

and install a new placard on the cover, per Boeing Alert Service Bulletin MD11-21A033, Revision 01, dated April 30, 2001, excluding Evaluation Form or Revision 02, dated December 4, 2002. The replacement must be done with part numbers that are specified in View C-C, Figure 1, of the service bulletin.

(c) Accomplishment of the actions specified in McDonnell Douglas Service Bulletin MD11-21-033, dated May 1, 1992, before the effective date of this AD, is considered acceptable for compliance with the requirements of paragraph (b) of this AD.

Spares

(d) As of the effective date of this AD, no person shall install a cover assembly, part number ABM7569-1, on any airplane.

Alternative Methods of Compliance

(e) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Los Angeles Aircraft Certification Office (ACO), FAA. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Los Angeles ACO.

Note 3: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Los Angeles ACO.

Special Flight Permits

(f) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Incorporation by Reference

(g) Unless otherwise specified by this AD, the actions shall be done in accordance with Boeing Alert Service Bulletin MD11-21A033, Revision 01, dated April 30, 2001, excluding Evaluation Form or Boeing Alert Service Bulletin MD11-21A033, Revision 02, dated December 4, 2002. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Boeing Commercial Aircraft Group, Long Beach Division, 3855 Lakewood Boulevard, Long Beach, California 90846, Attention: Data and Service Management, Dept. C1-L5A (D800-0024). Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

Effective Date

(h) This amendment becomes effective on May 27, 2003.

Issued in Renton, Washington, on April 10, 2003.

Ali Bahrami,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2000-SW-50-AD; Amendment 39-13123; AD 2001-13-03 R1]

RIN 2120-AA64

Airworthiness Directives; Kaman Aerospace Corporation Model K-1200 Helicopters

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment revises an existing airworthiness directive (AD) for Kaman Aerospace Corporation (Kaman) Model K-1200 helicopters that currently requires reducing the life limit of the rotor shaft and teeter pin assembly and establishing a life limit for the flap clevis. This amendment retains those requirements but removes a flap clevis part number from the applicability and, as a result of a comment, changes the application of the life limit from the flap clevis to the flap clevis assembly. This amendment is prompted by the determination after an analysis of testing results that a certain flap clevis assembly should have an unlimited life. The actions specified by this revision are intended to remove the life limit for a specified flap clevis assembly. The actions specified by this AD are intended to prevent fatigue failure of the rotor shaft, teeter pin assembly, and flap clevis assembly, and subsequent loss of control of the helicopter.

DATES: Effective May 27, 2003.

FOR FURTHER INFORMATION CONTACT:

Richard Noll, Aviation Safety Engineer, Boston Aircraft Certification Office, 12 New England Executive Park, Burlington, MA 01803, telephone (781) 238-7160, fax (781) 238-7170.

SUPPLEMENTARY INFORMATION: A proposal to amend 14 CFR part 39 by revising AD 2001-13-03, Amendment 39-12283 (66 FR 34102, June 27, 2001), for 1 Kaman Model K-1200 helicopters, was published in the **Federal Register** on May 13, 2002 (67 FR 31992). The action proposed retaining the existing life limit for each rotor shaft, teeter pin assembly, and flap clevis, except flap

clevis, part number (P/N) K911049-021. That action was prompted by the determination after an analysis of testing results that flap clevis, P/N K911049-021, should have an unlimited life.

Interested persons have been afforded an opportunity to participate in the making of this amendment. Due consideration has been given to the comment received. The one commenter, the manufacturer, states that the flap clevis assembly part numbers should be identified instead of the flap clevis part numbers to be consistent with actual current maintenance practices. The FAA agrees because we have approved a revision to the Airworthiness Limitations of the Kaman Model K-1200 helicopter maintenance manual that imposes the life limit on the flap clevis assembly part numbers not the flap clevis part numbers. The proposed change will make this AD consistent with the Airworthiness Limitations section. Additionally, two part numbers for the flap clevis were incorrectly stated in paragraph (b) of the proposed AD; however, this change to flap clevis assembly parts number also corrects that error.

After careful review of the available data, including the comments noted above, the FAA has determined that air safety and the public interest require adopting the rule with the changes described previously. The FAA has determined that these changes will neither increase the economic burden on any operator nor increase the scope of the AD.

The FAA estimates that this AD will affect nine helicopters of U.S. registry. No additional costs will be incurred to accomplish the proposed actions because it would relieve a previously-imposed AD life limit for flap clevis, P/N K911049-021.

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory