

proposed AD does not go undetected for a long period of time, the compliance is presented in calendar time instead of hours TIS.

### Regulatory Impact

#### *Would This Proposed AD Impact Various Entities?*

The regulations proposed herein would not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this proposed rule would not have federalism implications under Executive Order 13132.

#### *Would This Proposed AD Involve a Significant Rule or Regulatory Action?*

For the reasons discussed above, I certify that this proposed action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative,

on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action has been placed in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption **ADDRESSES**.

### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

### The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

### PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### **§ 39.13 [Amended]**

2. FAA amends § 39.13 by removing Airworthiness Directive (AD) 94-18-04

R1, Amendment 39-9173 (60 FR 62321, March 14, 1995), and by adding a new AD to read as follows:

**Univair Aircraft Corporation:** Docket No. 2002-CE-58-AD; Supersedes AD 94-18-04 R1, Amendment 39-9173.

(a) *What airplanes are affected by this AD?* This AD affects the following airplane models and serial numbers that are certificated in any category:

Models	Serial No.
Alon A-2 and A2-A .....	All.
ERCO 415-C, 415-CD, 415-D, 415-E, and 415-G.	All.
Forney F-1 and F-1A .....	All.
Mooney M10 .....	All.

(b) *Who must comply with this AD?*

Anyone who wishes to operate any of the airplanes identified in paragraph (a) of this AD must comply with this AD.

(c) *What problem does this AD address?*

The actions specified by this AD are intended to prevent wing damage caused by a corroded wing outer panel structural component, which, if not detected and corrected, could progress to the point of structural failure.

(d) *What actions must I accomplish to address this problem?* To address this problem, you must accomplish the following:

Actions	Compliance	Procedures
(1) Install inspection openings in the outer wing panels and inspect the wing outer panel internal structural components for corrosion and unrepaired corrosion damage.	Within the next 12 calendar months after March 24, 1995 (the effective date of AD 94-18-04 R1), unless already accomplished.	In accordance with Univair Service Bulletin No. 29, Revision B, dated January 2, 1995, or Univair Service Bulletin No. 29, Revision C, dated July 8, 1999, and Advisory Circular 43-4A, Corrosion Control for Aircraft.
(2) If corrosion or corrosion damage is found during the inspection required in paragraph (d)(1) of this AD, repair or replace components of the wing outer panel structure.	Repair or replace prior to further flight after the inspection required in paragraph (d)(1) of this AD.	In accordance with Univair Aircraft Corporation Service Bulletin No. 29, Revision B, dated January 2, 1995, or Univair Service Bulletin No. 29, Revision C, dated July 8, 1999, the applicable maintenance manual, and Advisory Circular 43-4A, Corrosion Control for Aircraft.
(3) Repetitively inspect the wing outer panel internal structural components for corrosion and unrepaired corrosion damage.	Initially inspect within the next 6 calendar months after the effective date of this AD unless the wing outer panel internal structure had been inspected for corrosion within the previous 6 calendar months immediately prior to the effective date of this AD. Repetitively inspect thereafter at intervals not to exceed 12 months after the last inspection.	In accordance with Univair Aircraft Corporation Service Bulletin No. 29, Revision C, dated July 8, 1999, and Advisory Circular 43-4A, Corrosion Control for Aircraft.
(4) At any time corrosion or corrosion damage is found, repair or replace components of the wing outer panel structure.	Repair or replace prior to further flight after the inspection in which the corrosion or corrosion damage is found. Continue with the repetitive inspection requirements of this AD.	In accordance with Univair Aircraft Corporation Service Bulletin No. 29, Revision B, dated January 2, 1995, or Univair Service Bulletin No. 29, Revision C, dated July 8, 1999, and Advisory Circular 43-4A, Corrosion Control for Aircraft.

**Note 1:** The compliance times specified in Univair Aircraft Corporation Service Bulletin No. 29, Revision B, dated January 2, 1995, or Univair Service Bulletin No. 29, Revision C, dated July 8, 1999, are different from those required by this AD. The compliance times in this AD take precedence over those in the service bulletin.

(e) *Can I comply with this AD in any other way?*

(1) To use an alternative method of compliance or adjust the compliance time, follow the procedures in 14 CFR 39.13. Send these requests to the Manager, Denver Aircraft Certification Office (ACO). For information on any already approved alternative methods of compliance, contact

Roger Caldwell, Aerospace Engineer, FAA, Denver Aircraft Certification Office, 26805 East 68th Avenue, Room 214, Denver, Colorado 80249-6361; telephone: (303) 342-1086; facsimile: (303) 342-1088.

(2) Alternative methods of compliance approved for the inspection required in AD 94-18-04 R1, which is superseded by this