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NAVAIR 00-110AF3-1

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Standard Aircraft Characteristics

NAVY MODEL
F-3B
AIRCRAFT

(F3H-2)

(TITLE UNCLASSIFIED)

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1 MAY 1955 IN PART AND ALL ADDENDA THERETO

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PUBLISHED BY DIRECTION OF THE COMMANDER OF THE NAVAL AIR SYSTEMS COMMAND

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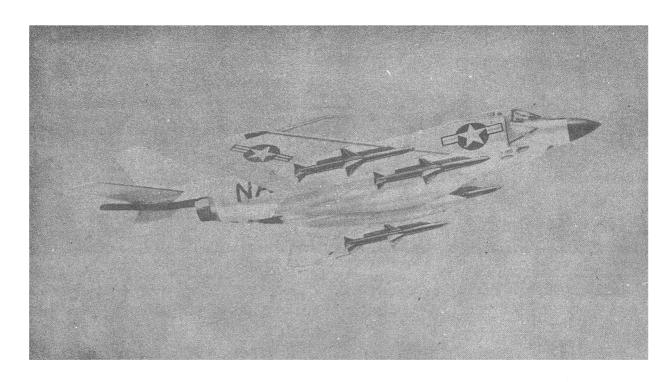
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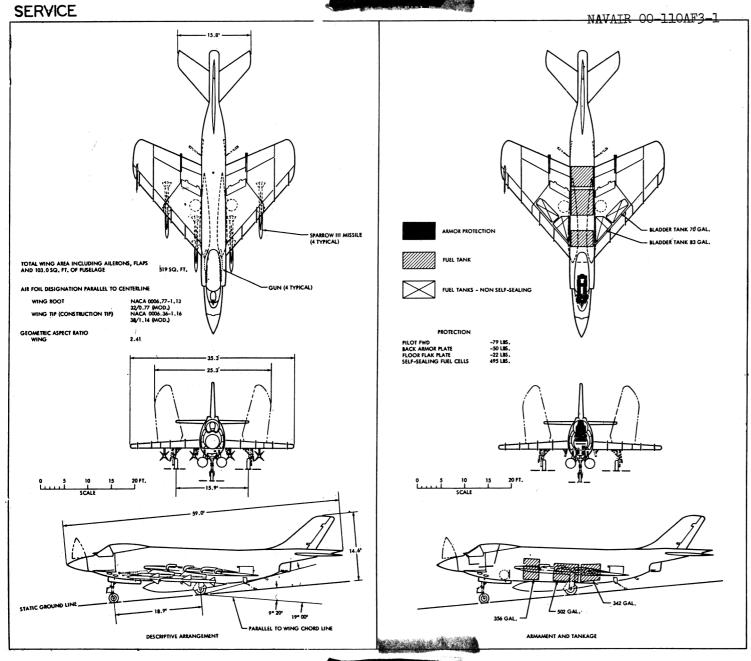


STANDARD AIRCRAFT CHARACTERISTICS

F-3B (F3H-2) "DEMON"
McDONNELL



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NAVATE OO-110AF3-1 POWER PLANT MISSION AND DESCRIPTION No. & Model..... (1) J71-A-2B The McDonnell F-3B (F3H-2) airplane is a Mfg..... Allison single place, swept-wing, jet-propelled Spec..... Allison No. 361 - F fighter and is designed for either land or Type......Axial Flow-Turbo Jet

RATINGS

Augmentation Afterburner

Length (Incl.A.B)......287.1"

Width 42.6"

Static Thurst at Sea Level

	Lbs.	RPM
Maximum (A/B)	14400	6175
Military	10000	6175
Normal	8700	6000(Est)
90% Normal	7920	5920
75% Normal	6600	.5820
Idle	714(Max	:)3950(Min)

ORDNANCE

4 - 20 mm. MK 12 aircraft guns 600 Rds. MK 12 20 mm. Ammo. (Normal Load)

760 Rds. MK 12 20 mm. Ammo. (for special loadings)

- 1 AFCS MK 21 Mod 0
- 1 AN-N-6A Gun Camera

External

- 4 Sparrow III Missiles
- 2 Douglas Ejector Racks (Fus.) (Aero 7A)
- 4 Sidewinder Missiles (Alternate Load)

carrier based operations.

The F-3B is equipped with the Sparrow III weapon system which consists of privisions for carrying four supersonic air-to-air guided missiles and special electronic gui- No dance equipment. Further armament consists of four forward firing 20 mm guns. Provisions are included for two heavy duty pylons on the underside of the fuselage for carrying various external stores.

The F-3B is privided with tricycle landing gear, catapult and arresting gear, and folding wing panels. Equipment includes an automatic pilot, ejection seat, and pressurized cockpit. An auxiliary power unit may be carried externally to provide for engine starting when operating from advance bases.

Power actuated leading edge slats and trailing edge plain flaps increase lift for landing and take-off, and fuselage mounted speed brakes provide drag control in flight, The primary control system incorporates power actuation with artificial feel forces. Provisions are included for obtaining moderate control forces during emergency operations. DEVELOPMENT

First Flight.....July 1957 Service Use September 1957

DIMENSIONS

Wing	
Area519	sq. ft.
Span	. 351 411
MAC	. 151 6"
Sweepback (\frac{1}{4}chord) Length (overall) 58	/30_121
Length (overall) 58	1711.5"
Height 1 Tread 15	4' 16.6"
Tread	10.4"

WE I WII	10	
Loading	Lbs.	L.F.
Empty	21287	
Basic	22293	
Empty Basic Design	26000	7.5
Combat Max.T.O. Max.—CarrierIand	31145	5.8
Max.T.O	39000	
MaxCarrierLand	26700	

WEIGHTS

SERVICE

MIL-F-5572

	FUEL F	IND OIL
. Tanks	Gal.	Location
1	356	Fuselage, Fwd.
1	502	Fuselage, Cen.
1	342	Fuselage, Aft.
2	166	Wing, Inboard
2	140	Wing, Outboard
2	282	External Tanks
Fuel Gr	adeCh	MIL-F-5624D or eapest MIL-F-5572
Fuel Sp		MIL-F-5624D or

Fuselage Tanks Self - Sealing Wing Tanks Bladder Type

OIL

Integral with Engine Spec..... MIL-L-7808

ELECTRONICS

UHF.	AN/ARC-27A
UHF DF AN/ARR-40 8	AN/ARA-25
Short Range Nav	AN/ARN-21
Radio Altimeter	. AN/APN-22
IFF Identifier	. AN/APX-6B
Radar	AN/APG-51B
Missile Guidance Set.	AN/APA-127
Missile Launching Set.	AN/ASA-23
Video Coder	AN/APA-89





NAVAIR OO 110AF3 1

PERFORMANCE SUMMARY					
TAKE-OFF LOADING CONDITION	**(1) (4)SPARROW III	(3)(4) SPARROWS!!! +(2)282 GAL. TKS.	(5) (2) SPARROW III	(7)(2)SPARROW !!! (2)282 GAL. TKS.	(9) (2) SIDEWINDERS
TAKE-OFF WEIGHT L	35239	38997	34203	38600	33424
FUEL INTERNAL/EXTERNAL (JP-5) LB./L	10241/	10241/3196	10241/	10241/3835	10241/
PAYLOAD AMMO./MISSILES LB./ L	432/1520	432/1520	432/760	432/760	432/310
WING LOADING LB:/SQ.L	67.9	75.1	65.9	74.4	64.4
STALL SPEED - POWER-OFF K	118	124	116	123	114
TAKE-OFF RUN AT S.L CALM (A)/(B) F	7770/~~~	9860/3030	4830/2265	9280/2970	4390/2150
TAKE-OFF RUN AT S.L.25kN. WIND(A)(B) F		7320/2070	3340/1500	6860/2030	3000/1415
TAKE-OFF TO CLEAR 50 FT CALM(A)(B) F	1 1400/2000	11800/4645	6640/3600	11200/4550	6160/3445
MAX. SPEED/ALTITUDE (A) M/F	.85/25000	.77 /20000	.88/25000	.80/25000	.89/25000
RATE OF CLIMB AT S.L. (A) FP	4330	3380	4840	3730	5140
TIME: S.L. TO20000 FT. (A) MI	7.19	10.82	6.28	7.33	5.89
TIME: S.L. TO 30000 FT. (A) MI	16.50	36,00	13.22	25.40	11.87
SERVICE CEILING (100 FPM.) F	32000	27150	34000	28950	35050
COMBAT RANGE N.M	- CAPA	1076	958	1277	1026
AVERAGE CRUISING SPEED]		.71	.76	7/	78
CRUISING ALTITUDE(S) FT/F	1 4/400/2/000	23200/34000	32000/37500	25400/36400	33400/38200
COMBAT RADIUS/MISSION TIME N.MI./H	~/4/~•00	420/2.69	336/2.21	501/3.01	363/2.24
AVERAGE CRUISING SPEED Mr		.73	.77	.75	.82
CAP LOITER TIME/DECK CYCLE TI.(C) HR/		1.34/2.76	.89/2.26	1.74/3.14	.97/2.33
IFR RADIUS/DECK CYCLE TIME N.MI/	IR 562/3.36	654/3.90	613/3.52	730/4.15	647/3.64
COMBAT LOADING CONDITION	(2) SPARROWS RETAINED	(4) TANKS OFF SPARROWS OFF	(6) SPARROWS RETAINED	(8) TANKS OFF SPARROWS OFF	(10) SIDEWINDERS RETAINED
COMBAT WEIGHT	31145	31839	29349	31944	29020
ENGINE POWER	Maximum	Maximum	Military	Military	Maximum
FUEL	0141	8065	6147	8446	6147
COMBAT SPEED/COMBAT ALTITUDE M/F	· 94/31000	.94/28500	.88/33700	.86/31000	96/35000
RATE OF CLIMB/COMBAT ALTITUDE FPM./F	4720/31000	5310/28500	890/33700	1010/31000	4740/35000
	43150	42880	35800	33950	45700
RATE OF CLIBEL AT S.L. FP	11110	12410	5860	5480	14350
MAX. SPEED AT S.L. Mn		.91	.83	.82	.94
MAX. SPEED/ALTITUDE MI/F	•94/35000	.94/35000	.89/30000	.87/30000	.97/35000
1. SDING WEIGHT	25425	25884	25144	25637	2/811
FUEL L	1947	2110	1942	2139	1938
STALL SPEED - POWER-OFF/APPR.POWER KN./K		99/94	97/93	98/94	97/93
DISTGRD ROLL/OVER 50 FT. OBS. FT./F		4775/6885	4710/6780	4755/6845	4685/6735
WATER DEDECOMANCE DAGTO CONT.	L J MARG TO				

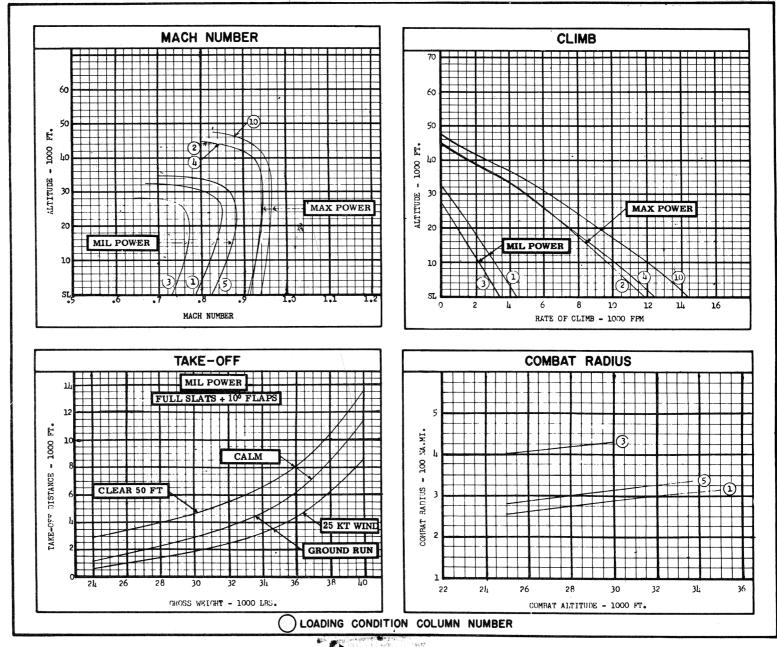
NOTES: PERFORMANCE BASIS: Contractor and NATC Flight Test RANGE AND/OR RADIUS are based on flight test fuel

consumption data.
*Nos. 1, 3, 5, 7, & 9 are General Purpose Fighters

(A) MILITARY RATED THRUST
(B) MAXIMUM RATED THRUST
(C) Combat Air Patrol-150 naut.mi. radius

All loadings include guns.





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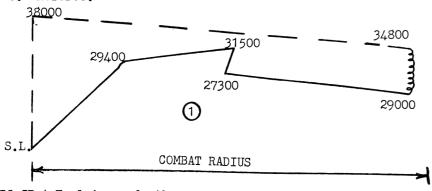
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NOTES

GENERAL PURPOSE FIGHTER WITH IN-FLIGHT REFUELING

- 1. Warm-up: Fuel for 5 min. with normal thrust.
- 2. Climb: With mil. thrust to opt. cruise alt.
- 3. Cruise-out: At alt. and speed for max. range.
- 4. Refueling allowance for hook-up and flight contingencies:
 Fuel for 5 min. at speed and altitude for max. endurance.
 No fuel used or distance gained during transfer.
- 5. Cruise-out: At alt. and speed for max. range. The remainder of the problem is the same as General Purpose fighter of condition 1.
- 6. Combat.
- 7. Cruise back.
- 8. Reserve.



following are applicable:	△WEIGHT	△ RANGE	▲ RADIUS	▲ DECK CYCLE TIME
G.P. Fighter-guns/4 Sparrows	-452	- 48	- 26	12
G.P. Fighter-guns/4 Sparrows-II	FR – 452	- 67	- 36	- .16
G.P. Fighter-guns/4 Sparrows plus 2-282 gal. tanks*	0	2	2	00
G.P. Fighter-guns/4 Sparrows	U	-3	- 2	02
plus 2-282 tksIFR*	0	- 52	-28	18
prab coc ons-iii	O	-)~	-20	10

*Airplane defueled to max. T. O. gross weight of 39,000 lbs.

O LOADING CONDITION COLUMN NUMBER

