NAVAIR 00-110AA3-4



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Standard Aircraft Characteristics

A-3B (CLE) AIRCRAFT

(TITLE UNCLASSIFIED)

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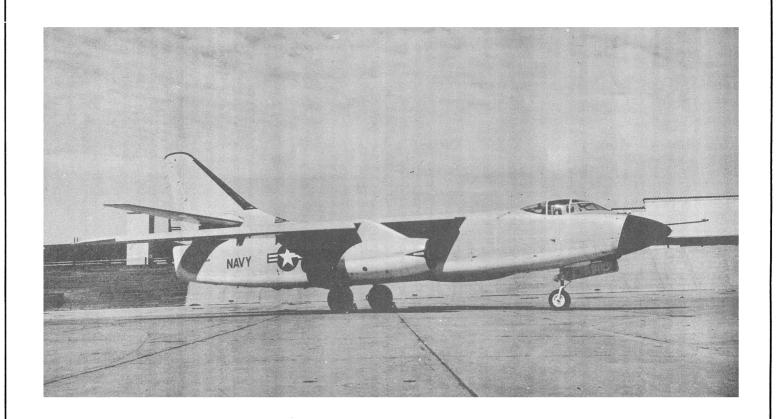
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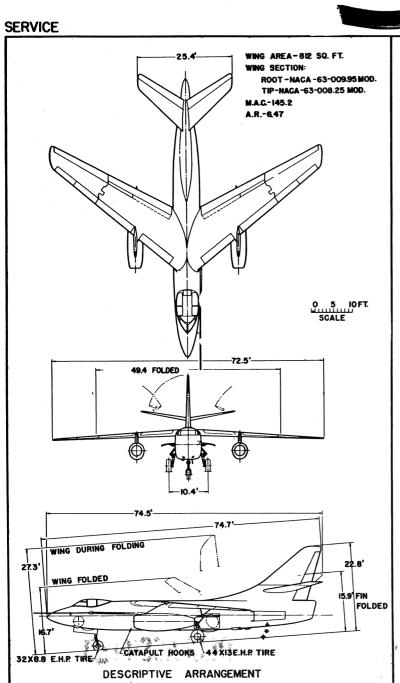
STANDARD AIRCRAFT CHARACTERISTICS

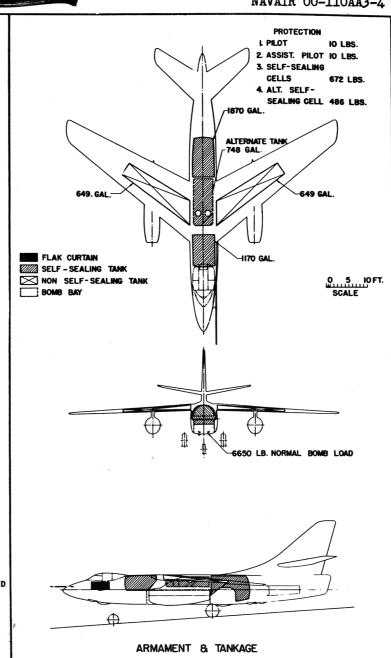
A-3B (CLE) SKYWARRIOR

DOUGLAS









TO ASSESS





VAVAIR UU-IIUAA)-4

POWER PLANT

No. & Model(2) J57-P-10
MfrPratt & Whitney
Eng. Spec. No N-1700-A (2-2-55)
Type Turbojet
ComprDual rotor, axial flow
Length 158 in.
Diameter 41 in.
No. & Type Assist12-5KS4500 JATO
Tail Pipe Nozzle Constant Exit
DATINCE Area

*N₁ Speed low press. compressor **N₂ Speed high press. compressor

DIMENSIONS

Wing: Area	812 sq.ft.*
Span	72.5 ft.
M.A.C	145.1 in.*
Sweepback	35 . 9°
Length	74.7 ft.
Height	22.8 ft.
Tread	10.4 ft.
*all aerodynamic da	ata are
based on the origi	nal wing
area of 779 sq.ft.	and MAC

ORDNANCE

of 140.1 inches.

Maximum Bomb Capa	city: 12,800 lb
BOMB	S
12-500 lb. G.P.	
6-1000 lb. G.P.	4-2000 lb. G.P.
MINE	ES
12-500 lb. Mk.50	6-1000 lb. Mk.52
6-1000 lb. Mk.36	12-374 lb. Mk.53
2-2000 lb. Mk.10	2-2000 lb. XA-4A
4-2000 1b. Mk.25	4-2000 lb. XG-7
Special Stores	

MISSION AND DESCRIPTION

The primary mission of the A3D-2 airplane is the attack and destruction of enemy ground and surface targets as required for mine-laying, reconnaissance, and high or low altitude attack missions. The airplane is designed to operate from land bases and from carriers.

The airplane has a conventional swept-wing structure. Two turbo-jet engines are enclosed in underwing nacelles. Provisions are made for a threeman crew; a pilot, a bomber-assistant pilot, and a gunner-navigator.

The tricycle landing gear, arresting gear, wingfold and tail-fold mechanisms, single-slotted wing flaps, fuselage speed brakes, and power mechanisms for rudder, elevator and ailerons are operated by hydraulic power. The horizontal stabilizer is adjustable for trim in flight. Leading edge slats are actuated automatically by aerodynamic loads.

The airplanes of Contract NOa(s) 59-0150 differs from the last 20 A3D-2 aircraft of Contract NOa(s) 55-190 in that four systems of electronic countermeasures have replaced the Aero-21B Tail Turret System and the ASB-7 Bomb Director System has replaced the ASB-1 system. The airplanes are the cambered wing leading edge configuration with tanker-receiver capabilities.

DEVELOPMENT

Contract NOa(s) 55-190c, 20 airplanes (delivered) Contract: NOa(s) 59-0150, 21 airplanes

First Flight: March 1960 Navy Acceptance: April 1960

Final Fleet Delivery: January 1961 (scheduled)

WEIGHTS				
Loading	Lbs.	<u>L.F.</u>		
Empty (C)	39,620			
Basic	40,120			
Design	55,942	3.4		
Combat	62,089	3.06		
Maximum T.O.				
a. Carrier	73,000	2.6		
b. Field	78,000	2.4		
Maximum Landing	5	1-1		
a. Carrier	49,000	1.79		
b. Field	56,000			

FUEL AND OIL	ı
Gal. No. Tanks Location	
3040 2 *Fuselage	
1298 2 Wing	
748 1 *Upper Bomb Ba Auxiliary	У
5086 Gal. Total usable fue	1
1224 1 **Lower Bomb Ba Auxiliary	y
Fuel GradeJP-4 or JP-	
Fuel SpecMIL-F-562	24
*Self-sealing	
**Service-Installed Kit (Tanker	r)
OIL	
Cal. No. Tanks Location	
11 2 Integral wit engines	h
Oil Specification MIL-L-78	80
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	.

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ELECTRONICS

VHF Trans-Receiver	AN/ARC-1
TACAN	AN/ARN-21
VOR Homing	AN/ARN-14F
UHF Trans-Receiver	AN/ARC-27A
HF Receiver-Trans	AN/ARC-38
Radio Altimeter	AN/APN-22
IFF Transponder	AN/APX-61
Interphone DAC Tran	
Coder	
UHF Direction Finder	
а	nd AN/ARN-6
DECM	AN/ALQ-19
DECM	AN/ALQ-32
	AIV ALW-D
DECM	AN/ALQ-38
BOMB DIRECTOR	AN/ASB-





		PERFORMA	NCE SUMMA	RY		
TAKE-OFF LOADING CONDITION	-		2 2050 lb stores	(5) Full Fuel	(7) Full Fuel Low Alt. Attack 3-1300 lb Stores	(9) Full Fuel High Alt. Attac
	7.	73,000	78,000	Upper BB Tank 83,259	75,081	77, 292
TAKE-OFF WEIGHT (A)	1b.		70,000	00,203	75,001	11,636
Fuel (JP-5)		27,278(B)	31,499(C)	34,584 6300	29,498	29,498
Payload	1b.	4100	4100		3900	6105
11216 2000000	sq. ft.	93.7	100.1	106.9	96.4	99.2
Stall speed - power-off (D)	kn.	124	129	133	126	128
Take-off run at S.L calm (D)	ft.	3680	4440	5300	3990	4320
Take-off run at S.L. 25 kn. win	d(D) ft.	2360	3000	3720	· 2630	2900
Take-off to clear 50 ft calm		5190	6130	7180	5580	6000
Max. speed/altitude	kn./ft.	556/2500	556/2500	556/2500	556/2500	556/2500
Rate of climb at S.L.	fpm	5440.	5020	4650	5260	5080
Time: S.L. to 20,000 ft.	min.	4.7	5.1	5.6	4.9	5.1
Time: S.L. to 30,000 ft.	min.	8.4	9.3	10.3	8.8	9.2
Service ceiling (100 fpm)	ft.	40,400	39,100	37, 700	39,900	39,300
Combat range	n.mi.	2270	2600	2740	2460	2380
Average cruising speed	km./M	436/.76	436/.76	436/.76	436/.76	436/.76
Cruising altitude(s)	ft.	36,000 - 43,400		33,200 - 42,000	35,400 - 43,400	34,800 - 42,500
	./n.mi.	1150/5.3 (E)	1310/6.0 (E)	1410/6.5	1120/5.3 (F)	1230/5.6
Average cruising speed	kn.	436/.76	436/.76	436/.76	436/.76	436/.76
	mi./hr. ./n.mi.	1660/7.9 (E) (G) 11,950/570	1870/8.8 (E) (H) 13,910/690	1900/9.0 (H) 12,900/750	1640/7.9 (H) 11,970/780	1740/8.2 (H) 12,270/770
COMBAT LOADING CONDITION	7	(2) 60% Fuel Stores Retained	(4) 60% Fuel Stores Retained	(6) 60% Fuel Stores Retained	(8) 60% Fuel Stores Retained	(10) 60% Fuel Stores Retaine
COMBAT WEIGHT	1b.	62, 089	65, 400	69, 425	63, 282	65, 493
Engine power		MILITARY	MILITARY	MILITARY	MILITARY	MILITARY
Fuel	lb.	16,367	18,899	20,750	17,699	17,699
	kn./ft.	470/42,600	469/41,600	467/40,500	556/S.L.	470/41,500
	fpm/ft.	400/48,600	395/41,600	380/40,500	6350/S.L.	400/41,500
Combat ceiling (500 fpm)	ft.	42,300	41,200	39,900	41,900	41,200
Rate of climb at S.L.	fpm	6510	6120	5720	6350	6120
Max. speed at S.L.	kn./M	556/.84	556/.84	556/.84	556/.84	556/.84
	kn./M	508/.88	506/.88	503/.87	507/-88	506/.88
LANDING WEIGHT	1b.	44,502	45,513	45,651	44,680	44,686
Fuel	lb.	2880	3112	3276	2997	2997
Stall speed - power-off/appr.pw	r. kn/kn	97/95	98/96	98/96	97/95	97/95
	2: 75	/	/	1000/0000	15.00 /5.000	1500/5000

(A) The limit catapult take-off weight of 73,000 pounds and the limit field take-off weight of 78,000 pounds are consistent with current operating bulletins. Under emergency conditions increased take-off weights may be utilized.

(B) Fuel limited to maintain T.O. weight. With full fuel of 29,498 lbs (T.O. Wt. = 75,220 lb), combat radius is 1240 n. mi. without refueling.

Land. Dist. GR. Roll/Over 50 ft. (I) ft./ft. 4550/5260

- mi. without refueling.

 (C) Fuel limited to maintain T.O. weight. With full fuel of 34,584 lbs (T.O.Wt=81,085 lbs), combat radius is 1430 n.mi.

 (D) Full flaps
- (E) For Low Alt. Attack mission, combat radius is decreased 125 n.mi. and mission time is decreased 0.5 hr.

4640/5350

(F) For Sea Level target approach, total radius is reduced:
65 n.m. for ea. 100 mi. cruise in at S.L. at maximum range
130 n.mi. for ea. 100 mi. run in at S.L. with military thrust
(G) One refueling from A3D-2 tanker. (Tanker T.O. Wt.=73,000 lbs)
(H) One refueling from A3D-2 tanker. (Tanker T.O. Wt.=78,000 lbs)
(I) Without chute. With chute land. dist. is decreased appr. 1850ft.
(J) All loadings include IFR probe.

4560/5270

4560/5270

4650/5360

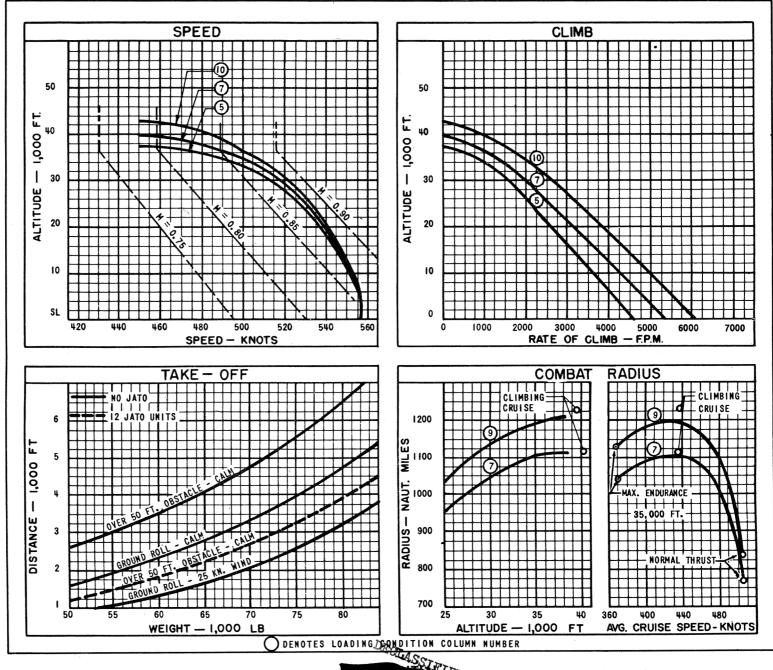
- (K) Perf. Basis: NATC & Contr. flt. test of prototype camb. wing A3D-2. Range & radii based on flt.test fuel consumption. SPOTTING: 27 A/C can be accommodated in landing spot on flt.
- and hanger decks of CVA-19 class angled deck carrier.



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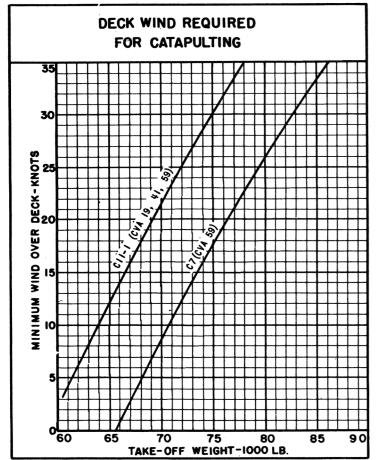
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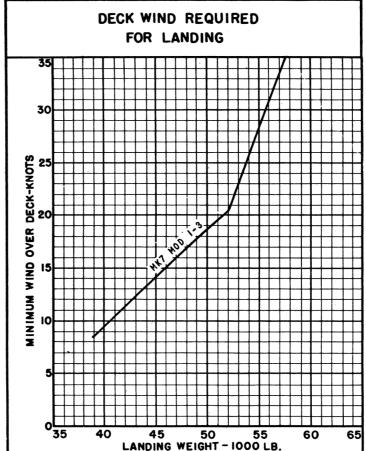






CARRIER SUITABILITY





Catapult take-off is based on NATC recommended minimum end airspeed. $\,$

Catapult end speed is limited by catapult capacity.

Approach speed is based on NATC recommended minimums.

Engaging speed limited by airplane strength limit as determined by maximum rate of sink.







NOTES

HIGH ALTITUDE ATTACK COMBAT RADIUS MISSION

START ENGINES, T.O. AND ACCELERATE: Fuel for 5 minutes sea level, normal static thrust

CLIMB OUT: At maximum rate of climb with military thrust, on course to optimum cruising altitude.

CRUISE-OUT: At speed for maximum range at optimum cruising altitude.

CLIMB: At maximum rate of climb with military thrust, on course to combat altitude. (Cruising ceiling, 300 fpm rate of climb with normal thrust).

BOMB RUN: Cruise in level flight to target for 15 minutes at maximum speed with normal thrust at combat altitude.

EVASIVE ACTION: Drop bombs. Evasive action for 2 minutes at maximum speed with normal thrust at combat altitude. (no distance credit).

ESCAPE: For 8 minutes at maximum speed with normal thrust at combat altitude. (Descent to optimum cruising altitude is accomplished in evasive action and escape periods).

CRUISE-BACK: At speed for maximum range at optimum cruising altitude.

RESERVE AND LANDING: 5% initial fuel load plus fuel for 20 minutes at sea level at speed for maximum endurance.

LOW ALTITUDE ATTACK COMBAT RADIUS MISSION

START ENGINES, T.O. AND ACCELERATE: Fuel for 5 minutes sea level, normal static thrust.

CLIMB-OUT: At maximum rate of climb with military thrust, on course to optimum cruise altitude.

CRUISE-OUT: At speed for maximum range at optimum cruising altitude.

DESCEND: To sea level (no fuel consumed-no distance covered) Drop bombs.

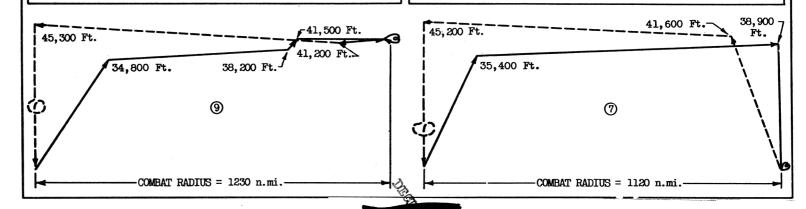
COMBAT: At sea level for 5 minutes with military thrust. No distance made good.

CLIMB-BACK: At maximum rate of climb with military thrust, on course to optimum cruise altitude.

CRUISE-BACK: At speed for maximum range at optimum cruising altitude.

DESCEND: To sea level (no fuel consumed, no distanced covered).

RESERVE AND LANDING: 5% initial fuel load plus fuel for 20 minutes at sea level at speed for maximum endurance.



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