

Actions	Compliance	Procedures
<p>(1) Incorporate the following flight restrictions into the Limitations Section of the flight manual:</p> <p>(i) "The operation of the engine will be limited to maximum 100% power (max. continuous power)." and</p> <p>(ii) "Hence the take-off procedure (take-off with take-off power 115%, see section 4.5.2.2, of the Flight Manual) must not be selectd. Alternative procedures (i.e., take-off with max. continuous power 100%) are published in the flight Manual."</p> <p>(2) Fabricate a placard that incorporates the following words (using at least 1/8-inch letters) and install this placard close to the throttle lever: "Operation above 100% continuous power is not allowed! (see SB A31-10-1065)."</p> <p>(3) As an alternative method of compliance to this AD, replace the lower cog wheel (P/N: 43.15.0028) with a modified cog wheel (P/N: 43.15.0043).</p>	<p>Within the next 10 days after October 20, 2003 (the effective date of this AD), unless already accomplished.</p> <p>Within the next 10 days after October 20, 2003 (the effective date of this AD), unless already accomplished.</p> <p>At any time as terminating action for the limitations and placard requirements of this AD.</p>	<p>Either insert a copy of this portion of the AD or Stemme Service Bulletin A31-10-065, Am.-Index: 02a, dated February 25, 2003, into the Limitations Section of this of the AFM. The owner/operator holding at least a private pilot certificate as authorized by section 43.7 Federal Aviation Regulations (14 CFR 43.7) may do this flight manual insertion. Make an entry into the aircraft records showing compliance with these portions of the AD in accordance with section 43.9 of the Federal Aviation regulations (14 CFR 43.9).</p> <p>No specific procedures are necessary for this action. Stemme Service Bulletin A31-10-065, Am.-Index: 02a, dated February 25, 2003, references this action. The owner/operator holding at least a private pilot certificate as authorized by section 43.7 of the Federal Aviation Regulations (14 CFR the throttle 43.7) may do the placard requirements. Make an entry into the aircraft records showing compliance with these portions of the AD in accordance with section 43.9 of the Federal Aviation (14 CFR 43.9).</p> <p>Use the instructions in Stemme Service Bulletin A31-10-065, Am.-Index: 02a, dated February 25, 2003.</p>

Why Is the FAA Not Mandating the Cog Wheel Replacement?

(f) We are not mandating the cog wheel replacement (as specified in the service information) in this AD action because of the "bootstrapping requirement." When we issue an AD that involves requirements affecting flight safety where we do not first provide notice and an opportunity for public comment, then we are only able to include a short-term action that immediately corrects the unsafe condition.

(1) The Administrative Procedures Act does not permit combining a long-term requirement with a short-term action when we do not provide prior public comment. The short-term action and the long-term action are analyzed separately for justification to bypass public notice.

(2) We may initiate future AD action with public comment to mandate the cog wheel replacement as terminating action for the AFM requirements of this AD. This cog wheel replacement is optional in this AD as terminating action.

What About Alternative Methods of Compliance?

(g) You may request a different method of compliance or a different compliance time for this AD by following the procedures in 14 CFR 39.13. Send your request to the Manager, Standards Office, Small Airplane Directorate, FAA. For information on any already approved alternative methods of compliance, contact Greg Davison, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4130; facsimile: (816) 329-4090.

Is There Material Incorporated by Reference?

(h) If you choose to do the replacement required by this AD, then you must use Stemme Service Bulletin A31-10-065, Am.-Index: 02a, dated February 25, 2003. The Director of the Federal Register approved the incorporation by reference of this service bulletin in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. You may get a copy from Stemme GmbH & Co. KG, Gustav-Meyer-Allee 25, D-13355 Berlin, Germany; telephone: 49.33.41.31.11.70; facsimile: 49.33.41.31.11.73. You may review copies at FAA, Central Region, Office of the Regional Counsel, 901 Locust, Room 506, Kansas City, Missouri 64106; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

Is There Other Information That Relates to This Subject?

(i) German AD Number 2002-389/2, Effective date: April 17, 2003, also addresses the subject of this AD.

Issued in Kansas City, Missouri, on September 30, 2003.

Michael Gallagher,

Manager, Small Airplane Directorate, Aircraft Certification Service.

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2003-CE-42-AD; Amendment 39-13333; AD 2003-20-15]

RIN 2120-AA64

Airworthiness Directives; Pilatus Aircraft Ltd. Models PC-12 and PC-12/45 Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule; request for comments.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for certain Pilatus Aircraft Ltd. (Pilatus) Models PC-12 and PC-12/45 airplanes. This AD requires you to inspect for certain installed fuel booster pumps and replace that fuel booster pump, inspect other certain fuel booster pumps for defects, and either install lead protection spiral wrap or replace the defective fuel booster pumps, depending on whether defects are found. This AD is the result of mandatory continuing airworthiness information (MCAI) issued by the airworthiness authority for Switzerland. We are issuing this AD to detect and correct any defective fuel booster pump, which could result in electrical arcing from the leads in an air/fuel mixture.