(GPS_RADALT_DATA_OUT/SF_RA))
GPC_TO_GPS_DATA_TWOi,(19 to 32) = 0
    
```

Table 4.17.6.4.3-1. NAV_AIDING_DATA Input Parameters							
Description	Name	Input Source	Type	Precision	Range	Units	Rate (Hz)
Body roll attitude euler angle	PHI_OUT	RCVR_AIDING_DATA, *	F	S	–	deg	6.25
Body yaw attitude euler angle	PSI_OUT	RCVR_AIDING_DATA, *	F	S	–	deg	6.25
Body pitch attitude euler angle	THETA_OUT	RCVR_AIDING_DATA, *	F	S	–	deg	6.25
Baro Measurement	GPS_BARO_DATA_OUT	GPC_ALT_DATA	F	S	–	ft	6.25
Baro Time Tag (Weeks)	T_GPS_BARO_WEEK_OUT	GPC_ALT_DATA	I	S	–	week	6.25
Baro Time Tag (Seconds)	T_GPS_BARO_SEC_OUT	GPC_ALT_DATA	F	DP	–	sec	6.25
Radar Altitude measurement	GPS_RADALT_DATA_OUT	GPC_ALT_DATA	F	S	–	ft	6.25
Radar Altitude time tag (Weeks)	T_GPS_RADALT_WEEK_OUT	GPC_ALT_DATA	I	S	–	week	6.25
Radar Altitude time tag (Seconds)	T_GPS_RADALT_SEC_OUT	GPC_ALT_DATA	F	DP	–	sec	6.25

Table 4.17.6.4.3-2. NAV_AIDING_DATA Output Parameters							
Description	Name	Destination	Type	Precision	Range	Units	Rate (Hz)
GPC to GPS receiver I data buffer two	GPC_TO_GPS_DATA_TWO(I)**	*	–	–	–	–	6.25

* See principal function output list in Table 4.17.6.10-2
 ** I = 1,2,3 (1 to GPS_SW_CAP)

4.17.6.5 GPS Display Computations (GPS_DISP_COMP)

The purpose of the GPS display computations function is to collect and compute GPS related information that are used for display only.

A. Functional Requirements. The GPS_DISP_COMP module is required to perform the following function:

1. Determine the GPS status value to be displayed for each receiver.
2. Determine the GPS operating mode to be displayed for each receiver.

B. Interface Requirements. The input and output parameters associated with GPS_DISP_COMP module are contained in Tables 4.17.6.5-1 and 4.17.6.5-2.

C. Detailed Requirements and Formulations. The following operations shall be performed by the GPS_DISP_COMP module for each receiver. Figure 4.17.6.5 presents a flow diagram of the GPS_DISP_COMP functional requirements. Table 4.17.6.5-3 defines the format of GPS_RCVR_STATUS word.

1. The GPS_RCVR_STATUS_i value for displaying of the receiver status is set up as shown below for each GPS receivers.

The displayed values are listed below in order of precedence:

GPS receiver is not installed.
GPS receiver data is missing due to COMMFAULT.
GPS receiver built-in test is in progress (self test).
GPS receiver processing unit failure.
GPS receiver battery level low warning.
GPS receiver navigation data is not valid.
GPS receiver is operating properly

Before the following assignments, GPS_RCVR_STATUS_{i,(1 to 3)} shall be blanked indicating receiver operating properly.

```
IF GPS_INSTALLEDi = 'OFF' THEN
  GPS_RCVR_STATUSi = NI
ELSE IF GPS_COMMFAULTi = ON THEN
  GPS_RCVR_STATUSi = M
ELSE IF GPS_BIT_SOPi = ON THEN
  GPS_RCVR_STATUSi = BIT

  ELSE IF BIT 3 of GPS_STATUS_FOM_GDOPi = 1 THEN
    GPS_RCVR_STATUSi = RPF
    RCVR_FAILi = ON
```

```
ELSE
    RCVR_FAILi = OFF
    IF BIT 5 of GPS_STATUS_FOM_GDOPi = 1 THEN
        GPS_RCVR_STATUSi = BATT
    ELSE IF GPS_NAV_DATA_VALIDi = OFF and
        GPS_MODE_CURRENTi = 'NAV' THEN
        GPS_RCVR_STATUSi = DGF
```

2. The current GPS operating mode to be displayed on the GPS STATUS display is prepared next. If GPS_MODE_CURRENT_i is 'NAV', then GPS_MODE_DISP_i shall be the last selected site if the flag, STATIONARY_MODE_SOP_i, is ON, otherwise it shall be the same as the GPS_NAV_SUBMODE_i. If GPS_MODE_CURRENT_i is not 'NAV', the GPS_MODE_DISP_i shall be the same as GPS_MODE_CURRENT_i.

```
IF GPS_MODE_CURRENTi = 'NAV' THEN
    IF STATIONARY_MODE_SOPi = ON THEN
        IF (MM_CODE = 101) or (MM_CODE = 901 AND
            OPS9_ROLLOUT_COMPLETE = OFF) THEN
            GPS_MODE_DISPi = SEL_TEST_ID(1 to 3)
        ELSE
            GPS_MODE_DISPi = SEL_SITE_ID(1 to 3)
```

```
ELSE
    GPS_MODE_DISPi = GPS_NAV_SUBMODEi
```

```
ELSE
    GPS_MODE_DISPi = GPS_MODE_CURRENTi
```

3. If GPS_BIT_DISP_ENABLE = OFF, then set GPS_FAIL_ST_i to 'Ø'; otherwise, test GPS_FAIL_ST_FLAG_i. If GPS_FAIL_ST_FLAG_i is ON then set GPS_FAIL_ST_i to '↓'; else, set GPS_FAIL_ST_i to '↑'.

```
IF GPS_BIT_DISP_ENABLEi = OFF THEN
    GPS_FAIL_STi = 'Ø'
ELSE IF GPS_FAIL_ST_FLAGi = ON THEN
    GPS_FAIL_STi = '↓'
ELSE
    GPS_FAIL_STi = '↑'
```

D. Special Considerations.

The RCVR_FAIL_i (i = 1 to 3) flags shall be initialized to OFF and the GPS_FAIL_ST_i (I = 1 to 3) indicators shall be initialized to 'Ø'. In addition, RCVR_FAIL_i shall retain their current values across all OPS transitions.

Table 4.17.6.5-1. GPS_DISP_COMP Input Parameters

Description	Name	Input Source	Type	Precision	Range	Units	Rate (Hz)
GPS commfault flag	GPS_COMMFAULT(I)**	*	D	–	ON/OFF	–	1.04
GPS BIT DISPLAY ENABLE Flag	GPS_BIT_DISP_ENABLE(I)**	GPS_TO_GPC	D	–	ON/OFF	–	6.25
GPS receiver self test in progress flag	GPS_BIT_SOP(I)**	GPS_TO_GPC	D	–	ON/OFF	–	1.04
Flag indicating navigation data valid	GPS_NAV_DATA_VALID(I)**	NAV_MODE_WORD_DATA	D	–	ON/OFF	–	1.04
GPS receiver status, figure of merit and GDOP word	GPS_STATUS_FOM_GDOP(I)**	NAV_MESSAGE_DATA	HMD	16 bits	–	–	1.04
GPS receiver I current operating mode indicator	GPS_MODE_CURRENT(I)**	NAV_MODE_WORD_DATA	CHAR	–	TEST/INIT/NAV/‘Ø’	–	1.04
GPS receiver I navigation submode indicator	GPS_NAV_SUBMODE(I)**	NAV_MODE_WORD_DATA	CHAR	–	PVA/INS/‘Ø’	–	1.04
Flag indicating GPS receiver I is in stationary mode	STATIONARY_MODE_SOP(I)**	NAV_MODE_WORD_DATA	D	–	ON/OFF	–	1.04
GPS receiver I fail self-test flag	GPS_FAIL_ST_FLAG(I)**	TEST_MESSAGE_DATA	D	–	ON/OFF	–	6.25
Major Mode Code	MM_CODE	*	I	S	–	–	1.04
Runway name for the selected landing site	SEL_SITE_ID	*	CHAR	–	–	–	1.04
Selected IMU calibration site name	SEL_TEST_ID	*	CHAR	–	–	–	1.04
Rollout complete flag	OPS9_ROLLOUT_COMPLETE	RCVR_AIDING_DATA	D	–	–	–	1.04

* See principal function input list in Table 4.17.6.10-1

** I = 1,2,3 (1 to GPS_SW_CAP)

Table 4.17.6.5-2. GPS_DISP_COMP Output Parameters

Description	Name	Output Destination	Type	Precision	Range	Units	Rate (Hz)
Status of GPS receiver I	GPS_RCVR_STATUS(I)**	*	I	S	0(NI) 1(M) 2(BIT) 3(RPF) 4(BATT) 5(DGF) 6('⌘')	–	6.25
Current GPS receiver I operating mode for display	GPS_MODE_DISP(I)**	*	CHAR	–	INIT, TEST, PVA, INS, '⌘', (site)	–	6.25
GPS I fail self-test indicator	GPS_FAIL_ST(I)**	*	VAR	–	'⌘', ↑, ↓	–	6.25
Flag indicating GPS receiver processing unit fail	RCVR_FAIL(I)**	*	D	–	ON/OFF	–	6.25

* See principal function output list in Table 4.17.6.10-2

** I = 1,2,3 (1 to GPS_SW_CAP)

Table 4.17.6.5-3. Definition of GPS_RCVR_STATUS_i word

Display	Value	Meaning	Related Parameter
NI	0	GPS receiver is not installed	GPS_INSTALLED _i is OFF
M	1	GPS receiver data is missing due to COMMFAULT	GPS_COMMFAULT _i is ON
BIT	2	GPS receiver Built In Test is in progress (commanded Self-Test)	GPS_BIT_SOP _i is ON
RPF	3	GPS receiver processing unit failure	BIT 3 of GPS_STATUS_FOM_GDOP _i is ON
BATT	4	GPS receiver battery is low	BIT 5 of GPS_STATUS_FOM_GDOP _i is ON
DGF	5	GPS receiver navigation data invalid	GPS_NAV_DATA_VALID _i
'⌘'	6	GPS receiver is operating properly	Blank field will be displayed if none of the above conditions is true

Note: The displayed values are listed in order of precedence.

4.17.6.6 GPS SOP Downlist Computations (GPS_SOP_DL_COMP)

The purpose of the GPS SOP downlist computation is to prepare parameters for downlist.

A. Functional Requirements. The GPS_SOP_DL_COMP module is required to perform the following function:

1. Prepare GPS data for downlist.

B. Interface Requirements. The input and output parameters associated with GPS_SOP_DL_COMP module are contained in Tables 4.17.6.6-1 and 4.17.6.6-2.

C. Detailed Requirements and Formulations. The following operations shall be performed by the GPS_SOP_DL_COMP module. Figure 4.17.6.6 presents the flow diagram of the GPS_SOP_DL_COMP functional requirements. Table 4.17.6.6-3 summarizes the bit assignment of GPS_SOP_COLLECTION_WORD.

1. The downlist parent words, GPS_SOP_COLLECTION_WORD and GPS_SOP_COLLECTION_WORD_TWO, are blanked first.

 BITS 0 to 15 of GPS_SOP_COLLECTION_WORD = 0
 BITS 0 to 15 of GPS_SOP_COLLECTION_WORD_TWO = 0
2. Bits 1, 2 and 3 of GPS_SOP_COLLECTION_WORD are set to GPS_COMMFAULT₁, GPS_COMMFAULT₂ and GPS_COMMFAULT₃.

 BIT 1 of GPS_SOP_COLLECTION_WORD = GPS_COMMFAULT₁
 BIT 2 of GPS_SOP_COLLECTION_WORD = GPS_COMMFAULT₂
 BIT 3 of GPS_SOP_COLLECTION_WORD = GPS_COMMFAULT₃
3. BIT 4 of GPS_SOP_COLLECTION_WORD is set to GPS_RA_RCVD.

 BIT 4 of GPS_SOP_COLLECTION_WORD = GPS_RA_RCVD
4. Bits 5 to 7 of GPS_SOP_COLLECTION_WORD are set to GPS_FAIL_ST_FLAG₁, GPS_FAIL_ST_FLAG₂, and GPS_FAIL_ST_FLAG₃.

 BIT 5 of GPS_SOP_COLLECTION_WORD = GPS_FAIL_ST_FLAG₁
 BIT 6 of GPS_SOP_COLLECTION_WORD = GPS_FAIL_ST_FLAG₂
 BIT 7 of GPS_SOP_COLLECTION_WORD = GPS_FAIL_ST_FLAG₃
5. Bit 8 of GPS_SOP_COLLECTION_WORD is set to GPS_TIME_ADJ_ENBL

 Bit 8 of GPS_SOP_COLLECTION_WORD = GPS_TIME_ADJ_ENBL
6. Flags indicating GPS receivers Built-in-Test in progress are packed into GPS_SOP_COLLECTION_WORD as follows:

 BIT 9 of GPS_SOP_COLLECTION_WORD = GPS_BIT_SOP₁
 BIT 10 of GPS_SOP_COLLECTION_WORD = GPS_BIT_SOP₂
 BIT 11 of GPS_SOP_COLLECTION_WORD = GPS_BIT_SOP₃

7. GPS receiver Time-Figure-of-Merits are packed as follows:

BITS 12 to 15 of GPS_SOP_COLLECTION_WORD = BITS 12 to 15 of GPS_TFOM₂

BITS 1 to 4 of GPS_SOP_COLLECTION_WORD_TWO = BITS 12 to 15 of GPS_TFOM₁

BITS 5 to 8 of GPS_SOP_COLLECTION_WORD_TWO = BITS 12 to 15 of GPS_TFOM₃

Table 4.17.6.6-1. GPS_SOP_DL_COMP Input Parameters							
Description	Name	Input Source	Type	Precision	Range	Units	Rate (Hz)
GPS commfault flag	GPS_COMMFault(I)**	*	D	1 Bit	ON/OFF		1.04
GPS receiver I Time Figure of Merit	GPS_TFOM(I)	NAV_MESSAGE_DATA	I	S	-	-	1.04
GPS receiver I self-test in progress flag	GPS_BIT_SOP(I)**	GPS_TO_GPC	D	-	ON/OFF	-	1.04
Crew initiated radar altimeter aiding flag	GPS_RA_RCVD	*	D	-	ON/OFF	-	6.25
GPS receiver I fail self-test flag	GPS_FAIL_ST_FLAG(I)**	TEST_MESSAGE_DATA	D	-	ON/OFF	-	6.25
GPS time adjustment enable ind	GPS_TIME_ADJ_ENBL	GPS_SOP	D	-	ON/OFF	-	6.25
* See principal function input list in Table 4.17.6.10-1							
** I = 1,2,3 (1 to GPS_SW_CAP)							

Table 4.17.6.6-2. GPS_SOP_DL_COMP Output Parameters							
Description	Name	Output Destination	Type	Precision	Range	Units	Rate (Hz)
GPS SOP collection word for downlist	GPS_SOP_COLLECTION_WORD	*	HMD	-	-	-	6.25
GPS SOP collection word two for downlist	GPS_SOP_COLLECTION_WORD_TWO	*	HMD	-	-	-	6.25
* See principal function output list in Table 4.17.6.10-2							

Table 4.17.6.6-3. Definition of GPS_SOP_COLLECTION_WORD					
Parameter Description	MSID	Bit No.	Bit Value	Meaning	Related Parameter
Spare		0			
GPS 1 commfault flag	V93X1237X	1	1	GPS receiver 1 commfaulted	GPS_COMMFAULT ₁
GPS 2 commfault flag	V93X1152X	2	1	GPS receiver 2 commfaulted	GPS_COMMFAULT ₂
GPS 3 commfault flag	V93X1238X	3	1	GPS receiver 3 commfaulted	GPS_COMMFAULT ₃
Crew initiated radar altimeter aiding flag	V93X1129X	4	1	Crew has initiated radar altimeter aiding	GPS_RA_RCVD
GPS 1 fail self-test	V93X3243X	5	1	GPS receiver 1 fail self-test	GPS_FAIL_ST_FLAG ₁
GPS 2 fail self-test	V93X3244X	6	1	GPS receiver 2 fail self-test	GPS_FAIL_ST_FLAG ₂
GPS 3 fail self-test	V93X3245X	7	1	GPS receiver 3 fail self-test	GPS_FAIL_ST_FLAG ₃
GPS time adjustment enable indicator	V94X0220X	8	1	GPC-to-GPS Time Adjustment capability is enabled.	GPS_TIME_ADJ_ENBL
GPS 1 Built-In Test in progress flag	V93X1002X	9	1	GPS receiver 1 BIT in progress	GPS_BIT_SOP ₁
GPS 2 Built-In Test in progress flag	V93X1003X	10	1	GPS receiver 2 BIT in progress	GPS_BIT_SOP ₂
GPS 3 Built-In Test in progress flag	V93X1004X	11	1	GPS receiver 3 BIT in progress	GPS_BIT_SOP ₃
GPS 2 Time Figure of Merit	V93U1001C	12 - 15	–	Binary number representing the Time Figure of Merit of GPS receiver 2	GPS_TFOM ₂

Table 4.17.6.6-4. Definition of GPS_SOP_COLLECTION_WORD_TWO					
Parameter Description	MSID	Bit No.	Bit Value	Meaning	Related Parameter
Spare		0			
GPS 1 Time Figure of Merit	V93U3001C	1-4	–	Binary number representing the Time Figure of Merit of GPS receiver 1	GPS_TFOM ₁
GPS 3 Time Figure of Merit	V93U9001C	5-8	–	Binary number representing the Time Figure of Merit of GPS receiver 3	GPS_TFOM ₃
Spare		14			
Spare		15			

4.17.6.7 GPS IMU Data Processing (GPS_IMU_SAVE)

The purpose of the GPS IMU data processing is (1) to reserve a set of selected accumulated IMU velocity counts from IMU RM and its associated time tag to be assigned to the GPS state vector, and (2) to choose the latest available GPS state vector which is bracketed in time by the available IMU data. The IMU data selection process requires that the chosen IMU data is the closest in time, but prior, to the corresponding GPS state. These saved IMU velocity counts shall be used by the GPS Receiver State Proc principal function to propagate the GPS state vector.

A. Functional Requirements. The GPS_IMU_SAVE module is required to perform the following function:

1. Store the most recent IMU data in buffer which accommodates 12 sets of selected IMU counts.
2. Determine the most appropriate IMU counts from the buffer to assign to the selected GPS states.

B. Interface Requirements. The input and output parameters associated with GPS_IMU_SAVE module are contained in Tables 4.17.6.7-1 and 4.17.6.7-2.

C. Detailed Requirements and Formulations. The following operations shall be performed by the GPS_IMU_SAVE module. Figure 4.17.6.7 presents a flow diagram of the GPS_IMU_SAVE functional requirements.

1. The time tag and the selected accumulated velocity counts shall be stored into the buffer in a cyclic manner. The buffer is designed to accommodate 12 sets of IMU data. With GPS SOP running at 6.25 Hz, the buffer size of 12 is equivalent to 1.92 seconds of IMU data. With aiding, the receiver outputs state vector at the state aiding time. The output state read by GPC can be up to 640 ms old. The 1.92 seconds of buffer size is well padded to cover the inherent latency and other system transport lags.

If the buffer index is equal to or greater than 12, it shall be set to 1 and the flag to indicate that the buffer is full shall be set to ON, otherwise, the index shall be increased by 1. The IMU data shall be stored in the appropriate buffers.

```
IF GPS_IMU_INDEX >= 12 THEN
    GPS_IMU_INDEX = 1
    GPS_IMU_BUFFER_FULL = ON
ELSE
    GPS_IMU_INDEX = GPS_IMU_INDEX + 1
```

```
 $\bar{V}_{IMU\_BUFFER_{GPS\_IMU\_INDEX}} = \bar{V}_{IMU\_GPS\_SOP}$ 
 $T_{IMU\_BUFFER_{GPS\_IMU\_INDEX}} = T_{IMU\_GPS\_SOP}$ 
```

2. Before the IMU buffers are fully stored, the GPS data good flags (GPS_DG_SOP_i) are overridden to the OFF state indicating that no IMU data is yet available for the GPS Receiver State Proc to propagate the GPS states properly.

```
IF GPS_IMU_BUFFER_FULL = OFF THEN GPS_DG_SOP = (OFF,OFF,OFF)
ELSE do step 3 below
```

3. The following shall be performed for each of the GPS receivers.

3.1 If the new GPS navigation data is not already IMU bracketed as indicated by the flag, IMU_BRACKET_NEW_i being OFF, then the following paragraphs 3.1.1 and 3.1.2 are performed.

3.1.1 The IMU time tag buffer shall be examined in order to find the most current IMU data that is NOT more recent than the GPS state. Firstly, the flag indicating search complete, FOUND_LOW, shall be set OFF. Next, the 12 elements of the T_IMU_BUFFER is checked against the new GPS time, T_GPS_NEW_i. This comparison shall be initiated at the most current IMU time which resides in the buffer pointed by the index, GPS_IMU_INDEX. If the GPS time is less than the IMU time, the search index will be lowered by one. However, if the index is zero or negative, it shall be added to the size of the buffer, 12, in order to continue the search. If the IMU time is found to be less than or equal to the GPS time, the search process terminates. The IMU data will be saved and assigned to the associated input GPS state.

```
FOUND_LOW = OFF
DO FOR k = 0 to 11 WHILE (FOUND_LOW = OFF)
    INDEX = GPS_IMU_INDEX - k
    IF INDEX <= 0 THEN
        INDEX = INDEX + 12
    IF T_GPS_NEWi >= T_IMU_BUFFERINDEX THEN
        FOUND_LOW = ON
```

```
 $\bar{V}_{IMU\_NEW}_i = \bar{IMU\_BUFFER}_{INDEX}$ 
 $T_{IMU\_NEW}_i = T_{IMU\_BUFFER}_{INDEX}$ 
```

3.1.2 If the loop index, k, in step 3.1.1 above is found to be greater than zero, indicating that the new GPS data is bracketed by IMU data in time, then the flag IMU_BRACKET_NEW_i is set ON and the associated input GPS state and IMU data are copied into the parameters for output to the GPS Receiver State Proc principal function.

```
IF k > 0 THEN
    IMU_BRACKET_NEWi = ON
    T_GPS_SOPi = T_GPS_NEWi
     $\bar{R}_{GPS\_ECEF\_SOP}_i = \bar{R}_{GPS\_ECEF\_NEW}_i$ 
     $\bar{V}_{GPS\_ECEF\_SOP}_i = \bar{V}_{GPS\_ECEF\_NEW}_i$ 
    GPS_DG_SOPi = ON
    GPS_FAIL_FOM_SOPi = GPS_FAIL_FOM_NEWi
    FOM_SOPi = FOM_NEWi
     $\bar{V}_{IMU\_SAVE\_SOP} = \bar{V}_{IMU\_NEW}_i$ 
    T_IMU_SAVE_SOPi = T_IMU_NEWi
```

D. Special Consideration.

1. The GPS_IMU_INDEX shall be initialized to zero at each OPS transition except OPS1 to OPS6. The GPS_IMU_BUFFER_FULL shall be initialized to OFF at each OPS transition except OPS1 to OPS6. When IMU SOP executes at 1.04 Hz, the IMU_BUFFER will store multiple copies of same IMU data, since the module is executed at 6.25 Hz.

2. The flag IMU_BRACKET_NEW_i shall be initialized to ON.
3. In the unlikely event that the GPS state is older in time than the oldest IMU data in the IMU BUFFER, the oldest IMU velocity counts are chosen to be associated with the GPS state which is then considered IMU-bracketed.

Table 4.17.6.7-1. GPS_IMU_SAVE Input Parameters

Description	Name	Input Source	Type	Precision	Range	Units	Rate (Hz)
Figure of merit associated with “NEW” GPS I data	FOM_NEW(I)**	NAV_MESSAGE_DATA	I	S	1 - 9	–	1.04
Flag indicating GPS I failed FOM threshold	GPS_FAIL_FOM_NEW(I)**	GPS_QA1_CHECK	D	–	ON/OFF	–	
Position vector of “NEW” GPS I data	R_GPS_ECEF_NEW(I)**	NAV_MESSAGE_DATA	V	DP	–	ft	1.04
Time tag of “NEW” GPS receiver I data	T_GPS_NEW(I)**	NAV_MESSAGE_DATA	S	DP	–	sec	1.04
Time tag of current accumulated IMU velocity counts	T_IMU_GPS_SOP	*	F	DP	–	sec	1.04/ 6.25
Velocity vector of “NEW” GPS I data	V_GPS_ECEF_NEW(I)**	NAV_MESSAGE_DATA	V	S	–	ft/sec	1.04
Current selected accumulated IMU velocity counts	\bar{V} _IMU_GPS_SOP	*	V	DP		ft/sec	1.04/ 6.25
Flag indicating GPS data has been time bracketed by IMU data	IMU_BRACKET_NEW(I)**	NAV_MESSAGE_DATA	D	–	ON/OFF	–	6.25

* See principal function input list in Table 4.17.6.10-1
** I = 1,2,3 (1 to GPS_SW_CAP)

Table 4.17.6.7-2. GPS_IMU_SAVE Output Parameters

Description	Name	Output Destination	Type	Precision	Range	Units	Rate (Hz)
Figure of merit associated with GPS data sent to GPS Receiver State Proc	FOM_SOP(I)**	*	I	S	1 - 9	–	6.25
Data good flag of GPS data sent to GPS Receiver State Proc	GPS_DG_SOP(I)**	*	D	–	ON/OFF		6.25
Failing FOM flag for GPS data sent to GPS Receiver State Proc	GPS_FAIL_FOM_SOP(I)**	*	D	–	ON/OFF		6.25
Flag indicating that the IMU buffer is full	GPS_IMU_BUFFER_FULL	NAV_MESSAGE_DATA	D	–	ON/OFF	–	6.25
Flag indicating GPS data has been time-bracketed by IMU data	IMU_BRACKET_NEW(I)**	NAV_MESSAGE_DATA	D	–	ON/OFF	–	6.25
Position vector of GPS data sent to GPS Receiver State Proc	$\bar{R}_{GPS_ECEF_SOP(I)**}$	*	V	DP	–	ft	6.25
Time tag of GPS data sent to GPS Receiver State Proc	T_GPS_SOP(I)**	*	S	DP	–	sec	6.25
Time tag of IMU counts associated with GPS data sent to GPS Receiver State Proc	T_IMU_SAVE_SOP(I)**	*	S	DP	–	sec	6.25
Velocity vector of GPS data sent to GPS Receiver State Proc	$\bar{V}_{GPS_ECEF_SOP(I)**}$	*	V	S	–	ft/sec	6.25
Accumulated IMU velocity count associated with the GPS receiver I states	$\bar{V}_{IMU_SAVE_SOP(I)**}$	*	V	DP		ft/sec	6.25

* See principal function output list in Table 4.17.6.10-2
** I = 1,2,3 (1 to GPS_SW_CAP)

4.17.6.8 GPS SOP Time Adjustment Uplink Processing (GPS_SOP_TIME_ADJ_UPLINK)

The flight software shall provide the capability to adjust the GPS SOP time calculations in response to a command uplink from the ground in OPS 9, OPS 1/6, MM201, and OPS 3. The GPS SOP shall use the uplinked GPS-to-GMT time adjustment factor (T_GPS_TO_GMT_ADJ) contained in this uplink during all GNC OPS phases where the GPS SOP is running.

A. Functional Requirements. The flight software is required to process the GPS SOP Time Adjustment uplink, update the GPS-to-GMT time adjustment parameter and make it available for use in the GPS SOP.

B. Interface Requirements. The input and output parameters associated with the GPS SOP Time Adjustment uplink processing are listed in Tables 4.17.6.8-1 and 4.17.6.8-2.

C. Detailed Requirements. The following operations shall be performed by the flight software when processing the GPS SOP Time Adjustment uplink:

1. The Op Code associated with the uplink shall be tested to see if this is a GPS SOP Time Adjustment uplink, indicated by the Op Code being set to 0111111 binary (63 decimal). If the OP Code is equal to 0111111 binary, then proceed with step 2. Otherwise, this is not a GPS SOP Time Adjustment uplink, and further processing is not required.

IF OP_CODE = 0111111, then do step 2.

2. Set the GPS-to-GMT time adjustment factor (T_GPS_TO_GMT_ADJ) equal to the uplinked value (GPS_TO_GMT_ADJ). Then, set the Op Code to 0000000 to prevent repeated processing of the uplink.

T_GPS_TO_GMT_ADJ = GPS_TO_GMT_ADJ
OP_CODE = 0000000

D. Special Considerations. Processing of the GPS SOP Time Adjustment uplink shall occur any time the uplink is received, regardless of whether the GPS SOP is running or not. The GPS-to-GMT time adjustment parameter (T_GPS_TO_GMT_ADJ) shall initially be set to zero, and shall be carried across all GN&C OPS transitions and used by the GPS SOP in all OPS where the GPS SOP is running.

E. Supplementary Information. Figure 4.17.6.8 is a flowchart illustrating the functional requirements for the GPS SOP Time Adjustment uplink processing.

Table 4.17.6.8-1. GPS SOP Time Adjustment Uplink Processing Input Parameters							
Description	Name	Input Source	Type	Precision	Range	Units	Rate (Hz)
Uplink operations code	OP_CODE	Uplink Processing	HXS	-	-	-	6.25
Buffered GPS to GMT time adjustment	GPS_TO_GMT_ADJ	Uplink Processing	F	DP	-	sec	On Demand

Table 4.17.6.8-2. GPS SOP Time Adjustment Uplink Processing Output Parameters							
Description	Name	Destination	Type	Precision	Range	Units	Rate (Hz)
Uplink operations code	OP_CODE	Uplink Processing	HXS	-	-	-	On Demand
GPS to GMT time adjustment	T_GPS_TO_GMT_ADJ	NAV_MESSAGE_DATA, RCVR_AIDING_DATA, GPS_ALT_DATA	F	DP	-	sec	On Demand

4.17.6.9 GPS SOP Time Adjustment (GPS_TIME_ADJ)

The flight software shall provide the capability to adjust the GPS SOP time when this capability is enabled.

A. Functional Requirements. This module computes the accumulated GPS time adjustment required to center the GPS time tag minus the GPC time tag at a predetermined I-Load value.

B. Interface Requirements. The input and output parameters associated with the GPS SOP Time Adjustment are listed in Table 4.17.6.9-1 and Table 4.17.6.9-2.

C. Detailed Requirements. The following operation shall be performed by the flight software when processing GPS SOP Time Adjustment.

1. GPS age-of-aiding data shall be selected based on GPS receiver communication fault statuses and the GPS receiver modes and data validities. An index (INDEX) with possible values from 0 to 7 is used as a pointer to the appropriate GPS_AOA selection algorithms. If INDEX equals 0, the software shall perform no age-of-aiding data selection. The INDEX is computed as follows:

INDEX =

1 ((NOT GPS_COMMFAULT₁) AND (GPS_MODE_CURRENT₁ = "NAV") AND
(NOT CREW_DES_RCVR_SOP₁) AND (GPS_NAV_DATA_VALID₁ = ON))
+ 2 ((NOT GPS_COMMFAULT₂) AND (GPS_MODE_CURRENT₂ = "NAV") AND
(NOT CREW_DES_RCVR_SOP₂) AND (GPS_NAV_DATA_VALID₂ = ON))
+ 4 ((NOT GPS_COMMFAULT₃) AND (GPS_MODE_CURRENT₃ = "NAV") AND
(NOT CREW_DES_RCVR_SOP₃) AND (GPS_NAV_DATA_VALID₃ = ON))

- 1.1 If INDEX equals 7, then the selected age-of-aiding (GPS_AOA_SEL) is the mid value of the three GPS receivers age-of-aiding.

IF (INDEX = 7) THEN
 GPS_AOA_SEL = MIDVAL(GPS_AOA₁, GPS_AOA₂, GPS_AOA₃)
END IF

- 1.2 If INDEX equals 6, then the selected age-of-aiding is the average of age-of-aiding from GPS receiver 2 and 3.

IF (INDEX = 6) THEN
 GPS_AOA_SEL = 0.5*(GPS_AOA₂ + GPS_AOA₃)
END IF

- 1.3 If INDEX equals 5, then the selected age-of-aiding is the average of age-of-aiding from GPS receiver 1 and 3.

IF (INDEX = 5) THEN
 GPS_AOA_SEL = 0.5*(GPS_AOA₁ + GPS_AOA₃)
END IF

- 1.4 If INDEX equals 4, then the selected age-of-aiding is the age-of-aiding from GPS receiver 3.

```
IF (INDEX = 4) THEN
    GPS_AOA_SEL = GPS_AOA3
END IF
```

- 1.5 If INDEX equals 3, then the selected age-of-aiding is the average of age-of-aiding from GPS receiver 1 and 2.

```
IF (INDEX = 3) THEN
    GPS_AOA_SEL = 0.5*(GPS_AOA1 + GPS_AOA2)
END IF
```

- 1.6 If INDEX equals 2, then the selected age-of-aiding is the age-of-aiding from GPS receiver 2.

```
IF (INDEX = 2) THEN
    GPS_AOA_SEL = GPS_AOA2
END IF
```

- 1.7 If INDEX equals 1, then the selected age-of-aiding is the age-of-aiding from GPS receiver 1.

```
IF (INDEX = 1) THEN
    GPS_AOA_SEL = GPS_AOA1
END IF
```

2. The remnant of the GPC-to-GPS time adjustment (T_GPS_GPC_REM) is equal to the selected age-of-aiding data minus the predetermined I-Load value (GPC_GPS_TARGET).

$$T_GPS_GPC_REM = GPS_AOA_SEL - GPC_GPS_TARGET$$

3. If the flag indicating GPS announced reset for the GPC-to-GPS time adjustment (GPS_ANNCD_RESET_TIME_ADJ) is ON, then the accumulated GPS time adjustment (T_GPS_TIME_ADJ_ACC) shall be incremented with the remnant of the GPC-to-GPS time adjustment. Furthermore, the flag indicating GPS announced reset for the GPC-to-GPS time adjustment shall be set to OFF.

```
IF (GPS_ANNCD_RESET_TIME_ADJ = ON) THEN

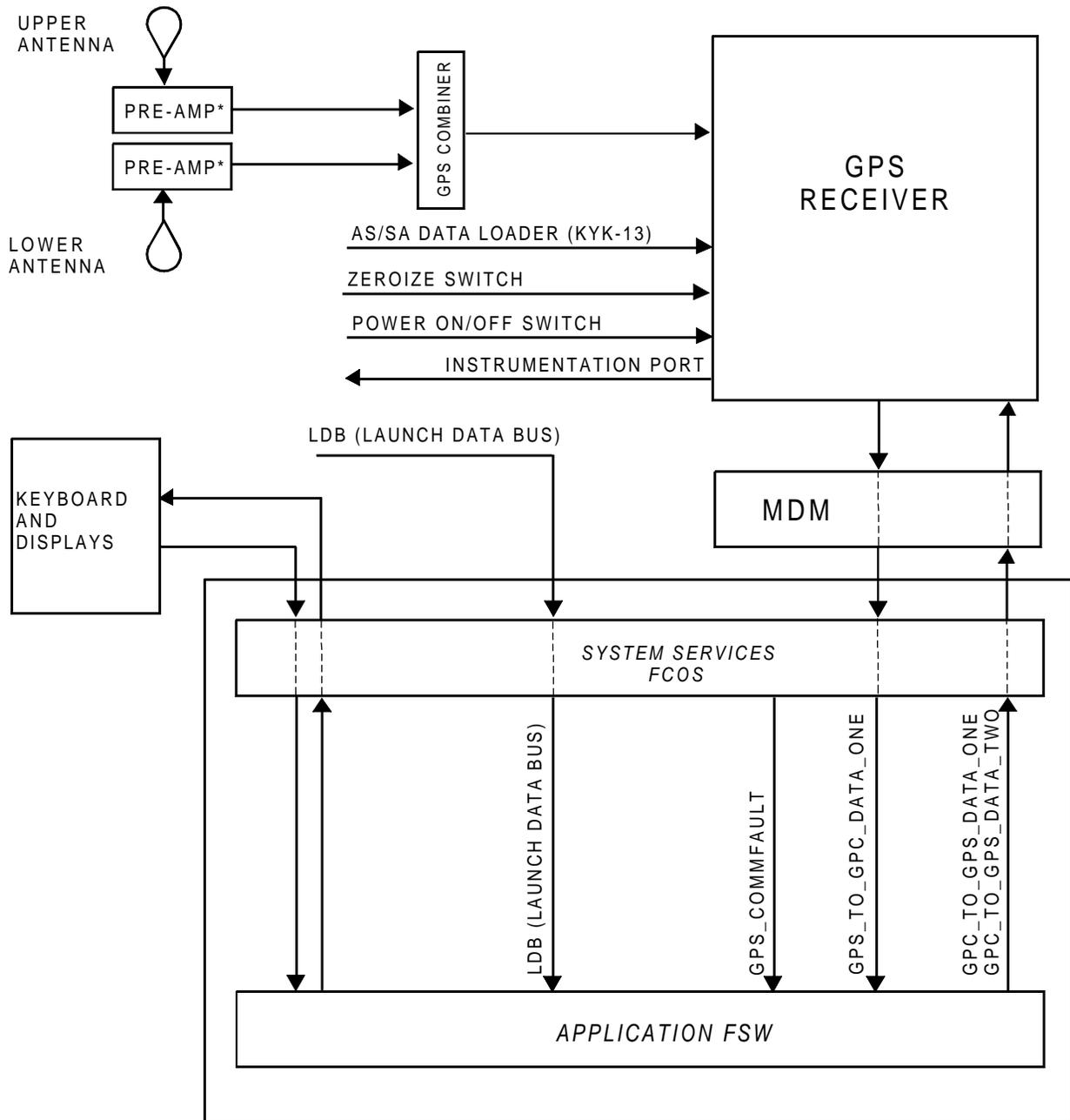
    T_GPS_TIME_ADJ_ACC = T_GPS_TIME_ADJ_ACC + T_GPS_GPC_REM
    GPS_ANNCD_RESET_TIME_ADJ = OFF
```

```
END IF
```

D. Special Considerations. The accumulated GPS time adjustment, T_GPS_TIME_ADJ_ACC, shall be a major-function-based parameter and initialized to 0 in OPS 1.

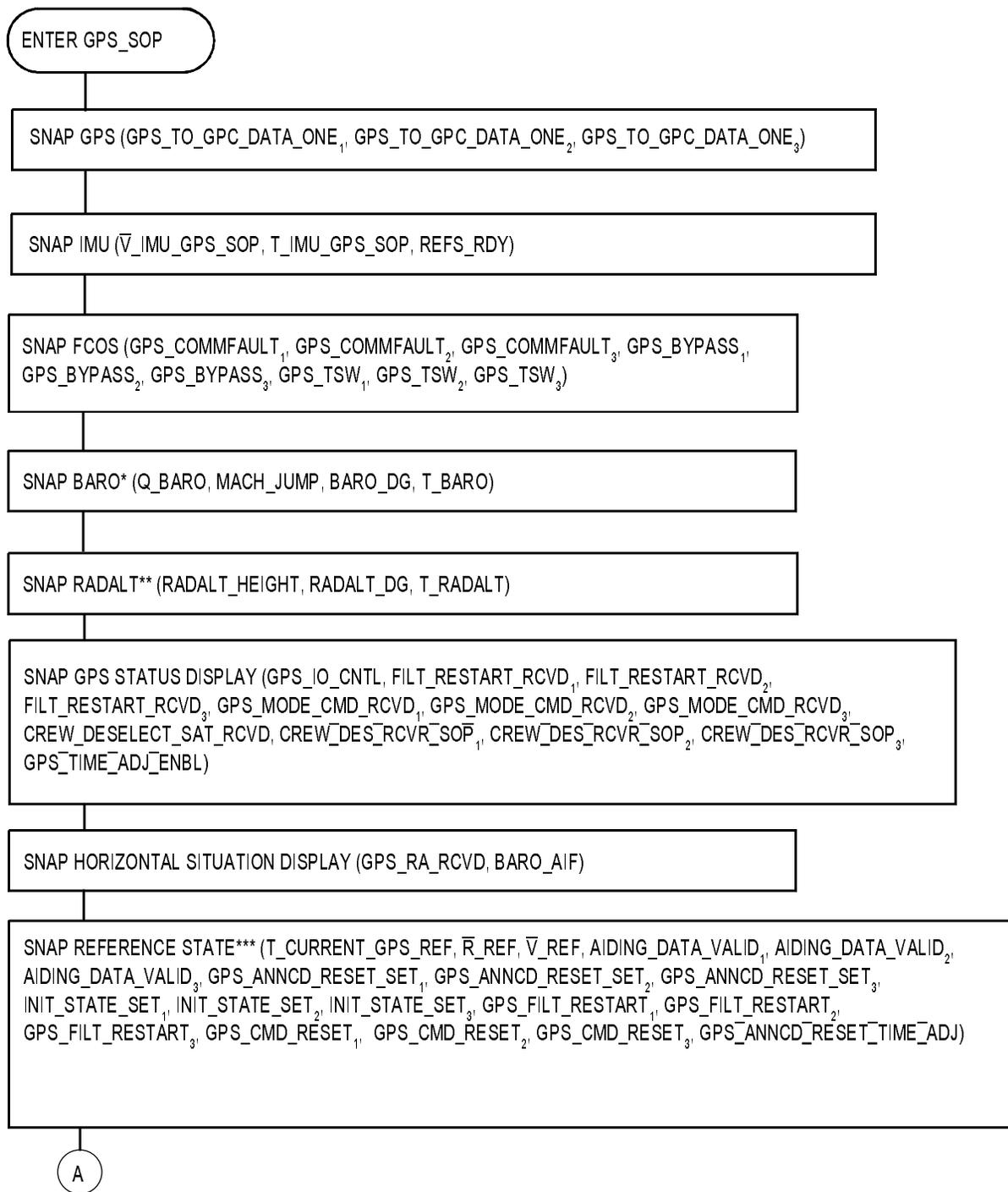
Table 4.17.6.9-1 GPS SOP Time Adjustment Input Parameters							
Description	Name	Input Source	Type	Precision	Range	Units	Rate (Hz)
Flag indicating commfault status for GPS receiver I	GPS_COMMFAULT (I)**	*	D	-	ON/OFF	-	1.04
Flag indicating an announced reset for the GPC-to-GPS time adjustment computation	GPS_ANNCD_RESET_TIME_ADJ	GPS_SOP,*	D	-	ON/OFF	-	6.25
Target value of GPS time minus GPC time	GPC_GPS_TARGET	***	I	S	-	MSEC	6.25
GPS receiver I current operating mode indicator	GPS_MODE_CURRENT(I)**	NAV_MODE_WORD_DATA	CHAR	-	TEST/INIT/NAV/ 'B'	-	6.25
Flag indicating valid GPS NAV receiver I data	GPS_NAV_DATA_VALID(I)**	NAV_MODE_WORD_DATA	D	-	ON/OFF	-	6.25
Crew deselected GPS receiver I	CREW_DES_RCVR_SOP(I)**	GPS_SOP	D	-	ON/OFF	-	6.25
Age-of-aiding data from GPS receiver I	GPS_AOA(I)**	NAV_MESSAGE_DATA	HXS	-	-	MSEC	6.25
<p>* See principal function input parameter listed in Table 4.17.6.10-1 ** I ranges from 1 to GPS_SW_CAP *** See principal function I-Loads parameters, Table 4.17.6.10-3</p>							

Table 4.17.6.9-2 GPS SOP Time Adjustment Output Parameters							
Description	Name	Destination	Type	Precision	Range	Units	Rate (Hz)
Selected age-of-aiding data from GPS receivers	GPS_AOA_SEL	*	HXS	-	-	MSEC	6.25
Accumulated GPC-to-GPS time adjustment	T_GPS_TIME_ADJ_ACC	*, RCVR_AIDING_DATA, GPS_ALT_DATA_CODE, NAV_MESSAGE_DATA	HXS	-	-	MSEC	6.25
* See principal function output parameter listed in Table 4.17.6.10-2							



* SINGLE STRING GPS ONLY

FIGURE 4.17.1.1-1. GPS Subsystem Block Diagram



* Air data shall be snapped in Major Mode 304, 305, 602 and 603 only.

** Radar altimeter data shall be snapped in Major Mode 305 and 603 only.

*** GPS Reference State Proc data shall not be collected in OPS 9.

FIGURE 4.17.6-1.(1) GPS_SOP (Sheet 1 of 3)

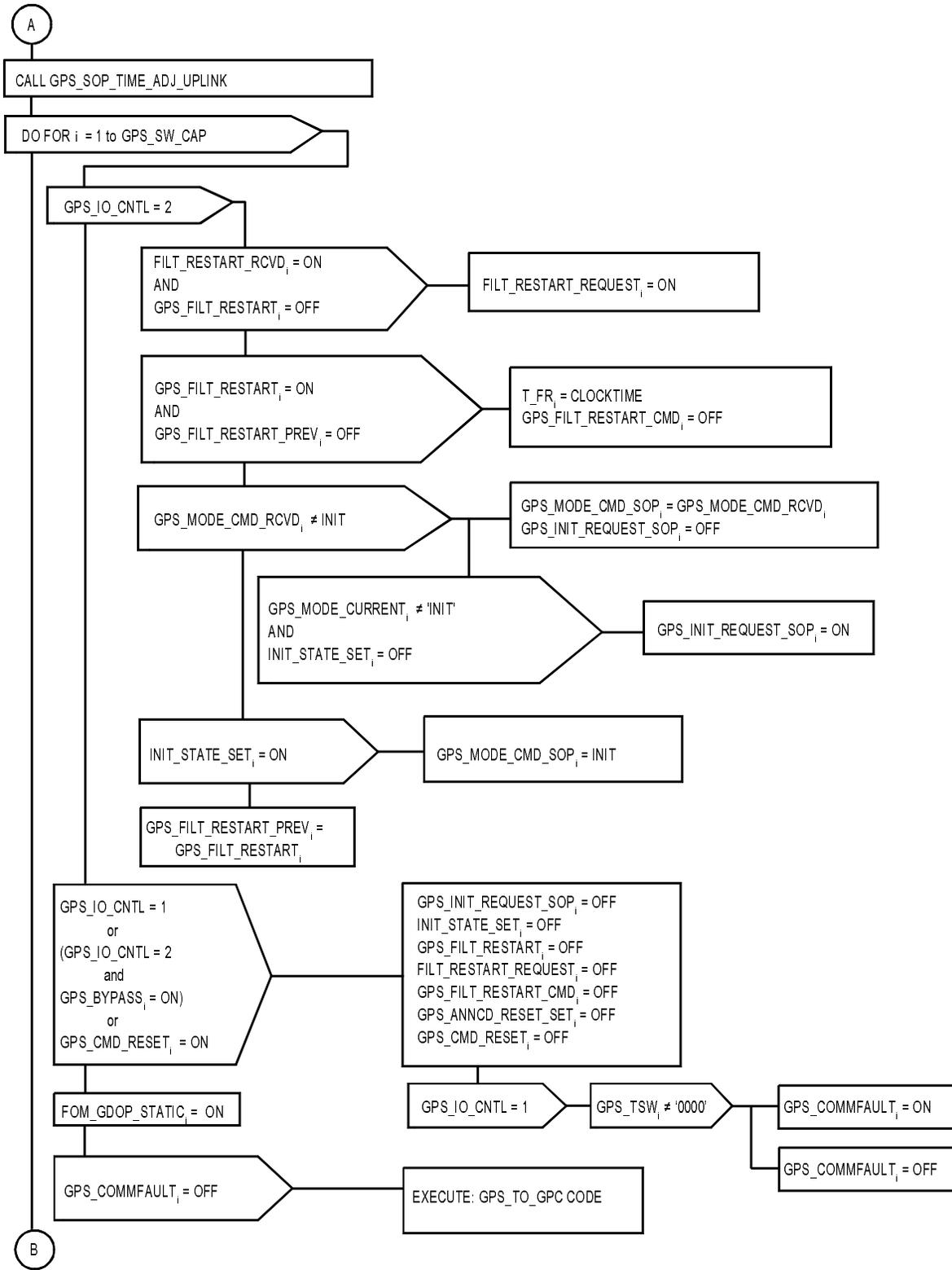


FIGURE 4.17.6-1.(2) GPS_SOP (Sheet 2 of 3)

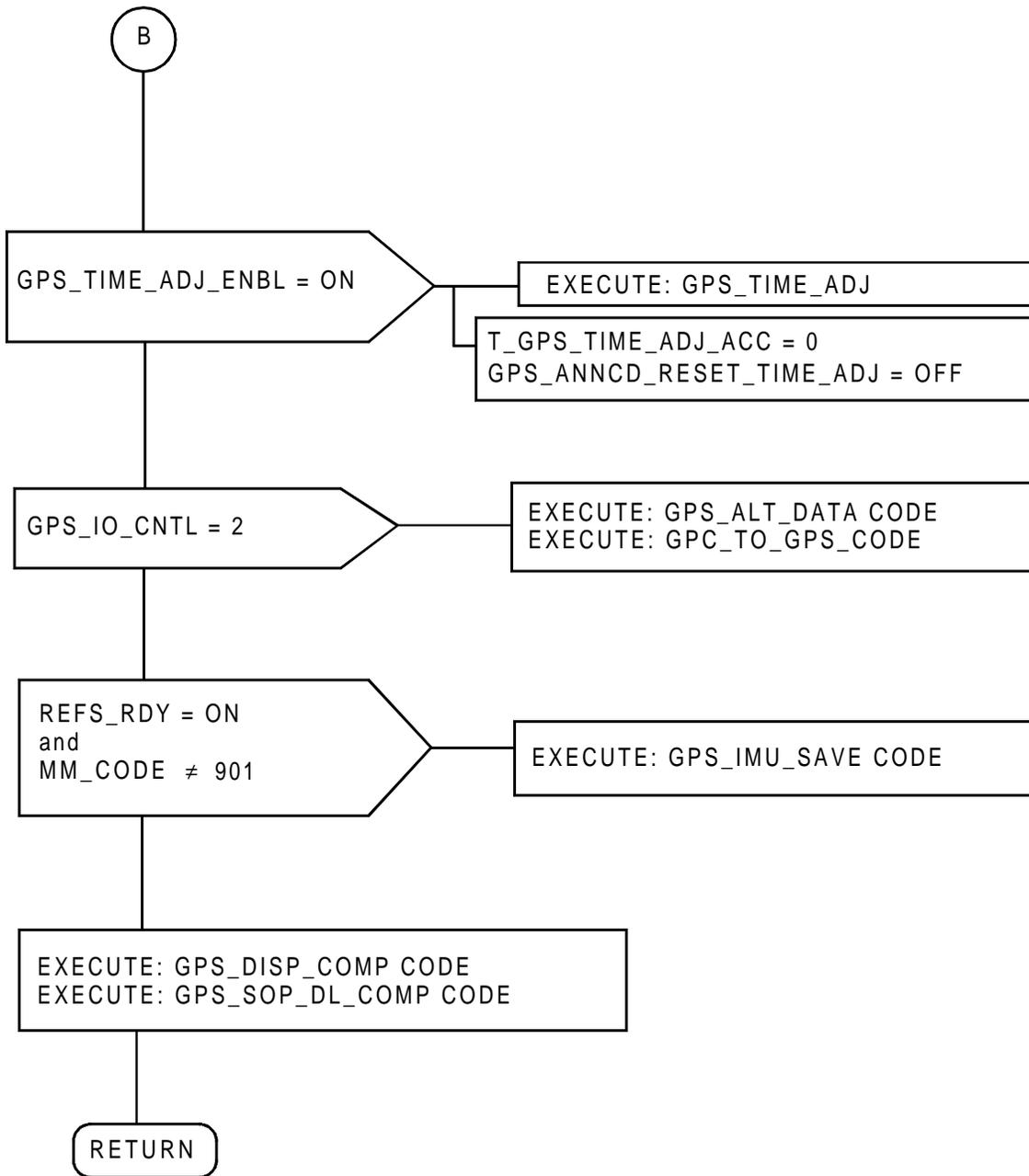


FIGURE 4.17.6-1.(3) GPS_SOP (Sheet 3 of 3)

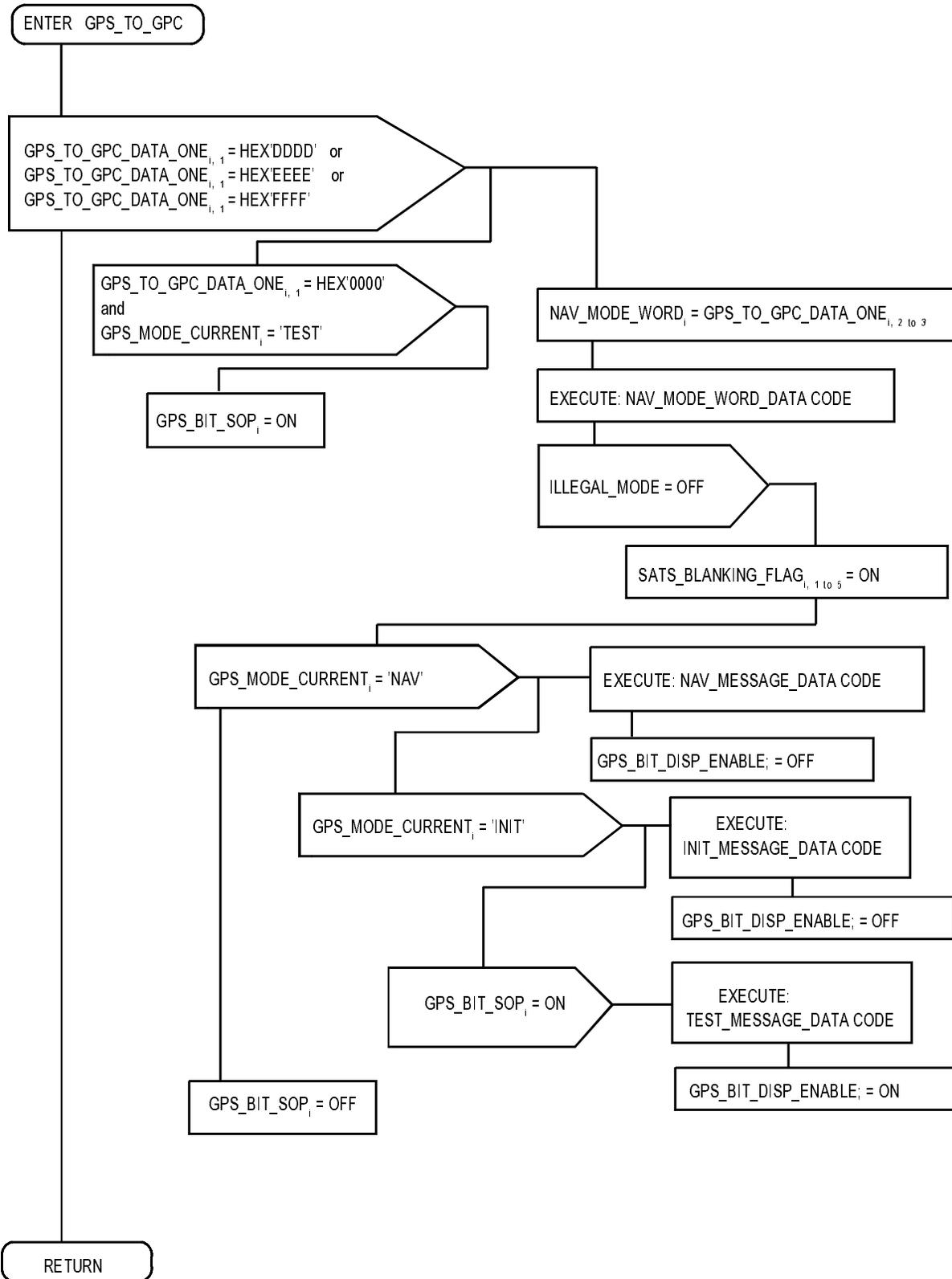


FIGURE 4.17.6.1. GPS_TO_GPC CODE

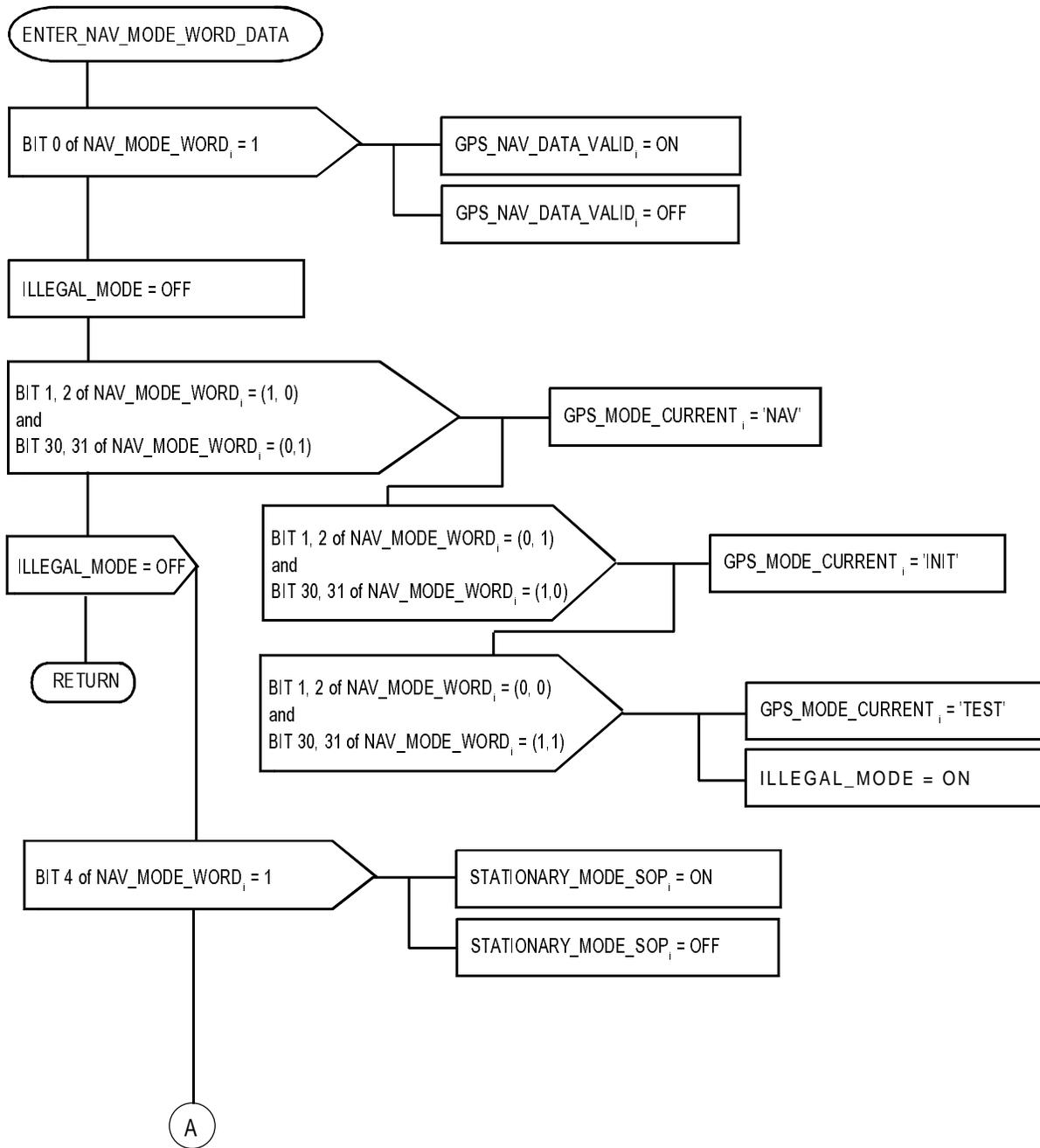


FIGURE 4.17.6.1.1.(1) NAV_MODE_WORD_DATA (Sheet 1 of 2)

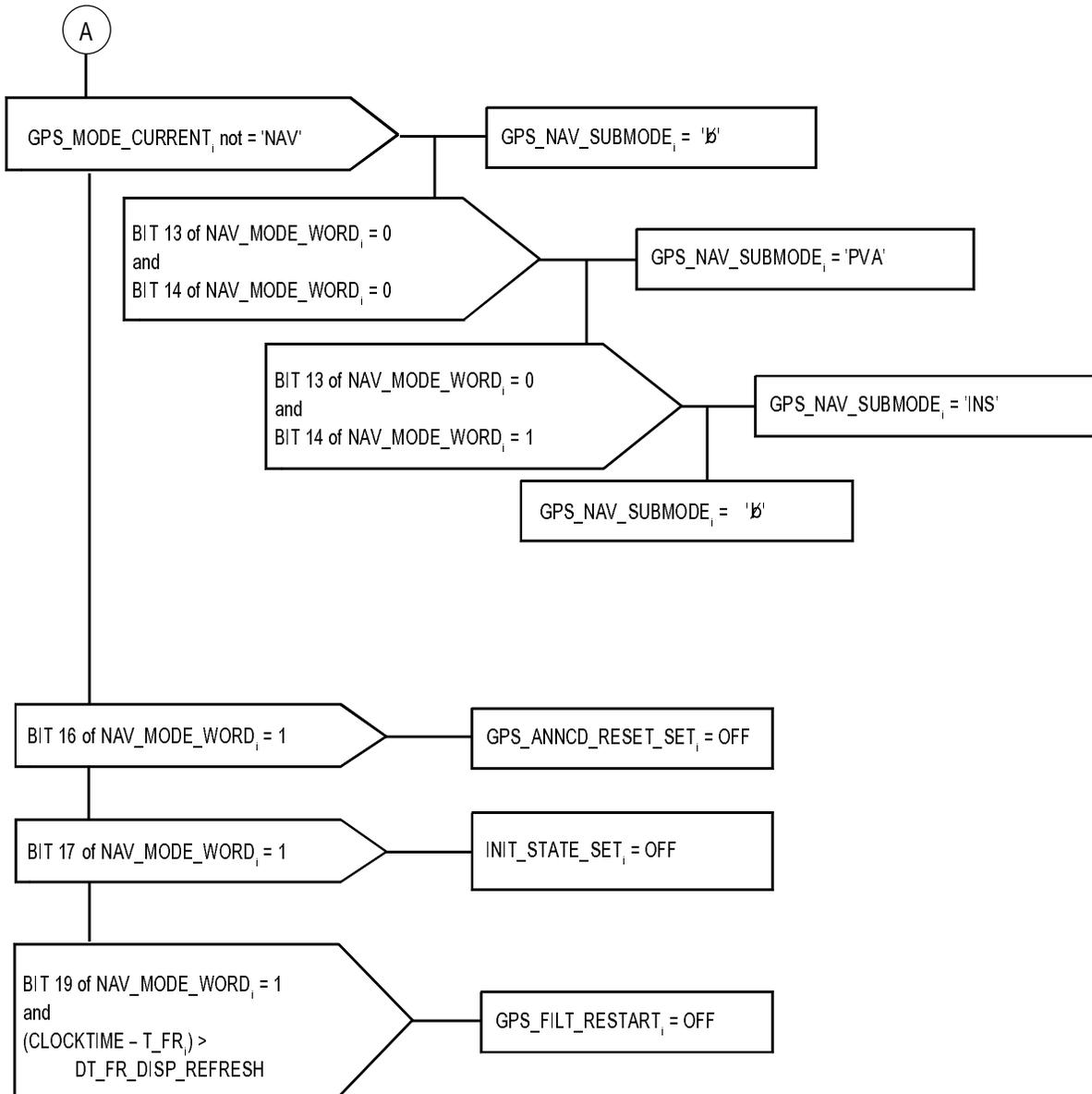


FIGURE 4.17.6.1.1.(2) NAV_MODE_WORD_DATA (Sheet 2 of 2)

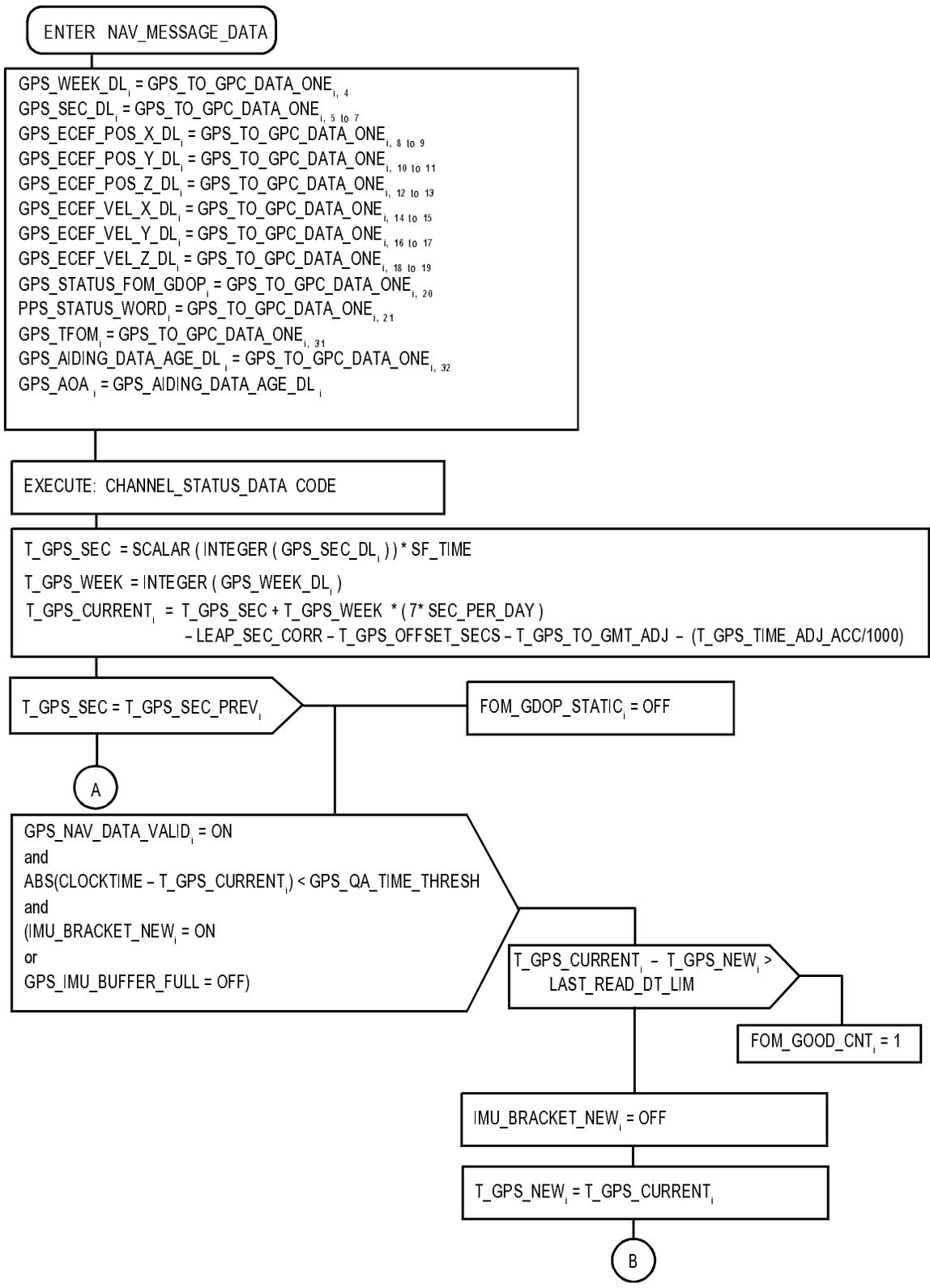


FIGURE 4.17.6.1.2.(1) NAV_MESSAGE_DATA CODE (Sheet 1 of 2)

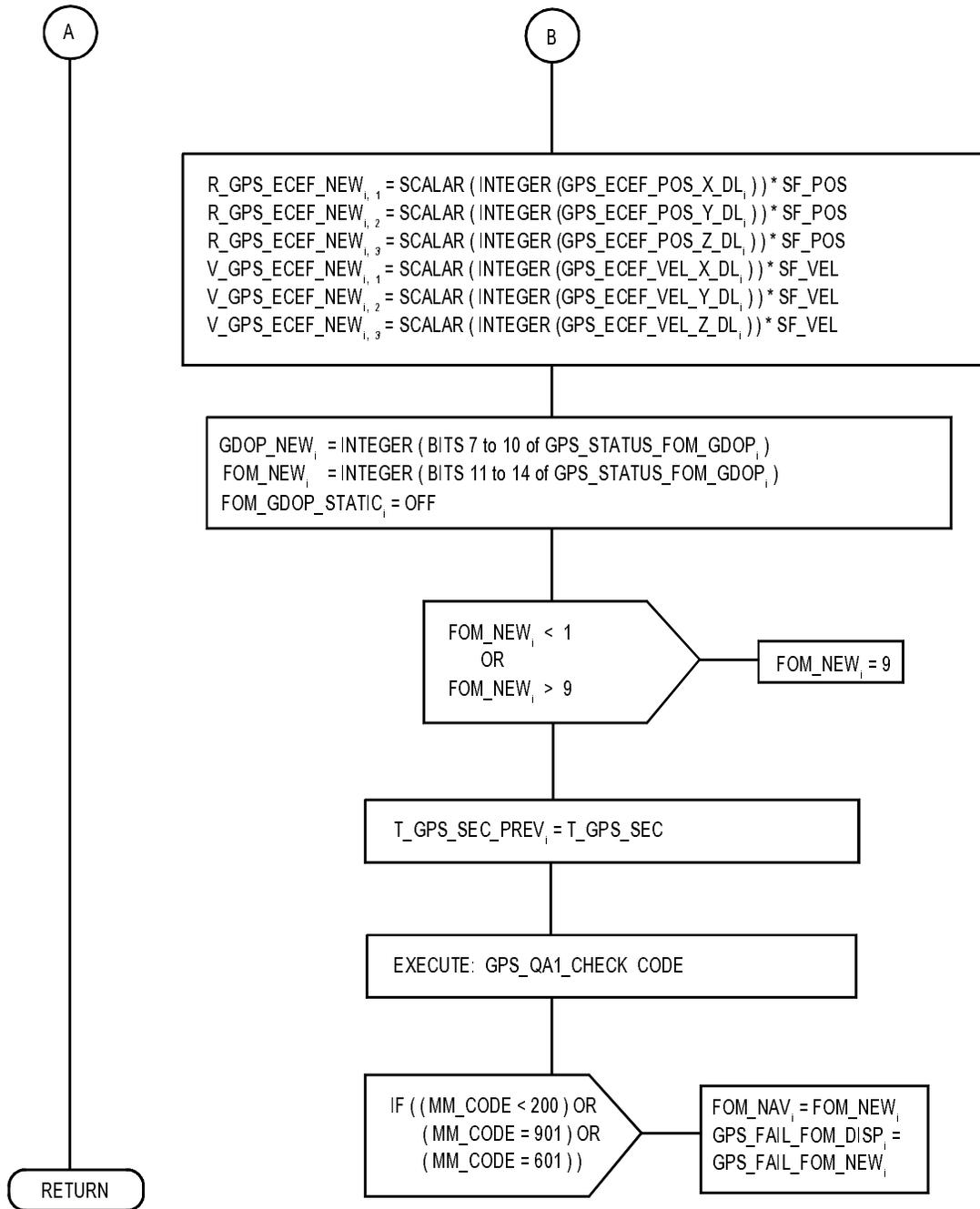


FIGURE 4.17.6.1.2.(2) NAV_MESSAGE_DATA CODE (Sheet 2 of 2)

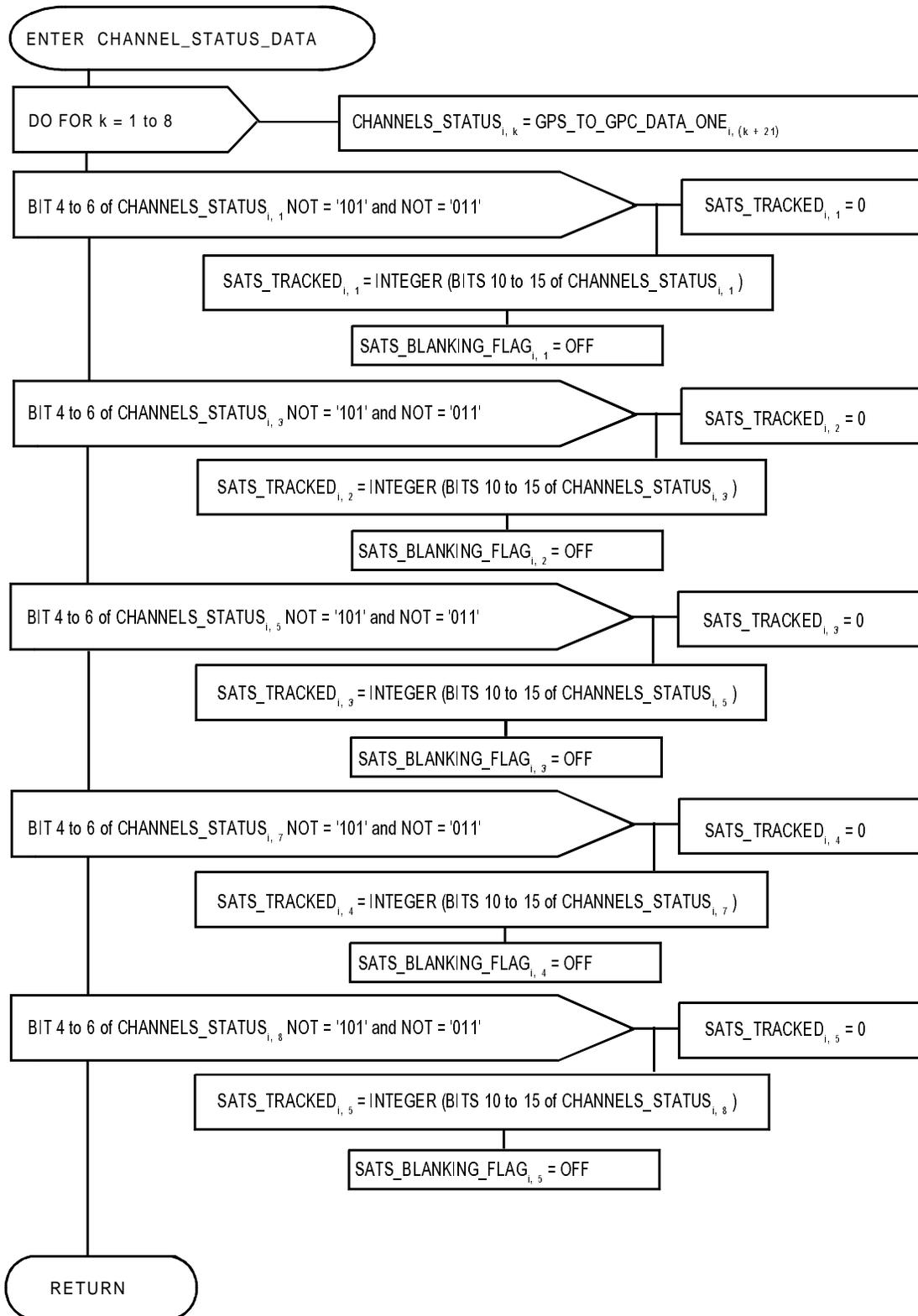


FIGURE 4.17.6.1.2.1. CHANNEL_STATUS_DATA CODE

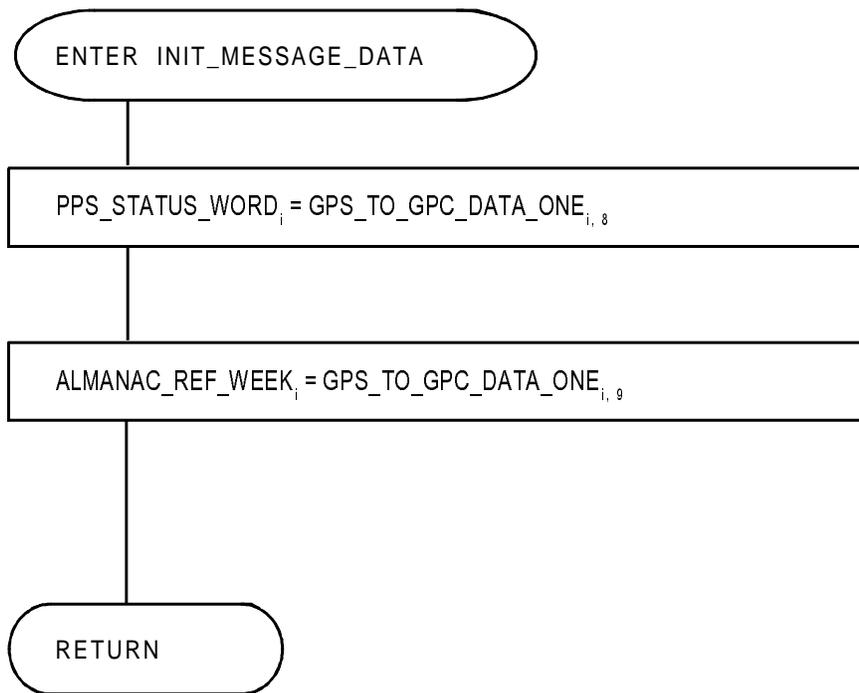


FIGURE 4.17.6.1.3. INIT_MESSAGE_DATA CODE

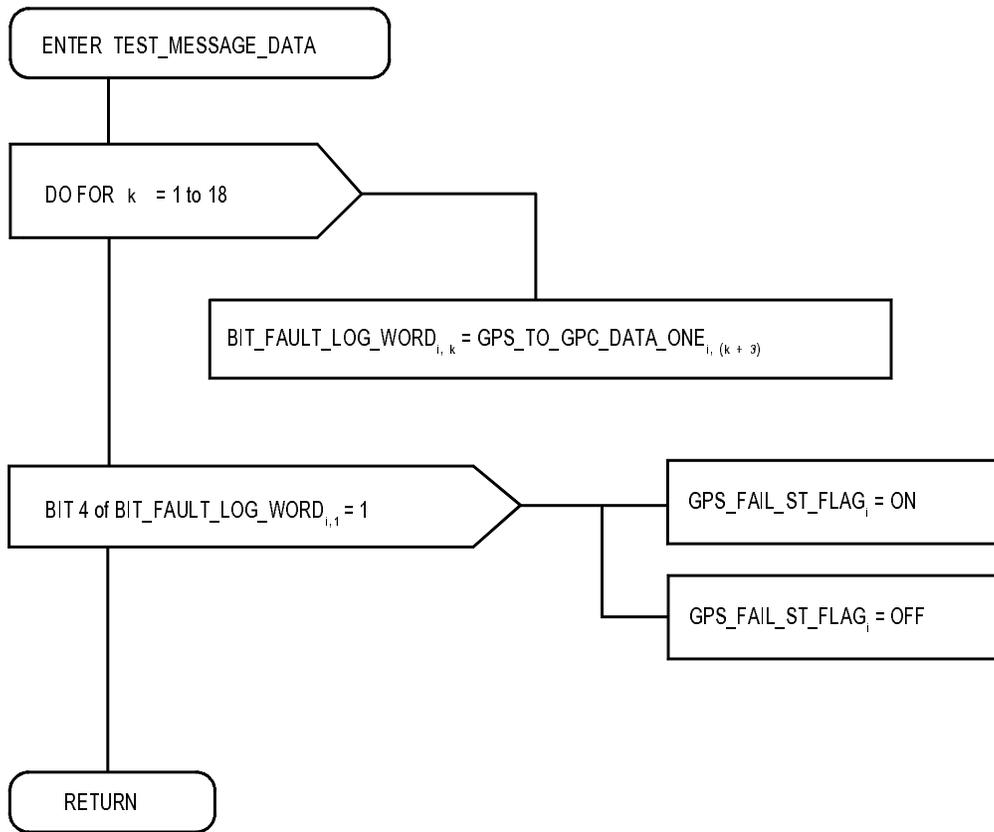


FIGURE 4.17.6.1.4. TEST_MESSAGE_DATA CODE

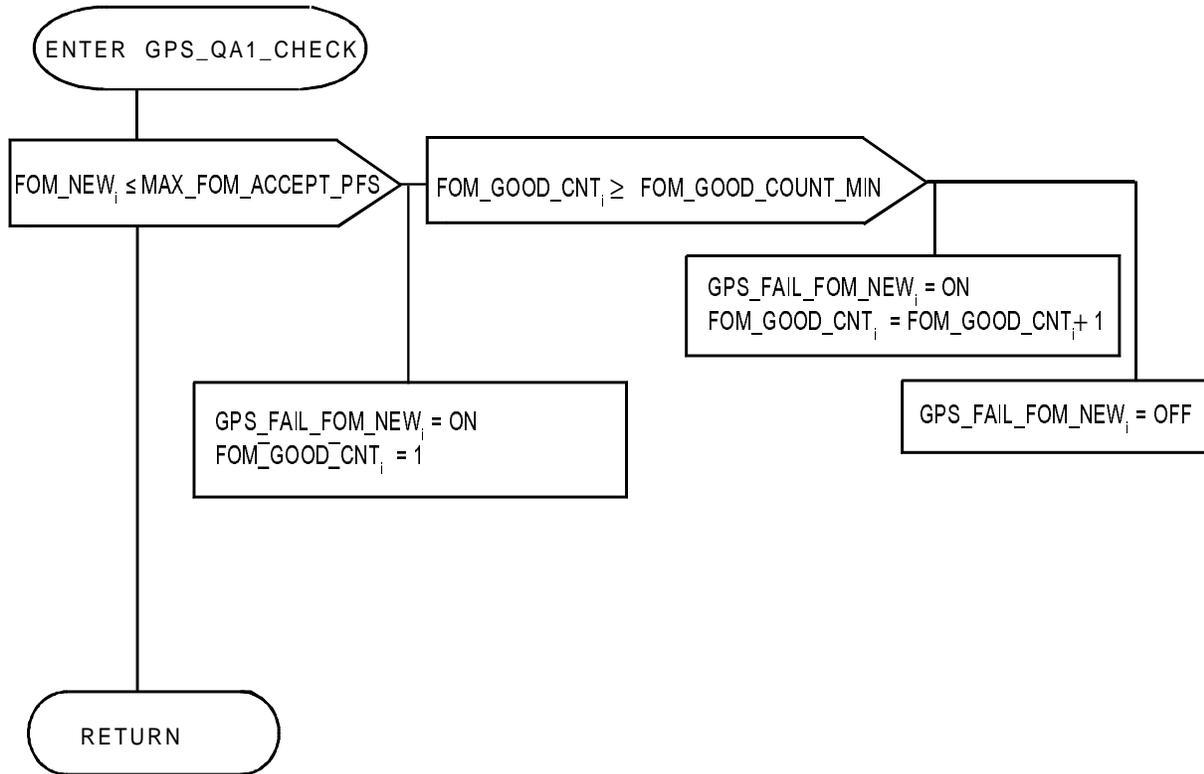


FIGURE 4.17.6.2. GPS_QA1_CHECK CODE

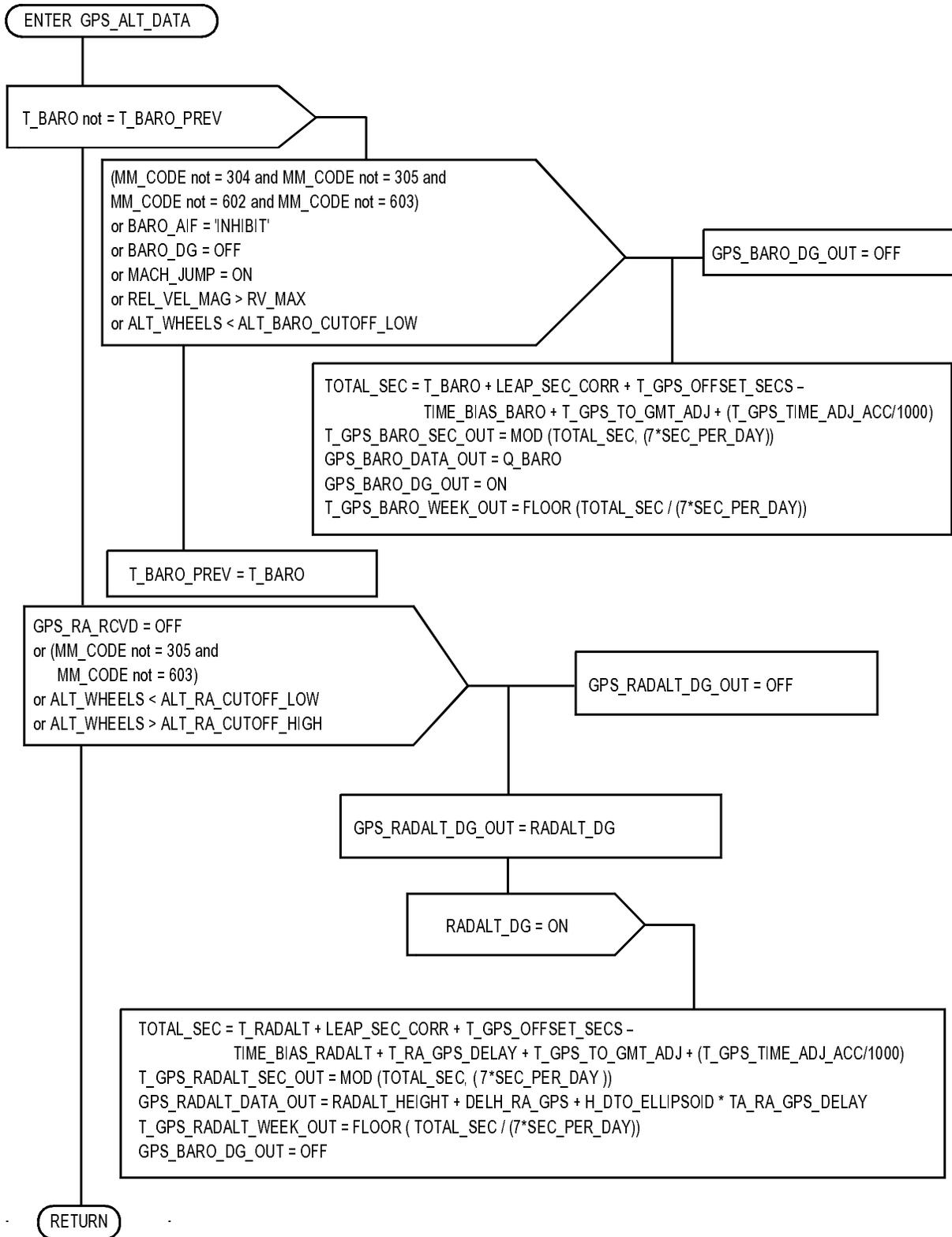


FIGURE 4.17.6.3. GPS_ALT_DATA CODE

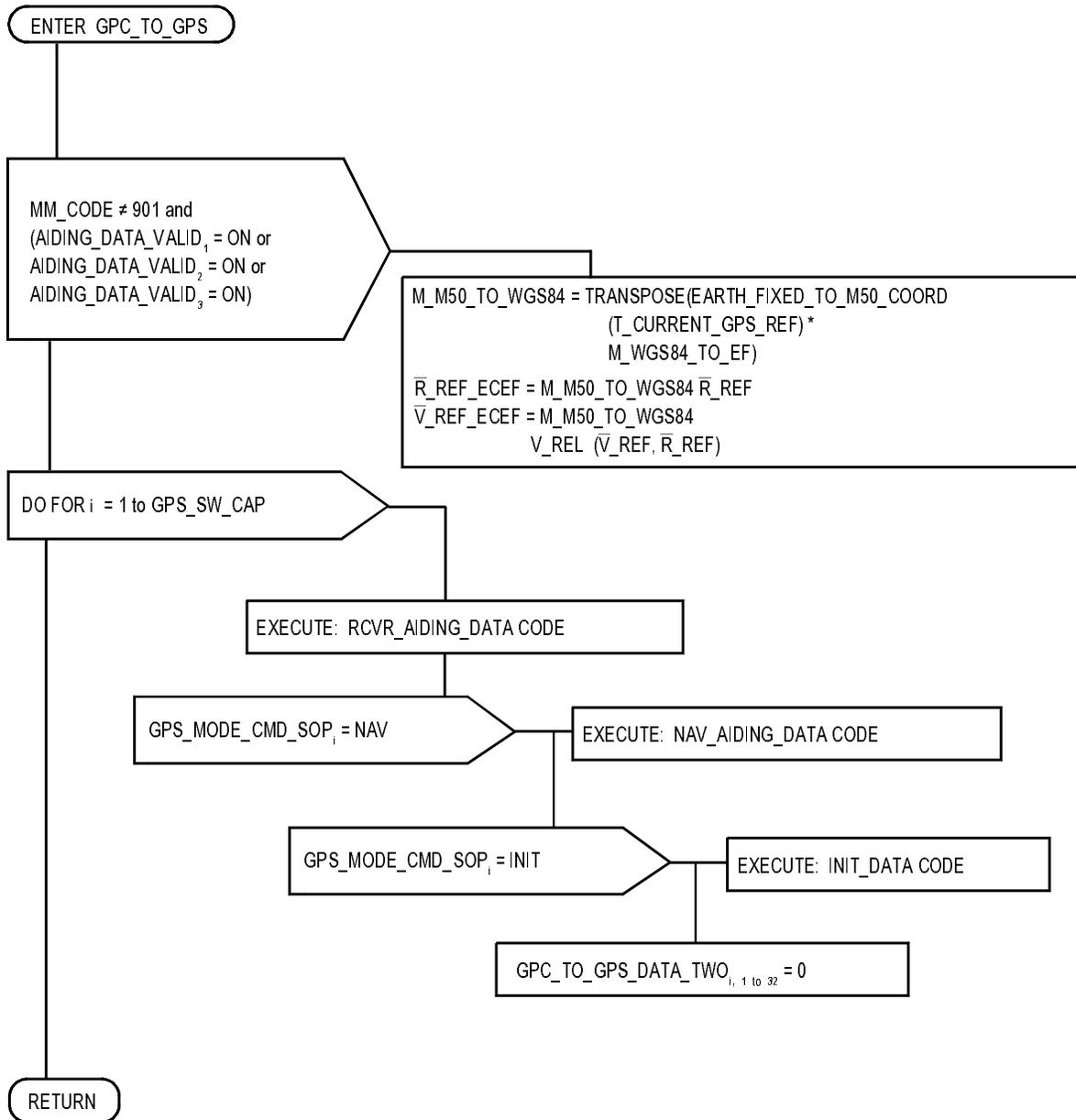


FIGURE 4.17.6.4. GPC_TO_GPS CODE

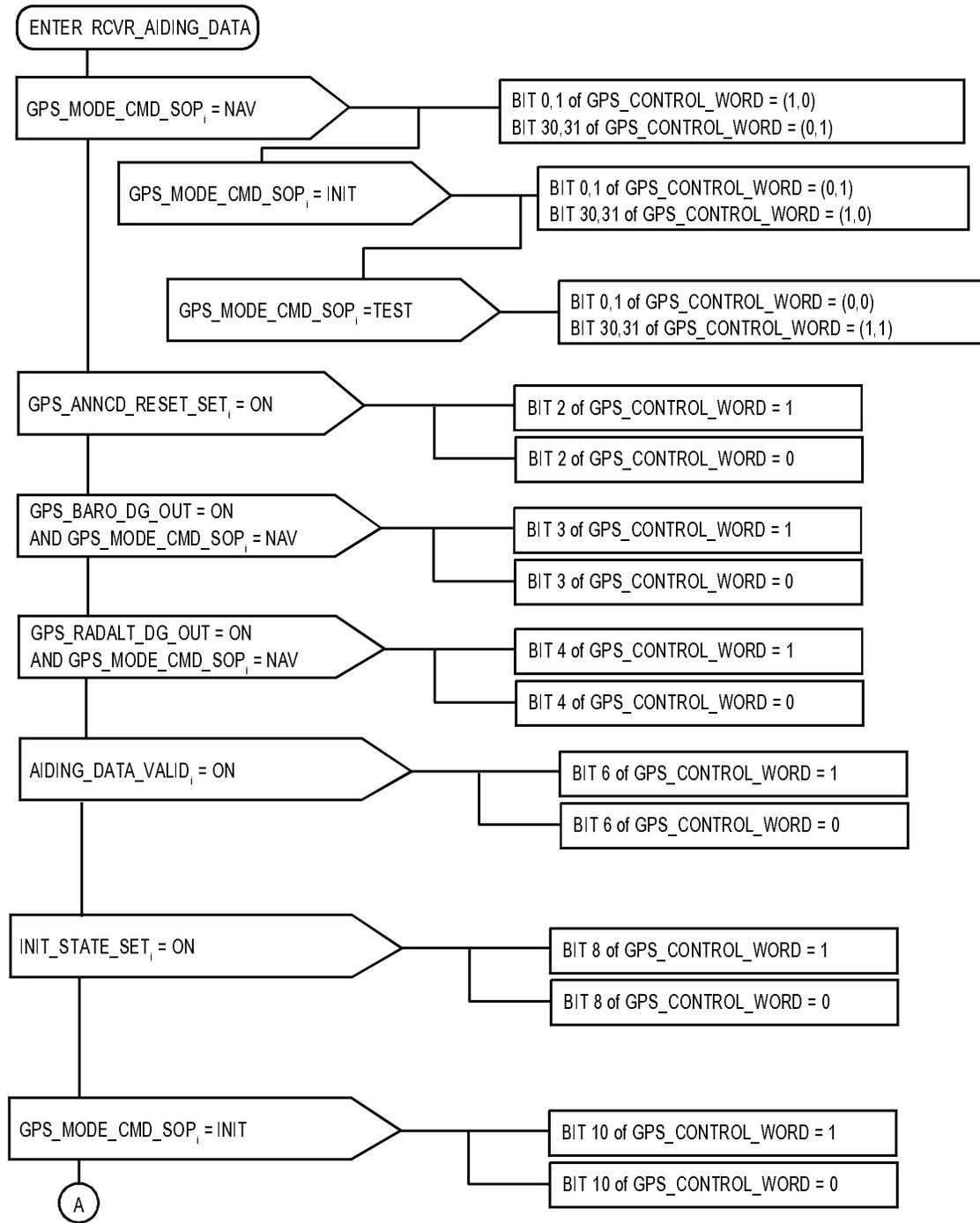


FIGURE 4.17.6.4.1.(1)RCVR_AIDING_DATA CODE (Sheet 1 of 4)

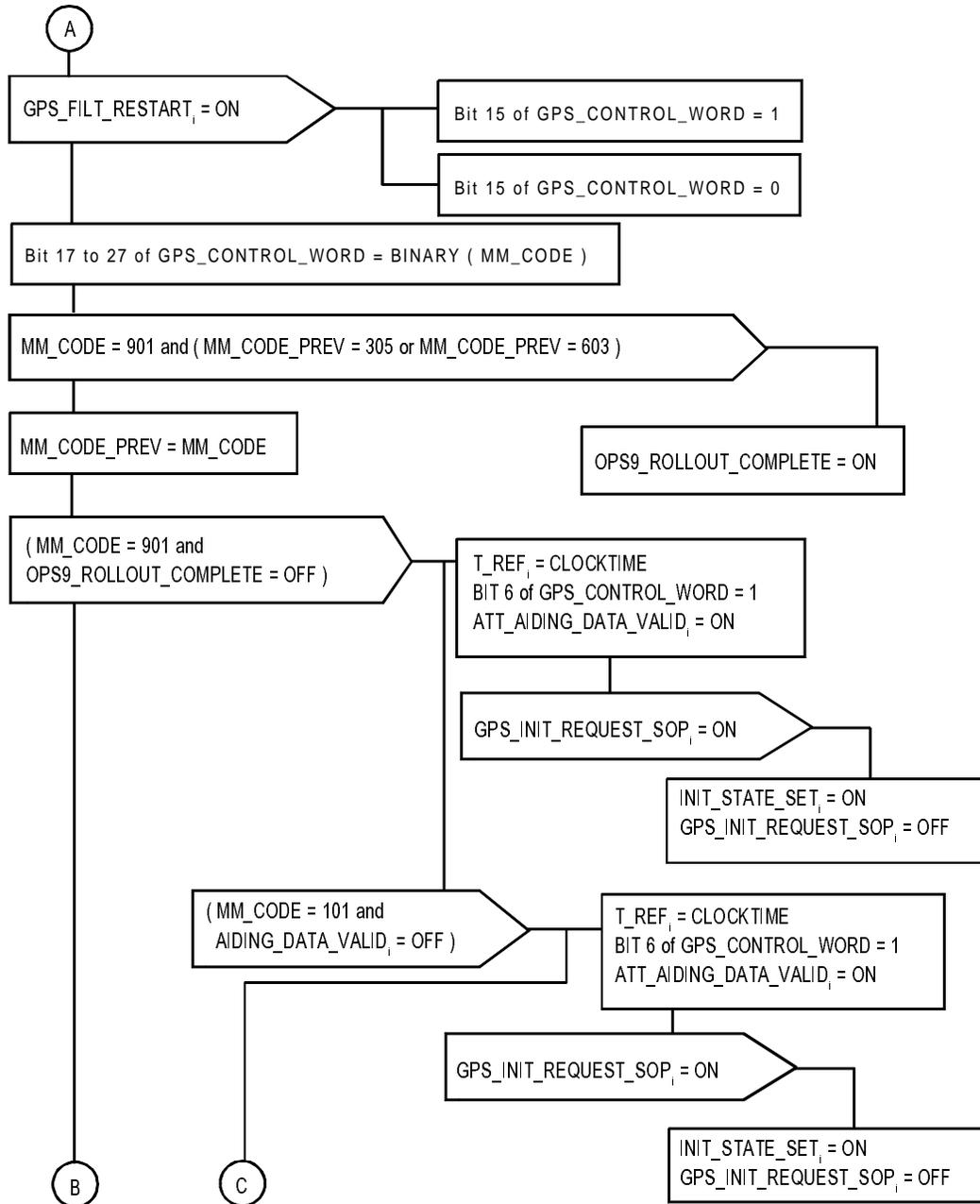


FIGURE 4.17.6.4.1.(2) RCVR_AIDING_DATA CODE (Sheet 2 of 4)

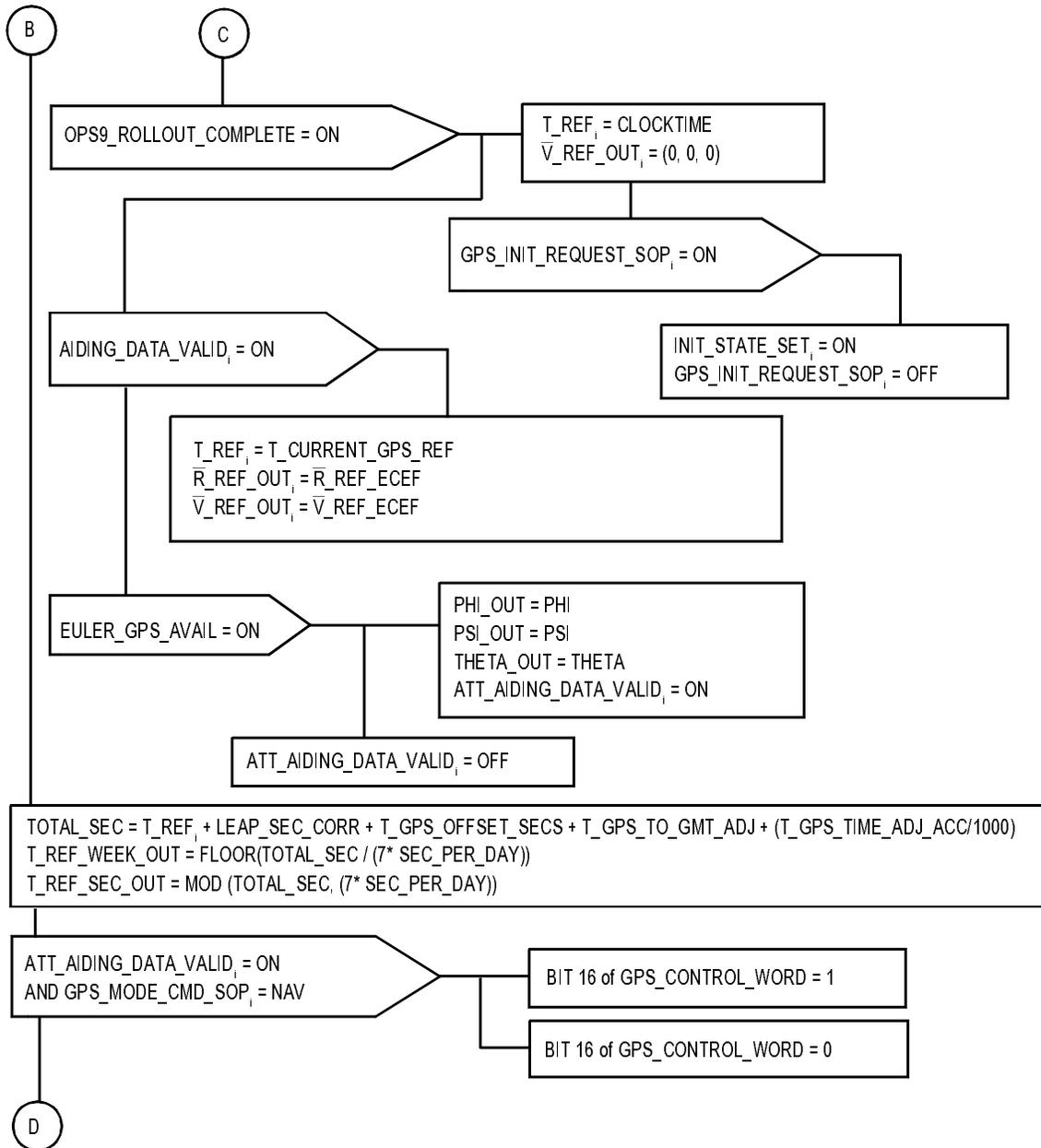


FIGURE 4.17.6.4.1.(3)RCVR_AIDING_DATA CODE (Sheet 3 of 4)

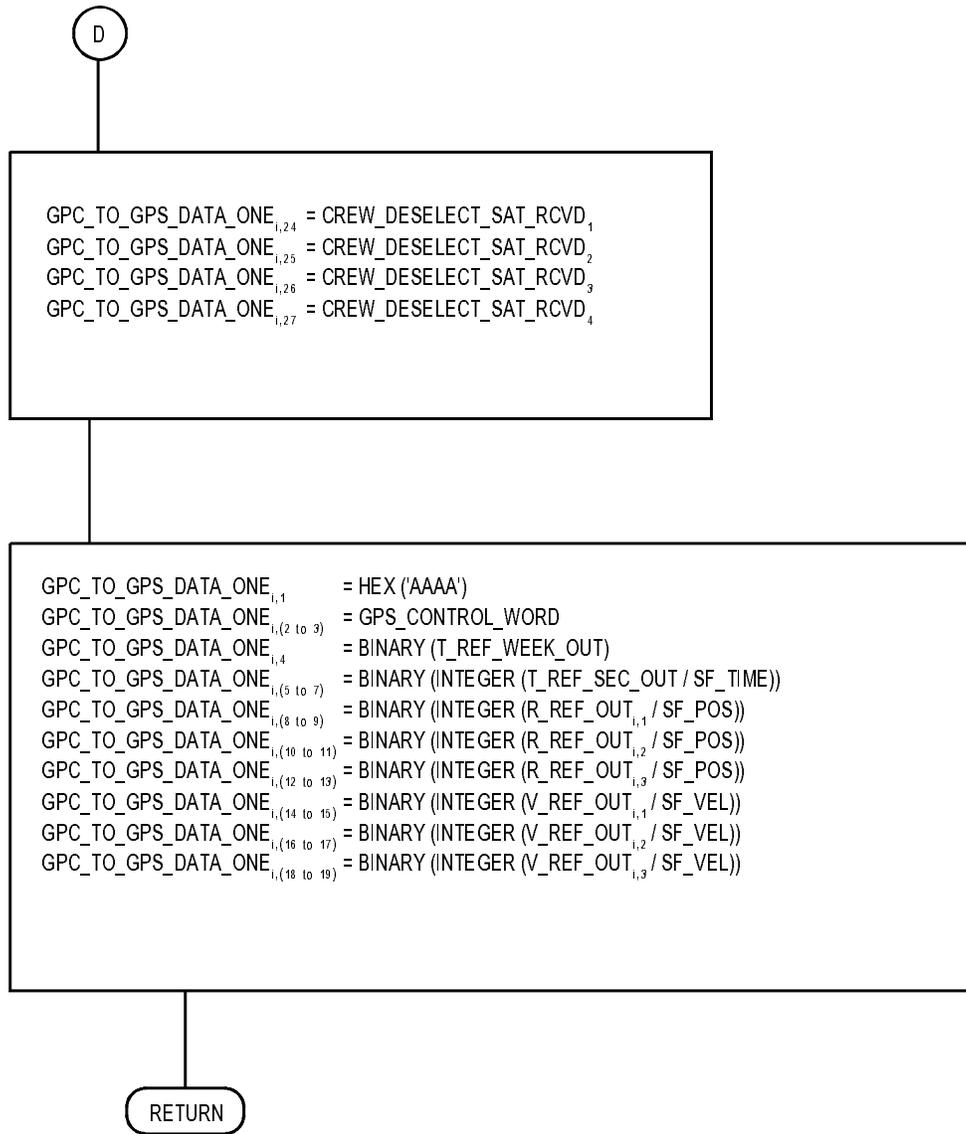


FIGURE 4.17.6.4.1.(4)RCVR_AIDING_DATA CODE (Sheet 4 of 4)



FIGURE 4.17.6.4.2. INIT_DATA CODE

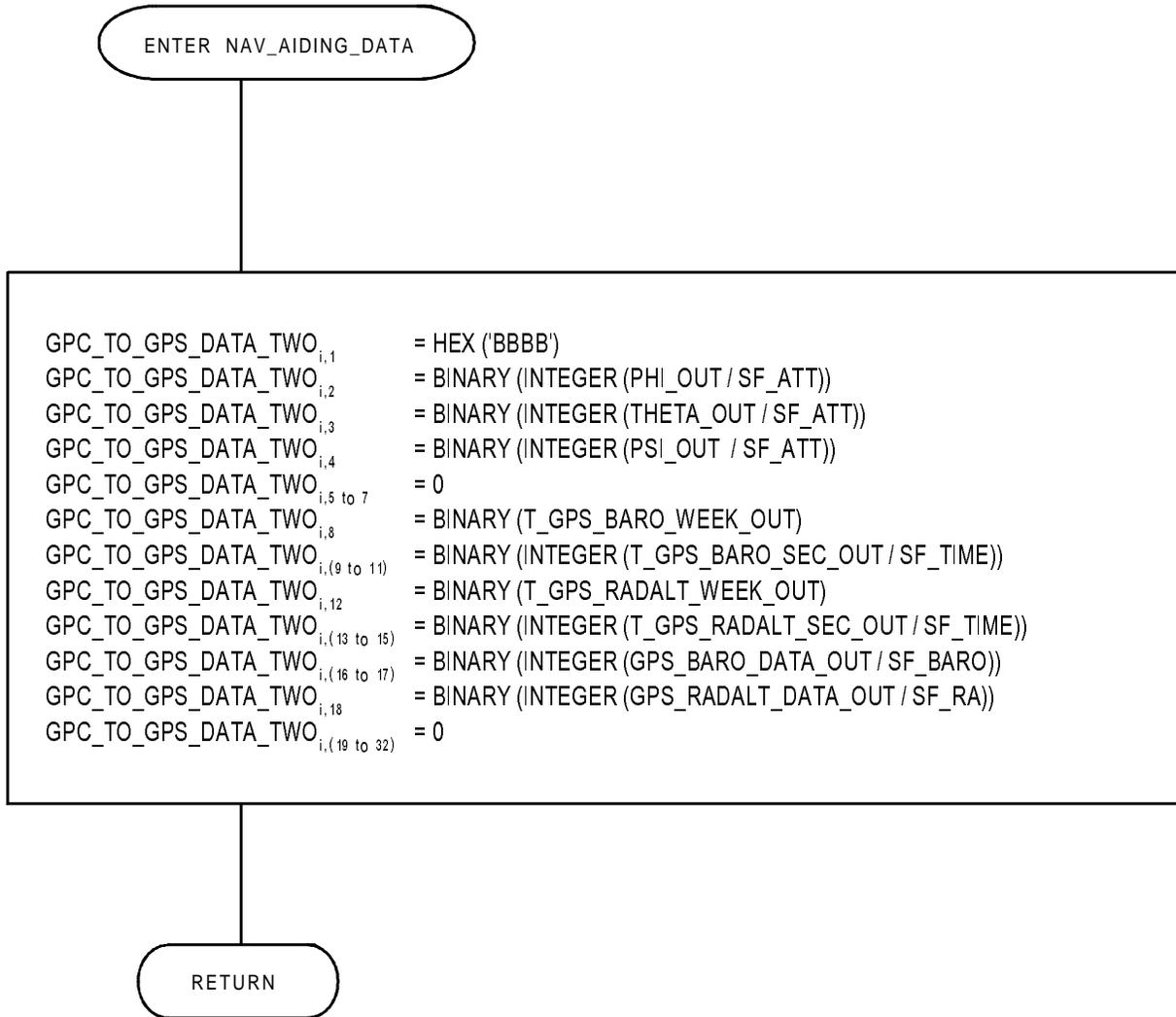


FIGURE 4.17.6.4.3. NAV_AIDING_DATA CODE

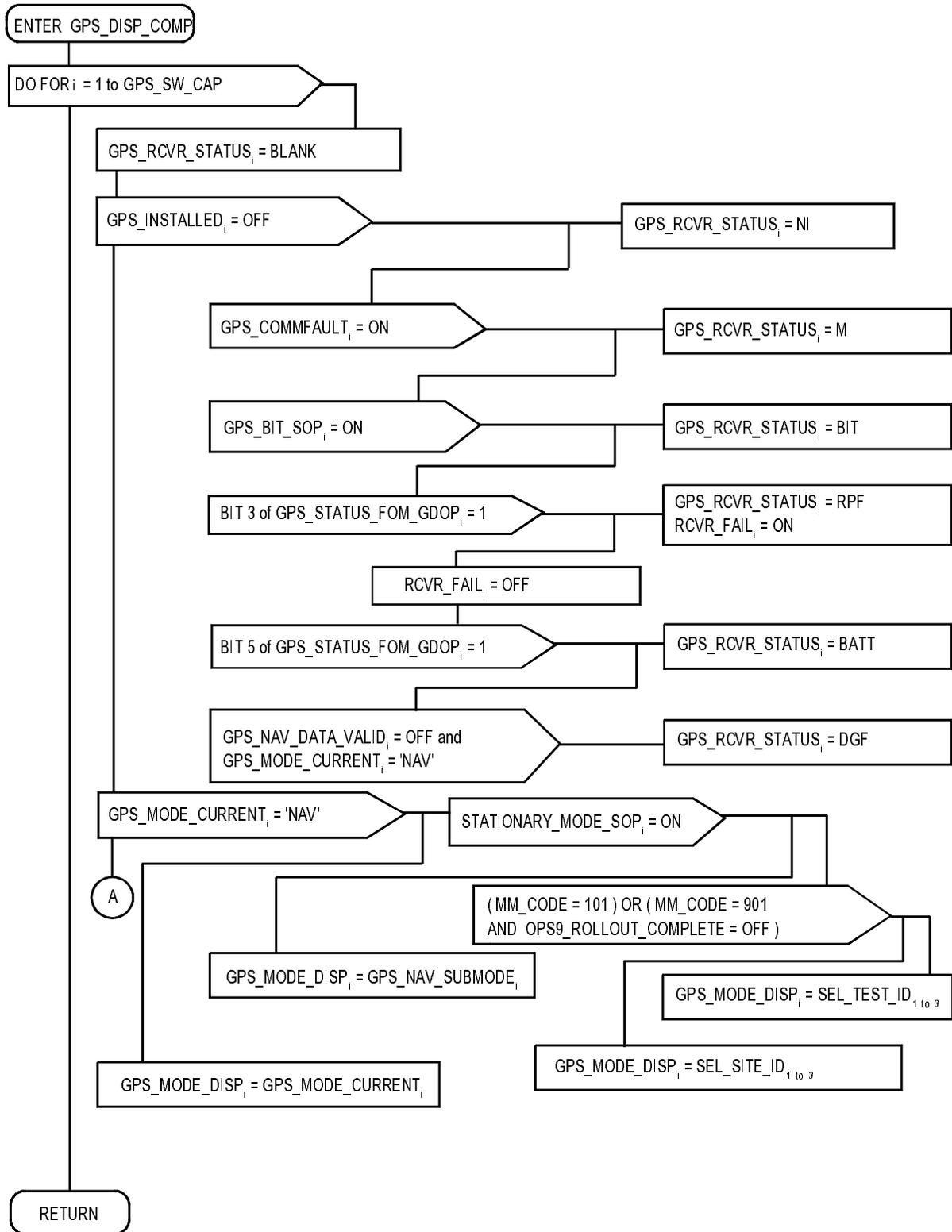


FIGURE 4.17.6.5.(1) GPS_DISP_COMP CODE (Sheet 1 of 2)

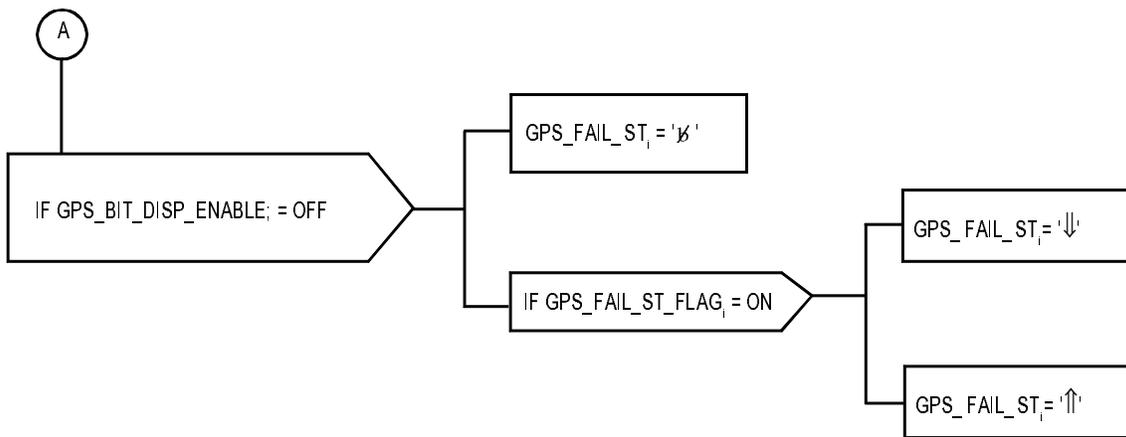


FIGURE 4.17.6.5.(2) GPS_DISP_COMP CODE (Sheet 2 of 2)

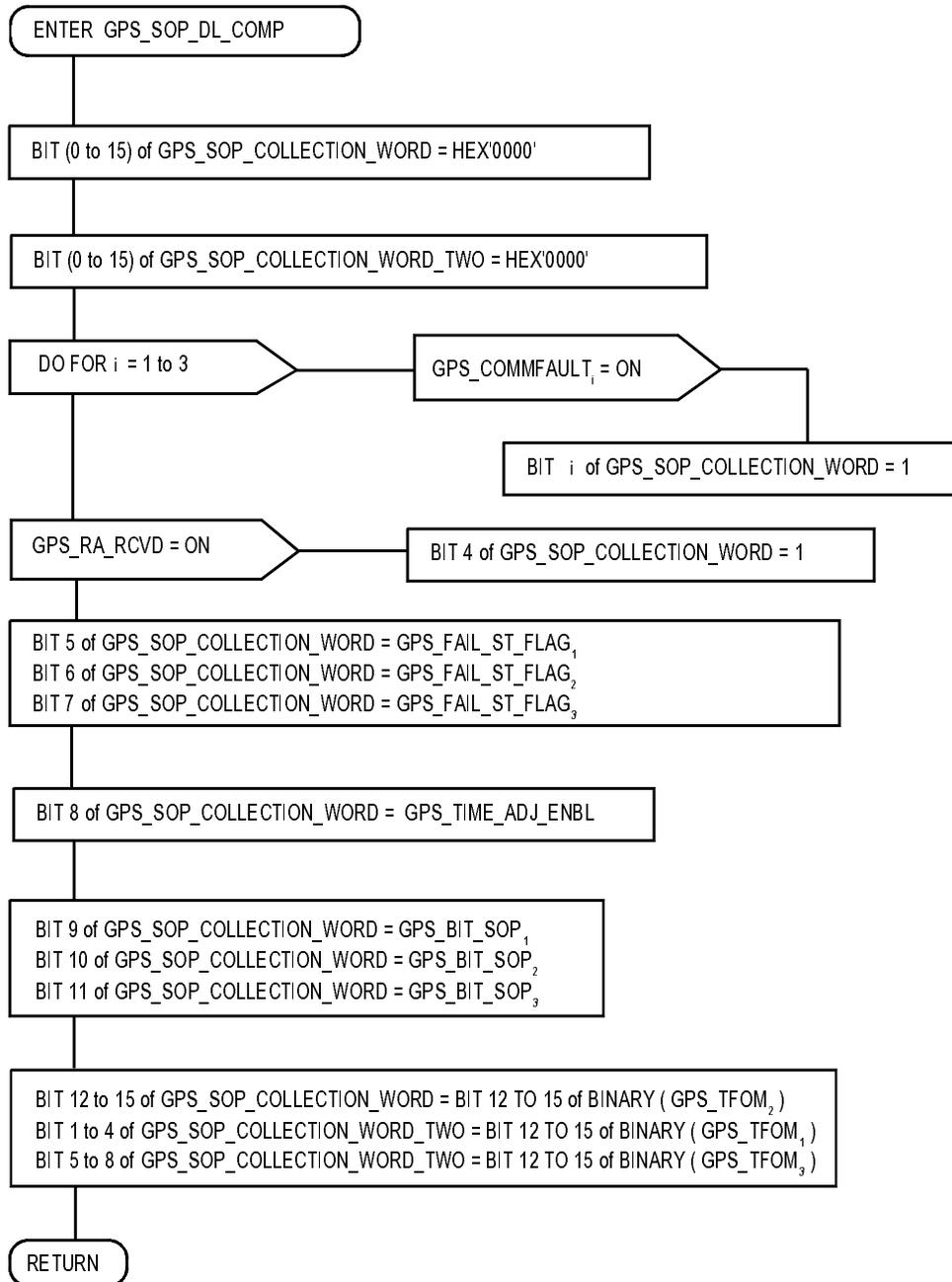


FIGURE 4.17.6.6. GPS_SOP_DL_COMP CODE

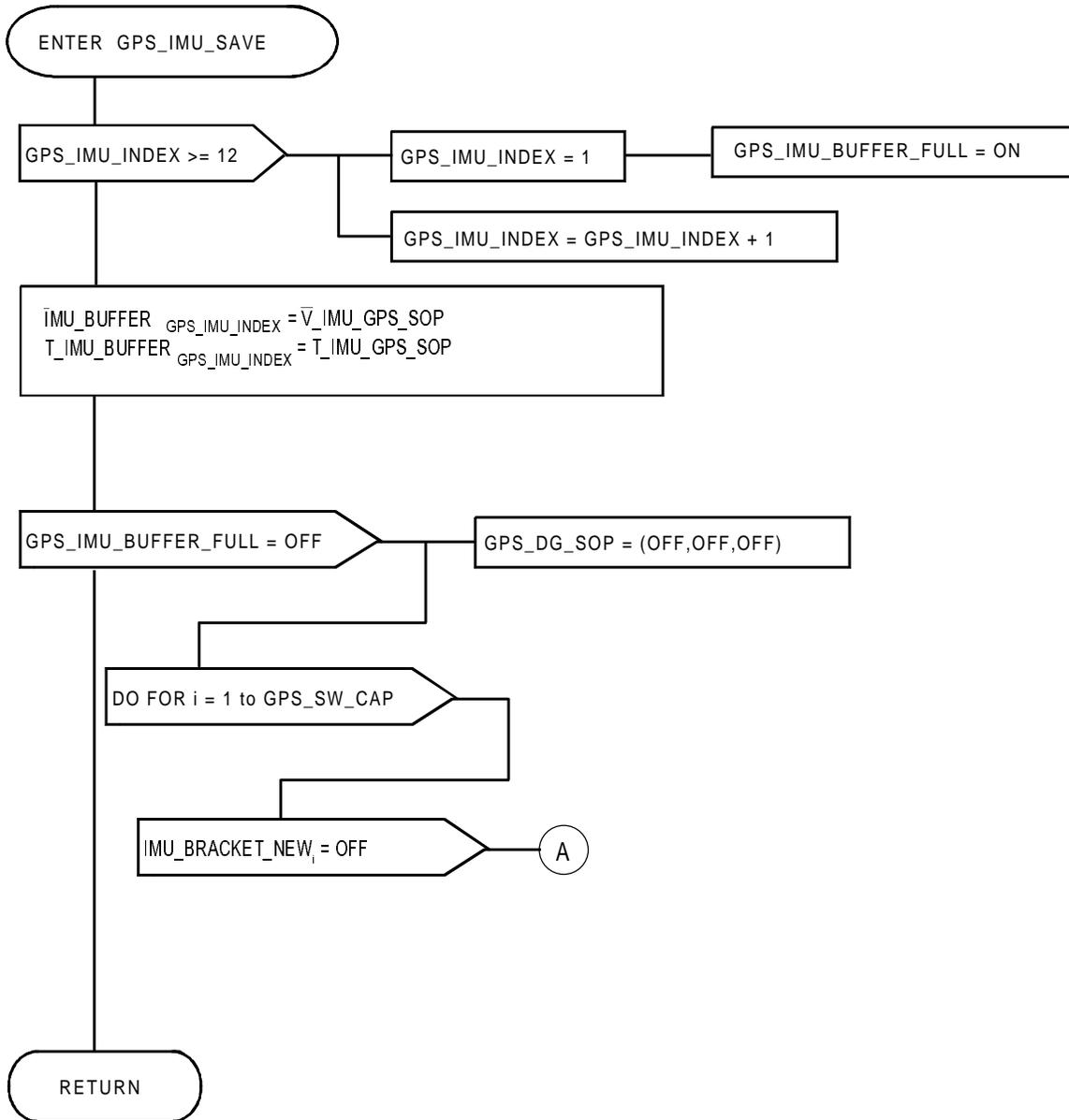


FIGURE 4.17.6.7.(1) GPS_IMU_SAVE CODE (Sheet 1 of 2)

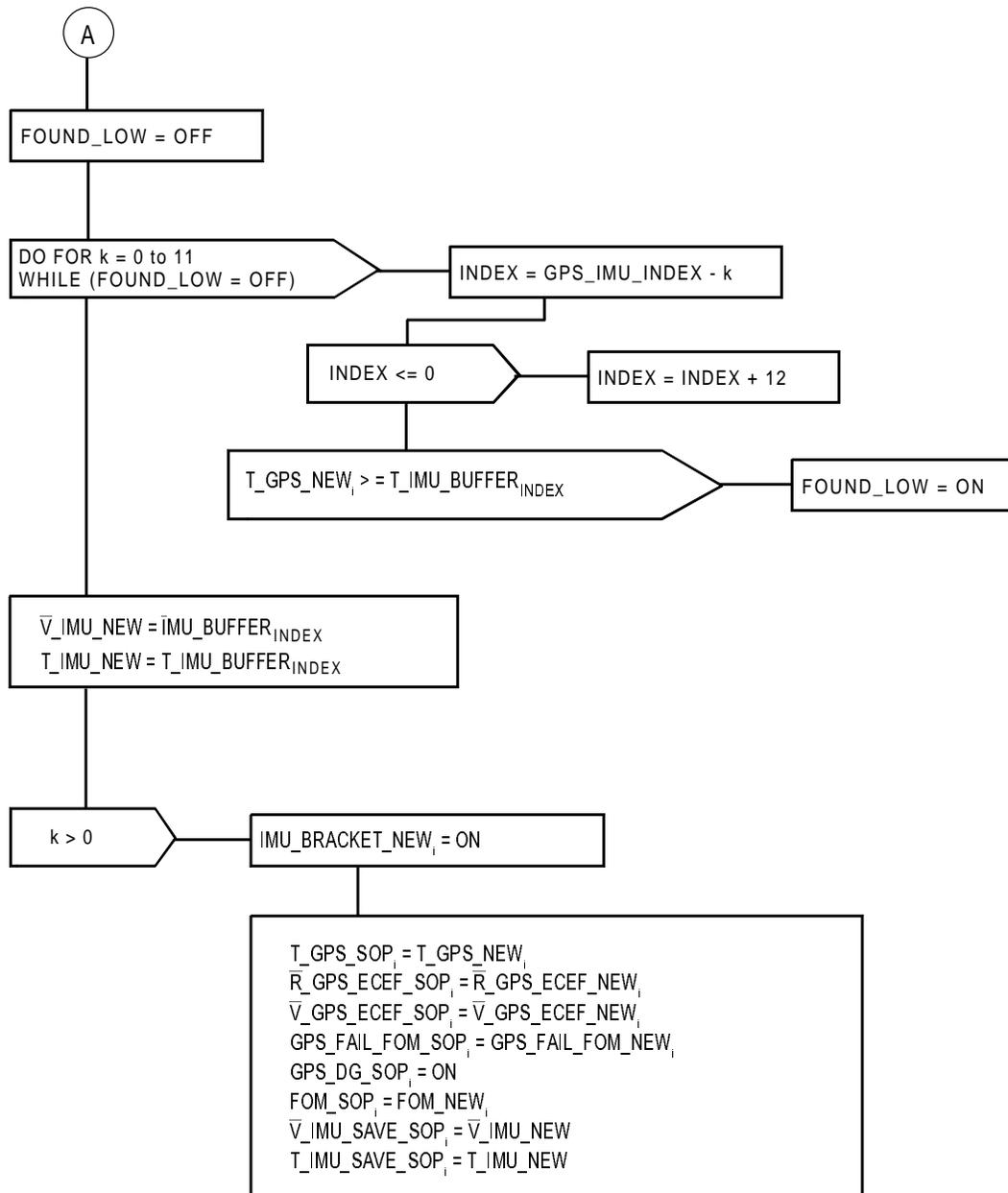


FIGURE 4.17.6.7.(2) GPS_IMU_SAVE CODE (Sheet 2 of 2)

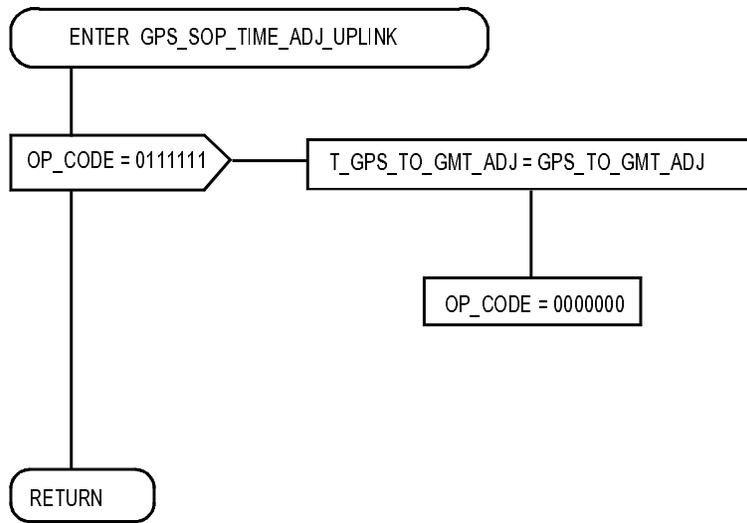


FIGURE 4.17.6.8. GPS_SOP_TIME_ADJ_UPLINK

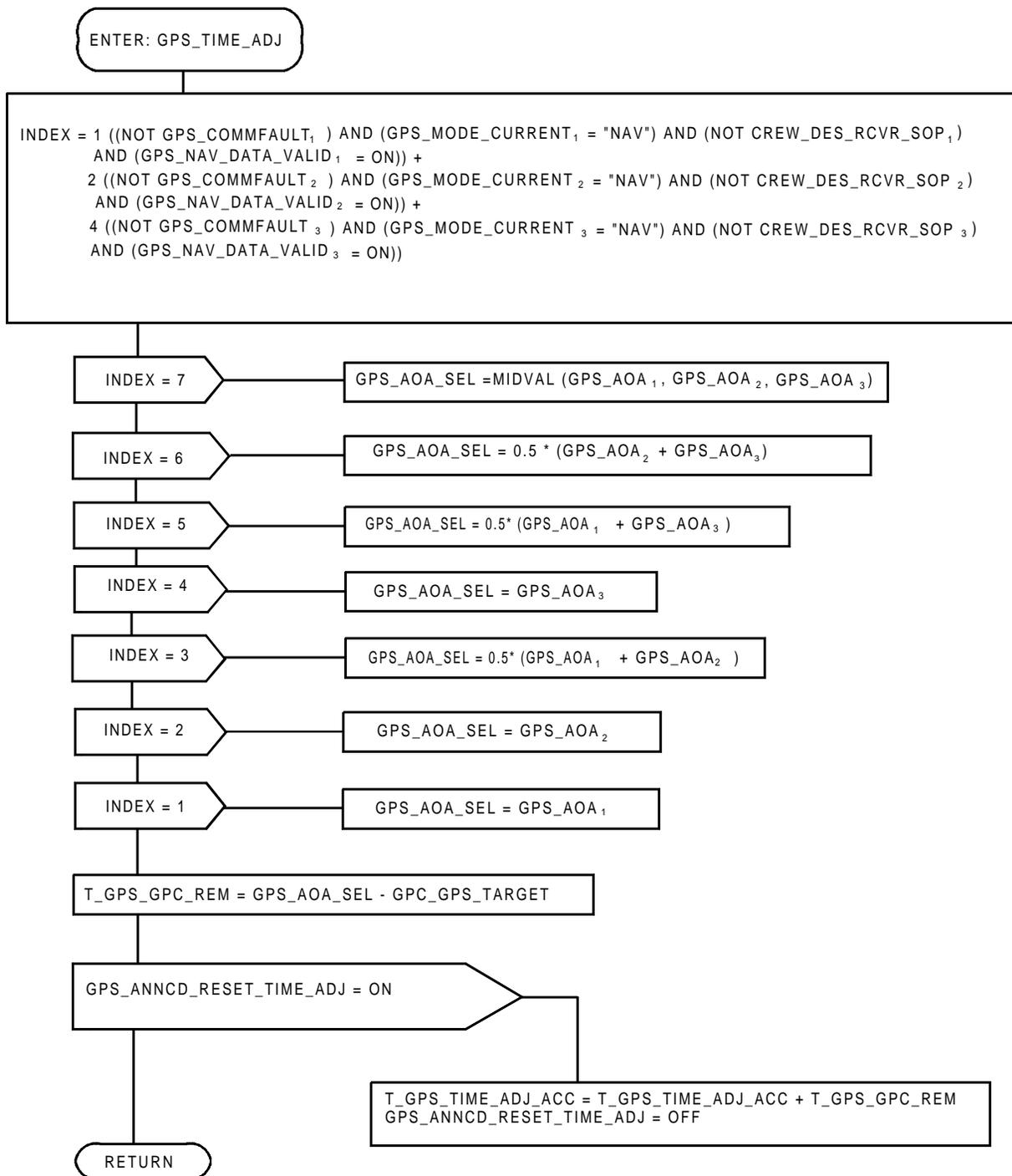


FIGURE 4.17.6.9 GPS_TIME_ADJ

4.17.6.10 GPS SOP (4.17) Principal Function Tables

This section contains the GPS SOP principal function input/output, I-load, K-load, constants and variable cross-reference tables which are presented in Tables 4.17.6.10-0 through 4.17.6.10-5 respectively.

Table 4.17.6.10-0. GPS SOP Principal Function Variable Cross-Reference	
FSSR Name	Source Name
BARO_DG	ALT_DG
CREW_DESELECT_SAT_RCVD	CREW_DESELECT_SAT
CREW_DES_RCVR_SOP	CREW_DESELECT_RCVR
FILT_RESTART_RCVD	GPS_FILT_RESTART_CMD
GPS_MODE_CMD_RCVD	GPS_MODE_CMD
GPS_RA_RCVD	GPS_RA_AID
GPS_TIME_ADJ_ENBL	GPS_TIME_ADJ_CMD
GPS_TO_GMT_ADJ	T_GPS_TO_GMT_ADJ
MACH_JUMP	ADS_MJ
OP_CODE	UP_OP_CODE
Q_BARO	HPC
PSI	PSI/PSI_TOPO
PHI	PHI/PHI_TOPO
T_BARO	T_HPC
T_IMU_GPS_SOP	TIME_V
T_RADALT	RA_TIME
THETA	THETA/THETA_TOPO
RADALT_DG	DG_RADALT
RADALT_HEIGHT	Q_RADALT
\bar{V} _IMU_GPS_SOP	\bar{S} EL_VEL_GPS_M50

TABLE 4.17.6.10-1. INPUT FUNCTIONAL PARAMETERS FOR GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	SOURCE	UNITS	DATA TYPE	P R E C	LAST CR	R E V
ADS_MJ	V95X3112X	ADS MACH JUMP FLAG	ADTA SOP		BD		93017G	*
AIDING_DATA_VALID(1)	V90X3266X	GPS 1 AIDING REF STATE VALID	GPS REF STATE PROC		BD		91051K	
AIDING_DATA_VALID(2)	V90X3267X	GPS 2 AIDING REF STATE VALID	GPS REF STATE PROC		BD		91051K	
AIDING_DATA_VALID(3)	V90X3268X	GPS 3 AIDING REF STATE VALID	GPS REF STATE PROC		BD		91051K	
ALT_DG	V95X3111X	PRESSURE ALTITUDE DATA GOOD FLAG	ADTA SOP		BD		93017G	*
ALT_WHEELS	V95H0165CA	ALT TO REAR WHEELS/NAV BASE(ENTRY)	RTLS UPP	FT			93017G	*
ALT_WHEELS	V95H0165CB	ALT TO REAR WHEELS/NAV BASE(ENTRY)	ENT UPP	FT	DPL		93017G	*
ALT_WHEELS	V95H0165CC	ALT TO REAR WHEELS/NAV BASE(ENTRY)	TAEM UPP	FT	DPL		93017G	*
ALT_WHEELS	V95H0165CD	ALT TO REAR WHEELS/NAV BASE(ENTRY)	A/L UPP	FT	DPL		93017G	*
BARO_AIF	V93J4473C	ADTA A/I/F EDIT CMD	HORIZ SIT DISP		HXS		93017G	*
CLOCK/CLOCKTIME	V91W5000C	CLOCK-COMPUTER(GMT)	FCOS	S			93160B	*
CREW_DESELECT_RCVR(1)	V93X1205XA	DESELECT GPS 1 RCVR CMD	GPS STATUS DISP		BD		92817C	
CREW_DESELECT_RCVR(1)	V93X1205XB	DESELECT GPS 1 RCVR CMD	HORIZ SIT DISP		BD		92817C	
CREW_DESELECT_RCVR(1)	V93X1205XC	DESELECT GPS 1 RCVR CMD	REL NAV DISP		BD		92817C	
CREW_DESELECT_RCVR(2)	V93X1206XA	DESELECT GPS 2 RCVR CMD	GPS STATUS DISP		BD		92817C	
CREW_DESELECT_RCVR(2)	V93X1206XB	DESELECT GPS 2 RCVR CMD	HORIZ SIT DISP		BD		92817C	
CREW_DESELECT_RCVR(2)	V93X1206XC	DESELECT GPS 2 RCVR CMD	REL NAV DISP		BD		92817C	
CREW_DESELECT_RCVR(3)	V93X1207XA	DESELECT GPS 3 RCVR CMD	GPS STATUS DISP		BD		92817C	
CREW_DESELECT_RCVR(3)	V93X1207XB	DESELECT GPS 3 RCVR CMD	HORIZ SIT DISP		BD		92817C	
CREW_DESELECT_RCVR(3)	V93X1207XC	DESELECT GPS 3 RCVR CMD	REL NAV DISP		BD		92817C	
CREW_DESELECT_SAT(1)	V94M6062P	CREW DESELECT SATELLITE WORD 1	GPS STATUS DISP		HMD		93017G	*
CREW_DESELECT_SAT(2)	V94M6063P	CREW DESELECT SATELLITE WORD 2	GPS STATUS DISP		HMD		93017G	*
CREW_DESELECT_SAT(3)	V94M6064P	CREW DESELECT SATELLITE WORD 3	GPS STATUS DISP		HMD		92067D	
CREW_DESELECT_SAT(4)	V94M6065P	CREW DESELECT SATELLITE WORD 4	GPS STATUS DISP		HMD		92067D	
DG_RADALT	V90X4003X	RALT HEIGHT DATA GOOD	SF		BD		93017G	*
EARTH_POLE(1)	V90U0190C	X-COMP UNIT VCTR ALONG EARTH AXIS	RNP MAT COMP	ND	DPL		93090E	
EARTH_POLE(2)	V90U0191C	Y-COMP UNIT VCTR ALONG EARTH AXIS	RNP MAT COMP	ND	DPL		93090E	
EARTH_POLE(3)	V90U0192C	Z-COMP UNIT VCTR ALONG EARTH AXIS	RNP MAT COMP	ND	DPL		93090E	
EULER_GPS_AVAIL	V90X2610XA	TOPODETTIC EULER ANGLES AVAIL FLAG	ORB ATT PROC		BD		92144C	
EULER_GPS_AVAIL	V90X2610XB	TOPODETTIC EULER ANGLES AVAIL FLAG	A/E ATT PROC		BD		92144C	
GPS_ANNCNCD_RESET_SET(1)	V90X3290XA	GPS 1 REF STATE ANNCNCD RESET	GPS REF STATE PROC		BD		92067D	
GPS_ANNCNCD_RESET_SET(2)	V90X3291XA	GPS 2 REF STATE ANNCNCD RESET	GPS REF STATE PROC		BD		92067D	
GPS_ANNCNCD_RESET_SET(3)	V90X3292XA	GPS 3 REF STATE ANNCNCD RESET	GPS REF STATE PROC		BD		92067D	
GPS_ANNCNCD_RESET_TIME_ADJ	V90X0437X	GPS ANNCNCD RESET FOR TIME ADJ	GPS REF STATE PROC		BD		92817C	
GPS_BYPASS(1)	V91X0026X	FF1 GPS1 BYPASS (GPS)	FCOS		BD		92289A	
GPS_BYPASS(2)	V91X0027X	FF2 GPS2 BYPASS (GPS)	FCOS		BD		92289A	
GPS_BYPASS(3)	V91X0028X	FF3 GPS3 BYPASS (GPS)	FCOS		BD		92289A	
GPS_CMD_RESET(1)	V90X0444X	GPS 1 COMMAND RESET FLAG	GPS REF STATE PROC		BD		92289A	
GPS_CMD_RESET(2)	V90X0445X	GPS 2 COMMAND RESET FLAG	GPS REF STATE PROC		BD		92289A	
GPS_CMD_RESET(3)	V90X0446X	GPS 3 COMMAND RESET FLAG	GPS REF STATE PROC		BD		92289A	
GPS_COMMFAULT(1)	V91X0015X	FF1 GPS1 COMMFAULT	FCOS		BD		92067D	
GPS_COMMFAULT(2)	V91X0016X	FF2 GPS2 COMMFAULT	FCOS		BD		92067D	
GPS_COMMFAULT(3)	V91X0017X	FF3 GPS3 COMMFAULT	FCOS		BD		92067D	
GPS_FILT_RESTART(1)	V90X3269XB	GPS1 FILTER RESTART FLAG	GPS REF STATE PROC		BD		92049C	
GPS_FILT_RESTART(2)	V90X3270XB	GPS2 FILTER RESTART FLAG	GPS REF STATE PROC		BD		92049C	
GPS_FILT_RESTART(3)	V90X3271XB	GPS3 FILTER RESTART FLAG	GPS REF STATE PROC		BD		92049C	

TABLE 4.17.6.10-1. INPUT FUNCTIONAL PARAMETERS FOR GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	SOURCE	UNITS	DATA TYPE	P R E C	LAST CR	R E V
GPS_FILT_RESTART_CMD(1)	V93X1220X	GPS 1 FILT RESTART CMD	GPS STATUS DISP		BD		91051K	
GPS_FILT_RESTART_CMD(2)	V93X1221X	GPS 2 FILT RESTART CMD	GPS STATUS DISP		BD		91051K	
GPS_FILT_RESTART_CMD(3)	V93X1222X	GPS 3 FILT RESTART CMD	GPS STATUS DISP		BD		91051K	
GPS_INIT_REQUEST_SOP(1)	V90X3018XA	GPS 1 REQUEST INITIAL STATE	GPS REF STATE PROC		BD		92067D	
GPS_INIT_REQUEST_SOP(2)	V90X3019XA	GPS 2 REQUEST INITIAL STATE	GPS REF STATE PROC		BD		92067D	
GPS_INIT_REQUEST_SOP(3)	V90X3020XA	GPS 3 REQUEST INITIAL STATE	GPS REF STATE PROC		BD		92067D	
GPS_IO_CNTL	V93J1231C	GPS I/O CONTROL	GPS STATUS DISP		HXS		92056D	
GPS_MODE_CMD(1)	V93U1223C	GPS 1 MODE CMD	GPS STATUS DISP		HXS		92069B	
GPS_MODE_CMD(2)	V93U1224C	GPS 2 MODE CMD	GPS STATUS DISP		HXS		92069B	
GPS_MODE_CMD(3)	V93U1225C	GPS 3 MODE CMD	GPS STATUS DISP		HXS		92069B	
GPS_RA_AID	V93X1204X	GPS RCVR USE RA DATA AIDING CMD	HORIZ SIT DISP		BD		91051K	
GPS_TIME_ADJ_CMD	V93X1111X	GPS TIME ADJUST CMD	GPS STATUS DISP	ND	BD		92817C	
GPS_TO_GPC_DATA_ONE(1,1)	V74U2318B	GPS1 INPUT HEADER	GPS1		BCH		93017G	*
GPS_TO_GPC_DATA_ONE(1,10)	V74U2327B	GPS 1 TO GPC DATA ONE HALFWORD 10	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,11)	V74U2328B	GPS 1 TO GPC DATA ONE HALFWORD 11	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,12)	V74U2329B	GPS 1 TO GPC DATA ONE HALFWORD 12	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,13)	V74U2330B	GPS 1 TO GPC DATA ONE HALFWORD 13	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,14)	V74U2331B	GPS 1 TO GPC DATA ONE HALFWORD 14	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,15)	V74U2332B	GPS 1 TO GPC DATA ONE HALFWORD 15	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,16)	V74U2333B	GPS 1 TO GPC DATA ONE HALFWORD 16	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,17)	V74U2334B	GPS 1 TO GPC DATA ONE HALFWORD 17	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,18)	V74U2335B	GPS 1 TO GPC DATA ONE HALFWORD 18	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,19)	V74U2336B	GPS 1 TO GPC DATA ONE HALFWORD 19	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,2)	V74M2319P	GPS1 NAV MODE WORD 1	GPS1		HMD		93017G	*
GPS_TO_GPC_DATA_ONE(1,20)	V74U2337B	GPS 1 TO GPC DATA ONE HALFWORD 20	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,21)	V74U2338B	GPS 1 TO GPC DATA ONE HALFWORD 21	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,22)	V74U2339B	GPS 1 TO GPC DATA ONE HALFWORD 22	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,23)	V74U2340B	GPS 1 TO GPC DATA ONE HALFWORD 23	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,24)	V74U2341B	GPS 1 TO GPC DATA ONE HALFWORD 24	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,25)	V74U2342B	GPS 1 TO GPC DATA ONE HALFWORD 25	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,26)	V74U2343B	GPS 1 TO GPC DATA ONE HALFWORD 26	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,27)	V74U2344B	GPS 1 TO GPC DATA ONE HALFWORD 27	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,28)	V74U2345B	GPS 1 TO GPC DATA ONE HALFWORD 28	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,29)	V74U2346B	GPS 1 TO GPC DATA ONE HALFWORD 29	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,3)	V74M2320P	GPS1 NAV MODE WORD 2	GPS1		HMD		93017G	*
GPS_TO_GPC_DATA_ONE(1,30)	V74U2347B	GPS 1 TO GPC DATA ONE HALFWORD 30	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,31)	V74U2348B	GPS 1 TO GPC DATA ONE HALFWORD 31	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,32)	V74U2349B	GPS 1 TO GPC DATA ONE HALFWORD 32	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,4)	V74U2321B	GPS 1 TO GPC DATA ONE HALFWORD 4	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,5)	V74U2322B	GPS 1 TO GPC DATA ONE HALFWORD 5	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,6)	V74U2323B	GPS 1 TO GPC DATA ONE HALFWORD 6	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,7)	V74U2324B	GPS 1 TO GPC DATA ONE HALFWORD 7	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,8)	V74U2325B	GPS 1 TO GPC DATA ONE HALFWORD 8	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(1,9)	V74U2326B	GPS 1 TO GPC DATA ONE HALFWORD 9	GPS1		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,1)	V74U1920B	GPS2 INPUT HEADER	GPS2		BCH		93017G	*
GPS_TO_GPC_DATA_ONE(2,10)	V74U2356B	GPS 2 TO GPC DATA ONE HALFWORD 10	GPS2		HMD		92069B	

TABLE 4.17.6.10-1. INPUT FUNCTIONAL PARAMETERS FOR GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	SOURCE	UNITS	DATA TYPE	P R E C	LAST CR	R E V
GPS_TO_GPC_DATA_ONE(2,11)	V74U2357B	GPS 2 TO GPC DATA ONE HALFWORD 11	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,12)	V74U2358B	GPS 2 TO GPC DATA ONE HALFWORD 12	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,13)	V74U2359B	GPS 2 TO GPC DATA ONE HALFWORD 13	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,14)	V74U2360B	GPS 2 TO GPC DATA ONE HALFWORD 14	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,15)	V74U2361B	GPS 2 TO GPC DATA ONE HALFWORD 15	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,16)	V74U2362B	GPS 2 TO GPC DATA ONE HALFWORD 16	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,17)	V74U2363B	GPS 2 TO GPC DATA ONE HALFWORD 17	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,18)	V74U2364B	GPS 2 TO GPC DATA ONE HALFWORD 18	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,19)	V74U2365B	GPS 2 TO GPC DATA ONE HALFWORD 19	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,2)	V74M1921P	GPS2 NAV MODE WORD 1	GPS2		HMD		93017G	*
GPS_TO_GPC_DATA_ONE(2,20)	V74U2366B	GPS 2 TO GPC DATA ONE HALFWORD 20	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,21)	V74U2367B	GPS 2 TO GPC DATA ONE HALFWORD 21	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,22)	V74U2368B	GPS 2 TO GPC DATA ONE HALFWORD 22	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,23)	V74U2369B	GPS 2 TO GPC DATA ONE HALFWORD 23	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,24)	V74U2370B	GPS 2 TO GPC DATA ONE HALFWORD 24	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,25)	V74U2371B	GPS 2 TO GPC DATA ONE HALFWORD 25	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,26)	V74U2372B	GPS 2 TO GPC DATA ONE HALFWORD 26	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,27)	V74U2373B	GPS 2 TO GPC DATA ONE HALFWORD 27	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,28)	V74U2374B	GPS 2 TO GPC DATA ONE HALFWORD 28	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,29)	V74U2375B	GPS 2 TO GPC DATA ONE HALFWORD 29	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,3)	V74M1934P	GPS2 NAV MODE WORD 2	GPS2		HMD		93017G	*
GPS_TO_GPC_DATA_ONE(2,30)	V74U2376B	GPS 2 TO GPC DATA ONE HALFWORD 30	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,31)	V74U2377B	GPS 2 TO GPC DATA ONE HALFWORD 31	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,32)	V74U2378B	GPS 2 TO GPC DATA ONE HALFWORD 32	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,4)	V74U2350B	GPS 2 TO GPC DATA ONE HALFWORD 4	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,5)	V74U2351B	GPS 2 TO GPC DATA ONE HALFWORD 5	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,6)	V74U2352B	GPS 2 TO GPC DATA ONE HALFWORD 6	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,7)	V74U2353B	GPS 2 TO GPC DATA ONE HALFWORD 7	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,8)	V74U2354B	GPS 2 TO GPC DATA ONE HALFWORD 8	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(2,9)	V74U2355B	GPS 2 TO GPC DATA ONE HALFWORD 9	GPS2		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,1)	V74U2379B	GPS3 INPUT HEADER	GPS3		BCH		93017G	*
GPS_TO_GPC_DATA_ONE(3,10)	V74U2388B	GPS 3 TO GPC DATA ONE HALFWORD 10	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,11)	V74U2389B	GPS 3 TO GPC DATA ONE HALFWORD 11	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,12)	V74U2390B	GPS 3 TO GPC DATA ONE HALFWORD 12	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,13)	V74U2391B	GPS 3 TO GPC DATA ONE HALFWORD 13	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,14)	V74U2392B	GPS 3 TO GPC DATA ONE HALFWORD 14	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,15)	V74U2393B	GPS 3 TO GPC DATA ONE HALFWORD 15	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,16)	V74U2394B	GPS 3 TO GPC DATA ONE HALFWORD 16	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,17)	V74U2395B	GPS 3 TO GPC DATA ONE HALFWORD 17	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,18)	V74U2396B	GPS 3 TO GPC DATA ONE HALFWORD 18	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,19)	V74U2397B	GPS 3 TO GPC DATA ONE HALFWORD 19	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,2)	V74M2380P	GPS3 NAV MODE WORD 1	GPS3		HMD		93017G	*
GPS_TO_GPC_DATA_ONE(3,20)	V74U2398B	GPS 3 TO GPC DATA ONE HALFWORD 20	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,21)	V74U2399B	GPS 3 TO GPC DATA ONE HALFWORD 21	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,22)	V74U2400B	GPS 3 TO GPC DATA ONE HALFWORD 22	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,23)	V74U2401B	GPS 3 TO GPC DATA ONE HALFWORD 23	GPS3		HMD		92069B	

TABLE 4.17.6.10-1. INPUT FUNCTIONAL PARAMETERS FOR GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	SOURCE	UNITS	DATA TYPE	P R E C	LAST CR	R E V
GPS_TO_GPC_DATA_ONE(3,24)	V74U2402B	GPS 3 TO GPC DATA ONE HALFWORD 24	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,25)	V74U2403B	GPS 3 TO GPC DATA ONE HALFWORD 25	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,26)	V74U2404B	GPS 3 TO GPC DATA ONE HALFWORD 26	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,27)	V74U2405B	GPS 3 TO GPC DATA ONE HALFWORD 27	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,28)	V74U2406B	GPS 3 TO GPC DATA ONE HALFWORD 28	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,29)	V74U2407B	GPS 3 TO GPC DATA ONE HALFWORD 29	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,3)	V74M2381P	GPS3 NAV MODE WORD 2	GPS3		HMD		93017G	*
GPS_TO_GPC_DATA_ONE(3,30)	V74U2408B	GPS 3 TO GPC DATA ONE HALFWORD 30	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,31)	V74U2409B	GPS 3 TO GPC DATA ONE HALFWORD 31	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,32)	V74U2410B	GPS 3 TO GPC DATA ONE HALFWORD 32	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,4)	V74U2382B	GPS 3 TO GPC DATA ONE HALFWORD 4	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,5)	V74U2383B	GPS 3 TO GPC DATA ONE HALFWORD 5	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,6)	V74U2384B	GPS 3 TO GPC DATA ONE HALFWORD 6	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,7)	V74U2385B	GPS 3 TO GPC DATA ONE HALFWORD 7	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,8)	V74U2386B	GPS 3 TO GPC DATA ONE HALFWORD 8	GPS3		HMD		92069B	
GPS_TO_GPC_DATA_ONE(3,9)	V74U2387B	GPS 3 TO GPC DATA ONE HALFWORD 9	GPS3		HMD		92069B	
GPS_TSW(1)	V91U0032C	GPS 1 TRANSACTION STATUS WORD	FCOS		HMD		91051K	
GPS_TSW(2)	V91U0033C	GPS 2 TRANSACTION STATUS WORD	FCOS		HMD		91051K	
GPS_TSW(3)	V91U0034C	GPS 3 TRANSACTION STATUS WORD	FCOS		HMD		91051K	
HPC	V95H3001C	ADS CORRECTED PRESSURE ALTITUDE	ADTA SOP	FT	SPL		93017G	*
H_DOT_ELLIPSOID/H_DOT	V95L0265CA	EST ALTITUDE RATE	RTLS UPP	FT/S	SPL		93017G	*
H_DOT_ELLIPSOID/H_DOT	V95L0265CB	EST ALTITUDE RATE	TAEM UPP	FT/S	SPL		93017G	*
H_DOT_ELLIPSOID/H_DOT	V95L0265CC	EST ALTITUDE RATE	A/L UPP	FT/S	SPL		93017G	*
H_DOT_ELLIPSOID/H_DOT	V95L0265CD	EST ALTITUDE RATE	ENT UPP	FT/S	SPL		93017G	*
INIT_STATE_SET(1)	V90X3278XA	GPS 1 REFERENCE STATE INITIALIZED	GPS REF STATE PROC		BD		92067D	
INIT_STATE_SET(2)	V90X3279XA	GPS 2 REFERENCE STATE INITIALIZED	GPS REF STATE PROC		BD		92067D	
INIT_STATE_SET(3)	V90X3280XA	GPS 3 REFERENCE STATE INITIALIZED	GPS REF STATE PROC		BD		92067D	
MM_CODE	V90Q8001C	MAJOR MODE CODE	UI		HXS		93017G	*
M_M50TOEF_AT_EPOCH(1,1)	V97U5738C	M50 TO EF TRANS MATRIX ELEM 1,1	RNP MAT COMP	ND	DPL		93090E	
M_M50TOEF_AT_EPOCH(1,2)	V97U5739C	M50 TO EF TRANS MATRIX ELEM 1,2	RNP MAT COMP	ND	DPL		93090E	
M_M50TOEF_AT_EPOCH(1,3)	V97U5740C	M50 TO EF TRANS MATRIX ELEM 1,3	RNP MAT COMP	ND	DPL		93090E	
M_M50TOEF_AT_EPOCH(2,1)	V97U5741C	M50 TO EF TRANS MATRIX ELEM 2,1	RNP MAT COMP	ND	DPL		93090E	
M_M50TOEF_AT_EPOCH(2,2)	V97U5742C	M50 TO EF TRANS MATRIX ELEM 2,2	RNP MAT COMP	ND	DPL		93090E	
M_M50TOEF_AT_EPOCH(2,3)	V97U5743C	M50 TO EF TRANS MATRIX ELEM 2,3	RNP MAT COMP	ND	DPL		93090E	
M_M50TOEF_AT_EPOCH(3,1)	V97U5744C	M50 TO EF TRANS MATRIX ELEM 3,1	RNP MAT COMP	ND	DPL		93090E	
M_M50TOEF_AT_EPOCH(3,2)	V97U5745C	M50 TO EF TRANS MATRIX ELEM 3,2	RNP MAT COMP	ND	DPL		93090E	
M_M50TOEF_AT_EPOCH(3,3)	V97U5746C	M50 TO EF TRANS MATRIX ELEM 3,3	RNP MAT COMP	ND	DPL		93090E	
PHI/PHI_TOPO	V90H2202C	BODY ROLL ATTITUDE EULER ANGLE	A/E ATT PROC	DEG	SPL		93017G	*
PHI/PHI_TOPO	V90H2202CA	BODY ROLL ATTITUDE EULER ANGLE	ORB ATT PROC	DEG	SPL		92144C	
PHI_OUT	V94H6066C	BODY ROLL EULER ANGLE AT SEL SITE	GROUND IMU DISP	DEG	SPL		92067D	
PSI/PSI_TOPO	V90H2230C	BODY YAW ATTITUDE EULER ANGLE	A/E ATT PROC	DEG	SPL		93017G	*
PSI/PSI_TOPO	V90H2230CA	BODY YAW ATTITUDE EULER ANGLE	ORB ATT PROC	DEG	SPL		92144C	
PSI_OUT	V94H6067C	BODY YAW EULER ANGLE AT SEL SITE	GROUND IMU DISP	DEG	SPL		92067D	
Q_RADALT	V90H4002C	SELECTED RALT HEIGHT-RAW	SF	FT	SPL		93017G	*
RA_TIME	V95W7000C	RADAR ALTIMETER TIME	RA SOP	S	DPL		93017G	*
REFS_RDY	V95X0780XA	REFSMAT COMPUTATION COMPLETED	IMU INT PROC		BD		93017G	*

TABLE 4.17.6.10-1. INPUT FUNCTIONAL PARAMETERS FOR GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	SOURCE	UNITS	DATA TYPE	P R E C	LAST CR	R E V
REL_VEL_MAG	V95L0151CA	GND REL VEL MAGNITUDE IN M50 SYS	RTLS UPP	FT/S	SPL		93017G	*
REL_VEL_MAG/V	V95L0151CB	GND REL VEL MAGNITUDE IN M50 SYS	A/L UPP	FT/S	SPL		93017G	*
REL_VEL_MAG/V	V95L0151CC	GND REL VEL MAGNITUDE IN M50 SYS	ENT UPP	FT/S	SPL		93017G	*
REL_VEL_MAG/V	V95L0151CD	GND REL VEL MAGNITUDE IN M50 SYS	TAEM UPP	FT/S	SPL		93017G	*
R_REF(1)	V90H3281C	X-COMP OF GPS REF STATE POS VEC	GPS REF STATE PROC	FT	DPL		92369B	
R_REF(2)	V90H3282C	Y-COMP OF GPS REF STATE POS VEC	GPS REF STATE PROC	FT	DPL		92369B	
R_REF(3)	V90H3283C	Z-COMP OF GPS REF STATE POS VEC	GPS REF STATE PROC	FT	DPL		92369B	
R_REF_OUT(1,1)	V90H3662C	X-COMP OF GPS 1 REF STATE POS VEC	GROUND IMU DISP	FT	DPL		92067D	
R_REF_OUT(1,2)	V90H3663C	Y-COMP OF GPS 1 REF STATE POS VEC	GROUND IMU DISP	FT	DPL		92067D	
R_REF_OUT(1,3)	V90H3664C	Z-COMP OF GPS 1 REF STATE POS VEC	GROUND IMU DISP	FT	DPL		92067D	
R_REF_OUT(2,1)	V90H3665C	X-COMP OF GPS 2 REF STATE POS VEC	GROUND IMU DISP	FT	DPL		92067D	
R_REF_OUT(2,2)	V90H3666C	Y-COMP OF GPS 2 REF STATE POS VEC	GROUND IMU DISP	FT	DPL		92067D	
R_REF_OUT(2,3)	V90H3667C	Z-COMP OF GPS 2 REF STATE POS VEC	GROUND IMU DISP	FT	DPL		92067D	
R_REF_OUT(3,1)	V90H3668C	X-COMP OF GPS 3 REF STATE POS VEC	GROUND IMU DISP	FT	DPL		92067D	
R_REF_OUT(3,2)	V90H3669C	Y-COMP OF GPS 3 REF STATE POS VEC	GROUND IMU DISP	FT	DPL		92067D	
R_REF_OUT(3,3)	V90H3670C	Z-COMP OF GPS 3 REF STATE POS VEC	GROUND IMU DISP	FT	DPL		92067D	
SEL_SITE_ID	V90U0378CA	SELECTED LANDING SITE ID	ASC SITE LOOKUP				92522D	
SEL_SITE_ID	V90U0378CB	SELECTED LANDING SITE ID	ENT SITE LOOKUP				92522D	
SEL_TEST_ID	V91U5085CA	CURRENT SITE SELECT	GROUND IMU DISP		VAR		91051K	
SEL_VEL_GPS_M50(1)	V90L2631C	SEL TOTAL M50 X VEL FOR GPS	IMU RM	FT/S	DPL		92369B	
SEL_VEL_GPS_M50(2)	V90L2632C	SEL TOTAL M50 Y VEL FOR GPS	IMU RM	FT/S	DPL		92369B	
SEL_VEL_GPS_M50(3)	V90L2633C	SEL TOTAL M50 Z VEL FOR GPS	IMU RM	FT/S	DPL		92369B	
THETA/THETA_TOPO	V90H2217C	BODY PITCH ATTITUDE EULER ANGLE	A/E ATT PROC	DEG	SPL		93017G	*
THETA/THETA_TOPO	V90H2217CA	BODY PITCH ATTITUDE EULER ANGLE	ORB ATT PROC	DEG	SPL		92144C	
THETA_OUT	V94H6068C	BODY PITCH EULER ANGLE AT SEL SITE	GROUND IMU DISP	DEG	SPL		92067D	
TIME_V	V90W2553CA	TIME OF SEL TOTAL M50 VEL	IMU RM	S			91051K	
T_CURRENT_GPS_REF	V90W3581C	TIME TAG FOR GPS REFERENCE STATES	GPS REF STATE PROC	S	DPL		92035D	
T_EPOCH	V97W5528C	TIME TAG FOR M_M50TOEF_AT_EPOCH	RNP MAT COMP	S	DPL		93090E	
T_GPS_OFFSET_SECS	V91W0029CA	GPS OFFSET SECONDS	D/L NAV SEQ	S	DPL		93160B	*
T_GPS_TO_GMT_ADJ	V91W0035C	GPS TO GMT TIME ADJUSTMENT	UL	S	DPL		92364D	
T_HPC	V95W3400C	ADS CORRECTED PRESS ALT TIME TAG	ADTA SOP	S	DPL		93017G	*
UP_OP_CODE	V96J3521C	OP CODE FOR UPLINK	UL		HXS		92067D	
V_REF(1)	V90L3308C	X-COMP OF GPS REF STATE VEL VEC	GPS REF STATE PROC	FT/S	DPL		92369B	
V_REF(2)	V90L3309C	Y-COMP OF GPS REF STATE VEL VEC	GPS REF STATE PROC	FT/S	DPL		92369B	
V_REF(3)	V90L3310C	Z-COMP OF GPS REF STATE VEL VEC	GPS REF STATE PROC	FT/S	DPL		92369B	

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
ALMANAC_REF_WEEK(1)	V74W4460B	GPS1 ALMANAC WEEK	DL FMT 44/1 DL FMT 45/1		HXU		93017G	*
ALMANAC_REF_WEEK(2)	V74W1460B	GPS2 ALMANAC WEEK	DL FMT 60/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1		HXU		93017G	*
ALMANAC_REF_WEEK(3)	V74W3460B	GPS3 ALMANAC WEEK	DL FMT 60/1 DL FMT 44/1 DL FMT 45/1		HXU		93017G	*
BIT_FAULT_LOG_WORD(1,1)	V74M4942P	GPS1 LOG WD1	DL FMT 60/1 DL FMT 22/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(1,10)	V74M4951P	GPS1 LOG WD10	DL FMT 60/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(1,11)	V74M4952P	GPS1 LOG WD11	DL FMT 60/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(1,12)	V74M4953P	GPS1 LOG WD12	DL FMT 60/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(1,13)	V74M4954P	GPS1 LOG WD13	DL FMT 60/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(1,14)	V74M4955P	GPS1 LOG WD14	DL FMT 60/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(1,15)	V74M4956P	GPS1 LOG WD15	DL FMT 60/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(1,16)	V74M4957P	GPS1 LOG WD16	DL FMT 60/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1		HMD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
BIT_FAULT_LOG_WORD(1,17)	V74M4958P	GPS1 LOG WD17	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(1,18)	V74M3098P	GPS1 LOG WD18	TLM		HMD		92067D	
BIT_FAULT_LOG_WORD(1,2)	V74M4943P	GPS1 LOG WD2	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(1,3)	V74M4944P	GPS1 LOG WD3	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(1,4)	V74M4945P	GPS1 LOG WD4	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(1,5)	V74M4946P	GPS1 LOG WD5	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(1,6)	V74M4947P	GPS1 LOG WD6	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(1,7)	V74M4948P	GPS1 LOG WD7	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(1,8)	V74M4949P	GPS1 LOG WD8	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(1,9)	V74M4950P	GPS1 LOG WD9	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(2,1)	V74M1942P	GPS2 LOG WD1	DL FMT 22/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
BIT_FAULT_LOG_WORD(2,10)	V74M1951P	GPS2 LOG WD10	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(2,11)	V74M1952P	GPS2 LOG WD11	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(2,12)	V74M1953P	GPS2 LOG WD12	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(2,13)	V74M1954P	GPS2 LOG WD13	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(2,14)	V74M1955P	GPS2 LOG WD14	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(2,15)	V74M1956P	GPS2 LOG WD15	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(2,16)	V74M1957P	GPS2 LOG WD16	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
BIT_FAULT_LOG_WORD(2,17)	V74M1958P	GPS2 LOG WD17	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(2,18)	V74M1397P	GPS2 LOG WD18	TLM		HMD		91051K	
BIT_FAULT_LOG_WORD(2,2)	V74M1943P	GPS2 LOG WD2	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(2,3)	V74M1944P	GPS2 LOG WD3	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(2,4)	V74M1945P	GPS2 LOG WD4	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(2,5)	V74M1946P	GPS2 LOG WD5	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(2,6)	V74M1947P	GPS2 LOG WD6	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(2,7)	V74M1948P	GPS2 LOG WD7	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
BIT_FAULT_LOG_WORD(2,8)	V74M1949P	GPS2 LOG WD8	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(2,9)	V74M1950P	GPS2 LOG WD9	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(3,1)	V74M5942P	GPS3 LOG WD1	DL FMT 22/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(3,10)	V74M5951P	GPS3 LOG WD10	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(3,11)	V74M5952P	GPS3 LOG WD11	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(3,12)	V74M5953P	GPS3 LOG WD12	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(3,13)	V74M5954P	GPS3 LOG WD13	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(3,14)	V74M5955P	GPS3 LOG WD14	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(3,15)	V74M5956P	GPS3 LOG WD15	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(3,16)	V74M5957P	GPS3 LOG WD16	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
BIT_FAULT_LOG_WORD(3,17)	V74M5958P	GPS3 LOG WD17	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(3,18)	V74M3108P	GPS3 LOG WD18	TLM		HMD		92067D	
BIT_FAULT_LOG_WORD(3,2)	V74M5943P	GPS3 LOG WD2	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(3,3)	V74M5944P	GPS3 LOG WD3	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(3,4)	V74M5945P	GPS3 LOG WD4	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(3,5)	V74M5946P	GPS3 LOG WD5	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(3,6)	V74M5947P	GPS3 LOG WD6	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(3,7)	V74M5948P	GPS3 LOG WD7	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(3,8)	V74M5949P	GPS3 LOG WD8	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
BIT_FAULT_LOG_WORD(3,9)	V74M5950P	GPS3 LOG WD9	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
CHANNELS_STATUS(1,1)	V74M4960P	GPS1 CHAN STATUS WD 1	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
CHANNELS_STATUS(1,2)	V74M4961P	GPS1 CHAN STATUS WD 2	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1		HMD		93017G	*
CHANNELS_STATUS(1,3)	V74M4962P	GPS1 CHAN STATUS WD 3	DL FMT 60/1 DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
CHANNELS_STATUS(1,4)	V74M4963P	GPS1 CHAN STATUS WD 4	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
CHANNELS_STATUS(1,5)	V74M4964P	GPS1 CHAN STATUS WD 5	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
CHANNELS_STATUS(1,6)	V74M4965P	GPS1 CHAN STATUS WD 6	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
CHANNELS_STATUS(1,7)	V74M4966P	GPS1 CHAN STATUS WD 7	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
CHANNELS_STATUS(1,8)	V74M4967P	GPS1 CHAN STATUS WD 8	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
CHANNELS_STATUS(2,1)	V74M1960P	GPS2 CHAN STATUS WD 1	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
CHANNELS_STATUS(2,2)	V74M1961P	GPS2 CHAN STATUS WD 2	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
CHANNELS_STATUS(2,3)	V74M1962P	GPS2 CHAN STATUS WD 3	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
CHANNELS_STATUS(2,4)	V74M1963P	GPS2 CHAN STATUS WD 4	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
CHANNELS_STATUS(2,5)	V74M1964P	GPS2 CHAN STATUS WD 5	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
CHANNELS_STATUS(2,6)	V74M1965P	GPS2 CHAN STATUS WD 6	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
CHANNELS_STATUS(2,7)	V74M1966P	GPS2 CHAN STATUS WD 7	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
CHANNELS_STATUS(2,8)	V74M1967P	GPS2 CHAN STATUS WD 8	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
CHANNELS_STATUS(3,1)	V74M5960P	GPS3 CHAN STATUS WD 1	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
CHANNELS_STATUS(3,2)	V74M5961P	GPS3 CHAN STATUS WD 2	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
CHANNELS_STATUS(3,3)	V74M5962P	GPS3 CHAN STATUS WD 3	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
CHANNELS_STATUS(3,4)	V74M5963P	GPS3 CHAN STATUS WD 4	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
CHANNELS_STATUS(3,5)	V74M5964P	GPS3 CHAN STATUS WD 5	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
CHANNELS_STATUS(3,6)	V74M5965P	GPS3 CHAN STATUS WD 6	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
CHANNELS_STATUS(3,7)	V74M5966P	GPS3 CHAN STATUS WD 7	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
CHANNELS_STATUS(3,8)	V74M5967P	GPS3 CHAN STATUS WD 8	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
FILT_RESTART_REQUEST(1)	V90X3006X	GPS 1 FILTER RESTART REQUEST FLAG	GPS REF STATE PROC		BD		91051K	
FILT_RESTART_REQUEST(2)	V90X3007X	GPS 2 FILTER RESTART REQUEST FLAG	GPS REF STATE PROC		BD		91051K	
FILT_RESTART_REQUEST(3)	V90X3008X	GPS 3 FILTER RESTART REQUEST FLAG	GPS REF STATE PROC		BD		91051K	
FOM_GDOP_STATIC(1)	V94X6142X	GPS 1 FOM/GDOP STATIC FLAG	GPS STATUS DISP HORIZ SIT DISP		BD		92369B	
FOM_GDOP_STATIC(2)	V94X6143X	GPS 2 FOM/GDOP STATIC FLAG	GPS STATUS DISP HORIZ SIT DISP		BD		92369B	
FOM_GDOP_STATIC(3)	V94X6144X	GPS 3 FOM/GDOP STATIC FLAG	GPS STATUS DISP HORIZ SIT DISP		BD		92369B	
FOM_NAV(1)	V94H1082CB	GPS 1 NAV FOM	GPS STATUS DISP HORIZ SIT DISP		HXS		92067D	
FOM_NAV(2)	V94H1083CB	GPS 2 NAV FOM	GPS STATUS DISP HORIZ SIT DISP		HXS		92067D	
FOM_NAV(3)	V94H1084CB	GPS 3 NAV FOM	GPS STATUS DISP HORIZ SIT DISP		HXS		92067D	
FOM_SOP(1)	V90H3009C	GPS 1 FIGURE OF MERIT	ENT NAV ORB/RND NAV PRELAND NAV RTLS NAV		HXS		92067D	
FOM_SOP(2)	V90H3010C	GPS 2 FIGURE OF MERIT	ENT NAV ORB/RND NAV PRELAND NAV RTLS NAV		HXS		92067D	
FOM_SOP(3)	V90H3011C	GPS 3 FIGURE OF MERIT	ENT NAV ORB/RND NAV PRELAND NAV RTLS NAV		HXS		92067D	
GDOP_NEW(1)	V94H0921C	GPS 1 CURRENT GDOP	GPS STATUS DISP		HXS		91051K	
GDOP_NEW(2)	V94H0922C	GPS 2 CURRENT GDOP	GPS STATUS DISP		HXS		91051K	
GDOP_NEW(3)	V94H0923B	GPS 3 CURRENT GDOP	GPS STATUS DISP		HXS		91051K	
GPC_TO_GPS_DATA_ONE(1,1)	V93U1240B	GPS1 OUTPUT HEADER	GPS1		BCH		92067D	
GPC_TO_GPS_DATA_ONE(1,10)	V93U1249C	GPC TO GPS 1 DATA ONE HALFWORD 10	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,11)	V93U1250C	GPC TO GPS 1 DATA ONE HALFWORD 11	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,12)	V93U1251C	GPC TO GPS 1 DATA ONE HALFWORD 12	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,13)	V93U1252C	GPC TO GPS 1 DATA ONE HALFWORD 13	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,14)	V93U1253C	GPC TO GPS 1 DATA ONE HALFWORD 14	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,15)	V93U1254C	GPC TO GPS 1 DATA ONE HALFWORD 15	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,16)	V93U1255C	GPC TO GPS 1 DATA ONE HALFWORD 16	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,17)	V93U1256C	GPC TO GPS 1 DATA ONE HALFWORD 17	GPS1		HMD		91051K	

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
GPC_TO_GPS_DATA_ONE(1,18)	V93U1257C	GPC TO GPS 1 DATA ONE HALFWORD 18	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,19)	V93U1258C	GPC TO GPS 1 DATA ONE HALFWORD 19	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,2)	V93M1241P	GPS1 CONTROL WORD 1	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1 GPS1		HMD		93017G	*
GPC_TO_GPS_DATA_ONE(1,20)	V93U1259C	GPC TO GPS 1 DATA ONE HALFWORD 20	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,21)	V93U1260C	GPC TO GPS 1 DATA ONE HALFWORD 21	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,22)	V93U1261C	GPC TO GPS 1 DATA ONE HALFWORD 22	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,23)	V93U1262C	GPC TO GPS 1 DATA ONE HALFWORD 23	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,24)	V93U1263C	GPC TO GPS 1 DATA ONE HALFWORD 24	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,25)	V93U1264C	GPC TO GPS 1 DATA ONE HALFWORD 25	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,26)	V93U1265C	GPC TO GPS 1 DATA ONE HALFWORD 26	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,27)	V93U1266C	GPC TO GPS 1 DATA ONE HALFWORD 27	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,28)	V93U1267C	GPC TO GPS 1 DATA ONE HALFWORD 28	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,29)	V93U1268C	GPC TO GPS 1 DATA ONE HALFWORD 29	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,3)	V93M1242P	GPS1 CONTROL WORD 2	GPS1 TLM		HMD		92067D	
GPC_TO_GPS_DATA_ONE(1,30)	V93U1269C	GPC TO GPS 1 DATA ONE HALFWORD 30	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,31)	V93U1270C	GPC TO GPS 1 DATA ONE HALFWORD 31	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,32)	V93U1271C	GPC TO GPS 1 DATA ONE HALFWORD 32	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,4)	V93U1243C	GPC TO GPS 1 DATA ONE HALFWORD 4	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,5)	V93U1244C	GPC TO GPS 1 DATA ONE HALFWORD 5	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,6)	V93U1245C	GPC TO GPS 1 DATA ONE HALFWORD 6	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,7)	V93U1246C	GPC TO GPS 1 DATA ONE HALFWORD 7	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,8)	V93U1247C	GPC TO GPS 1 DATA ONE HALFWORD 8	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(1,9)	V93U1248C	GPC TO GPS 1 DATA ONE HALFWORD 9	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,1)	V93U1186B	GPS2 OUTPUT HEADER	GPS2		BCH		92067D	
GPC_TO_GPS_DATA_ONE(2,10)	V93U1278C	GPC TO GPS 2 DATA ONE HALFWORD 10	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,11)	V93U1279C	GPC TO GPS 2 DATA ONE HALFWORD 11	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,12)	V93U1280C	GPC TO GPS 2 DATA ONE HALFWORD 12	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,13)	V93U1281C	GPC TO GPS 2 DATA ONE HALFWORD 13	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,14)	V93U1282C	GPC TO GPS 2 DATA ONE HALFWORD 14	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,15)	V93U1283C	GPC TO GPS 2 DATA ONE HALFWORD 15	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,16)	V93U1284C	GPC TO GPS 2 DATA ONE HALFWORD 16	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,17)	V93U1285C	GPC TO GPS 2 DATA ONE HALFWORD 17	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,18)	V93U1286C	GPC TO GPS 2 DATA ONE HALFWORD 18	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,19)	V93U1287C	GPC TO GPS 2 DATA ONE HALFWORD 19	GPS2		HMD		91051K	

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
GPC_TO_GPS_DATA_ONE(2,2)	V93M0556P	GPS2 CONTROL WORD 1	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1 GPS2		HMD		93017G	*
GPC_TO_GPS_DATA_ONE(2,20)	V93U1288C	GPC TO GPS 2 DATA ONE HALFWORD 20	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,21)	V93U1289C	GPC TO GPS 2 DATA ONE HALFWORD 21	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,22)	V93U1290C	GPC TO GPS 2 DATA ONE HALFWORD 22	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,23)	V93U1291C	GPC TO GPS 2 DATA ONE HALFWORD 23	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,24)	V93U1292C	GPC TO GPS 2 DATA ONE HALFWORD 24	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,25)	V93U1293C	GPC TO GPS 2 DATA ONE HALFWORD 25	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,26)	V93U1294C	GPC TO GPS 2 DATA ONE HALFWORD 26	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,27)	V93U1295C	GPC TO GPS 2 DATA ONE HALFWORD 27	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,28)	V93U1296C	GPC TO GPS 2 DATA ONE HALFWORD 28	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,29)	V93U1297C	GPC TO GPS 2 DATA ONE HALFWORD 29	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,3)	V93M0557P	GPS2 CONTROL WORD 2	GPS2 TLM		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,30)	V93U1298C	GPC TO GPS 2 DATA ONE HALFWORD 30	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,31)	V93U1299C	GPC TO GPS 2 DATA ONE HALFWORD 31	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,32)	V93U1300C	GPC TO GPS 2 DATA ONE HALFWORD 32	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,4)	V93U1272C	GPC TO GPS 2 DATA ONE HALFWORD 4	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,5)	V93U1273C	GPC TO GPS 2 DATA ONE HALFWORD 5	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,6)	V93U1274C	GPC TO GPS 2 DATA ONE HALFWORD 6	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,7)	V93U1275C	GPC TO GPS 2 DATA ONE HALFWORD 7	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,8)	V93U1276C	GPC TO GPS 2 DATA ONE HALFWORD 8	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(2,9)	V93U1277C	GPC TO GPS 2 DATA ONE HALFWORD 9	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,1)	V93U1305B	GPS3 OUTPUT HEADER	GPS3		BCH		92067D	
GPC_TO_GPS_DATA_ONE(3,10)	V93U1314C	GPC TO GPS 3 DATA ONE HALFWORD 10	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,11)	V93U1315C	GPC TO GPS 3 DATA ONE HALFWORD 11	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,12)	V93U1316C	GPC TO GPS 3 DATA ONE HALFWORD 12	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,13)	V93U1317C	GPC TO GPS 3 DATA ONE HALFWORD 13	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,14)	V93U1318C	GPC TO GPS 3 DATA ONE HALFWORD 14	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,15)	V93U1319C	GPC TO GPS 3 DATA ONE HALFWORD 15	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,16)	V93U1320C	GPC TO GPS 3 DATA ONE HALFWORD 16	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,17)	V93U1321C	GPC TO GPS 3 DATA ONE HALFWORD 17	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,18)	V93U1322C	GPC TO GPS 3 DATA ONE HALFWORD 18	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,19)	V93U1323C	GPC TO GPS 3 DATA ONE HALFWORD 19	GPS3		HMD		91051K	

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
GPC_TO_GPS_DATA_ONE(3,2)	V93M1306P	GPS3 CONTROL WORD 1	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1 GPS3		HMD		93017G	*
GPC_TO_GPS_DATA_ONE(3,20)	V93U1324C	GPC TO GPS 3 DATA ONE HALFWORD 20	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,21)	V93U1325C	GPC TO GPS 3 DATA ONE HALFWORD 21	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,22)	V93U1326C	GPC TO GPS 3 DATA ONE HALFWORD 22	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,23)	V93U1327C	GPC TO GPS 3 DATA ONE HALFWORD 23	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,24)	V93U1328C	GPC TO GPS 3 DATA ONE HALFWORD 24	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,25)	V93U1329C	GPC TO GPS 3 DATA ONE HALFWORD 25	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,26)	V93U1330C	GPC TO GPS 3 DATA ONE HALFWORD 26	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,27)	V93U1331C	GPC TO GPS 3 DATA ONE HALFWORD 27	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,28)	V93U1332C	GPC TO GPS 3 DATA ONE HALFWORD 28	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,29)	V93U1333C	GPC TO GPS 3 DATA ONE HALFWORD 29	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,3)	V93M1307P	GPS3 CONTROL WORD 2	GPS3 TLM		HMD		92067D	
GPC_TO_GPS_DATA_ONE(3,30)	V93U1334C	GPC TO GPS 3 DATA ONE HALFWORD 30	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,31)	V93U1335C	GPC TO GPS 3 DATA ONE HALFWORD 31	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,32)	V93U1336C	GPC TO GPS 3 DATA ONE HALFWORD 32	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,4)	V93U1308C	GPC TO GPS 3 DATA ONE HALFWORD 4	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,5)	V93U1309C	GPC TO GPS 3 DATA ONE HALFWORD 5	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,6)	V93U1310C	GPC TO GPS 3 DATA ONE HALFWORD 6	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,7)	V93U1311C	GPC TO GPS 3 DATA ONE HALFWORD 7	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,8)	V93U1312C	GPC TO GPS 3 DATA ONE HALFWORD 8	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_ONE(3,9)	V93U1313C	GPC TO GPS 3 DATA ONE HALFWORD 9	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,1)	V93U1337B	GPS1 DATA TWO OUTPUT HEADER	GPS1 TLM	ND	BCH		92067D	
GPC_TO_GPS_DATA_TWO(1,10)	V93U1346C	GPC TO GPS 1 DATA TWO HALFWORD 10	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,11)	V93U1347C	GPC TO GPS 1 DATA TWO HALFWORD 11	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,12)	V93U1348C	GPC TO GPS 1 DATA TWO HALFWORD 12	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,13)	V93U1349C	GPC TO GPS 1 DATA TWO HALFWORD 13	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,14)	V93U1350C	GPC TO GPS 1 DATA TWO HALFWORD 14	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,15)	V93U1351C	GPC TO GPS 1 DATA TWO HALFWORD 15	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,16)	V93U1352C	GPC TO GPS 1 DATA TWO HALFWORD 16	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,17)	V93U1353C	GPC TO GPS 1 DATA TWO HALFWORD 17	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,18)	V93U1354C	GPC TO GPS 1 DATA TWO HALFWORD 18	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,19)	V93U1355C	GPC TO GPS 1 DATA TWO HALFWORD 19	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,2)	V93U1338C	GPC TO GPS 1 DATA TWO HALFWORD 2	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,20)	V93U1356C	GPC TO GPS 1 DATA TWO HALFWORD 20	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,21)	V93U1357C	GPC TO GPS 1 DATA TWO HALFWORD 21	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,22)	V93U1358C	GPC TO GPS 1 DATA TWO HALFWORD 22	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,23)	V93U1359C	GPC TO GPS 1 DATA TWO HALFWORD 23	GPS1		HMD		91051K	

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
GPC_TO_GPS_DATA_TWO(1,24)	V93U1360C	GPC TO GPS 1 DATA TWO HALFWORD 24	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,25)	V93U1361C	GPC TO GPS 1 DATA TWO HALFWORD 25	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,26)	V93U1362C	GPC TO GPS 1 DATA TWO HALFWORD 26	GPS1		HMS		91051K	
GPC_TO_GPS_DATA_TWO(1,27)	V93U1363C	GPC TO GPS 1 DATA TWO HALFWORD 27	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,28)	V93U1364C	GPC TO GPS 1 DATA TWO HALFWORD 28	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,29)	V93U1365C	GPC TO GPS 1 DATA TWO HALFWORD 29	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,3)	V93U1339C	GPC TO GPS 1 DATA TWO HALFWORD 3	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,30)	V93U1366C	GPC TO GPS 1 DATA TWO HALFWORD 30	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,31)	V93U1367C	GPC TO GPS 1 DATA TWO HALFWORD 31	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,32)	V93U1368C	GPC TO GPS 1 DATA TWO HALFWORD 32	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,4)	V93U1340C	GPC TO GPS 1 DATA TWO HALFWORD 4	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,5)	V93U1341C	GPC TO GPS 1 DATA TWO HALFWORD 5	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,6)	V93U1342C	GPC TO GPS 1 DATA TWO HALFWORD 6	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,7)	V93U1343C	GPC TO GPS 1 DATA TWO HALFWORD 7	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,8)	V93U1344C	GPC TO GPS 1 DATA TWO HALFWORD 8	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(1,9)	V93U1345C	GPC TO GPS 1 DATA TWO HALFWORD 9	GPS1		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,1)	V93U1369B	GPS2 DATA TWO OUTPUT HEADER	GPS2	ND	BCH		92067D	
			TLM					
GPC_TO_GPS_DATA_TWO(2,10)	V93U1378C	GPC TO GPS 2 DATA TWO HALFWORD 10	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,11)	V93U1379C	GPC TO GPS 2 DATA TWO HALFWORD 11	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,12)	V93U1380C	GPC TO GPS 2 DATA TWO HALFWORD 12	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,13)	V93U1381C	GPC TO GPS 2 DATA TWO HALFWORD 13	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,14)	V93U1382C	GPC TO GPS 2 DATA TWO HALFWORD 14	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,15)	V93U1383C	GPC TO GPS 2 DATA TWO HALFWORD 15	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,16)	V93U1384C	GPC TO GPS 2 DATA TWO HALFWORD 16	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,17)	V93U1385C	GPC TO GPS 2 DATA TWO HALFWORD 17	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,18)	V93U1386C	GPC TO GPS 2 DATA TWO HALFWORD 18	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,19)	V93U1387C	GPC TO GPS 2 DATA TWO HALFWORD 19	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,2)	V93U1370C	GPC TO GPS 2 DATA TWO HALFWORD 2	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,20)	V93U1388C	GPC TO GPS 2 DATA TWO HALFWORD 20	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,21)	V93U1389C	GPC TO GPS 2 DATA TWO HALFWORD 21	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,22)	V93U1390C	GPC TO GPS 2 DATA TWO HALFWORD 22	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,23)	V93U1391C	GPC TO GPS 2 DATA TWO HALFWORD 23	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,24)	V93U1392C	GPC TO GPS 2 DATA TWO HALFWORD 24	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,25)	V93U1393C	GPC TO GPS 2 DATA TWO HALFWORD 25	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,26)	V93U1394C	GPC TO GPS 2 DATA TWO HALFWORD 26	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,27)	V93U1395C	GPC TO GPS 2 DATA TWO HALFWORD 27	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,28)	V93U1396C	GPC TO GPS 2 DATA TWO HALFWORD 28	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,29)	V93U1397C	GPC TO GPS 2 DATA TWO HALFWORD 29	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,3)	V93U1371C	GPC TO GPS 2 DATA TWO HALFWORD 3	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,30)	V93U1398C	GPC TO GPS 2 DATA TWO HALFWORD 30	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,31)	V93U1399C	GPC TO GPS 2 DATA TWO HALFWORD 31	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,32)	V93U1400C	GPC TO GPS 2 DATA TWO HALFWORD 32	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,4)	V93U1372C	GPC TO GPS 2 DATA TWO HALFWORD 4	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,5)	V93U1373C	GPC TO GPS 2 DATA TWO HALFWORD 5	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,6)	V93U1374C	GPC TO GPS 2 DATA TWO HALFWORD 6	GPS2		HMD		91051K	

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
GPC_TO_GPS_DATA_TWO(2,7)	V93U1375C	GPC TO GPS 2 DATA TWO HALFWORD 7	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,8)	V93U1376C	GPC TO GPS 2 DATA TWO HALFWORD 8	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(2,9)	V93U1377C	GPC TO GPS 2 DATA TWO HALFWORD 9	GPS2		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,1)	V93U1551B	GPS3 DATA TWO OUTPUT HEADER	GPS3	ND	BCH		92067D	
			TLM					
GPC_TO_GPS_DATA_TWO(3,10)	V93U1560C	GPC TO GPS 3 DATA TWO HALFWORD 10	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,11)	V93U1561C	GPC TO GPS 3 DATA TWO HALFWORD 11	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,12)	V93U1562C	GPC TO GPS 3 DATA TWO HALFWORD 12	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,13)	V93U1563C	GPC TO GPS 3 DATA TWO HALFWORD 13	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,14)	V93U1564C	GPC TO GPS 3 DATA TWO HALFWORD 14	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,15)	V93U1565C	GPC TO GPS 3 DATA TWO HALFWORD 15	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,16)	V93U1566C	GPC TO GPS 3 DATA TWO HALFWORD 16	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,17)	V93U1567C	GPC TO GPS 3 DATA TWO HALFWORD 17	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,18)	V93U1568C	GPC TO GPS 3 DATA TWO HALFWORD 18	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,19)	V93U1569C	GPC TO GPS 3 DATA TWO HALFWORD 19	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,2)	V93U1552C	GPC TO GPS 3 DATA TWO HALFWORD 2	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,20)	V93U1570C	GPC TO GPS 3 DATA TWO HALFWORD 20	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,21)	V93U1571C	GPC TO GPS 3 DATA TWO HALFWORD 21	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,22)	V93U1572C	GPC TO GPS 3 DATA TWO HALFWORD 22	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,23)	V93U1573C	GPC TO GPS 3 DATA TWO HALFWORD 23	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,24)	V93U1574C	GPC TO GPS 3 DATA TWO HALFWORD 24	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,25)	V93U1575C	GPC TO GPS 3 DATA TWO HALFWORD 25	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,26)	V93U1576C	GPC TO GPS 3 DATA TWO HALFWORD 26	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,27)	V93U1577C	GPC TO GPS 3 DATA TWO HALFWORD 27	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,28)	V93U1578C	GPC TO GPS 3 DATA TWO HALFWORD 28	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,29)	V93U1579C	GPC TO GPS 3 DATA TWO HALFWORD 29	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,3)	V93U1553C	GPC TO GPS 3 DATA TWO HALFWORD 3	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,30)	V93U1580C	GPC TO GPS 3 DATA TWO HALFWORD 30	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,31)	V93U1581C	GPC TO GPS 3 DATA TWO HALFWORD 31	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,32)	V93U1582C	GPC TO GPS 3 DATA TWO HALFWORD 32	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,4)	V93U1554C	GPC TO GPS 3 DATA TWO HALFWORD 4	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,5)	V93U1555C	GPC TO GPS 3 DATA TWO HALFWORD 5	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,6)	V93U1556C	GPC TO GPS 3 DATA TWO HALFWORD 6	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,7)	V93U1557C	GPC TO GPS 3 DATA TWO HALFWORD 7	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,8)	V93U1558C	GPC TO GPS 3 DATA TWO HALFWORD 8	GPS3		HMD		91051K	
GPC_TO_GPS_DATA_TWO(3,9)	V93U1559C	GPC TO GPS 3 DATA TWO HALFWORD 9	GPS3		HMD		91051K	
GPS_AIDING_DATA_AGE_DL(1)	V74W3477B	GPS1 AGE OF AIDING DATA	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1	MS	HXS		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
GPS_AIDING_DATA_AGE_DL(2)	V74W3478B	GPS2 AGE OF AIDING DATA	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1	MS	HXS		93017G	*
GPS_AIDING_DATA_AGE_DL(3)	V74W3479B	GPS3 AGE OF AIDING DATA	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1	MS	HXS		93017G	*
GPS_ANNCD_RESET_SET(1)	V90X3290XB	GPS 1 REF STATE ANNCD RESET	GPS REF STATE PROC		BD		92067D	
GPS_ANNCD_RESET_SET(2)	V90X3291XB	GPS 2 REF STATE ANNCD RESET	GPS REF STATE PROC		BD		92067D	
GPS_ANNCD_RESET_SET(3)	V90X3292XB	GPS 3 REF STATE ANNCD RESET	GPS REF STATE PROC		BD		92067D	
GPS_AOA_SEL	V90W0435C	SEL GPS AGE-OF-AIDING DATA	DL FMT 21/A DL FMT 22/A DL FMT 23/A	MS	HXS		93017G	*
GPS_DG_SOP(1)	V90X3012X	GPS 1 DATA GOOD FLAG	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP		BD		92067D	
GPS_DG_SOP(2)	V90X3013X	GPS 2 DATA GOOD FLAG	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP		BD		92067D	
GPS_DG_SOP(3)	V90X3014X	GPS 3 DATA GOOD FLAG	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP		BD		92067D	

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
GPS_ECEF_POS_X_DL(1)	V74H4453B	GPS1 WGS 84 ECEF POSITION X	DL FMT 21/1	FT	FXS		93017G	*
			DL FMT 22/1					
			DL FMT 23/1					
			DL FMT 32/1					
			DL FMT 44/1					
			DL FMT 45/1					
GPS_ECEF_POS_X_DL(2)	V74H1453B	GPS2 WGS 84 ECEF POSITION X	DL FMT 21/1	FT	FXS		93017G	*
			DL FMT 22/1					
			DL FMT 23/1					
			DL FMT 32/1					
			DL FMT 42/1					
			DL FMT 43/1					
GPS_ECEF_POS_X_DL(3)	V74H3453B	GPS3 WGS 84 ECEF POSITION X	DL FMT 21/1	FT	FXS		93017G	*
			DL FMT 22/1					
			DL FMT 23/1					
			DL FMT 32/1					
			DL FMT 44/1					
			DL FMT 45/1					
GPS_ECEF_POS_Y_DL(1)	V74H4454B	GPS1 WGS 84 ECEF POSITION Y	DL FMT 21/1	FT	FXS		93017G	*
			DL FMT 22/1					
			DL FMT 23/1					
			DL FMT 32/1					
			DL FMT 44/1					
			DL FMT 45/1					
GPS_ECEF_POS_Y_DL(2)	V74H1454B	GPS2 WGS 84 ECEF POSITION Y	DL FMT 21/1	FT	FXS		93017G	*
			DL FMT 22/1					
			DL FMT 23/1					
			DL FMT 32/1					
			DL FMT 42/1					
			DL FMT 43/1					
GPS_ECEF_POS_Y_DL(3)	V74H3454B	GPS3 WGS 84 ECEF POSITION Y	DL FMT 21/1	FT	FXS		93017G	*
			DL FMT 22/1					
			DL FMT 23/1					
			DL FMT 32/1					
			DL FMT 44/1					
			DL FMT 45/1					
			DL FMT 60/1					

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
GPS_ECEF_POS_Z_DL(1)	V74H4455B	GPS1 WGS 84 ECEF POSITION Z	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1	FT	FXS		93017G	*
GPS_ECEF_POS_Z_DL(2)	V74H1455B	GPS2 WGS 84 ECEF POSITION Z	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1	FT	FXS		93017G	*
GPS_ECEF_POS_Z_DL(3)	V74H3455B	GPS3 WGS 84 ECEF POSITION Z	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1	FT	FXS		93017G	*
GPS_ECEF_VEL_X_DL(1)	V74L4456B	GPS1 WGS 84 ECEF VELOCITY X	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1	FT/S	FXS		93017G	*
GPS_ECEF_VEL_X_DL(2)	V74L1456B	GPS2 WGS 84 ECEF VELOCITY X	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1	FT/S	FXS		93017G	*
GPS_ECEF_VEL_X_DL(3)	V74L3456B	GPS3 WGS 84 ECEF VELOCITY X	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1	FT/S	FXS		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
GPS_ECEF_VEL_Y_DL(1)	V74L4457B	GPS1 WGS 84 ECEF VELOCITY Y	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1	FT/S	FXS		93017G	*
GPS_ECEF_VEL_Y_DL(2)	V74L1457B	GPS2 WGS 84 ECEF VELOCITY Y	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1	FT/S	FXS		93017G	*
GPS_ECEF_VEL_Y_DL(3)	V74L3457B	GPS3 WGS 84 ECEF VELOCITY Y	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1	FT/S	FXS		93017G	*
GPS_ECEF_VEL_Z_DL(1)	V74L4458B	GPS1 WGS 84 ECEF VELOCITY Z	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1	FT/S	FXS		93017G	*
GPS_ECEF_VEL_Z_DL(2)	V74L1458B	GPS2 WGS 84 ECEF VELOCITY Z	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1	FT/S	FXS		93017G	*
GPS_ECEF_VEL_Z_DL(3)	V74L3458B	GPS3 WGS 84 ECEF VELOCITY Z	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1	FT/S	FXS		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
GPS_FAIL_FOM_DISP(1)	V94X1155XB	GPS 1 FAIL FOM DISP	GPS STATUS DISP		BD		92067D	
GPS_FAIL_FOM_DISP(2)	V94X1156XB	GPS 2 FAIL FOM DISP	HORIZ SIT DISP		BD		92067D	
GPS_FAIL_FOM_DISP(3)	V94X1157XB	GPS 3 FAIL FOM DISP	GPS STATUS DISP		BD		92067D	
GPS_FAIL_FOM_SOP(1)	V90X3015X	GPS 1 FOM THRESH FAIL FLAG	HORIZ SIT DISP		BD		92067D	
GPS_FAIL_FOM_SOP(2)	V90X3016X	GPS 2 FOM THRESH FAIL FLAG	ENT NAV		BD		92067D	
GPS_FAIL_FOM_SOP(3)	V90X3017X	GPS 3 FOM THRESH FAIL FLAG	ORB/RND NAV		BD		92067D	
GPS_FAIL_ST(1)	V94J6059C	GPS 1 FAIL SELF-TEST IND	PRELAND NAV		VAR		92067D	
GPS_FAIL_ST(2)	V94J6060C	GPS 2 FAIL SELF-TEST IND	RTLS NAV		VAR		92067D	
GPS_FAIL_ST(3)	V94J6061C	GPS 3 FAIL SELF-TEST IND	ENT NAV		VAR		92067D	
GPS_FILT_RESTART(1)	V90X3269XA	GPS 1 FILTER RESTART FLAG	ORB/RND NAV		BD		92049C	
GPS_FILT_RESTART(2)	V90X3270XA	GPS 2 FILTER RESTART FLAG	PRELAND NAV		BD		92049C	
GPS_FILT_RESTART(3)	V90X3271XA	GPS 3 FILTER RESTART FLAG	RTLS NAV		BD		92049C	
GPS_INIT_REQUEST_SOP(1)	V90X3018XB	GPS 1 REQUEST INITIAL STATE	GPS STATUS DISP		BD		92067D	
GPS_INIT_REQUEST_SOP(2)	V90X3019XB	GPS 2 REQUEST INITIAL STATE	GPS REF STATE PROC		BD		92067D	
GPS_INIT_REQUEST_SOP(3)	V90X3020XB	GPS 3 REQUEST INITIAL STATE	GPS REF STATE PROC		BD		92067D	
GPS_MODE_CMD_RCVD(1)	V94J0925C	GPS 1 MODE CMD FDBK	GPS STATUS DISP		HXS		91051K	
GPS_MODE_CMD_RCVD(2)	V94J0926C	GPS 2 MODE CMD FDBK	GPS STATUS DISP		HXS		91051K	
GPS_MODE_CMD_RCVD(3)	V94J0927C	GPS 3 MODE CMD FDBK	GPS STATUS DISP		HXS		91051K	
GPS_MODE_CURRENT(1)	V90U3021C	GPS 1 CURRENT OPERATING MODE	GPS REF STATE PROC		VAR		92132C	
GPS_MODE_CURRENT(2)	V90U3022C	GPS 2 CURRENT OPERATING MODE	GPS STATUS DISP		VAR		92132C	
GPS_MODE_CURRENT(3)	V90U3023C	GPS 3 CURRENT OPERATING MODE	GPS REF STATE PROC		VAR		92132C	
GPS_MODE_DISP(1)	V94J0928C	GPS 1 MODE	GPS STATUS DISP		VAR		91051K	
GPS_MODE_DISP(2)	V94J0929C	GPS 2 MODE	GPS STATUS DISP		VAR		91051K	
GPS_MODE_DISP(3)	V94J0930C	GPS 3 MODE	GPS STATUS DISP		VAR		91051K	
GPS_NAV_SUBMODE(1)	V90U2616C	GPS 1 NAV SUBMODE INDICATOR	GPS REF STATE PROC		VAR		91051K	
GPS_NAV_SUBMODE(2)	V90U2617C	GPS 2 NAV SUBMODE INDICATOR	GPS REF STATE PROC		VAR		91051K	
GPS_NAV_SUBMODE(3)	V90U2618C	GPS 3 NAV SUBMODE INDICATOR	GPS REF STATE PROC		VAR		91051K	
GPS_RA_RCVD	V94X0931X	GPS RA FLAG FDBK	HORIZ SIT DISP		BD		91051K	

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
GPS_RCVR_STATUS(1)	V94J0932C	GPS 1 RCVR STATUS FOR DISPLAY	GPS STATUS DISP HORIZ SIT DISP REL NAV DISP		VAR		92067D	
GPS_RCVR_STATUS(2)	V94J0933C	GPS 2 RCVR STATUS FOR DISPLAY	GPS STATUS DISP HORIZ SIT DISP REL NAV DISP		VAR		92067D	
GPS_RCVR_STATUS(3)	V94J0934C	GPS 3 RCVR STATUS FOR DISPLAY	GPS STATUS DISP HORIZ SIT DISP REL NAV DISP		VAR		92067D	
GPS_SEC_DL(1)	V74W4452B	GPS1 SECONDS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1	S	BSU		93017G	*
GPS_SEC_DL(2)	V74W1452B	GPS2 SECONDS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1	S	BSU		93017G	*
GPS_SEC_DL(3)	V74W3452B	GPS3 SECONDS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1	S	BSU		93017G	*
GPS_SOP_COLLECTION_WORD	V93M1151P	GPS SOP COLLECTION WORD	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
GPS_SOP_COLLECTION_WORD_TWO	V93M3151P	GPS SOP COLLECTION WORD TWO	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
GPS_STATUS_FOM_GDOP(1)	V74M4461P	GPS1 STATUS FOM GDOP WORD 1	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
GPS_STATUS_FOM_GDOP(2)	V74M1461P	GPS2 STATUS FOM GDOP WORD 1	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
GPS_STATUS_FOM_GDOP(3)	V74M3461P	GPS3 STATUS FOM GDOP WORD 1	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
GPS_TIME_ADJ_ENBL	V94X0220X	GPS TIME ADJ CMD RCVD	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1 GPS STATUS DISP	EVENT	BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
GPS_WEEK_DL(1)	V74W4451B	GPS1 WEEKS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1		HXU		93017G	*
GPS_WEEK_DL(2)	V74W1451B	GPS2 WEEKS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HXU		93017G	*
GPS_WEEK_DL(3)	V74W3451B	GPS3 WEEKS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HXU		93017G	*
INIT_STATE_SET(1)	V90X3278XB	GPS 1 REFERENCE STATE INITIALIZED	GPS REF STATE PROC		BD		92067D	
INIT_STATE_SET(2)	V90X3279XB	GPS 2 REFERENCE STATE INITIALIZED	GPS REF STATE PROC		BD		92067D	
INIT_STATE_SET(3)	V90X3280XB	GPS 3 REFERENCE STATE INITIALIZED	GPS REF STATE PROC		BD		92067D	
NUM_GPS_INSTALLED	V90U3671C	NUMBER OF GPS RCVR INSTALLED	A/L UPP ASC AREA NAV E/L EV AN PROC ENT AREA NAV ENT UPP GPS RCVR STATE PROC RTLS UPP TACAN RM TAEM UPP		HXS		92067D	
PPS_STATUS_WORD(1)	V74M4459P	GPS1 PPS STATUS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
PPS_STATUS_WORD(2)	V74M1459P	GPS2 PPS STATUS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
PPS_STATUS_WORD(3)	V74M3459P	GPS3 PPS STATUS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		HMD		93017G	*
RCVR_FAIL(1)	V94X0935X	GPS 1 RCVR FAIL IND	GAX		BD		92067D	
RCVR_FAIL(2)	V94X0936X	GPS 2 RCVR FAIL IND	GAX		BD		92069B	
RCVR_FAIL(3)	V94X0937X	GPS 3 RCVR FAIL IND	GAX		BD		92067D	
R_GPS_ECEF_NEW(1,1)	V90H1346C	GPS 1 NEW WGS84 POS VECTOR X-COMP	ASC AREA NAV ENT AREA NAV	FT	DPL		92067D	
R_GPS_ECEF_NEW(1,2)	V90H1347C	GPS 1 NEW WGS84 POS VECTOR Y-COMP	ASC AREA NAV ENT AREA NAV	FT	DPL		92067D	
R_GPS_ECEF_NEW(1,3)	V90H1348C	GPS 1 NEW WGS84 POS VECTOR Z-COMP	ASC AREA NAV ENT AREA NAV	FT	DPL		92067D	
R_GPS_ECEF_NEW(2,1)	V90H1349C	GPS 2 NEW WGS84 POS VECTOR X-COMP	ASC AREA NAV ENT AREA NAV	FT	DPL		92067D	
R_GPS_ECEF_NEW(2,2)	V90H1350C	GPS 2 NEW WGS84 POS VECTOR Y-COMP	ASC AREA NAV ENT AREA NAV	FT	DPL		92067D	
R_GPS_ECEF_NEW(2,3)	V90H1351C	GPS 2 NEW WGS84 POS VECTOR Z-COMP	ASC AREA NAV ENT AREA NAV	FT	DPL		92067D	
R_GPS_ECEF_NEW(3,1)	V90H1352C	GPS 3 NEW WGS84 POS VECTOR X-COMP	ASC AREA NAV ENT AREA NAV	FT	DPL		92067D	
R_GPS_ECEF_NEW(3,2)	V90H1353C	GPS 3 NEW WGS84 POS VECTOR Y-COMP	ASC AREA NAV ENT AREA NAV	FT	DPL		92067D	
R_GPS_ECEF_NEW(3,3)	V90H1354C	GPS 3 NEW WGS84 POS VECTOR Z-COMP	ASC AREA NAV ENT AREA NAV	FT	DPL		92067D	

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
R_GPS_ECEF_SOP(1,1)	V90H3036C	GPS 1 WGS84 POS VECTOR X-COMP	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP	FT	DPL		92067D	
R_GPS_ECEF_SOP(1,2)	V90H3037C	GPS 1 WGS84 POS VECTOR Y-COMP	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP	FT	DPL		92067D	
R_GPS_ECEF_SOP(1,3)	V90H3038C	GPS 1 WGS84 POS VECTOR Z-COMP	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP	FT	DPL		92067D	
R_GPS_ECEF_SOP(2,1)	V90H3039C	GPS 2 WGS84 POS VECTOR X-COMP	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP	FT	DPL		92067D	
R_GPS_ECEF_SOP(2,2)	V90H3040C	GPS 2 WGS84 POS VECTOR Y-COMP	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP	FT	DPL		92067D	

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
R_GPS_ECEF_SOP(2,3)	V90H3041C	GPS 2 WGS84 POS VECTOR Z-COMP	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP	FT	DPL		92067D	
R_GPS_ECEF_SOP(3,1)	V90H3042C	GPS 3 WGS84 POS VECTOR X-COMP	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP	FT	DPL		92067D	
R_GPS_ECEF_SOP(3,2)	V90H3043C	GPS 3 WGS84 POS VECTOR Y-COMP	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP	FT	DPL		92067D	
R_GPS_ECEF_SOP(3,3)	V90H3044C	GPS 3 WGS84 POS VECTOR Z-COMP	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP	FT	DPL		92067D	
SATS_BLANKING_FLAG(1,1)	V94X0185X	GPS 1 SAT 1 DISP BLANKING	GPS STATUS DISP		BD		92281D	
SATS_BLANKING_FLAG(1,2)	V94X0186X	GPS 1 SAT 2 DISP BLANKING	GPS STATUS DISP		BD		92281D	
SATS_BLANKING_FLAG(1,3)	V94X0187X	GPS 1 SAT 3 DISP BLANKING	GPS STATUS DISP		BD		92281D	
SATS_BLANKING_FLAG(1,4)	V94X0188X	GPS 1 SAT 4 DISP BLANKING	GPS STATUS DISP		BD		92281D	
SATS_BLANKING_FLAG(1,5)	V94X0189X	GPS 1 SAT 5 DISP BLANKING	GPS STATUS DISP		BD		92281D	
SATS_BLANKING_FLAG(1,6)	V94X0202X	GSP 1 SAT 6 DISP BLANKING	GPS STATUS DISP		BD		92281D	
SATS_BLANKING_FLAG(2,1)	V94X0190X	GPS 2 SAT 1 DISP BLANKING	GPS STATUS DISP		BD		92281D	
SATS_BLANKING_FLAG(2,2)	V94X0191X	GPS 2 SAT 2 DISP BLANKING	GPS STATUS DISP		BD		92281D	
SATS_BLANKING_FLAG(2,3)	V94X0192X	GPS 2 SAT 3 DISP BLANKING	GPS STATUS DISP		BD		92281D	
SATS_BLANKING_FLAG(2,4)	V94X0193X	GSP 2 SAT 4 DISP BLANKING	GPS STATUS DISP		BD		92281D	
SATS_BLANKING_FLAG(2,5)	V94X0194X	GPS 2 SAT 5 DISP BLANKING	GPS STATUS DISP		BD		92281D	
SATS_BLANKING_FLAG(2,6)	V94X0203X	GPS 2 SAT 6 DISP BLANKING	GPS STATUS DISP		BD		92281D	
SATS_BLANKING_FLAG(3,1)	V94X0195X	GSP 3 SAT 1 DISP BLANKING	GPS STATUS DISP		BD		92281D	
SATS_BLANKING_FLAG(3,2)	V94X0196X	GSP 3 SAT 2 DISP BLANKING	GPS STATUS DISP		BD		92281D	

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
SATS_BLANKING_FLAG(3,3)	V94X0197X	GPS 3 SAT 3 DISP BLANKING	GPS STATUS DISP		BD		92281D	
SATS_BLANKING_FLAG(3,4)	V94X0198X	GPS 3 SAT 4 DISP BLANKING	GPS STATUS DISP		BD		92281D	
SATS_BLANKING_FLAG(3,5)	V94X0199X	GPS 3 SAT 5 DISP BLANKING	GPS STATUS DISP		BD		92281D	
SATS_BLANKING_FLAG(3,6)	V94X0204X	GPS 3 SAT 6 DISP BLANKING	GPS STATUS DISP		BD		92281D	
SATS_TRACKED(1,1)	V94U0938C	GPS 1 SAT NO 1 TRACKED	GPS STATUS DISP		HXS		91051K	
SATS_TRACKED(1,2)	V94U0939C	GPS 1 SAT NO 2 TRACKED	GPS STATUS DISP		HXS		91051K	
SATS_TRACKED(1,3)	V94U0940C	GPS 1 SAT NO 3 TRACKED	GPS STATUS DISP		HXS		91051K	
SATS_TRACKED(1,4)	V94U0941C	GPS 1 SAT NO 4 TRACKED	GPS STATUS DISP		HXS		91051K	
SATS_TRACKED(1,5)	V94U0942C	GPS 1 SAT NO 5 TRACKED	GPS STATUS DISP		HXS		91051K	
SATS_TRACKED(1,6)	V94U0943C	GPS 1 SAT NO 6 TRACKED	GPS STATUS DISP		HXS		91051K	
SATS_TRACKED(2,1)	V94U0944C	GPS 2 SAT NO 1 TRACKED	GPS STATUS DISP		HXS		91051K	
SATS_TRACKED(2,2)	V94U0945C	GPS 2 SAT NO 2 TRACKED	GPS STATUS DISP		HXS		91051K	
SATS_TRACKED(2,3)	V94U0946C	GPS 2 SAT NO 3 TRACKED	GPS STATUS DISP		HXS		91051K	
SATS_TRACKED(2,4)	V94U0947C	GPS 2 SAT NO 4 TRACKED	GPS STATUS DISP		HXS		91051K	
SATS_TRACKED(2,5)	V94U0948C	GPS 2 SAT NO 5 TRACKED	GPS STATUS DISP		HXS		91051K	
SATS_TRACKED(2,6)	V94U0949C	GPS 2 SAT NO 6 TRACKED	GPS STATUS DISP		HXS		91051K	
SATS_TRACKED(3,1)	V94U0950C	GPS 3 SAT NO 1 TRACKED	GPS STATUS DISP		HXS		91051K	
SATS_TRACKED(3,2)	V94U0951C	GPS 3 SAT NO 2 TRACKED	GPS STATUS DISP		HXS		91051K	
SATS_TRACKED(3,3)	V94U0952C	GPS 3 SAT NO 3 TRACKED	GPS STATUS DISP		HXS		91051K	
SATS_TRACKED(3,4)	V94U0953C	GPS 3 SAT NO 4 TRACKED	GPS STATUS DISP		HXS		91051K	
SATS_TRACKED(3,5)	V94U0954C	GPS 3 SAT NO 5 TRACKED	GPS STATUS DISP		HXS		91051K	
SATS_TRACKED(3,6)	V94U0955C	GPS 3 SAT NO 6 TRACKED	GPS STATUS DISP		HXS		91051K	
T_GPS_NEW(1)	V90W1355C	GPS 1 NEW STATE TIME TAG(GMT)	ASC AREA NAV ENT AREA NAV	S	DPL		92067D	
T_GPS_NEW(2)	V90W1356C	GPS 2 NEW STATE TIME TAG(GMT)	ASC AREA NAV ENT AREA NAV	S	DPL		92067D	
T_GPS_NEW(3)	V90W1357C	GPS 3 NEW STATE TIME TAG(GMT)	ASC AREA NAV ENT AREA NAV	S	DPL		92067D	
T_GPS_OFFSET_SECS	V91W0029C	GPS OFFSET SECONDS	D/L NAV SEQ	S	DPL		93160B	*
T_GPS_SOP(1)	V90W3045C	GPS 1 STATE TIME TAG(GMT)	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP	S	DPL		92067D	
T_GPS_SOP(2)	V90W3046C	GPS 2 STATE TIME TAG(GMT)	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP	S	DPL		92067D	

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
T_GPS_SOP(3)	V90W3047C	GPS 3 STATE TIME TAG(GMT)	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP	S	DPL		92067D	
T_GPS_TIME_ADJ_ACC	V90W0436C	ACCUM GPC-TO-GPS TIME ADJ	DL FMT 21/A DL FMT 22/A DL FMT 23/A	MS	HXS		93017G	*
T_IMU_SAVE_SOP(1)	V90W3048C	IMU TIME TAG FOR CURNT GPS 1 STATE	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP	S	DPL		92067D	
T_IMU_SAVE_SOP(2)	V90W3049C	IMU TIME TAG FOR CURNT GPS 2 STATE	TLM A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP	S	DPL		92067D	
T_IMU_SAVE_SOP(3)	V90W3050C	IMU TIME TAG FOR CURNT GPS 3 STATE	TLM A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP	S	DPL		92067D	
V_GPS_ECEF_NEW(1,1)	V90L1521C	GPS 1 NEW WGS84 VEL VECTOR X-COMP	TLM ASC AREA NAV ENT AREA NAV	FT/S	SPL		92389C	
V_GPS_ECEF_NEW(1,2)	V90L1522C	GPS 1 NEW WGS84 VEL VECTOR Y-COMP	ASC AREA NAV ENT AREA NAV	FT/S	SPL		92389C	
V_GPS_ECEF_NEW(1,3)	V90L1523C	GPS 1 NEW WGS84 VEL VECTOR Z-COMP	ASC AREA NAV ENT AREA NAV	FT/S	SPL		92389C	
V_GPS_ECEF_NEW(2,1)	V90L1524C	GPS 2 NEW WGS84 VEL VECTOR X-COMP	ASC AREA NAV ENT AREA NAV	FT/S	SPL		92389C	

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
V_GPS_ECEF_NEW(2,2)	V90L1525C	GPS 2 NEW WGS84 VEL VECTOR Y-COMP	ASC AREA NAV ENT AREA NAV	FT/S	SPL		92389C	
V_GPS_ECEF_NEW(2,3)	V90L1526C	GPS 2 NEW WGS84 VEL VECTOR Z-COMP	ASC AREA NAV ENT AREA NAV	FT/S	SPL		92389C	
V_GPS_ECEF_NEW(3,1)	V90L1527C	GPS 3 NEW WGS84 VEL VECTOR X-COMP	ASC AREA NAV ENT AREA NAV	FT/S	SPL		92389C	
V_GPS_ECEF_NEW(3,2)	V90L1528C	GPS 3 NEW WGS84 VEL VECTOR Y-COMP	ASC AREA NAV ENT AREA NAV	FT/S	SPL		92389C	
V_GPS_ECEF_NEW(3,3)	V90L1529C	GPS 3 NEW WGS84 VEL VECTOR Z-COMP	ASC AREA NAV ENT AREA NAV	FT/S	SPL		92389C	
V_GPS_ECEF_SOP(1,1)	V90L3051C	GPS 1 WGS84 VEL VECTOR X-COMP	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP	FT/S	SPL		92067D	
V_GPS_ECEF_SOP(1,2)	V90L3052C	GPS 1 WGS84 VEL VECTOR Y-COMP	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP	FT/S	SPL		92067D	
V_GPS_ECEF_SOP(1,3)	V90L3053C	GPS 1 WGS84 VEL VECTOR Z-COMP	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP	FT/S	SPL		92067D	
V_GPS_ECEF_SOP(2,1)	V90L3054C	GPS 2 WGS84 VEL VECTOR X-COMP	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP	FT/S	SPL		92067D	

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
V_GPS_ECEF_SOP(2,2)	V90L3055C	GPS 2 WGS84 VEL VECTOR Y-COMP	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP	FT/S	SPL		92067D	
V_GPS_ECEF_SOP(2,3)	V90L3056C	GPS 2 WGS84 VEL VECTOR Z-COMP	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP	FT/S	SPL		92067D	
V_GPS_ECEF_SOP(3,1)	V90L3057C	GPS 3 WGS84 VEL VECTOR X-COMP	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP	FT/S	SPL		92067D	
V_GPS_ECEF_SOP(3,2)	V90L3058C	GPS 3 WGS84 VEL VECTOR Y-COMP	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP	FT/S	SPL		92067D	
V_GPS_ECEF_SOP(3,3)	V90L3059C	GPS 3 WGS84 VEL VECTOR Z-COMP	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP	FT/S	SPL		92067D	

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
V_IMU_SAVE_SOP(1,1)	V95L0133C	IMU ACCUM X VEL FOR CURRENT GPS 1	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP TLM	FT/S	DPL		92067D	
V_IMU_SAVE_SOP(1,2)	V95L0134C	IMU ACCUM Y VEL FOR CURRENT GPS 1	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP TLM	FT/S	DPL		92067D	
V_IMU_SAVE_SOP(1,3)	V95L0135C	IMU ACCUM Z VEL FOR CURRENT GPS 1	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP TLM	FT/S	DPL		92067D	
V_IMU_SAVE_SOP(2,1)	V95L0136C	IMU ACCUM X VEL FOR CURRENT GPS 2	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP TLM	FT/S	DPL		92067D	
V_IMU_SAVE_SOP(2,2)	V95L0137C	IMU ACCUM Y VEL FOR CURRENT GPS 2	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP TLM	FT/S	DPL		92067D	

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
V_IMU_SAVE_SOP(2,3)	V95L0138C	IMU ACCUM Z VEL FOR CURRENT GPS 2	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP TLM	FT/S	DPL		92067D	
V_IMU_SAVE_SOP(3,1)	V95L0139C	IMU ACCUM X VEL FOR CURRENT GPS 3	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP TLM	FT/S	DPL		92067D	
V_IMU_SAVE_SOP(3,2)	V95L0140C	IMU ACCUM Y VEL FOR CURRENT GPS 3	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP TLM	FT/S	DPL		92067D	
V_IMU_SAVE_SOP(3,3)	V95L0141C	IMU ACCUM Z VEL FOR CURRENT GPS 3	A/L UPP ENT NAV ENT UPP ORB/RND NAV PRELAND NAV RTLS NAV RTLS UPP TAEM UPP TLM	FT/S	DPL		92067D	
	V74E1978B	GPS2 CHAN 1 SIGNAL TO NOISE RATIO	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74E1980B	GPS2 CHAN 2 SIGNAL TO NOISE RATIO	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74E1989B	GPS2 CHAN 3 SIGNAL TO NOISE RATIO	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74E1991B	GPS2 CHAN 4 SIGNAL TO NOISE RATIO	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74E2080B	GPS2 CHAN 5 SIGNAL TO NOISE RATIO	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74E3080B	GPS3 CHAN 5 SIGNAL TO NOISE RATIO	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74E4978B	GPS1 CHAN 1 SIGNAL TO NOISE RATIO	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74E4980B	GPS1 CHAN 2 SIGNAL TO NOISE RATIO	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74E4989B	GPS1 CHAN 3 SIGNAL TO NOISE RATIO	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74E4991B	GPS1 CHAN 4 SIGNAL TO NOISE RATIO	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74E5978B	GPS3 CHAN 1 SIGNAL TO NOISE RATIO	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74E5980B	GPS3 CHAN 2 SIGNAL TO NOISE RATIO	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E V C	LAST CR	R E V
	V74E5989B	GPS3 CHAN 3 SIGNAL TO NOISE RATIO	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74E5991B	GPS3 CHAN 4 SIGNAL TO NOISE RATIO	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74E8080B	GPS1 CHAN 5 SIGNAL TO NOISE RATIO	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U1465B	GPS2 GEOMETRIC DILUTION OF PREC	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U1466B	GPS2 FIGURE OF MERIT	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74U1975B	GPS2 CHAN 1 STATE	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U1976B	GPS2 CHAN 1 HDWARE CHAN NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U1977B	GPS2 CHAN 1 SATELLITE NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U1979B	GPS2 CHAN 1 CODE TRACKING STATUS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U1981B	GPS2 CHAN 2 CODE TRACKING STATUS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74U1986B	GPS2 CHAN 2 STATE	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U1987B	GPS2 CHAN 2 HDWARE CHAN NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U1988B	GPS2 CHAN 2 SATELLITE NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U1990B	GPS2 CHAN 3 CODE TRACKING STATUS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U1992B	GPS2 CHAN 4 CODE TRACKING STATUS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74U1997B	GPS2 CHAN 3 STATE	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U1998B	GPS2 CHAN 3 HDWARE CHAN NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U1999B	GPS2 CHAN 3 SATELLITE NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U2081B	GPS2 CHAN 5 CODE TRACKING STATUS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U2088B	GPS2 CHAN 4 STATE	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74U2089B	GPS2 CHAN 4 HDWARE CHAN NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U2090B	GPS2 CHAN 4 SATELLITE NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U2095B	GPS2 CHAN 5 STATE	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U2096B	GPS2 CHAN 5 HDWARE CHAN NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U2097B	GPS2 CHAN 5 SATELLITE NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74U3081B	GPS3 CHAN 5 CODE TRACKING STATUS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1		BSU		93017G	*
	V74U3088B	GPS3 CHAN 4 STATE	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U3089B	GPS3 CHAN 4 HDWARE CHAN NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U3090B	GPS3 CHAN 4 SATELLITE NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U3095B	GPS3 CHAN 5 STATE	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U3096B	GPS3 CHAN 5 HDWARE CHAN NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74U3097B	GPS3 CHAN 5 SATELLITE NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U3465B	GPS3 GEOMETRIC DILUTION OF PREC	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U3466B	GPS3 FIGURE OF MERIT	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U4465B	GPS1 GEOMETRIC DILUTION OF PREC	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U4466B	GPS1 FIGURE OF MERIT	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U4975B	GPS1 CHAN 1 STATE	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74U4976B	GPS1 CHAN 1 HDWARE CHAN NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U4977B	GPS1 CHAN 1 SATELLITE NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U4979B	GPS1 CHAN 1 CODE TRACKING STATUS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U4981B	GPS1 CHAN 2 CODE TRACKING STATUS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U4986B	GPS1 CHAN 2 STATE	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U4987B	GPS1 CHAN 2 HDWARE CHAN NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74U4988B	GPS1 CHAN 2 SATELLITE NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U4990B	GPS1 CHAN 3 CODE TRACKING STATUS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U4992B	GPS1 CHAN 4 CODE TRACKING STATUS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U4997B	GPS1 CHAN 3 STATE	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U4998B	GPS1 CHAN 3 HDWARE CHAN NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U4999B	GPS1 CHAN 3 SATELLITE NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74U5975B	GPS3 CHAN 1 STATE	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U5976B	GPS3 CHAN 1 HDWARE CHAN NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U5977B	GPS3 CHAN 1 SATELLITE NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U5979B	GPS3 CHAN 1 CODE TRACKING STATUS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U5981B	GPS3 CHAN 2 CODE TRACKING STATUS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U5986B	GPS3 CHAN 2 STATE	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74U5987B	GPS3 CHAN 2 HDWARE CHAN NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U5988B	GPS3 CHAN 2 SATELLITE NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U5990B	GPS3 CHAN 3 CODE TRACKING STATUS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U5992B	GPS3 CHAN 4 CODE TRACKING STATUS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U5997B	GPS3 CHAN 3 STATE	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U5998B	GPS3 CHAN 3 HDWARE CHAN NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74U5999B	GPS3 CHAN 3 SATELLITE NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U8081B	GPS1 CHAN 5 CODE TRACKING STATUS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U8088B	GPS1 CHAN 4 STATE	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U8089B	GPS1 CHAN 4 HDWARE CHAN NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U8090B	GPS1 CHAN 4 SATELLITE NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U8095B	GPS1 CHAN 5 STATE	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74U8096B	GPS1 CHAN 5 HDWARE CHAN NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74U8097B	GPS1 CHAN 5 SATELLITE NUMBER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74W1475B	GPS2 PPS-MISSION DURATION FOR KEYS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74W3475B	GPS3 PPS-MISSION DURATION FOR KEYS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74W4475B	GPS1 PPS-MISSION DURATION FOR KEYS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V74X1398B	GPS2 LOG WD18 CHILD	TLM		BD		91051K	

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X1462B	GPS2 INCORP LESS THAN 4 MEAS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X1463B	GPS2 RPU FAIL	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X1464B	GPS2 BATTERY LOW	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X1467B	GPS2 PPS-DAILY KEY IN USE VERIFIED	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X1468B	GPS2 PPS-DAILY KEY IN USE UNVERIF	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X1469B	GPS2 PPS-DAILY KEY IN USE INCORR	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X1470B	GPS2 PPS-KEY FAILED PARITY	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X1471B	GPS2 PPS-GUV USER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X1472B	GPS2 PPS-2 HOUR ALERT	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X1473B	GPS2 PPS-INSUFFICIENT KEYS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X1474B	GPS2 PPS-RECEIVER CONTAINS KEYS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X1476B	GPS2 PPS-ERASE KEY FAIL	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1	EVENT	BD		93017G	*
	V74X1971B	GPS2 CHAN 1 FAULT FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X1972B	GPS2 CHAN 1 ANTENNA SELECT	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X1973B	GPS2 CHAN 1 FREQUENCY	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X1974B	GPS2 CHAN 1 EPHEM DOWNLOAD FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X1982B	GPS2 CHAN 2 FAULT FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X1983B	GPS2 CHAN 2 ANTENNA SELECT	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X1984B	GPS2 CHAN 2 FREQUENCY	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X1985B	GPS2 CHAN 2 EPHEM DOWNLOAD FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X1993B	GPS2 CHAN 3 FAULT FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X1994B	GPS2 CHAN 3 ANTENNA SELECT	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X1995B	GPS2 CHAN 3 FREQUENCY	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X1996B	GPS2 CHAN 3 EPHEM DOWNLOAD FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X2054B	GPS2 LOG WD1 CHILD	DL FMT 22/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X2055B	GPS2 LOG WD2 CHILD	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X2056B	GPS2 LOG WD3 CHILD	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X2057B	GPS2 LOG WD4 CHILD	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X2058B	GPS2 LOG WD5 CHILD	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X2059B	GPS2 LOG WD6 CHILD	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X2060B	GPS2 LOG WD7 CHILD	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X2061B	GPS2 LOG WD8 CHILD	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X2062B	GPS2 LOG WD9 CHILD	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X2063B	GPS2 LOG WD10 CHILD	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X2064B	GPS2 LOG WD11 CHILD	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X2065B	GPS2 LOG WD12 CHILD	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X2066B	GPS2 LOG WD13 CHILD	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X2067B	GPS2 LOG WD14 CHILD	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X2068B	GPS2 LOG WD15 CHILD	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X2069B	GPS2 LOG WD16 CHILD	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X2070B	GPS2 LOG WD17 CHILD	DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X2084B	GPS2 CHAN 4 FAULT FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X2085B	GPS2 CHAN 4 ANTENNA SELECT	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X2086B	GPS2 CHAN 4 FREQUENCY	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X2087B	GPS2 CHAN 4 EPHEM DOWNLOAD FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1		BD		93017G	*
	V74X2091B	GPS2 CHAN 5 FAULT FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X2092B	GPS2 CHAN 5 ANTENNA SELECT	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X2093B	GPS2 CHAN 5 FREQUENCY	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X2094B	GPS2 CHAN 5 EPHEM DOWNLOAD FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X3084B	GPS3 CHAN 4 FAULT FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3085B	GPS3 CHAN 4 ANTENNA SELECT	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3086B	GPS3 CHAN 4 FREQUENCY	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3087B	GPS3 CHAN 4 EPHEM DOWNLOAD FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3091B	GPS3 CHAN 5 FAULT FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3092B	GPS3 CHAN 5 ANTENNA SELECT	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X3093B	GPS3 CHAN 5 FREQUENCY	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3094B	GPS3 CHAN 5 EPHEM DOWNLOAD FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3099B	GPS1 LOG WD18 CHILD	TLM		BD		92067D	
	V74X3109B	GPS3 LOG WD18 CHILD	TLM		BD		92067D	
	V74X3462B	GPS3 INCORP LESS THAN 4 MEAS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3463B	GPS3 RPU FAIL	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3464B	GPS3 BATTERY LOW	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3467B	GPS3 PPS-DAILY KEY IN USE VERIFIED	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X3468B	GPS3 PPS-DAILY KEY IN USE UNVERIF	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3469B	GPS3 PPS-DAILY KEY IN USE INCORR	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3470B	GPS3 PPS-KEY FAILED PARITY	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3471B	GPS3 PPS-GUV USER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3472B	GPS3 PPS-2 HOUR ALERT	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X3473B	GPS3 PPS-INSUFFICIENT KEYS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3474B	GPS3 PPS-RECEIVER CONTAINS KEYS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3476B	GPS3 PPS-ERASE KEY FAIL	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1	EVENT	BD		93017G	*
	V74X3480X	GPS1 CHAN 1 PSEUDO RANGE EDIT IND	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3481X	GPS1 CHAN 2 PSEUDO RANGE EDIT IND	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X3482X	GPS1 CHAN 3 PSEUDO RANGE EDIT IND	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3483X	GPS1 CHAN 4 PSEUDO RANGE EDIT IND	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3484X	GPS1 CHAN 1 DELTA RANGE EDIT IND	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3485X	GPS1 CHAN 2 DELTA RANGE EDIT IND	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3486X	GPS1 CHAN 3 DELTA RANGE EDIT IND	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3487X	GPS1 CHAN 4 DELTA RANGE EDIT IND	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X3488X	GPS2 CHAN 1 PSEUDO RANGE EDIT IND	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3489X	GPS2 CHAN 2 PSEUDO RANGE EDIT IND	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3490X	GPS2 CHAN 3 PSEUDO RANGE EDIT IND	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3491X	GPS2 CHAN 4 PSEUDO RANGE EDIT IND	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3492X	GPS2 CHAN 1 DELTA RANGE EDIT IND	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X3493X	GPS2 CHAN 2 DELTA RANGE EDIT IND	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3494X	GPS2 CHAN 3 DELTA RANGE EDIT IND	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3495X	GPS2 CHAN 4 DELTA RANGE EDIT IND	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3496X	GPS3 CHAN 1 PSEUDO RANGE EDIT IND	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3497X	GPS3 CHAN 2 PSEUDO RANGE EDIT IND	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X3498X	GPS3 CHAN 3 PSEUDO RANGE EDIT IND	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3499X	GPS3 CHAN 4 PSEUDO RANGE EDIT IND	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3500X	GPS3 CHAN 1 DELTA RANGE EDIT IND	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3501X	GPS3 CHAN 2 DELTA RANGE EDIT IND	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3502X	GPS3 CHAN 3 DELTA RANGE EDIT IND	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X3503X	GPS3 CHAN 4 DELTA RANGE EDIT IND	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X4054B	GPS1 LOG WD1 CHILD	DL FMT 22/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4055B	GPS1 LOG WD2 CHILD	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4056B	GPS1 LOG WD3 CHILD	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4057B	GPS1 LOG WD4 CHILD	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4058B	GPS1 LOG WD5 CHILD	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4059B	GPS1 LOG WD6 CHILD	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4060B	GPS1 LOG WD7 CHILD	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4061B	GPS1 LOG WD8 CHILD	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4062B	GPS1 LOG WD9 CHILD	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4063B	GPS1 LOG WD10 CHILD	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4064B	GPS1 LOG WD11 CHILD	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X4065B	GPS1 LOG WD12 CHILD	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4066B	GPS1 LOG WD13 CHILD	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4067B	GPS1 LOG WD14 CHILD	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4068B	GPS1 LOG WD15 CHILD	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4069B	GPS1 LOG WD16 CHILD	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4070B	GPS1 LOG WD17 CHILD	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4462B	GPS1 INCORP LESS THAN 4 MEAS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4463B	GPS1 RPU FAIL	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4464B	GPS1 BATTERY LOW	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X4467B	GPS1 PPS-DAILY KEY IN USE VERIFIED	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4468B	GPS1 PPS-DAILY KEY IN USE UNVERIF	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4469B	GPS1 PPS-DAILY KEY IN USE INCORR	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4470B	GPS1 PPS-KEY FAILED PARITY	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4471B	GPS1 PPS-GUV USER	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X4472B	GPS1 PPS-2 HOUR ALERT	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4473B	GPS1 PPS-INSUFFICIENT KEYS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4474B	GPS1 PPS-RECEIVER CONTAINS KEYS	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4476B	GPS1 PPS-ERASE KEY FAIL	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1	EVENT	BD		93017G	*
	V74X4971B	GPS1 CHAN 1 FAULT FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X4972B	GPS1 CHAN 1 ANTENNA SELECT	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4973B	GPS1 CHAN 1 FREQUENCY	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4974B	GPS1 CHAN 1 EPHEM DOWNLOAD FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4982B	GPS1 CHAN 2 FAULT FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4983B	GPS1 CHAN 2 ANTENNA SELECT	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4984B	GPS1 CHAN 2 FREQUENCY	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X4985B	GPS1 CHAN 2 EPHEM DOWNLOAD FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4993B	GPS1 CHAN 3 FAULT FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4994B	GPS1 CHAN 3 ANTENNA SELECT	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4995B	GPS1 CHAN 3 FREQUENCY	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X4996B	GPS1 CHAN 3 EPHEM DOWNLOAD FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X5054B	GPS3 LOG WD1 CHILD	DL FMT 22/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X5055B	GPS3 LOG WD2 CHILD	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X5056B	GPS3 LOG WD3 CHILD	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1		BD		93017G	*
	V74X5057B	GPS3 LOG WD4 CHILD	DL FMT 60/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1		BD		93017G	*
	V74X5058B	GPS3 LOG WD5 CHILD	DL FMT 60/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1		BD		93017G	*
	V74X5059B	GPS3 LOG WD6 CHILD	DL FMT 60/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1		BD		93017G	*
	V74X5060B	GPS3 LOG WD7 CHILD	DL FMT 60/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1		BD		93017G	*
	V74X5061B	GPS3 LOG WD8 CHILD	DL FMT 60/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1		BD		93017G	*
	V74X5062B	GPS3 LOG WD9 CHILD	DL FMT 60/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1		BD		93017G	*
	V74X5063B	GPS3 LOG WD10 CHILD	DL FMT 60/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1		BD		93017G	*
	V74X5064B	GPS3 LOG WD11 CHILD	DL FMT 60/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1		BD		93017G	*
	V74X5065B	GPS3 LOG WD12 CHILD	DL FMT 60/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1		BD		93017G	*
	V74X5066B	GPS3 LOG WD13 CHILD	DL FMT 60/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X5067B	GPS3 LOG WD14 CHILD	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X5068B	GPS3 LOG WD15 CHILD	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X5069B	GPS3 LOG WD16 CHILD	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X5070B	GPS3 LOG WD17 CHILD	DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X5971B	GPS3 CHAN 1 FAULT FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X5972B	GPS3 CHAN 1 ANTENNA SELECT	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X5973B	GPS3 CHAN 1 FREQUENCY	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X5974B	GPS3 CHAN 1 EPHEM DOWNLOAD FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X5982B	GPS3 CHAN 2 FAULT FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X5983B	GPS3 CHAN 2 ANTENNA SELECT	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X5984B	GPS3 CHAN 2 FREQUENCY	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X5985B	GPS3 CHAN 2 EPHEM DOWNLOAD FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X5993B	GPS3 CHAN 3 FAULT FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X5994B	GPS3 CHAN 3 ANTENNA SELECT	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X5995B	GPS3 CHAN 3 FREQUENCY	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X5996B	GPS3 CHAN 3 EPHEM DOWNLOAD FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X8084B	GPS1 CHAN 4 FAULT FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X8085B	GPS1 CHAN 4 ANTENNA SELECT	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X8086B	GPS1 CHAN 4 FREQUENCY	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X8087B	GPS1 CHAN 4 EPHEM DOWNLOAD FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X8091B	GPS1 CHAN 5 FAULT FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X8092B	GPS1 CHAN 5 ANTENNA SELECT	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X8093B	GPS1 CHAN 5 FREQUENCY	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V74X8094B	GPS1 CHAN 5 EPHEM DOWNLOAD FLAG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V93U1001C	GPS2 TIME FIGURE OF MERIT	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V93U1187C	GPS2 MODE CMD	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V93U1199C	GPS2 MODE CMD CHECK	GPS2 TLM		BSU		91051K	
	V93U3001C	GPS1 TIME FIGURE OF MERIT	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V93U3187C	GPS1 MODE CMD	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V93U3199C	GPS1 MODE CMD CHECK	GPS1 TLM		BSU		92067D	
	V93U9001C	GPS3 TIME FIGURE OF MERIT	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V93U9187C	GPS3 MODE CMD	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BSU		93017G	*
	V93U9199C	GPS3 MODE CMD CHECK	GPS3 TLM		BSU		92067D	

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V93X1002X	GPS1 BIT IN PROGRESS SOP	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V93X1003X	GPS2 BIT IN PROGRESS SOP	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V93X1004X	GPS3 BIT IN PROGRESS SOP	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V93X1043X	GPS1 SAT DE-SELECT ENABLE	GPS1 TLM		BD		92276A	
	V93X1044X	GPS2 SAT DE-SELECT ENABLE	GPS2 TLM		BD		92276A	
	V93X1045X	GPS3 SAT DE-SELECT ENABLE	GPS3 TLM		BD		92276A	
	V93X1129X	GPS RA FLAG FDBK DL	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V93X1152X	GPS2 COMMFAULT DL	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V93X1188X	GPS2 ANNCD RESET CMD	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V93X1189X	GPS2 BARO AIDING DG	GPS2 DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V93X1190X	GPS2 RADALT AIDING DG	GPS2 DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V93X1191X	GPS2 DIFFERENTIAL GPS CMD	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1 GPS2		BD		93017G	*
	V93X1192X	GPS2 STATE AIDING DATA VALID	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1 GPS2		BD		93017G	*
	V93X1194X	GPS2 INIT STATE AVAIL	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1 GPS2		BD		93017G	*
	V93X1196X	GPS2 LEVER ARM AVAIL	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1 GPS2		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V93X1197X	GPS2 FILTER RESTART CMD	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V93X1198X	GPS2 ATT AIDING VALID	GPS2 TLM		BD		91051K	
	V93X1237X	GPS1 COMMFAULT DL	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V93X1238X	GPS3 COMMFAULT DL	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V93X3188X	GPS1 ANNC D RESET CMD	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V93X3189X	GPS1 BARO AIDING DG	GPS1 DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V93X3190X	GPS1 RADALT AIDING DG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1 GPS1		BD		93017G	*
	V93X3191X	GPS1 DIFFERENTIAL GPS CMD	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1 GPS1		BD		93017G	*
	V93X3192X	GPS1 STATE AIDING DATA VALID	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1 GPS1		BD		93017G	*
	V93X3194X	GPS1 INIT STATE AVAIL	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1 GPS1		BD		93017G	*
	V93X3196X	GPS1 LEVER ARM AVAIL	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1 GPS1		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V93X3197X	GPS1 FILTER RESTART CMD	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V93X3198X	GPS1 ATT AIDING VALID	GPS1 GPS1 TLM		BD		92067D	
	V93X3243X	GPS1 FAIL SELF-TEST DL	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1	EVENT	BD		93017G	*
	V93X3244X	GPS2 FAIL SELF-TEST DL	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1	EVENT	BD		93017G	*
	V93X3245X	GPS3 FAIL SELF-TEST DL	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1	EVENT	BD		93017G	*
	V93X9188X	GPS3 ANNCD RESET CMD	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1 GPS3		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V93X9189X	GPS3 BARO AIDING DG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1 GPS3		BD		93017G	*
	V93X9190X	GPS3 RADALT AIDING DG	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1 GPS3		BD		93017G	*
	V93X9191X	GPS3 DIFFERENTIAL GPS CMD	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1 GPS3		BD		93017G	*
	V93X9192X	GPS3 STATE AIDING DATA VALID	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1 GPS3		BD		93017G	*
	V93X9194X	GPS3 INIT STATE AVAIL	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1 GPS3		BD		93017G	*

TABLE 4.17.6.10-2. OUTPUT FUNCTIONAL PARAMETERS FROM GLOBAL POSITIONING SYSTEM SOP (G4.17)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V93X9196X	GPS3 LEVER ARM AVAIL	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1		BD		93017G	*
	V93X9197X	GPS3 FILTER RESTART CMD	DL FMT 21/1 DL FMT 22/1 DL FMT 23/1 DL FMT 32/1 DL FMT 44/1 DL FMT 45/1 DL FMT 60/1 GPS3		BD		93017G	*
	V93X9198X	GPS3 ATT AIDING VALID	GPS3 GPS3 TLM		BD		92067D	

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

**Rate Codes: (HDR Only) 1 = 1 sample/sec 2 = 5 samples/sec 3 = 12.5 samples/sec 4 = 25 samples/sec 5 = 100 samples/sec
(HDR and LDR) A = 1 sample/sec B = 5 samples/sec C = 12.5 samples/sec D = 25 samples/sec E = 100 samples/sec**

TABLE 4.17.6.10-3. GLOBAL POSITIONING SYSTEM SOP (G4.17) I-LOADS

FSSR NAME	MSID	ENG UNITS	DT	PR	D	S	PR FCTN	MC	CAT	LAST CR	R E V
ALT_BARO_CUTOFF_LOW	V99U8720C	FT	F	S	D	P	G4.127 G4.144 G4.17 G4.205	00	ZFT	91051K	
ALT_RA_CUTOFF_HIGH	V97U2827C	FT	F	S	D	P	G4.17 G6.21	00	GP3	92139	
ALT_RA_CUTOFF_LOW	V97U2828C	FT	F	S	D	P	G4.17	00	GP3	92139	
DELH_RA_GPS	V97U2829C	FT	F	S	D	P	G4.17	00	GP3	92139	
DT_FR_DISP_REFRESH	V97U3641C	SEC	F	S	D	P	G4.17	00	GPH	92102B	
FOM_GOOD_COUNT_MIN	V97U2238C	ND	I	S	D	C	G4.17	00	GPH	92430B	
GPC_GPS_TARGET	V97U3673C	MSEC	I	S	D	C	G4.17	00	GPH	92817C	
GPS_DR_EDIT_MAX_HOLD	V97U2815C	ND	D		D	P	G4.17	00	GPE	92630A	
GPS_DR_EDIT_MAX_STEP	V97U2816C	ND	I	S	D	P	G4.17	00	GPE	92630A	
GPS_DR_EDIT_STEP_SIZE	V97U2817C	ND	I	S	D	P	G4.17	00	GPE	92630A	
GPS_DR_EDIT_THRESHOLD	V97U2818C	ND	I	S	D	P	G4.17	00	GPE	92630A	
GPS_INSTALLED(1)	V97U2924C	ND	D		M	X	G4.17 G4.20 G4.21 G4.23 G4.24 G4.25 G6.11 G6.21 G6.31 G6.51 V5.1.4	00	GPI	92292B	
GPS_INSTALLED(2)	V97U2831C	ND	D		M	X	G4.17 G4.20 G4.21 G4.23 G4.24 G4.25 G6.11 G6.21 G6.31 G6.51 V5.1.4	00	GPI	92292B	

TABLE 4.17.6.10-3. GLOBAL POSITIONING SYSTEM SOP (G4.17) I-LOADS

FSSR NAME	MSID	ENG UNITS	DT	PR	D	S	PR FCTN	MC	CAT	LAST CR	R E V
GPS_INSTALLED(3)	V97U2925C	ND	D		M	X	G4.17 G4.20 G4.21 G4.23 G4.24 G4.25 G6.11 G6.21 G6.31 G6.51 V5.1.4	00	GPI	92292B	
GPS_LOWER_ANTENNA_BORESIGHT(1,1)	V97U2832C	ND	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_BORESIGHT(1,2)	V97U2833C	ND	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_BORESIGHT(1,3)	V97U2834C	ND	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_BORESIGHT(2,1)	V97U2835C	ND	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_BORESIGHT(2,2)	V97U2836C	ND	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_BORESIGHT(2,3)	V97U2837C	ND	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_BORESIGHT(3,1)	V97U2838C	ND	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_BORESIGHT(3,2)	V97U2839C	ND	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_BORESIGHT(3,3)	V97U2840C	ND	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_CG(1,1)	V97U2841C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_CG(1,2)	V97U2842C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_CG(1,3)	V97U2843C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_CG(2,1)	V97U2844C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_CG(2,2)	V97U2845C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_CG(2,3)	V97U2846C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_CG(3,1)	V97U2847C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_CG(3,2)	V97U2848C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_CG(3,3)	V97U2849C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_NB(1,1)	V97U2850C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_NB(1,2)	V97U2851C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_NB(1,3)	V97U2852C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_NB(2,1)	V97U2853C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_NB(2,2)	V97U2854C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_NB(2,3)	V97U2855C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_NB(3,1)	V97U2856C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_NB(3,2)	V97U2857C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_LOWER_ANTENNA_NB(3,3)	V97U2858C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_PR_EDIT_MAX_HOLD	V97U2811C	ND	D		D	P	G4.17	00	GPE	92630A	
GPS_PR_EDIT_MAX_STEP	V97U2812C	ND	I	S	D	P	G4.17	00	GPE	92630A	
GPS_PR_EDIT_STEP_SIZE	V97U2813C	ND	I	S	D	P	G4.17	00	GPE	92630A	
GPS_PR_EDIT_THRESHOLD	V97U2814C	ND	I	S	D	P	G4.17	00	GPE	92630A	
GPS_QA_TIME_THRESH	V97U3573C	SEC	F	S	D	P	G4.17	00	GPE	91051K	
GPS_UPPER_ANTENNA_BORESIGHT(1,1)	V97U2859C	ND	F	S	M	P	G4.17	00	GPH	92102B	
GPS_UPPER_ANTENNA_BORESIGHT(1,2)	V97U2860C	ND	F	S	M	P	G4.17	00	GPH	92102B	
GPS_UPPER_ANTENNA_BORESIGHT(1,3)	V97U2861C	ND	F	S	M	P	G4.17	00	GPH	92102B	
GPS_UPPER_ANTENNA_BORESIGHT(2,1)	V97U2862C	ND	F	S	M	P	G4.17	00	GPH	92102B	

TABLE 4.17.6.10-3. GLOBAL POSITIONING SYSTEM SOP (G4.17) I-LOADS

FSSR NAME	MSID	ENG UNITS	DT	PR	D	S	PR FCTN	MC	CAT	LAST CR	R E V
GPS_UPPER_ANTENNA_BORESIGHT(2,2)	V97U2863C	ND	F	S	M	P	G4.17	00	GPH	92102B	
GPS_UPPER_ANTENNA_BORESIGHT(2,3)	V97U2864C	ND	F	S	M	P	G4.17	00	GPH	92102B	
GPS_UPPER_ANTENNA_BORESIGHT(3,1)	V97U2865C	ND	F	S	M	P	G4.17	00	GPH	92102B	
GPS_UPPER_ANTENNA_BORESIGHT(3,2)	V97U2866C	ND	F	S	M	P	G4.17	00	GPH	92102B	
GPS_UPPER_ANTENNA_BORESIGHT(3,3)	V97U2867C	ND	F	S	M	P	G4.17	00	GPH	92102B	
GPS_UPPER_ANTENNA_CG(1,1)	V97U2868C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_UPPER_ANTENNA_CG(1,2)	V97U2869C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_UPPER_ANTENNA_CG(1,3)	V97U2870C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_UPPER_ANTENNA_CG(2,1)	V97U2871C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_UPPER_ANTENNA_CG(2,2)	V97U2872C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_UPPER_ANTENNA_CG(2,3)	V97U2873C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_UPPER_ANTENNA_CG(3,1)	V97U2874C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_UPPER_ANTENNA_CG(3,2)	V97U2875C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_UPPER_ANTENNA_CG(3,3)	V97U2876C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_UPPER_ANTENNA_NB(1,1)	V97U2884C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_UPPER_ANTENNA_NB(1,2)	V97U2885C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_UPPER_ANTENNA_NB(1,3)	V97U2886C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_UPPER_ANTENNA_NB(2,1)	V97U2887C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_UPPER_ANTENNA_NB(2,2)	V97U2888C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_UPPER_ANTENNA_NB(2,3)	V97U2889C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_UPPER_ANTENNA_NB(3,1)	V97U2890C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_UPPER_ANTENNA_NB(3,2)	V97U2891C	FT	F	S	M	P	G4.17	00	GPH	92102B	
GPS_UPPER_ANTENNA_NB(3,3)	V97U2892C	FT	F	S	M	P	G4.17	00	GPH	92102B	
LAST_READ_DT_LIM	V97U2239C	SEC	F	S	D	P	G4.17	00	GPH	92430B	
LEAP_SEC_CORR	V97U2130C	SEC	I	S	M	C	G4.17	00	GPT	92102B	
MAX_FOM_ACCEPT_PFS	V97U2830C	ND	I	S	D	P	G4.17	02	GP2	92369B	
								03	GP3		
								08	GP2		
								01	GP3		
								09	GP3		
M_WGS84_TO_EF(1,1)	V97U2927C	ND	F	D	M	P	G4.17 G4.20 G4.21 G4.220 G4.23 G4.237 G4.24 G4.25 V5.1.8	00	WGS	92067D	

TABLE 4.17.6.10-3. GLOBAL POSITIONING SYSTEM SOP (G4.17) I-LOADS

FSSR NAME	MSID	ENG UNITS	DT	PR	D	S	PR FCTN	MC	CAT	LAST CR	R E V
M_WGS84_TO_EF(1,2)	V97U2928C	ND	F	D	M	P	G4.17 G4.20 G4.21 G4.220 G4.23 G4.237 G4.24 G4.25 V5.1.8	00	WGS	92067D	
M_WGS84_TO_EF(1,3)	V97U2929C	ND	F	D	M	P	G4.17 G4.20 G4.21 G4.220 G4.23 G4.237 G4.24 G4.25 V5.1.8	00	WGS	92067D	
M_WGS84_TO_EF(2,1)	V97U2930C	ND	F	D	M	P	G4.17 G4.20 G4.21 G4.220 G4.23 G4.237 G4.24 G4.25 V5.1.8	00	WGS	92067D	
M_WGS84_TO_EF(2,2)	V97U2931C	ND	F	D	M	P	G4.17 G4.20 G4.21 G4.220 G4.23 G4.237 G4.24 G4.25 V5.1.8	00	WGS	92067D	
M_WGS84_TO_EF(2,3)	V97U2932C	ND	F	D	M	P	G4.17 G4.20 G4.21 G4.220 G4.23 G4.237 G4.24 G4.25 V5.1.8	00	WGS	92067D	

TABLE 4.17.6.10-3. GLOBAL POSITIONING SYSTEM SOP (G4.17) I-LOADS

FSSR NAME	MSID	ENG UNITS	DT	PR	D	S	PR FCTN	MC	CAT	LAST CR	R E V
M_WGS84_TO_EF(3,1)	V97U2933C	ND	F	D	M	P	G4.17 G4.20 G4.21 G4.220 G4.23 G4.237 G4.24 G4.25 V5.1.8	00	WGS	92067D	
M_WGS84_TO_EF(3,2)	V97U2934C	ND	F	D	M	P	G4.17 G4.20 G4.21 G4.220 G4.23 G4.237 G4.24 G4.25 V5.1.8	00	WGS	92067D	
M_WGS84_TO_EF(3,3)	V97U2935C	ND	F	D	M	P	G4.17 G4.20 G4.21 G4.220 G4.23 G4.237 G4.24 G4.25 V5.1.8	00	WGS	92067D	
RV_MAX	V99U7596C	FT/SEC	F	S	D	C	G4.15 G4.17 G4.205	00	ZFT	92080B	
SF_ANT	V97U3549C	ND	F	S	L	P	G4.17	00	GPH	92102B	
SF_ATT	V97U2911C	DEG	F	S	L	P	G4.17	00	GPH	92102B	
SF_BARO	V97U2912C	ND	F	S	L	P	G4.17	00	GPH	92102B	
SF_BORE	V97U3550C	ND	F	S	L	P	G4.17	00	GPH	92102B	
SF_POS	V97U2914C	ND	F	D	L	P	G4.17 G4.21	00	GPH	92067D	
SF_RA	V97U2915C	ND	F	S	L	P	G4.17	00	GPH	92102B	
SF_TIME	V97U2918C	ND	F	D	L	P	G4.17	00	GPH	92067D	
SF_VEL	V97U2919C	ND	F	S	L	P	G4.17 G4.21	00	GPH	92067D	
T_GPS_OFFSET_DAYS	V97U2131C	DAYS	I	S	M	C	G4.17	00	GPT	92102B	
T_RA_GPS_DELAY	V97U2922C	SEC	F	S	D	P	G4.17	00	GP3	92139	

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

TABLE 4.17.6.10-4. GLOBAL POSITIONING SYSTEM SOP (G4.17) K-LOADS

FSSR NAME DESCRIPTION	MSID	MC	KLOAD VALUE	ENG UNITS	DT	P R	S	PR FCTN	LAST CR	R E V
GPS_SW_CAP MAX NUM OF GPS LRU THAT SOFTWARE CAN HANDLE	V97U2923C	00	+3	ND	I		S P	G4.17 G4.18 G4.20 G4.21 G4.23 G4.24 G4.25	92069B	
TIME_BIAS_BARO PAD BIAS CORR BARO TIME TAG	V97U5544C	00	+1.790	E-01 SEC	F		D C	G4.127 G4.144 G4.17 G4.205	92069B	
TIME_BIAS_RADALT PAD LOAD CORR RAD ALT TIME	V97U5546C	00	+1.50	E-02 SEC	F		D P	G4.17 G4.20 G4.23 G4.24 G4.25	92069B	

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

TABLE 4.17.6.10-5. GLOBAL POSITIONING SYSTEM SOP (G4.17) CONSTANTS

FSSR NAME DESCRIPTION	MSID	MC	CONSTANT VALUE	ENG UNITS	DT	P R	S	PR FCTN	LAST CR	R E V
EARTH_RATE EARTH'S ROTATIONAL RATE	V97U4379C	00	+7.29211514646E-05	RAD/SEC	F		D C	A6.9 G4.126 G4.127 G4.139 G4.144 G4.148 G4.15 G4.17 G4.19 G4.20 G4.205 G4.21 G4.213 G4.22 G4.236 G4.237 G4.4 G5.26	93090E	

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

4.43 TACAN SUBSYSTEM OPERATING PROGRAM

This section defines the detailed functional requirements and formulations for the tactical air navigation (TACAN) subsystem operating program (SOP). The TACAN SOP defines software associated with TACAN range and bearing data processing, auto station command and response processing, upper/lower antenna selection, and self-test processing.

4.43.1 TACAN Requirements Overview

4.43.1.1 TACAN Functional Overview

TACAN is a military L-band air navigation system that provides slant range and magnetic bearing information from user aircraft to ground beacon. The desired ground beacon is selected by choosing a channel number (specific frequency). Although each ground beacon can be uniquely identified by its broadcasted Morse code ID tone, channel numbers can be ambiguous. However, the intent of the orbiter design is to use functional or real clear-channel ground beacons. A single TACAN subsystem is depicted in Figure 4.43.1.1-1 along with related interfacing equipment. Six major components are identified that are required for the TACAN function: controls on the panel, an MDM (multiplexer/demultiplexer) that contains the TACAN control register, a receiver-transmitter (R/T), an upper antenna, a lower antenna, and the on-board computer. The keyboard and the display electronics unit are also required for TACAN self-test command and data monitoring.

TACAN Control Head. Ultimate control of the TACAN subsystem is via the control head. The control head has a four-position function switch and four thumbwheels for channel selection. The four positions of the function switch are as follows: OFF, RECEIVE, TRANSMIT/RECEIVE, and GPC.

Power is applied to the R/T when the switch is in any position except OFF. A power-present discrete returns from the R/T to the GPC. When in RECEIVE or TRANSMIT/RECEIVE positions, the channel number is determined by the thumbwheel positions. Channel numbers 1 through 126 are recognized by the R/T. The fourth thumbwheel determines the mode of the ground beacon (X or Y) that completes the channel selection definition. When in RECEIVE mode, the R/T outputs only bearing data. When in TRANSMIT/RECEIVE mode, the R/T outputs both bearing and range data. When the switch is in the GPC position, the channel number, mode, and self-test state are determined by the GPC.

Antenna Control Switch. Antenna selection may be accomplished by allowing the GPC to automatically switch antennas when the antenna switch is in the AUTO position. The auto function may be inhibited by choosing the UPPER or LOWER switch position. The R/T will cause selection of the upper or lower antenna according to switch position.

TACAN Control Word. Mechanization of the command to the R/T, whether it be from the control head or the software, is via the TACAN control register in the MDM. Once every 32 milliseconds, the contents of the TACAN control register are transferred to an output register and are then sent in a serialdigital control word to the TACAN R/T. The format of the control register is shown in Figure 4.43.1.1.2-1. The TACAN discrete input register contains the control head switch settings with one exception, the four-position function switch GPC mode. The format of the discrete input word is shown in Figure 4.43.1.1.2-2.

When the mode switch is set to RECEIVE or TRANSMIT/RECEIVE, the contents of the TACAN discrete input register are converted to control register format and transferred to the TACAN control register. This takes place during the 32-millisecond cycle immediately prior to transfer of the TACAN control register contents to the output register and takes precedence over any control words transmitted from the GPC to the MDM.

When the mode switch is set to GPC, the control register update from the discrete input register does not take place. A control register command sent from the TACAN SOP is then permitted to be transferred to the output register and to command the TACAN T/R.

When read by the TACAN SOP, the contents of the TACAN control register depend upon setting of the mode switch, and upon previous TACAN SOP writes to the control register. When the mode switch is set to GPC, the R/T control word contains the most recent control word update from the TACAN SOP since the start of GPC mode. If the TACAN SOP has not updated the control word since the start of the GPC mode, the R/T control word will contain the last control head switch settings from manual mode. When the mode setting is RECEIVE or TRANSMIT/RECEIVE, the R/T control word read by the TACAN SOP will contain the current control head switch settings unless a control word was written from the TACAN SOP to the MDM in the previous 32 milliseconds. Under this condition, the contents of the R/T control word read by the TACAN SOP will be unpredictable.

TACAN Self-Test. Self-test of the TACAN R/T is initiated through the keyboard or directly by the software. Self-test of the R/T can be executed properly only if the control head mode switch is in GPC position and remains in GPC for the duration of the test. Upon completion of self-test, the TACAN outputs range, bearing, and status discrete to indicate the health of the R/T.

4.43.1.1.1 TACAN Operating Modes

The TACAN system operating modes are as follows:

- OFF mode
- RECEIVE mode
- TRANSMIT/RECEIVE mode
- GPC mode

OFF mode occurs when the power-on discrete from the TACAN transceiver is in the false or off state.

RECEIVE mode is not used during orbiter operations.

TRANSMIT/RECEIVE mode occurs when the transceiver is in operation as indicated by the power-on discrete and the control switch is in T/R position. The channel selection is controlled by thumbwheels on the control panel.

GPC mode occurs when the transceiver is in operation as indicated by the power-on discrete, and the control switch is in the GPC position. The GPC controls channel selection and self-test operation.

4.43.1.1.2 TACAN Digital Interfaces

The TACAN interfaces with the GPC via the serial digital multiplex/demultiplex input and output data channel. The digital I/O words are 16 bits in size, plus 3 bits for synchronization and 1 bit for odd parity. Throughout the TACAN SOP, the synchronization and parity bits are disregarded, and the most significant data bit in the word is designated as bit zero with the least significant bit being bit 15. The contents, format, and bit values of the TACAN data interfaces are illustrated in:

Figure 4.43.1.1.2-1, TACAN Control Register Format

Figure 4.43.1.1.2-2, TACAN Discrete Word Format

Figure 4.43.1.1.2-3, TACAN to MDM Range Word

Figure 4.43.1.1.2-4, MDM to GPC Range Words

Figure 4.43.1.1.2-5, TACAN to MDM Bearing Word

Figure 4.43.1.1.2-6, MDM to GPC Bearing Word

4.43.1.1.3 TACAN Discrete Interfaces

TACAN provides two discretes to GPC software: power-on discrete and test status monitor discrete.

The power-on discrete is an indication to the GPC that the transceiver power supply is operating. The test status monitor discrete provides an indication of the health of part of the transceiver and is used in conjunction with self-test for fault detection. The test status monitor discrete may change state only when self-test is active.

4.43.1.2 TACAN SOP Functional Overview

The TACAN SOP controls and monitors the operation of the three TACAN systems and processes TACAN data for use by other program elements. The inputs and outputs from the TACAN SOP modules are stored in locations labeled as to parameter and transceiver.

The TACAN SOP is partitioned into four modules which are as follows:

- A. TACAN Self-Test Processing. The TAC_TEST module initiates TACAN self-test and monitors self-test outputs to determine the health of the TACAN and MDM data link.
- B. TACAN Data Processing. The TAC_DATA module processes TACAN range, bearing, and channel data.
- C. TACAN Antenna Selection Processing. The TAC_ANT module performs the processing associated with TACAN upper or lower antenna selection.
- D. TACAN Channel Command Processing. The TAC_CHAN module generates the TACAN control word in support of navigation-controlled TACAN channel selection.

4.43.1.3 TACAN Functional Module Summary

The functional modules that constitute the TACAN SOP are identified in Table 4.43.1.3–1. Each of the modules, including functions and formulations are presented in detail in Section 4.43.6.

4.43.1.4 Reference Documents

MC409–0014 Receiver/Transmitter, Air Navigation Set, Tactical; Rockwell

MC409–0184 Receiver/Transmitter, Air Navigation Set, Tactical; Rockwell

4.43.1.5 Software Protection

The software implementation of the TACAN SOP shall protect against program checks and error interrupts caused by transient–type errors or bad data inputs. Specifically, the TACAN SOP shall protect against all of the error conditions identified in paragraph 3.6 that are applicable to the TACAN software design and code.

4.43.2 Definitions

See the glossary in Section 5.1.

4.43.3 Functional Requirements Traceability

The detailed requirements and formulations contained herein satisfy the GN&C, Level B, CPDS functional requirements for the TACAN SOP.

4.43.4 TACAN SOP Interfaces

This section defines the interfaces associated with the TACAN SOP. Table 4.43.4–1 contains the TACAN SOP inputs and outputs.

4.43.5 Parameter Initialization

This section defines the I–load parameters associated with the TACAN SOP. Table 4.43.5–1 contains the I–load parameters and identifies typical I–load values. The actual values will be specified later in the official I–load document. The codes designated in the constant column of Table 4.43.5–1 are to be interpreted as follows: “SX” indicates a soft constant that may change as the software is refined; “MX” indicates a mission/LRU dependent constant that is expected to change from mission to mission; and “X” indicates a hard constant that probably will not change.

4.43.6 Detailed Requirements and Formulations

This section defines the TACAN SOP functional modules in sufficient detail to create a functional design and generate computer code. Figure 4.43.6–1 presents a flow diagram of the TACAN SOP. Each module defines the functional requirements, interface requirements, detailed requirements and formulations, and special considerations if applicable. A flow diagram is also included for most modules; however, the flow diagrams are supplementary material and are not intended to convey requirements or to place restrictions on functional design or coding.

The TACAN SOP shall function as follows:

1. TACAN Time Tag. A homogeneous set of TACAN data shall be read into the GPC prior to each execution of the TAC_DATA module. These data shall consist of range, bearing, control register data, discrete register data, and the control switch discrete for each of the three TACAN's. The TACAN data shall be time-tagged with a single tag that corresponds to the time associated with the input transaction for the three-TACAN set. Whenever the TAC_DATA module is executed in support of TACAN RM function (0.25-Hz cycle), the current TACAN time tag shall be stored for use by navigation.
2. Commfault/Power Check. For each LRU (1, 2, and 3), determine whether or not a commfault exists. If a commfault does not exist, then set the TACAN power-on signal, TACPON_i, equal to the TACAN power discrete, TACPOD_i. If TACPON_i is false, then set the range and bearing comlocks (TRNG_CMLK_i and TBRG_CMLK_i) to false; otherwise, continue SOP execution for that LRU.

If a commfault does exist for an LRU, set the range and bearing comlocks for that LRU (TRNG_CMLK_i and TBRG_CMLK_i) to false and terminate processing on that LRU on that cycle. SOP processing for the remaining LRU or LRU's will continue.

4.43.6.1 TACAN Self-Test Processing (TAC_TEST)

The purpose of the TAC_TEST functional module is to initiate TACAN self-test and to monitor the TACAN self-test outputs to determine the health of the TACAN and MDM data link.

A. Functional Requirements. The TAC_TEST functional module is required to perform the following functions:

1. Set the TACAN self-test bit to the required state in response to a self-test request from the TACAN RM function.
2. Monitor the TACAN range and bearing data on completion of self-test. Limit check the data to detect failures.
3. Monitor the TACAN test status monitor discrettes to detect failures.
4. Provide range and bearing self-test fail flags for use by TACAN RM.
5. Provide a self-test in progress flag for use by TACAN RM.

B. Interface Requirements. The input and output parameters associated with the TAC_TEST functional modules are contained in Table 4.43.4-1. I-load parameters are contained in Table 4.43.5-1.

C. Detailed Requirements and Formulations. The TAC_TEST module shall perform the following operations in response to a self-test request from the TACAN RM function. Figure 4.43.6.1-1 presents a flow diagram of the TAC_TEST functional requirements.

1. Self-Test Initiation. Upon receipt of a self-test request signal from TACAN RM, determine if all TACAN's to be tested are in the GPC mode. If so, set the self-test-in-progress flag to logic 1, set all TACAN self-test failure flags to logic 0, and initiate TACAN self-test on the requested TACAN's by setting bit 15 of the applicable TACAN control words to logic 1 for output to the applicable MDM control registers via the TAC_CHAN module. Self-test shall be initiated within 1 second of the TACAN RM request for self-test. If any of the requested TACAN's are not in the GPC mode, set the range and bearing fail flags for the requested TACAN's to logic 1 (fail), set the self-test-in-progress flag to logic 0, and set all bits in the self-test request signal to logic 0.
2. Self-Test Monitor. The following operations shall be performed for each TACAN that has been commanded to self-test. After TAC_STTIM seconds have elapsed, the range and bearing data from the applicable TACAN's shall be read and limit checked to the following I-load limits:

	<u>Low Limit</u>	<u>High Limit</u>
Range (A B)	TAC_RNGLO	TAC_RNGHI
Bearing	TAC_BEARLO	TAC_BEARHI

In addition, the applicable TACAN test status monitor discretess shall be checked. If either the TACAN range limit check fails or the test status monitor discrete indicates a failure (logic 0), the applicable range self-test failure flag shall be set to logic 1. If either the TACAN bearing limit check fails or the test status monitor discrete indicates a failure, the applicable bearing self-test failure flag shall be set to logic 1. Note that the TACAN self-test fail flags must be initialized to logic 0 whenever the TACAN SOP is first invoked.

3. Self-Test Completion. Upon completion of self-test, the self-test-in-progress flag shall be set to logic 0, the self-test bits in all TACAN control words shall be set to logic 0, and all bits in the self-test request signal shall be set to logic 0.

D. Special Considerations. The following special considerations are applicable to TACAN self-test operation.

1. In order for the TACAN SOP to initiate TACAN self-test, the TACAN control head switches on the overhead panel in the cockpit must be set to the GPC position. This enables the GPC to output the self-test command word to the TACAN control registers in the MDM.

4.43.6.2 TACAN Data Processing (TAC_DATA)

The purpose of the TAC_DATA functional module is to process TACAN input data.

A. Functional Requirements. The TAC_DATA module is required to perform the following functions:

1. Time tag TACAN input data for use by navigation.
2. Compute the TACAN range values by concatenating the two range words and performing the required scaling and conversion to obtain floating point data in units of feet.

3. Compute the TACAN bearing values by performing the required scaling and conversion to obtain floating point data in units of radians.
4. Set the TACAN range and bearing lock-on flags and commfault/lock-on flags to the correct state.
5. Compute the current TACAN channel identification for use by other software functions.

B. Interface Requirements. The input and output parameters associated with the TAC_DATA functional module are contained in Table 4.43.4-1. I-load parameters are contained in Table 4.43.5-1.

C. Detailed Requirements and Formulation. The TAC_DATA module shall perform the following operations whenever the TACAN SOP is active. This module shall be executed prior to TACAN RM processing at a rate of 1.04 Hz. The TAC_DATA module shall also be executed once every 320 milliseconds (3.125 Hz) prior to area navigation processing if the HSI source and receiver switches (left or right) indicate TACAN. All three TACAN's shall be processed prior to TACAN RM processing. Only the HSI-selected TACAN's are required to be processed in support of area navigation. Figure 4.43.6.2-1 presents a flow diagram of the TAC_DATA functional requirements.

1. TACAN Range. Compute the TACAN range parameter, TACRANGE_i, as follows:
 - a. Concatenate input parameters TACRANGA_i and TACRANGB_i into a single 32-bit word as shown in Figure 4.43.1.1.2-4. Extract bits 2 through 19 and convert these bits from BCD to fixed-point notation using the following scaling:
 - Bits 2 through 3 = 100's nmi
 - Bits 4 through 7 = 10's nmi
 - Bits 8 through 11 = 1's nmi
 - Bits 12 through 15 = 1/10's nmi
 - Bits 16 through 19 = 1/100's nmi
 - b. Note that input values up to 400 nmi are to be accommodated. The range value shall then be converted from nautical miles to feet using the conversion factor TNM_FT, changed to floating point, and made available for use by other software.
2. TACAN Range Lock-On and Comlock. If bits 0 and 1 of TACRANGA_i are both zero; set the TACAN range lock-on flag, TACRANGDG_i, and TRNG_CMLK_i to true; else, set TACRANGDG_i and TRNG_CMLK_i to false.
3. TACAN Bearing. Compute the TACAN bearing parameter, TACBEAR_O_i, as follows:
 - a. Extract bits 3 through 14 of TACBEARA_i as shown in Figure 4.43.1.1.2-6. Note that TACBEARA_i binary data is scaled in degrees with bit 3 (MSB) equal to 90 degrees. If bits 1 and 2 of TACBEARA_i equal binary "11," then increase the value of the bearing data by 180 degrees.
 - b. The bearing value shall then be converted from degrees to radians, changed to floating point, and made available for use by other software.

4. TACAN Bearing Lock-On and Comlock. If bits 1 and 2 of TACBEARA_i are either binary “00” or binary “11,” set the TACAN bearing lock-on flag, TACBEARDG_i, and TBRG_CMLK_i to true; otherwise, set TACBEARDG_i and TBRG_CMLK_i to false.
5. TACAN Channel Word. Compute the TACAN channel word, TACCHAN_i, as follows:
 - a. If in GPC mode (TAC_GPC_i = 1), or if GPC input/output design precludes reading of a TACAN transceiver within 34 milliseconds of a write to the same transceiver, then extract bits 7 through 14 of the TACAN control word, TACCFM_i, and convert bits to fixed-point notation in accordance with the following bit scaling (refer to Figure 4.43.1.1.2-1):

Bits 7 through 10 = units (max = 9)

Bits 11 through 14 = tens (max = 120)

Otherwise, proceed to step 6b.
 - b. If in manual mode (TAC_GPC_i = 0) and the above input/output constraint is not met, then extract bits 0 and 7 through 14 of the TACAN discrete word, TAC_DW_i, and convert these bits to fixed-point notation in accordance with the following bit scaling (refer to FIGURE 4.43.1.1.2-2):

Bits 7 through 10 = units (max = 9)

Bits 11 through 14 = tens (max = 90)

Bit 0 = one hundred
 - c. If bit 2 of TACCFM_i (or TAC_DW_i) is true, indicating the TACAN is operating in the Y mode, set TACCHAN_i negative; otherwise, set TACCHAN_i positive.
 - d. If TACCHAN_i has changed since the previous cycle, set the TACAN channel change indicator, TCH_CHNG_i, equal to true; otherwise, set TCH_CHNG_i equal to false.

4.43.6.3 TACAN Antenna Selection Processing (TAC_ANT)

The purpose of the TAC_ANT functional module is to perform the processing associated with TACAN upper or lower antenna selection.

A. Functional Requirements. The TAC_ANT module is required to perform the following functions for each TACAN:

1. Select lower antenna when a new TACAN channel is commanded and call SITE LOOKUP with appropriate LRU_ID and NEW_STATION_ID.
2. Monitor the range lock-on bits to determine if range lock-on has been achieved.
3. If the TACAN has not achieved range lock-on, delay several seconds and then switch to the opposite antenna in an attempt to obtain range lock-on.

4. Set the antenna selection discrettes and the antenna inhibit discrettes to the proper state, and output these discrettes to the TACAN.

B. Interface Requirements. The input and output parameters associated with the TAC_ANT functional module are contained in Table 4.43.4–1. I-load parameters are contained in Table 4.43.5–1.

C. Detailed Requirements and Formulation. The following operations shall be performed by the TAC_ANT module whenever the TACAN SOP is active. Figure 4.43.6.3–1 presents a flow diagram of the TAC_ANT functional requirements.

1. Perform the following operations for each TACAN at a rate of once per second:
 - a. If TACCHAN_i has changed since the last cycle, set the TACANSEL_i discrete to logic 0 (lower) for TACAN_i; also set the range lock on timer, RL_TIMER_i, to zero for TACAN_i. Then perform steps e and g.
 - b. If TACAN_i currently has range lock-on (TACRANGA_i bits 0 and 1 are both zero), reset the TACAN range lock-on timer, RL_TIMER_i, to 0; otherwise continue with step c
 - c. Increment the range lock-on timer, RL_TIMER_i, by 1 second.
 - d. If the range lock-on timer, RL_TIMER_i, is greater than or equal to TAC_RLTIM seconds, change the state of the upper/lower antenna discrete, TACANTSEL_i, from 1 to 0 (lower) or 0 to 1 (upper) as applicable.
 - e. Set the TACAN antenna selection inhibit discrete, TACANTINH_i, to logic 1. Note that this discrete can be initialized to logic 1 when the TACAN SOP is first invoked and maintained at logic 1 for all TACAN SOP processing.
 - f. Set the range lock-on timer, RL_TIMER_i, to 0.
 - g. Output the TACANTSEL_i and TACANTINH_i discrettes within 0.5 second.
2. TACAN_i antenna selection processing is not required during TACAN_i self-test (TAC_TEST) processing.

D. Special Considerations. The following special considerations are applicable to TACAN SOP antenna selection.

1. During normal utilization of the TACAN's, the TACAN SOP will determine the proper selection of either the upper or lower antenna based upon achieving range lock-on within a predefined time interval. In order for the SOP to actually control the antenna selection, the TACAN antenna control switches on the overhead panel in the cockpit must be set at the auto position. This enables the GPC to control the antenna inhibit discrettes and the upper/lower selection discrettes.
2. The crew can override the TACAN SOP auto antenna selection by manually placing the TACAN antenna control switches in either the upper or lower antenna position. The SOP antenna selection processing will continue; however, the outputs from the GPC will be inhibited, and antenna selection will be in accordance with the antenna control switch position.

4.43.6.4 TACAN Channel Command Processing (TAC_CHAN)

The purpose of the TAC_CHAN functional module is to generate the control word output to the TACAN's in support of GPC-controlled TACAN channel selection. Note that TACAN channel selection is performed by the navigation function.

A. Functional Requirements. The TAC_CHAN functional module is required to perform the following functions:

1. Convert the TACAN channel number selected by navigation into the required format for output to the TACAN control registers.
2. Format the TACAN control word to specify the proper channel, mode, and test condition and output the control word to the TACAN control registers.

B. Interface Requirements. The input and output parameters associated with the TAC_CHAN functional module are contained in Table 4.43.4-1.

C. Detailed Requirements and Formulations. The following operations shall be performed by the TAC_CHAN module whenever the TACAN SOP is active. Figure 4.43.6.4-1 presents a flow diagram of the TAC_CHAN functional requirements.

1. Perform the following operations for each TACAN that is in GPC mode at a rate of once per second:
 - a. To command TACAN channel selection from the GPC, a 16-bit control word shall be generated and output to the MDM control register associated with the respective TACAN LRU. As defined in Figure 4.43.1.1.2-1, the bit configuration of the control word shall be as follows:

(Bit 0) 1 OMO 100 XXX XYY YYZ (Bit 15)

Where M is a mode select bit (X/Y), XXXX (units) and YYYY (tens) represent the desired channel number requested by the navigation function, and Z represents the self-test command bit (logic state controlled by the TAC_TEST module). Note that the Y-mode channels are represented by negative channel numbers in the navigation and display functions. Other bits in the control word correspond to the following:

Function = T/R

- b. Output the control word to the MDM within 0.5 second.
2. TAC_CHAN processing shall follow the channel selection processing in navigation.

D. Special Considerations. The following special considerations are applicable to GPC-controlled TACAN channel selection.

1. During normal utilization of the TACAN's, the navigation function and the TACAN SOP will select the TACAN channels based upon factors such as available stations, current state vector, flight path geometry, etc. In order for the SOP to actually control the channel section, the TACAN control head switches on the overhead panel in the cockpit must be set to the GPC

position. This enables the GPC to output the selected channel number to the TACAN control register in the MDM.

2. The crew can override the GPC-controlled TACAN channel selection by placing the TACAN control head switches in the T/R positions. This enables the crew to manually select the TACAN channel via the thumb-wheel switches. It should be noted that manual channel selection may be incompatible with TACAN selection filter processing and redundancy management.

4.43.6.5 Applicable Tables for TACAN SOP

The IDD inputs and outputs are listed in Tables 4.43.6.5-1 and 4.43.6.5-2. Values for the I-loads are contained in the I-loads requirements document (JSC-19350); however, I-load definitions applicable to this principal function are listed in Table 4.43.6.5-3. K-loads are listed in Table 4.43.6.5-4. Constants are listed in Table 4.43.6.5-5.

Table 4.43.1.3-1. TACAN SOP Functional Modules		
Module Name	Applicable Paragraph	Module Description
TAC_TEST	4.43.6.1	TACAN self-test
TAC_DATA	4.43.6.2	TACAN data processing
TAC_ANT	4.43.6.3	TACAN antenna selection
TAC_CHAN	4.43.6.4	TACAN channel command

Table 4.43.4-1.(1) TACAN SOP Interface Table

INPUT							
Name	MSID	Description	Source	Signal Type	Range	Units	Rate (Hz)
TACRANGA(1)***	V74M1550P	TACAN NO. 1 RANGE WORD A	TACAN 1	FIXED PT	0 – 400	NM	3.125
TACRANGA(2)***	V74M1650P	TACAN NO. 2 RANGE WORD A	TACAN 2	FIXED PT	0 – 400	NM	3.125
TACRANGA(3)***	V74M1750P	TACAN NO. 3 RANGE WORD A	TACAN 3	FIXED PT	0 – 400	NM	3.125
TACRANGB(1)***	V74M1551P	TACAN NO. 1 RANGE WORD B	TACAN 1	FIXED PT	0 – 400	NM	3.125
TACRANGB(2)***	V74M1651P	TACAN NO. 2 RANGE WORD B	TACAN 2	FIXED PT	0 – 400	NM	3.125
TACRANGB(3)***	V74M1751P	TACAN NO. 3 RANGE WORD B	TACAN 3	FIXED PT	0 – 400	NM	3.125
TACBEARA(1)***	V74M1530P	TACAN NO. 1 BEARING WORD A	TACAN 1	FIXED PT	0 – 360	DEG	3.125
TACBEARA(2)***	V74M1630P	TACAN NO. 2 BEARING WORD A	TACAN 2	FIXED PT	0 – 360	DEG	3.125
TACBEARA(3)***	V74M1730P	TACAN NO. 3 BEARING WORD A	TACAN 3	FIXED PT	0 – 360	DEG	3.125
TACBEARB(1)***	V74M1531P	TACAN NO. 1 BEARING WORD B	TACAN 1	FIXED PT	0 – 360	DEG	3.125
TACBEARB(2)***	V74M1631P	TACAN NO. 2 BEARING WORD B	TACAN 2	FIXED PT	0 – 360	DEG	3.125
TACBEARB(3)***	V74M1731P	TACAN NO. 3 BEARING WORD B	TACAN 3	FIXED PT	0 – 360	DEG	3.125
TACCFM(1)	V72M6598P	MDM FF01 DSCR IN CD00 CH6	HWDR CONT REG NO. 1	FIXED PT	NA	NA	3.125
TACCFM(2)	V72M6648P	MDM FF02 DSCR IN CD00 CH6	HWDR CONT REG NO. 2	FIXED PT	NA	NA	3.125
TACCFM(3)	V72M6698P	MDM FF03 DSCR IN CD00 CH6	HWDR CONT REG NO. 3	FIXED PT	NA	NA	3.125
TAC_TSM(1)	V74X0070X	TACAN NO. 1 TEST STATUS MONITOR	TACAN 1	DISCRETE	0,1	NA	3.125
TAC_TSM(2)	V74X0080X	TACAN NO. 2 TEST STATUS MONITOR	TACAN 2	DISCRETE	0,1	NA	3.125
TAC_TSM(3)	V74X0090X	TACAN NO. 3 TEST STATUS MONITOR	TACAN 3	DISCRETE	0,1	NA	3.125
TAC_GPC(1)	V72K8410X	TACAN NO. 1 AUTO DISCRETE (CMPTR GPC=LOGIC 1)	TACAN 1 D&C/ MDM	DISCRETE	0,1	NA	3.125
TAC_GPC(2)	V72K8430X	TACAN NO. 2 AUTO DISCRETE (CMPTR GPC=LOGIC 1)	TACAN 2 D&C/ MDM	DISCRETE	0,1	NA	3.125
TAC_GPC(3)	V72K8450X	TACAN NO. 3 AUTO DISCRETE (CMPTR GPC=LOGIC 1)	TACAN 3 D&C/ MDM	DISCRETE	0,1	NA	3.125
TACPOD(1)	V74X0071X	TACAN NO. 1 POWER STATUS	TACAN 1	DISCRETE	0,1	NA	3.125
TACPOD(2)	V74X0081X	TACAN NO. 2 POWER STATUS	TACAN 2	DISCRETE	0,1	NA	3.125
TACPOD(3)	V74X0091X	TACAN NO. 3 POWER STATUS	TACAN 3	DISCRETE	0,1	NA	3.125

Table 4.43.4-1.(1) TACAN SOP Interface Table

INPUT							
Name	MSID	Description	Source	Signal Type	Range	Units	Rate (Hz)
TAC_DW(1)	V72M6599P	MDM FF01 DSCR IN CD00 CH5	HWDR DISC REG NO. 1	FIXED PT	NA	NA	3.125
TAC_DW(2)	V72M6649P	MDM FF02 DSCR IN CD00 CH5	HWDR DISC REG NO. 2	FIXED PT	NA	NA	3.125
TAC_DW(3)	V72M6699P	MDM FF03 DSCR IN CD00 CH5	HWDR DISC REG NO. 3	FIXED PT	NA	NA	3.125
RCVR_LEFT*	V96J2105C	LH HSI RECEIVER SELECT	SW RM 4.180	INTEGER	1, 2, 3	NA	3.125
RCVR_RIGHT*	V96J2205C	RH HSI RECEIVER SELECT	SW RM 4.180	INTEGER	1, 2, 3	NA	3.125
SOURCE_LEFT*	V96X2111X	LH HSI SOURCE SELECT	SW RM 4.180	LOGICAL	0,1	NA	3.125
SOURCE_RIGHT*	V96X2211X	RH HSI SOURCE SELECT	SW RM 4.180	LOGICAL	0,1	NA	3.125
AUTO_CHANNEL_ID	V95J5022C	TACAN CHANNEL SELECTION	ASCENT ENTRY SITE LOOKUP 4.124/ 4.238	FIXED PT	+/-126	NA	1.04
TAC_STREQ	V90J3602C	TACAN SELF TEST REQUEST	TACAN RM 4.77	BITS	3 BITS	NA	OD
TAC_CF(1)	V91X3500X	TACAN 1 COMMFAULT	FCOS	LOGICAL	0,1	NA	3.125
TAC_CF(2)	V91X3510X	TACAN 2 COMMFAULT	FCOS	LOGICAL	0,1	NA	3.125
TAC_CF(3)	V91X3520X	TACAN 3 COMMFAULT	FCOS	LOGICAL	0,1	NA	3.125
CLOCK	V91W5000C	CLOCK-COMPUTER (GMT)	FCOS	FLOAT PT	NA	SEC	3.125

* These parameters are not directly used in TACAN SOP.

** This time tag is not used anywhere in TACAN SOP.

***See Figure 4.43.1.1.2-3 and 4.43.1.1.2-5 for contents of this parent word.

Table 4.43.4-1.(2) TACAN SOP Interface Table

OUTPUT							
Name	MSID	Description	Destination	Signal Type	Range	Units	Rate (Hz)
TACRANGE0(1)	V95H5102C	TACAN NO. 1 RANGE OUT	AREA NAV 4.220 TLM TACAN RM 4.77	FLOAT PT	0-2432K	FEET	1.04
TACRANGE0(2)	V95H5202C	TACAN NO. 2 RANGE OUT	AREA NAV 4.220 TLM TACAN RM 4.77	FLOAT PT	0-2432K	FEET	1.04
TACRANGE0(3)	V95H5302C	TACAN NO. 3 RANGE OUT	AREA NAV 4.220 TLM TACAN RM 4.77	FLOAT PT	0-2432K	FEET	1.04
TACBEARO(1)	V95H5122C	TACAN NO. 1 BEARING OUT	AREA NAV 4.220 TLM TACAN RM 4.77	FLOAT PT	0 - 6.28	RAD	1.04
TACBEARO(2)	V95H5222C	TACAN NO. 2 BEARING OUT	AREA NAV 4.220 TLM TACAN RM 4.77	FLOAT PT	0 - 6.28	RAD	1.04
TACBEARO(3)	V95H5322C	TACAN NO. 3 BEARING OUT	AREA NAV 4.220 TLM TACAN RM 4.77	FLOAT PT	0 - 6.28	RAD	1.04
TACRANGDG(1)	V95X5104X	TACAN NO. 1 RANGE LOCKED ON	TACAN RM 4.77 GAX 4.110 TLM	LOGICAL	0,1	NA	1.04
TACRANGDG(2)	V95X5204X	TACAN NO. 2 RANGE LOCKED ON	TACAN RM 4.77 GAX 4.110 TLM	LOGICAL	0,1	NA	1.04
TACRANGDG(3)	V95X5304X	TACAN NO. 3 RANGE LOCKED ON	TACAN RM 4.77 GAX 4.110 TLM	LOGICAL	0,1	NA	1.04
TACBEARDG(1)	V95X5124X	TACAN NO. 1 BEARING LOCKED ON	TACAN RM GAX 4.110 TLM	LOGICAL	0,1	NA	1.04
TACBEARDG(2)	V95X5224X	TACAN NO. 2 BEARING LOCKED ON	TACAN RM GAX 4.110 TLM	LOGICAL	0,1	NA	1.04

Table 4.43.4-1.(2) TACAN SOP Interface Table

OUTPUT							
Name	MSID	Description	Destination	Signal Type	Range	Units	Rate (Hz)
TACBEARDG(3)	V95X5324X	TACAN NO. 3 BEARING LOCKED ON	TACAN RM GAX 4.110 TLM	LOGICAL	0,1	NA	1.04
TACPON(1)	V95X5145X	TACAN NO. 1 BUFFERED POWER STATUS	TACAN RM	LOGICAL	0,1	NA	1.04
TACPON(2)	V95X5245X	TACAN NO. 2 BUFFERED POWER STATUS	TACAN RM	LOGICAL	0,1	NA	1.04
TACPON(3)	V95X5345X	TACAN NO. 3 BUFFERED POWER STATUS	TACAN RM	LOGICAL	0,1	NA	1.04
TRNG_CMLK(1)	V95X5140X	TACAN 1 RANGE COMMFAULT/LOCKON FLAG	AREA NAV 4.220, GAX TLM	LOGICAL	0,1	NA	1.04
TRNG_CMLK(2)	V95X5240X	TACAN 2 RANGE COMMFAULT/LOCKON FLAG	AREA NAV 4.220, GAX TLM	LOGICAL	0,1	NA	1.04
TRNG_CMLK(3)	V95X5340X	TACAN 3 RANGE COMMFAULT/LOCKON FLAG	AREA NAV 4.220, GAX TLM	LOGICAL	0,1	NA	1.04
TBRG_CMLK(1)	V95X5141X	TACAN 1 BEAR COMMFAULT/LOCKON FLAG	AREA NAV 4.220, GAX TLM	LOGICAL	0,1	NA	1.04
TBRG_CMLK(2)	V95X5241X	TACAN 2 BEAR COMMFAULT/LOCKON FLAG	AREA NAV 4.220, GAX TLM	LOGICAL	0,1	NA	1.04
TBRG_CMLK(3)	V95X5341X	TACAN 3 BEAR COMMFAULT/LOCKON FLAG	AREA NAV 4.220, GAX TLM	LOGICAL	0,1	NA	1.04
TACCHAN(1)	V95J5130C	TACAN NO. 1 CHANNEL W/MODE	TACAN RM 4.77 4.220 TLM	FIXED PT	+/-126	NA	1.04 1.04
TACCHAN(2)	V95J5230C	TACAN NO. 2 CHANNEL W/MODE	TACAN RM 4.77 4.220 TLM	FIXED PT	+/-126	NA	1.04 1.04
TACCHAN(3)	V95J5330C	TACAN NO. 3 CHANNEL W/MODE	TACAN RM 4.77 4.220 TLM	FIXED PT	+/-126	NA	1.04 1.04
TAC_TIM_TAG**	V95W5001C	TACAN TIME TAG	TLM	FLOAT PT	NA	SEC	1.04
TAC_CNTRL(1)	V74M1510P	TACAN 1 CONTROL WORD	TACAN 1, TLM	FIXED PT	NA	NA	O.D.
TAC_CNTRL(2)	V74M1610P	TACAN 2 CONTROL WORD	TACAN 2, TLM	FIXED PT	NA	NA	O.D.
TAC_CNTRL(3)	V74M1710P	TACAN 3 CONTROL WORD	TACAN 3, TLM	FIXED PT	NA	NA	O.D.

Table 4.43.4-1.(2) TACAN SOP Interface Table

OUTPUT							
Name	MSID	Description	Destination	Signal Type	Range	Units	Rate (Hz)
TACANTSEL(1)	V74K0178X	TACAN 1 ANTENNA SELECTION	TACAN 1, TLM	DISCRETE	0,1	NA	O.D
TACANTSEL(2)	V74K0188X	TACAN 2 ANTENNA SELECTION	TACAN 2, TLM	DISCRETE	0,1	NA	O.D
TACANTSEL(3)	V74K0198X	TACAN 3 ANTENNA SELECTION	TACAN 3, TLM	DISCRETE	0,1	NA	O.D
TACANTINH(1)	V74K0141X	TACAN 1 ANTENNA SWITCH INHIBIT	TACAN 1	DISCRETE	0,1	NA	O.D
TACANTINH(2)	V74K0151X	TACAN 2 ANTENNA SWITCH INHIBIT	TACAN 2	DISCRETE	0,1	NA	O.D.
TACANTINH(3)	V74K0161X	TACAN 3 ANTENNA SWITCH INHIBIT	TACAN 3	DISCRETE	0,1	NA	O.D.
TAC_STIP	V95X5006X	TACAN SELF TEST IN PROGRESS	TACAN RM 4.77 GAX	LOGICAL	0,1	NA	O.D.
TAC_BRGFL(1)	V95X5143X	TACAN NO. 1 BEARING SELF-TEST FAIL	TACAN RM 4.77	LOGICAL	0,1	NA	O.D.
TAC_BRGFL(2)	V95X5243X	TACAN NO. 2 BEARING SELF-TEST FAIL	TACAN RM 4.77	LOGICAL	0,1	NA	O.D.
TAC_BRGFL(3)	V95X5343X	TACAN NO. 3 BEARING SELF-TEST FAIL	TACAN RM 4.77	LOGICAL	0,1	NA	O.D.
TAC_RNGFL(1)	V95X5144X	TACAN NO. 1 RANGE SELF-TEST FAIL	TACAN RM 4.77	LOGICAL	0,1	NA	O.D.
TAC_RNGFL(2)	V95X5244X	TACAN NO. 2 RANGE SELF-TEST FAIL	TACAN RM 4.77	LOGICAL	0,1	NA	O.D.
TAC_RNGFL(3)	V95X5344X	TACAN NO. 3 RANGE SELF-TEST FAIL	TACAN RM 4.77	LOGICAL	0,1	NA	O.D.
TCH_CHNG(1)	V95X5146X	TACAN 1 CHANNEL CHANGE INDICATOR	TACAN RM 4.77	LOGICAL	0,1	NA	1.04
TCH_CHNG(2)	V95X5246X	TACAN 2 CHANNEL CHANGE INDICATOR	TACAN RM 4.77	LOGICAL	0,1	NA	1.04
TCH_CHNG(3)	V95X5346X	TACAN 3 CHANNEL CHANGE INDICATOR	TACAN RM 4.77	LOGICAL	0,1	NA	1.04
NEW_STATION_ID	V95J5020C	TACAN 1, 2, OR 3 NEW STATION CHANNEL NUMBER	SITE LOOK UP	FIXED PT	+/-126	NA	1.04
LRU_ID	V95J5021C	TACAN LRU NUMBER	SITE LOOK UP	FIXED PT	1,2,3	NA	1.04

Table 4.43.4-1.(2) TACAN SOP Interface Table

OUTPUT							
Name	MSID	Description	Destination	Signal Type	Range	Units	Rate (Hz)
T_TACAN	V90W3596C	TACAN TIME TAG	ENT NAV 4.127 PRELAND NAV 4.144 RTLS NAV 4.205				
<p>* These parameters are not directly used in TACAN SOP. ** This time tag is not used anywhere in TACAN SOP. ***See Figure 4.43.1.1.2-3 and 4.43.1.1.2-5 for contents of this parent word.</p>							

Table 4.43.5-1. TACAN SOP I-Load Parameters						
Name	MSID	Description	Const	Type	Value	Units
TAC_RNGLO	V99U5718C	TACAN RANGE LOW LIMIT	SX	SCALAR	*	NMI
TAC_RNGHI	V99U5717C	TACAN RANGE HI LIMIT	SX	SCALAR	*	NMI
TAC_BEARHI	V99U5715C	TACAN BEARING HI LIMIT	SX	SCALAR	*	DEG
TAC_BEARLO	V99U5716C	TACAN BEARING LOW LIMIT	SX	SCALAR	*	DEG
TNMFT	V99U5522C	TACAN NMI TO FEET CONV FACTOR	X	SCALAR	*	FT/NMI
TAC_RLTIM	V99U5520C	TACAN RANGE LOCK TIME LIMIT	SX	SCALAR	*	SEC
TAC_STTIM	V99U5521C	TACAN SELF-TEST	SX	SCALAR	*	SEC

* These values found in JSC 19350.

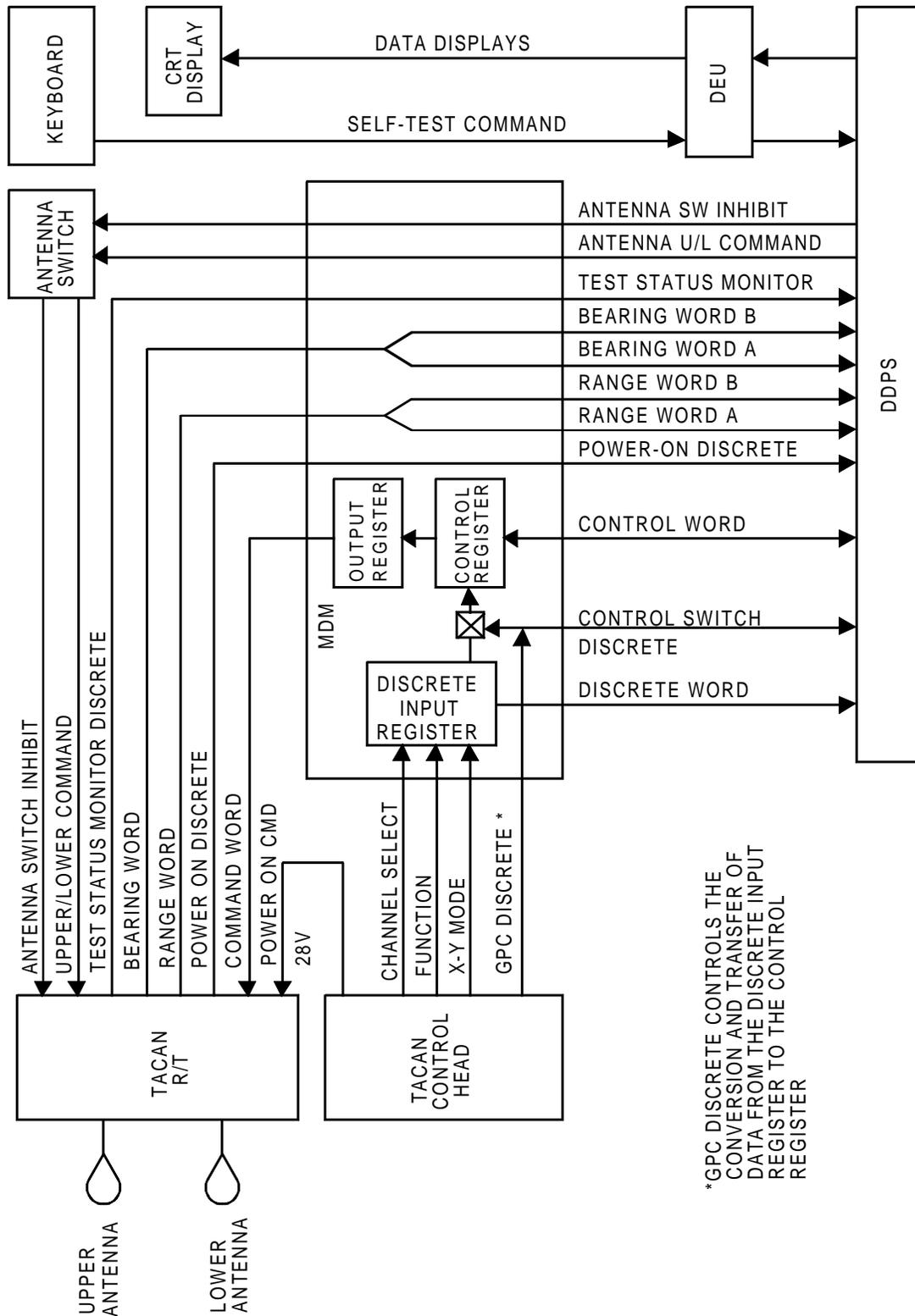


FIGURE 4.43.1.1-1. TACAN Subsystem Block Diagram

<u>BIT NO.</u>	<u>FUNCTION</u>	<u>COMMENT</u>
0	SYNC	ALWAYS 1
1	SPARE	*
2	X/Y MODE	1 = Y, 0 = X
3	A/A	0 = NOT AIR TO AIR
4	T/R, R	1 = T/R, 0 = R
5	SPARE	*
6	SPARE	*
7	CHANNEL SELECTION	1 UNITS
8	CHANNEL SELECTION	2 UNITS
9	CHANNEL SELECTION	4 UNITS
10	CHANNEL SELECTION	8 UNITS
11	CHANNEL SELECTION	10 TENS
12	CHANNEL SELECTION	20 TENS
13	CHANNEL SELECTION	40 TENS
14	CHANNEL SELECTION	80 TENS
15	SELF TEST	1 = TEST
16-31	SET TO 0 BY MDM	

* DON'T CARE BITS IN HARDWARE

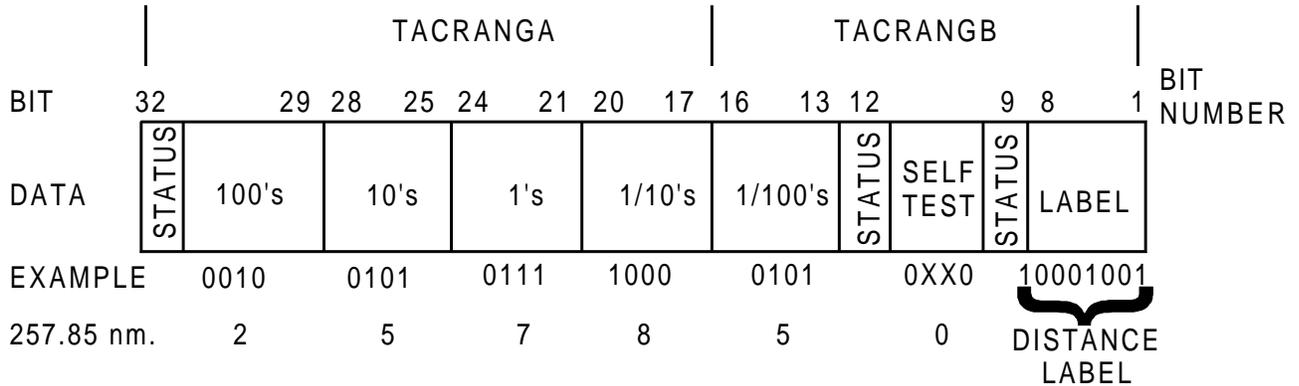
FIGURE 4.43.1.1.2-1. TACAN Control Register Format

<u>BIT NO.</u>	<u>FUNCTION</u>	<u>COMMENT</u>
0	CHANNEL SELECTION	100 HUNDREDS
1	POWER ON DISCRETE	1 = PWR ON
2	X/Y MODE	1 = Y, 0 = X
3	TEST STATUS MONITOR	1 = GOOD
4	T/R, R	1 = T/R, 0 = R
5	RADAR ALTIMETER	
6	RADAR ALTIMETER	
7	CHANNEL SELECTION	1 UNITS
8	CHANNEL SELECTION	2 UNITS
9	CHANNEL SELECTION	4 UNITS
10	CHANNEL SELECTION	8 UNITS
11	CHANNEL SELECTION	10 TENS
12	CHANNEL SELECTION	20 TENS
13	CHANNEL SELECTION	40 TENS
14	CHANNEL SELECTION	80 TENS
15	SPARE	

NOTE: FOR TACAN CHANNEL NUMBERS ≥100, BITS 11-14 OF THE CONTROL DIFFER FROM BITS 11-14 AND 0 OF THE DISCRETE WORD

FIGURE 4.43.1.1.2-2. TACAN Discrete Word Format

RANGWORD



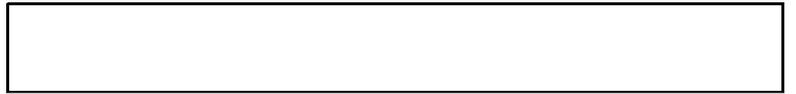
32	31	12	11	10	9	BIT
0	0	-	-	-	-	VALID DATA (TRACK OR SHORT-TERM MEMORY)
0	1	-	-	-	-	SELF-TEST FAILURE
1	0	-	-	-	-	NO COMPUTED DATA (SEARCH OR LONG-TERM MEMORY)
1	1	-	-	-	-	NOT USED
-	-	-	0	0	-	NOT IN SELF-TEST
-	-	-	1	1	-	IN SELF-TEST
-	-	-	0	1	-	NOT DEFINED
-	-	-	1	0	-	NOT DEFINED
-	-	0	-	-	-	PASSED AUTO BITE TEST *
-	-	1	-	-	-	FAILED AUTO BITE TEST *
-	-	-	-	-	0	SUPPRESSION PULSE NOT PRESENT *
-	-	-	-	-	1	SUPPRESSION PULSE PRESENT *

* APPLIES TO MC409-0184 TACAN ONLY

FIGURE 4.43.1.1.2-3. TACAN-to-MDM Range Word

RANGEWORD

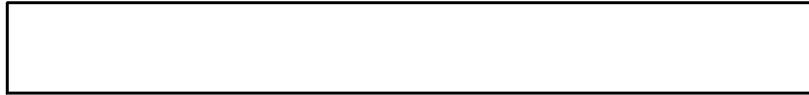
TACRANGA



0

15

TACRANGB



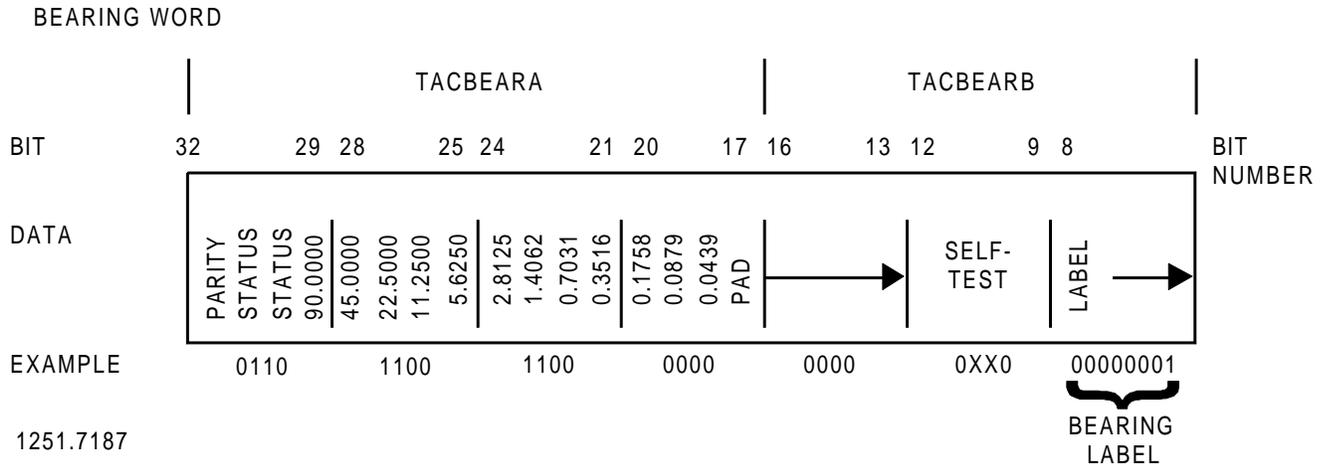
16

31

BITS	
0 AND 1	= "00" VALID DATA = "01" SELF-TEST FAILURE = "10" NO COMPUTED DATA = "11" NOT USED
2 AND 3	100'S N. MILES
4 THROUGH 7	10'S N. MILES
8 THROUGH 11	1'S N. MILES
12 THROUGH 15	1/10'S N. MILES
16 THROUGH 19	1/100'S N. MILES
20	*AUTO BITE TEST STATUS = "0" PASS = "1" FAIL
21 AND 22	= "00" NOT IN SELF-TEST = "11" IN SELF-TEST = "10" NOT USED = "01" NOT USED
23	*SUPPRESSION PULSE PRESENT = "0" PULSE NOT PRESENT = "1" PULSE PRESENT
24 THROUGH 31	LABEL = 10001001

* APPLIES TO MC409-0184 TACAN ONLY

FIGURE 4.43.1.1.2-4. MDM-to-GPC Range Words



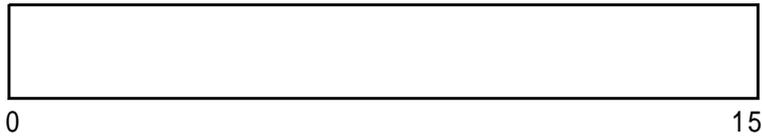
31	30	11	10	BIT
0	0	-	-	LESS THAN 180 DEGREES (TRACK OR SHORT-TERM MEMORY)
0	1	-	-	SELF-TEST FAILURE
1	0	-	-	NO COMPUTED DATA (NO OR UNUSEABLE SIGNAL)
1	1	-	-	180 DEGREES OR GREATER (TRACK OR SHORT-TERM MEMORY)
-	-	0	0	NOT IN SELF-TEST
-	-	1	1	IN SELF-TEST
-	-	0	1	NOT DEFINED
-	-	1	0	NOT DEFINED

NOTE: BIT 32 IS ALWAYS SET FOR ODD PARITY
BINARY VALUES ROUNDED.

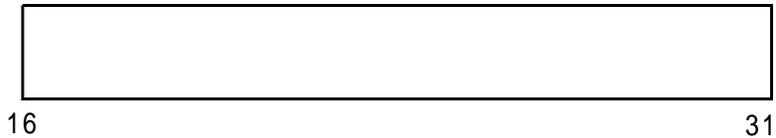
FIGURE 4.43.1.1.2-5. TACAN-to-MDM Bearing Word

BEARING WORD

TACBEARA



TACBEARB



BITS

0	=	PARITY	
1 and 2	=	'00'	LESS THAN 180 DEGREES
	=	'01'	SELF-TEST FAILURE
	=	'10'	NO COMPUTED DATA
	=	'≥11'	180 DEGREES OR GREATER
3	=	90.0000°	
4	=	45.0000°	
5	=	22.5000°	
6	=	11.2500°	
7	=	5.6250°	
8	=	2.8125°	
9	=	1.40625°	
10	=	0.703125°	
11	=	0.3515625°	
12	=	0.1757812°	
13	=	0.0878956°	
14	=	0.0439453°	
15 through 20	*		
21 and 22	=	'00'	NOT IN SELF TEST
	=	'11'	IN SELF TEST
	=	'10'	NOT USED
	=	'01'	NOT USED
23	*		
24 through 31		LABEL	= 0000 0001

* DON'T CARE BITS IN H/W

FIGURE 4.43.1.1.2-6. MDM-to-GPC Bearing Word

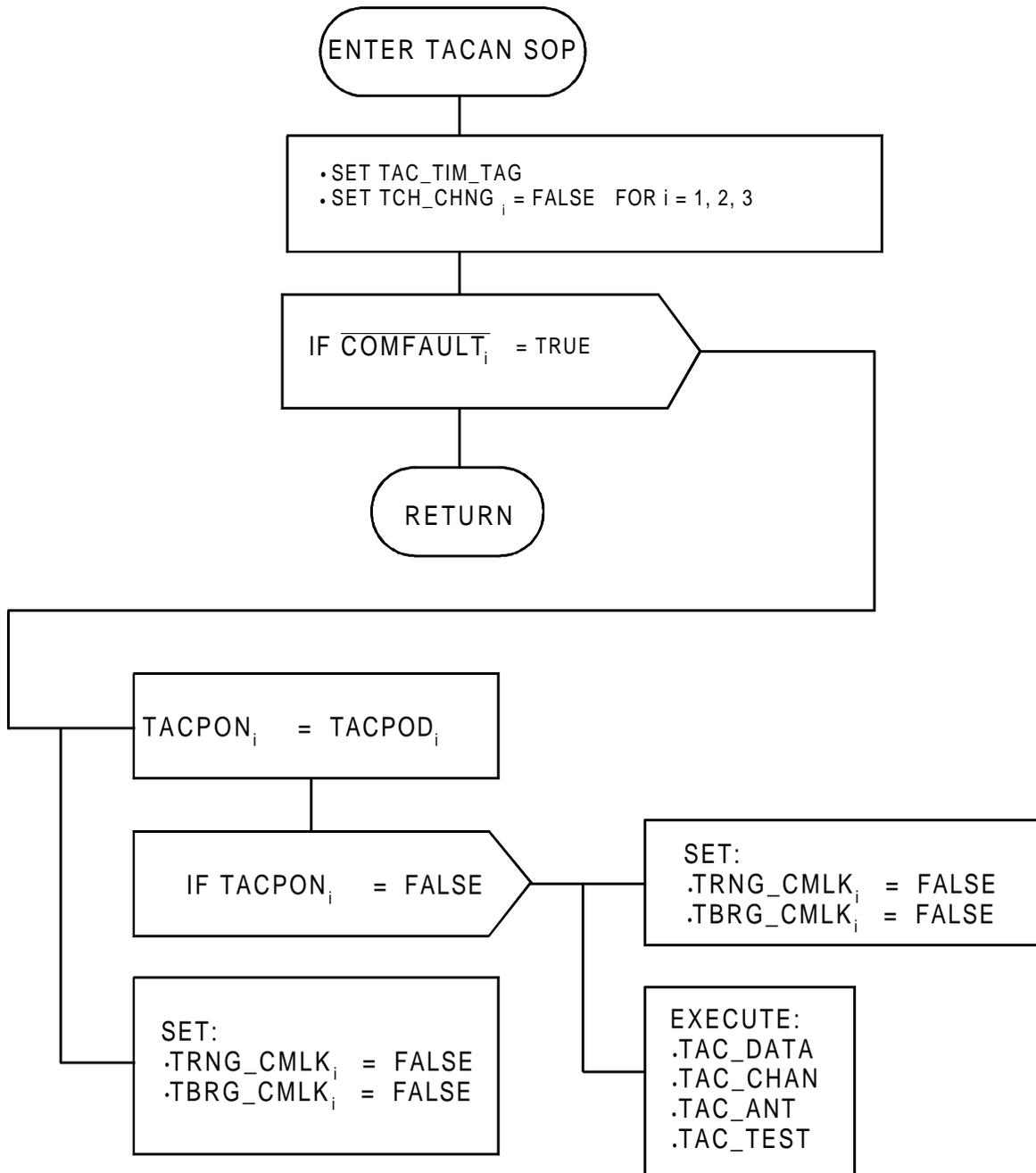


FIGURE 4.43.6-1.TACAN SOP

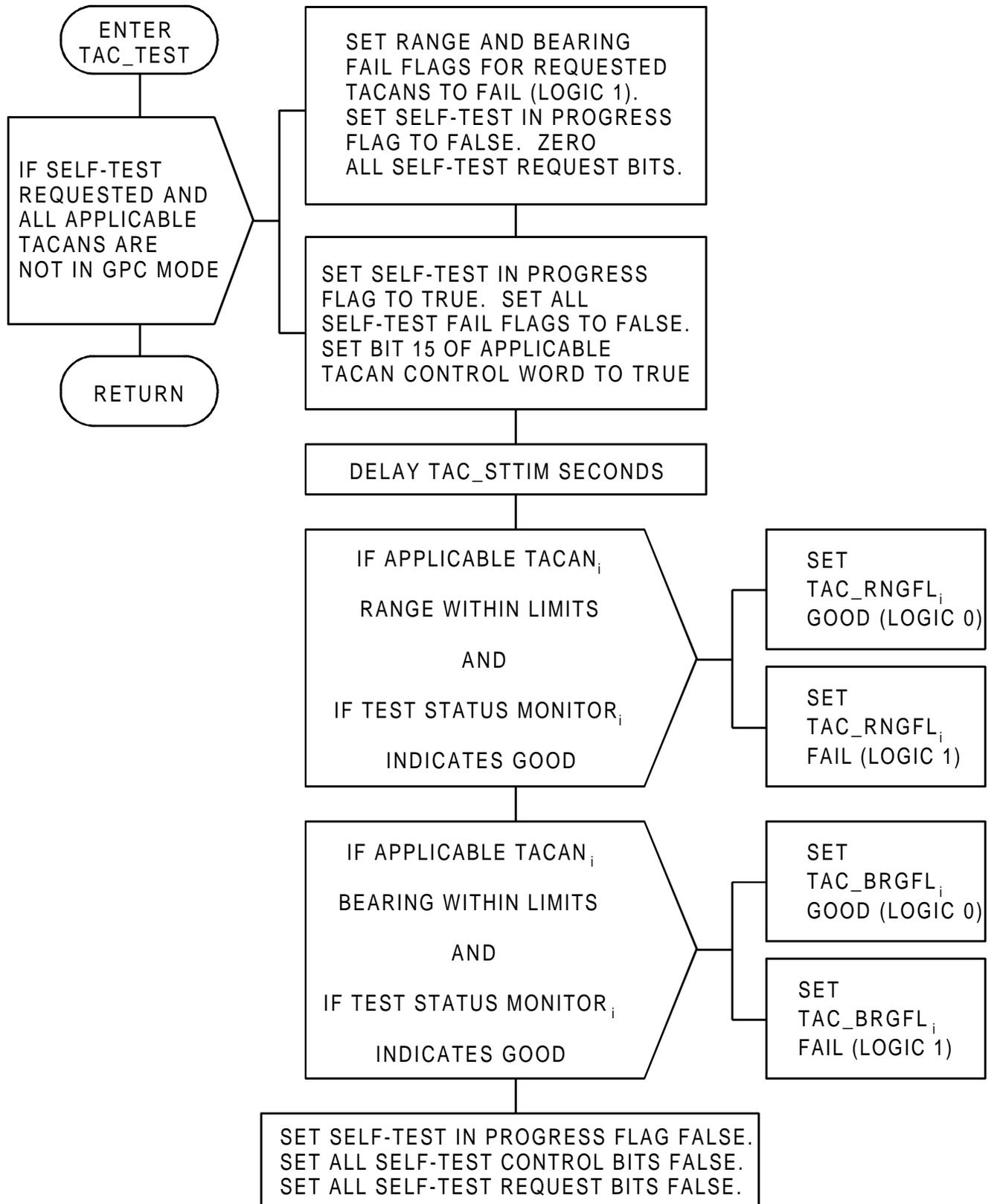


FIGURE 4.43.6.1-1. TAC_TEST Flow Diagram

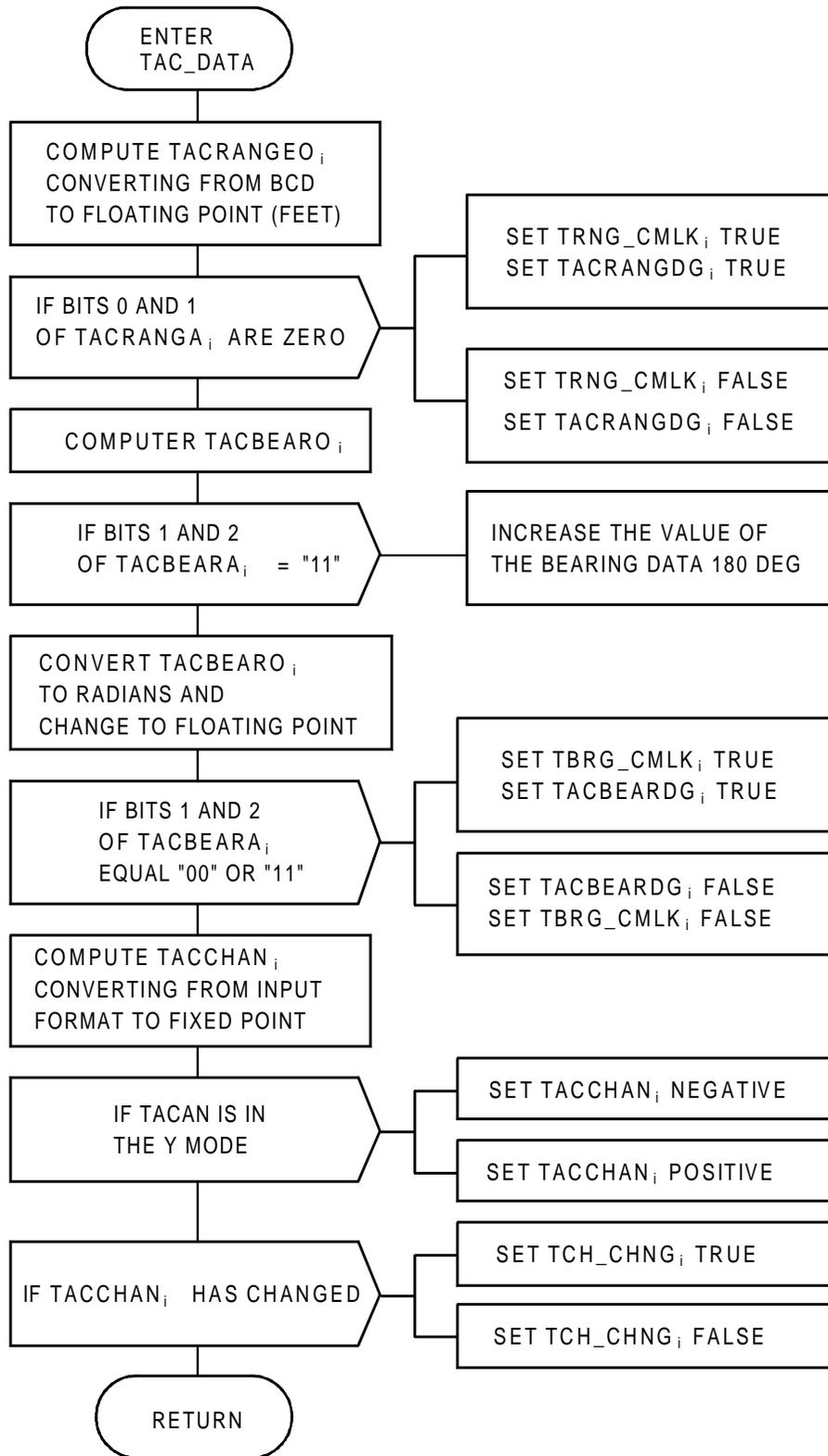


FIGURE 4.43.6.2-1. TAC_DATA Flow Diagram

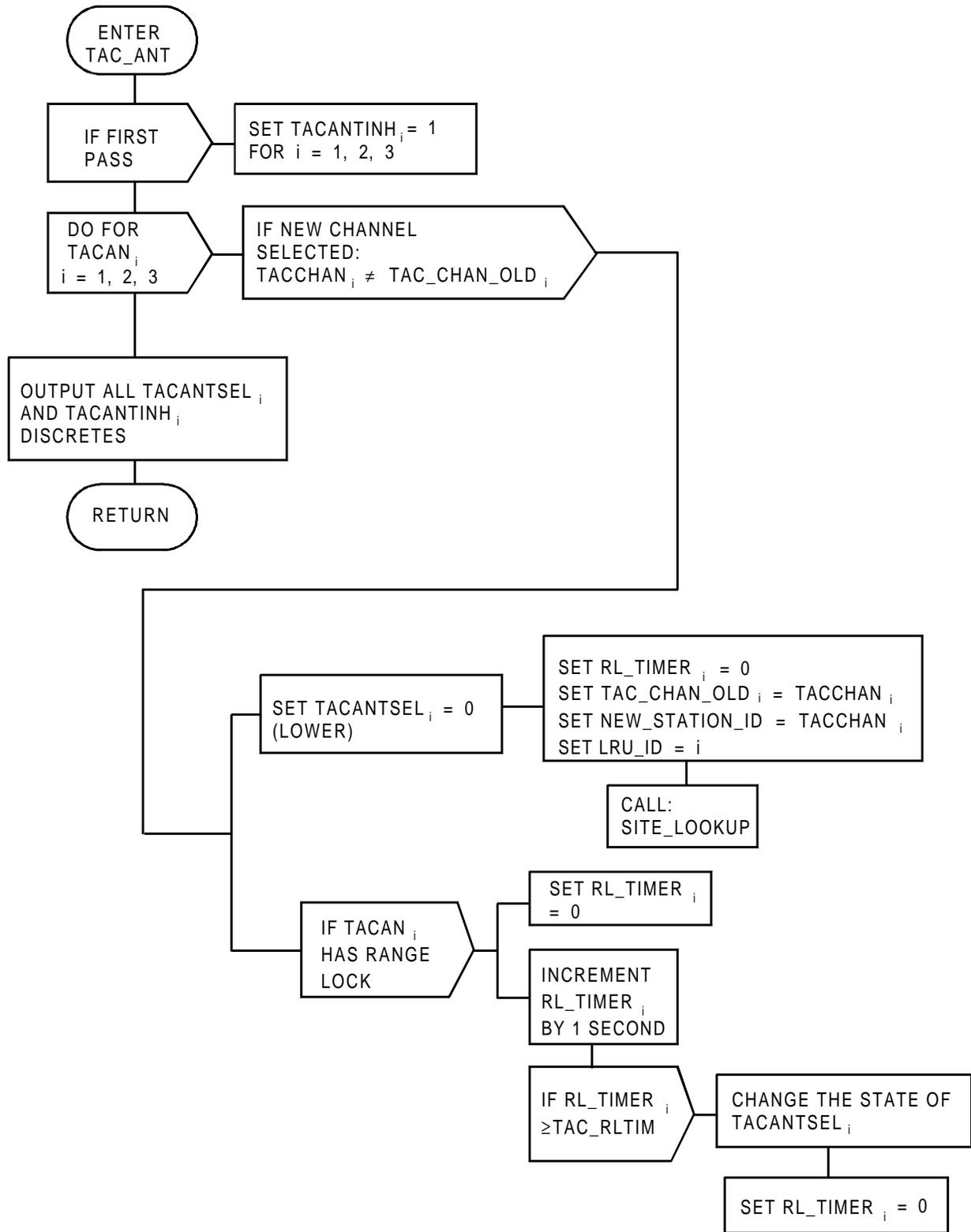


FIGURE 4.43.6.3-1. TAC_ANT Flow Diagram

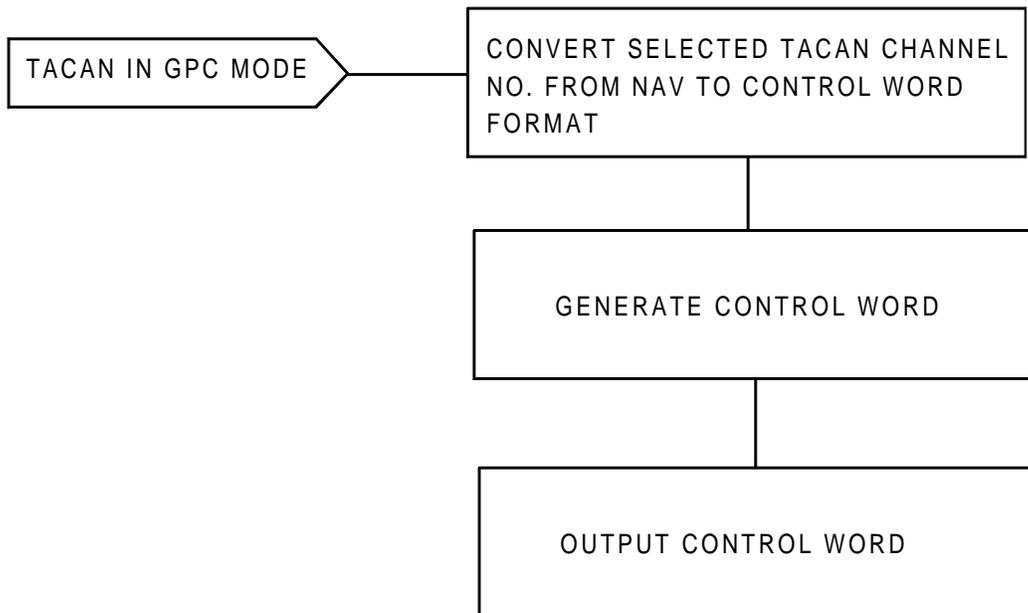


FIGURE 4.43.6.4-1. TAC_CHAN Flow Diagram

TABLE 4.43.6.5-1. INPUT FUNCTIONAL PARAMETERS FOR TACTICAL AIR NAVIGATION SOP (G4.43)

FSSR NAME	M/S ID	NOMENCLATURE	SOURCE	UNITS	DATA TYPE	P R E C	LAST CR	R E V
AUTO_CHANNEL_ID	V95J5022CA	AUTO TACAN CHANNEL SELECTION	ENT SITE LOOKUP					
AUTO_CHANNEL_ID	V95J5022CB	AUTO TACAN CHANNEL SELECTION	ASC SITE LOOKUP					
CLOCK/CLOCKTIME	V91W5000C	CLOCK-COMPUTER (GMT)	FCOS	S			93160B	
RCVR_LEFT	V96J2105C	LH HSI RECEIVER SELECT	GN&C SW RM					
RCVR_RIGHT	V96J2205C	RH HSI RECEIVER SELECT	GN&C SW RM					
SOURCE_LEFT	V96X2111X	LH HSI SOURCE SELECT	GN&C SW RM					
SOURCE_RIGHT	V96X2211X	RH HSI SOURCE SELECT	GN&C SW RM					
TACBEAR (1)	V74M1529P	TACAN 1 BEARING 32 BITS	TACAN 1		FMD		92297B	
TACBEAR (2)	V74M1629P	TACAN 2 BEARING 32 BITS	TACAN 2		FMD		92297B	
TACBEAR (3)	V74M1729P	TACAN 3 BEARING 32 BITS	TACAN 3		FMD		92297B	
TACBEARA (1)	V74M1530P	TACAN 1 BEARING WORD 1	WD1 TACAN 1	DEG	HMD		93017G	*
TACBEARA (2)	V74M1630P	TACAN 2 BEARING WORD 1	WD1 TACAN 2	DEG	HMD		93017G	*
TACBEARA (3)	V74M1730P	TACAN 3 BEARING WORD 1	WD1 TACAN 3	DEG	HMD		93017G	*
TACBEARB (1)	V74M1531P	TACAN 1 BEARING WORD 2	WD2 TACAN 1	DEG	HMD		93017G	*
TACBEARB (2)	V74M1631P	TACAN 2 BEARING WORD 2	WD2 TACAN 2	DEG	HMD		93017G	*
TACBEARB (3)	V74M1731P	TACAN 3 BEARING WORD 2	WD2 TACAN 3	DEG	HMD		93017G	*
TACCFM (1)	V72M6598P	MDM FF01 DSCR IN CD 00 CH 6	HDWR		HMD		92025B	
TACCFM (2)	V72M6648P	MDM FF02 DSCR IN CD 00 CH 6	HDWR		HMD		92025B	
TACCFM (3)	V72M6698P	MDM FF03 DSCR IN CD 00 CH 6	HDWR		HMD		92025B	
TACPOD (1)	V74X0071X	TACAN NO 1 POWER STATUS	TACAN 1		BD		93017G	*
TACPOD (2)	V74X0081X	TACAN NO 2 POWER STATUS	TACAN 2		BD		93017G	*
TACPOD (3)	V74X0091X	TACAN NO 3 POWER STATUS	TACAN 3		BD		93017G	*
TACRANG (1)	V74M1549P	TACAN 1 RANGE 32 BITS	TACAN 1		FMD		93017G	*
TACRANG (2)	V74M1649P	TACAN 2 RANGE 32 BITS	TACAN 2		FMD		93017G	*
TACRANG (3)	V74M1749P	TACAN 3 RANGE 32 BITS	TACAN 3		FMD		93017G	*
TACRANGA (1)	V74M1550P	TACAN 1 RANGE WORD A	WD1 TACAN 1	NM	HMD		92297B	
TACRANGA (2)	V74M1650P	TACAN 2 RANGE WORD A	WD1 TACAN 2	NM	HMD		92297B	
TACRANGA (3)	V74M1750P	TACAN 3 RANGE WORD A	WD1 TACAN 3	NM	HMD		92297B	
TACRANGB (1)	V74M1551P	TACAN 1 RANGE WORD B	WD2 TACAN 1	NM	HMD		92297B	
TACRANGB (2)	V74M1651P	TACAN 2 RANGE WORD B	WD2 TACAN 2	NM	HMD		92297B	
TACRANGB (3)	V74M1751P	TACAN 3 RANGE WORD B	WD2 TACAN 3	NM	HMD		92297B	
TAC_CF (1)	V91X3500X	TACAN 1 COMMFAULT	FCOS				92025B	
TAC_CF (2)	V91X3510X	TACAN 2 COMMFAULT	FCOS				92025B	
TAC_CF (3)	V91X3520X	TACAN 3 COMMFAULT	FCOS				92025B	
TAC_CHAN_OLD (1)	V95U5023CA	PREV VALUE TAC CHAN FOR LRU 1	ENT SITE LOOKUP					
TAC_CHAN_OLD (1)	V95U5023CB	PREV VALUE TAC CHAN FOR LRU 1	ASC SITE LOOKUP					
TAC_CHAN_OLD (2)	V95U5024CA	PREV VALUE TAC CHAN FOR LRU 2	ENT SITE LOOKUP					
TAC_CHAN_OLD (2)	V95U5024CB	PREV VALUE TAC CHAN FOR LRU 2	ASC SITE LOOKUP					
TAC_CHAN_OLD (3)	V95U5025CA	PREV VALUE TAC CHAN FOR LRU 3	ENT SITE LOOKUP					
TAC_CHAN_OLD (3)	V95U5025CB	PREV VALUE TAC CHAN FOR LRU 3	ASC SITE LOOKUP					
TAC_CNTRL (1)	V74M1510PA	TACAN NO 1 CONTROL WORD	D&C PNL				92025B	
TAC_CNTRL (2)	V74M1610PA	TACAN NO 2 CONTROL WORD	D&C PNL				92025B	
TAC_CNTRL (3)	V74M1710PA	TACAN NO 3 CONTROL WORD	D&C PNL				92025B	
TAC_DW (1)	V72M6599P	MDM FF01 DSCR IN CD 00 CH 5	HDWR		HMD		93017G	*
TAC_DW (2)	V72M6649P	MDM FF02 DSCR IN CD 00 CH 5	HDWR		HMD		93017G	*
TAC_DW (3)	V72M6699P	MDM FF03 DSCR IN CD 00 CH 5	HDWR		HMD		93017G	*

TABLE 4.43.6.5-1. INPUT FUNCTIONAL PARAMETERS FOR TACTICAL AIR NAVIGATION SOP (G4.43)

FSSR NAME	M/S ID	NOMENCLATURE	SOURCE	UNITS	DATA TYPE	P R E C	LAST CR	R E V
TAC_GPC(1)	V72K8410X	TACAN NO 1 AUTO DISCRETE(COMPUTER)	HDWR		BD		93017G	*
TAC_GPC(2)	V72K8430X	TACAN NO 2 AUTO DISCRETE(COMPUTER)	HDWR		BD		93017G	*
TAC_GPC(3)	V72K8450X	TACAN NO 3 AUTO DISCRETE(COMPUTER)	HDWR		BD		93017G	*
TAC_STREQ	V90J3602C	TACAN SELF TEST REQUEST	TACAN RM	BITS			92025B	
TAC_TSM(1)	V74X0070X	TACAN NO 1 TEST STATUS MONITOR-GO	TACAN 1		BD		93017G	*
TAC_TSM(2)	V74X0080X	TACAN NO 2 TEST STATUS MONITOR-GO	TACAN 2		BD		93017G	*
TAC_TSM(3)	V74X0090X	TACAN NO 3 TEST STATUS MONITOR-GO	TACAN 3		BD		93017G	*
	V72M6597P	MDM FF01 DSCR IN CD 09 CH 02	HDWR		HMD		93017G	*
	V72M6647P	MDM FF02 DSCR IN CD 09 CH 02	HDWR		HMD		93017G	*
	V72M6697P	MDM FF03 DSCR IN CD 09 CH 02	HDWR		HMD		93017G	*
	V74H1534B	TACAN 1 BEARING/STATUS	WD1B01-02 TACAN 1	DEG	BMD		93017G	*
	V74H1535B	TACAN 1 BEARING DATA	WD1B03-14 TACAN 1	DEG	BSU		93017G	*
	V74H1554B	TACAN 1 RANGE (WD-A)	WD1B02-15 TACAN 1	NM	BCD		92297B	
	V74H1555B	TACAN 1 RANGE (WD-B)	WD2B00-03 TACAN 1	NM	BCD		92297B	
	V74H1634B	TACAN 2 BEARING/STATUS	WD1B01-02 TACAN 2	DEG	BMD		93017G	*
	V74H1635B	TACAN 2 BEARING DATA	WD1B03-14 TACAN 2	DEG	BSU		93017G	*
	V74H1654B	TACAN 2 RANGE (WD-A)	WD1B02-15 TACAN 2	NM	BCD		92297B	
	V74H1655B	TACAN 2 RANGE (WD-B)	WD2B00-03 TACAN 2	NM	BCD		92297B	
	V74H1734B	TACAN 3 BEARING/STATUS	WD1B01-02 TACAN 3	DEG	BMD		93017G	*
	V74H1735B	TACAN 3 BEARING DATA	WD1B03-14 TACAN 3	DEG	BSU		93017G	*
	V74H1754B	TACAN 3 RANGE (WD-A)	WD1B02-15 TACAN 3	NM	BCD		92297B	
	V74H1755B	TACAN 3 RANGE (WD-B)	WD2B00-03 TACAN 3	NM	BCD		92297B	
	V74J1536B	TACAN 1 BEARING SELF TEST	WD2B04-07 TACAN 1		BMD		93017G	*
	V74J1537B	TACAN 1 BEARING LABEL	WD2B08-15 TACAN 1		BSU		92297B	
	V74J1552B	TACAN 1 RANGE STATUS	WD1B00-01 TACAN 1		BMD		93017G	*
	V74J1557B	TACAN 1 RANGE LABEL	WD2B08-15 TACAN 1		BSU		92297B	
	V74J1559B	TACAN 1 RANGE SELF TEST	WD2B05-06 TACAN 1		BMD		93017G	*
	V74J1636B	TACAN 2 BEARING SELF TEST	WD2B04-07 TACAN 2		BMD		93017G	*
	V74J1637B	TACAN 2 BEARING LABEL	WD2B08-15 TACAN 2		BSU		92297B	
	V74J1652B	TACAN 2 RANGE STATUS	WD1B00-01 TACAN 2		BMD		93017G	*
	V74J1657B	TACAN 2 RANGE LABEL	WD2B08-15 TACAN 2		BSU		92297B	
	V74J1659B	TACAN 2 RANGE SELF TEST	WD2B05-06 TACAN 2		BMD		93017G	*
	V74J1736B	TACAN 3 BEARING SELF TEST	WD2B04-07 TACAN 3		BMD		93017G	*
	V74J1737B	TACAN 3 BEARING LABEL	WD2B08-15 TACAN 3		BSU		92297B	
	V74J1752B	TACAN 3 RANGE STATUS	WD1B00-01 TACAN 3		BMD		93017G	*
	V74J1757B	TACAN 3 RANGE LABEL	WD2B08-15 TACAN 3		BSU		92297B	
	V74J1759B	TACAN 3 RANGE SELF TEST	WD2B05-06 TACAN 3		BMD		93017G	*
	V74U1533B	TACAN 1 BEARING 14 BITS	TACAN 1		BSS		92297B	
	V74U1553B	TACAN 1 RANGE 18 BITS	TACAN 1	NM	BCD		93017G	*
	V74U1633B	TACAN 2 BEARING 14 BITS	TACAN 2		BSS		92297B	
	V74U1653B	TACAN 2 RANGE 18 BITS	TACAN 2	NM	BCD		93017G	*
	V74U1733B	TACAN 3 BEARING 14 BITS	TACAN 3		BSS		92297B	
	V74U1753B	TACAN 3 RANGE 18 BITS	TACAN 3	NM	BCD		93017G	*
	V74X1532B	TACAN 1 BEARING DATA PARITY	WD1B00 TACAN 1		BD		93017G	*
	V74X1558B	TACAN 1 RNG BLT-IN TST STAT	WD2B04 TACAN 1		BD		93017G	*
	V74X1560B	TACAN 1 RNG SUPR PULS PRST	WD2B07 TACAN 1		BD		93017G	*

TABLE 4.43.6.5-1. INPUT FUNCTIONAL PARAMETERS FOR TACTICAL AIR NAVIGATION SOP (G4.43)

FSSR NAME	M/S ID	NOMENCLATURE	SOURCE	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X1632B	TACAN 2 BEARING DATA PARITY	WD1B00	TACAN 2	BD		93017G	*
	V74X1658B	TACAN 2 RNG BLT-IN TST STAT	WD2B04	TACAN 2	BD		93017G	*
	V74X1660B	TACAN 2 RNG SUPR PULS PRST	WD2B07	TACAN 2	BD		93017G	*
	V74X1732B	TACAN 3 BEARING DATA PARITY	WD1B00	TACAN 3	BD		93017G	*
	V74X1758B	TACAN 3 RNG BLT-IN TST STAT	WD2B04	TACAN 3	BD		93017G	*
	V74X1760B	TACAN 3 RNG SUPR PULS PRST	WD2B07	TACAN 3	BD		93017G	*

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

TABLE 4.43.6.5-2. OUTPUT FUNCTIONAL PARAMETERS FROM TACTICAL AIR NAVIGATION SOP (G4.43)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
LRU_ID	V95J5021C	TACAN LRU NUMBER	ASC SITE LOOKUP					
NEW_STATION_ID	V95J5020CA	TACAN 1,2,OR 3 NEW STATION CHAN NO	ENT SITE LOOKUP					
TACANTINH(1)	V74K0141XA	TACAN NO 1 ANTENNA SELECT INHIBIT	ASC SITE LOOKUP ENT SITE LOOKUP				93017G	*
			DL FMT 21/1 DL FMT 23/1 DL FMT 32/1 TACAN 1					
TACANTINH(2)	V74K0151XA	TACAN NO 2 ANTENNA SELECT INHIBIT	DL FMT 21/1 DL FMT 23/1 DL FMT 32/1 TACAN 2				93017G	*
TACANTINH(3)	V74K0161XA	TACAN NO 3 ANTENNA SELECT INHIBIT	DL FMT 21/1 DL FMT 23/1 TACAN 3				93017G	*
TACANTSEL(1)	V74K0178XA	TACAN NO 1 UPPER/LOWER ANT LOCK	DL FMT 21/1 DL FMT 23/1 DL FMT 32/1 TACAN 1				93017G	*
TACANTSEL(2)	V74K0188XA	TACAN NO 2 UPPER/LOWER ANT LOCK	DL FMT 21/1 DL FMT 23/1 DL FMT 32/1 TACAN 2				93017G	*
TACANTSEL(3)	V74K0198XA	TACAN NO 3 UPPER/LOWER ANT LOCK	DL FMT 21/1 DL FMT 23/1 TACAN 3				93017G	*
TACBEARDG(1)	V95X5124XA	TACAN NO 1 BEARING LOCKED ON	TACAN RM TLM					
TACBEARDG(2)	V95X5224XA	TACAN NO 2 BEARING LOCKED ON	TACAN RM TLM					
TACBEARDG(3)	V95X5324XA	TACAN NO 3 BEARING LOCKED ON	TACAN RM TLM					
TACBEARO(1)	V95H5122CA	TACAN NO 1 BEARING OUT	A/L UPP ASC AREA NAV ENT AREA NAV ENT UPP RTLS UPP TACAN RM TAEM UPP TLM	RAD				

TABLE 4.43.6.5-2. OUTPUT FUNCTIONAL PARAMETERS FROM TACTICAL AIR NAVIGATION SOP (G4.43)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
TACBEARO(2)	V95H5222CA	TACAN NO 2 BEARING OUT	A/L UPP ASC AREA NAV ENT AREA NAV ENT UPP RTLS UPP TACAN RM TAEM UPP TLM	RAD				
TACBEARO(3)	V95H5322CA	TACAN NO 3 BEARING OUT	A/L UPP ASC AREA NAV ENT AREA NAV ENT UPP RTLS UPP TACAN RM TAEM UPP TLM	RAD				
TACCHAN(1)	V95J5130CA	TACAN NO 1 CHANNEL W/MODE	TACAN RM TLM					
TACCHAN(2)	V95J5230CA	TACAN NO 2 CHANNEL W/MODE	TACAN RM TLM					
TACCHAN(3)	V95J5330CA	TACAN NO 3 CHANNEL W/MODE	TACAN RM TLM					
TACPON(1)	V95X5145XA	TACAN NO 1 BUFFERED POWER STATUS	TACAN RM				90182	
TACPON(2)	V95X5245XA	TACAN NO 2 BUFFERED POWER STATUS	TACAN RM				90182	
TACPON(3)	V95X5345XA	TACAN NO 3 BUFFERED POWER STATUS	TACAN RM				90182	
TACRANGDG(1)	V95X5104XA	TACAN NO 1 RANGE LOCKED ON	TACAN RM TLM					
TACRANGDG(2)	V95X5204XA	TACAN NO 2 RANGE LOCKED ON	TACAN RM TLM					
TACRANGDG(3)	V95X5304XA	TACAN NO 3 RANGE LOCKED ON	TACAN RM TLM					
TACRANGE0(1)	V95H5102CA	TACAN NO 1 RANGE OUT	A/L UPP ASC AREA NAV ENT AREA NAV ENT UPP RTLS UPP TACAN RM TAEM UPP TLM	FT				

TABLE 4.43.6.5-2. OUTPUT FUNCTIONAL PARAMETERS FROM TACTICAL AIR NAVIGATION SOP (G4.43)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
TACRANGE0(2)	V95H5202CA	TACAN NO 2 RANGE OUT	A/L UPP ASC AREA NAV ENT AREA NAV ENT UPP RTLS UPP TACAN RM TAEM UPP	FT				
TACRANGE0(3)	V95H5302CA	TACAN NO 3 RANGE OUT	TLM A/L UPP ASC AREA NAV ENT AREA NAV ENT UPP RTLS UPP TACAN RM TAEM UPP	FT				
TAC_BRGFL(1)	V95X5143X	TACAN NO 1 BEARING SELF TEST FAIL	TACAN RM					
TAC_BRGFL(2)	V95X5243X	TACAN NO 2 BEARING SELF TEST FAIL	TACAN RM					
TAC_BRGFL(3)	V95X5343X	TACAN NO 3 BEARING SELF TEST FAIL	TACAN RM					
TAC_CHAN(1)	V74K1514BA	TACAN NO 1 CHANNEL SELECT	DL FMT 21/1 DL FMT 23/A DL FMT 32/A DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 52/1 DL FMT 53/1 DL FMT 60/1 TACAN 1				93017G	*
TAC_CHAN(2)	V74K1614BA	TACAN NO 2 CHANNEL SELECT	DL FMT 21/1 DL FMT 23/A DL FMT 32/A DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 52/1 DL FMT 53/1 DL FMT 60/1 TACAN 2				93017G	*

TABLE 4.43.6.5-2. OUTPUT FUNCTIONAL PARAMETERS FROM TACTICAL AIR NAVIGATION SOP (G4.43)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
TAC_CHAN(3)	V74K1714BA	TACAN NO 3 CHANNEL SELECT	DL FMT 21/1 DL FMT 23/A DL FMT 32/A DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 52/1 DL FMT 53/1 DL FMT 60/1				93017G	*
TAC_CNTRL(1)	V74M1510PB	TACAN NO 1 CONTROL WORD AUTO	TACAN 3 DL FMT 21/1 DL FMT 23/A DL FMT 32/A DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 52/1 DL FMT 53/1 DL FMT 60/1				93017G	*
TAC_CNTRL(2)	V74M1610PB	TACAN NO 2 CONTROL WORD AUTO	TACAN 1 DL FMT 21/1 DL FMT 23/A DL FMT 32/A DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 52/1 DL FMT 53/1 DL FMT 60/1				93017G	*
TAC_CNTRL(3)	V74M1710PB	TACAN NO 3 CONTROL WORD AUTO	TACAN 2 DL FMT 21/1 DL FMT 23/A DL FMT 32/A DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 52/1 DL FMT 53/1 DL FMT 60/1				93017G	*
TAC_RNGFL(1)	V95X5144X	TACAN NO 1 RANGE SELF TEST FAIL	TACAN 3					
TAC_RNGFL(2)	V95X5244X	TACAN NO 2 RANGE SELF TEST FAIL	TACAN RM					

TABLE 4.43.6.5-2. OUTPUT FUNCTIONAL PARAMETERS FROM TACTICAL AIR NAVIGATION SOP (G4.43)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
TAC_RNGFL(3) TAC_STIP	V95X5344X V95X5006X	TACAN NO 3 RANGE SELF TEST FAIL TACAN SELF TEST IN PROGRESS	TACAN RM GAX		BD			
TAC_TIM_TAG	V95W5001CA	TACAN TIME TAG	HORIZ SIT DISP TACAN RM DL FMT 21/1 DL FMT 23/A	S			93017G	*
TBRG_CMLK(1)	V95X5141XA	TACAN 1 BEAR COMFAULT/LOCKON FLAG	TACAN RM ASC AREA NAV DL FMT 21/1 DL FMT 23/A ENT AREA NAV GAX				93017G	*
TBRG_CMLK(2)	V95X5241XA	TACAN 2 BEAR COMFAULT/LOCKON FLAG	HORIZ SIT SPEC ASC AREA NAV DL FMT 21/1 DL FMT 23/A ENT AREA NAV GAX				93017G	*
TBRG_CMLK(3)	V95X5341XA	TACAN 3 BEAR COMFAULT/LOCKON FLAG	HORIZ SIT SPEC ASC AREA NAV DL FMT 21/1 DL FMT 23/A ENT AREA NAV GAX				93017G	*
TCH_CHNG(1) TCH_CHNG(2) TCH_CHNG(3) TRNG_CMLK(1)	V95X5146X V95X5246X V95X5346X V95X5140XA	TACAN 1 CHANNEL CHANGE INDICATOR TACAN 2 CHANNEL CHANGE INDICATOR TACAN 3 CHANNEL CHANGE INDICATOR TACAN 1 RANGE COMFAULT/LOCKON FLAG	TACAN RM TACAN RM TACAN RM ASC AREA NAV DL FMT 21/1 DL FMT 23/A ENT AREA NAV GAX				93017G	*
TRNG_CMLK(2)	V95X5240XA	TACAN 2 RANGE COMFAULT/LOCKON FLAG	HORIZ SIT SPEC ASC AREA NAV DL FMT 21/1 DL FMT 23/A ENT AREA NAV GAX				93017G	*
TRNG_CMLK(3)	V95X5340XA	TACAN 3 RANGE COMFAULT/LOCKON FLAG	HORIZ SIT SPEC ASC AREA NAV DL FMT 21/1 DL FMT 23/A ENT AREA NAV GAX HORIZ SIT SPEC				93017G	*

TABLE 4.43.6.5-2. OUTPUT FUNCTIONAL PARAMETERS FROM TACTICAL AIR NAVIGATION SOP (G4.43)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
T_TACAN	V90W3596CB	TACAN TIME TAG	ENT NAV PRELAND NAV RTLS NAV	S				
	V72M6636PA	MDM FF01 DSCR OUT SET	CD 00 CH 7 DL FMT 21/1 DL FMT 23/1 DL FMT 32/1 TACAN 1				93017G	*
	V72M6686PA	MDM FF02 DSCR OUT SET	CD 00 CH 7 DL FMT 21/1 DL FMT 23/1 DL FMT 32/1 TACAN 2				93017G	*
	V72M6736PA	MDM FF03 DSCR OUT SET	CD 00 CH 7 DL FMT 21/1 DL FMT 23/1 TACAN 3				93017G	*
	V74K1511BA	TACAN NO 1 MODE X/Y	DL FMT 21/1 DL FMT 23/A DL FMT 32/A DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 52/1 DL FMT 53/1 DL FMT 60/1 TACAN 1				93017G	*
	V74K1512BA	TACAN NO 1 A/A	DL FMT 21/1 DL FMT 23/A DL FMT 32/A DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 52/1 DL FMT 53/1 DL FMT 60/1 TACAN 1				93017G	*

TABLE 4.43.6.5-2. OUTPUT FUNCTIONAL PARAMETERS FROM TACTICAL AIR NAVIGATION SOP (G4.43)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74K1513BA	TACAN NO 1 T/R+RCV ONLY	DL FMT 21/1 DL FMT 23/A DL FMT 32/A DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 52/1 DL FMT 53/1 DL FMT 60/1				93017G	*
	V74K1611BA	TACAN NO 2 MODE X/Y	TACAN 1 DL FMT 21/1 DL FMT 23/A DL FMT 32/A DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 52/1 DL FMT 53/1 DL FMT 60/1				93017G	*
	V74K1612BA	TACAN NO 2 A/A	TACAN 2 DL FMT 21/1 DL FMT 23/A DL FMT 32/A DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 52/1 DL FMT 53/1 DL FMT 60/1				93017G	*
	V74K1613BA	TACAN NO 2 T/R+RCV ONLY	TACAN 2 DL FMT 21/1 DL FMT 23/A DL FMT 32/A DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 52/1 DL FMT 53/1 DL FMT 60/1				93017G	*

TABLE 4.43.6.5-2. OUTPUT FUNCTIONAL PARAMETERS FROM TACTICAL AIR NAVIGATION SOP (G4.43)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74K1711BA	TACAN NO 3 MODE X/Y	DL FMT 21/1 DL FMT 23/A DL FMT 32/A DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 52/1 DL FMT 53/1 DL FMT 60/1				93017G	*
	V74K1712BA	TACAN NO 3 A/A	TACAN 3 DL FMT 21/1 DL FMT 23/A DL FMT 32/A DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 52/1 DL FMT 53/1 DL FMT 60/1				93017G	*
	V74K1713BA	TACAN NO 3 T/R+RCV ONLY	TACAN 3 DL FMT 21/1 DL FMT 23/A DL FMT 32/A DL FMT 42/1 DL FMT 43/1 DL FMT 44/1 DL FMT 45/1 DL FMT 52/1 DL FMT 53/1 DL FMT 60/1				93017G	*
	V95M2864P	CGYB_TBRG_CMLK\$(1:)	TACAN 3 DL FMT 21/1 DL FMT 23/A				93017G	*
	V95M2865P	CGYB_TBRG_CMLK\$(2:)	DL FMT 21/1 DL FMT 23/A				93017G	*
	V95M2866P	CGYB_TBRG_CMLK\$(3:)	DL FMT 21/1 DL FMT 23/A				93017G	*
	V95M2867P	CGYB_TRNG_CMLK\$(1:)	DL FMT 21/1 DL FMT 23/A				93017G	*
	V95M2868P	CGYB_TRNG_CMLK\$(2:)	DL FMT 21/1 DL FMT 23/A				93017G	*
	V95M2869P	CGYB_TRNG_CMLK\$(3:)	DL FMT 21/1 DL FMT 23/A				93017G	*

TABLE 4.43.6.5-2. OUTPUT FUNCTIONAL PARAMETERS FROM TACTICAL AIR NAVIGATION SOP (G4.43)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
-----------	--------	--------------	-------------	-------	-----------	------------------	---------	-------------

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

**Rate Codes: (HDR Only) 1 = 1 sample/sec 2 = 5 samples/sec 3 = 12.5 samples/sec 4 = 25 samples/sec 5 = 100 samples/sec
 (HDR and LDR) A = 1 sample/sec B = 5 samples/sec C = 12.5 samples/sec D = 25 samples/sec E = 100 samples/sec**

This page intentionally left blank.

TABLE 4.43.6.5-3. TACTICAL AIR NAVIGATION SOP (G4.43) I-LOADS

FSSR NAME	MSID	ENG UNITS	DT	PR	D	S	PR FCTN	MC	CAT	LAST CR	R E V
TAC_BEARHI	V99U5715C	DEG	F	S	D	P	G4.43	00	TSI	DC0755	
TAC_BEARLO	V99U5716C	DEG	F	S	D	P	G4.43	00	TSI	DC0755	
TAC_RLTIM	V99U5520C	SEC	F	S	L	X	G4.43	00	TSI	DC0755	
TAC_RNGHI	V99U5717C	NMI	F	S	L	P	G4.43	00	TSI	DC0755	
TAC_RNGLO	V99U5718C	NMI	F	S	L	P	G4.43	00	TSI	DC0755	
TAC_STTIM	V99U5521C	SEC	F	S	L	P	G4.43	00	TSI	DC0755	
TNMFT	V99U5522C	FT/NMI	F	S	L	P	G4.43	00	TSI	92025B	

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

TABLE 4.43.6.5-4. TACTICAL AIR NAVIGATION SOP (G4.43) K-LOADS

FSSR NAME DESCRIPTION	MSID	MC	KLOAD VALUE	ENG UNITS	DT	P R	S	PR FCTN	LAST CR	R E V
--------------------------	------	----	-------------	-----------	----	--------	---	---------	---------	-------------

NO REQUIREMENTS

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

TABLE 4.43.6.5-5. TACTICAL AIR NAVIGATION SOP (G4.43) CONSTANTS

FSSR NAME DESCRIPTION	MSID	MC	CONSTANT VALUE	ENG UNITS	DT	P R	S	PR FCTN	LAST CR	R E V
--------------------------	------	----	----------------	-----------	----	--------	---	---------	---------	-------------

NO REQUIREMENTS

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

This page intentionally left blank.

4.44 MICROWAVE SCAN BEAM LANDING SYSTEM SOP

This section defines the detailed functional requirements and formulations for the microwave scan beam landing system (MSBLS) subsystem operating program (SOP). The MSBLS SOP defines software associated with MSBLS azimuth, elevation, and range data processing.

4.44.1 MSBLS Requirements Overview

4.44.1.1 MSBLS Functional Overview.

The MSBLS is a military Ku-band precision landing navigation aid that is used by the orbiter during final approach and landing. This flight phase is nominally from 12,000 feet altitude through touchdown and roll-out. By manual selection of the channel number (specific frequency) associated with the targeted runway approach, the orbiter will receive the range, azimuth, and elevation data that are provided by the MSBLS ground stations. The orbiter is equipped with three MSBLS transceivers for redundancy purposes. A single MSBLS subsystem is depicted in Figure 4.44.1.1-1, along with related interfacing equipment. Five major components are identified that are required for the MSBLS function: a control head, an antenna, a transceiver, a multiplexer/demultiplexer (MDM), and the on-board computer. The keyboard and the display electronics unit are also required for the crew to initiate self-test of the MSBLS transceiver.

Control of the MSBLS is via the control head two-position function switch and channel select thumbwheel. The two positions of the function switch are ON and OFF. When the function switch is placed in the ON position, power is applied to the transceiver and a power-present discrete returns from the transceiver to the MDM and then to the computer. Channel selection can be implemented only by the control head thumbwheel, and selection verification can only be accomplished via a visual check of the thumbwheel position by the crew (no channel number is available to the computer). Channel numbers of 1 to 10 are recognized by the transceiver. One thumbwheel per transceiver is provided, and the three thumbwheels are operated independently of each other.

The only software control of the MSBLS subsystem is the self-test of the transceiver. The software for the MSBLS self-test is defined in VU requirements.

MSBLS Data Processing (MSB_DATA). The MSBLS data processing module performs bit-stripping of the flag bits in the MSBLS input words. The module also performs fixed-point to floating-point conversion of the azimuth, elevation, and range data and scaling of azimuth and elevation data to radians and range data from nautical miles to feet.

MSBLS data for use in navigation shall be processed by the MSBLS SOP at a 320-millisecond interval during the same 160-millisecond period as the selection filter. The MSBLS SOP processing shall precede the selection filter processing.

MSBLS data for use in area navigation shall be processed by the MSBLS SOP at a 320-millisecond interval during the same 160-millisecond period as area navigation is executed. The MSBLS SOP processing shall precede the area navigation execution.

4.44.1.1.1 MSBLS Operating Modes.

The MSBLS operating modes are as follows:

- OFF mode
- ON mode
- SELF-TEST mode

The OFF mode occurs when the power-on discrete from the MSBLS transceiver is in the false or off state. The ON mode occurs when the transceiver is in operation as indicated by the power-on discrete. The self-test mode occurs when the transceiver is ON and is commanded to self-test by VU software.

4.44.1.1.2 MSBLS Digital Interfaces.

The MSBLS interfaces with the GPC via the serial digital multiplex/demultiplex input and output data channel. The digital I/O words are 16 bits in size, plus 3 bits for synchronization and 1 bit for odd parity. Throughout the MSBLS SOP the synchronization and parity bits are disregarded, and the most significant data bit in the word is designated bit 0, with the least significant bit being bit 15. The contents, format, and bit values of the azimuth, elevation, and range data words transmitted from the MSBLS to the MDM are illustrated in Figure 4.44.1.1.2-1. The MSBLS data words transmitted from the MDM to GPC are illustrated in Figures 4.44.1.1.2-2, -3, and -4.

Azimuth Angle Data Word. The 16 bits of the azimuth data word passed on to the computer are divided among three functions: (1) azimuth data, (2) self-test status, and (3) data validity. The first bit (bit 0) of the 14-bit azimuth data is the sign bit. As the orbiter approaches the runway, a positive azimuth signal indicates that the orbiter is positioned to the right of the runway centerline. Conversely, a negative value indicates that the orbiter is to the left of the centerline. The value of the azimuth data (in degrees) is relative to the centerline and is referenced to the azimuth ground scanner location. Bit 1 is the most significant bit (16 degrees) of the binary-coded data.

Bit 14 of the data word indicates whether the transceiver is in self-test or not. Bit 15 indicates the validity of the azimuth data whether the transceiver is in self-test or not.

Elevation Angle Data Word. The 16 bits of the elevation angle data word are divided among four functions: (1) Bit 0 represents the status of the automatic gain control (AGC) and the mixer crystal current, (2) bits 1 through 13 are the elevation angle data, (3) bit 14 is the self-test mode indicator, and (4) bit 15 is the elevation angle data validity. Bit 0 can change from zero to one during self-test if an AGC failure is indicated by built-in test equipment or at any time if the mixer crystal current monitor indicates a failure. The elevation angle data (in degrees) are relative to the horizontal with the elevation ground scanner as reference point. Bit 1 (16 degrees) is the most significant bit in this binary-coded word.

Range Data Word. The 16 bits of the range data word are divided among four functions: (1) Bit 0 represents the status of AGC and the mixer crystal current, (2) bits 1 through 13 are the range data, (3) bit 14 is the self-test mode indicator, and (4) bit 15 is range data validity. Range word bit 0 provides an identical function to elevation word bit 0. The range data (in nautical miles) are relative to the ground distance measuring equipment antenna. Bit 1 (10.36 nmi) is the most significant bit in the binary-coded data word.

4.44.1.1.3 Other MSBLS Interfaces.

The MSBLS transceiver requires the following interfaces:

- Power-on discrete
- Self-test command

The power-on discrete is generated by the MSBLS and indicates the transceiver is in operation.

The self-test command is generated by the VU software and causes the transceiver to perform self-test. The self-test condition is triggered in the transceiver anytime a serial digital word is output to the MSBLS transceiver.

4.44.1.2 MSBLS SOP Overview.

The MSBLS SOP processes MSBLS data for use by other program elements. The inputs and outputs for the MSBLS SOP modules are stored in locations labeled as to parameter and transceiver.

The MSBLS SOP processing is performed at the highest rate required by the user: selection filter at 3.125 Hz, area navigation at 3.125 Hz, and FDIR at 0.25 Hz. The MSBLS SOP examines the dedicated display source selection signals to determine execution of the MSBLS SOP. Refer to Figure 4.44.1.2-1 for a MSBLS SOP signal flow diagram.

MSBLS Input Processing (MSB_IN). The MSBLS input processing module identifies and accepts new inputs and controls execution or nonexecution of the MSBLS data processing module. The MSBLS input processing functional module shall input the range, azimuth, and elevation. If the input data from one or more of the LRU's are changed by one or more bits, the transaction time shall be stored as the time tag for those parameters; and on the next cycle, the input data shall be made available to the user and also stored as old data. (Azimuth shall be updated after 320 milliseconds but no more than 480 milliseconds with or without a detectable change.)

4.44.1.3 MSBLS SOP Functional Module Summary.

The functional modules that constitute the MSBLS SOP are identified in Table 4.44.1.3-1. Each of the modules, including functions and formulation, are presented in detail in Section 4.44.6.

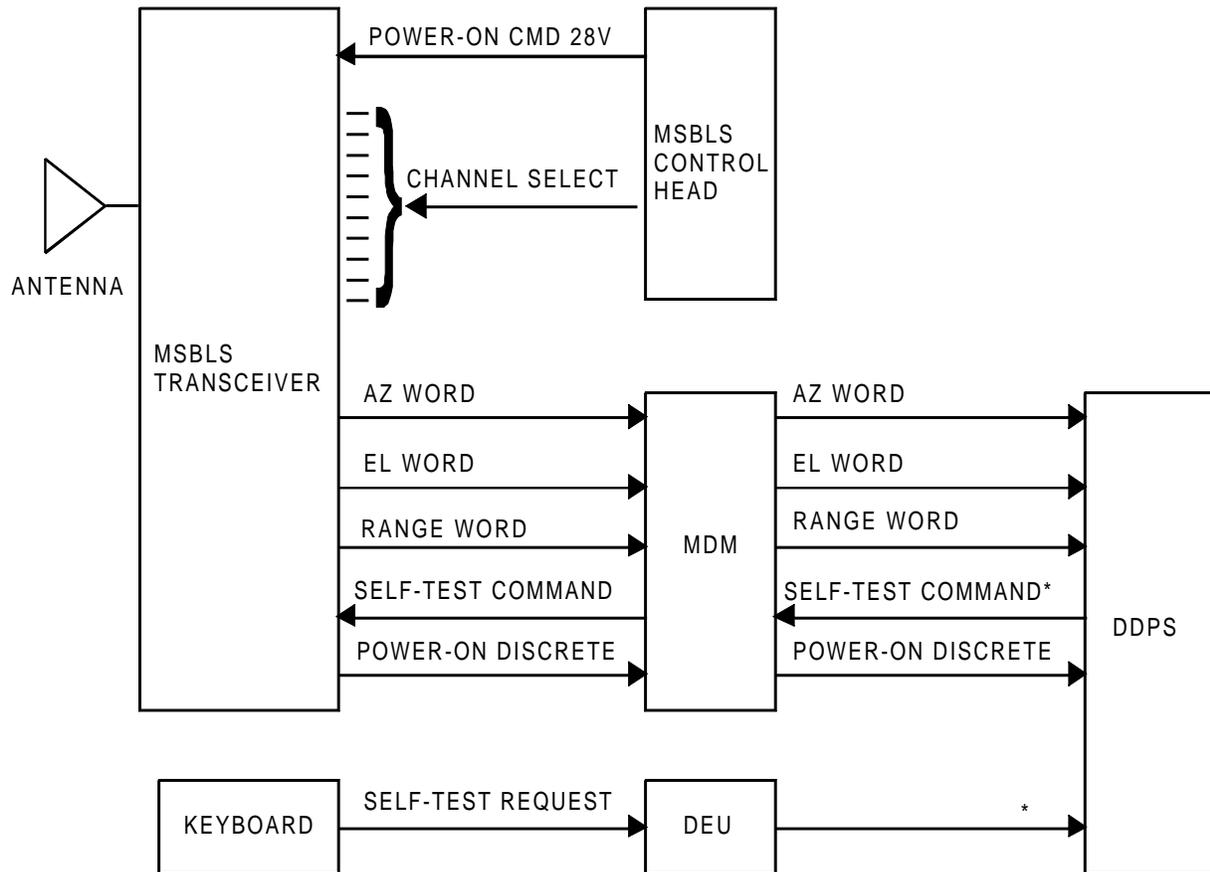
4.44.1.4 Reference Documents.

MC409-0017 Navigation Set, Microwave Scan Beam Landing System.

4.44.1.5 Software Protection.

The software implementation of the MSBLS SOP shall protect against program checks and error interrupts caused by transient-type errors or bad data inputs. Specifically, the MSBLS SOP shall protect against all of the error conditions identified in paragraph 3.6 that are applicable to the MSBLS software design and code.

Table 4.44.1.3-1. MSBLS SOP Functional Modules		
Module Name	Applicable Paragraph	Module Description
MSB_IN	4.44.6.1	MSBLS Input Processing
MSB_DATA	4.44.6.2	MSBLS Data Processing



* SOFTWARE REQUIREMENTS FOR SELF-TEST ARE SPECIFIED IN THE VU CPDS.

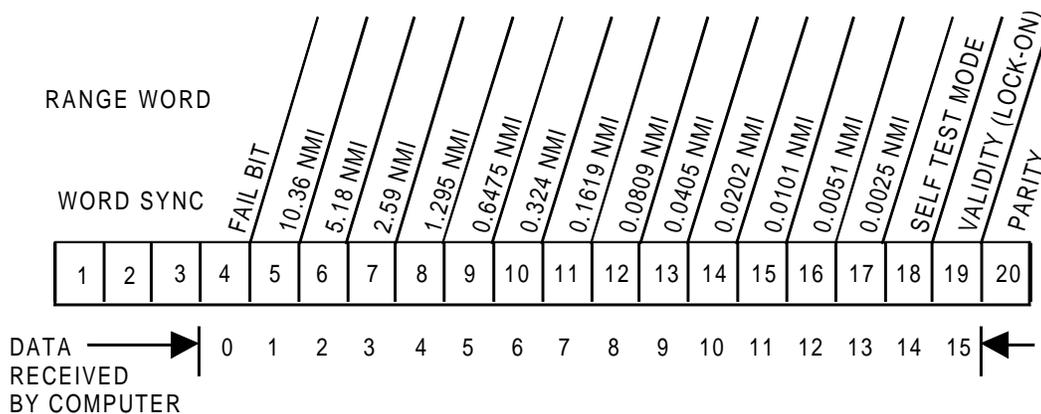
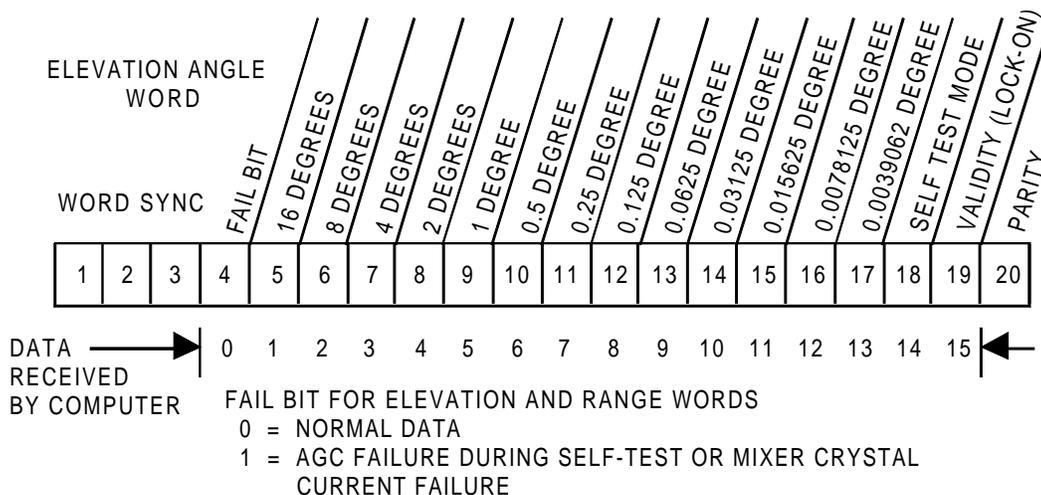
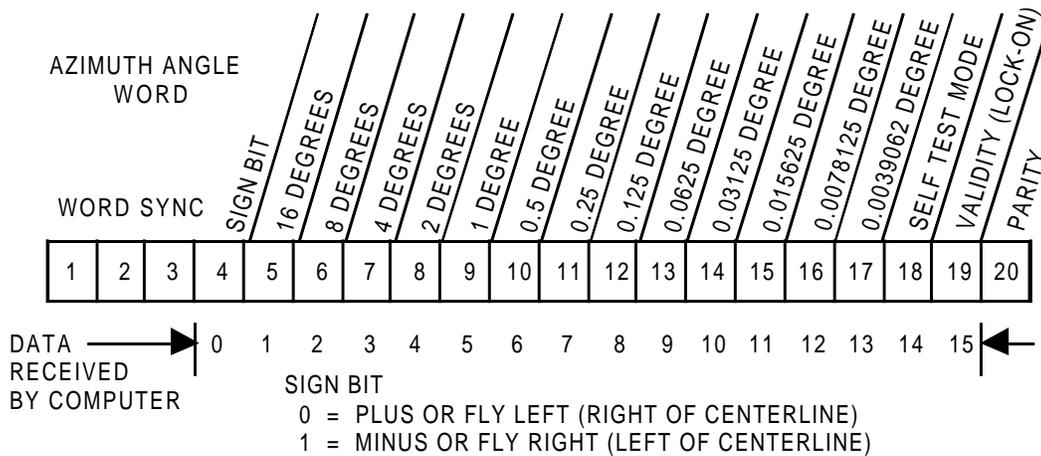
FIGURE 4.44.1.1-1. Microwave Scanning Beam Landing System, Single-String Diagram

SELF-TEST STATUS BITS (ALL)

0 = NOT IN SELF-TEST
1 = PERFORMING SELF-TEST

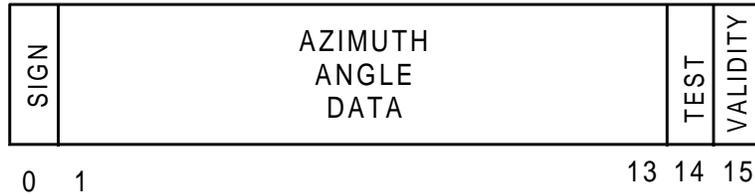
VALIDITY BITS (ALL)

0 = VALID DATA
1 = NONVALID DATA



NOTE: BINARY VALUES ARE ROUNDED

FIGURE 4.44.1.1.2-1. MSBLS Serial Data Words (MSBLS to MDM)



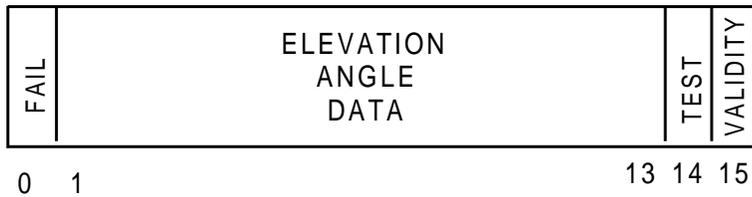
BIT 0 SIGN 0 = PLUS OR FLY LEFT
 1 = MINUS OR FLY RIGHT

1	16 DEG
2	8 DEG
3	4 DEG
4	2 DEG
5	1 DEG
6	0.5 DEG
7	0.25 DEG
8	0.125 DEG
9	0.0625 DEG
10	0.03125 DEG
11	0.015625 DEG
12	0.0078125 DEG
13	0.0039062 DEG

14	TEST	0 = NOT IN SELF-TEST
		1 = IN SELF-TEST

15	VALIDITY	0 = VALID DATA
		1 = INVALID DATA

FIGURE 4.44.1.1.2-2. MSBLS Azimuth Angle Word Format (MDM to GPC)



BIT 0 FAIL 0 = NORMAL DATA
 1 = AGC FAILURE (DURING SELF-TEST) OR MIXER
 CRYSTAL CURRENT FAILURE

1	16 DEG
2	8 DEG
3	4 DEG
4	2 DEG
5	1 DEG
6	0.5 DEG
7	0.25 DEG
8	0.125 DEG
9	0.0625 DEG
10	0.03125 DEG
11	0.015625 DEG
12	0.0078125 DEG
13	0.0039062 DEG
14	TEST 0 = NOT IN TEST 1 = IN SELF-TEST
15	VALIDITY 0 = VALID DATA 1 = INVALID DATA

FIGURE 4.44.1.1.2-3. MSBLS Elevation Angle Word Format (MDM to GPC)



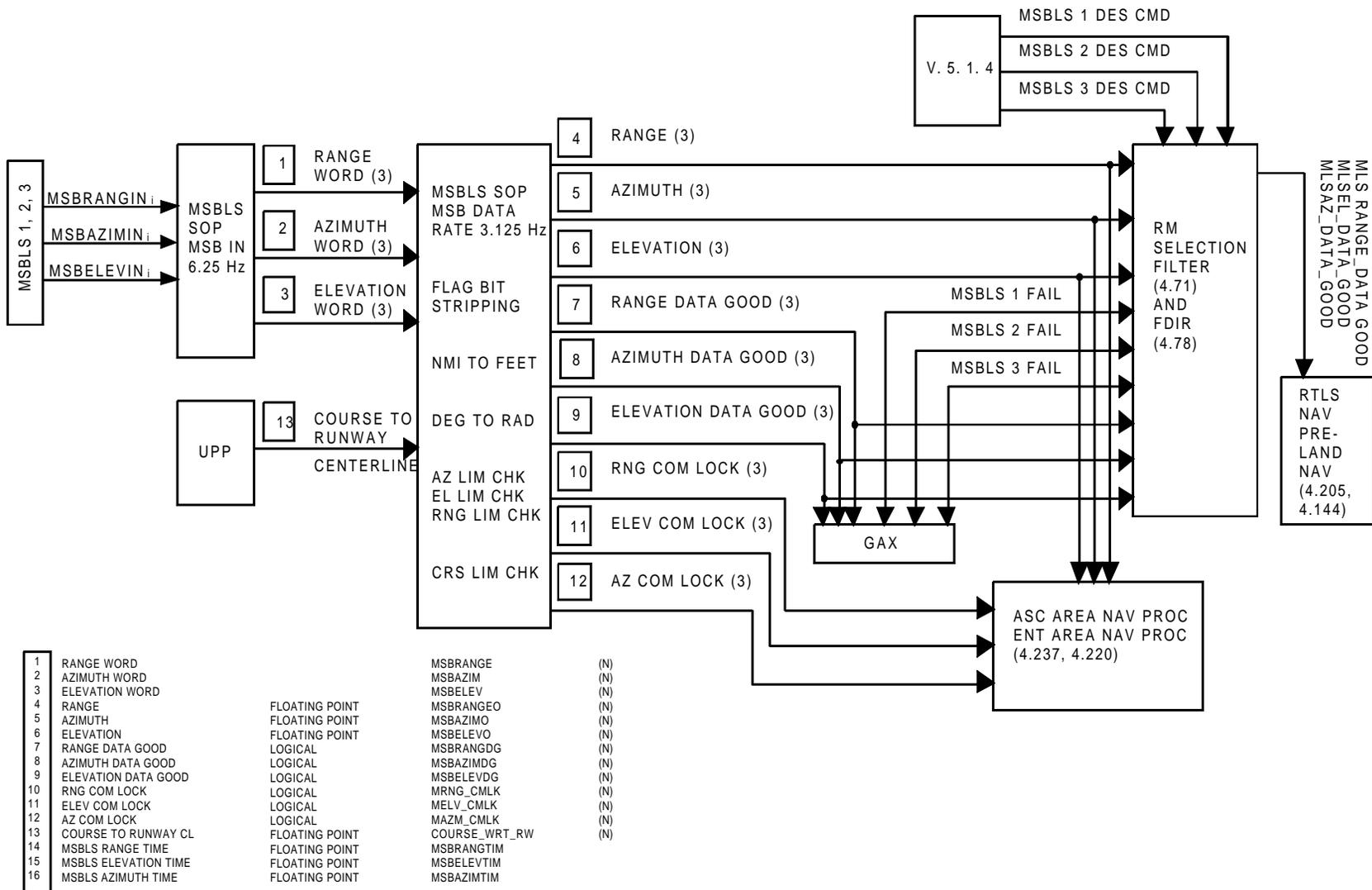
BIT 0 FAIL 0 = NORMAL DATA
 1 = AGC FAILURE (DURING SELF-TEST) OR
 MIXER CRYSTAL CURRENT FAILURE

1	10.36 NMI
2	5.18 NMI
3	2.59 NMI
4	1.295 NMI
5	0.6475 NMI
6	0.324 NMI
7	0.1619 NMI
8	0.0809 NMI
9	0.0405 NMI
10	0.0202 NMI
11	0.0101 NMI
12	0.0051 NMI
13	0.0025 NMI

14	TEST STATUS	0 = NOT IN SELF-TEST
		1 = IN SELF-TEST
15	VALIDITY	0 = VALID DATA
		1 = INVALID DATA

FIGURE 4.44.1.1.2-4. MSBLS Range Word Format (MDM to GPC)

This page intentionally left blank.



* FDIR IS AVAILABLE IN OPS 3 ONLY

FIGURE 4.44.1.2-1. MSBLS Data Signal Flow

4.44.2 Definitions.

This section is not applicable to the MSBLS SOP.

4.44.3 Functional Requirements Traceability.

The detailed requirements and formulations contained herein satisfy the GN&C, Level B, CPDS functional requirements for the MSBLS SOP.

4.44.4 Interface Parameters.

The interface parameters required to properly execute the MSBLS functional module are given in Table 4.44.4-1.

4.44.5 Parameter Initialization.

Table 4.44.5-1 presents the conversion factor for nautical miles to feet taken from the Level A document.

This section defines the I-load parameters associated with the MSBLS SOP. Table 4.44.5-2 contains the I-load parameters and identifies typical I-load values. The actual values will be specified later in the official I-load document. The codes designated in the constant column of Table 4.44.5-2 are to be interpreted as follows: "SX" indicates a soft constant that may change as the software is refined; "MX" indicates a mission/LRU dependent constant that is expected to change from mission to mission; and "X" indicates a hard constant that probably will not change.

4.44.6 Detailed Requirements and Formulation.

This section defines the MSBLS SOP functional modules in sufficient detail to create a software design and generate computer code. Each module defines the functional requirements, interface requirements, detailed requirements and formulations, and special considerations if applicable. A flow diagram is also included for most modules. However, the flow diagrams are supplementary material and are not intended to convey requirements or to place restrictions on software design or coding.

4.44.6.1 MSBLS Input Processing (MSB_IN).

The purpose of the MSB_IN functional module is to input data and control execution or nonexecution of the MSB_DATA functional module.

A. Functional Requirements. The MSB_IN module is required to perform the following function for each MSBLS:

Evaluate input data words for any change of value as compared to the last sampling. If a change has occurred, the transaction time tag will be stored as the measurement time tag, and on the next cycle, the input data will be stored and made available to the user. Since the azimuth data are changing slowly when the vehicle is lining up with the runway center line, the azimuth data will be accepted without a change if the last accepted input data are over 320 milliseconds old, but no more than 480 milliseconds.

B. Interface Requirements. The input and output parameters associated with the MSB_IN functional module are contained in Table 4.44.4-1.

C. Detailed Requirements and Formulations. The following sequence of operations shall be performed for each MSBLS transceiver at a rate of 6.25 times per second:

1. If this is not a range update cycle (RANGE_UPDATE=OFF), then compare each MSBLS input range word (MSBRANGIN) to the range word (MSBRANGE). If the input range word differs from the range word, store CLOCKTIME as the range time (RANGE_TIME) and set the range update flag to on. However, if this is a range update cycle (RANGE_UPDATE=ON), then update MSBRANGE with MSBRANGIN, MSBRANGTIM with RANGE_TIME and set the range update flag to off.
2. If this is not an elevation update cycle (ELEVATION_UPDATE = OFF), then compare each MSBLS input elevation word (MSBELEVIN) to the elevation word (MSBELEV). If the input elevation word differs from the elevation word, store CLOCKTIME as the elevation time (ELEVATION_TIME) and set the elevation update flag to on. However, if this is an elevation update cycle (ELEVATION_UPDATE=ON), then update MSBELEV with MSBELEVIN, MSBELEVTIM with ELEVATION_TIME and set the elevation update flag to off.
3. If this is not an azimuth update cycle (AZIMUTH_UPDATE = OFF), then compare each MSBLS input azimuth word (MSBAZIMIN) to the azimuth word (MSBAZIM) and perform the test on AZIMUTH_PASS flag. If the input azimuth word differs from the azimuth word or this is a second cycle since the last update (AZIMUTH_PASS = ON), then store CLOCKTIME as the azimuth time (AZIMUTH_TIME), set the azimuth update flag to on and set AZIMUTH_PASS to off, else set AZIMUTH_PASS to on. However, if this is an azimuth update cycle (AZIMUTH_UPDATE=ON), then update MSBAZIM with MSBAZIMIN, MSBAZIMTIM with AZIMUTH_TIME and set the azimuth update flag to off.

Figure 4.44.6.1-1 presents a flow diagram for the MSB_IN functional module.

4.44.6.2 MSBLS Data Processing (MSB_DATA).

The purpose of this MSB_DATA functional module is to process the MSBLS input data.

A. Functional Requirements. The MSB_DATA module is required to perform the following functions for each MSBLS:

1. Process the data words provided by the MSB_IN functional module and remove the flag bits, using these bits to drive flags for other functional modules.
2. Check the values of the range, heading, azimuth, and elevation for validity and magnitude within the specified limits.

Set the validity flags on words going to the selection filter and area nav.

3. Compute the MSBLS range, azimuth, and elevation values by performing the required scaling and conversion to obtain floating point data in required units.

B. Interface Requirements. The input and output parameters associated with the MSB_DATA functional module are contained in Table 4.44.4-1.

C. Detailed Requirements and Formulations. The following operations shall be performed for each applicable MSBLS transceiver at a rate of 3.125 Hz:

1. MSBLS Range Out. The MSBLS range data (MSBRANGE_O) are contained in the MSBLS range word (MSBRANGIN) as a 13-bit binary word (Figure 4.44.1.1.2-4). The most significant bit is bit 1 with a value of 10.36 nmi. Compute MSBRANGE_O in floating point and scale the value in feet using the conversion factor ft-per-nmi (MLSNMI_FT). Set MSBLS range lock-on and MSBLS range commfault flags to false.
2. MSBLS Range Lock-On and Comlock. The MSBLS range lock-on (MSBRANGDG) is driven by bit 15 of the MSBLS range word (MSBRANGE). MSBRANGDG is a two-state flag indicating data good or bad. True indicates data good. If MSBRANGE_i bit 15 is zero, then first test to determine if there is no range commfault and set MRNG_CMLK_i flag to true; else leave MRNG_CMLK_i false. Then set MSBRANGDG_i to true; else leave MSBRANGDG false.
3. MSBLS Elevation Out. The MSBLS elevation data (MSBELEVO) are contained in the MSBLS elevation word (MSBELEV) as a 13-bit binary word (Figure 4.44.1.1.2-3). The most significant bit is bit 1 with a value of 16 degrees. Compute MSBELEVO_i in floating point and scale in radians. Set MSBLS elevation lock-on and MSBLS elevation commfault flags to false.
4. MSBLS Elevation Lock-On and Comlock. The MSBLS elevation lock-on (MSBELEVDG) is driven by bit 15 of the MSBLS elevation word (MSBELEV). MSBELEVDG is a two-state flag indicating data good or bad. True indicates data good.

If MSBELEV bit 15 is zero, then first test to determine if there is no elevation COMFAULT_i and set MELV_CMLK_i to true; otherwise leave false. Next, test to determine if the absolute course with respect to the runway | COURSE_WRT_RW adjusted to the range -180 to +180 | is less than COURSE_WRT_RW_LIM, the desired limit; test to determine if MRNG_CMLK_i is true; and finally test whether the elevation is within the desired minimum MSBELMIN_LIM and desired maximum MSBELMAX_LIM.

If all these conditions are met, set MSBELEVDG_i to true.

5. MSBLS Fail Flag.

If MSBELV_i bit 0 = 1

Then set MSBELVSTB_i to true;

else set MSBELVSTB_i to false

The MSBLS failure flag (MSBELVSTB) is driven by crystal current failure or AGC failure during self-test

6. MSBLS Azimuth Out. The MSBLS azimuth data (MSBAZIMO) are contained in the MSBLS azimuth word (MSBAZIM) as a 13-bit binary word plus sign (Figure 4.44.1.1.2-2). The most significant bit is bit 1 with a value of 16 degrees. The sign is in bit 0 with + = 0 and - = 1. Plus indicates fly left (i.e., orbiter to the right of the centerline), and minus indicates fly right (i.e., orbiter to the left of the centerline). Compute MSBAZIMO_i in floating point and scale in radians. Set MSBLS azimuth lock-on and MSBLS azimuth commfault flags to false.
7. MSBLS Azimuth Lock-On and Comlock. The MSBLS azimuth lock-on (MSBAZIMDG) is driven by bit 15 of the MSBLS azimuth word (MSBAZIM). MSBAZIMDG is a two-state flag indicating data good or bad. True indicates data good.

If MSBAZIM bit 15 is zero, then first test to determine if there is no azimuth COMMFAULT_i and set MAZM_CMLK_i to true; otherwise leave false. Next, test to determine if the absolute course with respect to the runway | COURSE_WRT_RW adjusted to the range -180 to +180 | is less than COURSE_WRT_RW_LIM, the desired limit. If the range comlock MRNG_CMK_i is true and if the range is less than a maximum limit MSBRNG_LIM and if the absolute value of the azimuth is less than a desired maximum MSBAZ_LIM, set azimuth lock-on MSBAZIMDG_i to true.

Note that comlock processing is only required on area navigation cycles. Figure 4.44.6.2-1 presents a flow diagram for the MSB_DATA functional module.

4.44.6.3 Applicable Tables for MSBLS SOP.

The IDD inputs and outputs are listed in Tables 4.44.6.3-1 and 4.44.6.3-2. Values for the I-loads are contained in the I-loads requirements document (JSC-19350); however, I-load definitions applicable to this principal function are listed in Table 4.44.6.3-3. K-loads are listed in Table 4.44.6.3-4. Constants are listed in Table 4.44.6.3-5.

Table 4.44.4-1. MSBLS SOP Interface Table								
Name	MSID/ Mnemonic	Description	Input Source	Output Destination	Signal Type	Range	Units	Rate (Hz)
MSBRANGIN(1)	V74M1050P G08206	MSBLS NO 1 RANGE WORD (16 BIT PARENT)	MSBLS 1		HMD		NA	6.25
	V74X1051B	MSBLS NO 1 RNG AGC-XTAL CUR FAIL	MSBLS 1		LOGICAL	0, 1	NA	6.25
	V74H1052B	MSBLS NO 1 RANGE DATA	MSBLS 1		FIXED PT	0-20.7	NM	6.25
	V74X1053B	MSBLS NO 1 RANGE TEST	MSBLS 1		LOGICAL	0, 1	NA	6.25
	V74X1054B	MSBLS NO 1 RANGE VALIDITY	MSBLS 1		LOGICAL	0, 1	NA	6.25
MSBRANGIN(2)	V74M1150P G08207	MSBLS NO 2 RANGE WORD (16 BIT PARENT)	MSBLS 2		HMD		NA	6.25
	V74X1151B	MSBLS NO 2 RNG AGC-XTAL CUR FAIL	MSBLS 2		LOGICAL	0, 1	NA	6.25
	V74H1152B	MSBLS NO 2 RANGE DATA	MSBLS 2		FIXED PT	0-20.7	NM	6.25
	V74X1153B	MSBLS NO 2 RANGE TEST	MSBLS 2		LOGICAL	0, 1	NA	6.25
	V74X1154B	MSBLS NO 2 RANGE VALIDITY	MSBLS 2		LOGICAL	0, 1	NA	6.25
MSBRANGIN(3)	V74M1250P G08208	MSBLS NO 3 RANGE WORD (16 BIT PARENT)	MSBLS 3		HMD		NA	6.25
	V74X1251B	MSBLS NO 3 RNG AGC-XTAL CUR FAIL	MSBLS 3		LOGICAL	0, 1	NA	6.25
	V74H1252B	MSBLS NO 3 RANGE DATA	MSBLS 3		FIXED PT	0-20.7	NM	6.25
	V74X1253B	MSBLS NO 3 RANGE TEST	MSBLS 3		LOGICAL	0, 1	NA	6.25
	V74X1254B	MSBLS NO 3 RANGE VALIDITY	MSBLS 3		LOGICAL	0, 1	NA	6.25
MSBELEVIN(1)	V74M1030P G08203	MSBLS NO 1 ELEVATION WORD (16 BIT PARENT)	MSBLS 1		HMD		NA	6.25
	V74X1031B	MSBLS NO 1 EL AGC-XTAL CUR FAIL	MSBLS 1		LOGICAL	0, 1	NA	6.25
	V74H1032B	MSBLS NO 1 ELEVATION DATA	MSBLS 1		FIXED PT	0-30	DEG	6.25
	V74X1033B	MSBLS NO 1 ELEVATION TEST	MSBLS 1		LOGICAL	0, 1	NA	6.25
	V74X1034B	MSBLS NO 1 ELEVATION VALIDITY	MSBLS 1		LOGICAL	0, 1	NA	6.25

Table 4.44.4-1. MSBLS SOP Interface Table

Name	MSID/ Mnemonic	Description	Input Source	Output Destination	Signal Type	Range	Units	Rate (Hz)
MSBELEVIN(2)	V74M1130P G08204	MSBLS NO 2 ELEVATION WORD (16 BIT PARENT)	MSBLS 2		HMD		NA	6.25
	V74X1131B	MSBLS NO 2 EL AGC-XTAL CUR FAIL	MSBLS 2		LOGICAL	0, 1	NA	6.25
	V74H1132B	MSBLS NO 2 ELEVATION DATA	MSBLS 2		FIXED PT	0-30	DEG	6.25
	V74X1133B	MSBLS NO 2 ELEVATION TEST	MSBLS 2		LOGICAL	0, 1	NA	6.25
	V74X1134B	MSBLS NO 2 ELEVATION VALIDITY	MSBLS 2		LOGICAL	0, 1	NA	6.25
MSBELEVIN(3)	V74M1230P G08205	MSBLS NO 3 ELEVATION WORD (16 BIT PARENT)	MSBLS 3		HMD		NA	6.25
	V74X1231B	MSBLS NO 3 EL AGC-XTAL CUR FAIL	MSBLS 3		LOGICAL	0, 1	NA	6.25
	V74H1232B	MSBLS NO 3 ELEVATION DATA	MSBLS 3		FIXED PT	0-30	DEG	6.25
	V74X1233B	MSBLS NO 3 ELEVATION TEST	MSBLS 3		LOGICAL	0, 1	NA	6.25
	V74X1234B	MSBLS NO 3 ELEVATION VALIDITY	MSBLS 3		LOGICAL	0, 1	NA	6.25
MSBAZIMIN(1)	V74M1010P G08200	MSBLS NO. 1 AZIMUTH WORD (16 BIT PARENT)	MSBLS 1		HMD		NA	6.25
	V74H1012B	MSBLS NO 1 AZIMUTH DATA	MSBLS 1		FIXED PT	0-31.2	DEG	6.25
	V74X1013B	MSBLS NO 1 AZIMUTH TEST	MSBLS 1		LOGICAL	0, 1	NA	6.25
	V74X1014B	MSBLS NO 1 AZIMUTH VALIDITY	MSBLS 1		LOGICAL	0, 1	NA	6.25
MSBAZIMIN(2)	V74M1110P G08201	MSBLS NO 2 AZIMUTH WORD (16 BIT PARENT)	MSBLS 2		HMD		NA	6.25
	V74H1112B	MSBLS NO 2 AZIMUTH DATA	MSBLS 2		FIXED PT	0-31.2	DEG	6.25
	V74X1113B	MSBLS NO 2 AZIMUTH TEST	MSBLS 2		LOGICAL	0, 1	NA	6.25
	V74X1114B	MSBLS NO 2 AZIMUTH VALIDITY	MSBLS 2		LOGICAL	0, 1	NA	6.25

Table 4.44.4-1. MSBLS SOP Interface Table

Name	MSID/ Mnemonic	Description	Input Source	Output Destination	Signal Type	Range	Units	Rate (Hz)
MSBAZIMIN(3)	V74M1210P G08202	MSBLS NO 3 AZIMUTH WORD (16 BIT PARENT)	MSBLS 3		HMD		NA	6.25
	V74H1212B	MSBLS NO 3 AZIMUTH DATA	MSBLS 3		FIXED PT	0-31.2	DEG	6.25
	V74X1213B	MSBLS NO 3 AZIMUTH TEST	MSBLS 3		LOGICAL	0, 1	NA	6.25
	V74X1214B	MSBLS NO 3 AZIMUTH VALIDITY	MSBLS 3		LOGICAL	0, 1	NA	6.25
MSBLS_CF(1)	V91X3550X G46263	MSBLS 1 COMMFAULT	FCOS		LOGICAL	0, 1	NA	3.125
MSBLS_CF(2)	V91X3560X G46264	MSBLS 2 COMMFAULT	FCOS		LOGICAL	0, 1	NA	3.125
MSBLS_CF(3)	V91X3570X G46265	MSBLS 3 COMMFAULT	FCOS		LOGICAL	0, 1	NA	3.125
CLOCK TIME	V91W5000C G07829	CLOCK COMPUTER (GMT)	FCOS		FLOAT PT	NA	SEC	6.25
MSBRANGE0(1)	V95H6102C G08119	MSBLS 1 RANGE OUT		SF (4.71)	FLOAT	0 – 125K	FEET	3.125
				MSBLS FDIR (4.78)	PT			0.26
				ASC AREA NAV PROC (4.237)				1.04
				ENT AREA NAV PROC (4.220)				1.04
MSBRANGE0(2)	V95H6202C G08120	MSBLS 2 RANGE OUT		SF (4.71)	FLOAT	0 – 125K	FEET	3.125
				MSBLS FDIR (4.78)	PT			0.26
				ASC AREA NAV PROC (4.237)				1.04
				ENT AREA NAV PROC (4.220)				1.04
MSBRANGE0(3)	V95H6302C G08121	MSBLS 3 RANGE OUT		SF (4.71)	FLOAT	0 – 125K	FEET	3.125
				MSBLS FDIR (4.78)	PT			0.26
				ASC AREA NAV PROC (4.237)				1.04
				ENT AREA NAV PROC (4.220)				1.04

Table 4.44.4-1. MSBLS SOP Interface Table

Name	MSID/ Mnemonic	Description	Input Source	Output Destination	Signal Type	Range	Units	Rate (Hz)
MSBELEVO(1)	V95H6142C G08116	MSBLS 1 ELEVATION OUT		SF (4.71)	FLOAT	0 – 0.6	RAD	3.125
				MSBLS FDIR (4.78)	PT			0.26
				ASC AREA NAV PROC (4.237)				1.04
				ENT AREA NAV PROC (4.220)				1.04
MSBELEVO(2)	V95H6242C G08117	MSBLS 2 ELEVATION OUT		SF (4.71)	FLOAT	0 – 0.6	RAD	3.125
				MSBLS FDIR (4.78)	PT			0.26
				ASC AREA NAV PROC (4.237)				1.04
				ENT AREA NAV PROC (4.220)				1.04
MSBELEVO(3)	V95H6342C G08118	MSBLS 3 ELEVATION OUT		SF (4.71)	FLOAT	0 – 0.6	RAD	3.125
				MSBLS FDIR (4.78)	PT			0.26
				ASC AREA NAV PROC (4.237)				1.04
				ENT AREA NAV PROC (4.220)				1.04
MSBAZIMO(1)	V95H6122C G08113	MSBLS 1 AZIMUTH OUT		SF (4.71)	FLOAT	+/-0.6	RAD	3.125
				MSBLS FDIR (4.78)	PT			0.26
				ASC AREA NAV PROC (4.237)				1.04
				ENT AREA NAV PROC (4.220)				1.04
MSBAZIMO(2)	V95H6222C G08114	MSBLS 2 AZIMUTH OUT		SF (4.71)	FLOAT	+/-0.6	RAD	3.125
				MSBLS FDIR (4.78)	PT			0.26
				ASC AREA NAV PROC (4.237)				1.04
				ENT AREA NAV PROC (4.220)				1.04

Table 4.44.4-1. MSBLS SOP Interface Table

Name	MSID/ Mnemonic	Description	Input Source	Output Destination	Signal Type	Range	Units	Rate (Hz)
MSBAZIMO(3)	V95H6322C G08115	MSBLS 3 AZIMUTH OUT		SF (4.71)	FLOAT	+/-0.6	RAD	3.125
				MSBLS FDIR (4.78)	PT			0.26
				ASC AREA NAV PROC (4.237)				1.04
				ENT AREA NAV PROC (4.220)				1.04
MSBRANGDG(1)	V95X6104X G08110	MSBLS 1 RANGE LOCKED ON		SF 4.71 GAX 4.110 MSB FDIR 4.78	LOGICAL	0, 1	NA	3.125 1.04 0.26
MSBRANGDG(2)	V95X6204X G08111	MSBLS 2 RANGE LOCKED ON		SF 4.71 GAX 4.110 MSB FDIR 4.78	LOGICAL	0, 1	NA	3.125 1.04 0.26
MSBRANGDG(3)	V95X6304X G08112	MSBLS 3 RANGE LOCKED ON		SF 4.71 GAX 4.110 MSB FDIR 4.78	LOGICAL	0, 1	NA	3.125 1.04 0.26
MSBELEVDG(1)	V95X6144X G08107	MSBLS 1 ELEVATION LOCKED ON		SF 4.71 GAX 4.110 MSB FDIR 4.78	LOGICAL	0, 1	NA	3.125 1.04 0.26
MSBELEVDG(2)	V95X6244X G08108	MSBLS 2 ELEVATION LOCKED ON		SF 4.71 GAX 4.110 MSB FDIR 4.78	LOGICAL	0, 1	NA	3.125 1.04 0.26
MSBELEVDG(3)	V95X6344X G08109	MSBLS 3 ELEVATION LOCKED ON		SF 4.71 GAX 4.110 MSB FDIR 4.78	LOGICAL	0, 1	NA	3.125 1.04 0.26
MSBAZIMDG(1)	V95X6124X G08104	MSBLS 1 AZIMUTH LOCKED ON		SF 4.71 GAX 4.110 MSB FDIR 4.78	LOGICAL	0, 1	NA	3.125 1.04 0.26
MSBAZIMDG(2)	V95X6224X G08105	MSBLS 2 AZIMUTH LOCKED ON		SF 4.71 GAX 4.110 MSB FDIR 4.78	LOGICAL	0, 1	NA	3.125 1.04 0.26
MSBAZIMDG(3)	V95X6324X G08106	MSBLS 3 AZIMUTH LOCKED ON		SF 4.71 GAX 4.110 MSB FDIR 4.78	LOGICAL	0, 1	NA	3.125 1.04 0.26
MSBELVSTB(1)	V95X6148X G46260	MSBLS NO. 1 ELEVATION SELF-TEST BAD		MSBLS FDIR 4.78	LOGICAL	0, 1	NA	0.26

Table 4.44.4-1. MSBLS SOP Interface Table

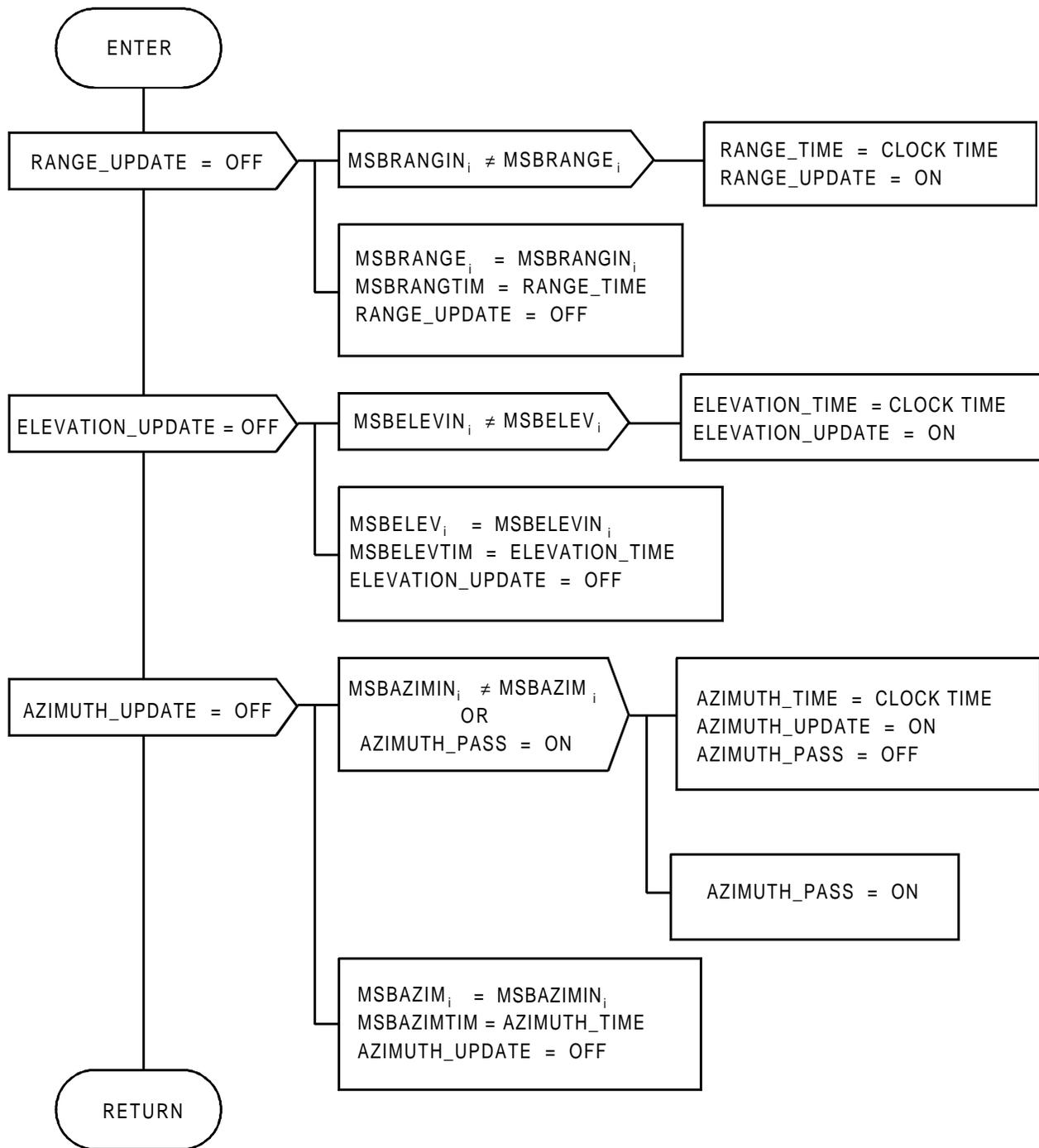
Name	MSID/ Mnemonic	Description	Input Source	Output Destination	Signal Type	Range	Units	Rate (Hz)
MSBELVSTB(2)	V95X6248X G46261	MSBLS NO. 2 ELEVATION SELF-TEST BAD		MSBLS FDIR 4.78	LOGICAL	0, 1	NA	0.26
MSBELVSTB(3)	V95X6348X G46262	MSBLS NO. 3 ELEVATION SELF-TEST BAD		MSBLS FDIR 4.78	LOGICAL	0, 1	NA	0.26
MSBRANTIM	V95W6000C G08122	MSBLS RANGE MEASUREMENT TIME		PRELAND NAV 4.144 TLM RTLS NAV 4.205	FLOAT PT	NA	SEC	3.125
MSBELEVTIM	V95W6020C G08124	MSBLS ELEVATION MEASUREMENT TIME		PRELAND NAV 4.144 TLM RTLS NAV 4.205	FLOAT PT	NA	SEC	3.125
MSBAZIMTIM	V95W6010C G08123	MSBLS AZIMUTH MEASUREMENT TIME		PRELAND NAV 4.144 TLM RTLS NAV 4.205	FLOAT PT	NA	SEC	3.125
MRNG_CMLK(1)	V95X6056X G50729	MSBLS 1 RANGE COMMFAULT/ LOCKON FLAG		ASC AREA NAV PROC (4.237) ENT AREA NAV PROC (4.220)	LOGICAL	0, 1	NA	1.04 1.04
MRNG_CMLK(2)	V95X6062X G50730	MSBLS 2 RANGE COMMFAULT/ LOCKON FLAG		ASC AREA NAV PROC (4.237) ENT AREA NAV PROC (4.220)	LOGICAL	0, 1	NA	1.04 1.04
MRGN_CMLK(3)	V95X6067X G50731	MSBLS 3 RANGE COMMFAULT/ LOCKON FLAG		ASC AREA NAV PROC (4.237) ENT AREA NAV PROC (4.220)	LOGICAL	0, 1	NA	1.04 1.04
MELV_CMLK(1)	V95X6057X G50723	MSBLS 1 ELEVATION COMMFAULT/ LOCKON FLAG		ASC AREA NAV PROC (4.237) ENT AREA NAV PROC (4.220)	LOGICAL	0, 1	NA	1.04 1.04

Table 4.44.4-1. MSBLS SOP Interface Table								
Name	MSID/ Mnemonic	Description	Input Source	Output Destination	Signal Type	Range	Units	Rate (Hz)
MELV_CMLK(2)	V95X6063X G50724	MSBLS 2 ELEVATION COMMFAULT/ LOCKON FLAG		ASC AREA NAV PROC (4.237)	LOGICAL	0, 1	NA	1.04
				ENT AREA NAV PROC (4.220)				1.04
MELV_CMLK(3)	V95X6068X G50725	MSBLS 3 ELEVATION COMMFAULT/ LOCKON FLAG		ASC AREA NAV PROC (4.237)	LOGICAL	0, 1	NA	1.04
				ENT AREA NAV PROC (4.220)				1.04
MAZM_CMLK(1)	V95X6058X G50726	MSBLS 1 AZIMUTH COMMFAULT/ LOCKON FLAG		ASC AREA NAV PROC (4.237)	LOGICAL	0, 1	NA	1.04
				ENT AREA NAV PROC (4.220)				1.04
MAZM_CMLK(2)	V95X6064X G50727	MSBLS 2 AZIMUTH COMMFAULT/ LOCKON FLAG		ASC AREA NAV PROC (4.237)	LOGICAL	0, 1	NA	1.04
				ENT AREA NAV PROC (4.220)				1.04
MAZM_CMLK(3)	V95X6069X G50728	MSBLS 3 AZIMUTH COMMFAULT/ LOCKON FLAG		ASC AREA NAV PROC (4.237)	LOGICAL	0, 1	NA	1.04
				ENT AREA NAV PROC (4.220)				1.04
SQ_AZMLS	V95H6021C G19901	SELECTED MSBLS AZIMUTH_SCLD		PRELAND NAV 4.144 RTLS NAV 4.205	FLOAT PT	+/-0.6	RAD	
SQ_ELMLS	V95H6022C G14402	SELECTED MSBLS ELEVATION_SCLD		PRELAND NAV 4.144 RTLS NAV 4.205	FLOAT PT	0, 6	RAD	
SQ_RANGE	V95H6023C G14403	SELECTED MSBLS RANGE_SCLD		PRELAND NAV 4.144 RTLS NAV 4.205	FLOAT PT	0, 125000	FT	

Table 4.44.5-1. MSBLS SOP Conversion Parameters				
Name	Description			Constant
MLSNMI_FT	MSBLS nmi to feet conversion factor			X
V97U6791C	Type	Value	Units	
	Scalar	6076.1155	ft/nmi	

Table 4.44.5-2. MSBLS SOP I-Load Parameters						
Name	MML No.	MSID Description	Constant	Type	Value	Units
MSBAZ_LIM	V99U5719C	MSBLS azimuth limit	X	Scalar	*	Radians
MSBELMAX_LIM	V99U5720C	MSBLS upper elevation limit	X	Scalar	*	Radians
MSBELMIN_LIM	V99U5721C	MSBLS lower elevation limit	X	Scalar	*	Radians
MSBRNG_LIM	V99U5722C	MSBLS range limit	X	Scalar	*	NMI
COURSE_WRT_ RW_LIM	V99U5723C	MSBLS heading limit	X	Scalar	*	Radians

* These values found in JSC 19350.



INITIALIZE RANGE_UPDATE, AZIMUTH_UPDATE, AZIMUTH_PASS, AND
 ELEVATION_UPDATE TO OFF

FIGURE 4.44.6.1-1. MSB_IN Flow Diagram

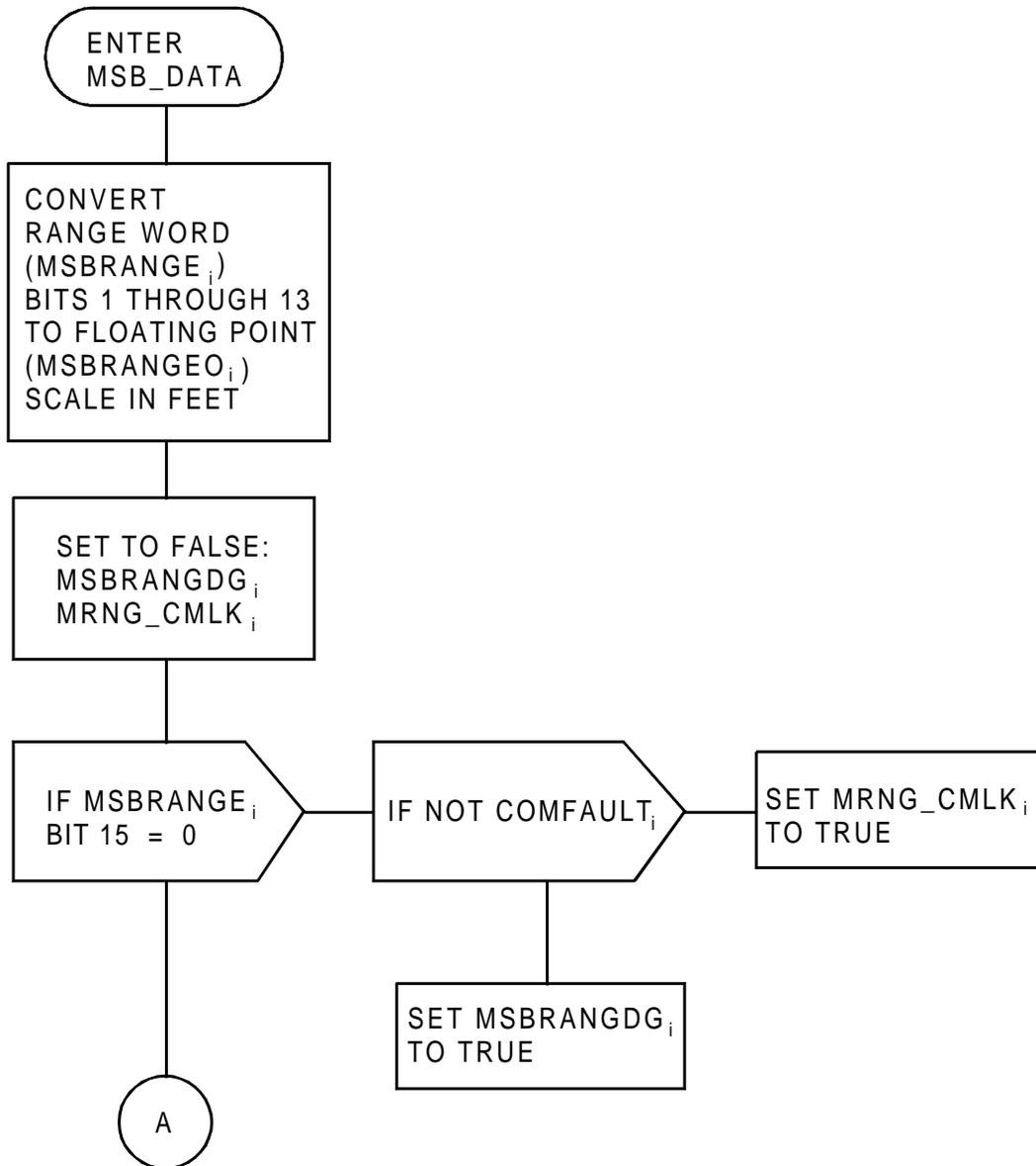


FIGURE 4.44.6.2-1.(1) MSB_DATA Flow Diagram (Sheet 1 of 3)

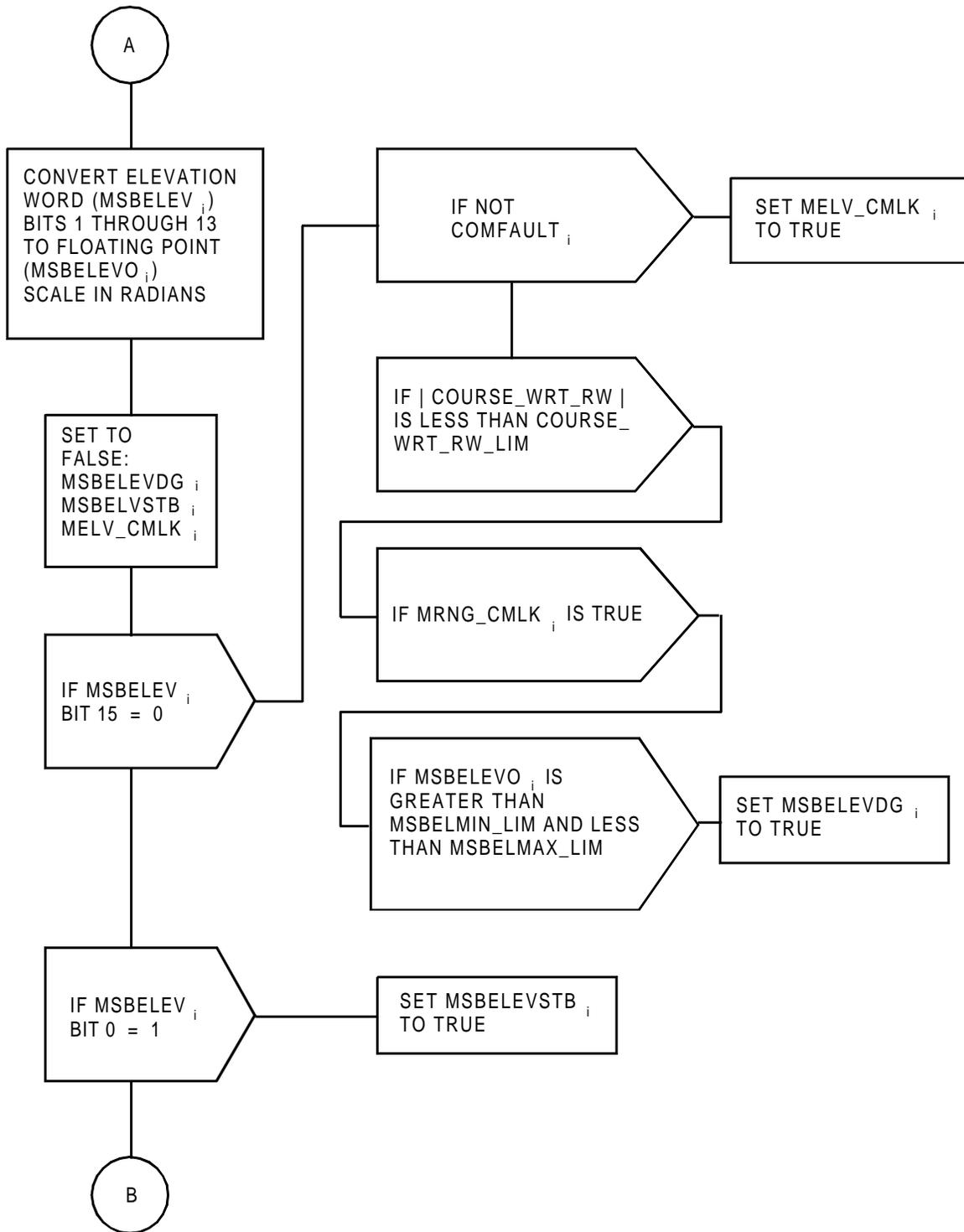


FIGURE 4.44.6.2-1.(2) MSB_DATA Flow Diagram (Sheet 2 of 3)

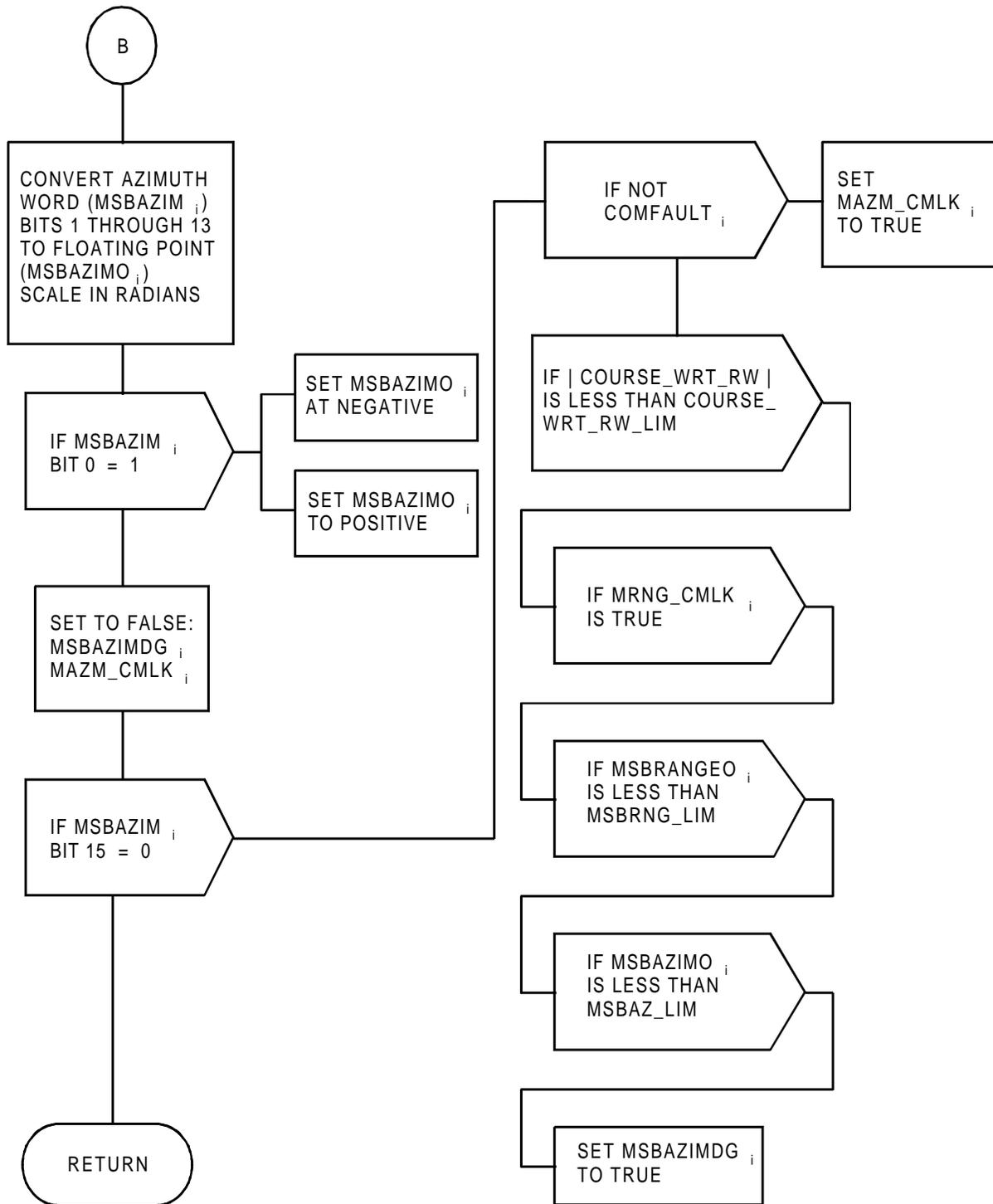


FIGURE 4.44.6.2-1.(3) MSB_DATA Flow Diagram (Sheet 3 of 3)

TABLE 4.44.6.3-1. INPUT FUNCTIONAL PARAMETERS FOR MICROWAVE SCAN BEAM LANDING SYSTEM SOP (G4.44)

FSSR NAME	M/S ID	NOMENCLATURE	SOURCE	UNITS	DATA TYPE	P R E C	LAST CR	R E V
CLOCK/CLOCKTIME	V91W5000C	CLOCK-COMPUTER (GMT)	FCOS	S			93160B	*
COURSE_WRT_RW/PSD	V95H0225CB	COURSE WRT RW CENTERLINE	TAEM UPP	DEG			93017G	*
COURSE_WRT_RW/PSD	V95H0225CC	COURSE WRT RW CENTERLINE	A/L UPP	DEG			93017G	*
COURSE_WRT_RW/PSD	V95H0225CD	COURSE WRT RW CENTERLINE	RTLS UPP	DEG			93017G	*
MSBAZIMIN (1)	V74M1010P	MSBLS NO 1 AZIMUTH WORD	MSBLS 1		HMD		93017G	*
MSBAZIMIN (2)	V74M1110P	MSBLS NO 2 AZIMUTH WORD	MSBLS 2		HMD		93017G	*
MSBAZIMIN (3)	V74M1210P	MSBLS NO 3 AZIMUTH WORD	MSBLS 3		HMD		93017G	*
MSBELEVIN (1)	V74M1030P	MSBLS NO 1 ELEVATION WORD	MSBLS 1		HMD		93017G	*
MSBELEVIN (2)	V74M1130P	MSBLS NO 2 ELEVATION WORD	MSBLS 2		HMD		93017G	*
MSBELEVIN (3)	V74M1230P	MSBLS NO 3 ELEVATION WORD	MSBLS 3		HMD		93017G	*
MSBLS_CF (1)	V91X3550X	MSBLS 1 COMMFAULT	FCOS					
MSBLS_CF (2)	V91X3560X	MSBLS 2 COMMFAULT	FCOS					
MSBLS_CF (3)	V91X3570X	MSBLS 3 COMMFAULT	FCOS					
MSBPOD (1)	V74X0110X	MSBLS NO 1 POWER STATUS	MSBLS 1		BD		93017G	*
MSBPOD (2)	V74X0120X	MSBLS NO 2 POWER STATUS	MSBLS 2		BD		93017G	*
MSBPOD (3)	V74X0130X	MSBLS NO 3 POWER STATUS	MSBLS 3		BD		93017G	*
MSBRANGIN (1)	V74M1050P	MSBLS NO 1 RANGE WORD	MSBLS 1		HMD		93017G	*
MSBRANGIN (2)	V74M1150P	MSBLS NO 2 RANGE WORD	MSBLS 2		HMD		93017G	*
MSBRANGIN (3)	V74M1250P	MSBLS NO 3 RANGE WORD	MSBLS 3		HMD		93017G	*
Q_AZMLS	V90H3002C	SELECTED MSBLS AZIMUTH-RAW	SF	DEG	SPL		93017G	*
Q_ELMLS	V90H3032C	SELECTED MSBLS ELEVATION-RAW	SF	DEG	SPL		93017G	*
Q_RANGE	V90H3062C	SELECTED MSBLS RANGE-RAW	SF	NM	SPL		93017G	*
	V72M6597P	MDM FF01 DSCR IN CD 09 CH 02	HDWR		HMD		93017G	*
	V72M6647P	MDM FF02 DSCR IN CD 09 CH 02	HDWR		HMD		93017G	*
	V72M6697P	MDM FF03 DSCR IN CD 09 CH 02	HDWR		HMD		93017G	*
	V74H1012B	MSBLS NO 1 AZIMUTH DATA	MSBLS 1	DEG	BMS		93017G	*
	V74H1032B	MSBLS NO 1 ELEVATION DATA	MSBLS 1	DEG	BSU		93017G	*
	V74H1052B	MSBLS NO 1 RANGE DATA	MSBLS 1	NM	BSU		93017G	*
	V74H1112B	MSBLS NO 2 AZIMUTH DATA	MSBLS 2	DEG	BMS		93017G	*
	V74H1132B	MSBLS NO 2 ELEVATION DATA	MSBLS 2	DEG	BSU		93017G	*
	V74H1152B	MSBLS NO 2 RANGE DATA	MSBLS 2	NM	BSU		93017G	*
	V74H1212B	MSBLS NO 3 AZIMUTH DATA	MSBLS 3	DEG	BMS		93017G	*
	V74H1232B	MSBLS NO 3 ELEVATION DATA	MSBLS 3	DEG	BSU		93017G	*
	V74H1252B	MSBLS NO 3 RANGE DATA	MSBLS 3	NM	BSU		93017G	*
	V74X1013B	MSBLS NO 1 AZIMUTH TEST	MSBLS 1		BD		93017G	*
	V74X1014B	MSBLS NO 1 AZIMUTH VALIDITY	MSBLS 1		BD		93017G	*
	V74X1031B	MSBLS NO 1 EL AGC-XTAL CUR FAIL	MSBLS 1		BD		93017G	*
	V74X1033B	MSBLS NO 1 ELEVATION TEST	MSBLS 1		BD		93017G	*
	V74X1034B	MSBLS NO 1 ELEVATION VALIDITY	MSBLS 1		BD		93017G	*
	V74X1051B	MSBLS NO 1 RNG AGC-XTAL CUR FAIL	MSBLS 1		BD		93017G	*
	V74X1053B	MSBLS NO 1 RANGE TEST	MSBLS 1		BD		93017G	*
	V74X1054B	MSBLS NO 1 RANGE VALIDITY	MSBLS 1		BD		93017G	*
	V74X1113B	MSBLS NO 2 AZIMUTH TEST	MSBLS 2		BD		93017G	*
	V74X1114B	MSBLS NO 2 AZIMUTH VALIDITY	MSBLS 2		BD		93017G	*
	V74X1131B	MSBLS NO 2 EL AGC-XTAL CUR FAIL	MSBLS 2		BD		93017G	*
	V74X1133B	MSBLS NO 2 ELEVATION TEST	MSBLS 2		BD		93017G	*

TABLE 4.44.6.3-1. INPUT FUNCTIONAL PARAMETERS FOR MICROWAVE SCAN BEAM LANDING SYSTEM SOP (G4.44)

FSSR NAME	M/S ID	NOMENCLATURE	SOURCE	UNITS	DATA TYPE	P R E C	LAST CR	R E V
	V74X1134B	MSBLS NO 2 ELEVATION VALIDITY	MSBLS 2		BD		93017G	*
	V74X1151B	MSBLS NO 2 RNG AGC-XTAL CUR FAIL	MSBLS 2		BD		93017G	*
	V74X1153B	MSBLS NO 2 RANGE TEST	MSBLS 2		BD		93017G	*
	V74X1154B	MSBLS NO 2 RANGE VALIDITY	MSBLS 2		BD		93017G	*
	V74X1213B	MSBLS NO 3 AZIMUTH TEST	MSBLS 3		BD		93017G	*
	V74X1214B	MSBLS NO 3 AZIMUTH VALIDITY	MSBLS 3		BD		93017G	*
	V74X1231B	MSBLS NO 3 EL AGC-XTAL CUR FAIL	MSBLS 3		BD		93017G	*
	V74X1233B	MSBLS NO 3 ELEVATION TEST	MSBLS 3		BD		93017G	*
	V74X1234B	MSBLS NO 3 ELEVATION VALIDITY	MSBLS 3		BD		93017G	*
	V74X1251B	MSBLS NO 3 RNG AGC-XTAL CUR FAIL	MSBLS 3		BD		93017G	*
	V74X1253B	MSBLS NO 3 RANGE TEST	MSBLS 3		BD		93017G	*
	V74X1254B	MSBLS NO 3 RANGE VALIDITY	MSBLS 3		BD		93017G	*

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

TABLE 4.44.6.3-2. OUTPUT FUNCTIONAL PARAMETERS FROM MICROWAVE SCAN BEAM LANDING SYSTEM SOP (G4.44)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
MAZM_CMLK(1)	V95X6058X	MSBLS 1 AZ COMFAULT/LOCKON FLAG	ASC AREA NAV ENT AREA NAV				92297B	
MAZM_CMLK(2)	V95X6064X	MSBLS 2 AZ COMFAULT/LOCKON FLAG	ASC AREA NAV ENT AREA NAV				92297B	
MAZM_CMLK(3)	V95X6069X	MSBLS 3 AZ COMFAULT/LOCKON FLAG	ASC AREA NAV ENT AREA NAV				92297B	
MELV_CMLK(1)	V95X6057X	MSBLS 1 EL COMFAULT/LOCKON FLAG	ASC AREA NAV ENT AREA NAV				92297B	
MELV_CMLK(2)	V95X6063X	MSBLS 2 EL COMFAULT/LOCKON FLAG	ASC AREA NAV ENT AREA NAV				92297B	
MELV_CMLK(3)	V95X6068X	MSBLS 3 EL COMFAULT/LOCKON FLAG	ASC AREA NAV ENT AREA NAV				92297B	
MRNG_CMLK(1)	V95X6056X	MSBLS 1 RANGE COMFAULT/LOCKON FLAG	ASC AREA NAV ENT AREA NAV				92297B	
MRNG_CMLK(2)	V95X6062X	MSBLS 2 RANGE COMFAULT/LOCKON FLAG	ASC AREA NAV ENT AREA NAV				92297B	
MRNG_CMLK(3)	V95X6067X	MSBLS 3 RANGE COMFAULT/LOCKON FLAG	ASC AREA NAV ENT AREA NAV				92297B	
MSBAZIMDG(1)	V95X6124X	MSBLS 1 AZIMUTH LOCKED ON	GAX MSBLS FDIR SF				92297B	
MSBAZIMDG(2)	V95X6224X	MSBLS 2 AZIMUTH LOCKED ON	GAX MSBLS FDIR SF				92297B	
MSBAZIMDG(3)	V95X6324X	MSBLS 3 AZIMUTH LOCKED ON	GAX MSBLS FDIR SF				92297B	
MSBAZIMO(1)	V95H6122C	MSBLS 1 AZIMUTH OUT	ASC AREA NAV ENT AREA NAV MSBLS FDIR SF	RAD			90278D	
MSBAZIMO(2)	V95H6222C	MSBLS 2 AZIMUTH OUT	ASC AREA NAV ENT AREA NAV MSBLS FDIR SF	RAD			90278D	
MSBAZIMO(3)	V95H6322C	MSBLS 3 AZIMUTH OUT	ASC AREA NAV ENT AREA NAV MSBLS FDIR SF	RAD			90278D	
MSBAZIMTIM	V95W6010C	MSBLS AZIMUTH MEASUREMENT TIME	DL FMT 23/1 PRELAND NAV RTL NAV	S	DPL		93017G	*
MSBELEVDG(1)	V95X6144X	MSBLS 1 ELEVATION LOCKED ON	GAX MSBLS FDIR SF				92297B	

TABLE 4.44.6.3-2. OUTPUT FUNCTIONAL PARAMETERS FROM MICROWAVE SCAN BEAM LANDING SYSTEM SOP (G4.44)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
MSBELEVDG(2)	V95X6244X	MSBLS 2 ELEVATION LOCKED ON	GAX MSBLS FDIR				92297B	
MSBELEVDG(3)	V95X6344X	MSBLS 3 ELEVATION LOCKED ON	SF GAX MSBLS FDIR				92297B	
MSBELEVO(1)	V95H6142C	MSBLS 1 ELEVATION OUT	SF ASC AREA NAV ENT AREA NAV MSBLS FDIR	RAD			90278D	
MSBELEVO(2)	V95H6242C	MSBLS 2 ELEVATION OUT	SF ASC AREA NAV ENT AREA NAV MSBLS FDIR	RAD			90278D	
MSBELEVO(3)	V95H6342C	MSBLS 3 ELEVATION OUT	SF ASC AREA NAV ENT AREA NAV MSBLS FDIR	RAD			90278D	
MSBELEVTIM	V95W6020C	MSBLS ELEVATION MEASUREMENT TIME	SF DL FMT 23/1 PRELAND NAV RTL5 NAV	S	DPL		93017G	*
MSBELVSTB(1)	V95X6148X	MSBLS NO 1 ELEVATION SELF TEST BAD	MSBLS FDIR					
MSBELVSTB(2)	V95X6248X	MSBLS NO 2 ELEVATION SELF TEST BAD	MSBLS FDIR					
MSBELVSTB(3)	V95X6348X	MSBLS NO 3 ELEVATION SELF TEST BAD	MSBLS FDIR					
MSBRANGDG(1)	V95X6104X	MSBLS 1 RANGE LOCKED ON	GAX MSBLS FDIR				92297B	
MSBRANGDG(2)	V95X6204X	MSBLS 2 RANGE LOCKED ON	SF GAX MSBLS FDIR				92297B	
MSBRANGDG(3)	V95X6304X	MSBLS 3 RANGE LOCKED ON	SF GAX MSBLS FDIR				92297B	
MSBRANGE0(1)	V95H6102C	MSBLS 1 RANGE OUT	SF ASC AREA NAV ENT AREA NAV MSBLS FDIR	FT			90278D	
MSBRANGE0(2)	V95H6202C	MSBLS 2 RANGE OUT	SF ASC AREA NAV ENT AREA NAV MSBLS FDIR	FT			90278D	
MSBRANGE0(3)	V95H6302C	MSBLS 3 RANGE OUT	SF ASC AREA NAV ENT AREA NAV MSBLS FDIR	FT			90278D	

TABLE 4.44.6.3-2. OUTPUT FUNCTIONAL PARAMETERS FROM MICROWAVE SCAN BEAM LANDING SYSTEM SOP (G4.44)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
MSBRANGTIM	V95W6000C	MSBLS RANGE MEASUREMENT TIME	DL FMT 23/1 PRELAND NAV RTLS NAV	S	DPL		93017G	*
SQ_AZMLS	V95H6021C	SELECTED MSBLS AZIMUTH-SCLD	PRELAND NAV RTLS NAV	RAD			89137C	
SQ_ELMLS	V95H6022C	SELECTED MSBLS ELEVATION-SCLD	PRELAND NAV RTLS NAV	RAD			89137C	
SQ_RANGE	V95H6023C	SELECTED MSBLS RANGE-SCLD	PRELAND NAV RTLS NAV	FT			89137C	

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

**Rate Codes: (HDR Only) 1 = 1 sample/sec 2 = 5 samples/sec 3 = 12.5 samples/sec 4 = 25 samples/sec 5 = 100 samples/sec
(HDR and LDR) A = 1 sample/sec B = 5 samples/sec C = 12.5 samples/sec D = 25 samples/sec E = 100 samples/sec**

TABLE 4.44.6.3-3. MICROWAVE SCAN BEAM LANDING SYSTEM SOP (G4.44) I-LOADS

FSSR NAME	MSID	ENG UNITS	DT	PR	D	S	PR FCTN	MC	CAT	LAST CR	R E V
COURSE_WRT_RW_LIM	V99U5723C	RAD	F	S	D	P	G4.44	00	MSI	DC0755	
MSBAZ_LIM	V99U5719C	RAD	F	S	D	P	G4.44	00	MSI	DC0755	
MSBELMAX_LIM	V99U5720C	RAD	F	S	D	P	G4.44	00	MSI	DC0755	
MSBELMIN_LIM	V99U5721C	RAD	F	S	D	P	G4.44	00	MSI	DC0755	
MSBRNG_LIM	V99U5722C	NMI	F	S	D	P	G4.44	00	MSI	DC0755	

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

TABLE 4.44.6.3-4. MICROWAVE SCAN BEAM LANDING SYSTEM SOP (G4.44) K-LOADS

FSSR NAME DESCRIPTION	MSID	MC	KLOAD VALUE	ENG UNITS	DT	P R	S	PR FCTN	LAST CR	R E V
--------------------------	------	----	-------------	-----------	----	--------	---	---------	---------	-------------

NO REQUIREMENTS

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

TABLE 4.44.6.3-5. MICROWAVE SCAN BEAM LANDING SYSTEM SOP (G4.44) CONSTANTS

FSSR NAME DESCRIPTION	MSID	MC	CONSTANT VALUE	ENG UNITS	DT	P R	S	PR FCTN	LAST CR	R E V
FT_PER_NMI NAUTICAL MILE TO FEET CONVERSION	V97U6791C	00	+6.0761155	E+03 FT/NMI	F		S P	A6.9 G4.213 G4.44	90374C	

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

This page intentionally left blank.

4.45 RADAR ALTIMETER SUBSYSTEM OPERATING PROGRAM.

This section defines the detailed functional requirements and formulations for the radar altimeter (RA) subsystem operating program (SOP). The RA SOP defines software associated with radar altimeter altitude (height above terrain) data processing.

4.45.1 Radar Altimeter Requirements Overview.

4.45.1.1 Radar Altimeter Functional Overview.

The RA is a C-band altimeter system that provides an indication of the orbiter altitude from 5000 feet to touchdown. The RA requires no ground station information as is the case with the other navigation sensors. The RA system provides altitude from the antenna to the nearest point on the terrain. The orbiter design contains two identical independent RA subsystems for redundancy purposes. A single RA subsystem is depicted in Figure 4.45.1.1-1 along with related interfacing equipment. Six major components are identified that are required for the RA function: a control head, an MDM (multiplexer/demultiplexer), a receiver-transmitter (R/T), a transmitting antenna, a receiving antenna, and the onboard computer. The keyboard and the display electronics unit are also required for the crew to initiate self-test command of the RA R/T.

The RA control head (panel switch) has a two-position function switch. The two positions of the function switch are ON and OFF. When the switch is in the ON position, power is applied to the R/T and a power-present discrete returns from the R/T to the MDM. Software control of the RA is limited to processing the self-test command. The RA self-test requirements are documented in the VU requirements.

4.45.1.1.1 Radar Altimeter Operating Modes.

The radar altimeter operating modes are as follows:

- OFF mode
- ON mode
- SELF-TEST mode

The OFF mode is indicated when the power-on discrete from the RA is in the false or off state.

The ON mode occurs when the power-on discrete from the RA is in the true or on state.

The SELF-TEST mode occurs when the power-on discrete is on and self-test is commanded from the keyboard. In this mode the transceiver tests the internal circuits and provides output data that indicates the health of the transceiver.

4.45.1.1.2 Radar Altimeter Digital Interfaces.

The radar altimeter interfaces with the GPC via the serial digital multiplex/demultiplex input and output data channels. The digital I/O words are 16 bits in size, plus 3 bits for synchronization and 1 bit for odd parity. Throughout the RA SOP, the synchronization and parity bits are disregarded and the most significant data bit in the word is designated as bit 0 with the least significant bit being bit 15. The contents, format, and bit values of the RA data words are illustrated in Figure 4.45.1.1.2-1 and Figure 4.45.1.1.2-2.

4.45.1.1.3 Radar Altimeter Discrete Interfaces.

The radar altimeter requires the following discrete interfaces: power-on discrete and self-test command.

The power-on discrete is generated by the RA and is true when power is applied and the RA is operating. The self-test command is generated by VU software. The self-test command causes the RA to perform self-test.

4.45.1.2 RA SOP Overview.

The RA SOP controls and monitors the operation of the two RA systems and processes RA data for other program elements. The inputs and outputs from the RA SOP modules are stored in locations labeled as to parameter and transceiver.

Each RA transceiver is serviced in sequence, and the altitude data and data good signals are made available to the selection filter and the AVVI processor. The time tag for radar altimeter data is made available to user parameter processing.

Refer to Figure 4.45.1.2-1 for a block diagram depicting the signal flow.

RA data processing is performed on each transceiver at a rate of 6.25 Hz. The RA_DATA functional module accepts RA outputs and removes the flag bits for use by other program elements and converts altitude from fixed-point to floating-point data.

4.45.1.3 Radar Altimeter SOP Functional Module Summary.

The functional modules that constitute the RA SOP are identified in Table 4.45.1.3-1. Each of the modules, including functions and formulations, are presented in detail in Section 4.45.6.

4.45.1.4 Reference Documents.

MC409-0015 Altimeter, Radar; Rockwell.

4.45.1.5 Software Protection.

The software implementation of the RA SOP shall protect against program checks and error interrupts caused by the transient-type errors or bad data inputs. Specifically, the RA SOP shall protect against all of the error conditions identified in paragraph 3.6 that are applicable to the RA software design and code.

4.45.2 Definitions.

This section is not applicable to the RA SOP.

4.45.3 Functional Requirements Traceability.

The detailed requirements and formulations contained herein satisfy the GN&C, Level B, CPDS functional requirements for the RA SOP.

4.45.4 Interface Parameters.

The interface parameters required to properly execute the RA SOP functional module are shown in Table 4.45.4-1.

4.45.5 Radar Altimeter Initialization.

No I-load parameters are required for RA data processing.

4.45.6 Detailed Requirements and Formulations.

This section defines the RA SOP functional module in sufficient detail to create a software design and to generate computer code. The module defines the functional requirements, interface requirements, detailed requirements and formulations, and special considerations if applicable. A flow diagram is also included; however, the flow diagram is supplementary material and is not intended to convey requirements or to place restrictions on software design or coding.

4.45.6.1 RA Data Processing (RA_DATA).

The purpose of the RA_DATA functional module is to process the RA input data from a powered RA.

A. Functional Requirements. The RA_DATA module is required to perform the following functions for each RA.

1. Accept the data words input from the RA and strip off the flag bits. Test data word input for odd parity and set RAPODD flag.
2. Generate the data good and comlock flags.
3. Convert altitude data from fixed point to floating point.

B. Interface Requirements. The input and output parameters associated with the RA_DATA functional module are contained in Table 4.45.4-1.

C. Functional Requirements and Formulations. The following operations shall be performed for each RA transceiver at a rate of 6.25 Hz.

1. RA Altitude Out. RA altitude out (RA_ALTO) is contained in the RA parent word, RA_ALTIN, as a 13-bit binary word (Figure 4.45.1.1.2-2). The most significant bit is bit 1. The least significant bit is bit 13 with a value of 1 foot. Extract bits 1 through 13 of the RA input word (RA_ALTIN_i) and form RA_ALTO_i in floating point in units of feet.

The RA parent word will be tested for odd parity. The 16 binary stages of the RA parent word will be summed and will be said to be true if the sum is an odd number. RAPODD will be set true if the sum is odd.

2. RA Altitude Lock-On and Comlock. The RA altitude lock-on, RALT_DG, is driven by bit 14 of the RA parent word (RA_ALTIN). RALT_DG is a two-state flag indicating data good or bad. True indicates good data. If RAPODD is true and if RA_ALTIN_i bit 14 = true, then set RALT_DG_i to true; else set RALT_DG_i to false.

If RA_i altitude is locked on and if there is no commfault, set RALT_CMLK_i flag to true; else set RALT_CMLK_i flag false.

Note that the RALT_CMLK_i flag status is only required to be computed for the AVVI processor at a rate of 1.04 Hz

3. RA Time Tag. Set the RA time tag parameter, RA_TIME, equal to the computer clock time, CLOCK, corresponding to the current RA input read transaction time.

Figure 4.45.6.1-1 presents a flow diagram for the RA_DATA functional module.

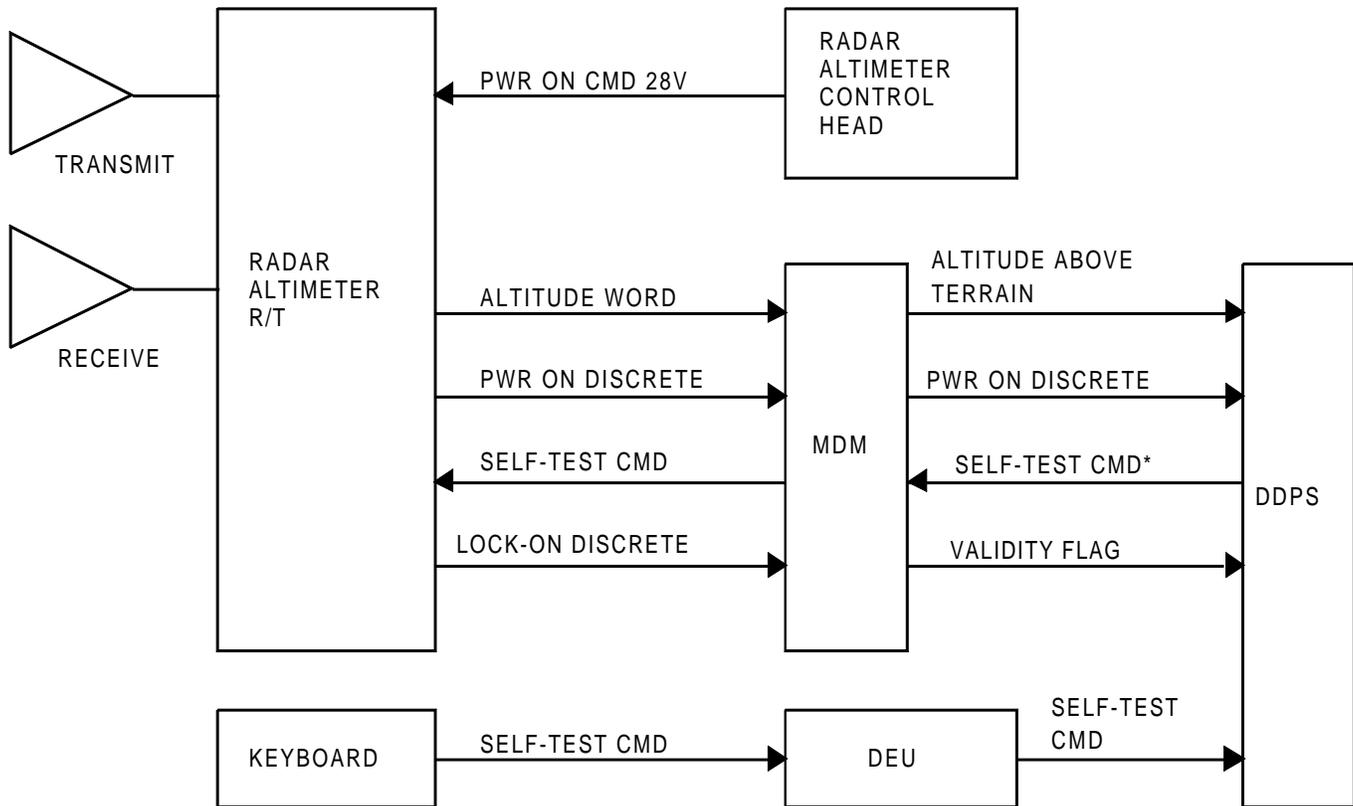
4.45.6.2 Applicable Tables for RA SOP.

The IDD inputs and outputs are listed in Tables 4.45.6.2-1. and 4.45.6.2-2 Values for the I-loads are contained in the I-loads requirements document (JSC-19350); however, I-load definitions applicable to this principal function are listed in Table 4.45.6.2-3. K-loads are listed in Table 4.45.6.2-4. Constants are listed in Table 4.45.6.2-5.

Table 4.45.1.3-1. RA SOP Functional Modules		
Module Name	Applicable Paragraph	Module Description
RA_DATA	4.45.6.1	Radar Altimeter Data Processing

Table 4.45.4-1. Radar Altimeter Interface Table

Name	MSID	Description	Input Source	Output Destination	Signal Type	Range	Units	Rate (Hz)
RA_ALTIN(1)	V74M1801P G08500	RADAR ALT NO. 1 PARENT WORD	RA1		FIXED PT	0, 8200	FEET	6.25
RA_ALTIN(2)	V74M1851P G08501	RADAR ALT NO. 2 PARENT WORD	RA2		FIXED PT	0, 8200	FEET	6.25
CLOCK	V91W5000C G07829	CLOCK-COMPUTER (GMT)	FCOS		FLOAT PT	NA	D:H:M:S	6.25
RA_ALTO(1)	V95H7100C G08405	RADAR ALTIMETER NO. 1 ALTITUDE		SF 4.71 AVVI PROC 4.106 TLM	FLOAT PT	0, 8200	FEET	6.25 6.25
RA_ALTO(2)	V95H7200C G08406	RADAR ALTIMETER NO. 2 ALTITUDE		SF 4.71 AVVI PROC 4.106 TLM	FLOAT PT	0, 8200	FEET	6.25 6.25
RALT_DG(1)	V95X7110X G08403	RADAR ALTIMETER NO. 1 LOCKED ON		SF 4.71 TLM	LOGICAL	0, 1	NA	6.25
RALT_DG(2)	V95X7210X G08404	RADAR ALTIMETER NO. 2 LOCKED ON		SF 4.71 TLM	LOGICAL	0, 1	NA	6.25
RALT_CF(1)	V91X3600X G16502	RADAR ALTIMETER 1 COMMFAULT	FCOS		LOGICAL	0, 1	NA	6.25
RALT_CF(2)	V91X3620X G16503	RADAR ALTIMETER 2 COMMFAULT	FCOS		LOGICAL	0, 1	NA	6.25
RA_TIME	V95W7000C G08407	RADAR ALTIMETER TIME		TAEM UPP 4.24 A/L UPP 4.25 RTLS UPP TLM	FLOAT PT	NA	SEC	6.25 6.25
RALT_CMLK(1)	V95X7130X	RA 1 COMMFAULT/ LOCKON FLAG		AVVI PROC 4.106	LOGICAL	0, 1	NA	1.04
RALT_CMLK(2)	V95X7230X	RA 2 COMMFAULT/ LOCKON FLAG		AVVI PROC 4.106	LOGICAL	0, 1	NA	1.04
RA_PWR_ON(1)	V74S0175E	RADAR ALTIMETER NO. 1 READY	RA1	—	DISCRETE	0, 1	NA	6.25
RA_PWR_ON(2)	V74S0185E	RADAR ALTIMETER NO. 2 READY	RA2	—	DISCRETE	0, 1	NA	6.25



*SOFTWARE REQUIREMENTS ARE SPECIFIED IN THE VU CPDS.

FIGURE 4.45.1.1-1. Radar Altimeter Subsystem Diagram

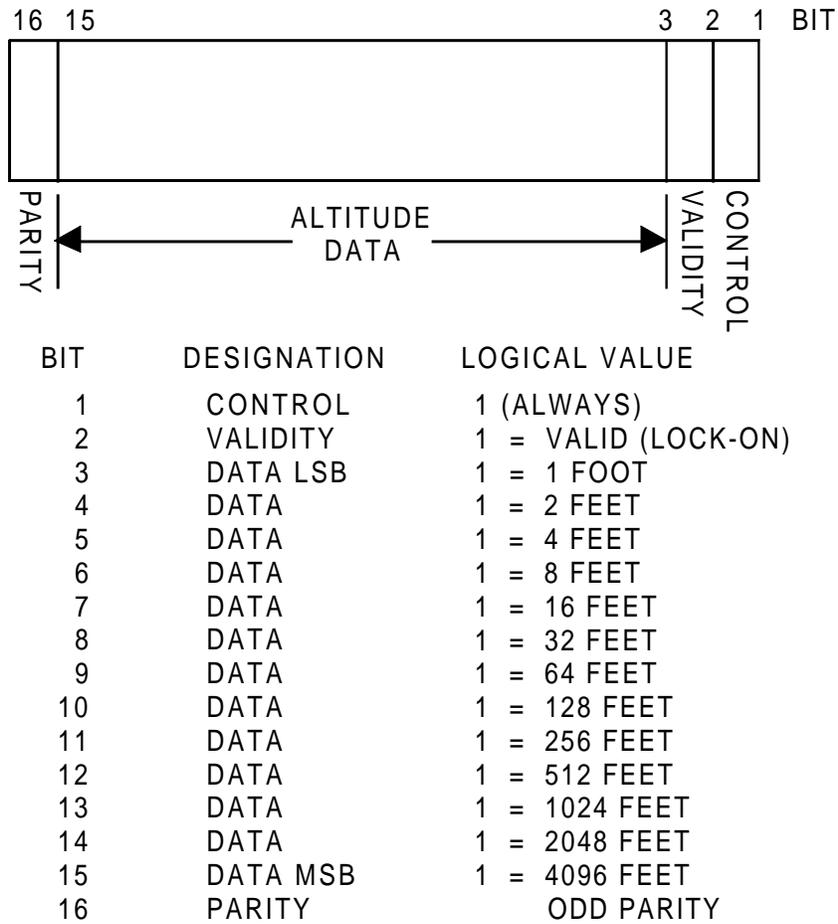
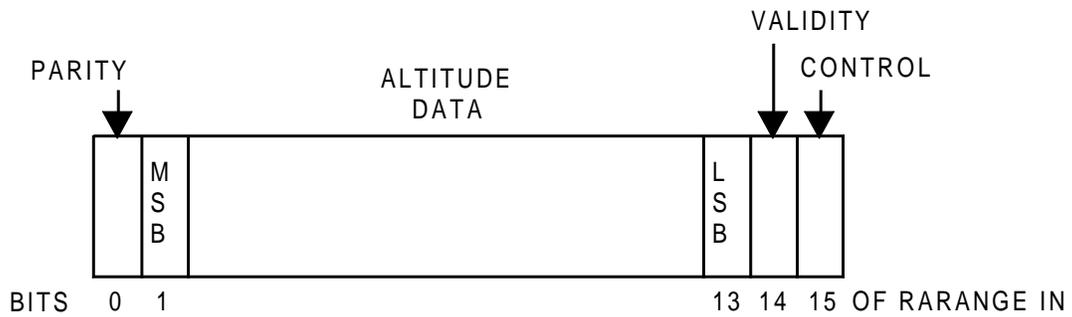


FIGURE 4.45.1.1.2-1. RA Altitude Data Word (From RA to MDM)



BIT	DESIGNATION	VALUE
0	PARITY	ODD PARITY
1	DATA MSB	1 = 4096 FT
2	DATA	1 = 2048 FT
3	DATA	1 = 1024 FT
4	DATA	1 = 512 FT
5	DATA	1 = 256 FT
6	DATA	1 = 128 FT
7	DATA	1 = 64 FT
8	DATA	1 = 32 FT
9	DATA	1 = 16 FT
10	DATA	1 = 8 FT
11	DATA	1 = 4 FT
12	DATA	1 = 2 FT
13	DATA LSB	1 = 1 FT
14	VALIDITY	1 = VALID (LOCK ON)
15	CONTROL	1 (ALWAYS)

FIGURE 4.45.1.1.2-2. RA Altitude Data Word (From MDM to GPC)

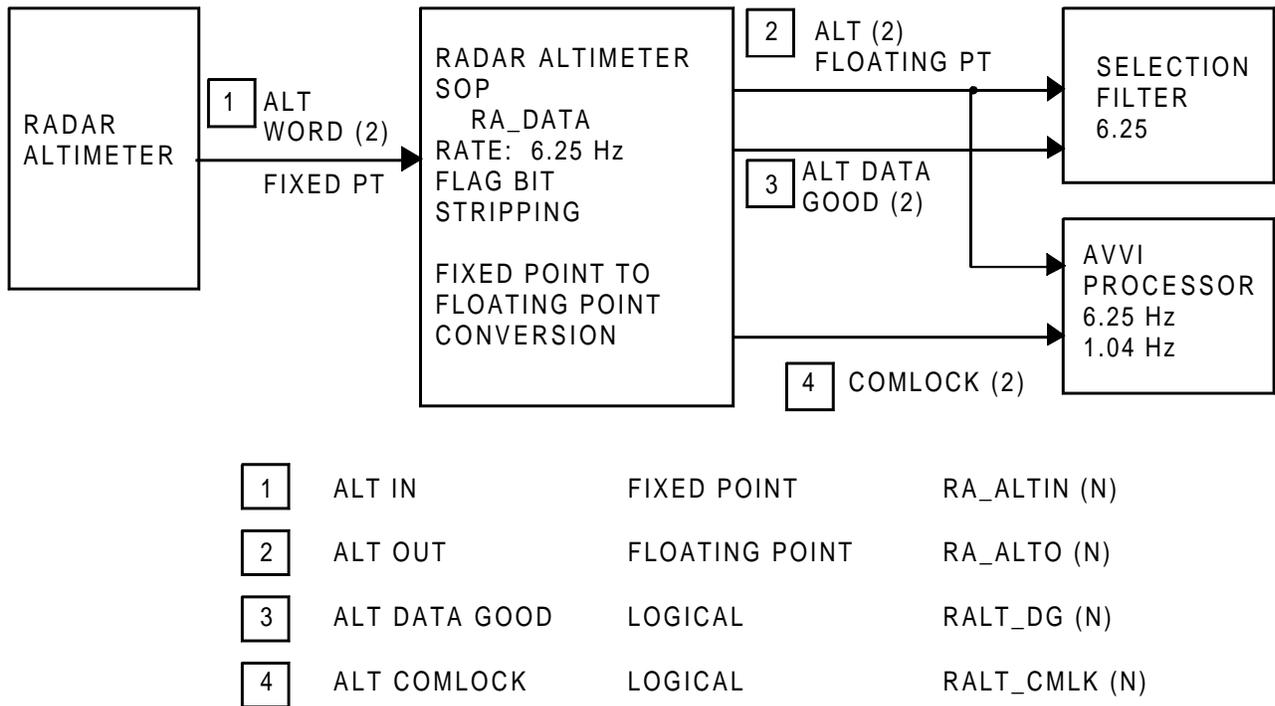


FIGURE 4.45.1.2-1. Radar Altimeter Data Signal Flow

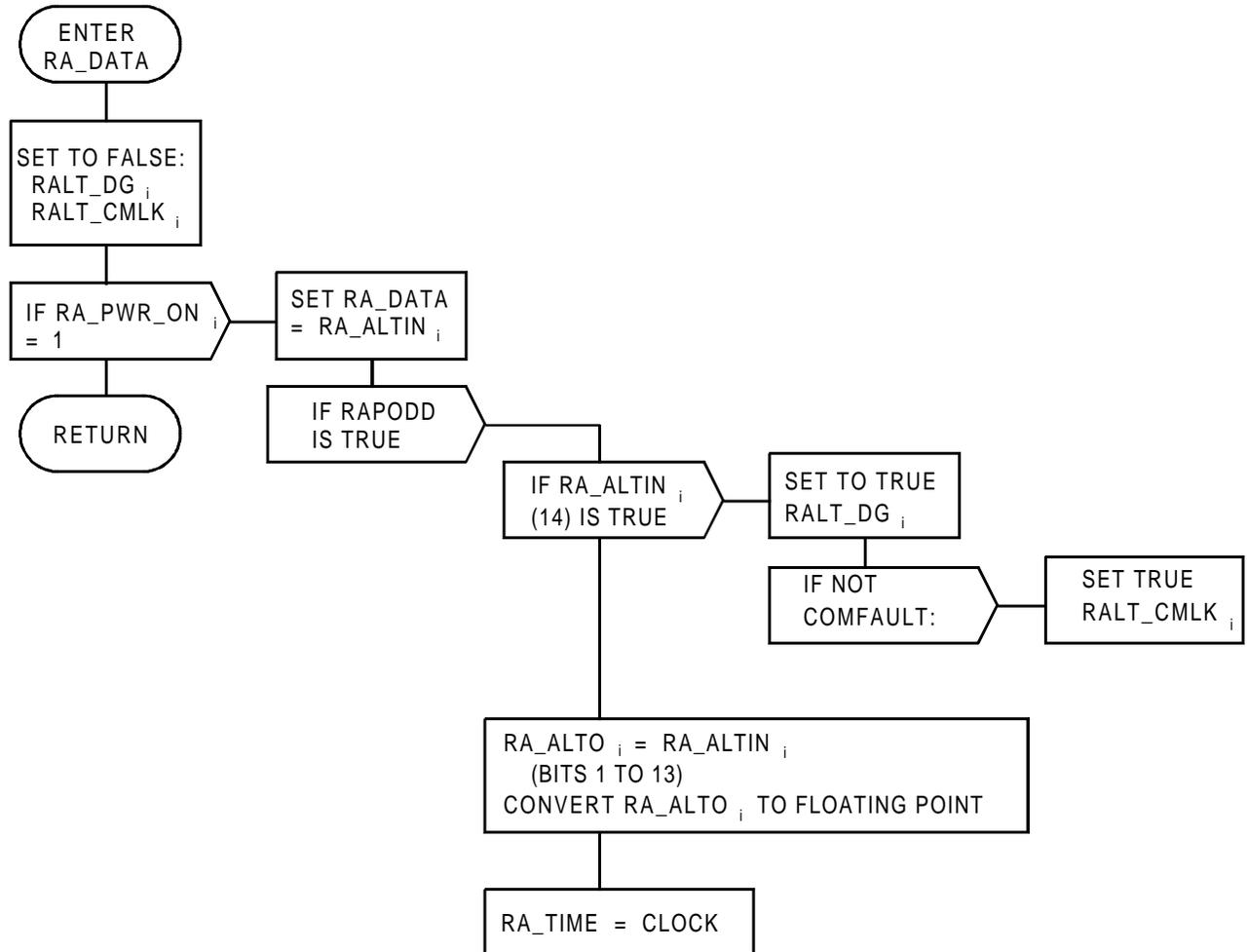


FIGURE 4.45.6.1-1. RA_DATA Flow Diagram

TABLE 4.45.6.2-1. INPUT FUNCTIONAL PARAMETERS FOR RADAR ALTIMETER SUBSYSTEM OPERATING PROGRAM (G4.45)

FSSR NAME	M/S ID	NOMENCLATURE	SOURCE	UNITS	DATA TYPE	P R E C	LAST CR	R E V
CLOCK/CLOCKTIME	V91W5000C	CLOCK-COMPUTER(GMT)	FCOS	S			93160B	*
RALT_CF(1)	V91X3600X	RADAR ALTIMETER 1 COMMFAULT	FCOS					
RALT_CF(2)	V91X3620X	RADAR ALTIMETER 2 COMMFAULT	FCOS					
RA_ALTIN(1)	V74M1801P	RADAR ALT NO 1 PARENT WORD	RA 1	FT	HMD		93017G	*
RA_ALTIN(2)	V74M1851P	RADAR ALT NO 2 PARENT WORD	RA 2	FT	HMD		93017G	*

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

TABLE 4.45.6.2-2. OUTPUT FUNCTIONAL PARAMETERS FROM RADAR ALTIMETER SUBSYSTEM OPERATING PROGRAM
(G4.45)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
RALT_CMLK(1)	V95X7130X	RA 1 COMMFault/LOCKON FLAG	AVVI PROC					
RALT_CMLK(2)	V95X7230X	RA 2 COMMFault/LOCKON FLAG	AVVI PROC					
RALT_DG(1)	V95X7110X	RADAR ALTIMETER NO 1 LOCKED ON	SF TLM					
RALT_DG(2)	V95X7210X	RADAR ALTIMETER NO 2 LOCKED ON	SF TLM					
RA_ALTO(1)	V95H7100C	RADAR ALTIMETER NO 1 ALTITUDE	AVVI PROC SF TLM	FT				
RA_ALTO(2)	V95H7200C	RADAR ALTIMETER NO 2 ALTITUDE	AVVI PROC SF TLM	FT				
RA_TIME	V95W7000C	RADAR ALTIMETER TIME	A/L UPP DL FMT 23/2 GPS SOP RTLS UPP TAEM UPP	S	DPL		93017G	*

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

**Rate Codes: (HDR Only) 1 = 1 sample/sec 2 = 5 samples/sec 3 = 12.5 samples/sec 4 = 25 samples/sec 5 = 100 samples/sec
(HDR and LDR) A = 1 sample/sec B = 5 samples/sec C = 12.5 samples/sec D = 25 samples/sec E = 100 samples/sec**

TABLE 4.45.6.2-3. RADAR ALTIMETER SUBSYSTEM OPERATING PROGRAM (G4.45) I-LOADS

FSSR NAME	MSID	ENG UNITS	DT	PR	D	S	PR FCTN	MC	CAT	LAST CR	R E V
-----------	------	-----------	----	----	---	---	---------	----	-----	---------	-------------

NO REQUIREMENTS

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

TABLE 4.45.6.2-4. RADAR ALTIMETER SUBSYSTEM OPERATING PROGRAM (G4.45) K-LOADS

FSSR NAME DESCRIPTION	MSID	MC	KLOAD VALUE	ENG UNITS	DT	P R	S	PR FCTN	LAST CR	R E V
--------------------------	------	----	-------------	-----------	----	--------	---	---------	---------	-------------

NO REQUIREMENTS

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

TABLE 4.45.6.2-5. RADAR ALTIMETER SUBSYSTEM OPERATING PROGRAM (G4.45) CONSTANTS

FSSR NAME DESCRIPTION	MSID	MC	CONSTANT VALUE	ENG UNITS	DT	P R	S	PR FCTN	LAST CR	R E V
--------------------------	------	----	----------------	-----------	----	--------	---	---------	---------	-------------

NO REQUIREMENTS

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

This page intentionally left blank.

4.122 NAVIGATION-DERIVED AIR DATA SOP.

This section documents the detailed level requirements for the navigation–derived air data (NAVDAD) SOP principal function.

4.122.1 Overview.

The purpose of the NAVDAD SOP is to calculate certain air data parameters based on navigation–supplied data. The navigation–derived air data parameters provided by the NAVDAD SOP are as follows:

- Dynamic pressure
- Equivalent airspeed
- Mach number

These parameters are calculated at the 1.04 Hz rate prior to the NAVDAD/ADS transition processing specified in the ADTA SOP.

These parameters are sent to the ADTA SOP to provide a central location for transitioning between NAVDAD and ADS air data parameters. It should be noted that the navigation–derived angle of attack (from attitude processing) and ground relative velocity magnitude (from the user parameter processing functions) are sent directly to the ADTA SOP for transitioning. The remaining navigation–derived air data parameters (altitude, altitude rate, drag acceleration, and sideslip angle) are calculated by navigation, attitude processing, or UPP.

The navigation–derived air data are needed during entry to drive guidance and flight control and the dedicated displays. After the crew has manually transferred to ADS air data, the navigation–derived air data are used as a backup source.

4.122.1.1 Functional Requirements Traceability.

The detailed requirements and formulations contained herein satisfy the GN&C, Level B, CPDS functional requirements for the NAVDAD SOP.

4.122.2 NAVDAD SOP Detail Level Requirements.

The NAVDAD SOP shall compute the following air data parameters based on nav–supplied data:

- Dynamic pressure
- Equivalent airspeed
- Mach number

These parameters shall be computed at the 1.04 Hz rate whenever the NAVDAD/ AUTO SPEC function indicates NAVDAD.

These parameters shall be computed prior to the NAVDAD/ADS transition function specified by the ADTA SOP. In addition, execution of this function shall occur in the computational cycle corresponding to the beginning of the AMI and AVVI principal functions' first-order hold periods. This is required to minimize the staleness of the AMI/AVVI data when NAVDAD data is being utilized and either air data source select switch is in the COMPUTER position.

The detailed requirements required to compute the above parameters are discussed in the following paragraphs. A functional block diagram of the NAVDAD SOP is shown in Figure 4.122-1. The functional modules are shown in Figures 4.122-2 through 4.122-5. A restriction of the execution sequence within the NAVDAD SOP is that the navigation-derived dynamic pressure shall be computed before the navigation-derived equivalent airspeed. In addition, at ground relative speeds less than 2000 feet per second, static air temperature shall be computed before the dynamic pressure subfunction.

The input/output parameters are shown in Tables 4.122-1 and 4.122-2, respectively. The initialization requirements are shown in Table 4.122-3 and the I-load parameters are shown in Table 4.122-4.

4.122.2.1 Navigation Derived Dynamic Pressure (DYPRS_N).

The purpose of the DYPRS_N subfunction shall be to compute dynamic pressure as a function of navigation-data. The vehicle's ground relative speed shall provide the basis for the dynamic pressure equations given below.

- A. For $REL_VEL_MAG \geq V_NAVDAD$ (NAVDAD velocity switching point), the subfunction shall first compute the drag coefficient as a function of navigation derived angle of attack and ground relative speed as follows:

$$C_D = KCD1 + KCD2 \exp - \left(\frac{REL_VEL_MAG - KCD3}{KCD4} \right) + KCD5 ALPHA_NAVDAD + KCD6 ALPHA_NAVDAD^2$$

where

C_D = Drag coefficient (nd)

REL_VEL_MAG = Ground relative velocity magnitude (ft/sec)

$ALPHA_NAVDAD$ = navigation derived angle of attack (deg)

KCD1 through KCD6 = Constants

Next, the dynamic pressure shall be computed as

$$QBAR_N = \frac{ACC_DRAG M}{C_D REF_AREA}$$

where

QBAR_N = navigation derived dynamic pressure (lb/sq ft)

ACC_DRAG = Drag acceleration (ft/sec²) from the appropriate user parameter processing (UPP) function

M = Current vehicle mass (slugs)

REF_AREA = Reference area (sq ft) for the vehicle associated with the model for C_D mentioned earlier

For OPS-3, the current vehicle mass is obtained from the DEORB GUID principal function (4.209), while for OPS 6 it is obtained from a premission value defined by GRTLS TAEM guidance.

B. For REL_VEL_MAG < V_NAVDAD: First, the air density shall be computed depending on the lapse rate for the Ith layer as follows:

1. For LM(I) = 0,

$$RHO_N = RHO_B(I) \exp - \left(G_OVR_R \frac{ALT - ALT_B(I)}{TMB(I)} \right)$$

where

LM(I) = Lapse rate in Ith layer (°K/ft)

RHO_N = Navigation-derived air density (slugs/ft³)

RHO_B(I) = Air density at the base of the Ith layer (slugs/ft³)

G_OVR_R = Sea level gravity divided by the universal gas constant of air (°K/ft)

ALT = Altitude above mean sea level (ft)

ALT_B(I) = Altitude at the base of the Ith layer (ft)

TMB(I) = Temperature at the base of the Ith layer (°K)

2. For LM(I) ≠ 0,

$$RHO_N = RHO_B(I) \left(\frac{TMB(I)}{SAT_N} \right)^{\left[\frac{G_OVR_R}{LM(I)} + 1 \right]}$$

where SAT_N is the navigation-derived static air temperature (°K) and other parameters are as defined earlier. The values of I and SAT_N are to be determined by the MACH_N subfunction of paragraph 4.122.2.3, and are based on a two-layered atmospheric model.

Next, QBAR_N shall be computed as follows:

$$QBAR_N = 0.5 RHO_N REL_VEL_MAG^2$$

4.122.2.2 Navigation-Derived Equivalent Airspeed (EQAS_N).

The purpose of the EQAS_N subfunction shall be to compute equivalent airspeed as a function of navigation-derived dynamic pressure. This subfunction shall not be executed until the DYPRS_N subfunction has been executed. The navigation derived equivalent airspeed shall be computed as follows:

$$EAS_N = \sqrt{\frac{2 QBAR_N}{RHO_SEA_LEVEL}}$$

where

EAS_N = Navigation-derived equivalent airspeed (ft/sec)

QBAR_N = Navigation-derived dynamic pressure (lb/ft²)

RHO_SEA_LEVEL = Sea level atmospheric density (slugs/ft³) (identical to RHO_B(1))

4.122.2.3 Navigation-Derived Mach Number (MACH_N).

The purpose of the MACH_N subfunction shall be to compute Mach number as a function of navigation-supplied altitude and velocity data. The ground relative speed shall provide the basis for the Mach equations as follows:

- A. For $REL_VEL_MAG \geq V_NAVDAD$

$$M_N = \frac{REL_VEL_MAG}{1000 \text{ ft / sec}}$$

where

M_N = Navigation-derived Mach number

REL_NAV_MAG = Ground relative velocity magnitude (ft/sec)

- B. For $REL_VEL_MAG < V_NAVDAD$, Mach shall be computed as a function of static air temperature and ground relative velocity. Static air temperature shall be determined by modeling the atmosphere into two layers. Within each layer a constant lapse rate with respect to altitude is assumed. This function shall first determine the layer, I, by comparing the vehicle altitude above mean sea level to the layer altitude boundaries. Then the constants describing that layer shall be used to compute static air temperature as follows:

$$SAT_N = TMB(I) + LM(I) (ALT - ALT_B(I)),$$

where

SAT_N = Navigation-derived static air temperature (°K)

TMB(I) = Temperature at base of layer (°K)

LM(I) = Lapse rate in layer (°K/ft)

ALT_B(I) = Altitude at the base of layer (ft)

ALT = Vehicle altitude above mean sea level (ft)

I = Layer identifier

The values of I and SAT_N are also provided to the dynamic pressure subfunction.

The Mach number shall then be computed as follows:

$$M_N = \frac{REL_VEL_MAG}{SPD_SND_CON \sqrt{SAT_N}}$$

where

M_N = Navigation-derived Mach number

REL_VEL_MAG = Ground relative velocity magnitude (ft/sec)

SAT_N = Navigation-derived static air temperature (°K)

SPD_SND_CON = Speed of sound constant for air (ft/sec/°K^{1/2})

4.122.2.4 Applicable Tables for NAVDAD SOP.

The IDD inputs and outputs are listed in Tables 4.122.2.4-1 and 4.122.2.4-2. Values for the I-loads are contained in the I-loads requirements document (JSC-19350); however, I-load definitions applicable to this principal function are listed in Table 4.122.2.4-3. K-loads are listed in Table 4.122.2.4-4. Constants are listed in Table 4.122.2.4-5.

Table 4.122-1. NAVDAD SOP Inputs							
Name	MSID	Description	Input Source	Signal Type	Range	Units	Rate (Hz)
NAV_DAD	G70317	NAVDAD/AUTO SELECT (0=NAVDAD)	HORZ SIT	DISCR ETE	0,1		1.04
REL_VEL_MAG	V95L0151C G03902	GND REL MAGNITUDE IN M50 SYS	ENTRY UPP TAEM UPP A/L UPP		0, 27K	FT/SEC	
ALT	V95H0175C G04207	CURR ORB ALTITUDE ABOVE REF ELLIPSOID	ENTRY UPP TAEM UPP A/L UPP, RTLS UPP	FLOAT PT	-10, +10,000	KFT	1.04
ALPHA_NAVDAD	V95H0233C	NAV DERIVED ANGLE OF ATTACK	RTLS, ENTRY, TAEM, A/L UPP	FLOAT PT		DEG	1.04
ACC_DRAG	V95A0160C G04205	DRAG DECEL	ENTRY UPP TAEM UPP RTLS UPP	FLOAT PT		FT/SEC /SEC	1.04
M*	V90U1961CC G48797	CURRENT VEHICLE MASS	DEORBIT GUID	FLOAT PT		SLUGS	

*Current vehicle mass obtained from DEORBGUID during OPS 3 only

Table 4.122-2. NAVDAD SOP Outputs							
Name	MSID	Description	Output Destination	Signal Type	Range	Units	Rate (Hz)
QBAR_N	V95P3170C G25207	NAV DERIVED DYNAMIC PRESSURE	ADTA SOP, TAEM GUID GRTLS TAEM GUID TLM	FLOAT PT	0,500	LB/SQ FT	1.04
EAS_N	V95L3175C G25203	NAV DERIVED EQUIVALENT AIRSPEED	ADTA SOP TLM	FLOAT PT	0,600	FT/SEC	1.04
M_N	V90L0847C G25202	NAV DERIVED MACH NUMBER	ADTA SOP TLM	FLOAT PT	0,27	ND	1.04

Table 4.122-3. NAVDAD SOP Initialization Parameters

Name	MSID	Description	Const	Type	Value	Units
SPD_SND_CON		SPEED OF SOUND CONSTANT	X	FLOAT PT	65.77018	FT/SEC/DEG K
G_OVR_R		SEA LEVEL GRAVITY DIVIDED BY THE UNIVERSAL GAS CONSTANT	X	FLOAT PT	1.044300444E-02	DEG K/FT
M*	V97U0443C	WEIGHT RTLS		FLOAT PT	**	SLUGS
			MX			
REF_AREA	V99U5625C	VEHICLE REFERENCE AREA FOR DRAG COEFFICIENT	X	FLOAT PT	2690.	SQ FT
KCD1	V99U5616C	DRAG COEFFICIENT CONSTANT 1	X	FLOAT PT	***	ND
KCD2	V99U5617C	DRAG COEFFICIENT CONSTANT 2	X	FLOAT PT	***	ND
KCD3	V99U5618C	DRAG COEFFICIENT CONSTANT 3	X	FLOAT PT	***	FT/SEC
KCD4	V99U5619C	DRAG COEFFICIENT CONSTANT 4	X	FLOAT PT	***	FT/SEC
KCD5	V99U5620C	DRAG COEFFICIENT CONSTANT 5	X	FLOAT PT	***	1/DEG
KCD6	V99U5621C	DRAG COEFFICIENT CONSTANT 6	X	FLOAT PT	***	1/DEG/DEG
ALT_B(1)	V99U5601C	ALTITUDE BASE OF 1ST LAYER	X	FLOAT PT	***	FT
ALT_B(2)	V99U5602C	ALTITUDE BASE OF 2ND LAYER (MC01)	X	FLOAT PT	***	FT
TMB(1)	V99U5695C	TEMPERATURE AT BASE OF 1ST LAYER	X	FLOAT PT	***	DEG K
TMB(2)	V99U5696C	TEMPERATURE AT BASE OF 2ND LAYER	X	FLOAT PT	***	DEG K
LM(1)	V99U5622C	LAPSE RATE IN 1ST LAYER	X	FLOAT PT	***	DEG K/FT
LM(2)	V99U5623C	LAPSE RATE IN 2ND LAYER	X	FLOAT PT	***	DEG K/FT
RHO_B_ND(1)	V99U5626C	DENSITY AT BASE OF 1ST LAYER	X	FLOAT PT	***	SLUG/FT ³
RHO_B_ND(2)	V99U5627C	DENSITY AT BASE OF 2ND LAYER	X	FLOAT PT	***	SLUG/FT ³
V_NAVDAD	V99U5710C	NAVDAD VELOCITY SWITCHING POINT	X	FLOAT PT	1400.	FT/SEC

*Current vehicle mass I-load parameter for OPS 6 only
 **Value to be obtained from GRTLS TAEM guidance I-load
 ***Henceforth, the current values shall be taken from JSC 19350.

Table 4.122-4. NAVDAD SOP I-Load Parameters

Name	MSID	Description	Const	Type	Value	Units
RHO_B_ND(1)	V99U5626C	DENSITY AT BASE OF 1ST LAYER	X	FLOAT PT	*	SLUG/FT ³
RHO_B_ND(2)	V99U5627C	DENSITY AT BASE OF 2ND LAYER	X	FLOAT PT	*	SLUG/FT ³

†
 *Henceforth, the current values shall be taken from JSC 19350

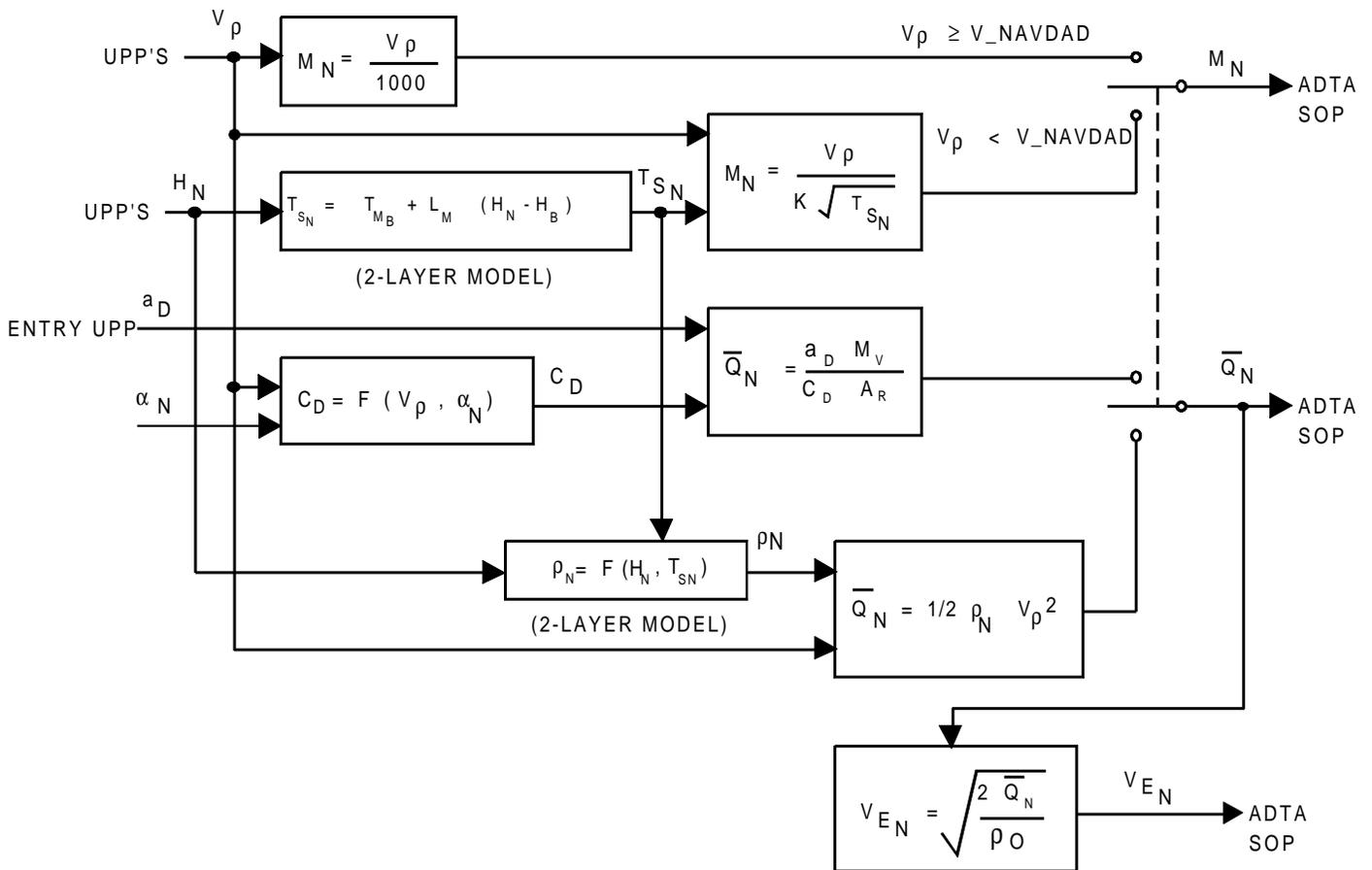


FIGURE 4.122-1. NAVDAD SOP Funtional Block Diagram

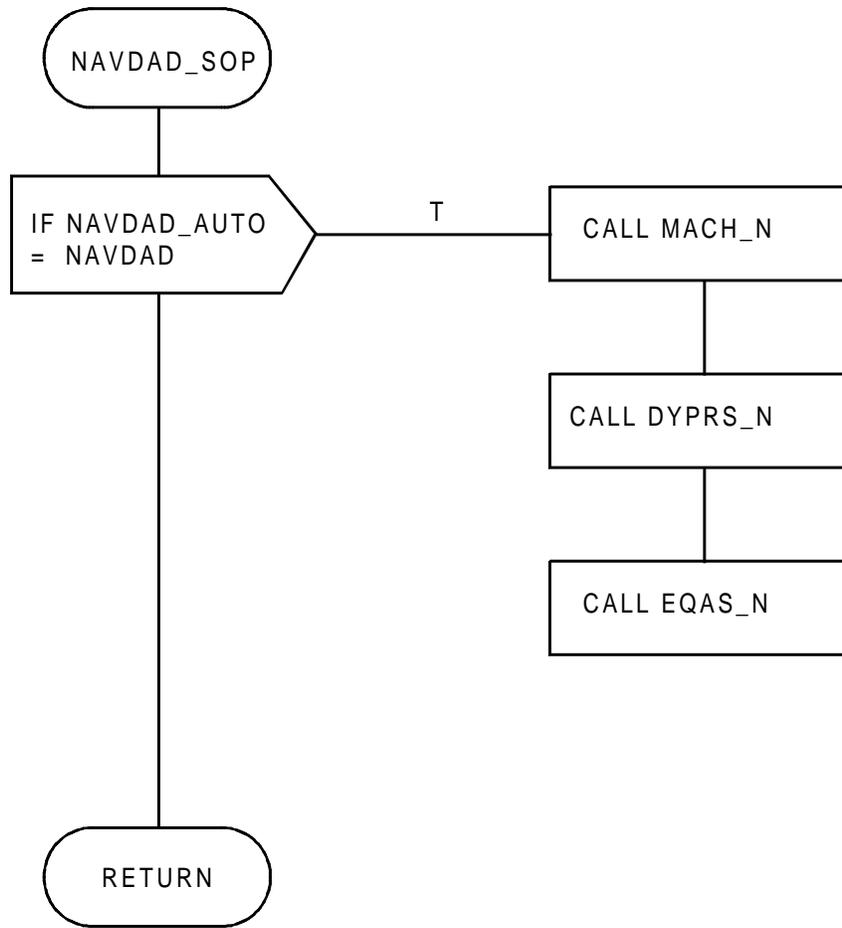


FIGURE 4.122-2. NAVDAD_SOP Functional Module

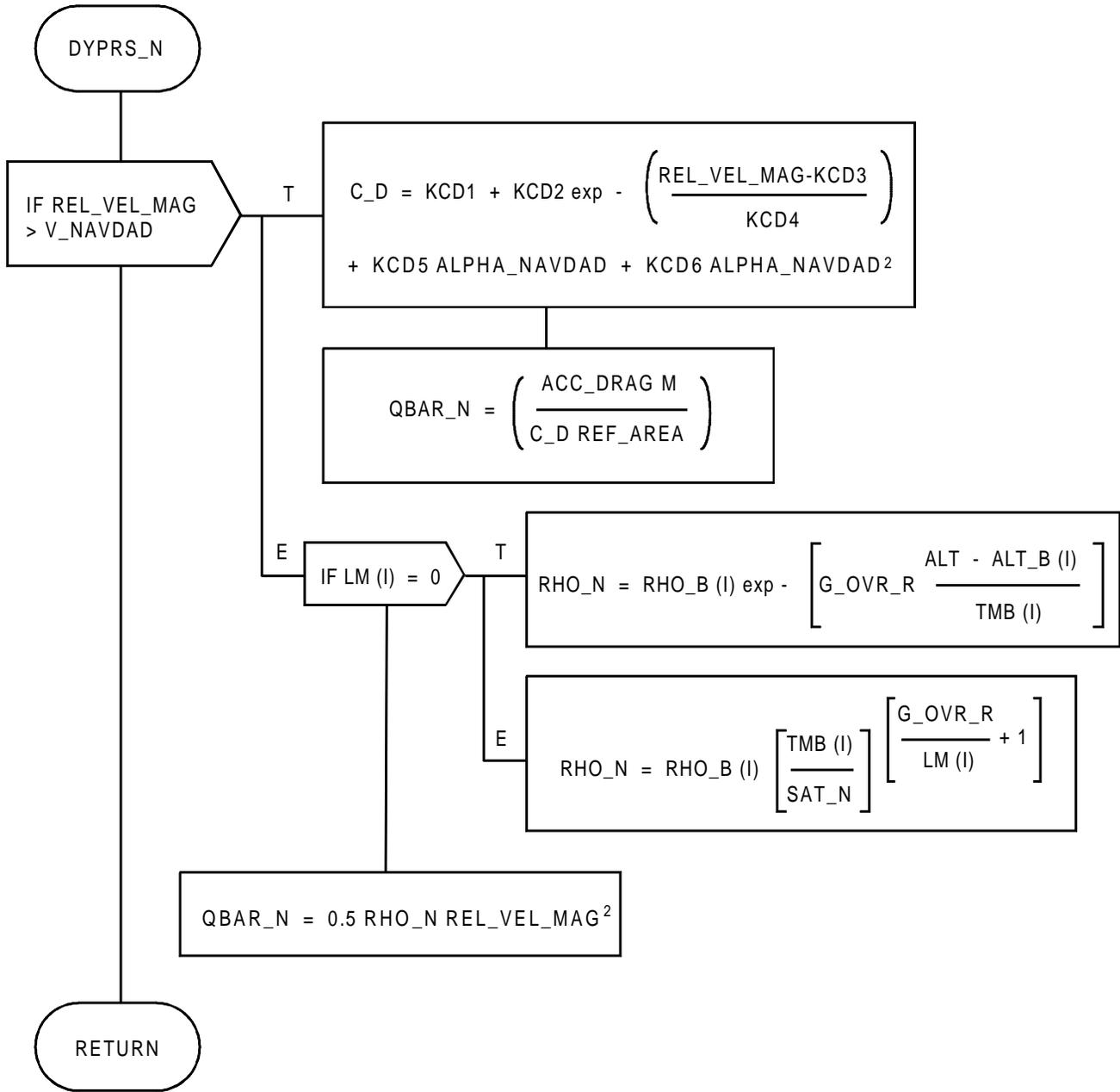


FIGURE 4.122-3. DYPRS_N Functional Module

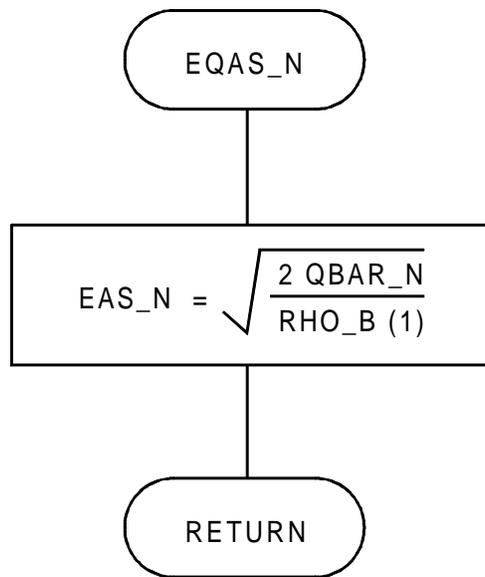


FIGURE 4.122-4. EQAS_N Functional Module

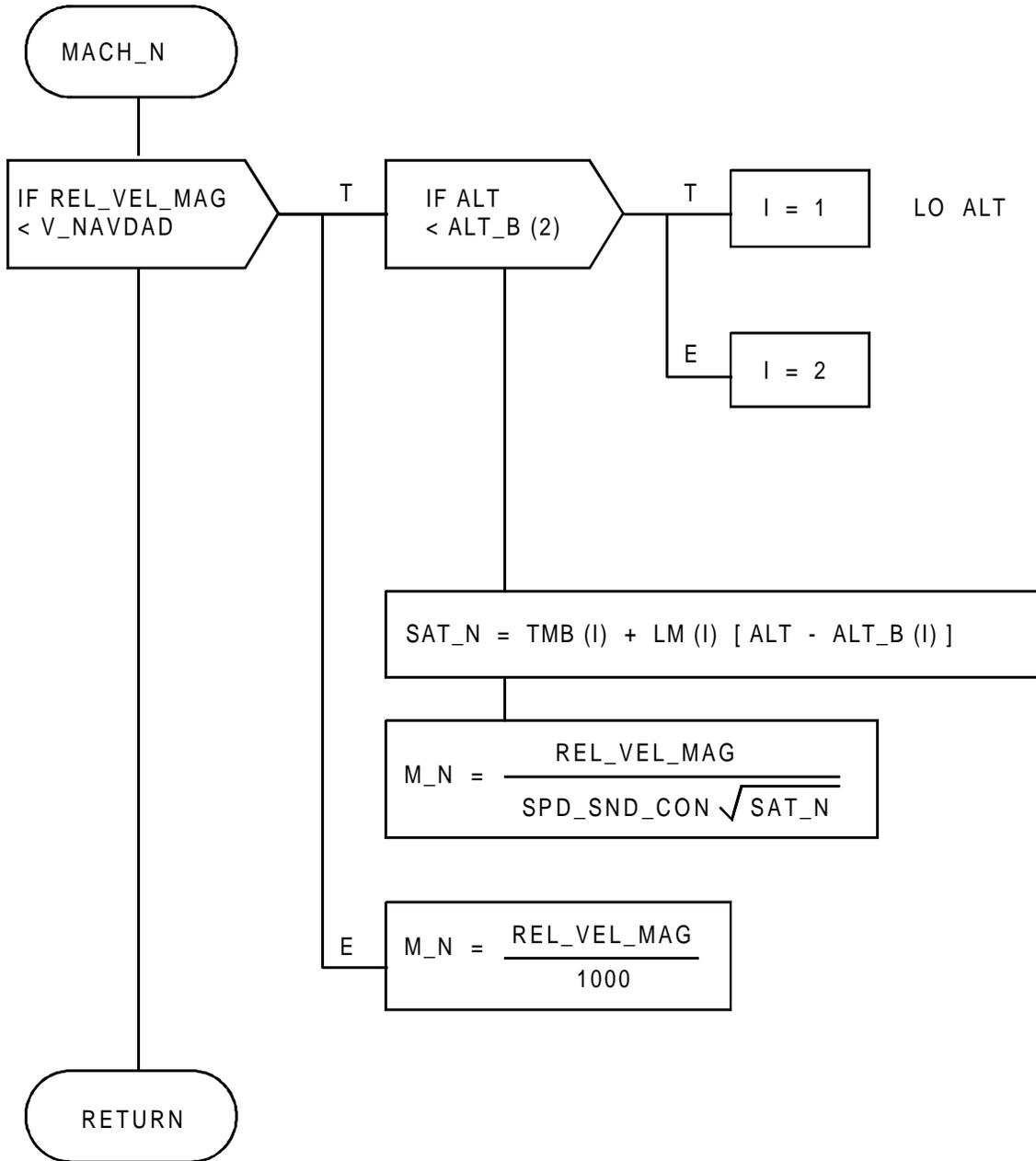


FIGURE 4.122-5. MACH_N Functional Module

This page intentionally left blank.

TABLE 4.122.2.4-1. INPUT FUNCTIONAL PARAMETERS FOR NAVIGATIONALY-DERIVED AIR DATA SOP (G4.122)

FSSR NAME	M/S ID	NOMENCLATURE	SOURCE	UNITS	DATA TYPE	P R E C	LAST CR	R E V
ACC_DRAG	V95A0160CA	DRAG DECEL	RTLS UPP				89979F	
ACC_DRAG	V95A0160CB	DRAG DECEL	ENT UPP	FT/S2			93017G	*
ACC_DRAG	V95A0160CC	DRAG DECEL	TAEM UPP	FT/S2			93017G	*
ALPHA_NAVDAD	V95H0233CA	AVG NAVDAD ANGLE OF ATTACK	ENT UPP	DEG				
ALPHA_NAVDAD	V95H0233CB	AVG NAVDAD ANGLE OF ATTACK	TAEM UPP	DEG				
ALPHA_NAVDAD	V95H0233CC	AVG NAVDAD ANGLE OF ATTACK	A/L UPP	DEG				
ALPHA_NAVDAD	V95H0233CD	AVG NAVDAD ANGLE OF ATTACK	RTLS UPP	DEG				
ALT	V95H0175CA	CURR ORB ALT ABOVE REF ELLIPSOID	RTLS UPP	FT	SPL		93017G	*
ALT	V95H0175CC	CURR ORB ALT ABOVE REF ELLIPSOID	ENT UPP	FT	SPL		93017G	*
ALT	V95H0175CD	CURR ORB ALT ABOVE REF ELLIPSOID	TAEM UPP	FT	SPL		93017G	*
ALT	V95H0175CE	CURR ORB ALT ABOVE REF ELLIPSOID	A/L UPP	FT	SPL		93017G	*
M/CURR_ORB_MASS/MASS	V90U1961CC	CURRENT VEHICLE MASS	DEORB GUID	SLUGS			93090E	
REL_VEL_MAG	V95L0151CA	GND REL VEL MAGNITUDE IN M50 SYS	RTLS UPP	FT/S	SPL		93017G	*
REL_VEL_MAG/V	V95L0151CB	GND REL VEL MAGNITUDE IN M50 SYS	A/L UPP	FT/S	SPL		93017G	*
REL_VEL_MAG/V	V95L0151CC	GND REL VEL MAGNITUDE IN M50 SYS	ENT UPP	FT/S	SPL		93017G	*
REL_VEL_MAG/V	V95L0151CD	GND REL VEL MAGNITUDE IN M50 SYS	TAEM UPP	FT/S	SPL		93017G	*

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

TABLE 4.122.2.4-2. OUTPUT FUNCTIONAL PARAMETERS FROM NAVIGATIONALY-DERIVED AIR DATA SOP (G4.122)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
EAS_N	V95L3175C	NAV DERIVED EQUIVALENT AIRSPEED	ADTA SOP DL FMT 21/1 DL FMT 23/1	FT/S	SPL		93017G	*
M_N	V90L0847C	NAV DERIVED MACH NUMBER	ADTA SOP DL FMT 21/1 DL FMT 23/1		SPL		93017G	*
QBAR_N	V95P3170C	NAV DERIVED DYNAMIC PRESSURE	ADTA SOP DL FMT 21/1 DL FMT 23/1 GRTLS TAEM GUID TAEM GUID	LB/FT2	SPL		93017G	*

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

**Rate Codes: (HDR Only) 1 = 1 sample/sec 2 = 5 samples/sec 3 = 12.5 samples/sec 4 = 25 samples/sec 5 = 100 samples/sec
(HDR and LDR) A = 1 sample/sec B = 5 samples/sec C = 12.5 samples/sec D = 25 samples/sec E = 100 samples/sec**

TABLE 4.122.2.4-3. NAVIGATIONALLY-DERIVED AIR DATA SOP (G4.122) I-LOADS

FSSR NAME	MSID	ENG UNITS	DT	PR	D	S	PR FCTN	MC	CAT	LAST CR	R E V
ALT_B(1)	V99U5601C	FT	F	S	D	C	G4.122	01	PR2	DC0599	
								03	PE2		
ALT_B(2)	V99U5602C	FT	F	S	D	C	G4.122	01	PR2	DC0599	
								03	PE2		
KCD(1)	V99U5616C	ND	F	S	D	C	G4.122	01	DN1	DC0599	
								03	DN3		
KCD(2)	V99U5617C	ND	F	S	D	C	G4.122	01	DN1	DC0599	
								03	DN3		
KCD(3)	V99U5618C	FT/SEC	F	S	D	C	G4.122	01	DN1	DC0599	
								03	DN3		
KCD(4)	V99U5619C	FT/SEC	F	S	D	C	G4.122	01	DN1	DC0599	
								03	DN3		
KCD(5)	V99U5620C	DEG** -1	F	S	D	C	G4.122	01	DN1	DC0599	
								03	DN3		
KCD(6)	V99U5621C	DEG** -2	F	S	D	C	G4.122	01	DN1	DC0599	
								03	DN3		
LM(1)	V99U5622C	DEG K/FT	F	S	D	C	G4.122	01	PR2	DC0599	
								03	PE2		
LM(2)	V99U5623C	DEG K/FT	F	S	D	C	G4.122	01	PR2	DC0599	
								03	PE2		
RHO_B_ND(1)	V99U5626C	SLUGS/FT**3	F	S	D	C	G4.122	01	PR2	DC0599	
								03	PE2		
RHO_B_ND(2)	V99U5627C	SLUGS/FT**3	F	S	D	C	G4.122	01	PR2	DC0599	
								03	PE2		
TMB(1)	V99U5695C	DEG K	F	S	D	C	G4.122	01	PR2	DC0599	
								03	PE2		
TMB(2)	V99U5696C	DEG K	F	S	D	C	G4.122	01	PR2	DC0599	
								03	PE2		
WEIGHT_RTLS	V97U0443C	SLUGS	F	S	M	C	G4.122 G4.204 G4.234	00	GWT	89979F	

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

TABLE 4.122.2.4-4. NAVIGATIONALY-DERIVED AIR DATA SOP (G4.122) K-LOADS

FSSR NAME DESCRIPTION	MSID	MC	KLOAD VALUE	ENG UNITS	DT	P R	S	PR FCTN	LAST CR	R E V
REF_AREA	V99U5625C	00	+2.6900000	E+03	FT**2	F	S	C G4.122	89792E	
VEH REF AREA FOR DRAG COEF								G4.123		
V_NAVDAD	V99U5710C	00	+1.400	E+03	FT/SEC	F	S	C G4.122	59972	
NAVDAD VEL SWITCHING POINT										

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

TABLE 4.122.2.4-5. NAVIGATIONALY-DERIVED AIR DATA SOP (G4.122) CONSTANTS

FSSR NAME DESCRIPTION	MSID	MC	CONSTANT VALUE	ENG UNITS	DT	P R	S	PR FCTN	LAST CR	R E V
G_OVR_R QUOTIENT MSL G/UNIV GAS CONST R	V99U5605C	00	+1.04129	E-02	DEG K/FT	F	S	C A6.9 G4.122	90374C	
SPD_SND_CON SPEED OF SOUND CONSTANT	V99U5628C	00	+6.577035	E+01	FT/SEC/(DEG K)**1/2	F	S	C A6.9 G4.122	90374C	

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

This page intentionally left blank.

4.229 RENDEZVOUS RADAR SOP.

This section defines the detailed functional requirements and formulations for the rendezvous radar (RR) subsystem operating program (SOP). The RR SOP specifies the software requirements associated with RR data input, data conversion, and with display processing. Note that requirements for RR control and antenna pointing are specified by the SM antenna management function and are supported by the GN&C Rendezvous Navigation function.

4.229.1 Rendezvous Radar Requirements Overview.

4.229.1.1 Rendezvous Radar Functional Overview.

The Ku-band Radar/ Communication system is a dual-purpose orbiter subsystem that is used during orbital operations. It functions as either a wide-band communications system for data interchange with the ground or as a rendezvous radar to detect and track detached payloads or other spacecraft before or during rendezvous maneuvers. The functions of communications and radar are not available simultaneously.

The rendezvous radar provides automatic target detection, acquisition, and tracking. It is a target-location and parameter-estimation device that assists in reducing target location uncertainties during rendezvous maneuvers. The targets are earth-orbiting satellites that are stabilized in three axes. The radar tracks a target automatically to determine range, range rate, roll, and pitch angle relative to the orbiter, as well as inertial target angle rates. All output parameters are provided simultaneously during the tracking phase. External assistance and operator intervention are normally required only for functions that initiate radar operation. To restrict search volume to target location uncertainties, angle search is performed about a direction (vector) furnished to the radar by the GPC.

Upon the initiation of search, the radar performs automatic search and indicates that search is in progress for a maximum of 1 minute. If target detection occurs, the search program stops, the search indication is removed, and the radar indicates that detection has occurred. The search indication and detection indication are provided in the status words. Figure 4.229.1.1-1 depicts a single RR subsystem block diagram.

4.229.1.1.1 Rendezvous Radar Digital Interfaces.

The rendezvous radar interfaces with the GPC's via the serial digital multiplex/demultiplex input and output data channels. The digital input words are 16 bits in size, plus 3 bits for synchronization and 1 bit for odd parity. Throughout the rendezvous radar SOP, the synchronization and parity bits are disregarded, and the most significant data bit in the word is designated as bit 0 with the least significant bit being bit 15.

Parameters that are transmitted from the RR to the GPC's are listed below.

- Status Word 1
- Status Word 2
- Roll Angle
- Pitch Angle
- Range Rate (2 words)
- Range (2 words)
- Roll Angle Rate
- Pitch Angle Rate

The contents, format, and bit values of the data words are illustrated in Figures 4.229.1.1.1–1 through 4.229.1.1.1–8.

4.229.1.2 Rendezvous Radar SOP Overview.

The rendezvous radar SOP receives the RR words input to the GN&C GPC and processes the data in preparation for navigation and CRT displays. The processing of range and angle data and rates is controlled by the state of certain status bits. Data good flags indicate the goodness of the RR data. Figure 4.229.1.2–1 illustrates a signal flow.

4.229.1.3 Reference Documents.

MC409–0025, Integrated Communications and Radar Equipment, Ku–Band.

4.229.2 Definitions.

This section is not applicable to the rendezvous radar SOP.

4.229.3 Functional Requirements Traceability.

The detailed requirements and formulations contained herein satisfy the GN&C, Level B, CPDS functional requirements for the rendezvous radar SOP.

4.229.4 Interface Requirements.

The interface parameters required to properly execute the RR SOP functional module are specified in Table 4.229.4–1.

4.229.5 Parameter Initialization.

No I–load parameters are required for the RR SOP.

4.229.6 Detailed Requirements and Formulations.

This section defines the RR SOP functional modules in sufficient detail to create a functional design and generate computer code. Each module defines the functional requirements, interface requirements, detailed requirements and formulations, and special considerations if applicable. A flow diagram is also included in Figure 4.229.6.1-1; however, the flow diagram is supplementary material and is not intended to convey requirements or to place restrictions on functional design or coding.

4.229.6.1 RR SOP.

The purpose of the RR SOP is to process range, range rate, line of sight (LOS) angle, and inertial LOS angle rate data and associated flags for navigation and displays.

A. Functional Requirements. The RR_SOP module is required to perform the following functions.

1. Acquire the RR digital data words, input data time tag, RR comfault flag, and Mean-of-50 to body quaternion from COMPOOL.
2. Monitor the RR commfault flag and status words to determine whether to process RR input data.
3. Perform RR data time tag processing.
4. Scale the range, range rate, roll, roll rate, pitch, and pitch rate data and convert from fixed point to floating point format.
5. Store the Mean-of-50 to body quaternion for use by navigation.
6. Identify self-test data, as such, for navigation.
7. Monitor the RR status words and process the RR range, range rate, and angle data good flags; the range, range rate, angle, and angle rate status words; the radar mode status word; and the RR VALID flag.

B. Interface Requirements. The input and output parameters associated with the RR_SOP module are contained in Table 4.229.4-1.

C. Detailed Requirements. The following operations are required for performance of the RR_SOP functional module.

1. Initialization. Before the RR SOP is first executed, the TRK_LAST_CYCLE flag should be set false, RR_STAT_DISP should be set equal to six and the RR_VALID flag should set to TRUE.
2. Data Acquisition. RR data shall be acquired from the radar electronics assembly (EA-1) via MDM FF3 at a rate of 6-1/4 Hz and made available to the RR SOP along with the RR commfault flag and input data time tag at a rate of 1.04 Hz. GN&C requests for data from the Ku-band system shall be nominally spaced with respect to each other and the SM requests for data from the Ku-band system so that a minimum of 122 μ seconds elapses after the last data word of a message and the first data word of the next response message at the MDM output(s) to the Ku-band system.

(NOTE: If the two requests for data from the Ku-band system occur such that the Ku-band system must respond via both SIO channels concurrently, then the Ku-band system may not respond properly to one or both data requests. The software contractor shall insure that the software will continue to execute properly when this occurs.)

Reference paragraph 6.3.3 of SS-P-0002-580, Level B SM CPDS and paragraph 20.3.1.2.4.2.1 of MC409-0025, Ku-band for more detail. The RR SOP shall also acquire the quaternion, M50 to measured body, from the ATT_PROC principal function at a rate of 1.04 Hz.

3. Data Processing. Set the RR_SELF_TEST flag false. Set the RR range, range rate, and angle data good flags false. Set the RR range, range rate, angle, and angle rate status words equal to 1. Monitor the RR_COM_FLT flag. If the flag is set true, discontinue processing. If the RR_COM_FLT flag is set false, monitor the radar on bit (bit 0 of status word 1). If the radar on bit is false, set RR_STAT_DISP to 0 (COMM mode) and discontinue processing. If the radar on bit is true, perform steps 3a through 3h and proceed to step 4:
 - a. Set the RR data time tag, RR_TIM, equal to the computer clock time, CLOCK, corresponding to the current RR data input transaction time.
 - b. Form the radar range output word, RR_RANGE_O, by combining RR_RANGE_A (16 bits) with the seven most significant bits of RR_RANGE_B and scaling to form a floating point word in units of feet.
 - c. Form the radar range rate output word RR_RANGRO, by using RR_RNGRTE_A (15 bits plus sign) and scaling to form a floating point word (with correct sign) in units of feet per second.
 - d. Form the radar roll angle output word, RR_ROLLO, by scaling and converting RR_ROLL_IN (12 bits plus sign) to floating point (with correct sign) in units of degrees.
 - e. Form the radar roll rate output word, RR_RRTEO, by scaling and converting RR_RRTE_IN (14 bits plus sign) to floating point (with correct sign) in units of milliradians per second.
 - f. Form the radar pitch angle output word, RR_PITCHO, by scaling and converting RR_PITCH_IN (11 bits plus sign) to floating point (with correct sign) in units of degrees.
 - g. Form the radar pitch rate output word, RR_PRTEO, by scaling and converting RR_PRTE_IN (14 bits plus sign) to floating point (with correct sign) in units of milliradians per second.
 - h. Update Q_M50BODY_RR (i)(i=1,4) with the currently acquired quaternion, M50 to measured body.
4. Mode Status. Examine the RR Status Word 1, (RR_STAT1), and set the RR mode status word, RR_STAT_DISP, in the following order.
 - a. If bit 9 of RR_STAT1 is true, set RR_STAT_DISP to 5 (self-test in progress), set RR_SELF_TEST true, and proceed to step 6; otherwise continue with step 4b.
 - b. If bit 3 of RR_STAT1 is true, set the RR_STAT_DISP to 3 (GPC acquisition mode) and proceed to step 5. Otherwise set TRK_LAST_CYCLE false and continue with step 4c.
 - c. If bit 2 of RR_STAT1 is true, set RR_STAT_DISP to 2 (GPC designate mode) and proceed to step 6; otherwise continue with step 4d.

- d. If bit 1 of RR_STAT1 is true, set RR_STAT_DISP to 1 (auto-track mode) and proceed to step 6. Otherwise set RR_STAT_DISP to 4 (manual mode) and proceed to step 6.
5. GAX Processing. Processing of the RR_VALID flag occurs only if the RR is in the GPC acquisition mode and proceeds as follows:
 - a. Monitor the RR track flag—bit 2 in the RR status word 2, RR_STAT2. If bit 2 of RR_STAT2 is true, set the RR_VALID flag true and proceed to step 5b. If bit 2 of RR_STAT2 is false, examine the flag indicating the status of tracking last cycle, TRK_LAST_CYCLE. If TRK_LAST_CYCLE is true, set the RR_VALID flag false and proceed to step 5b. If TRK_LAST_CYCLE is false, proceed to step 5b.
 - b. Set the TRK_LAST_CYCLE flag equal to bit 2 of RR_STAT2 and proceed to step 6.
6. RR Data Status and Data Good Processing. Perform steps 6a through 6d, then discontinue processing. (Note that all data goods are set false in paragraph 4.229.6.1.C3.)
 - a. If bit 7 of RR_STAT1 is true, set RR_ANG_DG true and ANG_STATUS to 0; otherwise, set ANG_STATUS to 2.
 - b. If bit 8 of RR_STAT1 is true, set RR_ANGR_DG true and ANGRT_STATUS to 0; otherwise set ANGRT_STATUS to 2.
 - c. If bit 14 of RR_STAT2 is true, set RNG_STATUS to 0; otherwise set RNG_STATUS to 2.
 - d. If bit 15 of RR_STAT2 is true, set RR_RNGR_DG true, and RNGRT_STATUS to 0; otherwise, set RNGRT_STATUS to 2.

D. Special Considerations. The following special considerations are applicable to the RR SOP software design and operational utilization.

1. The software implementation of the RR SOP shall protect against program checks and error interrupts caused by transient type errors or ad data inputs. Specifically, the RR SOP shall protect against all of the error conditions identified in paragraph 3.6 which are applicable to the RR software design and code.
2. The requirement for the data good flags could be satisfied by utilizing the range, range rate, angle, and angle rate status words. The only difference is that zero is used to indicate good data (true), and 1 and 2 values drive different displays both of which are flags of missing or bad data (false).

3. The integers associated with the range, range rate, angle, and angle rate status words are intended to convey the following information to the REL NAV display.

<u>Integer</u>	<u>REL NAV Response</u>
0	Normal data – display
1	Blank data and display an “M” (commfault or in COMM mode)
2	Display data with a “↓”

4. The purpose of the requirements in paragraph 4.229.6.1C5, GAX processing, is to generate an alarm only in the case where a loss of track occurs in the GPC acquisition mode after the RR has initially acquired and tracked the target. GAX sets off a Class 3 alert when it receives an RR_VALID flag set to the false state.
5. The integer associated with the radar mode status word (RR_STAT_DISP) is intended to convey the following information to the REL NAV:

<u>Integer</u>	<u>REL NAV Response</u>
0	COMM mode
1	AUTO_TRACK mode
2	GPC_DES mode
3	GPC_ACQ mode
4	MANUAL mode
5	SELF_TEST mode
6	INITIALIZATION

6. The RR_SELF_TEST flag is intended to notify rendezvous navigation not to use the radar data regardless of the states of the individual data good discretets when the RR is in the self-test mode.
7. Referring to paragraphs 4.229.6.1.C.3.b–g, several of the radar serial data words do not use a full 16 bits for transmission of numerical data. These bits are reserved for use by the radar hardware and should be ignored when converting data to GPC floating point words.

4.229.7 Applicable Tables for Rendezvous Radar SOP.

The IDD inputs and outputs are listed in Tables 4.229.7-1 and 4.229.7-2. Values for the I-loads are contained in the I-loads requirements document (JSC-19350); however, I-load definitions applicable to this principal function are listed in Table 4.229.7-3. K-loads are listed in Table 4.229.7-4. Constants are listed in Table 4.229.7-5.

Table 4.229.4-1. Rendezvous Radar SOP Interface Table

Name	MSID	Description	Input Source	Output Destination	Signal Type	Range	Units	Rate (Hz)
RR_PITCH_IN	V74H2920B G33318	KU BAND A CH 2 PITCH WORD 4 WD 4	RR	TLM	FIXED PT	+/-180	DEG	6.25
RR_RNGRTE_A	V74R2921B G33314	KU BAND A CH 2 RANGE RATE WD 5	RR	TLM	FIXED PT	+/-1638	FT/ SEC	6.25
RR_RANGE_A	V74H2924B G33312	KU BAND A CH 2 RANGE WORD 7 WD 7	RR	TLM	FIXED PT	40.0 2.6 E+6	FEET	6.25
RR_RANGE_B	V74H2925B G33313	KU BAND A CH 2 RANGWORD 8 WD 8	RR	TLM	FIXED PT	0.31- 39.7	FEET	6.25
RR_PRTE_IN	V74R2927B G33319	KU BAND A CH 2 PITCH RATE WD 10	RR	TLM	FIXED PT	+/-16.4	DEG/ SEC	6.25
RR_ROLL_IN	V74R2918B G33316	KU BAND A CH 2 ROLL WORD 3 WD 3	RR	TLM	FIXED PT	+/-360	DEG	6.25
RR_RRTE_IN	V74R2926B G33317	KU BAND A CH 2 ROLL RATE WD 9	RR	TLM	FIXED PT	+/-16.4	DEG/ SEC	6.25
RR_STAT1	V74M2882P G33324	KU-A CH 2 STATUS WORD 1 WD 1	RR	TLM	FIXED PT	NA	NA	6.25
RR_STAT2	V74M2900P G33325	KU-A CH 2 STATUS WORD 2 WD 2	RR	TLM	FIXED PT	NA	NA	6.25
RR_COM_FLT	V91X2467X	R RADAR COMMFault	FCOS		LOGICAL	0,1	NA	1.04
CLOCK	V91W5000C G07829	CLOCK-COMPUTER (GMT)	FCOS		FLOAT PT	NA	D:H:M: S	1.04
Q_BOD_M50(1)	V90U2240CB G19849	M50 TO BODY QUAT MEASURED ELEMENT 1	ATT_PROC 4.97		FLOAT PT	NA	NA	1.04
Q_BOD_M50(2)	V90U2241CB G19850	M50 TO BODY QUAT MEASURED ELEMENT 2	ATT_PROC 4.97		FLOAT PT	NA	NA	1.04
Q_BOD_M50(3)	V90U2242CB G19851	M50 TO BODY QUAT MEASURED ELEMENT 3	ATT_PROC 4.97		FLOAT PT	NA	NA	1.04
Q_BOD_M50(4)	V90U2243CB G19852	M50 TO BODY QUAT MEASURED ELEMENT 4	ATT_PROC 4.97		FLOAT PT	NA	NA	1.04
RR_RNG_DG	V95X4293X G33310	R RADAR RANGE DATA GOOD		ORB/RND NAV 4.126 TLM	LOGICAL	0,1	NA	O.D.

Table 4.229.4-1. Rendezvous Radar SOP Interface Table

Name	MSID	Description	Input Source	Output Destination	Signal Type	Range	Units	Rate (Hz)
RR_ENGR_DG	V95X4294X G33311	R RADAR RANGE RATE DATA GOOD		ORB/RND NAV 4.126 TLM	LOGICAL	0,1	NA	O.D.
RR_ANG_DG	V95X4295X G33326	R RADAR ANGLE DATA GOOD		ORB/RND NAV 4.126 TLM	LOGICAL	0,1	NA	O.D.
RR_RANGEO	V95H4287C G33305	R RADAR RANGE OUT		ORB/RND NAV 4.126 REL NAV 6.11	FLOAT PT	0-2.8 X 10**6	FEET	O.D. 1.04
RR_RNGRO	V95R4288C G33306	R RADAR RANGE RATE OUT		ORB/RND NAV 4.126 REL NAV 6.11	FLOAT PT	+/-2800	FT/SEC	O.D. 1.04
RR_PITCHO	V95H4281C G33301	R RADAR PITCH ANGLE OUT		ORB/RND NAV 4.126 REL NAV 6.11	FLOAT PT	+/-180	DEG	O.D. 1.04
RR_PRTEO	V95R4282C G33302	R RADAR PITCH RATE OUT		REL NAV 6.11	FLOAT PT	+/-286.2	MILLI RAD/ SEC	1.04 O.D.
RR_ROLLO	V95H4284C G33303	R RADAR ROLL ANGLE OUT		ORB/RND NAV 4.126 REL NAV 6.11	FLOAT PT	+/-360	DEG	O.D. 1.04
RR_RRTEO	V95R4285C G33304	R RADAR ROLL RATE OUT		REL NAV 6.11	FLOAT PT	+/-286.2	MILLI RAD/ SEC	1.04 O.D.
RR_TIM	V95W4292C G50298	R RADAR TIME		ORB/RND NAV 4.126	FLOAT PT	NA	SEC	O.D.
RR_VALID	V95X4283X G50297	R RADAR DATA VALID		GAX 4.110	LOGICAL	0,1	NA	1.04
RR_SELF_TEST	V95X4309X G50380	R RADAR SELF TEST PROGRESS		ORB/RND NAV 4.126	LOGICAL	0,1	NA	O.D.
RNG_STATUS	V95J4298C G50293	R RADAR RANGE STATUS WORD		REL NAV 6.11	INTEGER	0,1,2	NA	1.04
RNGRT_STATUS	V95J4299C G50294	R RADAR RANGE RATE STATUS WORD		REL NAV 6.11	INTEGER	0,1,2	NA	1.04
ANG_STATUS	V95J4300C G50295	R RADAR ANGLE STATUS WORD		REL NAV 6.11	INTEGER	0,1,2	NA	1.04
ANGRT_STATUS	V95J4302C G50296	R RADAR ANGLE RATE STATUS WORD		REL NAV 6.11	INTEGER	0,1,2	NA	1.04
RR_STAT_DISP	V95J4303C G50948	RADAR MODE STATUS WORD		REL NAV 6.11	INTEGER	0-6	NA	1.04
Q_M50BODY_ RR(1)	V95U4312C G50376	M50 TO BODY RR ELEMENT 1		ORB/RND NAV 4.126	FLOAT PT	NA	NA	O.D.

Table 4.229.4-1. Rendezvous Radar SOP Interface Table

Name	MSID	Description	Input Source	Output Destination	Signal Type	Range	Units	Rate (Hz)
Q_M50BODY_RR(2)	V95U4313C G50377	M50 TO BODY RR ELEMENT 2		ORB/RND NAV 4.126	FLOAT PT	NA	NA	O.D.
Q_M50BODY_RR(3)	V95U4314C G50378	M50 TO BODY RR ELEMENT 3		ORB/RND NAV 4.126	FLOAT PT	NA	NA	O.D.
Q_M50BODY_RR(4)	V95U4315C G50379	M50 TO BODY RR ELEMENT 4		ORB/RND NAV 4.126	FLOAT PT	NA	NA	O.D.

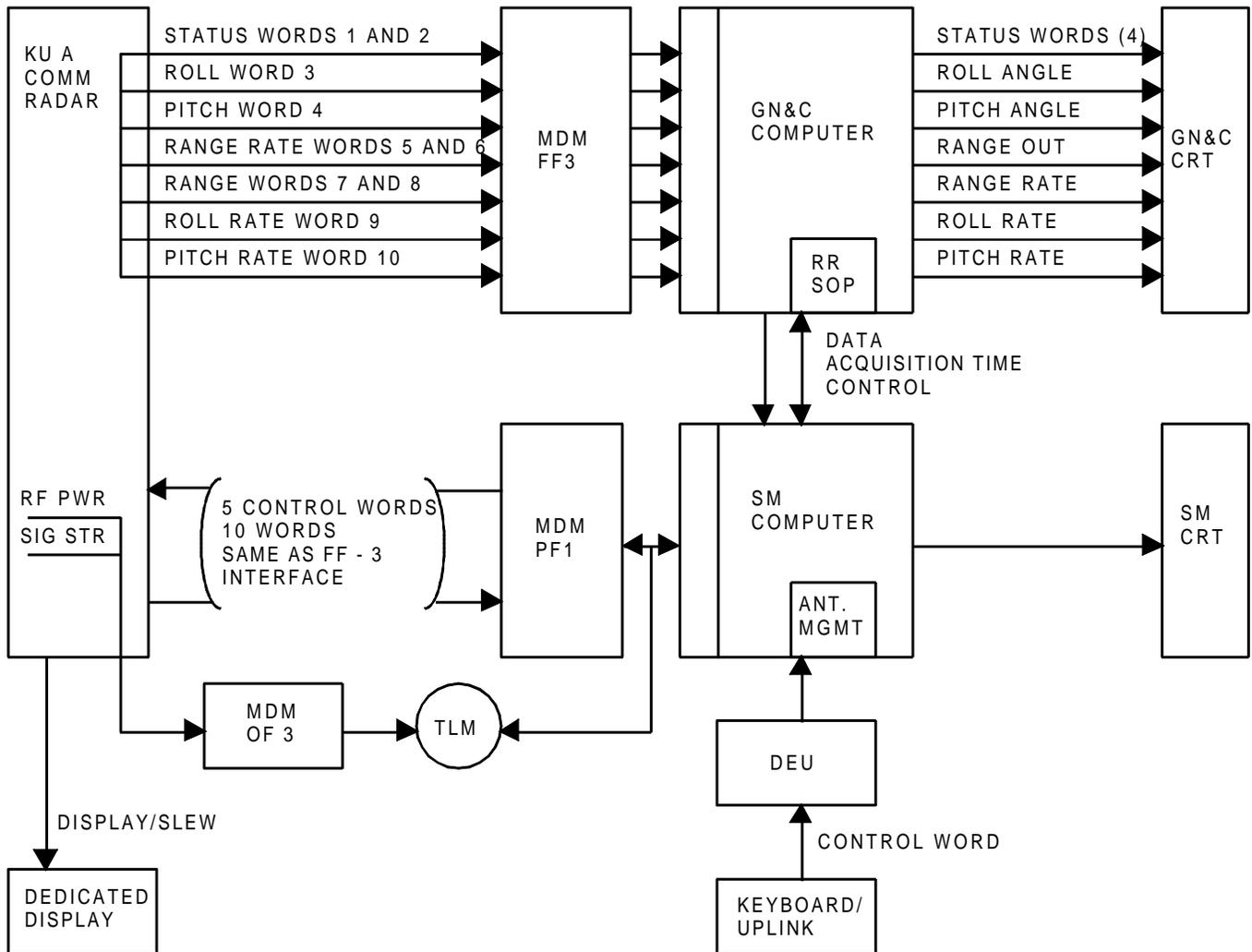


FIGURE 4.229.1.1-1. RR Subsystem Block Diagram

<u>BIT NO.</u>	RR_RANGE_A (WORD 1)	<u>VALUE (FT)</u>
0		1,310,720
1		655,360
2		327,680
3		163,840
4		81,920
5		40,960
6		20,480
7		10,240
8		5,120
9		2,560
10		1,280
11		640
12		320
13		160
14		80
15	RR_RANGE_B (WORD 2)	40
0		20
1		10
2		5
3		5/2
4		5/4
5		5/8
6		5/16

FIGURE 4.229.1.1.1-1. RR Range Words

<u>BIT NO.</u>	<u>EXACT VALUE</u>	<u>ROUNDED VALUE</u>
0	+/- (0/1)	
1	$180^\circ \times 2^0$	180.0
2	$180^\circ \times 2^{-1}$	90.0
3	$180^\circ \times 2^{-2}$	45.0
4	$180^\circ \times 2^{-3}$	22.5
5	$180^\circ \times 2^{-4}$	11.25
6	$180^\circ \times 2^{-5}$	5.625
7	$180^\circ \times 2^{-6}$	2.8125
8	$180^\circ \times 2^{-7}$	1.40625
9	$180^\circ \times 2^{-8}$	0.703125
10	$180^\circ \times 2^{-9}$	0.3515625
11	$180^\circ \times 2^{-10}$	0.1757812
12	$180^\circ \times 2^{-11}$	0.0878906
13	NOT USED	
14	NOT USED	
15	NOT USED	

FIGURE 4.229.1.1.1-2. RR Roll Angle Word

<u>BIT NO.</u>	<u>EXACT VALUE</u>	<u>ROUNDED VALUE</u>
0	+/- (0/1)	
1	$90^\circ \times 2^0$	90.0
2	$90^\circ \times 2^{-1}$	45.0
3	$90^\circ \times 2^{-2}$	22.5
4	$90^\circ \times 2^{-3}$	11.25
5	$90^\circ \times 2^{-4}$	5.625
6	$90^\circ \times 2^{-5}$	2.8125
7	$90^\circ \times 2^{-6}$	1.40625
8	$90^\circ \times 2^{-7}$	0.703125
9	$90^\circ \times 2^{-8}$	0.3515625
10	$90^\circ \times 2^{-9}$	0.1757812
11	$90^\circ \times 2^{-10}$	0.0878906
12	NOT USED	
13	NOT USED	
14	NOT USED	
15	NOT USED	

FIGURE 4.229.1.1.1-3. RR Pitch Angle Word

<u>BIT NO.</u>	<u>RR_RNGRTE_A (WORD 1)</u>	<u>VALUE (FT/SEC)</u>
0		+/- (0/1)
1		819.2
2		409.6
3		204.8
4		102.4
5		51.2
6		25.6
7		12.8
8		6.4
9		3.2
10		1.6
11		0.8
12		0.4
13		0.2
14		0.1
15		0.05
	<u>RR_RNGRTE_B (WORD 2)</u>	
0		NOT USED
1		NOT USED
2		NOT USED

FIGURE 4.229.1.1.1-4. RR Range Rate Words

<u>BIT NO.</u>	<u>VALUE_(°/SEC)</u>
0	+/- (0/1)
1	8.192
2	4.096
3	2.048
4	1.024
5	0.512
6	0.256
7	0.128
8	0.064
9	0.032
10	0.016
11	0.008
12	0.004
13	0.002
14	0.001
15	NOT USED

FIGURE 4.229.1.1.1-5. RR Roll Angle Rate Word

<u>BIT NO.</u>	<u>VALUE_(°/SEC)</u>
0	+/- (0/1)
1	8.192
2	4.096
3	2.048
4	1.024
5	0.512
6	0.256
7	0.128
8	0.064
9	0.032
10	0.016
11	0.008
12	0.004
13	0.002
14	0.001
15	NOT USED

FIGURE 4.229.1.1.1-6. RR Pitch Angle Rate Word

<u>BIT NO.</u>	<u>PARAMETER</u>
0	COMM A/RADAR ON (CR)
1	ANTENNA STEERING AUTO MODE (CR)
2	ANTENNA STEERING GPC DESIGNATE MODE (CR)
3	ANTENNA STEERING GPC ACQ MODE (CR)
4	PARITY ERROR (CR)
5	SYNC ERROR (CR)
6	VALIDITY ERROR (CR)
7	RADAR ANGLE DATA GOOD (R)
8	RADAR ANGLE RATE DATA GOOD (R)
9	RADAR SELF-TEST IN PROGRESS (R)
10	Ku A/B SELECT (C)
11	RDR THLD VAR/FIX (R)
12	OBSC INTLK OVRD (CR)
13	SPARE
14	SPARE
15	SPARE
	(C) COMMUNICATION FUNCTION
	(CR) COMMUNICATION AND RADAR FUNCTION
	(R) RADAR FUNCTION

FIGURE 4.229.1.1.1-7. RR Status Word 1

<u>BIT NO.</u>	<u>PARAMETER</u>
0	SEARCH (CR)
1	DETECT (CR)
2	TRACK (CR)
3	OPERATE (CR)
4	EA-1 RADAR TEST STATUS (R)
5	DA RADAR TEST STATUS (R)
6	EA-2 RADAR TEST STATUS (CR)
7	SYSTEM TEST STATUS (R)
8	TDRS WEST (C)
9	TDRS EAST (C)
10	SPARE
11	FWD LINK DATA FRAME SYNC (C)
12	DATA PRESENT (C)
13	SCAN WARN LIMIT (CR)
14	RADAR RANGE DATA GOOD (R)
15	RADAR RANGE RATE DATA GOOD (R)
	(C) COMMUNICATION FUNCTION
	(CR) COMMUNICATION AND RADAR FUNCTION
	(R) RADAR FUNCTION

FIGURE 4.229.1.1.1-8. RR Status Word 2

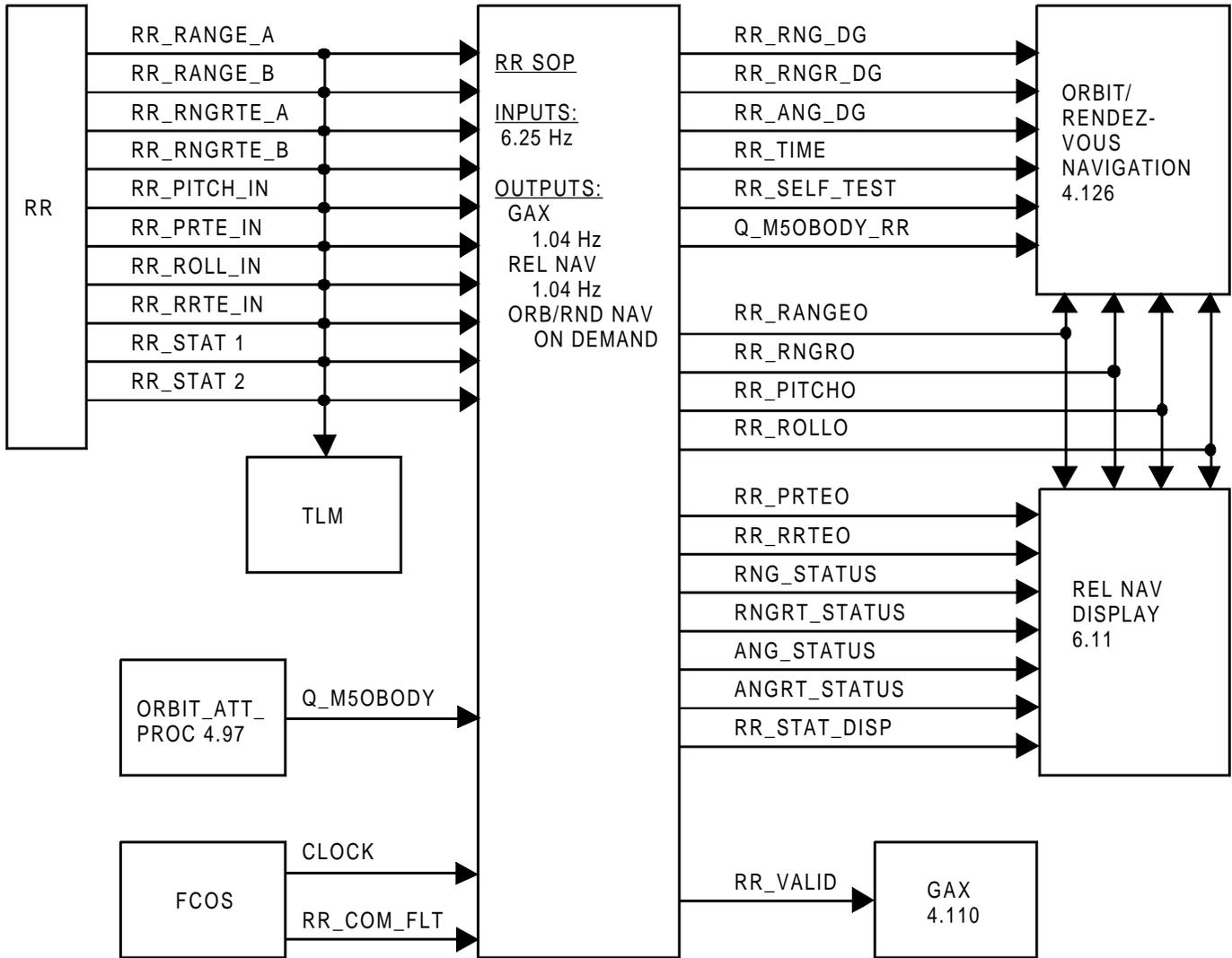


FIGURE 4.229.1.2-1. RR SOP Signal Flow Diagram

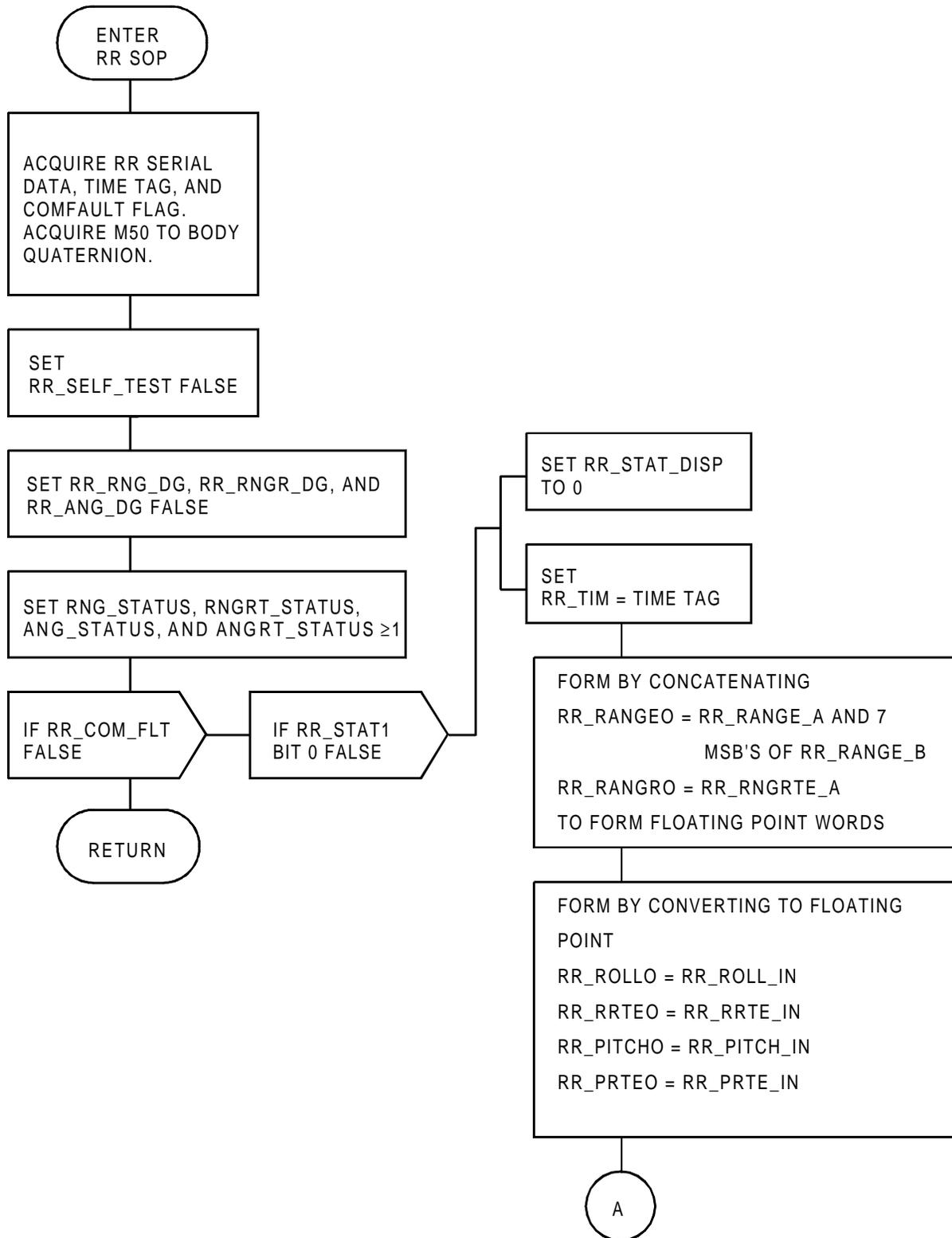


FIGURE 4.229.6.1-1.(1) RR SOP Flow Diagram (Sheet 1 of 3)

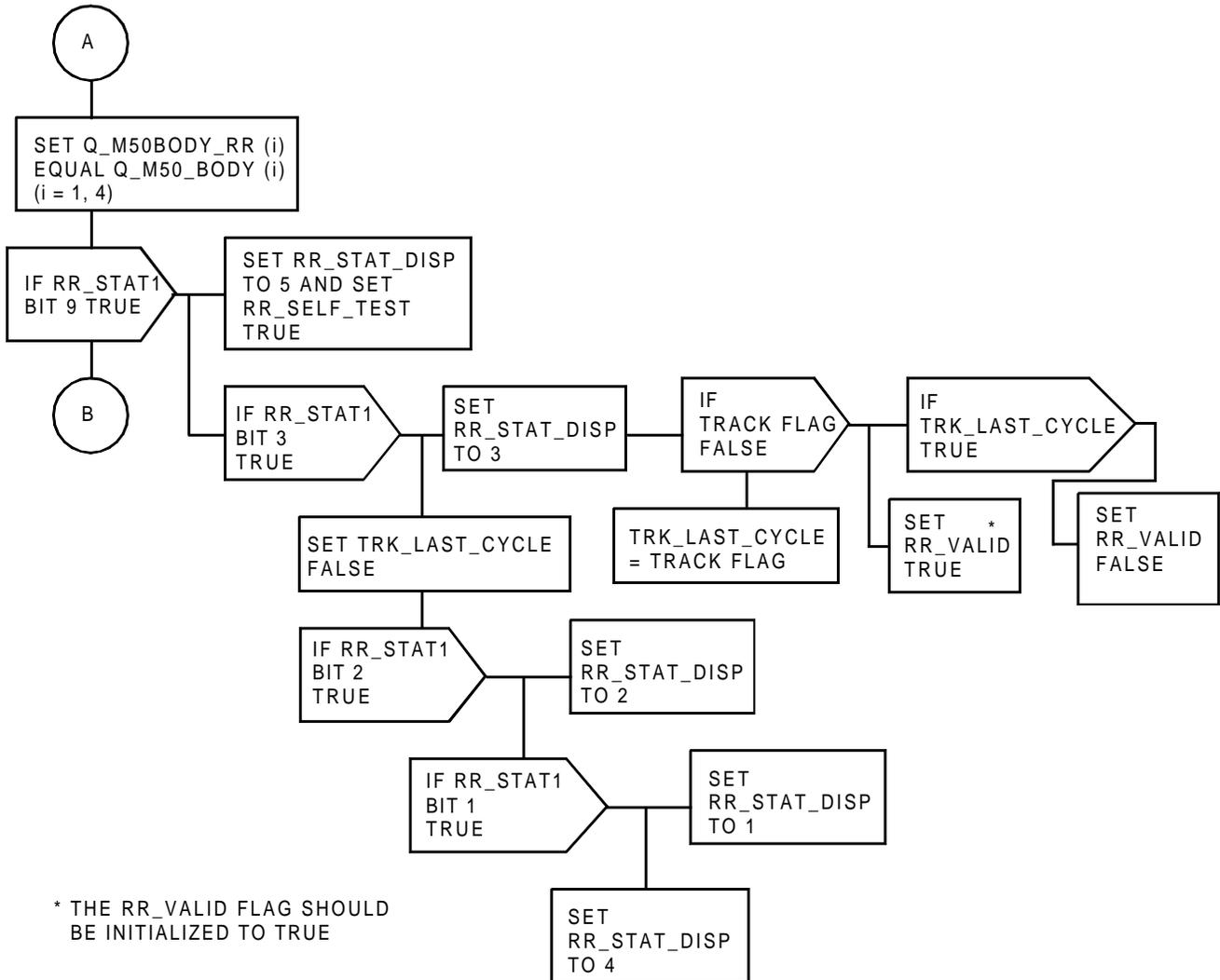


FIGURE 4.229.6.1-1.(2) RR SOP Flow Diagram (Sheet 2 of 3)

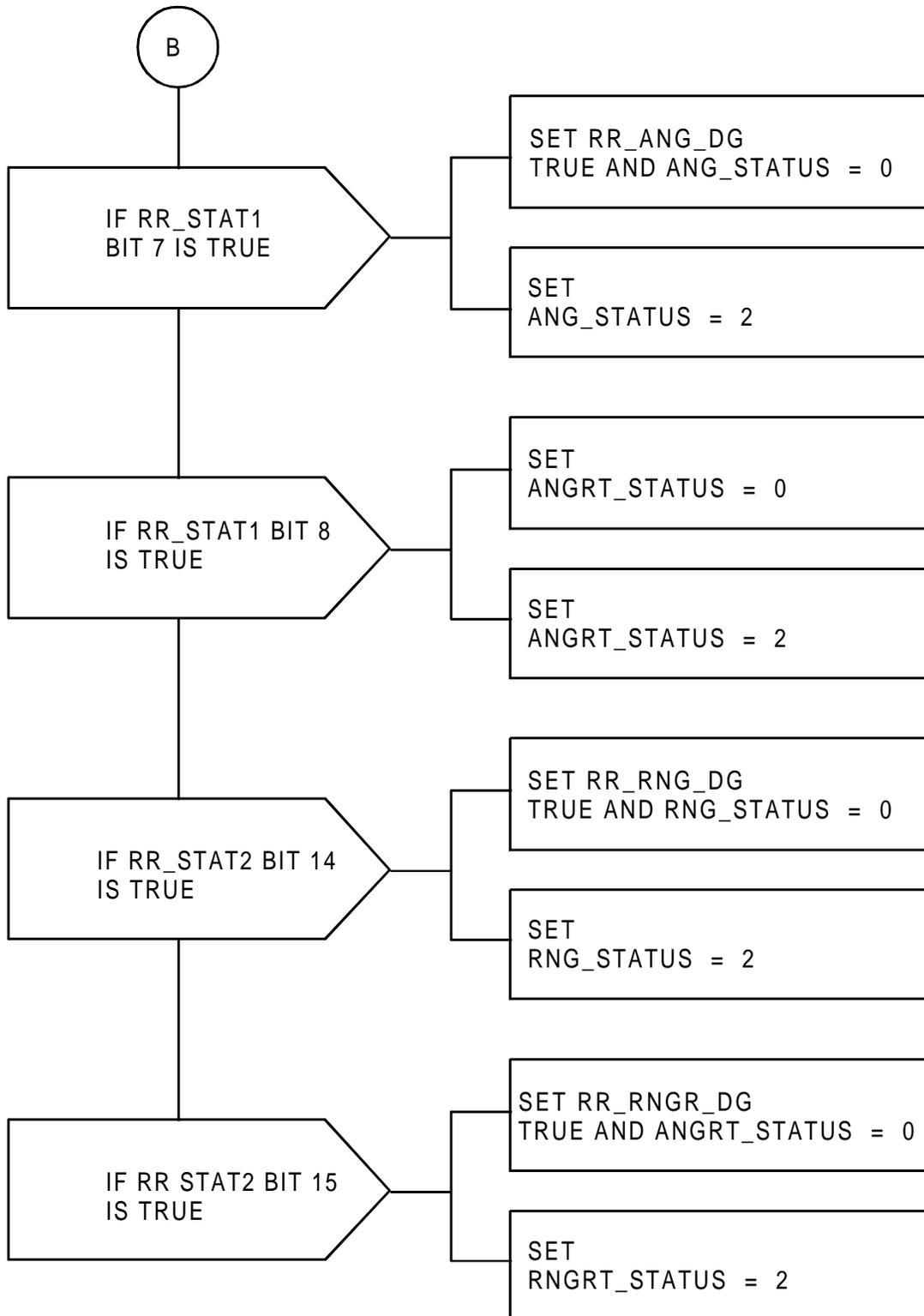


FIGURE 4.229.6.1-1.(3) RR SOP Flow Diagram (Sheet 3 of 3)

TABLE 4.229.7-1. INPUT FUNCTIONAL PARAMETERS FOR RENDEZVOUS RADAR SUBSYSTEM OPERATING PROGRAM (G4.229)

FSSR NAME	M/S ID	NOMENCLATURE	SOURCE	UNITS	DATA TYPE	P R E C	LAST CR	R E V
CLOCK/CLOCKTIME	V91W5000C	CLOCK-COMPUTER(GMT)	FCOS	S			93160B	*
Q_BOD_M50(1)/Q_B_M50(1)	V90U2240CB	M50 TO BODY QUAT MEASURED ELEM 1	ORB ATT PROC		SPL		93017G	*
Q_BOD_M50(2)/Q_B_M50(2)	V90U2241CB	M50 TO BODY QUAT MEASURED ELEM 2	ORB ATT PROC		SPL		93017G	*
Q_BOD_M50(3)/Q_B_M50(3)	V90U2242CB	M50 TO BODY QUAT MEASURED ELEM 3	ORB ATT PROC		SPL		93017G	*
Q_BOD_M50(4)/Q_B_M50(4)	V90U2243CB	M50 TO BODY QUAT MEASURED ELEM 4	ORB ATT PROC		SPL		93017G	*
RR_COM_FLT	V91X2467X	R RADAR COMMFAULT	FCOS					
RR_PITCH_IN	V74H2920B	KU BAND A CH 2 PITCH WORD 4	WD4 KU BAND A EA 1	DEG				
RR_STAT1	V74M2882P	KU-A CH2 STATUS WORD 1	WD1 KU BAND A EA 1		HMD			
RR_STAT2	V74M2900P	KU-A CH2 STATUS WORD 2	WD2 KU BAND A EA 1		HMD			
	V74H2918B	KU BAND A CH2 ROLL WORD 3	WD3 VU				93017G	*
	V74H2924B	KU-A CH 2 RDR RANGE BITS 01-16	WD7 VU				93017G	*
	V74H2925B	KU-A CH 2 RDR RANGE BITS 01-07	WD8 VU				93017G	*
	V74R2921B	KU-A CH2 RANGE RATE BITS 01-16	WD5 VU				93017G	*
	V74R2926B	KU BAND A CH 2 ROLL RATE	WD9 VU				93017G	*
	V74R2928B	KU BAND A CH 2 PITCH RATE	WD10 VU				93017G	*

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

TABLE 4.229.7-2. OUTPUT FUNCTIONAL PARAMETERS FROM RENDEZVOUS RADAR SUBSYSTEM OPERATING PROGRAM
(G4.229)

FSSR NAME	M/S ID	NOMENCLATURE	DESTINATION	UNITS	DATA TYPE	P R E C	LAST CR	R E V
ANGRT_STATUS	V95J4302C	R RADAR ANGLE RATE STATUS WORD	REL NAV DISP		VAR			
ANG_STATUS	V95J4300C	R RADAR ANGLE STATUS WORD	REL NAV DISP		VAR			
Q_M50BODY_RR(1)	V95U4312C	M50 TO BODY R RADAR ELEM 1	ORB/RND NAV					
Q_M50BODY_RR(2)	V95U4313C	M50 TO BODY R RADAR ELEM 2	ORB/RND NAV					
Q_M50BODY_RR(3)	V95U4314C	M50 TO BODY R RADAR ELEM 3	ORB/RND NAV					
Q_M50BODY_RR(4)	V95U4315C	M50 TO BODY R RADAR ELEM 4	ORB/RND NAV					
RNGRT_STATUS	V95J4299C	R RADAR RANGE RATE STATUS WORD	REL NAV DISP		VAR			
RNG_STATUS	V95J4298C	R RADAR RANGE STATUS WORD	REL NAV DISP		VAR			
RR_ANG_DG	V95X4295X	R RADAR ANGLE DATA GOOD	ORB/RND NAV TLM					
RR_PITCHO	V95H4281C	R RADAR PITCH ANGLE	ORB/RND NAV REL NAV DISP	DEG	VAR		92297B	
RR_PRTEO	V95R4282C	R RADAR PITCH RATE	REL NAV DISP	MRAD/S	VAR		92297B	
RR_RANGEO	V95H4287C	R RADAR RANGE	ORB/RND NAV REL NAV DISP	FT	VAR		92297B	
RR_RNGRO	V95R4288C	R RADAR RANGE RATE	ORB/RND NAV REL NAV DISP	FT/S	VAR		92297B	
RR_RNGR_DG	V95X4294X	R RADAR RANGE RATE DATA GOOD	ORB/RND NAV TLM					
RR_RNG_DG	V95X4293X	R RADAR RANGE DATA GOOD	ORB/RND NAV TLM					
RR_ROLLO	V95H4284C	R RADAR ROLL ANGLE	ORB/RND NAV REL NAV DISP	DEG	VAR		92297B	
RR_RRTEO	V95R4285C	R RADAR ROLL RATE	REL NAV DISP	MRAD/S	VAR		92297B	
RR_SELF_TEST	V95X4309X	R RADAR SELF TEST ON FLAG	ORB/RND NAV				92297B	
RR_STAT_DISP	V95J4303C	R RADAR MODE STATUS WORD	REL NAV DISP		VAR			
RR_TIM	V95W4292C	R RADAR TIME	ORB/RND NAV	S			92297B	
RR_VALID	V95X4283X	R RADAR ERROR FLAG	GAX					

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

Rate Codes: (HDR Only) 1 = 1 sample/sec 2 = 5 samples/sec 3 = 12.5 samples/sec 4 = 25 samples/sec 5 = 100 samples/sec
(HDR and LDR) A = 1 sample/sec B = 5 samples/sec C = 12.5 samples/sec D = 25 samples/sec E = 100 samples/sec

TABLE 4.229.7-3. RENDEZVOUS RADAR SUBSYSTEM OPERATING PROGRAM (G4.229) I-LOADS

FSSR NAME	MSID	ENG UNITS	DT	PR	D	S	PR FCTN	MC	CAT	LAST CR	R E V
-----------	------	-----------	----	----	---	---	---------	----	-----	---------	-------------

NO REQUIREMENTS

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

TABLE 4.229.7-4. RENDEZVOUS RADAR SUBSYSTEM OPERATING PROGRAM (G4.229) K-LOADS

FSSR NAME DESCRIPTION	MSID	MC	KLOAD VALUE	ENG UNITS	DT	P R	S	PR FCTN	LAST CR	R E V
--------------------------	------	----	-------------	-----------	----	--------	---	---------	---------	-------------

NO REQUIREMENTS

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

TABLE 4.229.7-5. RENDEZVOUS RADAR SUBSYSTEM OPERATING PROGRAM (G4.229) CONSTANTS

FSSR NAME DESCRIPTION	MSID	MC	CONSTANT VALUE	ENG UNITS	DT	P R	S	PR FCTN	LAST CR	R E V
--------------------------	------	----	----------------	-----------	----	--------	---	---------	---------	-------------

NO REQUIREMENTS

NOTE: REFERENCE TABLE 1.3-1 FOR EXPLANATION OF DATA ELEMENTS.

This page intentionally left blank.

4.253 Deleted

This page intentionally left blank.

5.0 SUPPLEMENTARY MATERIAL

5.1 GLOSSARY

Terms and acronyms that occur in the nav aid SOP's in this volume are identified on the following pages.

GLOSSARY							
Acronym/Name	Definition Description	Application					
		GPS	TACAN	MSBLS	RA	NAVDAD	RR
AA	Air-to-air		X				
A/L	Approach and landing					X	
ACC_DRAG	Drag deceleration from UPP (ft/sec ²)					X	
ADS	Air data system					X	
ADTA	Air data transducer assembly	X				X	
AGC	Automatic gain control			X			
ALPHA_NAVDAD	Navigation-derived angle of attack (deg)					X	
ALT	Altitude above mean sea level (ft)					X	
ALT_B(I)	Altitude at base of Ith layer (ft)					X	
AMI	Alpha/mach indicator. Panel instrument giving equivalent airspeed, alpha, mach indicator, accelerometer scale					X	
ANG_STATUS	Rendezvous radar angle status word						X
ANGRT_STATUS	Rendezvous radar angle rate status word						X
ATT_PROC	Attitude processing						X
AUTO_CHANNEL_ID	TACAN channel selection		X				
AUTO_TRACK	Auto track mode						X
AUTOSPEC	Automatic specification					X	
AVVI	Altitude, vertical velocity indicator. Panel instrument giving baro/nav altitude, vertical velocity, acceleration, and radar altitude				X	X	
BCD	Binary coded data						
BITE	Built-in test equipment		X	X	X		X
C	Communication function						X
C-BAND	7-cm electromagnetic radiation region				X		
C_D	Drag coefficient					X	
CL	Centerline						
CLOCK	Clock-computer (GMT)		X	X	X		X
COAS	Crewman optical alignment sight						
COMMFAULT	Communication fault	X	X	X			X
COMLOCK	No comfault and locked on to signal		X	X			
COMM	Communication						X
COMPOOL	Common data pool						

GLOSSARY							
Acronym/Name	Definition Description	Application					
		GPS	TACAN	MSBLS	RA	NAVDAD	RR
COURSE_WRT_RW	Absolute course with respect to runway CL			X			
COURSE_WRT_RW_LIM	Limiting value of course with respect to runway CL			X			
CPDS	Computer program development specification	X					
CR	Communication and radar function						X
CR	Change request		X	X	X	X	X
D&C	Display and controls						
DA	Deployment assembly						X
DDPS	Digital data processing system		X	X	X		
DEU	Display electronics unit		X	X	X		
DPS	Data processing subsystem						
DYPRS_N	Subfunction to calculate dynamic pressure as a function of navigation data					X	
EA-1	Electronics assembly No. 1						X
EA-2	Electronics assembly No. 2						X
EAS_N	Navigation-derived equivalent airspeed (ft/sec)					X	
EQAS	Subfunction to calculate equivalent airspeed from navigation data					X	
FCOS	Flight control operating system	X					
FDIR	Fault detection, identification, and recovery		X	X	X		
FWD	Forward						X
FOM	Figure of merit	X					
G_OVR_T	Sea-level gravity divided by the universal gas constant ($^{\circ}$ K/ft)					X	
GAX	GN&C annunciator interface		X	X			X
GDOP	Geometric Dilution of Precision	X					
GPC_ACQ	GPC acquisition mode						X
GPC_DES	GPC designate mode						X
GPS	Global Positioning System	X					
GRTLS	Glide return to launch site						
GUID	Guidance						
H/W	Hardware						
HSI	Horizontal situation indicator		X	X			

GLOSSARY							
Acronym/Name	Definition Description	Application					
		GPS	TACAN	MSBLS	RA	NAVDAD	RR
I	The Ith layer in the atmospheric model					X	
ID	Identification		X				
IMU	Inertial measurement unit	X					
INITIALIZATION	Initialization mode						X
KCD()	Constant in the atmospheric model					X	
KU-BAND	2-cm microwave radiation region			X			X
L-BAND	28-cm electromagnetic radiation region		X				
LDG	Landing			X			
LM (I)	Lapse rate in the Ith layer (°K/ft)					X	
LOS	Line of sight						X
LRU	Line replaceable unit	X	X	X	X		X
LRU_ID	TACAN LRU number		X				
LSB	Least significant bit						
M	Current vehicle mass (slugs)					X	
M_N	Navigation-derived Mach number					X	
MACH_N	Subfunction to calculate Mach number from navigation data					X	
MANUAL	Manual mode						X
MAZM_CMLK	MLS azimuth commfault/lock-on flag			X			
MAZVOMLK	Azimuth comlock			X			
MDM	Multiplexer/demultiplexer	X	X	X	X		X
MELVCMLK	MLS elevation commfault/lock-on flag			X			
MELVOMLK	Elevation comlock			X			
MLS	Combining form for MSBLS			X			
MLS_CF	MLS comfault			X			
MLSNMI_FT	MLS conversion: nautical miles to feet (ft/nmi)			X			
MML	Master measurement list						
MRNG_CMLK	MLS range commfault/lock-on flag			X			
MRNGOMLK	Range comlock			X			
MSABZIN	MLS azimuth in			X			
MSB	Most significant bit						
MSB	Combining form for MSBLS			X			

GLOSSARY							
Acronym/Name	Definition Description	Application					
		GPS	TACAN	MSBLS	RA	NAVDAD	RR
MSB_DATA	MLS data processing			X			
MSB_IN	MLS input processing			X			
MSBAZ_LIM	Maximum allowed limit on absolute value of MLS azimuth			X			
MSBAZIMDG	MLS azimuth lock-on flag			X			
MSBAZIMIN	MLS azimuth word			X			
MSBAZIMO	MLS azimuth out			X			
MSBAZIMTIM	MLS azimuth measurement time (sec)			X			
MSBELEV	MLS elevation in			X			
MSBELEVDG	MLS elevation locked on			X			
MSBELEVIN	MLS elevation word			X			
MSBELEVO	MLS elevation out			X			
MSBELEVTIM	MLS elevation measurement time (sec)			X			
MSBELMAX_LIM	Upper limit on allowable elevation			X			
MSBELMIN_LIM	Lower limit on allowable elevation			X			
MSBELVSTB	MLS failure: elevation self-test bad			X			
MSBLS	Microwave scanning beam landing system			X			
MSBRANGDG	MLS range locked on			X			
MSBRANGE	MLS range in			X			
MSBRANGE0	MLS range out			X			
MSBRANGIN	MLS range word			X			
MSBRANGTIM	MLS range measurement time (sec)			X			
MSBRNG_LIM	Maximum allowed limit on MLS range			X			
MSID	Master stimulus identification	X	X	X	X	X	X
MTU	Master timing unit			X			
MX	Mission/LRU dependent constant		X	X	X	X	X
NAVDAD	Navigation-derived air data					X	
NAVDAD_AUTO	Selection flag for NAVDAD					X	
NEW_STATION_ID	TACAN 1, 2, or 3 new station channel number		X				
NMI	Nautical miles		X	X	X		X
OFT	Orbital flight test						
OWDE	One-way Doppler extractor						

GLOSSARY							
Acronym/Name	Definition Description	Application					
		GPS	TACAN	MSBLS	RA	NAVDAD	RR
PCN	Page change notification		X	X	X	X	X
PROC	Processor, processing						
Q_M50BODY	M50-to-body quaternion element						
Q_M50BODY_RR	M50-to-body RR element						X
QBAR_N	Navigation-derived dynamic pressure (lb/ft ²) (dynamic pressure is pressure from forward motion of the vehicle)						X
R	Radar function						X
R	Receive mode		X				
R/T	Receiver-transmitter		X		X		
RA	Radar altimeter	X			X		
RA_ALTIN	Radar altimeter parent word				X		
RA_ALTO	Radar altimeter altitude				X		
RA_DATA	Radar altimeter data processing				X		
RA_PWR_ON	Radar altimeter ready				X		
RA_TIME	Radar altimeter time				X		
RALT_CMLK	RA commfault/lock-on flag				X		
RALT_DG	Radar altimeter locked on				X		
RAPODD	Parity sum of radar altimeter parent word				X		
RCVR_LEFT	HSI receiver select		X	X			
RCVR_RIGHT	HSI receiver select		X	X			
RDR	Radar						X
REF_AREA	Reference area (ft ²) for vehicle model associated with C_D					X	
REL NAV	Relative navigation						X
REL_MOT	Relative motion						X
REL_VEL_MAG	Ground relative velocity magnitude (ft/sec)					X	
RENDZ	Rendezvous						X
RHO_B(I)	Air density at the base of the Ith layer (slugs/ft ³)					X	
RHO_N	Navigation-derived air density (slugs/ft ³)					X	
RHO_SEA_LEVEL	Sea-level air density (slugs/ft ³)					X	
RL_TIMER	Range lock-on timer		X				
RM	Redundancy management		X	X	X		

GLOSSARY							
Acronym/Name	Definition Description	Application					
		GPS	TACAN	MSBLS	RA	NAVDAD	RR
RND RDR	Rendezvous radar						X
RNG_STATUS	Rendezvous radar range status word						X
RNGRT_STATUS	Rendezvous radar range rate status word						X
RR	Rendezvous radar						X
RR_ANG_DG	Rendezvous radar angle data good						X
RR_ANGR_DG	Rendezvous radar angle rate data good						X
RR_COM_FLT	Rendezvous radar commfault						X
RR_PITCH_IN	Ku-band A channel 2 pitch WD4 WD4						X
RR_PITCHO	Rendezvous radar pitch angle out						X
RR_PRTE_IN	Ku-band A channel 2 pitch rate WD10						X
RR_PRTEO	Rendezvous radar pitch rate out						X
RR_RANGE_A	Ku-band A channel 2 range WD7 WD7						X
RR_RANGE_B	Ku-band A channel 2 range WD8 WD8						X
RR_RANGEO	Rendezvous radar range out						X
RR_RANGRO	Rendezvous radar range rate out						X
RR_RNG_DG	Rendezvous radar range rate data good						X
RR_RNGRTE_A	Ku-band A channel 2 range rate WD5						X
RR_RNGRTE_B	Ku-band A channel 2 range rate WD6						X
RR_ROLL_IN	Ku-band A channel 2 roll rate WD3 WD3						X
RR_ROLLO	Rendezvous radar roll angle out						X
RR_RRTE_IN	Ku-band A channel 2 roll rate WD9						X
RR_RRTEO	Rendezvous radar roll rate out						X
RR_SELF_TEST	Rendezvous radar self-test in progress						X
RR_STAT	Ku-band A channel 2 status word						X
RR_STAT_DISP	Radar mode status word						X
RR_VALID	Rendezvous radar data valid						X
RTLS	Return to launch site		X	X	X	X	

GLOSSARY							
Acronym/Name	Definition Description	Application					
		GPS	TACAN	MSBLS	RA	NAVDAD	RR
SAT_N	Navigation-derived static air temperature (°K)					X	
SELF_TEST	Self_test mode		X	X	X		X
SF	Selection filter		X	X	X		
SITE_LOOKUP	Site location of TACAN station from mini-table		X				
SM	System management						X
SOP	Subsystem operating program	X	X	X	X	X	X
SOURCE_LEFT	HSI source select		X	X			
SOURCE_RIGHT	HSI source select		X	X			
SPD_SND_CON	Speed-of-sound constant for air (ft/sec °K ^{-1/2})					X	
ST	Star tracker						
STS	Space Transportation System						
SW	Software		X	X	X	X	X
SX	Soft constant		X	X	X	X	X
T/R	Transmit/receive		X				
TAC_ANT	TACAN antenna selection		X				
TAC_BEARHI	TACAN bearing high limit (deg)		X				
TAC_BEARLO	TACAN bearing low limit (deg)		X				
TAC_BIT	TACAN built-in test		X				
TAC_BRGFL	TACAN bearing self-test fail		X				
TAC_CHAN	TACAN channel command		X				
TAC_CNTRL	TACAN control word		X				
TAC_DATA	TACAN data processing		X				
TAC_DW	MDM FFO() discrete in CD00 channel 5		X				
TAC_GPC	TACAN auto discrete		X				
TAC_RLTIM	TACAN range lock-on time limit		X				
TAC_RNGFL	TACAN range self-test fail		X				
TAC_RNGHI	TACAN range high limit (nmi)		X				
TAC_RNGLO	TACAN range low limit (nmi)		X				
TAC_STIP	Self-test in progress		X				
TAC_STTIM	TACAN self-test time (sec)		X				
TAC_SUPP_ACT	TACAN suppression active		X				
TAC_TEST	TACAN self-test processing		X				
TAC_TIM_TAG	TACAN time tag		X				

GLOSSARY							
Acronym/Name	Definition Description	Application					
		GPS	TACAN	MSBLS	RA	NAVDAD	RR
TACAN	Tactical air navigation		X				
TACANTINH	TACAN antenna switch inhibit		X				
TACANTSEL	TACAN channel selection		X				
TACBEARA	TACAN bearing word A		X				
TACBEARB	TACAN bearing word B		X				
TACBEARDG	TACAN bearing locked on		X				
TACBEARO	TACAN bearing out		X				
TACCFM	MDM FFO discrete in CD00 channel 6		X				
TACCHAN	TACAN channel with mode		X				
TACPOD	TACAN power discrete		X				
TACPON	TACAN buffered power status		X				
TACRANGA	TACAN range word A		X				
TACRANGB	TACAN range word B		X				
TACRANGDG	TACAN range locked on		X				
TACRANGEO	TACAN range out		X				
TAEM	Terminal area energy management				X		
TBRG_CMLK	TACAN bearing commfault/lock-on flag		X				
TCH_CHNG	TACAN channel change indicator		X				
TDRS	Tracking and data relay satellite						X
TLM	Telemetry		X	X	X	X	X
TMB(I)	Temperature at base of Ith layer (°K)					X	
TNMFT	TACAN nautical miles to feet conversion (ft/nmi)		X				
TRK_LAST_CYCLE	Track flag for rendezvous						X
TRNG_CMLK	TACAN range commfault/lock-on flag		X				
UPLINK	Data relayed from ground to orbiter						
UPP	User parameter processor	X		X		X	
V_NAVDAD	Switching point from drag model to $1/2 \rho v^2$					X	
VU	Vehicle utility (display)			X	X		
WGS84	World Geodetic Survey 1984	X					
WR_RTLS	Current vehicle mass (slugs) for entry guidance						
WRT	With respect to						

GLOSSARY							
Acronym/Name	Definition Description	Application					
		GPS	TACAN	MSBLS	RA	NAVDAD	RR
X	Hard constant		X	X	X	X	X
X/Y	TACAN ground beacon mode function: X or Y		X				
X	TACAN ground beacon: direct mode						
Y	TACAN ground beacon: reverse mode		X				

5.2 NONREQUIREMENT FLOW CHARTS/DIAGRAMS

This section contains nonrequirement flow charts and diagrams applicable to the nav aid SOP. Flow diagrams applicable to a specific functional module are included in that module section.

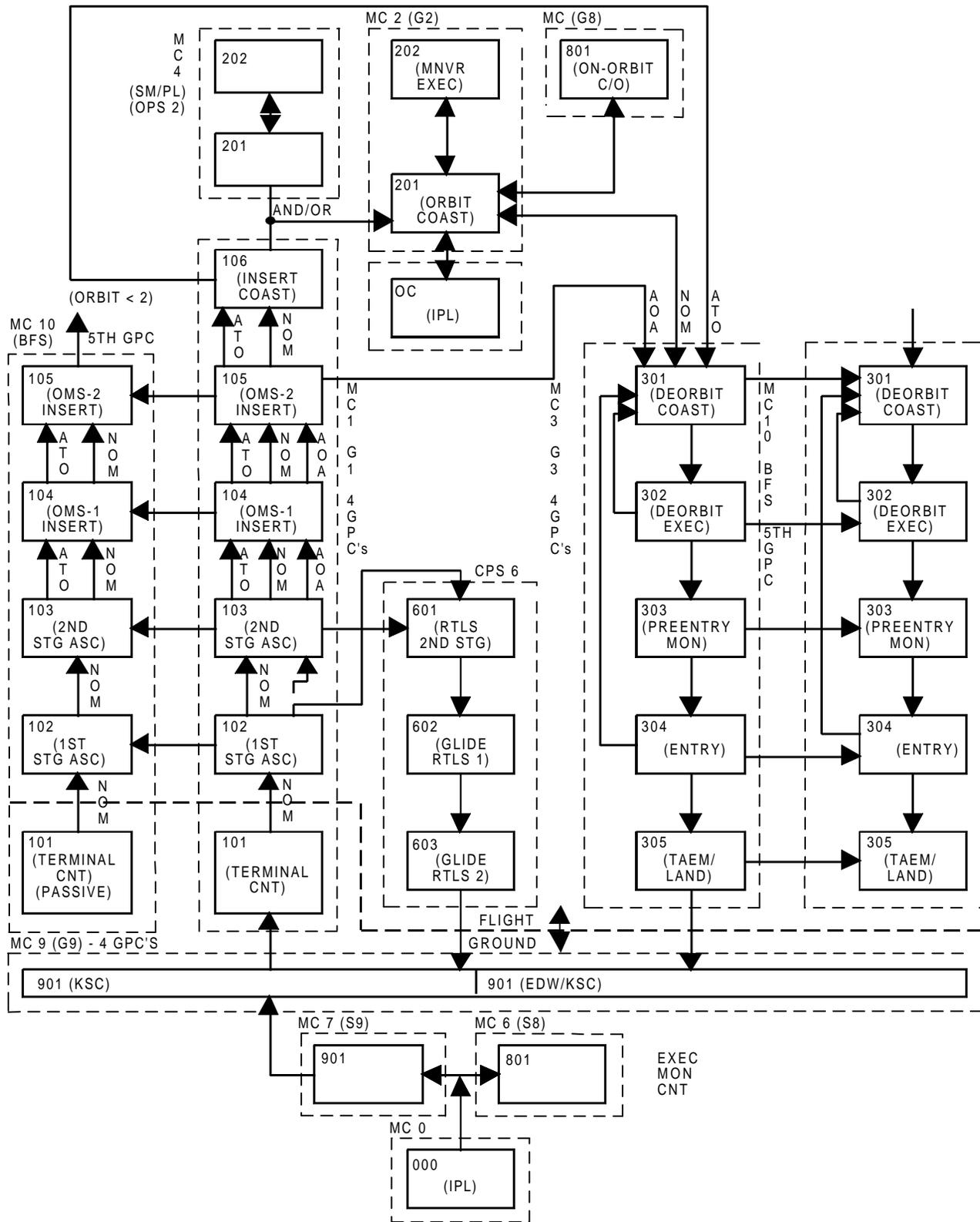


FIGURE 5.2-1. Major Modes and Transitions

APPENDIX R CHANGE AUTHORIZATION REFERENCES

R.1 CR REFERENCE SUMMARY

Table R.1-1. lists the paragraphs, tables (T), and figures (F) that have been changed as a result of approved change requests commencing with OI-26 up to the current OI. Changes incorporated prior to OI-26 are listed in the Historical Change Request Summary.

Table R.1-1. CR Reference Summary				
Paragraph	CR No.	OI	CR Title	
N/A	92103A	26B	FSSR Parameter Table Upgrade	
	92307	29	FSSR Format Change - Documentation Only	
3.3	91051K	26B	Streamlined Shuttle GPS Capability	
	92025B	26B	Onorbit Tacan Deletion	
4.0	92069B	26B	Single-String GPS Documentation Cleanup	
	92025B	26B	Onorbit Tacan Deletion	
4.0.1-1.(T)	92067D	27	Three String GPS Capability	
	92069B	26B	Single-String GPS Documentation Cleanup	
	92364D	29	Just-In-Time Flight Design (JITFD) I-Load Uplink Expansion	
	92529B	30	GPS Capability Enhancements for OI-30	
4.0.1-2.(T)	92069B	26B	Single-String GPS Documentation Cleanup	
4.17	91051K	26B	Streamlined Shuttle GPS Capability	
4.17.1	91051K	26B	Streamlined Shuttle GPS Capability	
4.17.1.1	91051K	26B	Streamlined Shuttle GPS Capability	
4.17.1.1-1.(F)	91051K	26B	Streamlined Shuttle GPS Capability	
4.17.1.1.1	91051K	26B	Streamlined Shuttle GPS Capability	
	92056D	26B	Remove Unnecessary GPS SOP Processing in G9	
	92067D	27	Three String GPS Capability	
	92144C	27	Three String GPS Document Cleanup	
	92277	27	GPS SOP Self-Test Bit SWAP	
	92289A	27	Post STS-91 OI-27 GPS Redesign	
	92369B	29	PASS GPS Operational Enhancements	
	92630A	30	I-Load Controls for GPS Residual Edit Logic	
	4.17.1.1.2	91051K	26B	Streamlined Shuttle GPS Capability
		92069B	26B	Single-String GPS Documentation Cleanup
4.17.1.1.2-1.(T)	91051K	26B	Streamlined Shuttle GPS Capability	
4.17.1.1.2-2.(T)	91051K	26B	Streamlined Shuttle GPS Capability	
	92035D	26B	GPS SOP Processing Relief	

Table R.1-1. CR Reference Summary			
Paragraph	CR No.	OI	CR Title
4.17.1.1.2-3.(T)	92049C	26B	Corrections to CR 91051K, Streamlined Shuttle GPS
	92067D	27	Three String GPS Capability
	91051K	26B	Streamlined Shuttle GPS Capability
4.17.1.1.2-4.(T)	92630A	30	I-Load Controls for GPS Residual Edit Logic
	91051K	26B	Streamlined Shuttle GPS Capability
4.17.1.1.2-5.(T)	92049C	26B	Corrections to CR 91051K, Streamlined Shuttle GPS
	92067D	27	Three String GPS Capability
	91051K	26B	Streamlined Shuttle GPS Capability
4.17.1.1.2-6.(T)	92067D	27	Three String GPS Capability
	91051K	26B	Streamlined Shuttle GPS Capability
4.17.1.1.2-7.(T)	92049C	26B	Corrections to CR 91051K, Streamlined Shuttle GPS
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
	91051K	26B	Streamlined Shuttle GPS Capability
	92049C	26B	Corrections to CR 91051K, Streamlined Shuttle GPS
4.17.1.1.3	92067D	27	Three String GPS Capability
	92529B	30	GPS Capability Enhancements for OI-30
4.17.1.2	91051K	26B	Streamlined Shuttle GPS Capability
	91051K	26B	Streamlined Shuttle GPS Capability
	92067D	27	Three String GPS Capability
	92364D	29	Just-In-Time Flight Design (JITFD) I-Load Uplink Expansion
4.17.1.3	92529B	30	GPS Capability Enhancements for OI-30
	92817C	32	GPS FSW Compensation for GPC Clock Drift
	91051K	26B	Streamlined Shuttle GPS Capability
4.17.1.3-1.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92067D	27	Three String GPS Capability
	92364D	29	Just-In-Time Flight Design (JITFD) I-Load Uplink Expansion
	92529B	30	GPS Capability Enhancements for OI-30
	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.1.4	91051K	26B	Streamlined Shuttle GPS Capability
4.17.1.5	91051K	26B	Streamlined Shuttle GPS Capability
4.17.2	91051K	26B	Streamlined Shuttle GPS Capability
4.17.3	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
4.17.4	91051K	26B	Streamlined Shuttle GPS Capability

Table R.1-1. CR Reference Summary				
Paragraph	CR No.	OI	CR Title	
4.17.5	92035D	26B	GPS SOP Processing Relief	
	92067D	27	Three String GPS Capability	
	92817C	32	GPS FSW Compensation for GPC Clock Drift	
	91051K	26B	Streamlined Shuttle GPS Capability	
	92067D	27	Three String GPS Capability	
4.17.6	92817C	32	GPS FSW Compensation for GPC Clock Drift	
	91051K	26B	Streamlined Shuttle GPS Capability	
	92035D	26B	GPS SOP Processing Relief	
	92049C	26B	Corrections to CR 91051K, Streamlined Shuttle GPS	
	92056D	26B	Remove Unnecessary GPS SOP Processing in G9	
	92069B	26B	Single-String GPS Documentation Cleanup	
	92067D	27	Three String GPS Capability	
	92144C	27	Three String GPS Document Cleanup	
	92289A	27	Post STS-91 OI-27 GPS Redesign	
	92364D	29	Just-In-Time Flight Design (JITFD) I-Load Uplink Expansion	
	92369B	29	PASS GPS Operational Enhancements	
	92529B	30	GPS Capability Enhancements for OI-30	
	92817C	32	GPS FSW Compensation for GPC Clock Drift	
	4.17.6-1.(F)	91051K	26B	Streamlined Shuttle GPS Capability
		92035D	26B	GPS SOP Processing Relief
92069B		26B	Single-String GPS Documentation Cleanup	
92067D		27	Three String GPS Capability	
92144C		27	Three String GPS Document Cleanup	
92289A		27	Post STS-91 OI-27 GPS Redesign	
92369B		29	PASS GPS Operational Enhancements	
4.17.6-1.(T)		91051K	26B	Streamlined Shuttle GPS Capability
		92069B	26B	Single-String GPS Documentation Cleanup
		92067D	27	Three String GPS Capability
	92144C	27	Three String GPS Document Cleanup	
	92289A	27	Post STS-91 OI-27 GPS Redesign	
	92369B	29	PASS GPS Operational Enhancements	
	4.17.6-1.(1)(F)	92392C	29	Doc-Only Corrections for OI-29 CRs 92329E, 92364D, and 92369B
		92817C	32	GPS FSW Compensation for GPC Clock Drift
		92817C	32	GPS FSW Compensation for GPC Clock Drift

Table R.1-1. CR Reference Summary			
Paragraph	CR No.	OI	CR Title
4.17.6-1.(3)(F)	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6-2.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
	92289A	27	Post STS-91 OI-27 GPS Redesign
	92369B	29	PASS GPS Operational Enhancements
	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.1	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
	92132C	27	Three String GPS GN&C Corrections
	92208C	28	GPS Issues for Correction on OI-28
	92281D	28	Post FACI OI-28 GPS Documentation Cleanup
	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.1.(F)	91051K	26B	Streamlined Shuttle GPS Capability
	92067D	27	Three String GPS Capability
	92208C	28	GPS Issues for Correction on OI-28
	92281D	28	Post FACI OI-28 GPS Documentation Cleanup
4.17.6.1-1.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.1-2.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
	92208C	28	GPS Issues for Correction on OI-28
	92281D	28	Post FACI OI-28 GPS Documentation Cleanup
4.17.6.1.1	91051K	26B	Streamlined Shuttle GPS Capability
	92067D	27	Three String GPS Capability
	92289A	27	Post STS-91 OI-27 GPS Redesign
	92369B	29	PASS GPS Operational Enhancements
4.17.6.1.1.(F)	91051K	26B	Streamlined Shuttle GPS Capability
	92035D	26B	GPS SOP Processing Relief
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
	92289A	27	Post STS-91 OI-27 GPS Redesign

Table R.1-1. CR Reference Summary			
Paragraph	CR No.	OI	CR Title
4.17.6.1.1-1.(T)	92369B	29	PASS GPS Operational Enhancements
	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
4.17.6.1.1-2.(T)	92817C	32	GPS FSW Compensation for GPC Clock Drift
	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
4.17.6.1.1-3.(T)	92369B	29	PASS GPS Operational Enhancements
	92817C	32	GPS FSW Compensation for GPC Clock Drift
	91051K	26B	Streamlined Shuttle GPS Capability
	92067D	27	Three String GPS Capability
4.17.6.1.2	92369B	29	PASS GPS Operational Enhancements
	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
	92132C	27	Three String GPS GN&C Corrections
	92289A	27	Post STS-91 OI-27 GPS Redesign
	92369B	29	PASS GPS Operational Enhancements
	92529B	30	GPS Capability Enhancements for OI-30
	92817C	32	GPS FSW Compensation for GPC Clock Drift
	4.17.6.1.2.(F)	91051K	26B
92067D		27	Three String GPS Capability
92289A		27	Post STS-91 OI-27 GPS Redesign
92369B		29	PASS GPS Operational Enhancements
92392C		29	Doc-Only Corrections for OI-29 CRs 92329E, 92364D, and 92369B
92430B		29	Protect Against GPS Receiver Non-Homogenous Data, DRs 109743/109744
4.17.6.1.2-1.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
	92364D	29	Just-In-Time Flight Design (JITFD) I-Load Uplink Expansion
	92529B	30	GPS Capability Enhancements for OI-30
	92817C	32	GPS FSW Compensation for GPC Clock Drift

Table R.1-1. CR Reference Summary			
Paragraph	CR No.	OI	CR Title
4.17.6.1.2-2.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92049C	26B	Corrections to CR 91051K, Streamlined Shuttle GPS
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
	92233	27	Clean-up of GPS_SEC_DL Data Types
	92369B	29	PASS GPS Operational Enhancements
	92389C	29	Spiral HAC in TAEM Area Navigation
	92430B	29	Protect Against GPS Receiver Non-Homogenous Data, DRs 109743/109744
	92529B	30	GPS Capability Enhancements for OI-30
	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.1.2-3.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92067D	27	Three String GPS Capability
4.17.6.1.2-4.(T)	91051K	26B	Streamlined Shuttle GPS Capability
4.17.6.1.2-5.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92067D	27	Three String GPS Capability
4.17.6.1.2-6.(T)	91051K	26B	Streamlined Shuttle GPS Capability
4.17.6.1.2.(1)(F)	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.1.2.1	91051K	26B	Streamlined Shuttle GPS Capability
	92281D	28	Post FACI OI-28 GPS Documentation Cleanup
4.17.6.1.2.1.(F)	91051K	26B	Streamlined Shuttle GPS Capability
	92281D	28	Post FACI OI-28 GPS Documentation Cleanup
4.17.6.1.2.1-1.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
	92281D	28	Post FACI OI-28 GPS Documentation Cleanup
	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.1.2.1-2.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
	92214A	27	Post-CI Three-String GPS Documentation Cleanup
	92281D	28	Post FACI OI-28 GPS Documentation Cleanup
	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.1.2.1-3.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup

Table R.1-1. CR Reference Summary			
Paragraph	CR No.	OI	CR Title
4.17.6.1.2.1-4.(T)	91051K	26B	Streamlined Shuttle GPS Capability
4.17.6.1.2.1-5.(T)	91051K	26B	Streamlined Shuttle GPS Capability
4.17.6.1.2.1-6.(T)	91051K	26B	Streamlined Shuttle GPS Capability
4.17.6.1.2.1-7.(T)	91051K	26B	Streamlined Shuttle GPS Capability
4.17.6.1.2.1-8.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92529B	30	GPS Capability Enhancements for OI-30
4.17.6.1.2.1-9.(T)	91051K	26B	Streamlined Shuttle GPS Capability
4.17.6.1.2.1-10.(T)	91051K	26B	Streamlined Shuttle GPS Capability
4.17.6.1.2.1-11.(T)	91051K	26B	Streamlined Shuttle GPS Capability
4.17.6.1.3	91051K	26B	Streamlined Shuttle GPS Capability
	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.1.3.(F)	91051K	26B	Streamlined Shuttle GPS Capability
4.17.6.1.3-1.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.1.3-2.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.1.3-3.(T)	91051K	26B	Streamlined Shuttle GPS Capability
4.17.6.1.4	91051K	26B	Streamlined Shuttle GPS Capability
	92067D	27	Three String GPS Capability
	92277	27	GPS SOP Self-Test Bit SWAP
	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.1.4.(F)	91051K	26B	Streamlined Shuttle GPS Capability
	92067D	27	Three String GPS Capability
	92277	27	GPS SOP Self-Test Bit SWAP
4.17.6.1.4-1.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
	92144C	27	Three String GPS Document Cleanup
	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.1.4-2.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup

Table R.1-1. CR Reference Summary			
Paragraph	CR No.	OI	CR Title
4.17.6.2	92067D	27	Three String GPS Capability
	92144C	27	Three String GPS Document Cleanup
	92817C	32	GPS FSW Compensation for GPC Clock Drift
	91051K	26B	Streamlined Shuttle GPS Capability
	92067D	27	Three String GPS Capability
	92369B	29	PASS GPS Operational Enhancements
	92430B	29	Protect Against GPS Receiver Non-Homogenous Data, DRs 109743/109744
4.17.6.2.(F)	92817C	32	GPS FSW Compensation for GPC Clock Drift
	91051K	26B	Streamlined Shuttle GPS Capability
	92369B	29	PASS GPS Operational Enhancements
	92430B	29	Protect Against GPS Receiver Non-Homogenous Data, DRs 109743/109744
4.17.6.2-1.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	90269B	26B	Streamlined Shuttle GPS Capability
	92430B	29	Protect Against GPS Receiver Non-Homogenous Data, DRs 109743/109744
4.17.6.2-2.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	90269B	26B	Streamlined Shuttle GPS Capability
	92067D	27	Three String GPS Capability
4.17.6.3	91051K	26B	Streamlined Shuttle GPS Capability
	92067D	27	Three String GPS Capability
4.17.6.3.(F)	92817C	32	GPS FSW Compensation for GPC Clock Drift
	91051K	26B	Streamlined Shuttle GPS Capability
	92067D	27	Three String GPS Capability
4.17.6.3-1.(T)	92817C	32	GPS FSW Compensation for GPC Clock Drift
	91051K	26B	Streamlined Shuttle GPS Capability
	92067D	27	Three String GPS Capability
	92364D	29	Just-In-Time Flight Design (JITFD) I-Load Uplink Expansion
4.17.6.3-2.(T)	92529B	30	GPS Capability Enhancements for OI-30
	92817C	32	GPS FSW Compensation for GPC Clock Drift
	91051K	26B	Streamlined Shuttle GPS Capability
4.17.6.4	91051K	26B	Streamlined Shuttle GPS Capability
	92035D	26B	GPS SOP Processing Relief
	92069B	26B	Single-String GPS Documentation Cleanup

Table R.1-1. CR Reference Summary

Paragraph	CR No.	OI	CR Title
4.17.6.4.(F)	92080B	26B	Single-String GPS UPP and SOP Corrections
	92067D	27	Three String GPS Capability
	92369B	29	PASS GPS Operational Enhancements
	92630A	30	I-Load Controls for GPS Residual Edit Logic
	92817C	32	GPS FSW Compensation for GPC Clock Drift
	91051K	26B	Streamlined Shuttle GPS Capability
4.17.6.4-1.(T)	92035D	26B	GPS SOP Processing Relief
	92080B	26B	Single-String GPS UPP and SOP Corrections
	92067D	27	Three String GPS Capability
	92369B	29	PASS GPS Operational Enhancements
	91051K	26B	Streamlined Shuttle GPS Capability
	92035D	26B	GPS SOP Processing Relief
4.17.6.4-2.(T)	92069B	26B	Single-String GPS Documentation Cleanup
	92080B	26B	Single-String GPS UPP and SOP Corrections
	92067D	27	Three String GPS Capability
	92329E	29	Onboard Calculation of RNP Matrix
	92369B	29	PASS GPS Operational Enhancements
	92392C	29	Doc-Only Corrections for OI-29 CRs 92329E, 92364D, and 92369B
	92817C	32	GPS FSW Compensation for GPC Clock Drift
	91051K	26B	Streamlined Shuttle GPS Capability
	92035D	26B	GPS SOP Processing Relief
	92069B	26B	Single-String GPS Documentation Cleanup
4.17.6.4.1	92080B	26B	Single-String GPS UPP and SOP Corrections
	92067D	27	Three String GPS Capability
	92369B	29	PASS GPS Operational Enhancements
	92817C	32	GPS FSW Compensation for GPC Clock Drift
	91051K	26B	Streamlined Shuttle GPS Capability
	92035D	26B	GPS SOP Processing Relief
	92056D	26B	Remove Unnecessary GPS SOP Processing in G9
	92080B	26B	Single-String GPS UPP and SOP Corrections
	92067D	27	Three String GPS Capability
	92144C	27	Three String GPS Document Cleanup
92214A	27	Post-CI Three-String GPS Documentation Cleanup	
92276A	27	GPS SOP Satellite De-Select Activation	

Table R.1-1. CR Reference Summary			
Paragraph	CR No.	OI	CR Title
4.17.6.4.1.(F)	92369B	29	PASS GPS Operational Enhancements
	92817C	32	GPS FSW Compensation for GPC Clock Drift
	91051K	26B	Streamlined Shuttle GPS Capability
	92035D	26B	GPS SOP Processing Relief
	92067D	27	Three String GPS Capability
4.17.6.4.1-1.(T)	92144C	27	Three String GPS Document Cleanup
	92369B	29	PASS GPS Operational Enhancements
	91051K	26B	Streamlined Shuttle GPS Capability
	92035D	26B	GPS SOP Processing Relief
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
	92144C	27	Three String GPS Document Cleanup
	92289A	27	Post STS-91 OI-27 GPS Redesign
	92364D	29	Just-In-Time Flight Design (JITFD) I-Load Uplink Expansion
	92369B	29	PASS GPS Operational Enhancements
4.17.6.4.1-2.(T)	92529B	30	GPS Capability Enhancements for OI-30
	92817C	32	GPS FSW Compensation for GPC Clock Drift
	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
4.17.6.4.1-3.(T)	92817C	32	GPS FSW Compensation for GPC Clock Drift
	91051K	26B	Streamlined Shuttle GPS Capability
	92035D	26B	GPS SOP Processing Relief
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
4.17.6.4.1.(3)(F)	92276A	27	GPS SOP Satellite De-Select Activation
4.17.6.4.1.(3)(F)	92369B	29	PASS GPS Operational Enhancements
4.17.6.4.1.(3)(F)	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.4.2	91051K	26B	Streamlined Shuttle GPS Capability
	92067D	27	Three String GPS Capability
	92630A	30	I-Load Controls for GPS Residual Edit Logic
	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.4.2.(F)	91051K	26B	Streamlined Shuttle GPS Capability
	92630A	30	I-Load Controls for GPS Residual Edit Logic
4.17.6.4.2-1.(T)	91051K	26B	Streamlined Shuttle GPS Capability

Table R.1-1. CR Reference Summary

Paragraph	CR No.	OI	CR Title
4.17.6.4.2-2.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.4.2-3.(T)	92630A	30	I-Load Controls for GPS Residual Edit Logic
4.17.6.4.2-4.(T)	92630A	30	I-Load Controls for GPS Residual Edit Logic
4.17.6.4.3	91051K	26B	Streamlined Shuttle GPS Capability
	92067D	27	Three String GPS Capability
	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.4.3.(F)	91051K	26B	Streamlined Shuttle GPS Capability
4.17.6.4.3-1.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92067D	27	Three String GPS Capability
4.17.6.4.3-2.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.5	91051K	26B	Streamlined Shuttle GPS Capability
	92056D	26B	Remove Unnecessary GPS SOP Processing in G9
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
	92144C	27	Three String GPS Document Cleanup
	92208C	28	GPS Issues for Correction on OI-28
4.17.6.5.(F)	91051K	26B	Streamlined Shuttle GPS Capability
	92056D	26B	Remove Unnecessary GPS SOP Processing in G9
	92067D	27	Three String GPS Capability
	92208C	28	GPS Issues for Correction on OI-28
4.17.6.5-1.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
	92208C	28	GPS Issues for Correction on OI-28
	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.5-2.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
	92817C	32	GPS FSW Compensation for GPC Clock Drift

Table R.1-1. CR Reference Summary			
Paragraph	CR No.	OI	CR Title
4.17.6.5-3.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92067D	27	Three String GPS Capability
4.17.6.6	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
	92369B	29	PASS GPS Operational Enhancements
	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.6.(F)	91051K	26B	Streamlined Shuttle GPS Capability
	92067D	27	Three String GPS Capability
	92369B	29	PASS GPS Operational Enhancements
	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.6-1.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
	92369B	29	PASS GPS Operational Enhancements
	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.6-2.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92067D	27	Three String GPS Capability
	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.6-3.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
	92132C	27	Three String GPS GN&C Corrections
	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.6-4.(T)	92067D	27	Three String GPS Capability
	92144C	27	Three String GPS Document Cleanup
	92369B	29	PASS GPS Operational Enhancements
4.17.6.7	91051K	26B	Streamlined Shuttle GPS Capability
	92056D	26B	Remove Unnecessary GPS SOP Processing in G9
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
	92132C	27	Three String GPS GN&C Corrections
4.17.6.7.(F)	92369B	29	PASS GPS Operational Enhancements
	91051K	26B	Streamlined Shuttle GPS Capability
	92214A	27	Post-CI Three-String GPS Documentation Cleanup

Table R.1-1. CR Reference Summary

Paragraph	CR No.	OI	CR Title
4.17.6.7-1.(T)	92369B	29	PASS GPS Operational Enhancements
	92369B	29	PASS GPS Operational Enhancements
	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
4.17.6.7-2.(T)	92369B	29	PASS GPS Operational Enhancements
	92817C	32	GPS FSW Compensation for GPC Clock Drift
	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
	92067D	27	Three String GPS Capability
4.17.6.8	92817C	32	GPS FSW Compensation for GPC Clock Drift
	91051K	26B	Streamlined Shuttle GPS Capability
	92035D	26B	GPS SOP Processing Relief
	92067D	27	Three String GPS Capability
	92132C	27	Three String GPS GN&C Corrections
	92364D	29	Just-In-Time Flight Design (JITFD) I-Load Uplink Expansion
	92392C	29	Doc-Only Corrections for OI-29 CRs 92329E, 92364D, and 92369B
	92529B	30	GPS Capability Enhancements for OI-30
4.17.6.8.(F)	92067D	27	Three String GPS Capability
	92364D	29	Just-In-Time Flight Design (JITFD) I-Load Uplink Expansion
	92529B	30	GPS Capability Enhancements for OI-30
4.17.6.8-0.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92035D	26B	GPS SOP Processing Relief
4.17.6.8-1.(T)	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
	92105B	26B	GPS MSID Documentation Corrections
	92067D	27	Three String GPS Capability
	92364D	29	Just-In-Time Flight Design (JITFD) I-Load Uplink Expansion
4.17.6.8-2.(T)	92529B	30	GPS Capability Enhancements for OI-30
	91051K	26B	Streamlined Shuttle GPS Capability
	92069B	26B	Single-String GPS Documentation Cleanup
	92105B	26B	GPS MSID Documentation Corrections
	92067D	27	Three String GPS Capability
	92364D	29	Just-In-Time Flight Design (JITFD) I-Load Uplink Expansion

Table R.1-1. CR Reference Summary				
Paragraph	CR No.	OI	CR Title	
4.17.6.8-3.(T)	92529B	30	GPS Capability Enhancements for OI-30	
	91051K	26B	Streamlined Shuttle GPS Capability	
	92069B	26B	Single-String GPS Documentation Cleanup	
	92080B	26B	Single-String GPS UPP and SOP Corrections	
	92098D	26B	Single-String GPS RCVR STATE PROC/Status Display Corrections	
4.17.6.8-4.(T)	92102B	26B	GPS I-Load Category Update	
	91051K	26B	Streamlined Shuttle GPS Capability	
4.17.6.8-5.(T)	91051K	26B	Streamlined Shuttle GPS Capability	
4.17.6.9	92067D	27	Three String GPS Capability	
	92817C	32	GPS FSW Compensation for GPC Clock Drift	
4.17.6.9(F)	92817C	32	GPS FSW Compensation for GPC Clock Drift	
4.17.6.9-0.(T)	92067D	27	Three String GPS Capability	
	92369B	29	PASS GPS Operational Enhancements	
4.17.6.9-1.(T)	92067D	27	Three String GPS Capability	
	92144C	27	Three String GPS Documentation Cleanup	
	92289A	27	Post STS-91 OI-27 GPS Redesign	
	92329E	29	Onboard Calculation of RNP Matrix	
	92353D	29	Automated Alpha-Beta Management During Mated Coast & MM104	
	92364D	29	Just-In-Time Flight Design (JITFD) I-Load Uplink Expansion	
	92369B	29	PASS GPS Operational Enhancements	
	92389C	29	Spiral HAC in TAEM Area Navigation	
	92392C	29	Doc-Only Corrections for OI-29 CRs 92329E, 92364D, and 92369B	
	92450C	29	Documentation Clean-up for CR 92389	
	92332D	30	Add Theta Limits Bracket to MEDS ADI Displays	
	92522D	30	Overhaul MEDS Composite Flight Instrument Display	
	92571B	30	Drive HSD MLS, TAC, DRAG and ADTA A/I/F Asterisks at Item Entry	
	4.17.6.9-2.(T)	92817C	32	GPS FSW Compensation for GPC Clock Drift
		92067D	27	Three String GPS Capability
92132C		27	Three String GPS GN&C Corrections	
92214A		27	Post-CI Three-String GPS Documentation Cleanup	
92233		27	Clean-up of GPS_SEC_DL Data Types	
92276A		27	GPS SOP Satellite De-Select Activation	

Table R.1-1. CR Reference Summary			
Paragraph	CR No.	OI	CR Title
4.17.6.9-3.(T)	92261C	28	Add PPS Erase KEY MSID's To Three String GPS Downlist
	92289A	27	Post STS-91 OI-27 GPS Redesign
	92281D	28	Post FACI OI-28 GPS Documentation Cleanup
	92369B	29	PASS GPS Operational Enhancements
	92389C	29	Spiral HAC in TAEM Area Navigation
	92529B	30	GPS Capability Enhancements for OI-30
	92817C	32	GPS FSW Compensation for GPC Clock Drift
	92067D	27	Three String GPS Capability
	92139	27	GPS I-Load Categorization Updates
	92292B	28	GPS Related Category Modifications for OI-28
	92329E	29	Onboard Calculation of RNP Matrix
	92369B	29	PASS GPS Operational Enhancements
4.17.6.9-4.(T)	92430B	29	Protect Against GPS Receiver Non-Homogenous Data, DRs 109743/109744
	92630A	30	I-Load Controls for GPS Residual Edit Logic
4.17.6.10	92067D	27	Three String GPS Capability
	92292B	28	GPS Related Category Modifications for OI-28
4.17.6.10-0.(T)	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.10-1.(T)	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.10-2.(T)	93090E	33	Bearing Displays
	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.10-3.(T)	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.10-4.(T)	92817C	32	GPS FSW Compensation for GPC Clock Drift
4.17.6.10-5.(T)	92817C	32	GPS FSW Compensation for GPC Clock Drift
	93090E	33	Bearing Displays
4.43.6.5-1.(T)	92025B	26B	Onorbit Tacan Deletion
	92067D	27	Three String GPS Capability
	92297B	27	Delete Erroneous TLM Destinations for NAV
	92522D	30	Overhaul MEDS Composite Flight Instrument Display
	93090E	33	Bearing Displays
4.43.6.5-2.(T)	92157	28	IDD Clarification to Remove Blanks from FSSR Names
4.43.6.5-3.(T)	92025B	26B	Onorbit Tacan Deletion
4.44.1.2-1.(F)	92289A	27	Post STS-91 OI-27 GPS Redesign
4.44.4-1.(T)	92297B	27	Delete Erroneous TLM Destinations for NAV

Table R.1-1. CR Reference Summary			
Paragraph	CR No.	OI	CR Title
4.44.6.3-1.(T)	92025B	26B	Onorbit Tacan Deletion
	92067D	27	Three String GPS Capability
	92389C	29	Spiral HAC in TAEM Area Navigation
	92450C	29	Documentation Clean-up for CR 92389
	92522D	30	Overhaul MEDS Composite Flight Instrument Display
	93090E	33	Bearing Displays
4.44.6.3-2. (T)	92297B	27	Delete Erroneous TLM Destinations for NAV
4.45.6.2-1.(T)	92025B	26B	Onorbit Tacan Deletion
	92067D	27	Three String GPS Capability
	92522D	30	Overhaul MEDS Composite Flight Instrument Display
	93090E	33	Bearing Displays
4.45.6.2-2.(T)	91051K	26B	Streamlined Shuttle GPS Capability
4.122.2.4-1.(T)	92067D	27	Three String GPS Capability
	92144C	27	Three String GPS Document Cleanup
	92389C	29	Spiral HAC in TAEM Area Navigation
	92450C	29	Documentation Clean-up for CR 92389
	93090E	33	Bearing Displays
4.229.7-1.(T)	92025B	26B	Onorbit Tacan Deletion
	92067D	27	Three String GPS Capability
	92522D	30	Overhaul MEDS Composite Flight Instrument Display
	93090E	33	Bearing Displays
4.229.7-2. (T)	92297B	27	Delete Erroneous TLM Destinations for NAV
4.253	92025B	26B	Onorbit Tacan Deletion
4.253.1	92025B	26B	Onorbit Tacan Deletion
4.253.1.1	92025B	26B	Onorbit Tacan Deletion
4.253.1.1.1	92025B	26B	Onorbit Tacan Deletion
4.253.1.1-1.(F)	92025B	26B	Onorbit Tacan Deletion
4.253.1.1.2	92025B	26B	Onorbit Tacan Deletion
4.253.1.1.2-1.(F)	92025B	26B	Onorbit Tacan Deletion
4.253.1.1.2-2.(F)	92025B	26B	Onorbit Tacan Deletion
4.253.1.1.2-3.(F)	92025B	26B	Onorbit Tacan Deletion
4.253.1.1.2-4.(F)	92025B	26B	Onorbit Tacan Deletion
4.253.1.1.2-5.(F)	92025B	26B	Onorbit Tacan Deletion
4.253.1.1.2-6.(F)	92025B	26B	Onorbit Tacan Deletion
4.253.1.1.3	92025B	26B	Onorbit Tacan Deletion

Table R.1-1. CR Reference Summary			
Paragraph	CR No.	OI	CR Title
4.253.1.2	92025B	26B	Onorbit Tacan Deletion
4.253.1.3	92025B	26B	Onorbit Tacan Deletion
4.253.1.3-1.(T)	92025B	26B	Onorbit Tacan Deletion
4.253.1.4	92025B	26B	Onorbit Tacan Deletion
4.253.1.5	92025B	26B	Onorbit Tacan Deletion
4.253.2	92025B	26B	Onorbit Tacan Deletion
4.253.3	92025B	26B	Onorbit Tacan Deletion
4.253.4	92025B	26B	Onorbit Tacan Deletion
4.253.4-1.(1)	92025B	26B	Onorbit Tacan Deletion
4.253.4-1.(2)	92025B	26B	Onorbit Tacan Deletion
4.253.5	92025B	26B	Onorbit Tacan Deletion
4.253.5-1	92025B	26B	Onorbit Tacan Deletion
4.253.6	92025B	26B	Onorbit Tacan Deletion
4.253.6-1.(F)	92025B	26B	Onorbit Tacan Deletion
4.253.6.1	92025B	26B	Onorbit Tacan Deletion
4.253.6.2	92025B	26B	Onorbit Tacan Deletion
4.253.6.2-1.(F)	92025B	26B	Onorbit Tacan Deletion
4.253.6.3	92025B	26B	Onorbit Tacan Deletion
4.253.6.3-1.(F)	92025B	26B	Onorbit Tacan Deletion
4.253.6.4	92025B	26B	Onorbit Tacan Deletion
4.253.6.4-1.(F)	92025B	26B	Onorbit Tacan Deletion
4.253.7	92025B	26B	Onorbit Tacan Deletion
4.253.7-1.(T)	92025B	26B	Onorbit Tacan Deletion
4.253.7-2.(T)	92025B	26B	Onorbit Tacan Deletion
4.253.7-3.(T)	92025B	26B	Onorbit Tacan Deletion
4.253.7-4.(T)	92025B	26B	Onorbit Tacan Deletion
4.253.7-5.(T)	92025B	26B	Onorbit Tacan Deletion
5.1	92069B	26B	Single-String GPS Documentation Cleanup

R.2 HISTORICAL CHANGE REQUEST SUMMARY

Table R.2-1. is an historical record of approved change requests incorporated since the initial baseline up to OI-26. Change requests commencing with OI-26 are reformatted and listed in the CR Reference Summary.

Table R.2-1. Historical Change Request Summary		
CR No.	OI	CR Title
2066		MDM Constraint on Tacan S/W Entry I/O
2166		LVL C SDC NAV DATA/GAX/MSC
2235A		NAV DATA/GAX/MSC
12019		Standardization of FSSR PRIN FUNC
12054B		Horiz Sit Disp Changes
12112		GAX-Annunciation Rewrite
2391		OWDE BCE Element Bypass Require
2602A		Radar Altimeter SOP Chgs
12177B		Dedicated Display Rate Reduction
12336		I-Load OFT-1 Baseline
12337		I-Load OFT-1 Baseline
12338		I-Load OFT-1 Baseline
12339		I-Load OFT-1 Baseline
12109		On ORB 1, I/O Update
12148		MSBLS SOP Changes
12266A		Tacan SOP & RM
12374		RR SOP I/F (1DD Only)
12316		GN&C Switch RM PROC Rate
12443		I-Load for Entry FACI
12403A		MSBLS Range Elev & Azimuth Limits
12419A		RR SOP Corrections
12442A		Integrated ENT NAV, ASC NAV & ORB NAV
12471A		TAEM GUID NAVDAD/DEFAULT
12698A		Tacan SOP/Site Look Up Interface
12749A		RR SOP Changes
12522		Route "Current Vehicle Mass" to NAVDAD SOP
12812A		Relative NAV Display
12918		Correct Command Seq to Self Test
19022		MODS to MPAD I-Load
12848H		Optional TAEM Targeting
12868A		I-Load Change to Reflect Mode TAEM Actions

Table R.2-1. Historical Change Request Summary		
CR No.	OI	CR Title
12948		Addition of Automatic Maneuver
19159		MSBLS SOP Limits Revised Specification
19222		ET SEP Switch Default & Overrid
19059		Tacan SOP & RM
19391		Change to I-Load Values
19100		FSSR Update
19335		Delete Tacan "AUTO" Discrete Test
19455		Measurement Attribute Correction
19361A		Rewrite On-ORB NAV FSSR
19300		FSSR Update II
19480		RA DATA Flow: Changes to RA SOP
19434A		Changes to RNDZ Radar
19611A		Provide Time Homogeneous Data to NAVDAD
19690A		Changes to I-Load Values for NAVDAD
19845A		Tacan Commonfault Addition
19900		FSSR Update 5
29107		Tacan Description Error
29123		Change Value of V_NAVDAD
29279		Correction to MSBLS NMI to Feet Conversion Rate (STS-2)
29149B		Add Program Interrupt Handling to LVLA
29349		Units Correction for NAV Related (REF)
29410A		Change RENDZ Radar STAT WD 1
29471B		HFE Overrun Definition
29478		RTLS NAVDAD Drag Coefficients (REF Only)
29500		FSSR Update 7
29611A		MOD to RENDZ Radar SOP
29660A		Current Vehicle MASS Update
29800		FSSR Update 8
29999A		OTT Rewrite (IDD)
39000		FSSR Update 9
39192A		Cycle 3.3.3 I-Load
39224A		Tacan CHAN Change Flag Reset
39264E		HUD Autoland Monitor Format
39400		FSSR Update 10 (R18)
39595		CR Implementation Baseline, August 1981
39640		FSSR Update 11 (R18), 2(R19)
39655		Tacan Y-Mode Selection Fix (STS-3 REF)

Table R.2-1. Historical Change Request Summary		
CR No.	OI	CR Title
39783		MSBLS SOP Low Elev Cut Off Lim (STS-4 REF)
39980		CR Implementation Baseline- Feb, 1982
39988F		STS-5 CY 3 DES/ENT I/L (STS-5; REF)
59220		CR Implementation Baseline- July, 1982
59375A		CR implementation Baseline - Sept, 1982
59408D		STS-8 CYC 1 NAV/ENT I-Load (IL REF Only)
59455G		Tacan Orbiter NAV-Data ACQ & D/L
59737A		RR Track Loss Annunciation
59951		Design Dependent I-Loads
59972		I-Load Scrub: Atmosphere
69056		STS-13 CYC 1 NAV I-Loads (REF Only)
69070		FSSR Update 3 (OI2, 3, and 4)
69119C		I-Load Scrub: Tacan Scale Factor
69125B		CR Implementation Baseline - November 1983
69154A		STS-14 Cycle 1 NAV I-Loads (REF Only)
69402C		Create A Category For GRTLS TAEM Mission - Unique I-Loads (REF Only)
69427		I-Load Units/Configuration Change - IMRB (22) (REF Only)
69550B		CR Implementation Baseline - March/April, 1984
69700		FSSR Update Baseline 4
69893		Cleanup of CR 39264E
79217A		Entry Tacan SOP Interface Documentation Cleanup
79425		CR Implementation Baseline - April, 1985
79442C		Inadvertent Mode Transitions
79470B		MSBLS SOP DOC I/F Cleanup
79499		FSSR Update No. 5
79790		CR Implementation Baseline - Nov, 1985
79991A		Errata STS 83-0014A V1
89137C		Delete HUD 2 (Autoland) FMT
89250B		Bailout Control System
89372B		Errata CR for FSSR 10
89491C		I-Load Owner Change
89497		STS 83-0014B, Vol. 1, B/L
89599C		Cleanup Level B GN&C Document
89662B		FSSR STS 83-0003B Errata Update
89672A		STS 83-0014B, Vol 1, PCN-1, Errata
89740B		FSSR-7, OI-8B, PCN-1, Errata

Table R.2-1. Historical Change Request Summary

CR No.	OI	CR Title
89845B		MSBLS Input Processing Time TAG Cleanup
90044A		STS 83-0014B, Vol 1, PCN-2, Errata
90182	21	SCM Scrub Completion
90185A		Add CR Field to FSSR I/O Tables
90278D	21	MSBLS Data Downlist Cleanup
90374C		Automated FSSR Constrants
90380C	23	Downlist Update for New Tacan
90587C	23	SOP Updates for New Collins Tacan
90606E	23	PFS/BFS ICD Baseline
90690A	24	Deletion of EQTN MSID Column from K-Load FSSR Tables
90705H	25	3 Engines Out Auto Contingency
90790	24	Add Memory Configuration to the FSSR I-Load Tables
90814C	24	Add Description of Database Generated FSSR Parameter Tables
90881A	25	FSSR STS 83-0001 Documentation Changes
90891	23	FSSR Doc Only Changes, NAV AIDS SOP
90964	25	Interface Clarification for 3 Engines Out – CR 90705H
90966A	24	Deletion of the Qualifier from the FSSR I-Load Tables
91066D	25	Standardize the FSSR Correlation Tables.
92027B	25	TDMS-2 Required FSSR Changes

This page intentionally left blank.