

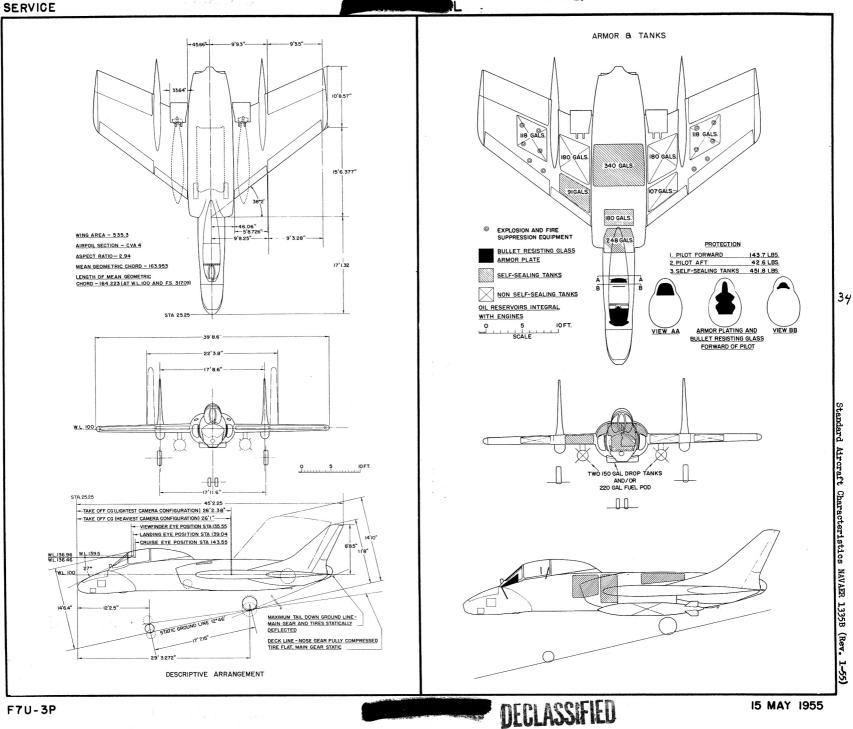
STANDARD AIRCRAFT CHARACTERISTICS

F7U-3P "CUTLASS"

CHANCE VOUGHT

640601-7585

DECLASSIFIED



15 MAY 1955

POWER PLANT

NO. & MODEL. (2) J46—WE-8B
MFR. Westinghouse
AFTERBURNER. Integral
TYPE. 12-Stage Axial Compr.
2 Stage Turbine
LENGTH. 200"
DIAMETER. 29"

RATINGS

LES @ RPM @ ALT.

MAXIMUM 5,725 10,100 S.S.L.

MILITARY 4,020 10,100 S.S.L.

NORMAL 3,620 10,100 S.S.L.

SPEC. NO. WAGT-24C10-14.

EQUIPMENT

No.	Description	Location
1	Camera, K-37	Fuselage
1	Camera, P-2	Fuselage
2	Photoflash Cartridge Ejectors, B-4	Fuselage
40	Photoflash Cartridges M=123	Fuselage

MISSION AND DESCRIPTION

The F7U-3P is a carrier-based, single-place photographic airplane. Its primary mission is reconnaissance.

Longitudinal and lateral control is provided by ailavators which combine the functions of elevators and ailerons. Full-span retractable slats in the leading edge of the wing are used in take-off and landing. Split-flap type speed brakes are located on the center section trailing edge of the wing. The main gear and nose gear retract into the fin stubs and fuselage, respectively. The cockpit is pressurized and is equipped with a pilot ejection seat.

Each of two wing center section pylons has provisions for carrying a 150 gallon drop tank. A 220 gallon fuel pod may be carried on the fuselage centerline.

DEVELOPMENT

First flight......September 1954

WEIGHTS

LOADINGS	LBS	L.F.
EMPTY		
BASIC		
DESIGN COMBAT		
MAX. LANDING		
(Arrest).	23,500	

All weights are estimated.

* Maximum Anticipated Weight.

FUEL AND OIL

No. Tanks	Tot. Gal.	Location
3	768	Fuselage, S.S.
1	91	Wing, S.S.
9	703	Wing, Bladder
2	300	Wing, Drop
1	220	Fuselage, Pod
FU	EL GRADE	JF-4
FU	EL SPEC. MI	L-F-5624

OIL

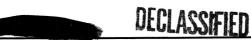
CAPACITY (Gals.)............6.5 SFEC.....MIL-L-7808

DIMENSIONS

WING	
AREA	
MAC	ġ۱۱
SWEEPBACK (chord)	
HEIGHT	
TREAD18' -	J"

ELECTRONICS

JHF COMMAN/ARC-27	Α
RADIO ALTIMETERAN/APN-2	2
NAVIGATION RADIOAN/ARR-2	A
RADIO COMFASSAN/ARN-	
IFFAN/APX-6	
CODER GROUPAN/APA-8	9
AN/ARN-21 to replace ARN-6 when	



PERFORMANCE SUMMARY							
TAKE-OFF LOADING CONDITION		(1) BASIC AIRFLANE		(4) BASIC AIRPLANE + TWO 150 GAL DAS TANKS (D)			AL FUEL TWO 150 S TANKS (D)
TAKE-OFF WEIGHT	lb.	29,828		32,334			34,064
Fuel	1b.	10,153	,	12,103			13,533
Fayload	1b.	608	77	608			608
Wing loading	lb./sq.ft.	55.8		60.4			63.7
Stall speeu - power-off	kn.	115		119			123
Take-off run at S.L calm	(C) ft.	2 , 750	,	3,200			3,600
Take-off run at S.L. 25 kn. wind	(C) ft.	1,850		2,200			2,450
Take-off to clear 50 ft calm	(c) ft.	3,800		4,450			4,950
Max. speed/altitude	(A) kn./ft.	470/17,000		418/13,000		4	03/10,000
Rate of climb at S.L.	(B) fpm.	3,520		2,850			2,550
Time: S.L. to 20,000 ft.	(B) min.	7.5		10.4			11.8
Time: S.L. to 30,000 ft.	(B) min.	15.2	}	22.4			26.7
Service ceiling (100 fpm)	(B) ft.	34,600		29,200			26,500
Combat range	n.mi.	775		920			990
Average cruising speed	kn.	450		450			435
Cruising altitude(s)	ft.	30,300-36,500		24,700-35,500		22,0	00-34,000
Combat radius	n.mi.	200		280			330
Average cruising speed	kn.	450		450			450
COMBAT LOADING CONDITION		(2) COMBAT	(3) COMBAT	(5) COMBAT	(6) COMBAT	(8) COMBAT	(9)COMBAT
COMBAT WEIGHT	lb.	25,767	25,767	27,239	27,239	28,397	28,397
Engine power		Military	Combat	Military	Combat	Military	Combat
Fuel	lb.	6,092	6,092	7,262	7,262	8,120	8,120
Combat speed/combat altitude	kn./ft.	490/32,000	530/32,000	488/30,000		481/28000	534/28000
Rate of climb/combat altitude	fpm/ft.	800/32,000	4,400/32,000	750/30,000	4,400/30,000	800/28000	4450/28000
Combat ceiling (500 fpm)	ft.	34,700	45,800	32,500	44,600	31,000	43,500
Rate of climb at S.L.	fpm.	4,250	10,700	3,900	9,900	3,670	9,300
Max. speed at S.L.	kn.	504	589	498	585	486	579
Max. speed/altitude	kn./ft.	515/15,000	589/S.L.	511/15,000	585/S.L.	498/15000	579/S.L.
LANDING WEIGHT	lb.	22,563		22,992			23,409
Fuel	_lb.	2,888		3,015			3,132
Stall speed - power-off	kn.	104		105			106
Stall speed - with approach power	kn.	99		100			101

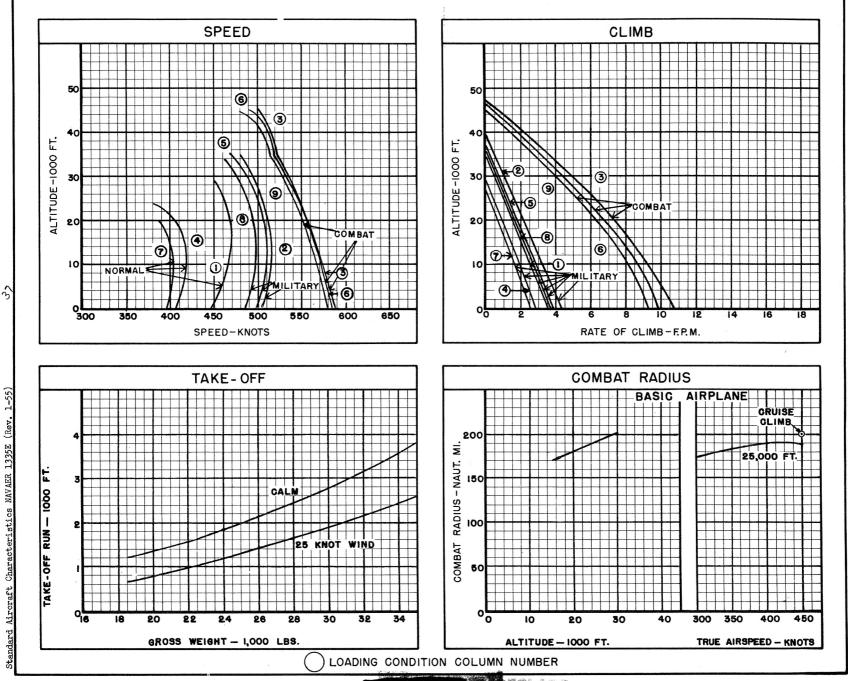
NOTES

- (A) Normal Rated Thrust
- (B) Military Rated Thrust
- (C) Combat Rated Thrust
- (D) Tanks dropped when empty

Performance based on Contractor's flight test

Range and radius are based on engine specification fuel consumption increased by 5%







RADIUS PROBLEM

WARM-UP, TAXI, TAKE-OFF: 5 minutes at normal thrust plus 1 minute at combat thrust CLIMB: To cruise ceiling at military thrust (cruising ceiling = altitude for 300 ft./min. rate of climb at normal thrust).

CRUISE-OUT: At V for long range at cruising ceiling

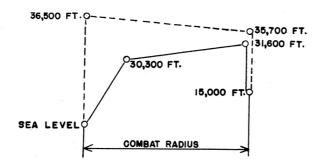
DESCEND: To 15,000 ft. (no fuel used, no distance gained).

CRUISE: At 15,000 ft. for 20 minutes at military thrust (no distance gained).

CLIMB: To cruise ceiling at military thrust.

CRUISE-BACK: At V for long range at cruise ceiling.

RESERVE: 5% of initial fuel plus fuel required for 2 engines for 20 minutes at V for maximum endurance at sea level.



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tandard Aircraft Characteristics NAVAER 1335F (Rev.

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