

AD is being issued to find and fix fatigue cracking of the wings, which could result in structural failure of the wings and consequent loss of control of the airplane. This AD requires repetitive inspections to find cracks in the lower rear cap of the wing front spar, front spar web, and lower skin of the wings; and repair or replacement of any cracked part with a new part; and follow-on inspections at new intervals.

#### Determination of Rule's Effective Date

Since a situation exists that requires the immediate adoption of this regulation, it is found that notice and opportunity for prior public comment hereon are impracticable, and that good cause exists for making this amendment effective in less than 30 days.

#### Comments Invited

Although this action is in the form of a final rule that involves requirements affecting flight safety and, thus, was not preceded by notice and an opportunity for public comment, comments are invited on this rule. Interested persons are invited to comment on this rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified under the caption **ADDRESSES**. All communications received on or before the closing date for comments will be considered, and this rule may be amended in light of the comments received. Factual information that supports the commenter's ideas and suggestions is extremely helpful in evaluating the effectiveness of the AD action and determining whether additional rulemaking action would be needed.

Submit comments using the following format:

- Organize comments issue-by-issue. For example, discuss a request to change the compliance time and a request to change the service bulletin reference as two separate issues.
- For each issue, state what specific change to the AD is being requested.
- Include justification (e.g., reasons or data) for each request.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the rule that might suggest a need to modify the rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report that summarizes each FAA-public contact concerned with the substance of this AD will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this rule must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 2003-NM-23-AD." The postcard will be date stamped and returned to the commenter.

#### Regulatory Impact

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

The FAA has determined that this regulation is an emergency regulation that must be issued immediately to correct an unsafe condition in aircraft, and that it is not a "significant regulatory action" under Executive Order 12866. It has been determined further that this action involves an emergency regulation under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979). If it is determined that this emergency regulation otherwise would be significant under DOT Regulatory Policies and Procedures, a final regulatory evaluation will be prepared and placed in the Rules Docket. A copy of it, if filed, may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

#### Adoption of the Amendment

■ Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

##### § 39.13 [Amended]

■ 2. Section 39.13 is amended by adding the following new airworthiness directive:

**2003-08-13 Various surplus military airplanes manufactured by Consolidated, Consolidated Vultee, and Convair:** Amendment 39-13126. Docket 2003-NM-23-AD.

**Applicability:** Including, but not limited to, all of the following surplus military airplanes, certificated in any category:

Consolidated Vultee Model PB4Y-1, P4Y-2, and LB-30 airplanes;  
Consolidated and Convair Model B-24 airplanes; and  
Consolidated Model C-109 and C-87 airplanes.

**Note 1:** This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

**Compliance:** Required as indicated, unless accomplished previously.

To find and fix fatigue cracking in the lower rear cap of the wing front spar, front spar web, and lower skin of the wings, which could result in structural failure of the wings and consequent loss of control of the airplane, accomplish the following:

#### Initial & Repetitive Inspections

(a) Within 30 days after the effective date of this AD, do the actions specified in paragraphs (a)(1) and (a)(2) of this AD per a method approved by the Manager, Los Angeles Certification Office (ACO), FAA.

(1) Do an inspection (between 39 and 63 inches outboard of the airplane center line on both the left and right sides of the wings) to find cracks in the lower rear cap of the wing front spar, front spar web, and lower skin of the wings localized under the front spar lower cap. Special detailed inspection procedures must be sufficiently reliable to determine the location, size, and orientation of the cracks.

(2) Develop repetitive inspection intervals that prevent crack growth from exceeding the minimum residual strength required to support limit load on the affected structure. The repetitive inspection intervals must be approved by the Manager, Los Angeles ACO. Thereafter, do the inspection approved per paragraph (a)(1) of this AD at the intervals approved per this paragraph.

(b) If any crack is found during any inspection required by this AD, before further flight, do the action(s) specified in paragraphs (b)(1) and (b)(2) of this AD per a method approved by the Manager, Los Angeles ACO.

(1) Repair or replace the cracked part or structure.

(2) Repeat the inspection required by paragraph (a)(1) of this AD at reduced intervals approved by the Manager, Los Angeles ACO to find cracks before the growth is critical and exceeds the minimum residual strength required to support limit load on the affected structure.