

require, among other actions, modifying the down drive brackets of the left- and right-hand inboard flap track 1 assembly, and testing the torque value of the nuts.

Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were submitted in response to the supplemental proposal or to the FAA's determination of cost to the public.

Conclusion

After careful review of the available data, the FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

Cost Impact

The FAA estimates that 9 airplanes of U.S. registry will be affected by this AD, that it will take approximately 13 work hours per airplane to accomplish the modifications and installations, and that the average labor rate is \$60 per work hour. Based on these figures, the cost impact of the AD on U.S. operators is estimated to be \$7,020, or \$780 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and

that no operator would accomplish those actions in the future if this AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

Regulatory Impact

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory

Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

2003-10-12 Airbus: Amendment 39-13157. Docket 2001-NM-142-AD.

Applicability: The airplanes specified in Table 1 of this AD, certificated in any category. Table 1 is as follows:

TABLE 1.—APPLICABILITY

Model—	Which have received—	Excluding airplanes—
A330 series airplanes	Airbus Modification 45326 in production.	Modified in production per Airbus Modification 47619, or modified in service per Airbus Service Bulletin A330-57-3067, Revision 03, dated August 7, 2002.
A340 series airplanes	Airbus Modification 45326 in production.	Modified in production per Airbus Modification 47619, or modified in service per Airbus Service Bulletin A340-57-4075, Revision 02, dated August 7, 2002.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been otherwise modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or

repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent failure of the bolts due to flexural loads caused by transmission jam loading, which could lead to a "flap-locked" condition, causing reduced controllability of the airplane, accomplish the following:

Modification and Testing

(a) At the times specified in Table 2 of this AD, modify the down drive brackets of the left- and right-hand inboard flap track 1 assembly and test the torque value of the nuts by accomplishing all actions specified in the Accomplishment Instructions of Airbus Service Bulletin A330-57-3067, Revision 03, dated August 7, 2002 (for Model A330 series airplanes); or Airbus Service Bulletin A340-57-4075, Revision 02, dated August 7, 2002 (for Model A340 series airplanes); as applicable. Table 2 is as follows: