

### *Would This Proposed AD Involve a Significant Rule or Regulatory Action?*

For the reasons discussed above, I certify that this proposed action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action has been placed in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption **ADDRESSES**.

### **List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Safety.

### **The Proposed Amendment**

Accordingly, under the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

### **PART 39—AIRWORTHINESS DIRECTIVES**

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### **§ 39.13 [Amended]**

2. FAA amends § 39.13 by removing Airworthiness Directive (AD) 2000–23–01, Amendment 39–11971 (65 FR 70645, November 27, 2000), and by adding a new AD to read as follows:

**Cessna Aircraft Company:** Docket No. 2002–CE–57–AD; Supersedes AD 2000–23–01, Amendment 39–11971.

(a) *What airplanes are affected by this AD?* This AD affects Models 402C and 414A airplanes, all serial numbers, that are certificated in any category.

(b) *Who must comply with this AD?* Anyone who wishes to operate any of the airplanes identified in paragraph (a) of this AD must comply with this AD.

(c) *What problem does this AD address?* The actions specified by this AD are intended to prevent wing spar cap failure due to undetected fatigue cracks. Such failure could result in loss of a wing with consequent loss of airplane control.

(d) *What actions must I accomplish to address this problem?* To address this problem, you must inspect the wing spar caps for fatigue cracks and repair or replace the wing spar caps as necessary and incorporate a spar strap modification on each wing spar in accordance with Cessna Service Bulletin MEB02–5, dated June 24, 2002, and Cessna Service Kit SK402–47, dated June 24, 2002, as follows:

Compliance times	Affected airplanes
(1) Inspect and modify at whichever of the following that occurs later and repair or replace as necessary prior to further flight after the inspection, unless already accomplished (no repetitive actions necessary): (i) Upon accumulating 8,500 hours time-in-service (TIS) on a wing spar; or (ii) Within the next 500 hours TIS after the effective date of this AD or 12 months after the effective date of this AD, whichever occurs first	Cessna Models 402C and 414A airplanes, serial number 414A0001 through 414A0047 and 414A0049 through 414A0200.
(2) Inspect and modify at whichever of the following that occurs first and repair or replace as necessary prior to further flight after the inspection, unless already accomplished (no repetitive actions necessary): (i) Upon accumulating 14,500 hours TIS on a wing spar; (ii) Within the next 500 hours TIS after the effective date of this AD or 12 months after the effective date of this AD, whichever occurs first	Cessna Models 402C and 414A airplanes, serial numbers 414A0201 or through 414A1212.

(e) *Can I comply with this AD in any other way?*

(1) To use an alternative method of compliance or adjust the compliance time, follow the procedures in 14 CFR 39.19. Send these requests to the Manager, Wichita Aircraft Certification Office (ACO). For information on any already approved alternative methods of compliance, contact Paul Nguyen, Aerospace Engineer, FAA, Wichita Aircraft Certification Office, 1801 Airport Road, Mid-Continent Airport, Wichita, Kansas 67209; telephone: (316) 946–4125; facsimile: (316) 946–4107.

(2) Alternative methods of compliance approved in accordance with AD 2000–23–01 and AD 99–11–13 are not approved as alternative methods of compliance with this AD.

(f) *How do I get copies of the documents referenced in this AD?* You may get copies of the documents referenced in this AD from the Cessna Aircraft Company, Product Support, P.O. Box 7706, Wichita, Kansas 67277; telephone: (316) 517–5800; facsimile: (316) 942–9006. You may view these

documents at FAA, Central Region, Office of the Regional Counsel, 901 Locust, Room 506, Kansas City, Missouri 64106.

(g) *Does this AD action affect any existing AD actions?* This amendment supersedes AD 2000–23–01, Amendment 39–11971.

Issued in Kansas City, Missouri, on September 26, 2003.

**Michael Gallagher,**

*Manager, Small Airplane Directorate, Aircraft Certification Service.*

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## **DEPARTMENT OF TRANSPORTATION**

### **Federal Aviation Administration**

#### **14 CFR Part 39**

**[Docket No. 2002–CE–05–AD]**

**RIN 2120–AA64**

### **Airworthiness Directives; Cessna Aircraft Company Models 401, 401A, 401B, 402, 402A, 402B, 411, and 411A Airplanes**

**AGENCY:** Federal Aviation Administration, Transportation.

**ACTION:** Supplemental notice of proposed rulemaking (NPRM); Reopening of the comment period.

**SUMMARY:** This document proposes to revise an earlier proposed airworthiness directive (AD) that would supersede Airworthiness Directive (AD) 79–10–15