

contacting the Rules Docket at the location provided under the caption **ADDRESSES**.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

#### The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the

Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

**Airbus:** Docket 2001–NM–216–AD.

*Applicability:* Airplanes as listed in Table 1 of this AD, certificated in any category.

TABLE 1.—APPLICABILITY

Airplane models	As listed in Airbus Service Bulletin—
A300 B2 and B4 series .....	A300–76–0007, Revision 06, dated August 23, 2001.
A300 B2 and B4 series .....	A300–76–0015, Revision 02, dated August 23, 2001.
A300 B2 and B4 series .....	A300–76–0016, Revision 03, dated August 23, 2001.
A300 B4–620, B4–622, C4–620 .....	A300–76–6002, Revision 02, dated August 23, 2001.
A300 B4–620 .....	A300–76–6003, Revision 04, dated February 26, 2002.
A300 B4–601, –603, and –605R .....	A300–76–6007, Revision 01, dated March 14, 2000.
A300 B4–601, B4–603, B4–605R, and C4–605R Variant F .....	A300–76–6009, Revision 02, dated October 29, 1999.
A310–203, –204, –221, and –222 .....	A310–76–2001, Revision 01, dated March 14, 2000.
A310–203 .....	A310–76–2004, Revision 03, dated August 23, 2001.
A310–203, –221, and –222 .....	A310–76–2005, Revision 01, dated March 14, 2000.
A310–221, –222, and –322 .....	A310–76–2006, Revision 03, dated February 26, 2002.
A310–204 and –304 .....	A310–76–2010, Revision 03, dated August 23, 2001.
A310–203, –204, and –304 .....	A310–76–2012, Revision 02, dated November 5, 2001.

*Compliance:* Required as indicated, unless accomplished previously.

To prevent hard points in the throttle control system that could lead to jamming of the throttle control cable, which could result in an asymmetric thrust condition and consequent reduced controllability of the airplane, accomplish the following:

#### Modifications

(a) Within 22 months after the effective date of this AD, do the actions specified in paragraphs (a)(1), (a)(2), (a)(3), (a)(4), (a)(5), and (a)(6) of this AD; as applicable.

(1) For airplanes listed in Airbus Service Bulletin A300–76–0007, Revision 06, dated August 23, 2001: Install a flexible ice protection boot on the upper fitting of the throttle and fuel shut-off valve control cables in each engine pylon, per the Accomplishment Instructions of that service bulletin.

(2) For airplanes listed in Airbus Service Bulletin A300–76–0015, Revision 02, dated August 23, 2001; or A310–76–2001, Revision 01, dated March 14, 2000: Install a heating system for the throttle control system in each engine pylon, per the Accomplishment Instructions of the applicable service bulletin.

(3) For airplanes listed in Airbus Service Bulletin A300–76–0016, Revision 03, dated August 23, 2001; A300–76–6002, Revision 02, dated August 23, 2001; or A310–76–2005, Revision 01, dated March 14, 2000: Replace, with new improved parts, the roller and rotation pin of the secondary relay lever of the throttle control system in each engine pylon. Accomplish the replacement per the Accomplishment Instructions of the applicable service bulletin.

(4) For airplanes listed in Airbus Service Bulletin A300–76–6007, Revision 01, dated March 14, 2000; or A310–76–2010, Revision

03, dated August 23, 2001. Install a new cooling duct and a new cooling shroud for the throttle control cable, per the instructions in the “Description” section of Airbus Service Bulletin A300–76–6007, Revision 01; or per the Accomplishment Instructions of A310–76–2010, Revision 03; as applicable.

**Note 1:** Airbus Service Bulletins A300–76–6007, Revision 01; and A310–76–2010, Revision 03; refer to GE CF6–80C2 Service Bulletins 71–088, Revision 03, dated March 15, 1991; and 75–021, Revision 03, dated August 5, 1992; for additional service information for accomplishing the installation of a new cooling duct and a new cooling shroud for the throttle control cable.

(5) For airplanes listed in Airbus Service Bulletin A300–76–6009, Revision 02, dated October 29, 1999; or A310–76–2012, Revision 02, dated November 5, 2001: Install an elastomer plug filled with grease on the end fitting of the throttle control cable in each engine pylon, per the Accomplishment Instructions of the applicable service bulletin.

(6) For airplanes listed in Airbus Service Bulletin A310–76–2004, Revision 03, dated August 23, 2001: Install a sealing sleeve (also called a sealing boot) on the flexible control ball joint of the throttle control cable in each engine pylon (including a detailed inspection for deterioration of the throttle control cable, and replacement of the throttle control cable, as applicable) by doing all actions in the Accomplishment Instructions of the service bulletin, per the Accomplishment Instructions of the service bulletin. Replacement of the throttle control cable, if required, must be accomplished before further flight.

**Note 2:** For the purposes of this AD, a detailed inspection is defined as: “An intensive visual examination of a specific

structural area, system, installation, or assembly to detect damage, failure, or irregularity. Available lighting is normally supplemented with a direct source of good lighting at intensity deemed appropriate by the inspector. Inspection aids such as mirror, magnifying lenses, etc., may be used. Surface cleaning and elaborate access procedures may be required.”

#### Accomplishment of Required Actions per Previous Service Bulletin Revisions

(b) Actions accomplished before the effective date of this AD per previous service bulletin revisions are acceptable for compliance with paragraph (a) of this AD; as specified in paragraph (b)(1), (b)(2), (b)(3), (b)(4), (b)(5), and (b)(6) of this AD; as applicable.

(1) Accomplishment of the installation required by paragraph (a)(1) of this AD per Airbus Service Bulletin A300–76–007, Revision 05, dated March 14, 2000; is acceptable for compliance with paragraph (a)(1) of this AD.

(2) Accomplishment of the replacement required by paragraph (a)(2) of this AD per Airbus Service Bulletin A300–76–0015, Revision 01, dated March 14, 2000, is acceptable for compliance with paragraph (a)(2) of this AD.

(3) Accomplishment of the replacement required by paragraph (a)(3) of this AD per Airbus Service Bulletin A300–76–016, Revision 02, dated March 14, 2000; or A300–76–6002, Revision 01, dated March 14, 2000; as applicable; is acceptable for compliance with paragraph (a)(3) of this AD.

(4) Accomplishment of the installation required by paragraph (a)(4) of this AD per Airbus Service Bulletin A310–76–2010, Revision 02, dated March 14, 2000, is acceptable for compliance with paragraph (a)(4) of this AD.