

• do not add any additional burden upon the public than was already proposed in the NPRM.

How does the revision to 14 CFR part 39 affect this AD? On July 10, 2002, FAA published a new version of 14 CFR part 39 (67 FR 47997, July 22, 2002), which governs FAA's AD system. This regulation now includes material that

relates to special flight permits, alternative methods of compliance, and altered products. This material previously was included in each individual AD. Since this material is included in 14 CFR part 39, we will not include it in future AD actions.

Cost Impact

How many airplanes does this AD impact? We estimate that this AD affects 24 airplanes in the U.S. registry.

What is the cost impact of this AD on owners/operators of the affected airplanes? We estimate the following costs to accomplish the initial inspection:

Labor cost	Parts cost	Total cost per airplane	Total cost on U.S. operators
250 hours × \$60 per hour = \$15,000.	No parts necessary to accomplish inspection.	\$15,000 per airplane.	\$15,000 × 24 airplanes = \$360,000.

The follow-up inspections will be substantially less than the initial inspection because the flight control rods only have to be removed in the initial inspection. Replacement control rods cost \$2,000. We have no way of determining the number of airplanes that may need such repair/replacement.

Compliance Time of This AD

What will be the compliance time of this AD? The initial inspection compliance time of this AD is "within the next 3 months after the effective date of this AD or within 24 months after the last inspection accomplished in accordance with Shorts Service Bulletin 27-74 (any revision level), whichever occurs later." The repetitive inspection compliance time of this AD is "thereafter at intervals not to exceed 24 months."

Why is the compliance time presented in calendar time instead of hours time-in-service (TIS)? The unsafe condition specified by this AD is caused by corrosion. Corrosion can occur regardless of whether the aircraft is in operation or is in storage. Therefore, to ensure that the unsafe condition specified in this AD does not go undetected for a long period of time, the compliance is presented in calendar time instead of hours TIS.

Regulatory Impact

Does this AD impact various entities? The regulations adopted herein will not

have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

Does this AD involve a significant rule or regulatory action? For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the final evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

■ Accordingly, under the authority delegated to me by the Administrator, the Federal Aviation Administration

amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. FAA amends § 39.13 by adding a new AD to read as follows:

2003-13-07 Short Brothers and Harland Ltd.: Amendment 39-13207; Docket No. 2003-CE-15-AD.

(a) *What airplanes are affected by this AD?* This AD affects Models SC-7 Series 2 and SC-7 Series 3 airplanes, all serial numbers, that are certificated in any category.

(b) *Who must comply with this AD?* Anyone who wishes to operate any of the airplanes identified in paragraph (a) of this AD must comply with this AD.

(c) *What problem does this AD address?* The actions specified by this AD are intended to prevent failure of any flight control system rod caused by cracks or corrosion. Such failure could lead to complete failure of the flight control system with consequent loss of control of the airplane.

(d) *What actions must I accomplish to address this problem?* To address this problem, you must accomplish the following:

Actions	Compliance	Procedures
(1) Inspect all flight control system rods for cracks and corrosion damage.	Initially inspect within the next 3 months after August 11, 2003 (the effective date of this AD or within 24 months after the last inspection accomplished in accordance with Shorts Service Bulletin 27-74 (any revision level), whichever occurs later, unless already accomplished. Repetitively inspect thereafter at intervals not to exceed 24 months.	In accordance with the ACCOMPLISHMENT INSTRUCTIONS section of Shorts Service Bulletin Number 27-77, Original Issue 27/FEB/03.