

Massachusetts Institute of Technology
Instrumentation Laboratory
Cambridge Massachusetts

LUMINARY Memo #60

To: Distribution
From: J. Kernan, P. Volante
Date: 9 January 1969
Subject: LUMINARY Level 5 Test Plan

Level 5 testing of the LUMINARY program consists of:

1. Rerunning selected Level 4 tests (some with modifications).
2. Specific tests requested by MSC as a result of the pre-FACI review.
3. Selected Level 3 tests.

All Level 4 reruns include the following instrument errors:

1. Radar

	Bias	Std. Deviation of Noise
Range	500 ft *	.3% Range
	80 ft **	Same
Range-Rate	.3 fps	1.3/3% Range Rate
Shaft & Trunnion	15 mr	.9 mr

2. IMU - The uncompensated IMU errors are:

Pipa Bias +.2 CM./Sec in each component
Pipa Scale Factor Error -120 PPM in each component
Gyro Bias Drift +2 meru in X
-2 meru in Y
+2 meru in Z

* If Range > 50.8 N. M.

** If Range < 50.8 N. M.

Input Axis Sensitive Drift	-8 meru/G in X
	+8 meru/G in Y
	-8 meru/G in Z
Spin Reference Axis Sensitive Drift	+5 meru/G in each axis

The tests being run in each category are:

1. Level 4 Reruns

Test Description	Responsible Engineer
5.1 Tests 4.3 and 4.5 combined	Eyles/Covelli
5.2 Test 4.4	Bernikowich/Covelli
5.3 Test 4.6	Batra/Covelli
5.4 Test 4.7	Finkelstein/Covelli
5.5 Test 4.10	Berman/Covelli
5.6 Tests 4.11 and 4.13 combined	White
5.7 Test 4.15 with TPI-TPM added	Connor/White
5.8 Tests 4.16 and 4.18 combined	Gamer/White
5.9 Test 4.19 with SOR-TPI-TPM added	Albert/White
5.10 Test 4.27	Hatch/White
5.11 Test 4.28	Wiggins/White
5.12 Test 4.30	Dunbar/White

2. MSC Requested Tests

5.13 P32 with N=5 C=0	White
5.14 Same as 5.13 with different initial conditions	White
5.15 Same as 5.13 except $C \neq 0$	White
5.16 Same as 5.14 except $C \neq 0$	White
5.17 P32 with N=5, C=0, $V_r = 6.8$ fps	White
5.18 Same as 5.17 except $V_r = 7.2$ fps	White
5.19 P32 - Alarm 602	White
5.20 P32 - Alarm 603	White
5.21 P00 Integration	Wiggins
5.22 4.13 Run with Navigation at ranges less than 5000 meters	White

5.23	P20 with V32 response to Alarm 503	Dunbar
5.24	P20 with V32 and V34 responses to Alarm 525 and to V06N05 (Delta Theta Display)	Dunbar
5.25	P22 in no update (with Downlink edit) - Test 4.8	Millard
5.26	P21 (LM option) with LM on lunar surface - Test 4.29	Dunbar
5.27	R04 in P47 - Test 4.1 (with Downlink edit)	Batra
5.28	A backward integration over more than one revolution	Dunbar
5.29	R29 downlink check - Test 4.10	Berman
5.30	Verify Alarm 1706 in P40 and P42 (This was done on the Hybrid)	Adler
5.31	Docked DPS external Delta V burn with 5 seconds at 10% thrust and 21 seconds at 40% thrust - Test 4.21	Fisher
5.32	Same as 5.31 except the burn is a Lambert targetted burn (P31-P40) - Test 4.2	Fisher
5.33	Run P47 from TPM to near intercept with PIPA bias error	Adler
5.34	A 30-second duration P42 burn with an astronaut slippage or TIG and astronaut termination. R30 continuously operated during burn - Test 4.25 (modified)	Braunhardt
5.35	P31-P42 burn with engine fail at the end of the burn	Kirven
5.36	P40 burn with desired V_G less than that accumulated during ullage	Kirven
5.37	Verify vertical rise logic with PCR 258 - Test 4.10	Berman
5.38	Verify RCS insertion logic (done with Level 3 test 12.3)	Berman
5.39	Same as 5.38, using Level 3 test 12.4	Berman

5. 40	Verify Alarms 511 and 523 and checklist code 500 in R12 (to be done on Hybrid)	Covelli
5. 41	Verify the reasonability test limits for velocity and altitude	Covelli
5. 42	Verify that no updating of the state vector occurs until four seconds after the detection of a data good discrete	Covelli
5. 43	Run P22 and do R21 designate when the CM is directly overhead	Dunbar
5. 44	Verify R21 operation when R21 calls R61 (Hybrid)	Volante
5. 45	Verify R21 operation during RR turn on sequence	Dunbar
5. 46	Verify that R21 does not operate when the RNDVZFLG is reset	Dunbar
5. 47	Verify that R22 calls R61 when the 30 ⁰ RR tracking limit is reached (Hybrid)	Volante
5. 48	Verify that Alarm 514 is properly displayed (Hybrid)	Volante
5. 49	Verify that the 520 alarm is called and displayed properly (Hybrid)	Volante
5. 50	Verify proper operation of the V34E response to the V16N80 display in R24	Dunbar
5. 51	Verify proper operation of R25 when the RR auto mode discrete goes from on-off and off-on	Dunbar
5. 52	Abort stage when vehicle is maneuvering at a high rate (modified Test 4. 7)	Braunhardt/Covelli

3. Level 3 Tests

5.53	IMU orientation determination	Millard/Moore
5.54	Same as 5.52 with recycle exercised	Millard/Moore
5.55	Re-align to preferred orientation	Millard/Moore
5.56	Re-align to nominal orientation using gyro torque coarse align option and realign	Millard/Moore
5.57	IMU orientation determination, re-align to landing site orientation, re-align to REFSMMAT	Millard/Moore
5.58	Re-align to preferred orientation, re-align to nominal orientation	Millard/Moore
5.59	Lunar surface preferred orientation (Level 3 Test P57.1)	Millard/Moore
5.60	Landing site orientation, option 1 (Level 3 Test P57.2)	Millard/Moore
5.61	Landing Site orientation, option 2 (Level 3 Test P57.3)	Millard/Moore
5.62	Landing site orientation, option 3 (Level 3 Test P57.4)	Millard/Moore