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NAVAIR 00-110AH1-5

PRE SERVICE

POWER PLANT	POWER PLANT MISSION AND DESCRIPTION				
No. & Model (1) T400-WV-402 Manufacturer . United Aircraft of Canada Pratt and Whitney) Engine Spec. No. 743A Type Twin Section Free Power Turbine with Reduction Gearbox Gear Reduction Ratios Main Rotor 21.288:1 Tail Rotor 4.520.1	The primary mission of this aircraft is that of an armed tactical helicopter capable of delivering weapons fire, low altitude high speed flights, search and target acquisition, re- connaissance by five, multiple weapons fire support, and troop helicopter support. The aircraft is capable of performing this mission from prepared or unpreprepared areas, day or night, and operation from ships at sea.	<u>Loading</u> LB. LF Empty 8030 - Basic 8076 - Design 10,000 3.5			
RATINGS	RATINGS The gas turbine powered twin sea cobra is of combat design,				
SHP RPM ALT Intermediate 1970 6600 0 Maximum Continuous 1673 6600 0 Single Power Section	featuring tandem seating to give both prot and gunner excellent visibility. Both crew stations have flight control and fire control systems permitting flexibility in division of functions under normal and emergency situations. The twin engine installation gives increased safety and re-	Maximum Landing 14,000 2.5			
Intermediate 970 6600 0 Maximum Continuous 824 6600 0 Transmission Limits	liability. The 2050 shy transmission can transmit the full rated engine power under all conditions except for density altitudes	FUEL AND OIL			
Takeoff (5 min) 2050 SHP Maximum Continuous 1850 SHP	well below sea level. A mission designed fuselage combined with the improved Model 214 rotor system gives increased load	FUEL			
ELECTRONICS	capability and speed. Four wing stores stations and integral chin turret provide good armament versatility with the capability of quickly mounting a wide variety of weapons to match the desired	Location No. Tanks Gals.			
UHF Command Set AN/ARC-159(V)1	mission. Most components have been proven in the AH-1J and Model 214, assuring reliability and maintainability.	Fuselage 2 306			
FM Tactical Set AN/ARC-114A Intercom AN/AIC-18 Tacan Nav Sys AN/ARN-84(V) UHF Direction Finder AN/ARA-50 Grosyn Compass AN/ASN-75B Direction Finder Set AN/ARN-83 Radar Altimeter AN/ARN-72 Radar Beacon AN/APN-171(V)		Grade JP-4 JP-5 JP-5 Specification MIL-T-5624 OIL Engine 3 5.0 Specification MIL-L-23699A			
Two Radio Set Controls C-8057/ARC Transponder Test Set TS-1843A/APX		ORDNANCE			
PROVISIONS FOR Two Communications Security Units TSEC/KY-28 Computer, Mark XII KIT-1A/TSEC	DEVELOPMENT Contract Placement June 23, 1975 First Flight May 20, 1976	20 MM CHIN TURRET WING STORES PYLONS (FOUR) any combination of the following: LAU 61/A, 9/A, 68/A, 68A/A, and 68B/A rocket pods on ita. 1-4. EU-55B fuel-air explosive on sta. 1 & 4 or i & 3. U-44 Flare Dispenser on sta. 1-4.			
· · · ·	DIMENSIONS	HTW, MK-155 - Mod O on sta. 1 & 4 using modified multiple sta. Bomb Rock (MSBR).			
	Rotor Diameter48.0 Ft.Length Rotors Operating58.0 Ft.Rotors Static58.0 Ft.Fuselage45.3 Ft.Span (max. lateral)10.6 Ft.	M-118 Grenade Dispenser on sta. 1 & 4. MK-45 Parachute Flare on MSBR. Maximum ammunition capacity for the 20 MM chin turret is 750 rounds. ACCOMODATIONS Basic, Medium or Heavy Combat			
APPII 1079	Height 14.3 Ft. Tread 7.0 Ft. Rotor ground clearance 9.0 Ft. (static against stops)	Pilot 1 Gunner 1 Clean Mission Pilot 1 Copilot 1 Copilot 1			

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	PER	FORMANCE S				
TAKE-OFF LOADING CONDITION	① CLEAN HIGH ALTITUDE	<pre>② ATTACK 2 LAU 68 PODS 750 RDS 20 MM</pre>	(3) ATTACK 2 LAU 68 PODS 2 MINIGUN PODS 750 RDS 20 MM	(4) CLOSE SUPPORT 2 LAU 68 PODS 2 MINIGUN PODS 750 RDS 20 MM	ATTACK 4 LAU 61A PODS 750 RDS 20 MM	6 FERRY CLEAN
TAKE-OFF WEIGHT	b. 10,705(A)	11,746	12,401	12,401	13,928	10,705(A)
Fuel internal/external (JP-5) Ib./I		2,081/0	2,081/0	2,081/0	2,081/0	2,081/0
Payload	b. 0	909	1,074	1,074	2,707	0
Disc loading Ib./sq.	ft. 5.9	6.5	6.9	6.9	7.7	5.9
Vertical rate of climb at S.L. (B) fp	m. 2,258	1,770	1,390	1,390	301	2,258
Absolute hovering ceiling (OGE) (B)	ft. 10,000 (B)	7,150	5,350	5,350	1,200	10,000 (B)
Max. rate of climb at S.L. (B) fp	m. 2,880	2,460	2,190	2,190	1,785	2,880
Service ceiling (C)	ft. 10,000 (C,D)	10,000 (C,D)	10,000 (C,D)	10,000 (C,D)	7,400 (D)	10,000 (C,D)
	i n . 165	161	157	157	149	165
Max speed/altitude (B) kn./	ft. 165/4,000	161/0	157/0	157/0	· 149/0	166/4,000
O.E.I. Service ceiling (B)	ft. 9,000	6,400	3,000	3,000	-100	9,000
Min. speed (O.E.I.) (B)	k n. 42	45	54	54	65	42
Max. speed (O.E.I.) (B)	k n. 112	109	105	105	95	112
Combat radius n. r	ni. 165	116	113	74	108	314
Mission time h	rs . 2.81	1.63	1.63	2.31	1.63	2.60
Average cruising speed	(n . 121	150	146	121	138	121
Cruising altitude	ft. 10,000	S.L.	S.L.	S.L.	S.L.	10,000
Range n. r	ni. 342	245	234	240	227	342
Average cruising speed	(n. 121	150	145	121	138	121
Cruising altitude	ft. 10,000	S.L.	S.L.	S.L.	S.L.	10,000
Maximum endurance h	rs. 3.42	2.73	2.62	2.62	2.41	3.42
Endurance speed	(n. 79	75	76	76	78	79
Endurance altitude	ft. 10,000	S.L.	S.L.	S.L.	S.L.	10,000
 (A) Includes ammunition boxes for which are required for proper (B) At Intermediate Power (C) Limited by oxygen requirement; (D) At maximum continuous power Performance Basis (1) All performance at standard da (2) Flight test data from modified (3) Engine specification fuel cons (4) 20 MM nose turret on all confi 	CG. at maximum continuous " ^{t.} y conditions Model 309, April 1974 umption increased 5%					
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