

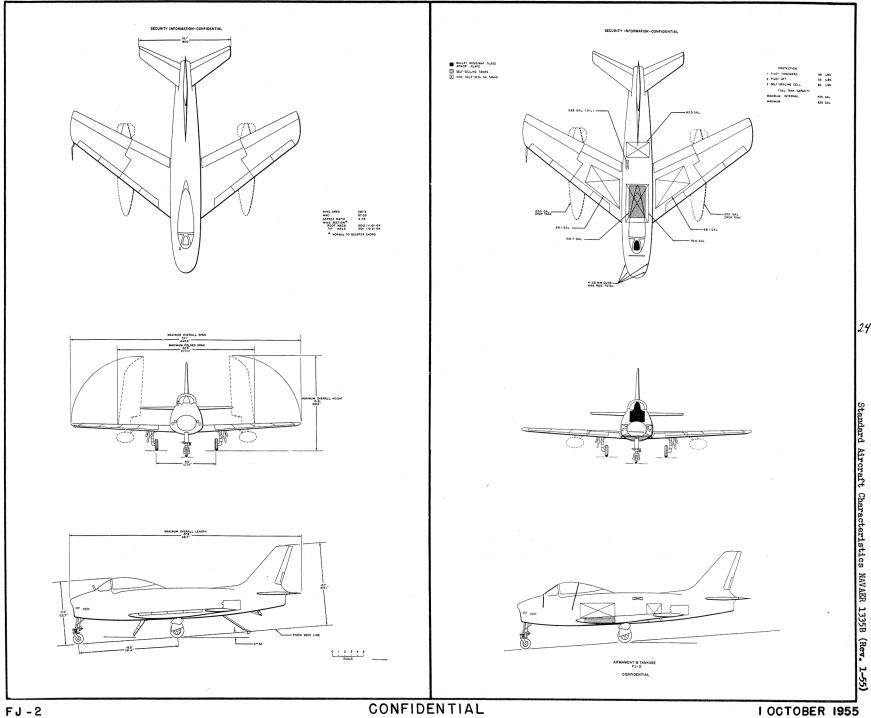
STANDARD AIRCRAFT CHARACTERISTICS

FJ-2 "FURY"

NORTH AMERICAN

rd Aircraft Characteristics NAVAER 1335A (REV. 1-55

CONFIDENTIAL



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I OCTOBER 1955

POWER PLANT

NO. & MODEL.....(1) J47-GE-2 MFR.....General Electric TYPE.....Axial Flow ENG. LENGTH.....148"

RATINGS

LBS. RPM ALT. 6,000 7,950 T.O. Static S.L. 7,950 6,000 MIL. Static S.L. NORM. 5,350 7,630 Static S.L.

SPEC. NO. E-610-B

ORDNANCE

GUNS

NO. SIZE LOCATION RDS. 20mm 600 Fuselage Forward

FIRE CONTROL

A.F.C.S.....MK.16 MOD. 5 Radar Ranging Equipment.....AN/APG-30

MISSION AND DESCRIPTION

The FJ-2 airplane is a single-seat, carrier-based, day fighter capable of operating from an H8, H4-1, C7 and C11 catapult and a MK 5 or MK 7 arresting gear.

Special features are electrically operated flaps, hydraulically operated speed brakes, automatic wing leading edge slats, hydraulic power-operated irreversible controls with artificial feel for the all-movable horizontal tail and ailerons, and a conventional pilot-operated rudder.

The cockpit is provided with 5 psi differential pressurization, a jettisonable canopy, an ejection type seat, and anti-G suit provisions.

Design maximum dive speed is 590 knots EAS at 5,000 ft. Design maximum Mach No. is 1.15 at 30,000 ft.

DEVELOPMENT

Mock-up date - - - - - - - - July 1951 First flight - - - - - - - December 1951 Service use to start - - - - December 1952

WEIGHTS

LOADINGS L.F. LBS. EMPTY.....11,802..... BASIC......12,314..... MAX. TAKE OFF*....19,000......5.6 MAX. LANDING.... (arrest)......16,000.......6.5

* Max. anticipated weight. All weights are estimated.

FUEL AND OIL

NO. TANKS TOTAL GAL. LOCATION 3 222 Fuselage 400 Wing, Drop FUEL GRADE.....JP-4 FUEL SPEC.....MIL-0-6081 OIL

CAPACITY (GALS.).....3.2 GRADE.....1010 SPEC.....MIL-0-6081

DIMENSIONS

TREAD.....91 - 0"

ELECTRONICS

UHF TRANSMTR-REC.....AN/ARC-27A UHF ADF.....AN/ARA-25 ADF......AN/ARN-21 (Replacement for AN/ARN-6)

		PERFORM	ANCE SUMMA	RY	
TAKE-OFF LOADING CONDITION	-	(1) FIGHTER 2-200 GAL. EXTERNAL TANKS			
TAKE-OFF WEIGHT	lb.	18,791			
Fuel	1b.	5,428			
Fayload	lb.			·	
Wing loading	lb./sq.ft.	65.3			
Stall speeu - power-off	kn.	115			
Take-off run at S.L calm	ft.	2,600			
Take-off run at S.L. 25 kn. wind	ft.	1,645			
Take-off to clear 50 ft calm	ft.	3,950			
Max. speed/altitude (A)	kn./ft.	533/5,000			
Rate of climb at S.L. (A)	fpm.	5,000			
Time: S.L. to 20,000 ft. (A)	min.	5.6			
Time: S.L. to 30,000 ft. (A)	min.	11.0	,		
Service ceiling (100 fpm) (A)	ft.	39,100			
Combat range	n.mi.	860			
Average cruising speed	kn.	450			
Cruising altitude(s)	ft.	37,000/42,000			
Combat radius	n.mi.	270			
Average cruising speed	kn.	440			
Cycle time	hrs.	1.9			
COMBAT LOADING CONDITION		(2) COMBAT			
COMBAT WEIGHT	lb.	15,813			
Engine power		Military			
Fuel	lb.	2,828			
Combat speed/combat altitude	kn./ft.	523/35,000			
Rate of climb/combat altitude	fpm/ft.	1,690/35,000	, , , , , , , , , , , , , , , , , , ,		
Combat ceiling (500 fpm)	ft.	41,700			
Rate of climb at S.L.	fpm.	7,230			
Max. speed at S.L.	kn.	587			•
Max. speed/altitude	kn./ft.	587/S.L.			
LANDING WEIGHT	1b.	14,125			
Fuel	1b.	885			
Stall speed - power-off	kn.	98.8			
Stall speed - with approach power	kn.	96.2			

NOTES

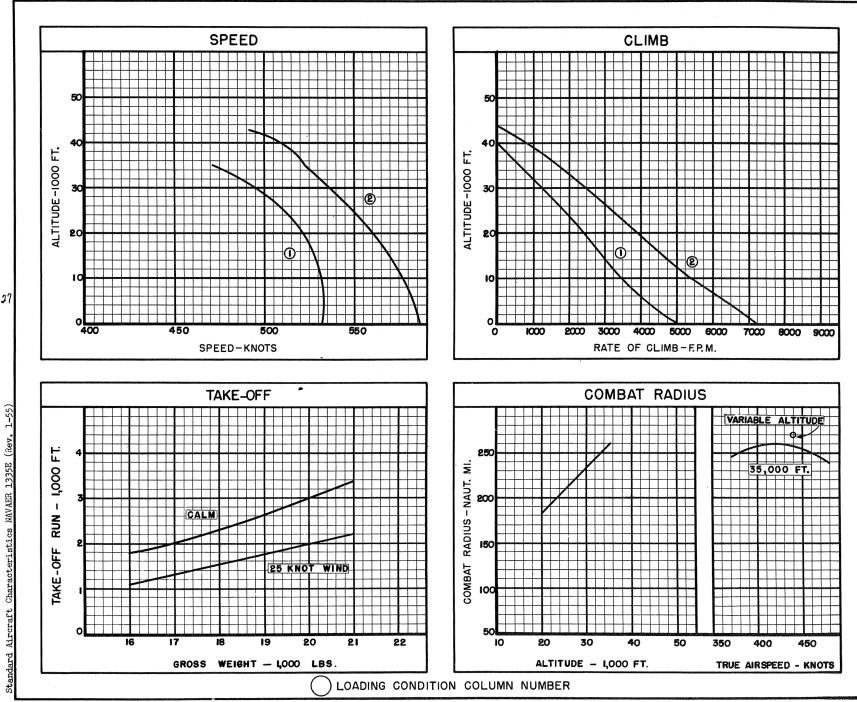
(A)	Military	Rated	Thrust
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In radius problem, tanks dropped prior to entering combat. If retained during combat until empty, radius would increase by 27 nautical miles.

Performance based on Flight Test

Range and radius based on flight test fuel consumption increased by 5%.

AVAER-1335D (Rev.



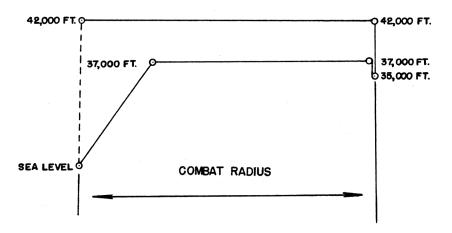
NOTES

GENERAL PURPOSE AND ESCORT FIGHTER COMBAT RADIUS PROBLEM (GAS TURBINE)

WARM-UP, TAXI, TAKE-OFF: 5 minutes at normal thrust.
CLIMB: To cruise altitude at military thrust.
CRUISE-OUT: At V for long range at cruise altitude.
COMBAT: At 35,000 ft. 20 minutes at military thrust (external tanks dropped prior

to combat)

CRUISE-BACK: At V for long range at cruise altitude.
RESERVE: Fuel for 20 minutes loiter at sea level at V for maximum endurance plus 5% initial fuel.



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