NAVAIR 00-110AF111-1

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Standard Aircraft Characteristics

F-111B AIRCRAFT

(TITLE UNCLASSIFIED)

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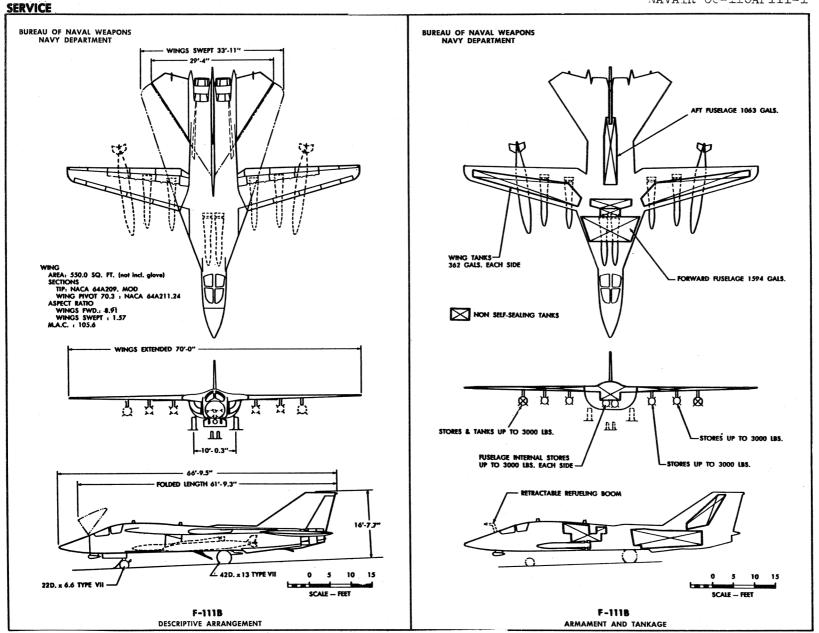
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STANDARD AIRCRAFT CHARACTERISTICS

F-IIIB

GENERAL DYNAMICS



DOWED DIANIT	AUCCIONI AND DECODIDEIONI	SERVICE		
POWER PLANT	MISSION AND DESCRIPTION	WEIGHTS		
No. and Model Manufacturer Specification Type Type Augmentation Length with A/B (Operating Temp) Diameter (Operating Temp) Dry Weight RATINGS Static Thrust at Sea Level Lb. RPM Manufacturer Fratt & Whitney P&W A-1795 12/5/64 Turbo Fan Modulated A/B 251,20 in 3880 lb RATINGS	The F-111B is a two-place, twin engine fighter whose primary mission is the destruction of enemy aircraft. The aircraft has the additional capacility of carrying conventional and special weapons for attack missions. The basic armament consists of 4 air-to-air PHDENIX missiles pylon mounted under the wings and 2 PHDENIX missiles in the weapons bay. Two 450 gal. external fuel tanks may be carried on the outboard wing stations. The F-111B features a variable sweep wing, 4 pivoting wing pylons which may be utilized at all sweep positions and 2 fixed pylons useable for the swept forward configuration. The high lift system is composed of full span double slotted flaps and slats with a rotating section in the leading edge of the wing glove. Lateral control is accomplished by spoilers and	LOADING WEIGHT(POUNDS) LOAD FACTOR		
Maximum (A/B) 18,500 14,200 Military 10,750 14,200 Normal 8,500 13,450	differential horizontal tail deflection. The aircraft uses an escape capsule containing the entire	FUEL AND OIL		
ORDNANCE	cockpit rather than ejection seats. F-111B features antiskid brakes. The F-111B is basically similar to the Air Force F-111A being	No. of Tanks Gallons Location 2 363 each Wing 1 1594 Forward Fuselage 1 1063 Aft Fuselage		
Missiles: Phoenix Sidewinder IC-IR Shrike	concurrently developed. Many components are common to both designs.	1 1063 Aft Fuselage 2 450 each Fixed Pylon Ext. Drop Grade JP-4 or JP-5		
Rocket Launchers: LAU-3A/A LAU-10/A LAU-32A/A Gun Pod: MK-4		Specification MIL-F-5624		
Nuclear Weapons: MK-43 MK-57 TX-61		2 4 each (3 useable) Specification MIL-L-7808 & MIL-L-23699		
Conventional Bombs: (For Multiple Carriage on MER/TER 7 Racks) MK-81, MK-82, MK-83 MK-84 (Single Carriage Only) MK-81 Snakeye I, MK-82 Snakeye I 220# AN/FRAG (Banded) 260# AN/FRAG (Banded) 250# ANGP (Banded) MK-77 MOD 2 Firebomb MK-79 Firebomb MK-24 Flare	DEVELOPMENT First Flight F-111A	### Company of the co		
CBU-1, CBU-2, CBU-3 Sadeye, Rockeye II, Fireye	DIMENSIONS			
Briteye, Bigeye, Weteye Fuel Tanks: Miscellaneous: MX-900 Chaff Dispenser MX-12 Smoke Tank ALQ-76- ZCM Pod AUX Power Unit (RCPP-10) AQM-37A Target Supersonic Aerial Tow Target	Wing Area 550 Sq. Ft. Most Forward Sweep (ldg. edge) 16 deg. Span 70 ft. M.A.C. 8 Ft. 10 in. Incidence 1 deg. Dihedral 1 deg. Airfoil Section Sta 134 NACA 64A211.24 Most Aft Sweep (ldg. edge) 72.5 deg. Span 33 Ft. 11 in. Length 66 Ft 9.5 in. Height 16 Ft 7.7 in. Wheelbase 24 Ft 3 in. Iread 10 Ft 0.4 in.	PHOENIX MISSILE SYSTEM AN/AWG-SHRIKE COMPUTER CP-74.1/LELECTRIC BOME FUZING AWN-ARMAMENT MONITOR AND CONTROL (AMAC) A/A24B-A		

PERFORMANCE SUMMARY						
TAKE-OFF LOADING CONDITION	(1) FIGHTER 6 Phoenix	(3) FIGHTER 2 Phoenix	(5) FERRY 2 450 Gal. Tanks			
TAKE-OFF WEIGHT Ib	77566	72421	77302			
Fuel internal/external (JP-5) lb./lb		23003/	23003/6210			
Payload Ib	6620	2207				
Wing loading lb./sq. ft	141.0	131.8	140.5			
Stall speed—power-off kn		110.3	114.0			
Take-off to clear 50 ft at S.L calm (A)	6000	5100	5950			
Take-off to clear 50 ft at S.I calm (B)		2700	3050			
Wind Cver Deck Required for Launch (C) km		+2.5	+10.0			
Max. speed/altitude (A) kn./fl		622/SL	(D)480/12000			
Rate of climb at S.L. (A)/(B) fpm		5505/23200	4720/			
Time: S.L. to 20,000 ft. (A)/(B) min	7.1/1.2	5.1/0.9	6.3/			
Time: S.L. to 30,000 ft. (E) (A)/(B) min	13.9/2.2	11.0/1.7	13.8/			
Service ceiling (100 fpm) (A) ft	32100	37600	34200			
Combat range n.mi	1830	2330	(F) 2760			
Average cruising speed kn		428	423			
Cruising altitude(s)	27000 to 39100	33200 to 40300	29300 to 38600			
Combat radius/mission time n.mi./hi	475/2,40	845/4.02				
Average cruising speed kn		428				
CAP loiter time/mission time (G) hr	(H)1.52/2.37	(I)3.25/4.02				
Average loiter speed (J) kr	365	367		·		
COMBAT LOADING CONDITION	(2) 6 Phoenix	(4) 2 Phoenix	(6) 2 450 Gal. Tanks			
COMBAT WEIGHT Ib	68365	63220	65617			
Engine power / Wing sweep	Maximum/Cptimum	Maximum/Optimum	Military/26°			
Fuel lb		13800	17528			
Combat speed/combat altitude kn./fl		(K) 1150/30000	(D) 448/30000			
Rate of climb at combat altitude (L)/M) fpm	7485/6212	9065/25100	1035/			
Combat ceiling (500 fpm) (L)/(M) ft	-/ 44900	-/ 52600	36100/			
Rate of climb at S.L. fpm	21300	27200	5740			
Max. speed at S.L. $(A)/(B)$ kn	575/678	(K) 625/793	(D) 400/400			
Max. speed/altitude kn./fi		(K) 1260/40000	(D) 480/12000			
Acceleration time, Vcruise to Vcombat mir	10.2	4.0				
LANDING WEIGHT (External Stores retained) Ib	56980	51613	50637			
Fuel , lb		2195	2548			
Stall speed—power-off/approach power kn./km	(11) 10000)	(N) 95.5/94.8	(P)96.6/95.8			
Landing distance-ground roll/over 50 ft. obst. ft./ft	1		2125/2840	the state of the s		

NOTES

DATA BASIS: Calculations Based on BUWEPS Estimated Performance As of 1 March 1965

Fuel Consumption Based on Engine Manufacturer's Specification Fuel Flows Increased 5%

Weight and Aerodynamic Data Based on Projected Fleet Configuration

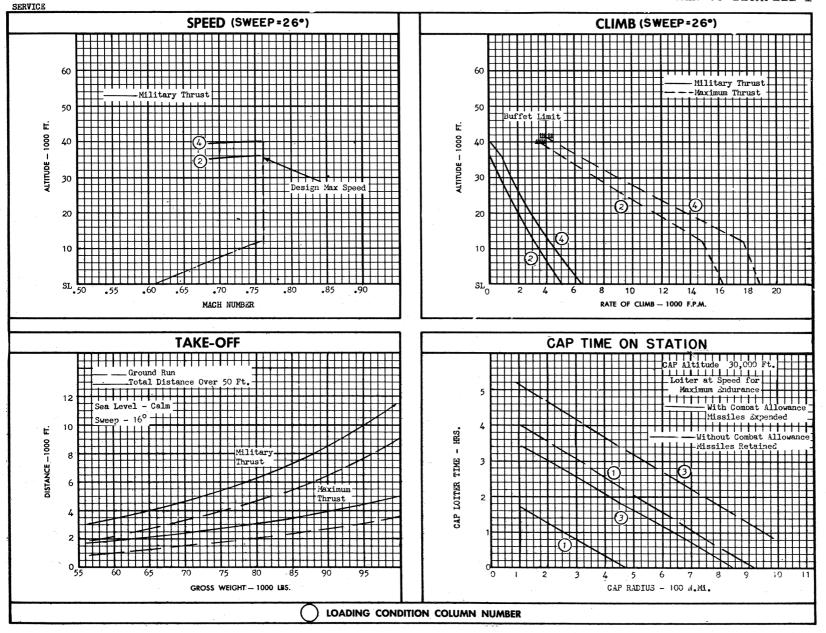


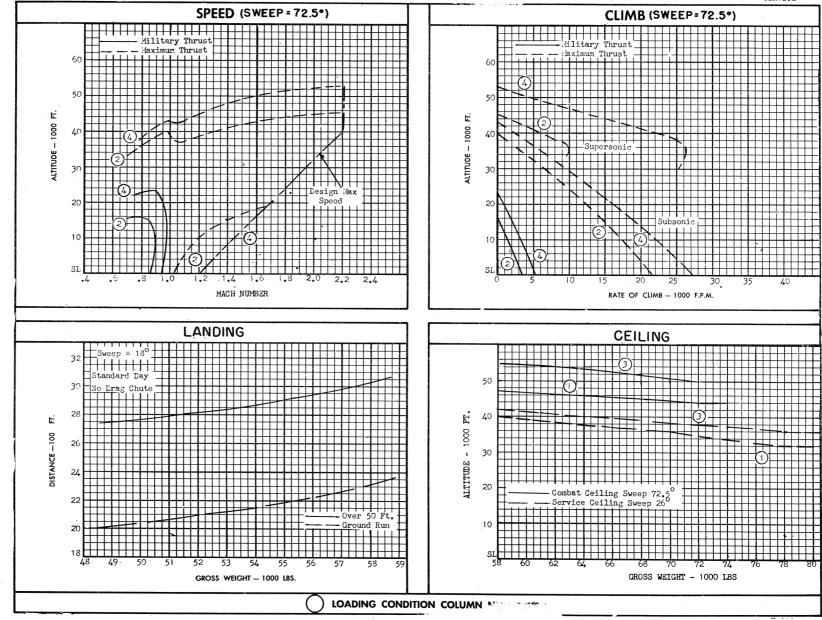
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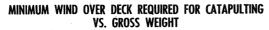
Footnotes For Performance Summary

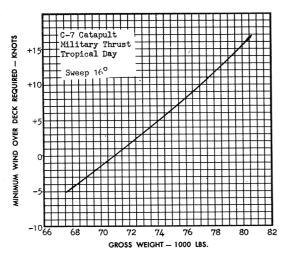
- (A) Military Rated Thrust
- (B) Maximum Rated Thrust
- (C) Minimum Catapult Take-Off Speed, C-7 Catapult, 90°F Ambient Temperature
- (D) Design Maximum Level Flight Speed for 26° Sweep per Contractor's FZM-12-929 Revised 6 Dec 1964 Not Thrust Limit*
- (E) 30,000 ft. or Cruise Ceiling, Whichever is Lower for Military Thrust Climbs
- (F) Tanks Retained, Range is 2997 N.Mi. if Tanks Dropped When Ampty
- (G) Combat Air Patrol Radius 150 N.Mi. CAP Altitude 30,000 ft.
- (H) 3.79 hrs. If No Combat Fuel Allowance is Made Mission Time 4.64 hrs.
- (I) 4.90 hrs. If No Combat Fuel Allowance is Made Mission Time 5.66 hrs.
- (J) Speed for Maximum Endurance, Maneuverability Limited to 1.27g by Buffet. Loiter Time Significantly Reduced at Higher Speed
- (K) Design Maximum Level Flight Speed for 72.5° Sweep per Contractor's FZM-12-929 Revised 6 Dec 1964; Not Thrust Limit
- (L) Subsonic Climb Speed
- (M) Supersonic Climb Speed (Design Maximum Speed for 72.5° Sweep)
- (N) 18° Sweep Required for Stability (21° if Missiles Not Retained) (5% Static Margin)
- (P) 21° Sweep Required for Stability (5% Static Margin)

^{*} Wing sweep limited to 26° when fixed pylon stations are loaded. External fuel tanks may be carried at fixed pylon stations only.

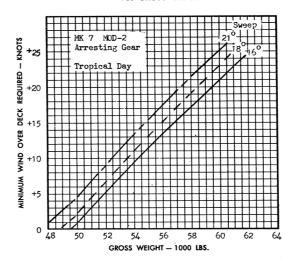








MINIMUM WIND OVER DECK REQUIRED FOR ARRESTING VS. GROSS WEIGHT



NOTES

These Curves Should be Used for Planing Purposes Only. Actual Catapult and Arresting Gear Operation Should be in Accordance with Applicable Aircraft Technical Orders, and Catapult and Arresting Gear Bulletins.

NOTES

GENERAL PURPOSE FIGHTER - MISSION DEFINITION

- 1) Warm-up, Take-off, Accelerate: 5 min. with normal thrust at sea level
- Climb: Cn course to cruise altitude with military thrust
- 3) Cruise-Cut: At altitude and speeds for maximum
- 4) Combat Fuel Allowance: Accelerate with maximum thrust at 30,000 ft. from cruise speed to 1.5 in and remain at this speed and altitude for 2 Min. at maximum power EXPEND MISSILES
- 5) Cruise-Back: At altitudes and speeds for maximum
- Reserve: 20 Min. at speed for maximum endurance at sea level (2 engines operating) plus 5% of initial fuel load.

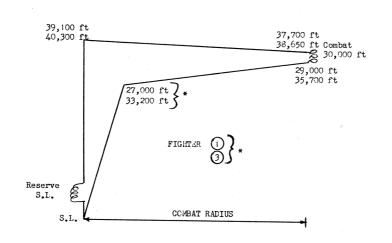
COMBAT AIR PATROL - MISSION DEFINITION

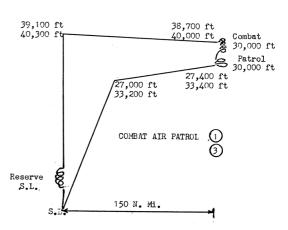
- 1) Warm-Up, Take-Cff, Accelerate: Same as G.P. Fighter 2) Climb: Same as G. P. Fighter
- 3) Cruise-Out: To a point 150 Naut. miles from base at altitudes and speeds for best range
- 4) Loiter: On station at 30,000 ft at speed for maximum endurance
- Combat Fuel Allowance: Same as G. P. Fighter
- 6) Cruise-Back: 150 Naut. miles to base at altitudes and speeds for best range
- 7) Reserve: Same as G.P. Fighter

COMBAT AIR PATROL - DESIGN MISSION

Same as above except:

5) Combat Fuel Allowance: None, Retain Missiles





*Flight altitudes correspond to the take-off loading conditions show on the Performance Summary page of this chart

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